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RADC GUIDE TO ENVIRONMENTAL STRESS SCREENING

Eugene Fiorentino

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| precipitate latent defects to | early failure. | Methods and | procedures | are provid | ed for |
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| report enables the user to play | n, monitor and (| control the s | creening pr | ocess so a | s to |
| achieve quantitative reliabili | ty objectives in | n a cost-effe | ctive manne | er | |
| The report has been prepared in | n MIL-HDBK forma | at so as to s | erve as a b | asis for a | OOD |
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Executive Summary

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1. The objective of this Guidebook is to provide techniques for planning and evaluating Environmental Stress Screening (ESS) programs. The guidance contained herein departs from other approaches to ESS in that quantitative methods are used to plan and control both the cost and effectiveness of ESS programs. Guidebook procedures and methodology were developed under RADC contractual and in-house studies. Contractual efforts were performed by the Hughes Aircraft Company of Fullerton, California, under the direction of Mr. A. E. Saari. The Guidebook has been prepared in MIL-HDBK format to facilitate its eventual release, when other DOD and Industry inputs become available, as a fully coordinated military handbook.

2. Environmental Stress Screening (ESS) programs, which are applied during the production phase, can yield significant improvements in field reliability and reductions in field maintenance costs. The benefits for the manufacturer include: a high degree of visibility as to the sources of reliability problems in the product or process, better control of rework costs, and the opportunity to determine corrective actions which eliminate the sources of reliability problems from the product or process.

3. ESS is an emerging technology and there are various approaches associated with the application of stress screens. Regardless of the approach used, the fundamental objective of ESS remains the same i.e. to remove latent defects from the product prior to field delivery. The quantitative methods contained in this guidebook extend this objective by focusing on the defects which remain in the product at delivery and their impact on field reliability. The goal of ESS programs thus becomes to reduce the latent defect population, at delivery, to a level which is consistent with the reliability requirements for the product. Reduction of the latent defect population in a production lot of electronic equipment, is accomplished by:

i

• Use of ESS results to isolate defect-failure causes followed by determining appropriate corrective actions. Effective corrective actions eliminate the source (cause) of the defect from the process or product, thereby improving manufacturing process capability.

• Use of ESS to precipitate flaws in the assembled hardware to a detectable level coupled with the use of thorough tests to facilitate their detection and removal.

4. The guidebook is organized according to the general sequence of events to be undertaken by the contractor in planning, monitoring and controlling a screening program. Five detailed procedures, contained in Section 5 of the guideline, are used to assist the user in accomplishing ESS planning and evaluation activities. The detailed procedures are entitled:

- Procedure A Planning Estimates of Incoming Defect Density
- Procedure B Screen Selection and Placement
- Procedure C Failure-Free Acceptance Tests
- Procedure D Cost Effectiveness Analysis
- Procedure E Monitoring, Evaluation and Control

Section 2 of the guidebook lists references used in its preparation, and Section 3 defines terms and acronyms used. Section 4 and the appendices contain general guidelines and the background information necessary for understanding the rationale for and use of the detailed procedures.

The product development phase is used to experiment with various stress screens and to define and plan a cost effective screening regimen for production. (Section 4.8 and 4.9). After the screening program is implemented during production, stress screening results are used to evaluate the screening process to establish whether program objectives are being achieved. (Section 4.11). Quantitative objectives for the screening program must be established early. Appendix B provides the rationale used for establishing quantitative goals which are related to reliability requirements for the product. Appendix A contains the mathematical relations and model descriptions used in the guidebook. A review of Appendix A will help the interested reader in gaining a quick understanding of the rationale and methodology of the guidebook. Appendix C provides the derivation of the Failure Free Acceptance Test.

5. It should be noted that is is not possible to remove all defects from the hardware through stress screening. The vast majority of parts in the hardware have failure rates sufficiently low so that they never fail throughout the life of the product. Gross latent defects tend to fail early and dominate the reliability of fielded products during early life. The objective is to remove as many of the gross defects from the hardware as is technically and economically feasible so as to achieve the designedin reliability. The guidebook implements these objectives through use of controls on the defects remaining in the hardware at delivery, the costs to precipitate and remove them, and the assurance needed that reliability objectives have been achieved.

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TABLE OF CONTENTS

| 1.1 1.2 1.3 1.3.1 1.3.2 1.3.3 1.3.3.1 1.3.4 1.3.5 | Scope Purpose. Application. General. What is ESS? Development and Production Phase Reliability Assurance. ESS Application and the Quantitative Approach. The Quantitative Approach. Benefits of the Quantitative Approach. Process Capability and Defect Density. Organization of the Guidebook. | 1 1 1 2 3 4 6 7 7 |
|---|--|---|
| 2.1 | Referenced Documents Government Documents Other Publications | 10 |
| 3.1 | Definitions and Acronyms Definitions Acronyms | 12 |
| 4.1 4.2 4.3 4.4 4.5 4.6 4.7 4.8 4.9 4.10 4.10.1 4.10.2 4.10.2.1 4.10.2.3 4.10.2.3.1 4.10.2.3.2 4.10.2.3.4 4.10.3 4.10.3.1 4.10.3.2 4.10.3.3.1 4.10.3.3.1 4.10.3.3.2 4.10.3.3.3 | General Guidelines.Relation of ESS to MIL-STD-785 Program Tasks.Contractual Aspects of ESS.Subcontractor and Supplier Screening.Costs of ESS vs Productivity Improvement.Air Force R&M 2000 ESS Policy - Part Fraction Defective.Part vs Assembly Fraction Defective.Part Level Screening and Rescreening.Development Phase Screening.ESS Experimentation - Pre Production Prototypes.Planning an ESS Program for Production.Establishing GoalsCategories of Defects.Categories of Defects.Factors which Impact Defect DensityPart Quality LevelField Stress Environment.Process MaturityScreen Selection and Placements - Factors Effecting.Goals on Remaining Defect DensityIncoming Defect Density.Screen EffectivenessPre and Post Screen Testing and Screen EffectivenessPre and Post Screen Testing During Production.Screen ParametersScreen Parameters< | 16171891912225229332457388999042344444 |

PAGE

÷U.

PAGI

| 4. 10.3.4.1 4. 10.3.4.2 4. 10.3.5 4. 10.3.6 4. 10.3.7 4. 10.3.8 4. 10.4 4. 10.4.1 4. 10.4.1 4. 10.4.2 4. 10.5 4. 11 4. 11.2 4. 11.3 4. 11.4 4. 11.4.1 4. 11.4.1 4. 11.4.3 | Power On vs Power Off. Thermal and Vibration Response Characteristics. Design Limits. Facilities and Costs. Failure-Free Acceptance Test. Preparation of ESS Plans. Development Phase Plan. Production Phase Plan. Guidelines for Initial Screen Selection and Placement. Production Phase Monitoring and Control. Data Collection. Failure Classification. Preliminary Analysis of Screen Fallout Data. Analysis of Screen Fallout Data. Quality Control Charts. Use of CDE Model | 48 55 55 55 55 55 55 55 55 55 55 55 55 55 |
|--|---|--|
| 5. 5.1 5.2 5.2.1 5.2.2 5.2.2.1 5.2.3 5.2.4 5.3 5.3.1 5.3.2 5.3.3 5.3.4 5.4 5.4.1 5.4.2 5.4.3 5.4.4 5.5 5.5.1 5.5.2 5.5.3 5.6 5.6.1 | Detailed Guidelines. ESS Planning, Monitoring and Control Procedures Procedure A - Planning Estimates of Incoming Defect Density Objective. Methodology Equipment Breakdown Procedure Steps. Part Fraction Defective Tables- Procedure B - Screen Selection and Placement. Objective. Methodology Procedure Steps. Screening Strength Tables. Procedure C - Failure-Free Acceptance Test (FFAT) Objective. Methodology. Procedure Steps. Tables of Lower Confidence Bound on Yield. Procedure D - Cost Effectiveness Analysis. Objective. Methodology. Procedure Steps. Tables of Lower Confidence Bound on Yield. Procedure D - Cost Effectiveness Analysis. Nethodology. Procedure Steps. Procedure E - Monitoring, Evaluation and Control. | 65 67 67 68 92 55 57 67 68 92 55 57 03 03 00 00 00 00 00 00 00 00 00 00 00 |
| 5.6.2 5.6.3 5.6.4 | Control - Methodology Procedure Steps | 08 11 16 |

PAGE

| 5.6.5 5.6.6 5.6.7 5.6.8 5.6.9 5.6.10 | Procedure Steps | 17 18 20 20 |
|---|--|----------------------------|
| Append is | <u>k A</u> - Stress Screening Mathematical Models | A 1 |
| 10. 10.1 10.2 10.3 10.4 10.5 10.6 10.7 | General Defect Density Defect Density and Yield Screening Strength Screening Strength and Yield Test Detection Efficiency Test Strength Relationship Between D _R and D _{IN} | A1 A2 A4 A4 A5 |
| 10.8 10.9 10.10 | Cost Effectiveness of ESS Programs Chance Defective Exponential Model (CDE) Relating D _R to Field Reliability | 8A |
| Append is | <u>x B</u> - Establishing Goals for Remaining Defect Density | B 1 |
| 10. 10.1 10.2 | General Failure Rate Function - CDE Model Remaining Part Fraction Defective Goal | B1 |
| Appendi: | <u>x C</u> - Development of Failure-Free Acceptance Test Requirements | C1 |
| 10. 10.1 | General | |

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List of Tables

Downloaded from http://www.everyspec.com

| <u>Table #</u> | Title | Page |
|--|--|--|
| Table 4.1 Table 4.2 | Manufacturer's Receiving Inspection Test Results Results of Retesting Incoming Receiving Test Micro circuit Rejects | . 22 . 23 |
| Table 4.3 Table 4.4 Table 4.5 | Remaining Defect Density Goals (D _p) Defect Types and Density vs Process Maturity Assembly Defect Types Precipitated by Thermal and | . 37 |
| Table 4.6 Table 4.7 Table 4.8 Table 5.1 Table 5.2 Table 5.2 Table 5.3 Table 5.4 Table 5.5 Table 5.6 Table 5.7 Table 5.8 Table 5.9 Table 5.10 Table 5.12 Table 5.13 Table 5.13 Table 5.14 | Vibration Screens Detection Efficiency vs Test Types Fault Coverage For Automatic Test Systems Fault Detection For a 1000 PCB Lot Size Guidelines for Initial Screen Selection and Placement Quality Levels and Grades Part Fraction Defective Microelectronic Devices. Part Fraction Defective Microelectronic Devices. Part Fraction Defective Transistors Part Fraction Defective Diodes Part Fraction Defective Resistors. Part Fraction Defective Resistors. Part Fraction Defective Capacitors Part Fraction Defective Inductive Device. Part Fraction Defective Rotating Devices. Part Fraction Defective Relays Part Fraction Defective Switches Part Fraction Defective Connections. Part Fraction Defective Connectors. Part Fraction Defective Printed Wiring Boards. Screening Strength and Failure Rates, Random Vibration. | . 49 . 49 . 56 . 70 . 73 . 74 . 75 . 76 . 77 . 78 . 80 . 81 . 82 . 83 . 84 |
| Table 5.15 Table 5.16 Table 5.17 | Screens Screening Strength, Temperature Cycling Screens Failure Rates, Temperature Cycling Screens Screening Strength and Failure Rates, Swept-Sine | . 92 |
| Table 5.18 | Vibration Screens Screening Strength and Failure Rates, Constant Temp erature Screens | |
| Table 5.19 | 90% Lower Confidence Bound on Yield $\frac{\overline{\lambda}_D}{\lambda_0}$ (.1-1.0) | . 98 |
| Table 5.20 | 90% Lower Confidence Bound on Yield $\frac{\tilde{\lambda}_{D}}{\lambda_{0}}$ (1-60) | . . 9 8 |
| Table 5.21 | 80% Lower Confidence Bound on Yield $\frac{\overline{\lambda}_D}{\lambda_0}$ (.1-1.0) | . 99 |

| | _ · | PAGE |
|------------|--|------|
| Table 5.22 | 80% Lower Confidence Bound on Yield $\frac{\overline{\lambda}_D}{\overline{\lambda}_0}$ (1-60) | 99 |
| Table 5.23 | 70% Lower Confidence Bound on Yield $\frac{\overline{\lambda}_D}{\lambda_D}$ (.1-1.0) | 100 |
| Table 5.24 | 70% Lower Confidence Bound on Yield $\frac{\overline{\lambda}_D}{\lambda_0}$ (1-60) | 100 |
| Table 5.25 | 60% Lower Confidence Bound on Yield $\frac{\overline{\lambda}_D}{\lambda_D}$ (.1-1.0) | 101 |
| Table 5.26 | 60% Lower Confidence Bound on Yield $\frac{\overline{\lambda}_D}{\lambda_0}$ (1-60) | 101 |
| Table 5.27 | 50% Lower Confidence Bound on Yield $\frac{\overline{\lambda}_{D}}{\lambda_{0}}$ (.1-1.0) | 102 |
| Table 5.28 | 50% Lower Confidence Bound on Yield $\frac{\overline{\lambda}_{D}}{\lambda_{0}}$ (1-60) | 102 |
| Table 5.29 | Comparison of Actual vs Planned Defect Density (D _{IN}) and Test Strength (TS) Values | 110 |
| Table 5.30 | 90% Control Probability Intervals | 121 |
| Table B-1 | Remaining Defect Density vs Specified Failure Rate | B4 |
| Table B-2 | Remaining Defect Density vs Specified Failure Rate | B5 |
| Table B-3 | Remaining Defect Density vs Failure Rate (PPM Method) | B7 |

<u>R</u>

ix

List of Figures

| Figure # | <u>Title</u> Pa | <u>ige</u> |
|--|---|-------------|
| Figure 1.1 Figure 1.2 Figure 1.3 | The Quantitative Problem Stress Screening Variables Sequence of Tasks in Planning, Evaluating and Controlling an ESS Program | 4 5 9 |
| Figure 4.1 | Fraction Defective Assemblies vs Remaining | 20 |
| Figure 4.2 Figure 4.3 Figure 4.4 | Defect Categories and Product Life Failures | 36 |
| | Defective Exponential Model | |
| Figure 5.1 | Equipment Breakdown Chart | 58 |
| Figure 5.2 Figure 5.3 | Unit Breakdown to Assembly Level | 38 50 |
| Figure 5.4 | Multilevel Screening Flow Chart | 36 |
| Figure 5.5 | Identification of Items to be Screened | 37 |
| Figure 5.6 | Cost Analyses Worksheet10 |)4 |
| Figure 5.7 | Screen Fallout Data Sample |)9 |
| Figure 5.8 | Control Chart I, Incoming Defect Density | 13 |
| Figure 5.9 Figure 5.10 | Control Chart II, Fallout | 13 |
| Figure 5.11 | Failure Rate Vs Time | 17 |
| Figure A-1 | Production Screening Process Flow ModelA6 | 5 5 |
| Figure A-2 | Field Failure Rate vs Defect Density | 12 |

x

1.0 Scope

<u>1.1 Purpose</u>. This guidebook provides uniform procedures, methods and techniques for planning, monitoring and controlling the cost effectiveness of ESS programs for electronic equipment. It is intended to support the requirements of MIL-STD-785, "Reliability Program for Systems and Equipment Development and Production", Task 301, "Environmental Stress Screening".

<u>1.2 Application</u>. The guidebook is intended for use by procuring activities and contractors during development and production. It is not intended that the guidebook procedures and techniques be used in a cookbook fashion. Knowledge of the equipment and the manufacturing process is essential for a properly planned and tailored ESS program. The data base needed for a systematic approach to ESS application is not fully developed. Use of the guidebook, by Government procuring agencies and military equipment manufacturers will foster the development of an improved and broader data base.

<u>1.3 General</u>. A properly applied ESS program can significantly impact the quality and reliability of electronic products delivered to the Government. ESS is interrelated with the requirements set forth in MIL-Q-9858, "Quality Program Requirements" and MIL-STD-785, "Reliability Program for Systems and Equipment Development and Production". Quality Control is a manufacturing function and Reliability Engineering is a design function. Although the Quality and Reliability disciplines are related, in practice, they are conducted as separate programs without common objectives. The guidebook uses the ESS program as a means for integrating Quality Control and Reliability Engineering tasks so as to assure achievement of reliability objectives during manufacture.

<u>1.3.1 What is ESS?</u> ESS is a process or series of processes in which environmental stimuli are applied to electronic items in order to precipitate latent defects to early failure. An equally important and inseparable aspect of the screening process is the testing which is done, either during or following the screen, to detect and isolate the defects

which have been precipitated to failure by the screen. The screening and testing process is basically a search for defects. Manufacturing techniques for modern electronic hardware consist of hundreds of individual operations and processes through which defects can be introduced into the product. Many of the defects can be detected without the need for stress screens by use of visual inspections, functional tests and other conventional quality assurance procedures. However, a small percentage of latent defects remain undetected by obvious means and, if not removed in the factory, will eventually manifest as early life failures during product use. The inability to find defects by obvious means is a consequence of the increased complexity of modern electronic products and the processes which are used in their manufacture. ESS is the vehicle by which latent defects are accelerated to early failure in the factory. ESS can thus be viewed as an extension of the quality control inspection and testing process.

1.3.2 Development and Production Phase Reliability Assurance. ESS is not a substitute for a sound reliability program conducted during the design and development phases. The inherent reliability of the product is driven primarily by the design. However, without a viable reliability assurance program during production, the reliability which is designed into the product can be seriously degraded. An equipment will eventually pass a MIL-STD-781 reliability demonstration test, either during development or on a sample basis during production. A single equipment passing the MIL-STD-781 test does not imply that all other equipments in the production lot have the same reliability. A relatively few latent defects contained in various equipments in the lot can significantly reduce the field reliability, especially for equipments with high reliability requirements. A production reliability assurance program which complements the design/development reliability program, is therefore essential to achieving reliability objectives. A properly planned, monitored and controlled stress screening program, structured as part of a production

reliability assurance program, is the vehicle through which product reliability in manufacture can be maintained. The procedures presented in this guideline are oriented toward achieving reliability objectives through use of quantitative methods for stress screening and production reliability assurance.

1.3.3 ESS Application & The Quantitative Approach. Historically there have been two basic approaches to the application of stress screens. In one approach, the Government explicitly specifies the screens and screening parameters to be used at various assembly levels. Failure-free periods are sometimes attached to the screens, as acceptance requirements, in order to provide assurance that the product is reasonably free of defects. Another approach is to have the contractor propose a screening program which is tailored to the product and is subject to the approval of the procuring activity. Although the latter approach is preferred, neither approach is adequate since explicit objectives and the relations between the screening program and quantitative reliability requirements are never defined. Costs are also uncontrolled because some of the screens might be more efficiently performed at lower assembly levels where rework costs are lower. In addition, screening levels may far exceed the design limits of the product and result in damage to the equipment.

There are several unknowns associated with the application of stress screens. How effective are the screens? What is considered acceptable or unacceptable fallout from a screen? How does the quantity of defects remaining in the equipment after delivery to the customer impact field reliability? The aforementioned ESS approaches do not fully address these questions. For example, if the screen fallout is "low", it is not known whether the equipment is "good" (i.e., defect-free) or whether the screen is not effective. On the other hand, if the fallout is "high", it is not known whether the incoming defect levels are inordinately high or whether the screen might be causing non-defectives to fail.

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Screens and tests are not perfect. At each stage of manufacture where screens and tests might be applied, from device level to the final system level, escapes to the next assembly stage occur and new opportunities for introducing defects are created. The number of defects which remain in the product at delivery and their impact on field reliability, however, is the primary concern.

<u>1.3.3.1 The Quantitative Approach</u>. The use of a quantitative approach to stress screening requires that the defect levels introduced into the product, the effectiveness of the screens, and reasonably acceptable values for the number of defects which remain and escape into the field be addressed. Figure 1.1 illustrates the quantitative aspects of stress screening.

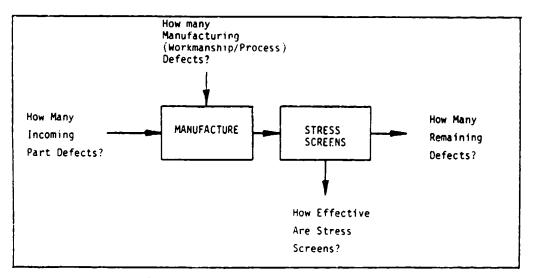


Figure 1.1 The Quantitative Problem

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When a quantitative approach to stress screening is used the variables of interest are the average number of defects per product which enter the screening process (D_{IN}) , the screen/test effectiveness or test strength (TS) and the average number of defects per product which escape the screen/test (D_{out}) . Figure 1.2 shows the relationships between these stress screening variables.

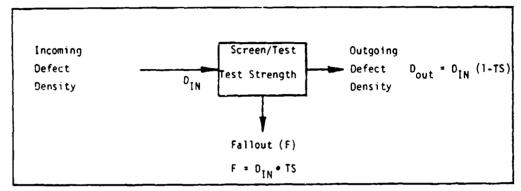


Figure 1.2 Stress Screening Variables

The number of defects remaining in the production lot at delivery is a function of three key factors:

• The quantity of design, part and manufacturing (workmanship and process) defects which initially reside in the hardware prior to assembly level screening.

The capability of the screens to precipitate flaws in assemblies to a detectable level.

• The thoroughness of the testing which is done, either during or after the screen, to assure detection and removal of the defects precipitated to failure by the screens.

None of the three factors which impact the reliability of delivered products is known with certainty. Without a basic knowledge of their quantitative value, however, effective screening programs cannot be properly planned and controlled. The procedures in the guideline are directed to obtaining both preliminary planning and measured estimates of the three factors in order to plan, monitor and control the screening process. Experience data gathered from previous screening programs, screening experiments conducted during the development phase and use of the guidebook procedures provides the methodology and information needed to plan and conduct effective screening programs.

Once a screening program is implemented during production, the results must be monitored and appropriate changes made in the screening regimen to assure that goals on remaining defects are achieved. The basic mechanism for assuring control is to compare the screening results with established goals so as to determine the need for corrective actions. Such actions could involve increasing screening or test detection capability so that more defects can be precipitated and detected or reducing incoming defect quantities through improved process controls. Changes which reduce or eliminate screening at some levels of assembly can also be taken to reduce costs, when it is found that the screens are ineffective or unnecessary.

1.3.4 Benefits of a Quantitative Approach. A quantitative approach to stress screening enables the establishment of explicit quantitative objectives and provides a basis for planning, monitoring and controlling the screening process to meet those objectives. A quantitative approach also facilitates Government and contractor communication on the status of the screening process and on the progress being made toward achieving objectives. Coupled with a good Failure Reporting Analysis and Corrective Action System (FRACAS), the quantitative approach also provides a more focused emphasis on the sources of latent reliability problems in the product or process as well as better control of costs.

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1.3.5 Process Capability and Defect Density. The use of a quantitative approach to stress screening requires addressing the capability of the manufacturing process to produce products which are reasonably free of Defects are introduced into a lot of manufactured products defects. through repeated assembly, handling and testing operations. The average number of defects per product (defect density) varies as a function of the degree of control which is exercised over the manufacturing process. When the variation is due only to random noncorrectable causes, the process is said to be in control. The range over which such variability occurs is often referred to as the process capability. In quantitative terms, the process capability can be defined in terms of a process mean (average defect density) and a standard deviation. Process capability determines defect density and not vice versa. Quality control studies are often performed to establish process capability. However, rather than ask the question: What is the process capability?, one should ask: What must the process capability be in order to meet quantitative reliability objectives? The use of a quantitative approach to stress screening focuses attention on the latter question. Analyses of screening failures should be directed to determining root causes of defects and corrective actions so as to improve the process capability. Process capability is improved only through reducing the number of failure causes which are falsely deemed to be noncorrectable.

<u>1.3.6 Organization of the Guidebook</u>. The guidebook is organized according to the general sequence of events to be undertaken by the contractor in planning, monitoring and controlling a screening program. Five detailed procedures, contained in Section 5 of the guideline, are used to assist the user in accomplishing ESS planning and evaluation activities. The detailed procedures are entitled:

- Procedure A Planning Estimates of Incoming Defect Density
- Procedure B Screen Selection and Placement

- Procedure C Failure-Free Acceptance Tests
- Procedure D Cost Effectiveness Analysis
- Procedure E Monitoring, Evaluation and Control

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Figure 1.3 shows the sequence of application of the various ESS tasks contained in the guidebook. References to the applicable sections of the guidebook are provided for each task included in the figure.

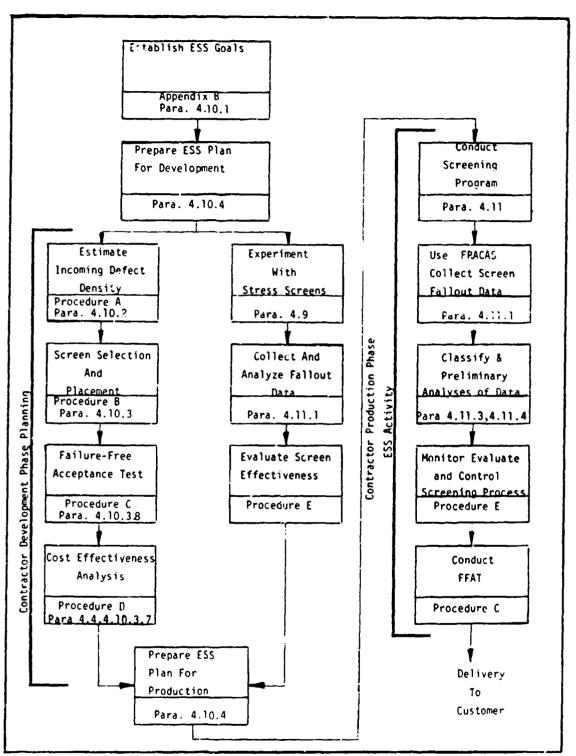


Figure 1.3 Task Sequence in Planning, Monitoring & Controlling an ESS Program

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<u>2. Referenced Documents</u> The documents cited in this Section are for guidance and information.

- 2.1 Government Documents.
- 2.1.2 MIL-STD-785, "Reliability Program For Systems and Equipment Development & Production"
- 2.1.3 MIL-STD-781, "Reliability Design Qualification and Production Acceptance Tests: Exponential Distribution"
- 2.1.4 MIL-STD-721, "Definition of Terms for Reliability and Maintainability"
- 2.1.5 MIL-HDBK-217, "Reliability Prediction of Electronic Equipment"
- 2.1.6 MIL-Q-9858, "Quality Program Requirements"
- 2.1.7 MIL-STD-883 "Test Methods and Procedures for Microelectronics"

2.2 Other Tublications.

- 2.2.1 Saari, A.E., Schafer, R.E., and VanDen Berg, S.J., "Stress Screening of Electronic Hardware", Hughes Aircraft Company, RADC-TR-82-87, May 82 (All8261)
- 2.2.2 Saari, A.E., Van Den Berg, S.J., and Angus, J.E., "Environmental Stress Screening", Hughes Aircraft Company, RADC-TR-86-149
- 2.2.3 NAVMAT P-9492, "Navy Manufacturing Screening Program", May 79
- 2.2.4 Institute of Environmental Sciences, "Environmental Stress Screening Guidelines for Assemblies", Sep 84
- 2.2.5 Institute of Environmental Sciences, "Environmental Stress Screening Guidelines", 1981
- 2.2.6 Anderson, J.R., "Environmental Burn-In Effectiveness", McDonnell Aircraft Company, St. Louis, MO, Report No. AFWAL TR-80-3086, Aug 80
- 2.2.7 Fertig, K.W., Murthy, V.K, "Models for Reliability Growth During Burn-In", Proceedings of the 1978 Annual R&M Symposium, pp. 504-509.

- 2.2.8 Bateson, J.T., "Board Test Strategies Production Testing in the Factory of the Future", Test and Measurement World, pp. 118-129, Dec 84.
- 2.2.9 Kube, F., Hirschberger, G., "An Investigation to Determine Effective Equipment Acceptance Test Methods", Grumman Aerospace Corporation, Report No., ADR 14-04-73, Apr 73
- 2.2.10 Brownlee, K.A. (1960), <u>Statistical Theory and Methodology in</u> <u>Science and Engineering</u>, New York, John Wiley and Sons

3. Definitions and Acronyms

3.1 Definitions. Definitions applicable to this guidebook are:

Detectable Failures A failure that can be detected with 100% test detection efficiency.

Failure-Free Period A contiguous period of time during which an item is to operate without the occurrence of a failure while under environmental stress.

Failure-Free Test A test to determine if an equipment can operate without failure for a predetermined time period under specific stress conditions.

Fallout Failures observed during, or immediately after, and attributed to stress screens. Symbol used F.

Part Fraction Defective The number of defects contained in a part population divided by the total number of parts in the population expressed in PPM.

Latent Defect An inherent weakness that has a high probability of resulting in an early life failure under field stress conditions.

Patent Defect A failure, usually one waiting to be detected without the need for stress screens.

Precipitation (of Defects) The process of transforming a latent defect into a patent defect.

Screen Effectiveness Generally, a measure of the ability of a screen to precipitate defects. Sometimes used specifically to mean screening strength.

Screen Parameters Parameters in screening strength equations which relate to screening strength, (e.g., vibration g-levels, temperature rate of change and time duration.)

Screening Regimen A combination of stress screens applied to an equipment, identified in the order of application (i.e., assembly, unit and system screens).

Screening Strength The probability that a screen will precipitate a defect, given that a defect susceptible to the screen is present. Symbol (SS)

Selection and Placement The process of systematically selecting the most effective stress screens and placing them at the appropriate levels of assembly.

Stress Screening The process of applying mechanical, electrical and/or thermal stresses to an equipment item for the purpose of precipitating latent part and workmanship defects to early failure.

Definitions Continued

Test Detection Efficiency A characteristic of a test measured by the ratio of the number of failure modes detectable to the total number of failure modes. The probability of detection. Symbol (DE). Test Strength The product of screening strength and test detection efficiency. The probability that a defect will be precipitated by a screen and detected in a test. Symbol (TS). Thermal Survey The measurement of thermal response characteristics at points of interest within an equipment when temperature extremes are applied to the equipment. Vibration Survey The measurement of vibration response characteristics at points of interest within an equipment when vibration excitation is applied to the equipment. Yield The probability that an equipment is free of defects when offered for acceptance. Defect Density Average number of defects per item. Symbols Used: D_{IN} and D_{R} for incoming and remaining defect density, respectively. Production Lot group of items manufactured under Α essentially same conditions and the processes. Assembly/Module A number of parts joined together to perform a specific function and capable of disassembly, For example a printed circuit board. An assembly of parts designed to function in conjunction with similar or different modules when assembled into a unit. (i.e. Printed Circuit Assembly, power supply module, core memory module.) Failure Rate The total number of failures within an item population, divided by the total number of life units expended by that population during a particular measurement interval under stated conditions. Symbol used λ . reliability measure related to MTBF. Escapes A proportion of incoming defect density which is not detected by a screen and test and which is passed on to the next level. Symbol (Dout) Item A non-specific term to denote any product includina systems, materials. parts. subassemblies, sets, accessories, etc.

(Definitions Continued)

Part

Any identifiable item within the product which can be removed or repaired (e.g., discrete semiconductor, resistor, IC, solder joint, connector).

System/Equipment A group of units interconnected or assembled to perform some overall electronic function (e.g., electronic flight control system, communications system).

Unit

A self-contained collection of parts and/or assemblies within one package performing a specific function or group of functions, and removable as a single package from an operating system (i.e., autopilot computer, vhf communications, transmitter).

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3.2 Acronyms

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G₁ -

| Procedure A of Section 5 |
|--|
| Description |
| Airborne Inhabited Cargo Airborne Inhabited Trainer Airborne Inhabited Bomber Airborne Inhabited Attach Airborne Uninhabited Fighter Airborne Uninhabited Cargo Airborne Uninhabited Trainer Airborne Uninhabited Bomber Airborne Uninhabited Attack Airborne Uninhabited Fighter Airborne Rotary Wing Cannon Launch Ground Benign Ground Fixed Ground Mobile Missile Free Flight Airbreathing Missile Flight Manpack Naval Sheltered Naval Unsheltered Naval Undersea Unsheltered Naval Submarine Naval Hydrofoil Space Flight |
| Undersea Launch |
| Description |
| Built In Test Cannot Duplicate Chance Defective Exponential Environmental Stress Screening Failure Reporting and Corrective Action System Fault Location Failure Mode & Effect Analysis Functional Board Tester In Circuit Tester In Circuit Tester In Circuit Analyzer Mean Time Between Failures Maximum Likelihood Estimate No Fault Found Parts Per Million Printed Wiring Assembly Performance Monitoring Retest OK Test Analyze & Fix |
| |

4. General Guidelines

4.1 Relation of ESS to Reliability Program Tasks. Planning an ESS program for the production phase is interrelated with many of the MIL-STD-785 reliability program tasks which are required to be performed during development and production. Every effort should be made to integrate the knowledge gained from MIL-STD-785 tasks into the planning of an ESS program MI_-STD-785 reliability program tasks which have a for production. particular bearing on ESS planning include: Reliability Prediction (Task 203), Allocation (Task 202), Qualification Tests (Task 303), Parts Control (Task 207), Failure Reporting Analysis and Corrective Action System (Task 104), Failure Modes and Criticality Analysis (Task 204), Reliability Growth Testing (Task 302), and of course, ESS (Task 301). Proper screen selection and placement is highly dependent on the reliability and stress Information derived from design characteristics of the equipment. reliability program tasks such as: predicted and demonstrated failure rates, quality level of parts, number and type of nonstandard and MILparts, number and type of interconnections, design capability, field stress environments, and critical items should be used in structuring an ESS program for production.

<u>4.2 Contractual Aspects of ESS</u>. ESS must remain an adaptive process so that the screening regimen can be changed to improve cost-effectiveness. Contract provisions for ESS programs should have flexibility to affect necessary modification of stress screens. During the initial stages of production more severe stress screens may be required. As the product and process mature, the screens may require adjustment such as by reducing the number of temperature cycles, the number of axes of vibration or by eliminating unnecessary screens. In early production, a number of unknowns precludes adoption of optimum stress screening. Some of the more significant unknowns are:

- residual design deficiencies
- manufacturing planning errors

- worker training
- new suppliers

Serie os in

Service of

- Intent defects in new part lots
- new process capability
- stress screening effectiveness
- testability (for defect detection)

The stress screening program, even if carefully planned, may produce unexpected results which should be addressed through modification of the screens. The principle of adaptive screening is to adjust the screens on the basis of observed screening results so that the screens are always most cost effective. Contract terms should be flexible enough to permit modification of screens or screen parameters when such modification can be shown to be beneficial.

In long term production the quantity and distribution of latent defects change with time and therefore contract terms should contain provisions for periodically reassessing the individual screens and the overall screening program. The overriding criterion for change should be the most cost effective achievement of objectives. Contracting arrangements should be made which permit such changes without having to resort to extensive renegotiation.

<u>4.3 Subcontractor and Supplier Stress Screening</u>. Items which are furnished by subcontractors or other equipment suppliers may require stress screening. There are several distinct advantages for the subcontractor or supplier to perform the stress screening rather than the prime contractor.

(a) Subcontractor/supplier concern for yield can be translated to profits which may force process improvements to minimize latent defects.

(b) Screening at receiving inspection/test, by the prime contractor, may involve returning defective items to the subcontractor/supplier and result in shortages and schedule slippages.

(c) Special stress screening facilities and test equipment do not have to be purchased, supported and operated by the prime contractor.

The procedures and methodology contained in the Guidebook can be imposed on the subcontractor/supplier. To assure that the subcontractor/supplier is able to perform the tasks required by the guidebook, the intent must be made known prior to production. In this manner, the subcontractor/supplier can prepare a screening plan, acquire the necessary capability or arrange for an external laboratory to perform the screening.

4.4 Costs of ESS vs Productivity Improvement. The costs of conducting a screening program during the production phase can be high. To a large extent, the costs can be offset by the increased productivity which results through proper screen selection and placement. Screening at the lowest possible level of assembly will always be the least costly alternative in terms of rework costs. The time and effort required to test, troubleshoot and repair items increase by at least an order of magnitude at each subsequent level of assembly. Significant cost savings or avoidance can accrue to the manufacturer by analyzing the cost benefits of various screen selection and placement alternatives and by striving to find defects at the lowest possible level of assembly. Cost savings to the Government will result through improved field reliability and corresponding reductions in field repair costs. The benefits of a properly conducted ESS programs to the Government go beyond field repair costs alone. Improved reliability during early life will also reduce over-buying of spares, since estimates of required spare quantities are based upon early life field performance. The opportunity for introducing new defect sources into the hardware during field maintenance and handling is also reduced.

There should be however, controls and constraints on the cost of conducting a screening program. Situations can arise where the cost of conducting a screening program far outweigh any benefits which may be derived. For example, for low complexity items the number of screenable defects which are likely to be present in the hardware may be relatively Conducting a full-scale screening program, in such cases, can small. result in very high costs per defect eliminated. Cost of \$10K to \$15K per defect eliminated may be justified for equipments which are used in critical missions with very high reliability requirements. On the other hand, such costs may be difficult to justify if the equipment is used in noncritical missions and if the costs of field maintenance are not severely affected by not screening. Each case, where a stress screening program is under consideration, must be judged individually as to the cost benefits to be derived from stress screening. Procedure D, in Section 5 of the Guidebook, is used to determine the cost-effectiveness of ESS programs.

<u>4.5 Air Force R&M 2000 ESS Policy-Part Fraction Defective</u>. Air Force R&M 2000 ESS policy requires that the manufacturing process begin with piece parts having a remaining part fraction defective below 1000 PPM (parts per million) by FY87 and below 100 PPM by FY90. ESS results from first assembly screens are used in the Guidebook procedures to evaluate the achivement of these goals. In terms of the reliability of delivered systems, the R&M 2000 ESS requirements can be extended to include goals on remaining part fraction defective for the system at delivery. Appendix B of the Guidebook discusses a method which uses a 50 PPM part fraction defective goal for delivered systems to establish quantitative ESS program goals.

<u>4.6 Part vs Assembly Fraction Defective</u>. The part fraction defective can have a significant impact on the assembly fraction objective depending upon the number of parts contained in the assembly. The Poisson approximation is used in Figure 4.1 to illustrate the expected assembly fraction defective as a function of the remaining part fraction defective and the number of parts per assembly.

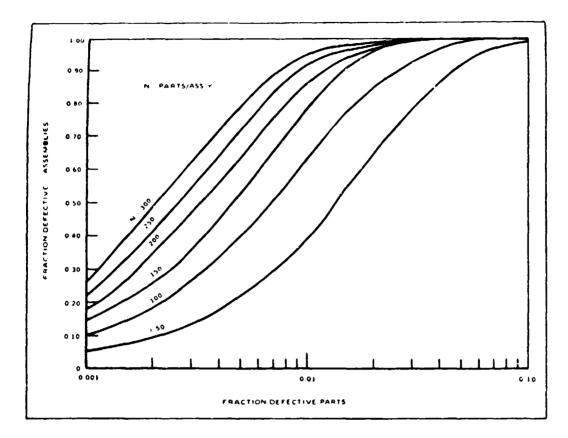


FIGURE 4.1. FRACTION OF DEFECTIVE ASSEMBLIES VS REMAINING PART FRACTION DEFECTIVE

As can be noted relatively small values of part fraction defective result in large values of assembly fraction defective depending upon the number of parts contained in the assembly. As an example, for a 150 part assembly containing parts with a fraction defective of .01 (10,000 PPM), the defect density is 1.5 and the assembly fraction defective is about .8. In terms of yield only about 20% of such assemblies, when subjected to first assembly test, would pass without failure. It is quite obvious that the part fraction defective must be much better than .01 if the costs of rework, retesting and handling of the assemblies are to be avoided. Elimination of defectives through part level screening is obviously the most cost effective course of action. However, the following questions can

be posed: How much better must the remaining part fraction defective be?; What level of part fraction defection is needed for delivered systems?; and Can such levels be achieved? A part fraction defective of .001 (1000 PPM) on a 300 part assembly, results in an assembly fraction of about .27. Although not shown in Figure 4.1, a part fraction defective of .001 (1000 PPM) on a 1000 part assembly or on a 10,000 part system gives an assembly/ system fraction defective of .63 and .99995 respectively. The foregoing implies that for the 1000 part assembly, an average of 37 out of 100 assemblies would be defect free and for the 10,000 part system only about 5 systems in 100,000 would be defect free. Extending the same example to the case where the part fraction defective is .0001 (100 PPM), then an average of 99 of 100, 1000 part assemblies would be defect free and for the 10,000 part system an average of 37 out of 100 systems would be defect free. It would, therefore, appear that levels of part fraction defective of less than 100 PPM are needed or should at least be established as goals for delivered systems. The degree to which such goals can be achieved is dependent upon the emphasis placed on finding defects during screening and eliminating their cause(s) through corrective action. A method for establishing goals on remaining defect density at delivery, which is discussed in Appendix B, uses a 50 PPM goal on part fraction defective for each system delivered to the field.

<u>4.7 Part Level Screening/Rescreening</u>. Screening at the part level is the most cost effective alternative for eliminating defects prior to the parts being assembled into the production hardware. A population of parts, even those procured to high quality levels, may appear to contain inordinately high fraction defective levels when the parts are retested. For example, microelectronic devices procured to the quality requirements of MIL-STD-883 receive 100% final electrical testing by the part manufacturer. None-theless, one manufacturer has found that about 1%, and as much as 4% of the parts will not pass a similar electrical test performed at the manufacturer's receiving inspection. There are several possible reasons for this including:

- the seller's and buyer's tests are different
- seller testing errors
- buyer testing errors
- device damage or degradation in handling
- inspection and sorting errors.

Table 4.1 shows the percent rejected for a large quantity of devices which were subjected to rescreening at the equipment manufacturers facility. The data indicate that the remaining part fraction defective (per cent rejected column) after the initial screening was done by the part vendor, is clearly unacceptable. However, as is typical with such data, it is not known whether the parts are truly of poor quality or whether testing errors or handling damage could be the cause.

Table 4.1 Manufacturer Receiving Inspection Test Results.

| Part Type | Quantity | Average Quality | Rejects | Per Cent Rejected |
|-------------------------|-----------|--------------------|---------|----------------------|
| Microcircuits | 1,419,581 | B-1 | 13,779 | 0.97/9706PPM |
| Discrete Semiconductors | 343,000 | Tx | 2,008 | 0.59/5854PPM |
| Passives | 1,296,200 | ER-M | 8,539 | 0.65/6538PPM |

Reference 2.2.1

The quantities of remaining defects in a population of screened or rescreened parts is, at best, uncertain. Screens and tests are not perfect and if a lot of parts are subjected to a series of screens, rescreens and tests, on a 100% basis, the observed fraction defective does not provide information on the remaining fraction defective. In fact, poor screens and tests will indicate very low observed part fraction defectives. Part level screening should result in a remaining part fraction defective of at least .01% (100 PPM) to avoid costly rework during manufacture and to ensure adequate reliability in the product. Most statistical sampling plans contain provisions which establish average outgoing quality, but the assumption underlying such plans is that the screens and tests are 100% effective. To determine the fraction of incoming microcircuit test rejects that were actually defective, another manufacturer performed a retest of 525 rejects from a population of 75,981 tested devices. The results are shown in Table 4.2.

| Table 4.2 Results | of Retesting | Incoming Receiving | Test Microcircuit | : Rejects |
|-------------------|--------------|--------------------|-------------------|-----------|
| | | | | |

| # of | | Total | Rejects | | | Verified | (See Note) |
|---------------|------|--------|---------|------|------|----------|------------|
| Supplier Lots | Qty. | Total | X | Pass | Fail | 7 Fail | |
| ^ | 25 | 8525 | 100 | 1.17 | 62 | 32 | 0.38 |
| E | 8 | 8435 | 22 | .26 | 15 | 7 | 0.08 |
| C | | 21826 | 166 | .70 | 120 | 46 | 0.21 |
| ้อ | 30 | 27295 | 144 | . 53 | 35 | 102 | 0.37 |
| E | 22 | 9471 | 96 | 1.01 | 31 | 63 | 0.67 |
| ŗ | 2 | 429 | 6 | 1.40 | : | 2 | 0.47 |
| TOTALS | 104 | 75,981 | 534 | 0.70 | 267 | 255 | 0.34 |

Reference 2.2.1

MOTE: S25 of the S34 rejects were retested. Percent failed shown in last column is the percent of the total quantity tested.

As can be noted, about 50% of the rejects were found to be defective but 50% could not be verified as rejects i.e. they passed the retest. Such large discrepancies can be the result of any one or a combination of the reasons listed above. On a lot basis, other data show that about 75-80% of the lots subjected to rescreening show zero failures and only 5% of the lots produce 90-95% of the failures. These data suggest that the differences are traceable primarily to lot problems and very likely, would be found to correlate with chronic problems specific to particular device types or manufacturers. Corrective action for specific problems should be determined rather than resorting to retesting of reasonably defect-free lots with the attendant possibility of handling damage or testing errors.

Screening at the assembly level is a costly means of finding and eliminating part defects from the hardware. There are always uncertainties as to whether the part defects which are found during assembly level screening are escapes from part level screens, or whether they are newly introduced defects due to handling, test and assembly operations. The part fallout from early screening at the assembly level can provide much of the information needed for resolving such uncertainties and taking corrective action. If the part fallout at assembly level screening is greater than 1000 PPM, at most, then rejection of suspect lots, changing vendors, or negotiation of corrective action with the part vendors should be made. A thorough failure analysis of the part fallout from assembly level screening can help in determining the types of screens which should be used, at the part level, for eliminating specific defect types.

4.8 Development Phase Screening. Screening during the development phase is primarily intended as an experimental activity to gather information on the quantity and type of defects likely to be present in the production hardware and the effectiveness of screens which might be applied. When the development hardware is similar to the production configuration, the knowledge gained from screen experimentation can be invaluable for coping with the problem during production. However, the development hardware can, in some instances, be an advanced development model in which a technical concept is being validated and the hardware used bears little resemblance to the production hardware. In addition, for some high volume production programs, a production engineering phase (PEP) may follow development in which major hardware design changes are made to enhance producibility. Suppliers and vendors used during development may also change for production. The system may contain many nonstandard parts substituted due to lead time problems. Screening fallout data for nonstandard parts would not be representative of production. It would also be difficult to obtain a measure of workmanship or process latent defects because the been fabricated in engineering laboratories. hardware may have

In addition, experience has shown that about one-half of development phase failures are design related. The lack of disciplined electrostatic discharge/electrical overstress (ESD/ECS) controls can result in failures during development testing, which may not occur under more controlled production conditions. The combination of one or more of these conditions during development will tend to overshadow information needed for planning a production screening program. It may therefore be somewhat unrealistic to expect that accurate screening data can be derived from some development phase screening for use in production planning.

4.9 ESS Experimentation - Pre-Froduction Prototypes. Use of the procedures contained in the guidebook in conjunction with stress screen experimentation on pre-production prototype equipment can provide invaluable data for planning. Estimates of the type and quantity of defects likely to be present in the hardware can be evaluated against experimental data. Screens can be designed based upon engineering evaluation, which provide the desired stess stimulation for suspected defect sites in the hardware. Test specifications can also be evaluated to ensure that possible failure modes, arising from various defect types and sources, can be detected by the tests performed either during or following the screens. Integration of the results from the MIL-STD-785 reliability program tasks can also be effectively accomplished. Early fallout from screens provides the maximum amount of information on likely defect sources and process capability. Corrective actions taken as a result of screen experimentation during development can aid significantly in stabilizing the process for produc-In addition, planning estimates of incoming density, screening tion. strength and test detection efficiency can be refined as a result of experimenting with stress screens.

<u>4.10 Planning a Stress Screening Program for the Production Phase</u>. Planning a stress screening program for production must begin early in the development phase. The success of a stress screening program is strongly dependent on knowledge of the product and the processes to be used in manufacture. The following must be kept in mind when planning a stress screening program using quantitative methods.

• The defects which can potentially reside in the product and the effectiveness of screens in precipitating the defects to failure are not known with certainty. By comparison of planned estimates for defect fallout with actual screen fallout, the screening and manufacturing process can be adapted to achieve desired goals.

• Screening experience data on equipment similar in composition, construction and degree of maturity, can provide very useful data for planning purposes. Information derived from the following sources should be used in planning an ESS program for production.

- a. Identification of hardware items (parts, assemblies) which have exhibited a high incidence of latent defectives on other programs.
- b. Identification of suppliers/vendors whose products have indicated high defect levels.
- c. Qualification test results.
- d. Supplier acceptance test results.
- e. Part receiving inspection, test and screening results.
- f. Screening and test records for previous programs.
- g. Reliability growth test results.

• A viable screening program must be dynamic, i.e. the screening process must be continuously monitored to ensure that it is both technically and cost effective. Changes to the screening process should be made, as necessary, based on analysis of screening fallout data and failure analysis so that quantitative screening objectives can be achieved.













The basic questions which must be addressed in planning a stress screening program are:

- a. What are the quantitative objectives of the program?
- b. What are the stress screens to be used and what level of assembly should the screens be placed to achieve the desired objectives?
- c. What are the costs associated with each of the possible alternative screening sequences and how can the screening program be made cost effective?
- d. How will one know if the screening program is proceeding according to plan? What assurances can be provided that program objectives have been achieved?
- e. What corrective actions must be taken to achieve desired screening program goals if the screening fallout data indicate significant departures from the planned program?

These questions are each addressed through use of the procedures in Section 5 of the Guidebook.

Planning an ESS program for the production phase includes the following major tasks:

Establishing Objective/Goals

- Obtaining Initial Estimates of Defect Density
- Selection and Placement of Screens
- Preparation of Production ESS Plan

A discussion of each of these major tasks which includes background, rationale and general guidelines for use of the detailed procedures is contained in the following Sections 4.10.1 - 4.10.5.

<u>4.10.1</u> Establishing Goals/Objectives. Expressed quantitatively, the objective or goal of a stress screening program is to reduce the incoming defect density in a production lot of equipment to an acceptable remaining defect density in a cost effective manner. Methods discussed in Appendix B provide the basis for establishing goals on remaining defect density. A set of sample calculations of remaining defect density is shown in Table 4.3, as a function of the specified or predicted failure rate (λ_0) in failures per operating hour.

| Specified Failure Rate ^λ ο | MTBF | D _R |
|---|--------|----------------|
| .1 | 10 | 10 |
| .01 | 100 | 1 |
| .005 | 200 | .5 |
| .002 | 500 | .2 |
| .001 | 1000 | .1 |
| .0005 | 2000 | .05 |
| .0002 | 5000 | .02 |
| .0001 | 10000 | .01 |
| .00001 | 100000 | .0001 |

Table 4.3 Remaining Defect Density Goals (D_R)

A simple relation for obtaining goals for remaining defect density is to multiply the predicted or specified failure rate λ_0 for the system by 100.

$$100 \lambda_0 = D_R$$

The remaining defect density D_R is directly related to yield, i.e. $D_R = -\ln p_R$ yield. Yield or D_R is the goal at which planning, monitoring and controlling the screening process is aimed.

<u>4.10.2</u> Obtaining Planning Estimates of Defect Density. The design of a stress screening program requires knowledge of the quantity and type of defects which are likely to reside in the prior to assembly level screening. The part fraction defective tables contained in Procedure A of Section 5 of the Guidebook are used to obtain planning estimates of defect density. Values in the tables are based upon studies of historical defect data from the factory and field for several part types. Extrapolations to other part types and field environments were made based upon correlations to MIL-HDBK-217 quality level and field environment factors. As more experience data on part fraction defective are gathered the estimates will be improved. Study results and methodology are contained in reference 2.2.2.

<u>4.10.2.1 Latent vs Patent Defects</u>. A common understanding of the nature of the defects which the screening program should be designed to precipitate, is essential for proper planning. The factors which impact incoming defect density and the rationale for the procedures used in obtaining planning estimates of defect density should also be understood. As will be shown, there can be no precise definition of the term "defect".

In a simple context, a defect can be defined in terms of an out-oftolerance or specification condition which can be readily detected by an inspection or test procedure. Such defects are termed patent defects. Patent defects represent the majority of the defect population in an equipment and are readily detected without the need for stress screens. A smaller percentage of defects however, cannot be detected by conventional means. Such defects are termed latent defects. Latent defects are characterized as an inherent or induced weakness or flaw in a material which will manifest as an early or premature failure in the operational environment.

Both patent and latent defects are introduced into the product during fabrication, assembly, handling and test operations. The patent defects pass through various assembly stages until they are detected by a test or inspection of sufficient thoroughness and are subsequently eliminated from the product. When good quality control test and inspection procedures are applied, all but the most subtle patent defects should be detected and eliminated prior to shipment. Some examples of patent defects are:

Patent Defects

- Parts
 - Broken or damaged in handling
 - Wrong part installed
 - Correct part installed incorrectly
 - Failure due to electrical overstress or to electrostatic discharge
 - Missing parts
- Interconnections
 - Incorrect wire termination
 - Open wire due to handling damage
 - Wire short to ground due to misrouting or insulation damage
 - Missing wire
 - Open etch on printed wiring board
 - Open plated through hole
 - Shorted etch
 - Solder bridge
 - Loose wire strand

Latent defects cannot be detected until they are transformed to patent defects by environmental stress applied over time. Stress screening is the vehicle by which latent defects are transformed into detectable failures. Some examples of latent defects are:

Latent Defects

- Parts
 - Partial damage through electrical overstress on electrostatic discharge
 - Partial physical damage during handling
 - Material or process induced hidden flaws
 - Damage inflicted during soldering operations (excessive heat)
- Interconnections
 - Cold solder joint
 - Inadequate/excessive solder
 - Broken wire strands
 - Insulation damage
 - Loose screw termination
 - Improper crimp
 - Unseated connector contract
 - Cracked etch
 - Poor Contact termination
 - Inadequate wire stress relief

<u>4.10.2.2</u> Categories of Defects. The majority of parts and connections within an electronic equipment are "good" and will never fail over the product's lifetime. The failures which occur during product life are traceable to design or externally induced causes, or to latent defects which were introduced into the product during manufacture. Not all latent defects however, are screenable i.e., capable of being removed from the equipment in the factory by use of stress screens. It is only those latent defects, whose failure threshold can be accelerated by the stresses imposed by the screens, which are screenable. Such screenable defects, if not removed from the field. It is the screenable early life failure which the stress screening program must be designed to remove. Figure 4.2 illustrates the categories of defects and their relationship to product life failures.

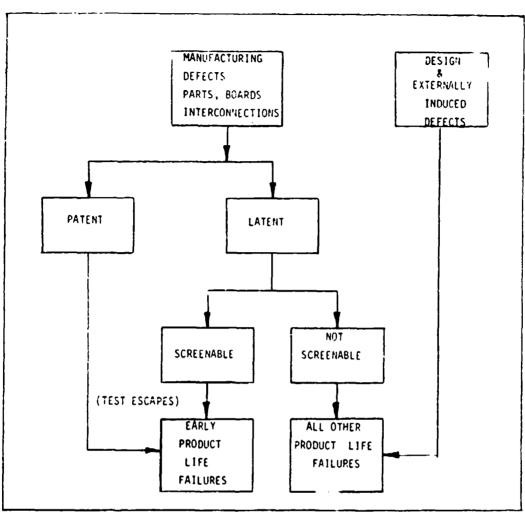


Figure 4.2 Defect Categories & Product Life Failures

<u>4.10.2.3 Factors Which Impact Defect Density</u>. The quantity and type of defects which are introduced into a product are dependent upon several factors. The first five factors, listed below, are related to product or program characteristics for which the manufacturing function within a company has little control. The last two factors are related to the manufacturing process for which the product manufacturer has direct control.

a. <u>Complexity</u> - The quantity and type of parts and interconnections used in the product effects defect density. Increased complexity creates more opportunities for defects. b. <u>Part Quality Level/Grade</u> - The quality levels of parts are established by Mil-Std part screening requirements. The number of defects which remain in a lot of screened parts is determined by the type and extent of screening and testing to which the parts are subjected under Mil-Std screening requirements.

c. <u>Field Stress Environment</u> - The stress conditions to which the equipment will be exposed in the field environment will affect the proportion of defects which should be screened from the product. A defect may be precipitated to early failure in a harsh field operating environment, but may survive product life in a benign field environment.

d. <u>Process Maturity</u> - New production requires time to identify and correct planning and process problems, train personnel and to establish vendor and process controls. Maturity is dependent on volume and time. Low production volume over a long period would have a low maturity rate and will thus impact defect density.

e. <u>Packaging Density</u> - Electronic assemblies with high part and wiring density are more susceptible to process, workmanship and temperature induced defects due to smaller error margins, increased rework difficulty and thermal control problems.

The following factors are under the direct control of the manufacturer. The degree of control exercised will determine defect density. Screen fallout data provide the necessary input for determining out-of-control conditions.

f. <u>Manufacturing Process Controls</u> - Good process controls will tend to reduce the number of defects which are introduced into the product. The criteria by which processes are considered to be in or out of control should be established by reliability requirements and monitored using the fallout from the screening process.

g. <u>Workmanship Quality Standards</u> - Stringent and properly enforced workmanship quality standards will enhance the reliability of the product through reduced introduction of workmanship defects into the product. The levels to which quality standards should be established and monitored must also be dictated by reliability requirements and made visible by the screening process.

4.10.2.3.1 Relationship of Part Fraction Defective to Quality Level. The number of defects which reside in electronic hardware is strongly dependent on the MIL-STD quality level of the parts used. An example, using microcircuit quality grades is presented to illustrate ideas. The failure rate of different populations of microcircuits, operating under identical conditions, can vary over an order of magnitude depending on quality level (e.g. Class S versus C-1). Major differences between the Class S die and the class C-1 die include the visual inspection acceptance criteria, level of process controls, extent of screening and the electrical tests to which the dies are subjected. Since screens and tests do not make devices more reliable (they improve lot quality by eliminating some latent defective parts), a "good" class C-1 die is as "good" as a class S die. Perhaps this can be extended to "good" class D-1 die as well. Therefore, it can be postulated that the difference in the failure rate of the two populations due solely to quality level, is also a direct measure of the difference in fraction defective of those populations.

To extend the example, consider a class S, hermetic flatpack MSI device of, say, 40 gates operating with $T_j = 25^{\circ}C$ in a benign ground environment. A failure rate of 0.0032 x 10^{-6} failures per hour is calculated using MIL-HDBK-217. Let 5,000 of such devices be used in an end item expected to operate 50,000 hours. The expected number of device failures during the end item life is less than 1. For this application, the device can be considered to be "good", i.e. free of latent defectives. If a class C-1 device were used on the end item instead of the class S device, an

additional 20 failures could be expected to occur during the same end item life, due solely to the difference in quality grade. The additional 20 failures can be viewed as representing latent defectives in the population. If the class S parts were operated with $T_i = 100^{\circ}$ C instead of 25°C the increase in failure rate would result in an additional two failures during the 50,000 hours. This may indicate that the class S lot contains latent defectives that were precipitated by the increased operating temperature. There can be no precise definition for a latent defective part because the inherent flaw can range from a minor flaw (which may not be subjected to sufficient stress to cause degradation of the flaw to a hard failure) to a major flaw, which requires only a slight stress. One view is that if a part fails during the life of the end item in which it resides it is, by definition, a latent defective part (excluding wearout, design or externally induced failures). The quality level of the parts, used in a product is a major factor affecting incoming defect density. Procedure A, in Section 5 of the Guidebook uses the quality level as a factor in obtaining planning estimates of defect density.

4.10.2.3.2 Screenable Latent Defects and the Field Stress Environment.

The notion of screenable latent defects must be further examined to fully understand the rationale used for the procedures contained in the Guide-The population of latent defects within newly manufactured book. electronic items can be viewed as a continuum which ranges from minor defects of small size to major defects of large size. Defects of large size will tend to fail prematurely under normal field operating stress Defects of small size will either eventually manifest as conditions. failures, or not fail at all during product life. It is the major flaw or defect which stress screening is intended to precipitate to failure. Good manufacturing process controls will tend to reduce the number of latent defects which are introduced into the hardware. However, it is important to note a somewhat controversial point, i.e., given the same manufacturing process, the number of screenable latent defects which may reside in the hardware will differ, dependent upon the operating environment to which

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the equipment will be exposed. The stress/time to which a latent defect is exposed will determine its time-to-failure or failure threshold. The failure threshold for a defect in a harsh environment will differ from the same defect in a benign environment. To illustrate, Figure 4.3 shows that a harsh uninhabited airborne environment has a smaller time-to-failure than a ground benign environment thus affecting the proportion of a latent defect population which should be screened from the equipment.

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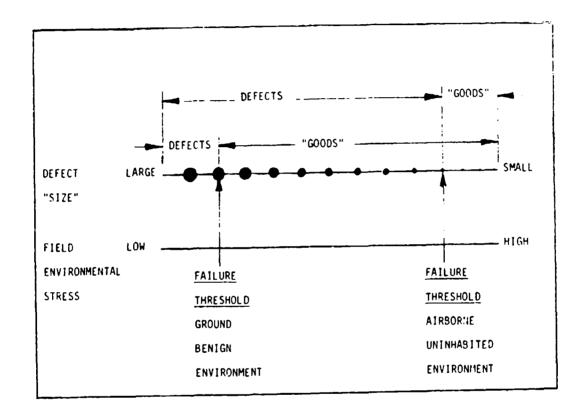


Figure 4.3 Defects and the Field Stress Environment

Obtaining an initial estimate of defect density for an equipment must take into consideration the field operating environment to which the equipment will be exposed during product life. The methods contained in Procedure A of Section 5, use the environments of MIL-HDBK-217 (Π_E Factors) as a defect density estimation factor.

4.10.2.3.3 Process Maturity and Defects. The maturity of both the product design and the manufacturing process can significantly impact the quantity and type of defects which can reside in the hardware. The data shown in Table 4.4 represent experience on several large development and production projects. As the data illustrate, the proportions of failures in a product which are traceable to design, part or manufacturing causes can differ substantially, depending upon the stage of maturity of the product and the manufacturing process. During the development phase, the major contributor to product failure is design ($\sim 50\%$), while parts may account for 20% of the failures. Unfortunately, design problems can still be present in the product when stress screens are being conducted during production. The proportion of failures in a product, attributable to design, would be expected to decrease as the process matures. As can be noted, part and workmanship problems tend to dominate early and late The overall defect density in the product would also be production. expected to decrease as the process matures. Maturity of the product and process should be taken into account when planning estimates of defect density are being determined in accordance with the Procedure A of Section 5. In such cases, the user may decide to modify some of the incoming part fraction defective values in Tables 5.2 through 5.13, of Procedure A either upward or downward, depending upon past experience and assessments of maturity.

| | Defec | Defect | | | |
|------------------|--------|---------------|-------|----------|--|
| Maturity | Design | Manufacturing | Parts | Density | |
| Development | 40-60 | 20-40 | 10-30 | High | |
| Early Production | 20-40 | 30~50 | 20-40 | Moderate | |
| Late Production | 5-15 | 20-30 | 60-70 | Low | |

Table 4.4 Defect Types & Density vs Process Maturity

Reference 2.2.1

<u>4.10.2.3.4 Packaging Density</u>. Assemblies with high part and wiring density are more likely to contain both patent and latent defects because of the proximity of devices and interconnections contained within a small volume. The effects of poor heat dissipation in densely packaged electronic assemblies can introduce latent defects. Difficulties in initially assembling or reworking the hardware can also make such assemblies more defect prone. Procedure A in Section 5, for estimating defect density, does not directly take into account the packaging density factor. It is recommended however, for those assemblies in an equipment which are judged to have high packaging density, that the tabled values of part fraction defective be increased in accordance with the manufacturer's experience.

<u>4.10.3 Screen Selection and Placement</u> The singular most important task in planning a stress screening program is the selection of appropriate screens and their placement at various levels of assembly so as to achieve a cost effective screening program. Listed below are the factors which affect screen selection and placement. Each of the factors are discussed in more detail in following sections.

a. <u>Goals on Remaining Defect Density</u> - The extent of screening required during the production phase will depend on MTBF requirements and on the goals established for remaining defect density (D_p) .

b. <u>Incoming Defect Density</u> - The quantity and type of defects which reside in the hardware at various assembly levels effects the type and extent of screening required.

c. <u>Screen Effectiveness</u> - Prior knowledge of the effectiveness of the screens in precipitating defects to failure.

d. <u>Test Detection Efficiency</u> - The tests which can be economically and feasibly used to detect defects which have been precipitated to failure by the screens.

e. <u>Thermal and Vibration Response Characteristics</u> - The structural, thermal and material properties of the items to be screened and their response to applied stress.

f. <u>Design Limits</u> - The environmental stress design limits of the items to be screened.

g. <u>Facilities</u> - The screening, test and instrumentation facilities available to the manufacturer to perform screening and test operations.

h. <u>Costs</u> - The costs to achieve screening program goals on remaining defect density.

i. <u>Failure-Free Acceptance Tests (FFAT)</u> - The use of a FFAT as an integral part of a system level screen to verify that goals have been achieved.

<u>4.10.3.1</u> Goals on Remaining Defect Density Equipments having high reliability requirements will have more stringent goals on remaining defect density and consequently increased screening emphasis is needed. Methods for determining goals on remaining defect density are discussed in Appendix B. Achieving low defect densities may require 100% screening at all assembly levels and use of a failure-free acceptance screen/test at the system level to provide assurance that goals have been achieved.

<u>4.10.3.2</u> Incoming Defect Density. A production program which begins with high levels of incoming defect density will require more extensive screening to reduce the defect density to acceptable levels. Every effort should be made to determine realistic estimates of incoming defect density based upon the manufacturers experience and use of the procedures contained in the Guidebook.

4.10.3.3 Screen Effectiveness. Screen effectiveness is characterized as the "screening strength" which is defined as: the probability that a screen will precipitate a defect to a detectable state given that a defect susceptible to the screen stress is present. A basic premise of stress screening is that under specific screening stresses applied over time, the failure rates of defectives are accelerated from that which would occur under normal field operating stress conditions. By subjecting electronic items to accelerated stresses, i.e. rapid temperature cycling and random vibration, latent defects are thus precipitated to early failure. More severe stresses will tend to accelerate failure mechanisms and the rate of defect failure. For example, the failure rate of a latent defect increases with more rapid rates of temperature change and larger temperature extremes. The screening strength of a random vibration screen increases as a function of the level and duration of the applied excitation.

Stress screens are not all equally effective in transforming latent The nature of defects varies with defects into detectable failures. equipment type, manufacturer and time. Screen effectiveness is achieved through proper application of screens which can only be realized through prior experience and experimentation. Stress screens are intended to precipitate latent part and workmanship defects. In a very broad sense, vibration screens are considered to be more effective for workmanship defects and thermal screens are considered more effective for part defects. There are also classes of defects which are responsive to both vibration and thermal excitation. Table 4.5 provides a listing of latent defect types and the screens believed to be effective is precipitating them to failure. Table 4.5 may be used as an aid in the selection of a screen type when prior knowledge on workmanship or part defects for similar assemblies is available.

| Table 4.5 Assembly | | | |
|--------------------|-----|---------------|------|
| by Therma | and | Vibration Scr | eens |

| Defect Type Detected | Thermal Screen | Vibration Screen |
|--------------------------------|----------------|------------------|
| Defective Part | X | X |
| Broken Part | X | X |
| Improperly Installed Part | X | X |
| Solder Connection | X | X |
| PCB etch, Shorts and Opens | X | x |
| Loose contact | | X |
| Wire Insulation | X | |
| Loose wire termination | X | x |
| Improper crimp or mating | X | |
| Contamination | X | |
| Debris | | X |
| Loose hardware | + | X |
| Chafed, pinched wires | | x |
| Parameter drift | X | |
| Hermetic seal failure | X | |
| Adjacent boards/parts shorting |] | X |

Table 4.5 indicates that vibration screens are generally more effective for loose contacts, debris and loose hardware while temperature cycling screens are not effective. Thermal screens are generally more effective for part parameter drift, contamination and improper crimp or mating type defects while vibration screens are not. For other defect classes listed in the table, both thermal and vibration screens are

Realized in the second seco

effective, but the relative degree of effectiveness of one screen type over the other is not precisely known. These are some of the uncertainties which must be dealt with in planning a screening program. Historically, on average, 20% of the defects are found to be responsive to vibration screens and 80% to temperature cycling screens. (Reference 2.2.4).

<u>4.10.3.3.1 Pre/Post Screen Testing and Screen Effectiveness</u>. In order to experimentally determine scress screen effectiveness, the following conditions are required:

a. The items subjected to stress screening must be tested thoroughly before the stress screen to assure that no detectable failures remain at the start of stress screening. When testing is not performed prior to stress screening, it is not known whether patent defects were present, which could have been detected without stress screening or whether latent defects were precipitated by the stress screen.

b. The items subjected to stress screening must be powered and exercised. Performance must be continuously monitored to assure that stress-dependent defects (e.g., intermittents, temperature and timing sensitive faults) are detected.

c. The items subjected to screening must be tested using the same test(s) both before and after the stress screen to assure that the failures detected are a result of the stresses imposed.

d. Data must be collected on defect fallout after the stress screen (i.e., during subsequent stress screens, tests, or early field operation) to obtain an estimate of the number of defects which were initially present.

When such data are available, the screen effectiveness can then be determined by use of the observed fallout from the screen and the number of defects initially present i.e.:

Fallout

Screening Strength =

of Initial Defects

If the screen effectiveness was known precisely then the number of incoming defects could be calculated directly using the observed fallout from the The remaining number of defects would also be known. screen. Such idealized conditions are difficult to realize in practice. We are thus compelled to use a modeling approach where screen effectiveness (strength) is based upon estimates derived from a combination of the actual screening program data, experiments, and the published literature. The screening strength models and values used in the guidebook tables of Procedure B in Section 5, were developed using such an approach. The results and methodology used for these studies are contained in references 2.2.1, 2.2.2. Additional information is also provided in references 2.2.6 and 2.2.9. As more experience data on stress screening are gathered, the screening strength estimates will be refined and improved.

<u>4.10.3.3.2</u> Pre and Post Screen Testing During Production. As was previously discussed, if an item is not tested prior to entering a screen it cannot be determined, even if a detailed failure analyses were performed, whether the defects were precipitated by the screen or whether they were present in the item (patent defects) before the screen. Testing items before they enter the screens and establishing that the items are functioning properly is essential. Evaluation and measurement of the effectiveness of the screens and the overall screening process should be based upon only those defects which are precipitated to failure by the stressed imposed by

the screen. Pre-screen testing should be done immediately prior to the screen to eliminate the uncertainties of latent defect introduction during such processes as cleaning, conformal coating and handling which may otherwise follow the pre-screen test. Relaxing pre-screen test requirements for economic reasons can be detrimental to achieving program objectives. If major changes take place during production such as in an assembly or fabrication process, personne! or production flow, then the defect density (both latent and patent) is likely to change and affect the fallout observed during screening. Under long term production, process improvements and other corrective actions taken as a result of the screening process are likely to change the quantity and distribution of latent defects present in the hardware. Workmanship and manufacturing process defects tend to dominate early production and part related defects dominate mature production. Screens have a different degree of effectiveness for different defect types and therefore screens which may have been effective during early production should be re-evaluated to ascertain their effectiveness. Without the use of pre-screen testing, evaluation and control of the screening process is not possible.

<u>1.10.3.3.3 Screen Parameters</u>. Screening strength and the failure rate of defects are a function of specific screen stresses (parameters) and the time duration of the stress application. Tables 5.14 thru 5.18 in Procedure B of Section 5 provide values for screening strength and defect failure rates as a function of relevant screening parameters. Temperature cycle, constant temperature, random and swept-sine screening parameters are defined as follows:

(A) Thermal Cycle Screen Parameters

1. <u>Maximum Temperature (Tmax)</u> - The maximum temperature to which the screened assembly will be exposed. This should not exceed the lowest of the maximum ratings of all the parts and materials comprising the assembly. Note that nonoperating temperature ratings for parts are higher than operating ratings.

2. <u>Minimum Temperature (Tmin)</u> - The minimum temperature to which the screened item will be exposed. This should not exceed the highest of the minimum ratings of all the parts and materials comprising the assembly.

3. <u>Range (R)</u> - The range is the difference between the maximum and minimum applied external (chamber) temperature (Tmax - Tmin). Temperatures are expressed in $^{\circ}C$.

4. <u>Temperature Rate of Change (T)</u> - This parameter is the average rate of change of the temperature of the item to be screened as it transitions between Tmax and Tmin and is given by:

$$\dot{T} = \left[\left(\frac{Tmax - Tmin}{t_1} \right) + \left(\frac{Tmax - Tmin}{t_2} \right) \right] \div 2$$

Where:

t₁ is the transition time from Tmin to Tmax in minutes

 t_2 is the transition time from Tmax to Tmin in minutes

5. <u>Dwell</u> - Maintaining the chamber temperature constant, once it has reached the maximum (or minimum) temperature, is referred to as dwell. Dwell at the temperature extremes may be required to allow the item being screened to achieve the chamber temperature at the extremes. The duration of the dwell is a function of the thermal mass of the item being screened. For assemblies which have low thermal mass, part case temperatures will track chamber temperatures closely thereby eliminating the need for dwell. Units and systems may have a greater thermal lag and achieving high rates of temperature change may be difficult. Dwells at temperature extremes are required in such instances.

6. <u>Number of Cycles</u> - The number of transitions between temperature extremes (Tmax or Tmin) divided by two.

(B) Constant Temperature Screen Parameters

1. <u>Temperature Delta (Δ T)</u> - The absolute value of the difference between the external (chamber) temperature at which the equipment is being screened and 25^oC.

$$\Delta T = T - 25^{\circ}C$$

Where T is the external chamber temperature

2. <u>Duration</u> - The time period over which the external temperature is applied to the item being screened, in hours.

(C) Vibration Screen Parameters

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1. grms level for Random Vibration - The rms value of the applied power spectral density over the vibration frequency spectrum.

2. <u>g-level for Swept Sine Vibration</u> - The constant acceleration applied to the equipment being screened throughout the frequency range above 40H₂. The g-level below 40H₂ may be less.

3. <u>Duration</u> - The time period over which the vibration excitation is applied to the item being screened, in minutes.

4. Axes of Vibration - This can be a single axis or multiple axes depending on the sensitivity of defects to particular axial inputs.

<u>4.10.3.4 Test Detection Efficiency</u>. Test detection efficiency is a measure of test coverage and is defined as a characteristic of a functional or operational test which is measured by the ratio of failure modes detectable by the test to the total number of possible failure modes. While stress screens may be effective in transforming a defect into a detectable failure, removal of the failed condition is dependent on the capability of the test used to detect and localize the failure.

Modern electronic equipment comprised of microprocessors, large memory and LS1 devices may contain defects so subtle that only the most thorough of tests can detect them. Printed wiring assemblies (PWA) have also become much more complex with associated higher defect densities. The costs of PWA fault isolation and repair at end item test and during field use can be 10-100 times greater than at the PWA level. Stress screening and testing at the PWA level even perhaps at the bare board level, thus becomes more cost effective. Investments in test equipment and in developing thorough tests with high test detection efficiency also becomes practical from an economic standpoint.

Care should be taken to ensure that tests have detection efficiencies as high as is technically and economically achievable. The screens may otherwise precipitate defects to failure which may go undetected by post screen tests. Effective screening at lower levels of casembly may not always be easily accomplished because of low detection efficiency. The difficulty in accurately simulating functional interfaces or the inability to establish meaningful acceptance criteria may make the development of tests with high detection efficiency at the assembly level very costly. A certain percentage of defects may <u>only</u> be detectable at the unit/system level when all or a majority of the system components are connected and operating as a system. Analysis and quantification of test detection efficiencies should be an integral part of the planning for a screening program.

<u>4.10.3.4.1</u> Determining Test Detection Efficiency. On some system procurements the probability of detection is a specified parameter for builtin-test (BIT), performance monitoring (PM) and fault location (FL) capability requirements. When the required BIT or PM/FL capability is used to verify performance of an item being screened, the specified values of detection efficiency should be used in developing the screening plan. On other system procurements, requirements to perform a failure modes and effects analysis (FMEA) are specified in the contract. In such cases, the FMEA should be used to estimate the fraction of failure modes detectable for a given test design.

When FMEA or BIT fault detection requirements are not specified in the contract, estimates of test detection efficiency should be made based upon experience data. The data should be gathered from fixed test positions and analyzed by test engineering personnel. Table 4.6 provides values of test detection efficiency for various tests which may be applied with stress screens. The values in the table were derived by production and engineering test personnel from a large DOD electronic system manufacturer. (Reference 2.2.1).

| Level Assembly | Test Type | Detection Efficiency |
|-------------------|-------------------------------------|-------------------------|
| | Production Line GO-NO GO Test | 0.85 |
| Assembly | Production Line In-Circuit Test | 0.90 |
| • | High Performance Automatic Tester | 0.95 |
| | Performance Verification Test (PVT) | 0.90 |
| Unit | Factory Checkout | 0.95 |
| • • • | Final Acceptance Test | 0.98 |
| | On-Line Performance Monitoring Test | 0.90 |
| System | Factory Checkout Test | 0.95 |
| - | Customer Final Acceptance Test | 0.99 |

TABLE 4.6 DETECTION EFFICIENCY vs TEST TYPES

| Table 4.7 | provides | fault | coverage | estimates | for | various | automatic | test |
|------------|------------|--------|----------|-------------|------|---------|-------------|------|
| systems us | ed by elec | tronic | s system | manufacture | ers. | (Refere | nce 2.2.8.) |) |

| | Automatic Test System Type | | | | | | | |
|--------------|--|---------------------------------|-------------------------------|---------------------------------------|--|--|--|--|
| Circuit Type | Loaded Board Shorts Tester (LBS) | In-Circuit Analyzer (ICA) | In-Circuit Tester (ICT) | , functional Board Tester (FBT) | | | | |
| Digital | 45% to 65% | 50% to 75% | 85% to 94% | 90% to 98% | | | | |
| Analog | 35% to 55% | 70% to 92% | 90% to 96% | 80% to 90% | | | | |
| Hybrid | 40% to 60% | 60% to 90% | 87% to 94% | 83% to 95% | | | | |

TABLE 4.7. FAULT COVERAGE FOR AUTOMATIC TEST SYSTEMS

*Reference 2.2.8

An illustration of fault coverage for a sample of 1000 PWA's subjected to various test strategies is also provided in Reference 2.2.8. The strategies employed include the use of each of four automatic testers independently and in combination. Table 4.8 provides a summary of the results.

| Fault Classification | Actual | LBS | ICA | ICT | FBT | ICA-ICT | ICA-FBT | ICT-FBT | IC 1C F0 |
|----------------------------------|--------|-----|-----|-----|-----|---------|---------|---------|----------------|
| Shorts | 261 | 261 | 261 | 261 | 261 | 261 | 261 | 261 | 26 |
| Opens Missing | 5 | 5 | 5 | 5 | 5 | 5 | 5 | 5 | |
| Components Wrong | 30 | | 25 | 28 | 25 | 29 | 27 | 29 | 3 |
| Components | 67 | | 53 | 61 | 55 | 64 | 59 | 60 | 6 |
| Reversed Components | 28 | | 26 | 23 | 25 | 27 | 28 | 25 | 2 |
| Bent Leads Analog | 43 | | 38 | 43 | 43 | 43 | 43 | 43 | |
| Specifications | 25 | | 13 | 21 | 18 | 21 | 21 | 22 | â |
| Digital Logic | 27 | | | 20 | 27 | 20 | 27 | 27 | 2 |
| Performance | 26 | | | | 26 | | 26 | 26 | ; |
| Total No. of Faults | 512 | 266 | 421 | 462 | 486 | 470 | 497 | 498 | 50 |
| | | | | | ÷ | | | | |
| Fault Coverage Fault Coverage | 100% | 52% | 82% | 90% | 95% | 92% | 97% | 97% | 99 |
| Increase | - | - | - | - | • | 2.2% | 2.3% | 2.5% | 4 |
| Rejected PCBs | 398 | 223 | 345 | 370 | 385 | 374 | 391 | 393 | 39 |
| Rework Yield Undetected | | 195 | 316 | 354 | 376 | 361 | 384 | 388 | 39 |
| Faulty PCB | | 203 | 82 | 44 | 22 | 37 | 14 | 10 | |
| Rework Yield Rework Yield | | 49% | 79% | 89% | 94% | 91% | 96% | 97% | 99 |
| Increase | - | - | - | - | - | 2% | 2.1% | 3.2% | 4 |
| Finished Units | | 805 | 918 | 956 | 978 | 963 | 986 | 990 | 99 |

TABLE 4.8 FAULT DETECTION FOR A 1000 PCB LOT SIZE*

*Reference 2.2.8

As can be noted from the table, using only a Functional Board Tester (FBT) provides 95% fault coverage but combining an In-Circuit Tester (ICT) with the FBT increases coverage to 97% and adding an In-Circuit Analyzer (ICA) to the sequence, increases coverage to 99%.

The faults detected are typical patent defects and do not cover the spectrum of defect types of interest in stress screening. The statistics provided in the table, however, provide a basis for developing estimates of test detection efficiency when a stress screening program is being planned. The data should also be helpful in selecting test strategies for use with stress screens.

4.10.3.4.2 Power-On Testing vs Power-Off. Application of power, exercising and monitoring equipment performance continuously during the screen will greatly enhance test detection efficiency. Subtle faults, such as contact intermittents or temperature sensitive parts, can only be detected with powered and monitored screens. With the increased complexity of modern electronics, fault sites may be confined to smaller areas and fault symptoms may appear only during certain tests or under a special set of external conditions. As a result, a greater incidence of "Cannot Duplicate" (CND), "No-Fault Found" (NFF) and "Retest OK" (RTOK) and similar intermittent or transient phenomena can occur. Latent defects which are precipitated to failure by stress screens can be categorized into three general types:

- <u>Type 1</u> Physical defects that are readily transformed from an inherent weakness to a hard failure by the stress screen.
- <u>Type 2</u> Physical defects that manifest as failures only while under thermal or mechanical stress. (e.g. intermittent caused by a cold solder joint)
- <u>Type 3</u> Functional defects that manifest as performance failures or anomolies only while under thermal or mechanical stress. (e.g. timing problems)

The type 1 defects are readily detected by post screen tests of sufficient thoroughness. Type 2 and Type 3 defects require thorough and continuously monitored tests so that they can be detected. Type 3 defects, which include problems such as timing, part parameter drift with temperature or tolerance build-up can only be detected with powered and monitored tests. Type 2 and Type 3 defects can comprise 50% and as much as 80% of the latent defects present in the hardware. (Reference 2.2.2)

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Developing tests and test strategies for use with stress screens and estimating their detection efficiency is a vitally important activity in planning a stress screening program. The use of tests with high detection efficiency is of equal importance to using effective screens in structuring a screening program for production.

4.10.3.5 Thermal and Vibration Response Characteristics. All assembled hardware consists of many paths along which a stress might be transmitted. The selection of screening parameters and methods of stress application must be suited to the stress transmission characteristics of the hardware design. As a part of the screen selection and placement process, in which thermal or vibration screens are to be used, a stress response survey of the item to be screened should be performed. Care should be exercised to ensure that hardware responses are large enough to generate an effective screen while not exceeding hardware design capability. Environmental stresses should be applied to the hardware and the response of critical hardware elements measured to determine whether maximum or minimum temperature limits are being exceeded, and whether suspected defect sites (parts, interconnections etc.) are responsive to the screen stress. In addition, normal design provisions for isolating the hardware from stress such as the use of shock mounting, vibration isolators or cooling air should also be evaluated. Application of environmental stress screening in such instances, should require bypassing the normal stress isolation provisions or may dictate the need for screening at lower assembly levels which do not include the stress isolation design features.

4.10.3.6 Design Limits. The use of screen parameters which impose stresses which exceed the design limits of the product is not recommended. Effective screening programs can be developed without having to resort to stresses which exceed the design capability of the hardware. Criteria for judging how much the design limits can be safely exceeded, without causing damage to the product, are non-existent or at least arbitrary. The impetus for exceeding the design limits is basically economic in nature because harsher screens tend to take less time to precipitate defects to failure. Using the procedures contained in the Guidebook, the manufacturer can focus on those items in which defects are most likely to reside and determine safe screening levels, within appropriate cost constraints, for precipitating them to failure.

4.10.3.7 Facilities and Costs. The facilities that the manufacturer has available for screening, instrumenting and testing the product affects screen selection and placement. A manufacturer may not have random vibration facilities or automatic test systems which can be used for the stress screening program. In such cases, the manufacturer may decide to impose less severe stresses for a longer duration or decide to use less expensive alternatives such as described in NAVMAT P-9492. The costs to purchase expensive screening or test equipment and perform screens at a given level of assembly may not be warranted, in terms of the number of defects which are likely to be found. The screening must be addressed in preparing the screening program plan and in the screen selection and placement process. Costs versus the benefits to be derived from screening should be addressed.

The criterion used in the guidebook to both limit costs and judge the cost effectiveness of the screening program is called the cost threshold. The cost threshold is based upon the average cost of repair in the field and can be viewed as a "not-to-exceed" cost. After determining the costs

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of conducting the screening program and estimating the expected fallout in accordance with the Procedures B & C, the manufacturer should compare the cost per defect eliminated against the cost threshold. If the cost per defect eliminated is found to be higher than the cost threshold, then the manufacturer should determine alternative methods which lower the costs of finding and eliminating the defects to a value which is less than or equal to the cost threshold. Alternatives might include reducing the incoming defect density by means other than assembly screening, (e.g., increase the quality level of parts used) increase the screening strength at lower assembly levels, or eliminate screens which may be of questionable value. In those cases where field reliability is an overriding requirement, the Government procuring activity must decide to what extent the cost threshold should be exceeded.

4.10.3.8 Failure-Free Acceptance Screen/Test. The use of failure-free periods or cycles, as a part of a stress screen, is intended to provide some degree of assurance for the user that screening is complete. failure-free period is a time interval during which the equipment must operate without failure while exposed to environmental stress. Arbitrary selection of failure-free periods does not provide any quantitative assurance that the remaining defect density goals have been achieved. Prior knowledge of defect density, the effectiveness of the screens to be used, and a quantitative goal for the remaining defect density must be available in order to establish failure-free acceptance test requirements. The quantity of primary interest is the average number of defects remaining (defect density) per equipment at delivery. Yield, which is directly related to remaining defect density, can be verified by conducting a failure-free screen/test for a predetermined period of time. The length of the failure-free period is dependent on the yield requirement or goal, the degree and type of stresses applied during the failure-free period and the statistical confidence needed to provide assurance that the yield goal has been achieved.

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The failure-free acceptance test can be used as an integral part of the system level screen or as part of a formal acceptance test for the system when a stress screen is not used at the system level. When a failure-free acceptance test is used, each system offered for acceptance must be subjected to the failure-free screen and test. Passing the test involves contiguous operation of the equipment for a time T, without failure while under screening stress. If a failure occurs, the failure is repaired and the equipment is again subjected to the same failure-free period starting at T=0. Appendix C of the Guidebook provides the mathematical derivation of the FFAT methods contained in the Guidebook. Procedure C in Section 5 contains the detailed procedures for tailoring a FFAT to program requirements.

4.10.4 Preparation of ESS Plans. The contractor should prepare ESS plans for both the development and production phases. The purpose of the development phase plan is to describe the proposed application of ESS during development and production. The development phase plan should be submitted as part of the Reliability Program Plan. A detailed ESS plan should be submitted for approval by the procuring activity prior to production.

<u>4.10.4.1</u> Development Phase Plan - The development phase plan should include the following:

a. identification of the reliability requirements for the product and the quantitative goals for the ESS program.

b. identification of the equipment to be screened and the respective production quantities.

c. description of the initial screens which will be applied and the screening experiments which will be conducted.

d. description of the data collection and analysis program which will be used.

e. description of subcontractor and supplier stress screening to be performed.

f. results of preliminary use of the Guidebook procedures.

g. identification of the organization elements that will be responsible for ESS planning and experimentation, and the conduct of development phase screening activity.

<u>4.10.4.2 Production Phase Plan</u>. The production phase plan shall include the following:

a. Quantitative objectives of the ESS program.

b. Detailed breakdown to the assembly level of the equipment which will be screened.

c. Description of the screens which will be applied, including screen parameters and exposure time.

d. Description of the results in applying Procedures A, B, C and D of the Guidebook including the rationale for achieving quantitative objectives in a cost effective manner.

e. Description of the FRACAS to be used and the analyses procedures which will be used to evaluate and control the screening process.

f. Description of the Failure-Free Acceptance Test to be performed for each system to verify achievement of objectives.

g. Identification of the organizational elements responsible for conducting and evaluating the effectiveness of the production ESS program.

<u>4.10.5 Guidelines for Initial Screen Selection and Placement</u>. An initial screening regimen should be selected for experimental use during the development phase in conjunction with the use of the Guidebook procedures. Table 4.9 is recommended as an aid in selecting and placing screens for a starting regimen.

TABLE 4.9 GUIDELINES FOR INITIAL SCREEN SELECTION AND PLACEMENT

| | | Select | 101 | | Placement | | | | | |
|-------------------------|--|-----------------|--|--------------|--|--|---|--|--|--|
| Level of Assembly | Temp Cycle | Const. Temp. | Rand Vib. | S.S. Vib. | | Advantages | | Disadvantages | | |
| Assy | CycleTemp.VID.VID.E1M2M3NE = EffectiveMarginally EffectiveM = Marginally EffectiveWotes:1. Particularly if power1s applied and perfor-mance is monitored attemperature extremes.2. Effective whereassemblies contain com-plex devices (RAMs, micro-processors, hybrids)3. Effectiveness highlydependent on assemblystructure.Not effectivefor small, stiff PWAs. | | Cost per flaw pre- cipitated is lowest (unpowered screens) Small size permits batch screening Low thermal mass allows high rates of temperature change Temperature range greater than oper- ating range | | Test detection efficiency is rel- atively low Test equipment cost for powered screens is high | | | | | |
| Unit | E | M | E | R | • | Relatively easy to power and monitor performance during screen Higher test detec- tion efficiency than assembly level Assembly inter- connections (e.g., wiring backplane) are screened | • | cludes high rates of change, or requires costly facilities Cost per flaw significantly higher than assembly level | | |
| System | Ε | H | ε | M | | All potential sources of flaws are screened Unit interopera- bility flaws detected High test detec- tion efficiency | • | costly to test at temperature extremes | | |

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<u>4.11 Production Phase - Monitoring, Evaluation and Control</u>. Once a screening program is implemented during the production phase, the screen fallout data and the screening process must be monitored and controlled to assure that program objectives are achieved. Use of a Failure Reporting Analysis, and Corrective Action System (FRACAS) should be an integral part of production phase monitoring and control tasks. The fallout from the screening process provides the necessary visibility regarding the sources of defects in the product and the manufacturing process. Finding defects, determining their root causes and ensuring that the sources of the defects are eliminated from either the process or product, is the basic mechanism by which process capability is improved.

Analyses of screen fallout data must be performed with specific objectives in mind. Well-defined monitoring, evaluation and control task objectives will ensure that the proper data is collected, classified and correctly analyzed to meet objectives. The objectives of the monitoringevaluation and control tasks are to establish assurance that remaining defect density and reliability goals are achieved through implementing improvements in manufacturing, screening and test process capability. Manufacturing process capability is improved through taking corrective actions which reduce the number of defects that are introduced into the Screening process capability is improved by increasing the product. stress or duration of a screen and ensuring that potential sites for defects in the product are being adequately stimulated by the screen. Testing process capability is improved by increasing test detection efficiencies when it is found that latent defects, precipitated to failure by a lower level screen, are escaping and being detected by tests at upper assembly levels.

Another goal of monitoring and control tasks is related to cost effectiveness. The initial screening program might have been based upon planning estimates which were overly pessimistic. Corrective actions might also have been taken during production to reduce the number of defects introduced into the product. In either case, if the screening

program is continued as planned, more screening than is necessary results, which impacts both cost and schedule. Decisions must be made to either reduce the screening regimen, resort to screening on a sample basis or to completely eliminate the screen. In a sense, the goal of monitoring and control tasks is to make the screening program unnecessary.

<u>4.11.1 Data Collection</u>. The importance of timely and accurate data collection to achieving screening program objectives cannot be overemphasized. The data elements listed below should be collected during the conduct of the screening program. Some of the data elements become available directly as observed events from the screening process. Other data elements will become available only after analysis or the failures and failure data, or after a batch of items have been exposed to screening.

a. Identification of the items exposed to the screen/test.

b. Number of like items exposed to the screen/test.

c. Number of like items passed/failed the screen/test.

d. Description of the type of defect found (part, workmanship/process, design)

e. Type and number of defects found in conjunction with the number of items exposed, passed/failed (data elements b, c, d).

f. Identification of the part, interconnection site where the defect was found.

g. Identification of the assembly level or manufacturing process operation where the defect was introduced.

h. Screen conditions under which the defect was found (e.g., high temperature, vertical axis of vibration etc.).

i. Time-to-failure relative to the start of the screen.

j. Failure analysis results which identifies the root cause of the defect.

h. Corrective action taken to eliminate the cause of the defect from the product and/or process.

<u>4.11.2 Failure Classification</u>. In order to establish a basis for the analysis of the screening fallout data, the failures must be properly classified. The following classification scheme is recommended.

a. <u>Part Defect</u> - A failure or malfunction which is attributable to a basic weakness or flaw in a part (diode, transistor, microcircuit, etc.).

b. <u>Manufacturing Defect</u> - A failure or malfunction attributable to workmanship or to the manufacturing process (cold solder joint, cracked etch, broken wire strands, etc.).

c. <u>Design Failure</u> - A failure or malfunction attributable to a design deficiency. Note that electrical or thermal overstress failures due to inadequate derating, are design problems. One would expect that all or most design problems would have been eliminated from the hardware prior to production. Nonetheless, a substantial proportion of failures during early production (\sim 30%) are found to be traceable to design.

d. <u>Externally Induced Failures</u> - A failure attributable to external influences such as prime power disturbances, test equipment, instrumentation malfunctions or test personnel.

e. <u>Dependent Failure</u> - A failure which is caused by the failure of another associated item which failed independently.

- f. <u>Software Failure</u> A failure attributable to an error in a computer program.
- g. <u>Unknown Cause Failure</u> An independent failure which requires repair and rework but which cannot be classified into any of the above categories.

4.11.3 Preliminary Analysis of Fallout Data. A preliminary analysis of the fallout data should be performed to insure that failure causes are properly established and to categorize the failures so that more detailed analysis related to the program objectives can be performed.

a. <u>Part and Interconnection Defects</u> - All failures classified as part, board and interconnection defects, which are precipitated and detected by a screen/test, should be considered to be latent defects provided that pre-screen testing was performed. These data should be used for monitoring and control purposes.

b. <u>Design Failures</u> - A predominance of design problems which is discovered during production screening operations is a matter of serious concern. Every effort should be made to determine corrective actions for design problems very early in production. It does no good to speculate that the design problems should have been eliminated from the hardware during the development stage. Stress screening, on a 100% basis, is an expensive and time consuming method for finding design problems. If the fallout from screening indicates persistent evidence of design problems, methods other than 100% stress screening should be used. Reliability growth and Test-Analyze-And-Fix (TAAF) techniques are recommended.

c. <u>Unknown Cause Failures</u> - Special attention should be given to unknown cause failures. Sufficient investigation should be made to establish that an intermittent condition does not exist. The number of failures classified as "Unknown Cause" should be kept to a minimum. Every effort should be made to correlate the failure circumstance data with the other similar failure incidents, as will as to use failure analysis so as to establish the cause of failure. d. <u>Dependent and Induced Failures</u> - Analyses of dependent and induced failures should be performed to determine necessary corrective actions.

<u>4.11.4 Analysis of Screen Fallout Data</u>. The analysis of screening fallout data is directed toward evaluating the screening process so as to achieve screening program goals on remaining defect density D_R . Yield goals are achieved by both improving manufacturing process capability through corrective action and by improving the screening and test process capability when it is found to be needed.

Manufacturing, screening and test process capability will determine the remaining defect density. The capability of these processes is measured and controlled by use of two important quantities, the incoming defect density (D_{IN}) and the test strength (TS). Neither one of these quantities is directly observable as a result of the screening process. The only observable statistic is the fallout from the screen/test, from which inferences regarding D_{IN} and TS must be drawn. The basic approach used in the Procedure E of Section 5, is to obtain estimates of D_{IN} and TS, using the screen fallout data and to statistically compare the observed data against the planning estimates. Based upon the comparisons, corrective actions are determined to eliminate the source of the defect from the process and/or to change the screens so as to achieve stated objectives.

Four complementary procedures are presented in Procedure E for performing monitoring and analyses tasks. Procedures El and E4 use Quality Control Charts and control intervals for monitoring and control. Procedures E2 and E3 use maximum likelihood and graphical techniques, applied to the Chance Defective Exponential model, to estimate $D_{\rm IN}$ and TS.

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<u>4.11.4.1 Quality Control Charts</u>. The use of control charts for defect control is a standard quality assurance technique. Control charts are used in Procedure El which are based upon the Poisson Probability distribution; i.e.,

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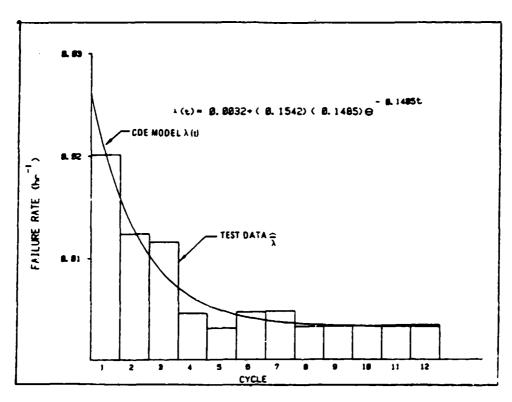
$$P(\chi=x) = \frac{e^{-D} D^{\chi}}{x!}$$

| Where: | D | 2 | defect density |
|--------|-----|---|-------------------------------------|
| | x | 3 | number of defects in an item |
| P(X- | =x) | = | probability of x defects in an item |

The mean of the Poisson distribution is D and the standard deviation is \overline{D} . 99% control limits on the control charts are established by $D \stackrel{+}{=} 3\sqrt{D}$. The primary purpose of the control chart technique is to establish baselines against which the process can be monitored and by which out-of-control conditions can be identified. Part fraction defective and defect density are calculated, using the fallout data, and compared against the control chart baselines. Part and workmanship (process) problems are rank ordered and corrective actions are required which eliminate the source of the defects from the product. Procedure El of Section 5 contains the detailed methodology for implementing the control chart technique.

<u>4.11.4.2 Use of the CDE Model to Evaluate Screening Results</u>. The Chance Defective Exponential (CDE) model was developed by Fertig and Muthy and is discussed in a paper contained in the 1978 Annual R&M Symposium (Reference 2.2.7). Appendix A of the guidebook, provides a description of the CDE model. The failure rate function of the CDE model can be fitted to the observed fallout data for a given screen so as to obtain estimates of the model parameters. The parameters of the CDE model provide estimates of the incoming defect density $D_{\rm IN}$, the screening strength (SS) and the failure

rate of the "good" part population for an equipment. Figure 4.4 is an extract from study report (Reference 2.2.6) which shows a histogram of the screen fallout from a 12 cycle -54° C to 71° C temperature cycle screen. The fallout per cycle is used to obtain maximum likelihood estimate (MLE) for the parameters of the CDE model.



Reference 2.2.6

FIGURE 4.4 TEMPERATURE CYCLING DATA FITTED TO THE CHANCE DEFECTIVE EXPONENTIAL MODEL

As the figure shows, the CDE model parameters estimated by the MLE procedure, are: incoming defect density (D_{IN}) equal to .1542 defects per item, the failure rate of a defect $(\overline{\lambda}_D)$ equal to .1485 failures per hour (which corresponds to a screening strength of .95) and a value of .0032 for

the failure rate of the main population (λ_0) . The MLE estimates of the model parameters should be compared against the planning estimates of D_{IN} and SS to determine appropriate corrective action. The parameter estimation procedure should be applied to several batches of screened items, and/or confidence limits should be calculated for the MLE parameters to verify that significant differences from planning estimates exist. Caution should be exercised in interpreting the MLE estimates of the CDE model parameters. In most instances, the time duration of a screen/test is insufficient to obtain any precision in the estimate of λ_0 , the failure rate of the "nondefective" population. It is therefore recommended, as a first step, that λ_0 be set to zero, or that a prior estimate of λ_0 be used. Prior estimates of λ_0 can be obtained from development phase reliability tasks, i.e., from a MIL-HDBK-217 prediction or from the results of a MIL-STD-781 demonstration test.

4.11.4.3 Comparing Observed and Planning Estimates of D_{TN} and TS. In practice the "true" values of incoming defect density and test strength can differ significantly from planning estimates. When significant difference exist, both the outgoing defect density and costs are effected. Under certain conditions, differences from planning estimates will jeopardize achieving goals on remaining defect density, whereas in other cases, the differences will have more of an impact on costs. The corrective action required to assure achievement of screening program goals will differ, depending upon the degree of departure from planned values and whether D_{IN}, TS or both are higher or lower than planning estimates. It must also be recognized that, given effective corrective actions, good process control and the removal of defects from the product, D_{TN} would be expected to decrease rather rapidly during the initial stages of screening. It is, therefore, necessary to establish monitoring schedules and lot sampling techniques, which correlate with major corrective action charges so that reductions in D_{IN} and the effectiveness of corrective actions can be measured.

5. Detailed Guidelines

5.1 ESS Planning, Monitoring and Control Procedures. Detailed procedures and methodologies for performing the major tasks involved in planning, monitoring and controlling the screening program are contained in the following sections. There are five basic procedures.

• <u>Procedure A entitled</u>, "Planning Estimates of Incoming Defect <u>Density</u>" is used to obtain initial estimates of the parameter D_{IN} . The procedure contains tabled values of part, board and connection fraction defective as a function of part quality level and field environmental stress. Other factors which impact incoming defect density, such as maturity and packaging density, should be factored into the estimates based upon experience and the recommendations contained in the guidebook.

• <u>Procedure B entitled</u>, "Screen Selection and Placement" uses the results obtained from Procedure A, to plan a screening program to achieve objectives on remaining defect density. The procedure contains tabled values of screening strength and defect failure rates as a function of the screen parameters and duration. Other factors which affect screen selection and placement, such as the quantity of defect type susceptible to temperature vs vibration screens, must be factored into the procedure based upon the manufacturer's experience and the recommendations contained in the guideline. Procedure B must be performed in conjunction with the following two procedures C and D, to develop a screening plan.

• <u>Procedure C entitled</u>, "Failure-Free Acceptance Tests" is used to establish failure-free acceptance periods which provide a lower confidence bound on yield or equivalently, the remaining defect density. The failurefree acceptance test can be made a part of the end item (system) level screen or used as part of a separate acceptance test procedure. In either case, the costs of conducting the FFAT must be factored into the screen selection and placement and cost estimating procedures.

• <u>Procedure D entitled</u>, "<u>Cost Effectiveness Analyses</u>" is used to estimate and compare the costs of various screen selection and placement alternatives in order to arrive at a cost effective screening program. The manufacturer's cost of conducting the screening program is normalized to a cost per defect eliminated. Comparison of the cost per defect eliminated by the screening program against a cost threshold value is used to determine cost effectiveness.

• <u>Procedure E entitled, "Monitoring, Evaluation and Control"</u> is used to obtain estimates of the defect density based upon the observed screen fallout data and to establish whether the observed defect density falls within or outside of predetermined control limits. Comparisons of observed part fraction defective and defect density are made against baseline criteria to prioritize and determine the need for corrective actions which improve manufacturing or screening process capability.

5.2 Procedure A - Planning Estimates of Incoming Defect Density

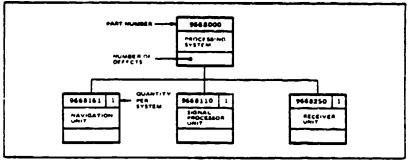
5.2.1 Objectives. The primary objective of this procedure is to obtain planning estimates of incoming defect density which will serve as a basis for planning a stress screening program. Another objective is to estimate the proportion of the incoming defect density which may be sensitive only to vibration or to temperature screens.

5.2.2 Methodology. The methodology is similar to the procedures used in MIL-HDBK-217 for estimating failure rates. Tables 5.2 through 5.13 are used in the procedure to obtain incoming defect density estimates as a function of the number of parts, boards and connections contained in the product, their quality level and the field stress environment to which the Other factors which may affect estimates of parts will be exposed. incoming defect density, such as the product or process maturity, packaging density or prior experience should be used, as may be appropriate, to tailor the estimate to the unique characteristics of a given product and process. Estimates can be scaled unward or downward when prior knowledge or experience data on specific part types or manufacturing processes are available. The proportion of incoming defect density which is responsive to either vibration or temperature screens should also be estimated. Historical data has shown that approximately 20% of the defects in a production lot are sensitive only to vibration type screens and 80% to temperature screens (Reference 2.2.5). Each situation, however, must be The defect density estimates obtained by this judged individually. procedure should be viewed as being representative of the user's (manufacturer's) average process capability. It should be recognized that the estimates obtained by this procedure are planning estimates only which are required for establishing a baseline screening program. Comparison of the planned estimates of defect density against observed values using Procedure E, is the vehicle by which defect density is controlled and the screening program objectives and production reliability assurance are achieved.

The procedure uses a three-level equipment breakdown structure, i.e. System, Unit and Assembly, to illustrate the methodology for planning a stress screening program. Other equipment breakdown structures are, of

course, possible and can be adapted to the structure used herein. Stress screening, excluding part level screening, is generally confined to three levels. However, if more levels are used, the methodology is equally applicable, requiring only the expansion of the three-level-worksheets.

5.2.2.1 Equipment Breakdown. The equipment to be screened should be depicted in chart form down to the assembly level as illustrated in Figures 5.1 and 5.2. Figure 5.1 shows the breakdown of a system to be screened into three units. Figure 5.2 shows the breakdown of one of the units into its constituent assemblies.



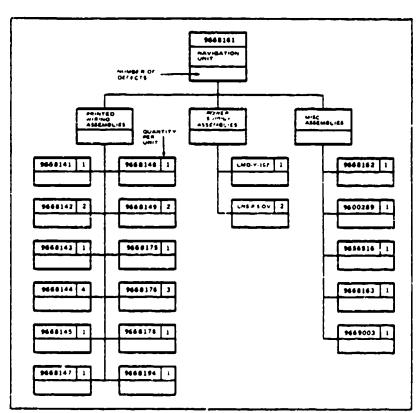


Figure 5.1 System Breakdown Chart

Figure 5.2 Unit Breakdown Chart

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5.2.3 Procedure Steps. Using the equipment breakdown charts and the defect estimation worksheets (Figure 5.3) the following steps should be performed.

Step 1 Assembly Defect Estimates. For each assembly identified in the equipment breakdown, as in Figure 5.2., a defect estimation worksheet as shown in Figure 5.3 should be completed.

| Program/Project | | System Nomenclature | | | | | | |
|--------------------|--------------|---------------------|--------------------------|------------|-----------------------|----------------------|--|--|
| Unit Identifier | Assembly Id | dentifier | Pr | repared By | | ate | | |
| Par | t Type | 1 | uality evel/ arade | Quantity | Fraction Defective | Estimated Defects | | |
| Microelectron | ic Devices | | | | | | | |
| Transistors | | | | | | | | |
| Diodes | | | | | | | | |
| Resistors | | | | | | | | |
| Capacitors | | | | | | | | |
| Inductive Dev | ices | | | | i | | | |
| Rotating Devi | ces | | | | | | | |
| Relays | | | | | | | | |
| Switches | | | | | | | | |
| Connectors | | | | | | | | |
| Printed Wirin | g Boards | | | | | | | |
| | | | | | | | | |
| Connections, | | | | | | | | |
| Connections, | | | | | | | | |
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| Connections, | Reflow Solde | er | | | | | | |

DEFECT ESTIMATION WORKSHEET

Defect Density Total

Figure 5.3 Worksheel for Estimating Defect Density

<u>Step 2 Part Type</u>. Determine the part types used in the item. Part types shown on the worksheet are the standard types included in MIL-HDBK-217. Miscellaneous part types can be added as necessary.

<u>Step 3 Quality Level/Grade</u>. Enter the appropriate quality level or grade for the part types as indicated by Table 5.1.

| Equivalent Quality Levels* | | | | | | | |
|----------------------------|---------------|--|-------------|--|--|--|--|
| Quality | | | Passive | | | | |
| Grade | Microcircuits | Semiconductors | Parts | | | | |
| 0 | S | JAKS | Ť | | | | |
| 1 | в | JANTXV | S | | | | |
| 2 | 8-0 | JANTX | R | | | | |
| 3 | 8-1 | 1 JAN, 2 JANTX** | P | | | | |
| 4 | B-? | 2 JAN, 1 JANTX** | 3 M, 2 P ** | | | | |
| 5 | с | JAN | м | | | | |
| 6 | C-1 | $\frac{1}{2}$ JAN, $\frac{1}{2}$ LOWER** | 1 L, 1 M ** | | | | |
| 7 | D | LOWER | ι | | | | |
| 8 | D-1 | PLASTIC | COMMERCIAL | | | | |

as defined in MIL-HDBK-217.

** Mixture of quality levels to obtain quality grade

Step 4 Quantity. Enter the quantity of each part and connection type.

<u>Step 5 Fraction Defective</u>. Determine the fraction defective in parts per million (PPM) for each part, connection, board and connector type using Tables 5.2 through 5.13. The field environment under which the equipment is intended to operate must be known.

<u>Step 6 Estimated Defects</u>. Determine the estimated defects by multiplying the Quantity in Step 4 by the Fraction Defective in Step 5 and enter on the worksheet.

<u>Step 7 Defect Density</u>. Enter the Defect Density for the Assembly by adding all the estimated defects for all the parts in the assembly. Enter the total Defect Density by multiplying the assembly Defect Density by the number of identical assemblies contained in the equipment.

<u>Step 8 Unit Defect Estimates</u>. For each unit identified in the System Breakdown Chart, a Unit Breakdown chart as shown in Figure 5.2 should be prepared. A Defect Estimation Worksheet should be completed for each unit, as was done for the assemblies, including only those parts and interconnections that were <u>not</u> included in the assemblies. Determine the estimated number of defects for each unit by summing the estimated defects for all the assemblies comprising the unit and the estimated unit flaws. Note that the quantity of identical assemblies or units in the system must be used in calculating defect density. Enter the totals on the Equipment Breakdown Chart in the spaces provided.

<u>Step 9 System Defect Estimates</u>. A Defect Estimation Worksheet should be completed for the system to estimate the number of defects not included in the Unit or Assembly level estimates. Determine the total estimated number of defects in the System by summing the Unit Defect Estimates and the quantity from the System Defect Estimates. This total is the incoming defect density for the system which is used as the planning estimate D_{IN} .

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<u>Step 10 Total Defects Production Lot</u>. The total defects for the production lot should be calculated by multiplying the system defect density obtained in Step 9 by the number of systems to be produced.

5.2.4 Part Fraction Defective Tables. Twelve tables are provided which contain the part fraction defective as a function of the part quality level and the field stress environment to which the equipment will be exposed. Part types included in the tables are:

- Microelectronic Devices
- Transistors
- Diodes

- Resistors
- Capacitors
- Inductive Devices

- Rotating Devices
- elays Relays
- Switches
- Connections
- Connectors
- Printed Wiring Boards

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| | Quality Level | | | | | | | | |
|------------------|---------------|--------|-------------|--------|---------|---------|---------|---------|--------|
| Environ- ment | S | 8 | B- 0 | B-1 | 8-2 | C | C-1 | 0 | 0-1 |
| 68 | 9.2 | 18.3 | 36.6 | 54.9 | 119.0 | 146.4 | 237.9 | 320.3 | 640.6 |
| GF | 19.4 | 38.7 | 77.4 | 116.1 | 251.6 | 309.6 | 503.2 | 677.3 | 1354.6 |
| G/4 | 27.5 | 55.1 | 110.1 | 165.2 | 351.9 | 440.5 | 715.8 | 963.6 | 1927.2 |
| MP . | 25.6 | 51.2 | 102.4 | 153.6 | 332.9 | 409.7 | 665.8 | 896.3 | 1792.5 |
| NSB | 26.6 | 53.1 | 106.3 | 159.4 | 345.4 | 425.1 | 690.8 | 929.9 | 1859.9 |
| NS | 26.6 | 53.1 | 106.3 | 159.4 | 345.4 | 425.1 | 690.8 | 929.9 | 1859.9 |
| NU | 34.7 | 69.5 | 139.0 | 208.5 | 451.7 | 556.0 | 903.5 | 1216.2 | 2432.5 |
| NH | 35.7 | 71.4 | 142.8 | 214.3 | 464.3 | 571.4 | 928.5 | 1249.9 | 2499.9 |
| NUU | 37.6 | 5.3 | 150.5 | 225.8 | 489.3 | 602.2 | 978.6 | 1317.3 | 2634.0 |
| ARW | 48.2 | 96.4 | 192.9 | 289.3 | 626.9 | 771.6 | 1253.8 | 1687.8 | 3375.0 |
| AIC | 19.4 | 38.7 | 77.4 | 116.1 | 251.6 | 309.6 | 503.2 | 671.3 | 1354.0 |
| AIT | 21.8 | 43.5 | 87.0 | 130.5 | 282.9 | 348.1 | 565.7 | 761.5 | 1523. |
| ALB | 31.4 | 62.8 | 125.5 | 188.3 | 408.0 | 502.1 | 815.9 | 1098.4 | 2196. |
| AIA | 26.6 | 53.1 | 106.3 | 159.4 | 345.4 | 425.1 | 690.8 | 929.9 | 1859. |
| AIF | 36.2 | 72.4 | 144.8 | 217.2 | 470.5 | 579.1 | 941.0 | 1266.8 | 2533. |
| AUC | 21.8 | 43.5 | 87.0 | 130.5 | 282.9 | 348.1 | 565.7 | 761.5 | 1523 |
| AUT | 26.5 | 53.1 | 106.3 | 159.4 | 345.4 | 425.1 | 690.8 | 929.9 | 1859. |
| AUB | 43.4 | 86.8 | 173.6 | 260.5 | 564.3 | 694.6 | 1127.7 | 1519.4 | 3038 |
| AUA | 36.2 | 12.4 | 144.8 | 217.2 | 470.5 | 579.1 | 941.0 | 1266.B | 2533. |
| AUF | 50.6 | 101.3 | 202.5 | 303.6 | 658.2 | 810.1 | 1316.4 | 1772.0 | 3544. |
| SF | 11.7 | 23.3 | 46.6 | 69.9 | 151.5 | 186.4 | 303.0 | 407.9 | 815. |
| MEE | 26.1 | 52.2 | 104.4 | 156.5 | 339.2 | 417.4 | 678.3 | 913.1 | 1826. |
| MFA | 33.3 | 66.6 | 133.2 | 199.8 | 433.0 | 532.9 | 866.0 | 1165.7 | 2331. |
| USL | 60.3 | 120.5 | 241.0 | 361.5 | 783.3 | 964.0 | 1566.6 | 2108.8 | 4217. |
| HL | 69.9 | 139.8 | 279.5 | 419.3 | 908.4 | 1118.0 | 1816.8 | 2445.7 | 4893. |
| CL | 1065.9 | 2131.8 | 4263.7 | 6395.5 | 13857.0 | 17054.8 | 27714.0 | 37307.4 | 74614. |

TABLE 5.2 PART FRACTION DEFECTIVE, MICROELECTRONIC DEVICES (PPM*)

* Parts Per Million

| | | | Quali | ty Level | |
|-------------|--------|--------|---------|----------|----------|
| Environment | JANTXV | JANTX | JAN | Lower | Plastic |
| GB | 10.9 | 21.9 | 109.3 | 546.6 | 1093.2 |
| GF | 34.6 | 69.2 | 346.0 | 1730.2 | 3460.4 |
| GM | 98.8 | 189.5 | 947.7 | 4738.5 | 9477.C |
| MP | 65.2 | 130.4 | 651.8 | 3259.0 | 6518.0 |
| NSB | 54.3 | 108.7 | 543.3 | 2716.5 | 5433.1 |
| NS | 54.3 | 108.7 | 543.3 | 2716.5 | 5433.1 |
| NU | 109.6 | 219.1 | 1095.7 | 5478.3 | 10956.6 |
| -NH | 99.7 | 199.4 | 997.0 | 4985.1 | 9970.2 |
| NUU | 104.6 | 209.3 | 1046.3 | 5231.7 | 10463.4 |
| ARW | 139.2 | 278.3 | 1391.6 | 6957.8 | 13915.6 |
| AIC | 52.9 | 105.7 | 528.5 | 2642.6 | 5285.1 |
| AIT | 80.0 | 160.0 | 799.8 | 3998.8 | 7997.5 |
| AIB | 178.6 | 357.2 | 1786.1 | 8930.5 | 17860.9 |
| AIA | 104.6 | 209.3 | 1046.3 | 5231.7 | 10463.4 |
| AIF | 203.3 | 406.5 | 2032.7 | 10163.4 | 20326.8 |
| AUC | 80.0 | 160.0 | 799.8 | 3998.8 | 7997.5 |
| AUT | 129.3 | 258.6 | 1292.9 | 6464.6 | 12929.2 |
| AUB | 301.9 | 603.8 | 3019.0 | 15095.1 | 30190.1 |
| AUA | 178.6 | 357.2 | 1786.1 | 8930.5 | 17860.9 |
| AUF | 326.6 | 653.1 | 3265.6 | 16328.0 | 32656.0 |
| SF | 8.0 | 15.9 | 79.7 | 398.6 | 797.3 |
| MFF | 65.2 | 130.4 | 651.8 | 3259.0 | 6518.0 |
| MFA | 89.8 | 179.7 | 898.4 | 4491.9 | 8983.9 |
| USL | 183.5 | 367.1 | 1835.4 | 9177.O | 18354.1 |
| ML | 208.2 | 416.4 | 2082.0 | 10410.0 | 20819.9 |
| CL | 3408.9 | 6817.7 | 34088.7 | 170443.3 | 340386.7 |

TABLE 5.3 PART FRACTION DEFECTIVE, TRANSISTORS (PPM*)

* PPM - Part Per Million

. **B** 1

| • | |
|---|--|

| | | | Qua | lity Lev | el | |
|-------------|-------|--------|--------|----------|---------|---------|
| Environment | JANS | JANTXV | JANTX | JAN | Lower | Plastic |
| GB | 1.2 | 5.9 | 11.8 | 59.2 | 296.2 | 592.3 |
| GF | 1.7 | 8.6 | 17.2 | 86.0 | 430.0 | 860.0 |
| GM | 4.3 | 21.6 | 43.2 | 216.2 | 1080.8 | 2161.5 |
| MP | 3.2 | 16.1 | 32.2 | 160.8 | 803.8 | 1607.7 |
| NSB | 1.9 | 9.4 | 18.9 | 94.2 | 471.5 | 943.1 |
| NS | 1.9 | 9.4 | 18.9 | 94.3 | 471.5 | 943.1 |
| NU | 4.9 | 24.4 | 48.8 | 243.8 | 1219.2 | 2438.5 |
| NH | 4.5 | 22.5 | 45.1 | 225.4 | 1126.9 | 2253.8 |
| NUU | 4.7 | 23.5 | 46.9 | 234.6 | 1173.1 | 2346.2 |
| ARW | 6.0 | 29.9 | 59.8 | 299.2 | 1496.2 | 2992.3 |
| AIC | 3.8 | 18.8 | 37.7 | 188.5 | 942.3 | 1884.6 |
| AIT | 4.7 | 23.5 | 46.9 | 234.6 | 1173.1 | 2346.2 |
| AIB | 6.5 | 32.7 | 65.4 | 326.9 | 1634.6 | 3269.2 |
| AIA | 5.6 | 28.1 | 56.2 | 280.8 | 1403.8 | 2807.7 |
| AIF | 7.5 | 37.3 | 74.6 | 373.1 | 1865.4 | 3730.8 |
| AUC | 5.6 | 28.1 | 56.2 | 280.8 | 1403.8 | 2807.7 |
| AUT | 6.5 | 32.7 | 65.4 | 326.9 | 1634.6 | 3269.2 |
| AUB | 10.2 | 51.2 | 102.3 | 511.5 | 2557.7 | 5115.4 |
| AUA | 8.4 | 41.9 | 83.8 | 419.2 | 2096.2 | 4192.3 |
| AUF | 10.2 | 51.2 | 102.3 | 511.5 | 2557.7 | 5115.4 |
| SF | 1.2 | 5.9 | 11.8 | 59.2 | 296.2 | 592.3 |
| MFF | 3.2 | 16.1 | 32.2 | 160.8 | 803.8 | 1607.7 |
| MFA | 4.1 | 20.7 | 41.4 | 206.9 | 1034.6 | 2069.2 |
| USL | 7.6 | 38.2 | 76.5 | 382.3 | 1911.5 | 3823.1 |
| ML | 8.6 | 42.8 | 85.7 | 428.5 | 2142.3 | 4284.6 |
| CL | 128.4 | 64ì.9 | 1283.6 | 6419.2 | 32096.2 | 64192.3 |

TABLE 5.4 PART FRACTION DEFECTIVE, DIODES (PPM*)

* PPM - Parts Per Million

| | Quality Level | | | | | | | | |
|-------------|---------------|-------|-------|--------|----------|--------|--|--|--|
| Environment | S | R | Ρ | м | MIL-SPEC | Lower | | | |
| GE | 0.4 | 1.2 | 3.7 | 12.3 | 61.4 | 184. | | | |
| GF | 0.6 | 2.0 | 6.1 | 20.3 | 101.7 | 305. | | | |
| GM | 1.5 | 5.1 | 15.4 | 51.5 | 257.4 | 772. | | | |
| MP | 1.7 | 5.7 | 17.2 | 57.2 | 286.2 | 858. | | | |
| NSB | 0.9 | 3.1 | 9.2 | 30.7 | 153.6 | 460. | | | |
| NS | 1.0 | 3.4 | 10.1 | 33.6 | 168.1 | 504. | | | |
| NU | 2.6 | 8.7 | 26.2 | 87.2 | 436.2 | 1308. | | | |
| NH | 2.6 | 8.7 | 26.2 | 87.2 | 436.2 | 1308. | | | |
| NUU | 2.8 | 9.3 | 27.9 | 93.0 | 465.0 | 1395. | | | |
| ARW | 3.5 | 11.6 | 34.8 | 116.1 | 580.3 | 1740. | | | |
| AIC | 0.6 | 2.1 | 6.3 | 20.9 | 104.6 | 313. | | | |
| AIT | 0.7 | 2.4 | 7.1 | 23.8 | 119.0 | 357. | | | |
| AIB | 1.3 | 4.4 | 13.2 | 44.0 | 219.9 | 659. | | | |
| AIA | 1.2 | 4.1 | 12.3 | 41.1 | 205.5 | 616. | | | |
| AIF | 1.8 | 5.8 | 17.5 | 58.4 | 292.0 | 876. | | | |
| AUC | 1.4 | 4.7 | 14.1 | 46.9 | 234.4 | 703. | | | |
| AUT | 1.3 | 4.4 | 13.2 | 44.0 | 219.9 | 659. | | | |
| AUB | 2.8 | 9.3 | 27.9 | 93.0 | 465.0 | 1395. | | | |
| AUA | 2.8 | 9.3 | 27.9 | 93.0 | 465.0 | 1395. | | | |
| AUF | 3.7 | 12.2 | 36.5 | 121.8 | 609.1 | 1827. | | | |
| SF | 0.3 | 0.9 | 2.6 | 8.8 | 44.1 | 132. | | | |
| MEF | 1.7 | 5.8 | 17.3 | 57.8 | 289.1 | 867. | | | |
| MFA | 2.3 | 7.6 | 22.7 | 75.7 | 378.5 | 1135. | | | |
| USL | 4.7 | 15.6 | 46.9 | 156.4 | 782.1 | 2346. | | | |
| ML | 5.4 | 17.9 | 53.8 | 179.5 | 897.4 | 2692. | | | |
| CL | 88.4 | 294.7 | 884.1 | 2947.0 | 14735.0 | 44205. | | | |

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TABLE 5.5 PART FRACTION DEFECTIVE, RESISTORS (PPM*)

*PPM - Parts Per Million

| | | | | Quality | Level | | |
|-------------|------------|--------|--------|---------|---------|----------|---------|
| Environment | Ś | R | Ρ | M | L | MIL-SPEC | Lower |
| GB | 1.2 | 3.8 | 11.5 | 38.4 | 115.3 | 115.3 | 384.4 |
| 6F | 1.8 | 6.2 | 18.4 | 61.5 | 184.5 | 184.5 | 615.0 |
| 6M | 9.0 | 30.0 | 89.9 | 299.8 | 899.4 | 899.4 | 2998. |
| MP | 12.7 | 42.3 | 126.8 | 422.8 | 1268.4 | 1268.4 | 4228. |
| NSB | 5.8 | 19.2 | 57.7 | 192.2 | 576.6 | 576.6 | 1921.9 |
| NS | 6.3 | 21.1 | 63.4 | 211.4 | 634.2 | 634.2 | 2114. |
| NU | 14.3 | 47.7 | 143.0 | 476.6 | 1429.9 | 1429.9 | 4766. |
| NH | 18.4 | 61.5 | 184.5 | 615.0 | 1845.0 | 1845.0 | 6150. |
| NUU | 20.8 | 69.2 | 207.6 | 691.9 | 2075.6 | 2075.6 | 6918. |
| ARW | 27.7 | 92.2 | 276.7 | 922.5 | 2767.5 | 2767.5 | 9225. |
| AIC | 3.5 | 11.5 | 34:6 | 115.3 | 345.9 | 345.9 | 1153. |
| AIT | 3.5 | 11.5 | 34.6 | 115.3 | 345.9 | 345.9 | 1153. |
| AIB | 5.8 | 19.2 | 57.7 | 192.2 | 576.6 | 576.6 | 1921. |
| AIA | 3.5 | 11.5 | 34.6 | 115.3 | 345.9 | 345.9 | 1153. |
| AIF | 6.9 | 23.1 | 69.2 | 230.6 | 691.9 | 691.9 | 2306. |
| AUC | ö.6 | 28.8 | 86.5 | 288.3 | 864.8 | 864.8 | 2882. |
| AUT | 9.2 | 30.7 | 92.2 | 307.5 | 922.5 | 922.5 | 3075. |
| AUB | 11.5 | 38.4 | 115.3 | 384.4 | 1153.1 | 1153.1 | 3843. |
| AUA | 9.2 | 30.7 | 92.2 | 307.5 | 922.5 | 922.5 | 3075. |
| AUF | 17.3 | 57.7 | 173.0 | 576.6 | 1729.7 | 1729.7 | 5765. |
| SF | 0.9 | 3.1 | 9.2 | 30.7 | 92.2 | 92.2 | 307. |
| MFF | 12.7 | 42.3 | 126.8 | 422.8 | 1268.4 | 1268.4 | 4228. |
| MFA | 17.3 | 57.7 | 173.0 | 576.6 | 1729.7 | 1729.7 | 5765. |
| USL | 36.9 | 123.0 | 369.0 | 1230.0 | 3690.0 | 3690.0 | 12300. |
| ML | 41.5 | 138.4 | 415.1 | 1383.7 | 4151.2 | 4151.2 | 13837. |
| CL | 703.4 | 2344.7 | 7034.1 | 23446.9 | 70340.6 | 70340.6 | 234468. |

TABLE 5.6 PART FRACTION DEFECTIVE, CAPACITORS (PPM*)

Downloaded from http://www.everyspec.com

*PPM - Parts Per Million

| | | Quality Level |
|-------------|----------|---------------|
| Environment | MIL-SPEC | Lower |
| GB | 537.2 | 1790.7 |
| GF | 1222.9 | 4076.4 |
| GM | 1996.1 | 7140.1 |
| MP | 2142.0 | 6653.8 |
| NSB | 1135.4 | 3784.6 |
| NS | 1222.9 | 4076.4 |
| NU | 2433.8 | 8112.7 |
| NH | 2725.6 | 9085.3 |
| NUU | 3017.4 | 10058.0 |
| ARW | 3892.7 | 12975.8 |
| AIC | 1047.8 | 3492.8 |
| AIT | 1266.7 | 4222.3 |
| AIB | 1266.7 | 4222.3 |
| AIA | 1266.7 | 4222.3 |
| Alf | 1704.4 | 5681.2 |
| AUC | 1339.6 | 4465.4 |
| AUT | 1339.6 | 4465.4 |
| AUB | 1485.5 | 4951.7 |
| Λυλ | 1485.5 | 4951.7 |
| AUF | 1850.3 | 6167.5 |
| SF | 537.2 | 1790.7 |
| MFF | 1996.1 | 6653.8 |
| MFA | 2579.7 | 8599.0 |
| USL | 5059.9 | 16866.2 |
| ML | 5643.4 | 18811.5 |
| CL | 89385.3 | 297951.1 |

TABLE 5.7 PARTS FRACTION DEFECTIVE, INDUCTIVE DEVICES (PPM*)

Downloaded from http://www.everyspec.com

*PPM - Parts Per Million

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Downloaded from http://www.everyspec.com

| nvironment | Fraction defective (Defects/10 ⁶) |
|------------|---|
| GB | 5935.2 |
| GF | 11663.1 |
| GM | 30168.5 |
| MP | 27965.5 |
| NSB | 14967.6 |
| NS | 16289.4 |
| NU | 34574.6 |
| ИН | 38980.6 |
| NUU | 43386.7 |
| ARW | 56604.8 |
| AIC | 12544.3 |
| AIT | 13645.8 |
| AIB | 15848.8 |
| AIA | 13645.8 |
| AIF | 23559.4 |
| AUC | 14747.3 |
| AUT | 18051.9 |
| AUB | 20254.9 |
| AUA | 18051.9 |
| AUF | 25762.5 |
| SF | 5935.2 |
| MFF | 27965.5 |
| USL | 74229.1 |
| ML | 83041.2 |
| CL | ***** |

TABLE 5.8 PART FRACTION DEFECTIVE, ROTATING DEVICES (PPM*)

*PPM - Parts Per Million

Downloaded from http://www.everyspec.com

| | | Quality Level |
|-------------|----------|---------------|
| Environment | MIL-SPEC | Lower |
| GB | 142.5 | 210.9 |
| GF | 231.4 | 388.8 |
| GM | 635.1 | 1784.5 |
| MP [| 1510.8 | 4384.3 |
| NSB | 621.4 | 1716.0 |
| NS | 621.4 | 1716.0 |
| NU | 1031.9 | 2673.9 |
| NH | 2263.4 | 6642.0 |
| NUU | 2400.2 | 6915.7 |
| ARW | 3221.2 | 9652.3 |
| AIC | 450.3 | 724.0 |
| AIT | 484.5 | 1100.3 |
| AIB | 758.2 | 1442.4 |
| AIA | 587.2 | 1100.3 |
| AIF | 758.2 | 1784.5 |
| AUC | 621.4 | 1442.4 |
| AUT | 689.8 | 1784.5 |
| AUB | 1100.3 | 2810.7 |
| AUA | 758.2 | 2126.5 |
| AUF | 1100.3 | 3152.8 |
| SF | 142.5 | 210.9 |
| MFF | 1510.8 | 4384.3 |
| MFA | 2058.1 | 5684.2 |
| USL | 4315.8 | 13073.1 |
| ML | 4931.6 | 14441.4 |
| CL | N/A | N/A |

TABLE 5.9 PART FRACTION DEFECTIVE, RELAYS (FPM*)

*PPM - Parts Per Million

| è | 5 - S - S |
|-----|----------------|
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| g | 10,11,13 |
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| Environment | NIL-SPEC | Lower | |
|-------------|----------|---------|--|
| GB | 1.4 | 24.4 | |
| GF | 2.4 | 44.0 | |
| GM | 8.8 | 158.4 | |
| MP | 12.8 | 230.6 | |
| NSB | 5.3 | 95.5 | |
| NS | 5.3 | 95.5 | |
| NU | 12.2 | 220.3 | |
| NH | 19.1 | 344.1 | |
| NUU | 20.3 | 364.7 | |
| ARW | 27.1 | 488.4 | |
| AIC | 5.4 | 96.6 | |
| AIT | 5.4 | 96.6 | |
| AIB | 9.4 | 168.8 | |
| AIA | 9.4 | 168.8 | |
| AIF | 12.2 | 220.3 | |
| AUC | 6.5 | 117.2 | |
| AUT | 5.5 | 117.2 | |
| AUB | 12.2 | 220.3 | |
| AUA | 12.2 | 220.3 | |
| AUF | 15.1 | 271.9 | |
| SF | 1.4 | 24.4 | |
| MFF | 12.3 | 230.6 | |
| MFA | 17.4 | 313.1 | |
| USL | 36.9 | 663.7 | |
| ML NL | 41.5 | 746.2 | |
| CL | 688.3 | 12388.6 | |

TABLE 5.10 PART FRACTION DEFECTIVE, SWITCHES (PPM*)

Quality Level

*Parts Per Million



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| | | | | | | | | Crimp | | |
|-------------|----------------|-------|--------------------|----------------------------|--------------|------------------|-------|----------------|---------------|----------------|
| Environment | Hand Solder | Weld | Solderless Wrap | Wrapped and Soldered | Clip Jerm | Reflow Solder | Auto | Man., Upper | Man., Std. | Man., Lower |
| 68 | 12. | 0.2 | 50.0 | 1. | 1. | 0.3 | 1.2 | 1.2 | 2.5 | 24.8 |
| 6F | 26. | 0.5 | 0.03 | 1. | 1. | 0.7 | 2.6 | 2.6 | 5.2 | 52.0 |
| 6M | 90. | 1.7 | 0.12 | 5. | 4. | 2.4 | 9.0 | 9.0 | 18.1 | 180.8 |
| МР | 90. | 1.7 | 0.12 | 5. | 4. | 2.4 | 9.0 | 9.0 | 18.1 | 180.0 |
| NSB | 43. | 0.8 | 0.06 | 2. | 2. | 1.1 | 4.3 | 4.3 | 8.7 | 86.7 |
| NS | 54. | 1.0 | 0.07 | 3. | 3. | 1.4 | 5.4 | 5.4 | 10.9 | 109.0 |
| NU | 123. | 2.4 | 0.16 | 1. | 6. | 3.3 | 12.3 | 12.3 | 24.5 | 245.1 |
| MH | 136. | Z.6 | 0.18 | 1. | 6. | 3.6 | 13.6 | 13.6 | 21.2 | 212.4 |
| NUU | 149. | 2.9 | 0.20 | 8. | 1 7. | 3.9 | 14.9 | 14.9 | 29.7 | 297.1 |
| ARM | 198. | 3.8 | 0.27 | 11. | 9. | 5.3 | 19.0 | 39.6 | 39.6 | 396.2 |
| ALC | 31. | 0.6 | 0.04 | 2. | 1. | 0.8 | 3.1 | 3.1 | 6.2 | 61.9 |
| A11 | 56. | 1.1 | 0.07 | 3. | 3. | 1.5 | 5.6 | 5.6 | (11.1 | 111.4 |
| A19 | 68. | 1.3 | 0.09 | 4. | 3. | 1.8 | 6.8 | 6.8 | 13.6 | 136.2 |
| ALA | 62. | 1.2 | 0.08 | 3. | 3. | 1.6 | 6.2 | 6.8 | 12.4 | 123.8 |
| AIF | 93. | 1.8 | 0.12 | 5. | 4. | 2.5 | 9.3 | 9.3 | 18.6 | 185.7 |
| AUC | 37. | 0.7 | 0.05 | 2. | 2. | 1.0 | 3.7 | 3.7 | 7.4 | 74.3 |
| AU1 | 14 | 1.4 | 0.10 | 4. | 3. | 2.0 | 7.4 | 7.4 | 14.9 | 148.6 |
| AUB | 93. | 1.8 | 0.12 | 5. | 4. | 2.5 | 9.3 | 9.3 | 18.6 | 105.7 |
| AUA | 87. | 1.7 | 0.12 | 5. | 4. | 2.3 | 8.7 | 8.7 | 17.3 | 173.3 |
| AUF | 118. | 2.3 | 0.16 | 6. | 5. | 3.1 | 11.8 | 11.8 | 23.5 | 235.2 |
| SF | 12. | 0.2 | 0.02 | 1. | 1. | 0.3 | 1.2 | 2.5 | 2.5 | 24.8 |
| MEE | 90. | 1.7 | 0.12 | 5. | 4. | 2.4 | 9.0 | 9.0 | 18.1 | 180.8 |
| MFA | 124. | 2.4 | 0.17 | 1. | 6. | 3.3 | 12.4 | 12.4 | 24.6 | 247.6 |
| USL | 212. | 5.2 | 0.37 | 15. | 13. | 1.2 | 27.2 | 21.2 | 54.5 | 544.8 |
| ML | 310. | 6.0 | 0.42 | 17. | 14. | 8.2 | 31.0 | 31.0 | 61.9 | 619.0 |
| CL | 5200. | 100.0 | 7.0 | 280. | 240. | 138.0 | 520.0 | 520.0 | 1040.0 | 10400.0 |

TABLE 5.11 PART FRACTION DEFECTIVE, CONNECTIONS (PPH*)

Connection Type

* Parts Per Million

Pinese -

| Г | | Aug 1444 1 4 40 |
|-------------|----------|-----------------|
| | | Quality Level |
| Environment | MIL-SPEC | Lower |
| GB | 73.7 | 97.3 |
| GF | 83.2 | 248.1 |
| GM | 417.7 | 1204.6 |
| MP | 427.1 | 827.7 |
| NSB | 219.8 | 408.3 |
| NS | 276.3 | 544.9 |
| NU | 639.2 | 1298.9 |
| NH | 639.2 | 1251.8 |
| NUU | 686.3 | 1346.0 |
| ARW | 921.9 | 1770.1 |
| AIC | 120.9 | 497.8 |
| AIT | 168.0 | 497.8 |
| AIB | 238.7 | 733.4 |
| AIA | 215.1 | 733.4 |
| AIF | 332.9 | 969.0 |
| AUC | 262.2 | 733.4 |
| AUT | 463.6 | 733.4 |
| AUB | 497.8 | 969.0 |
| AUA | 474.3 | 969.0 |
| AUF | 733.4 | 1440.2 |
| SF | 73.7 | 97.3 |
| MFF | 427.1 | 827.7 |
| MFA | 592.1 | 1157.5 |
| USL | 1204.6 | 2382.7 |
| ML | 1393.1 | 2759.6 |
| CL | 23115.8 | 45733.8 |

TABLE 5.12 PART FRACTION DEFECTIVE, CONNECTORS (PPM*)

*Parts Per Million

Downloaded from http://www.everyspec.com

| | | Quality Level |
|-------------|----------|---------------|
| Environment | MIL-SPEC | Lower |
| GB | 425.0 | 4250.0 |
| GF | 690.3 | 6903.2 |
| GM | 1792.4 | 17924.3 |
| MP | 1629.2 | 16291.5 |
| NSB | 1057.7 | 10576.9 |
| NS | 1302.6 | 13026.0 |
| NU | 2670.0 | 26700.3 |
| NH | 2874.1 | 28741.2 |
| NUU | 3078.2 | 30782.2 |
| ARW | 4098.7 | 40986.9 |
| AIC [| 731.1 | 7311.4 |
| AIT | 1139.3 | 11393.2 |
| AIB | 1853.7 | 18536.5 |
| AIA | 1567.9 | 15679.2 |
| AIF | 2261.8 | 22618.4 |
| AUC | 1751.6 | 17516.1 |
| AUT | 3282.3 | 32823.1 |
| AUB | 5323.3 | 53232.5 |
| AUA | 4302.8 | 43027.8 |
| AUF | 7364.2 | 73641.9 |
| SF | 425.0 | 4250.0 |
| MFF | 1996.5 | 19955.2 |
| MFA | 2670.0 | 26700.3 |
| USL | 5527.3 | 55273.5 |
| ML | 6139.6 | 61396.3 |
| CL | 102267.9 | ****** |

TABLE 5.13 PART FRACTION DEFECTIVE, PRINTED WIRING BOARDS (PPM*)

*Parts Per Million

5.3 Procedure B - Screen Selection and Placement

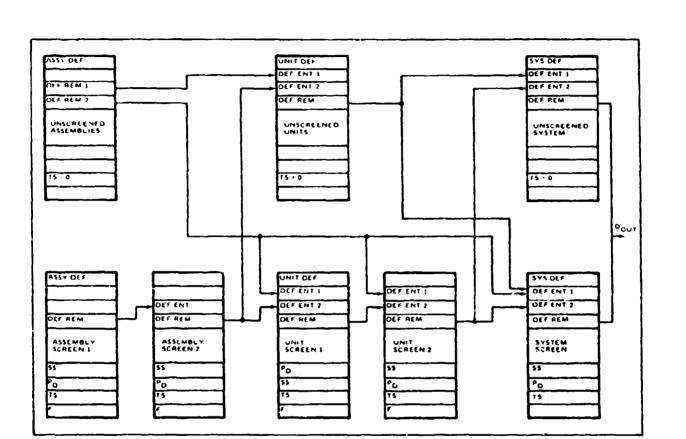
5.3.1 Objective. The objective of this procedure is to select and place screens at appropriate levels of assembly so as to develop a screening program plan for achieving program objectives in a cost effective manner.

5.3.2 Methodology. Procedure steps, outlined below, should be performed iteratively and in conjunction with the next two Procedures C & D.

Iterative application of the procedure should be as follows:

- Initial Screen Selection and Placement (Based upon engineering evaluation, available facilities and procedure B)
- Failure-Free Acceptance Tests (Procedure C)
- Screen Selection and Placement Modification
- Cost Effectiveness Analysis (Procedure D)
- Remaining Defect Density Calculations
- Goals on remaining defect density achieved within given cost constraints

Table 4.9 should be used as a guide for initial screen selection and placement. A diagram of similar defect flow chart, as shown in Figure 5.4, should be used in calculating the remaining defects for various possible screening sequences.



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Figure 5.4 Multilevel Screening Flow Chart

5.3.3 Procedure Steps. Instructions for use of the flow chart in Figure 5.4 are as follows:

<u>Step 1</u>. For each trial screening sequence, identify the units and assemblies that will be screened at their respective levels and those that will not be screened. (See Figure 5.5)

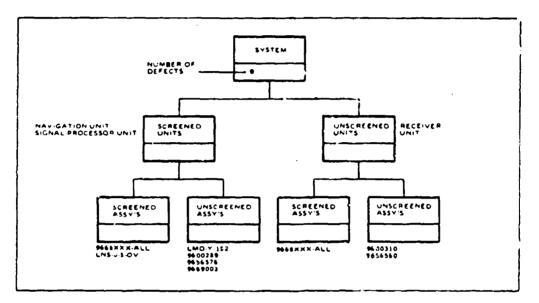


Figure 5.5 Identification of Equipment to be Screened

<u>Step 2</u>. From the Defect Estimation Worksheets of Procedure A, or from the Unit Breakdown Charts, total the estimated number of defects in assemblies to be screened and enter in the block "ASS'Y DEF" for ASSEMBLY SCREEN 1.

<u>Step 3</u>. Similarly, total the estimated number of defects in assemblies that are not to be screened and enter in the block "ASS'Y DEF" for UNSCREENED ASSEMBLIES.

Step 4. Repeat Steps 2 and 3 for Unit and System levels.

<u>Step 5</u>. Select candidate screens using the guidelines. Determine screening strengths for selected screens from Tables 5.14 through 5.18.

<u>Step 6</u>. Determine and enter the Detection Efficiency (D_E) of the tests to be performed during and after screening. For guidance, see Section 4.10.3.4 of the guidebook.

<u>Step 7</u>. Compute test strengths by multiplying screening strengths by their respective detection efficiencies (SS x D_E) and enter.

<u>Step 8</u>. Identify the unscreened assemblies that are installed in unscreened units and enter the total estimated number of defects for those assemblies in the UNSCREENED ASSEMBLIES block DEF REM 1 and in the block DEF ENT 1 of UNSCREENED UNITS. Enter the balance of estimated defects for unscreened assemblies in DEF REM 2.

<u>Step 9.</u> Determine which unscreened assemblies (DEF REM 2) will be installed in units that will first enter UNIT SCREEN 1, UNIT SCREEN 2., or SYSTEM SCREEN. Enter the number of estimated defects into the corresponding DEF ENT 1 block(s).

<u>Step 10</u>. In the ASSEMBLY SCREEN 1 block, calculate the screening fallout, F, by multiplying the ASS'Y DEF by test strength, TS, and enter in block F. Subtract F from ASS'Y DEF and enter difference in DEF REM and DEF ENT in ASSEMBLY SCREEN 2.

NOTE: If a second assembly scheen is not considered, the test strength for ASSEMBLY SCREEN 2 is zero and the defects remaining (DEF REM) will be the same as the defects entering (DEF ENT).

torining he A

<u>Step 11</u>. If TS \neq 0 for ASSEMBLY SCREEN 2, calculate F by multiplying DEF ENT by TS. Subtract F from DEF ENT and enter in DEF REM.

<u>Step 12</u>. Determine which of the screened assemblies will be installed in Units that will enter UNIT SCREEN 1 and those that will be installed in unscreened units. Enter the number of estimated defects into the corresponding DEF ENT 2 block(s).

<u>Step 13</u>. In the UNIT SCREEN 1 block, calculate F by multiplying the sum of DEF ENT 1 and DEF ENT 2 by TS subtract F from the sum of DEF ENT 1 and DEF ENT 2 and enter in DEF REM and in the block DEF ENT 2 of UNIT SCREEN 2.

<u>Step 14</u>. In the UNIT SCREEN 2 block, repeat step 13. Enter the value in DEF REM 2 in the block DEF ENT 2 of SYSTEM SCREEN if the System is to be screened or in the corresponding block in UNSCREENED SYSTEM, if the system is not to be screened.

<u>Step 15</u>. In the UNSCREENED UNITS block, add the values in UNIT DEF, DEF ENT 1, and DEF ENT 2 and enter the sum in DEF REM.

<u>Step 16</u>. Determine which unscreened units will be screened as part of the system screen. Add the estimated defects for those units to the value in DEF ENT 1 of the SYSTEM SCREEN block. Enter the balance of estimated defects for unscreened units in DEF ENT 1 of the UNSCREENED SYSTEM block.

<u>Step 17</u>. In the SYSTEM SCREEN block, calculate F and subtract from the sum of DEF ENT 1 and DEF ENT 2. Enter the difference in DEF REM.

<u>Step 18</u>. In the UNSCREENED SYSTEM block, add the values in DEF ENT 1 and DEF ENT 2 and enter the sum in DEF REM.

<u>Step 19</u>. Add the values in the DEF REM blocks of UNSCREENED SYSTEM and SYSTEM SCREEN blocks. The sum is D_{out} , an estimate of the number of defects remaining after completing the candiate screening sequence. The value of D_{out} must be equal to or less than D_R to satisfy the specified yield requirement.

The above 19 steps complete the initial process of screen selection/placement and remaining defect calculation. The process shall be repeated with alternate or modified screens since more than one screening sequence may quality as a candidate for subsequent cost tradeoff analysis.

5.3.4 Screening Strength Tables. Tables are provided which contain the screening strength of various screen types as a function of the screening parameters and time duration of the screen. The failure rates for defects, as a function of the stress level are also provided. Screen types included are:

- Random Vibration
- Temperature Cycling
- Swept-Sine Vibration
- Constant Temperature

| | | G-RMS Level | | | | | | | | | | | | |
|-----------------------|-------|-------------|---------------|-------|-------|-------|-------|--------|-------|-------|------------------------|-------|-------|-------|
| Duration (minutes) | 0.5 | 1.0 | 1.5 | 2.0 | 2.5 | 3.0 | 3.5 | 4.0 | 4.5 | 5.0 | 5.5 | 6.0 | 6.5 | 7.0 |
| 5 | 0.007 | 0.023 | 0.045 | 0.072 | 0.104 | 0.140 | 0.178 | Q. 218 | 0.260 | 0.303 | ^{0.346} | 0.389 | 0.431 | 0.473 |
| 10 | 0.014 | 0.045 | 0.088 | 0.140 | 0.198 | 0.260 | 0.324 | 0.389 | 0.452 | 0.514 | 0.5/2 | 0.627 | 0.677 | 0.723 |
| 15 | 0.021 | 0.067 | 0.129 | 0.202 | 0.282 | 0.363 | 0.444 | 0.522 | 0.595 | 0.661 | 0.720 | 0.172 | 0.816 | 0.854 |
| 20 | 0.028 | 0.088 | 0.168 | 0.260 | 0.356 | 0.452 | 0.543 | 0.626 | 0,100 | 0.764 | 0.817 | 0.861 | 0.896 | 0.923 |
| 25 | 0.035 | 0.109 | 0.206 | 0.314 | 0.424 | 0.529 | 0.625 | 0.708 | 0.778 | 0.835 | 0.680 | 0.915 | 0.941 | 0.959 |
| 30 | 0.041 | 0.129 | 0.241 | 0.363 | 0.484 | 0.595 | 0.691 | 0.112 | 0.836 | 0.885 | 0.922 | 0.948 | 0.966 | 0.979 |
| 35 | 0.048 | 0,149 | 0.215 | 0.409 | 0.538 | 0.651 | 0.746 | 0.822 | 0.878 | 0.920 | 0.949 | 0.968 | 0.901 | 0.989 |
| 40 | 0.055 | 0,168 | 0. 308 | 0.452 | 0.586 | 0.700 | 0.791 | 0.860 | 0.910 | 0.944 | 0.966 | 0.981 | 0.989 | 0.994 |
| 45 | 0.061 | 0.187 | 0.339 | 0.492 | 0.629 | 0.742 | 0.829 | 0.891 | 0.933 | 0.961 | 0.978 | 0.988 | 0.994 | 0.997 |
| 50 | 0.068 | 0.205 | 0.369 | 0.529 | U.668 | 0.778 | 0.859 | 0.915 | 0.951 | 0.973 | 0.986 | 0.993 | 0.996 | 0.998 |
| 55 | 0.074 | 0.224 | 0.397 | 0.563 | 0.702 | 0.809 | 0.684 | 0.933 | 0.964 | 0.981 | 0.991 | 0.996 | 0.998 | 0.999 |
| 60 | 0.081 | 0.241 | 0.424 | 0.595 | 0.734 | 0.836 | 0.905 | 0.948 | 0.973 | 0.987 | 0.994 | 0.997 | 0.999 | 1.000 |
| ۲ ^۵ | 0.084 | 0.276 | 0.552 | 0.903 | 1.322 | 1.806 | 2.351 | 2.954 | 3.653 | 4.327 | 5.092 | 5.905 | 6.776 | 7.692 |

TABLE 5.14 SCREENING STRENGTH AND FAILURE RATES $\vec{\lambda}_{D},$ RANDOM VIBRATION SCREENS

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| Number | Temp Rate of | | | Temp | erature | Range F | ₹ (⁰ C) | | | |
|--------------|------------------------------|----------------|------------------------|----------------|----------------|------------------------|---------------------|----------------|---|----------------|
| of Cycles | Change ^O C/Min | 20. | 40. | 60. | 80. | 100. | 120. | 140. | 160. | 180. |
| 2 | i | | | | | | | | | |
| | T 5 | .1633 | .2349 | . 2886 | . 3324 | . 3697 | . 4023 | .4312 | .4572 | . 4809 |
| | 10 | . 2907 | .4031 | .4812 | .5410 | .5891 | . 6290 | .6629 | . 6920 | .7173 |
| | 15 20 | .3911 .4707 | .5254 .6155 | .6124 .7034 | .6752 .7636 | .7232 .8075 | .7612 .8407 | .7920 .8665 | .8175 .8871 | .8388 .9037 |
| 4 | | | | | | | | | | |
| | T 5 | . 2998 | .4147 | . 4939 | . 5543 | . 6027 | .6427 | . 6765 | . 7054 | . 7305 |
| | 10 | . 4969 | .6437 | . 7308 | .7893 | .8312 | .8624 | . 8863 | . 9051 | .9201 |
| | 15 20 | .6292 .7198 | .77 48 .8522 | .8498 .9120 | .8945 | .9234 .9629 | .9430 .9746 | .9567 .9822 | .9667 .9873 | .9740 .990? |
| | 20 | . / 130 | .0522 | . 9120 | . 3441 | . 3023 | . 3740 | . 3022 | . 3073 | . 330 : |
| 6 | Ť | | | | | | | | | |
| | 5 | . 4141 | . 5522 | . 6400 | . 7025 | . 7496 | . 7864 | .8160 | .8401 | .8601 |
| | 10 | .6431 | . 1873 | .8603 | .9033 | .9306 | .9489 | .9617 | .9708 | .9774 |
| | 15 20 | .7742 .8517 | .8931 .9432 | .9418 .9739 | .9657 .9868 | .9788 .9929 | .9864 .9960 | .9910 .9976 | .9939 .9986 | .9958 .9991 |
| 8 | Ť | | | | | | | | | |
| | 1 5 | . 5098 | .6574 | . 7439 | .8014 | .8422 | .8723 | . 8953 | .9132 | .9274 |
| | 10 | .7469 | .8731 | .9275 | .9556 | .9715 | .9811 | . 9871 | .9910 | . 9936 |
| | 15 20 | .8625 | .9493 .9781 | .9774 .9923 | .9889 | .9941 | .9967 .9994 | .9981 .9997 | .9989 | .9993 .9999 |
| | 20 | . 92 1 3 | i . 3701 | | i . 3 9 0 9 | | } | . , , , , , , | .,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,, | |
| 10 | ŧ | | i | | 1 | ; | | | i | |
| | 5 | . 5898 | . 1379 | .8178 | .8674 | .9005 | .9237 | .9405 | .9529 | .9623 |
| | 10 | . 8204 | .9242 | .9624 | .9796 | .9883 | .9930 | .9956 | .9972 | .9982 |
| | 15 20 | .9163 .9585 | .9759 .9916 | .9913 .9977 | .9964 .9993 | .9 984 .9997 | .9992 .9999 | .9996 .9999 | .9998 .9999 | .9999 .9999 |
| 12 | | | | | | | | | | |
| | • T 5 | | | | | | | | 0.144 | 0.004 |
| 1 | 5 10 | .6568 .8726 | .7994 | .8704 | .9115 | .9373 | .9544 .9974 | .9661 .9985 | .9744 .9991 | .9804 .9995 |
| | 15 | .9490 | .9886 | .9966 | . 9988 | .9996 | . 9998 | . 9999 | .9999 | . 9999 |
| | 20 | .9780 | .9968 | .9993 | . 9998 | .9999 | . 9999 | . 9999 | .9999 | .9999 |

| TABLE | 5.15 | SCREENING | STRENGTH | TEMPERATURE | CYCLING | SCREENS |
|-------|------|-----------|----------|-------------|---------|---------|
|-------|------|-----------|----------|-------------|---------|---------|

TABLE 5.16 FAILURE RATES TO, TEMPERATURE CYCLING SCREENS

| Temp. Rate of | | | Tempe | erature | Range | R (°C) | <u> </u> | | |
|---------------------|--------|--------|--------|---------|--------|--------|----------|--------|------------------|
| Change ºC/Min | 20. | 40. | 60. | 80. | 100. | 120. | 140. | 160. | 180. |
| i | | | | | | | | | |
| 5 10 15 | Q.1717 | 0.2580 | 0.3281 | 0.3893 | 0.4447 | 0.4958 | 0.5436 | 0.5888 | 0.3278 0.6317 |
| 20 | 0.2480 | 0.3726 | 0.4739 | 0.5623 | 0.6423 | 0.7161 | 0.7852 | 0.8504 | 0.9125 |

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| | G Level | | | | | | | | | | | | |
|-----------|---------|--------|--------|--------|--------|--------|--------|--------|-------------|-----------|--------|--------|--------|
| | | | | | | | | | | | | | |
| Duration | | | | | | | | | | | | | |
| (minutes) | 0.5 | 1.0 | 1.5 | 2.0 | 2.5 | 3.0 | 3.5 | 4.0 | 4.5 5.0 | 5.5 | 6.0. | 6.5 | 1.0 |
| | | | | | | | | | | | | | |
| 5 | 0.0020 | 0.0036 | 0.0051 | 0.0066 | 0.0080 | 0.0093 | 0.0107 | 0.0120 | 0.0132 0.01 | 45 0.0157 | 0.0169 | 0.0181 | 0.0193 |
| 10 | 0.0040 | 0.0072 | 0.0103 | 0.0131 | 0.0159 | 0.0186 | 0.0212 | 0.0238 | 0.0263 0.02 | 87 0.0312 | 0.0335 | 0.0359 | 0.0382 |
| 15 | 0.0060 | 0.0108 | 0.0154 | 0.0196 | 0.0238 | 0.0278 | 0.0316 | 0.0354 | 0.0391 0.04 | 28 0.0464 | 0.0499 | 0.0534 | 0.0568 |
| 20 | 0.0080 | 0.0144 | 0.0204 | 0.0261 | 0.0316 | 0.0368 | 0.0420 | 0.0470 | 0.0519 0.09 | 66 0.0614 | 0.0660 | 0.0705 | 0.0750 |
| 25 | 0.0099 | 0.0180 | 0.0255 | 0.0325 | 0.0393 | 0.0458 | 0.0522 | 0.0584 | 0.0644 0.0 | 03 0.0761 | 0.0818 | 0.0874 | 0.0929 |
| 30 | 0.0119 | 0.0216 | 0.0305 | 0 0389 | 0.0470 | 0.0547 | 0.0623 | 0.0696 | 0.0768 0.0 | 38 0.0906 | 0.0973 | 0.1039 | 0.1104 |
| 35 | 0.0139 | 0.0251 | 0.0355 | 0.0452 | 0.0546 | 0.0636 | 0.0723 | 0.0807 | 0.0890 0.0 | 70 0.1049 | 0.1126 | 0.1201 | 0.1225 |
| 40 | 0.0159 | 0.0287 | 0.0404 | 0.0515 | 0.0621 | 0.0723 | 0.0822 | 0.0917 | 0.1010 0.1 | 01 0.1189 | 0.1276 | 0.1361 | 0.1444 |
| 45 | í | i I | | 1 | (I | 1 | (| (| 0.1129 0.13 | 1 | 1 | 1 1 | 1 (|
| 50 | | 1 | | | | | | | 0.1246 0.1 | | ļ | | 1 1 |
| 55 | ł | | | | | • | i | | 0.1362 0.1 | 1 | | | 1 1 |
| 60 | 1 | ł | | 1 | 1 | 1 | 1 | 1 | 0.1476 0.1 | 1 | ł | 1 | ł ł |
| | | | | 1 | 1 | 1 | | | 0.1597 0.1 | | 1 | 1 | |
| , D | 0.0240 | 0.0430 | 0.0019 | 0.0193 | 0.0902 | 0.1120 | 0.1200 | | | | | | |

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| TABLE 5.18 SCREENING STRENGTH & FAILURE RATES | \mathbf{X}_{D} , constant temperature screens |
|---|---|
|---|---|

| Time | Temperature Delta (DT) | | | | | | | | |
|----------------|------------------------|--------|---------------|---------|--------|--------|--------|----------------|--------|
| -in Hours | 0. | 10. | 20. | 30. | 40. | 50. | 60. | 70. | 80. |
| 10 | 0.0124 | 0.0677 | 0.0991 | 0.1240 | 0.1452 | 0.1639 | 0.1809 | 0.1964 | 0.2108 |
| 20 | 0.0247 | 0.1308 | 0.1885 | 0.2326 | 0.2693 | 0.3010 | 0.3290 | 0.3542 | 0.3772 |
| 30 | 0.0368 | 0.1896 | 0.2689 | 0.3278 | 0.3754 | 0.4156 | 0.4504 | 0.4810 | 0.5084 |
| 40 | 0.0488 | 0.2445 | 0.3414 | 0.4112 | 0.4661 | 0.5114 | 0.5498 | 0.5830 | 0.6121 |
| 50 | 0.0 606 | 0.2956 | 0.4067 | 0.4842 | 0.5436 | 0.5915 | 0.6312 | 0.6649 | 0.6938 |
| 60 | 0.0723 | 0.3433 | 0.4655 | 0.5481 | 0.6099 | 0.6584 | 0.6979 | 0.7 307 | 0.7584 |
| 70 | 0.0839 | 0.3877 | 0.5185 | 0.6042 | 0.6665 | 0.7144 | 0.7525 | 0.7836 | 0.8093 |
| 80 | 0.0953 | 0.4292 | 0.5663 | 0.6533 | 0.7149 | 0.7612 | 0.7973 | 0.8261 | 0.8495 |
| 90 | 0.1065 | 0.4678 | 0.6093 | 0.6963 | 0.7563 | 0.8004 | 0.8339 | 0.8602 | 0.8812 |
| 100 | 0.1176 | 0.5038 | 0.6480 | 0.7339 | 0.7917 | 0.8331 | 0.8640 | 0.8877 | 0.9063 |
| 110 | 0.1286 | 0.5374 | 0.6829 | 0.7669 | 0.8219 | 0.8605 | 0.8886 | 0.9097 | 0.9260 |
| 120 | 0.1394 | 0.5687 | 0.7144 | 0.7958 | 0.8478 | 0.8833 | 0.9087 | 0.9275 | 0.9416 |
| 130 | 0.1501 | 0.5979 | 0.7427 | 0.8211 | 0.8699 | 0.9025 | 0.9252 | 0.9417 | 0.9539 |
| 140 | 0.1607 | 0.6251 | 0.7682 | 0.8433 | 0.8888 | 0.9184 | 0.9388 | 0.9532 | 0.9636 |
| 150 | 0.1711 | 0.6505 | 0.7912 | 0.8628 | 0.9049 | 0.3318 | 0.9498 | 0.9624 | 0.9713 |
| 160 | 0.1814 | 0.6742 | : ! 0.8119 | 0.8798 | 0.9187 | 0.9430 | 0.9589 | 0.9697 | 0.9774 |
| 170 | 0.1916 | 0.6962 | 0.8305 | 0.8947 | 0.9305 | 0.9523 | 0.9663 | 0.9757 | 0.9821 |
| 180 | 0.2017 | | 0.8473 | 0.9077 | 0.9406 | 0.9602 | 0.9724 | 0.9805 | 0.9859 |
| 190 | 0.2116 | 0.7360 | 0.8625 | 0.9192 | 0.9492 | 0.9667 | 0.9774 | 0.9843 | 0.9889 |
| 200 | 0.2214 | 0.7538 | 0.8761 | 0.9292 | 0.9566 | 0.9721 | 0.9815 | 0.9874 | 0.9912 |
| ν _D | 0.0013 | 0.0070 | 0.0104 | 0.'0132 | 0.0157 | 0.0179 | 0.0199 | 0.0219 | 0.0237 |

5.4 Procedure C - Failure-Free Acceptance Test (FFAT).

<u>5.4.1 Objective</u>. The objective of this procedure is to determine the length T of a failure-free period which provides a given statistical confidence that the yield goal (remaining defect density) has been achieved.

5.4.2 <u>Methodology</u>. The values of three parameters should be determined in establishing failure-free acceptance test requirements.

- (a) $\overline{\lambda}_0$ The predicted or specified failure rate for the system (per MIL-HDBK-217)
- (b) $\overline{\lambda}_D$ The average failure rate of a defect under the stress screen to be used in the FFAT. (Tables 5.14 to 5.18 in Procedure B)
- (c) $\frac{\overline{\lambda}D}{\lambda}$ The ratio of the failure rate of a defect and the predicted failure rate of the system.

Tables 5.19 through 5.28 provide 90, 80, 70, 60 and 50% lower confidence bounds on yield as a function of the parameters defined above and T, the length of the failure-free period.

5.4.3 Procedure Steps. The following outlines the procedural steps involved in determining the failure-free period.

<u>Step 1</u>. Determine the predicted failure rate for the system in accordance with MIL-HDBK-217. The prediction should be based upon the more detailed MIL-HDBK stress analysis procedures rather than simple part count estimation procedures.

<u>Step 2</u>. Establish the average defect density entering the system level screen, D_{IN_3} based upon prior screening results.

<u>Step 3</u>. Determine the screen type most appropriate for use at the system level, based upon prior knowledge of screen effectiveness and the type of defects expected to be present.

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<u>Step 4</u>. Translate the yield requirement or goal into defect density which will remain in the equipment upon completion of the failure-free acceptance test. (i.e. $D_p = \ln yield$).

<u>Step 5</u>. Determine the required test strength for the screen/test to reduce D_{IN_2} to D_R . i.e.,

$$D_R = D_{IN_3} (1-TS)$$

and

TS = 1

<u>Step 6</u>. Use the estimated test detection efficiency (DE) for the tests which will be applied at the system level.

<u>Step 7</u>. Determine the required screening strength SS for the screen which will be used during the failure-free acceptance test. i.e.,

$$SS = -$$

DE

<u>Step 8</u>. Select a screen with the required SS determined in the previous step (7) from the Tables 5.14 through 5.18 of Procedure B. Note that the screen should not be selected based upon screening strength alone. The FFAT screen should be selected based upon analyses of screen fallout data at lower assembly levels, the quantity and type of defects expected to be present in the final system product prior to the FFAT and the screen type believed to be most effective for those defects.

<u>Step 9</u>. Determine the failure rate of a defect $(\overline{\lambda}_D)$ for the screen selected in Step 8, using the same Tables 5.14 through 5.18 of Procedure B.

<u>Step 10</u>. Determine the failure rate ratio $\frac{\lambda_D}{\lambda_0}$ and the statistical confidence required for verifying the yield requirement.

<u>Step 11</u>. Using Tables 5.19 through 5.28, select the table corresponding to the statistical confidence desired.

<u>Step 12</u>. Find the column in the table corresponding to the ratio $\frac{\lambda_D}{\lambda_o}$ and proceeding down that column, find the value of yield which corresponds to the requirement or goal.

<u>Step 13</u>. Find the value of $\overline{\lambda}_D$ in the left most column of the table which corresponds to the yield value found by Step 12.

<u>Step 14</u>. Divide the value of $\overline{\lambda}_D T$ found in Step 13 by $\overline{\lambda}_D$, the defect failure rate of Step 9, to determine the length of the failure-free period T.

<u>Step 15</u>. Successful completion of the failure-free acceptance test will provide x% confidence that the actual yield is not less than the required value.

5.4.4 Tables for % Lower Confidence Bound on Yield. Tables for 50, 60, 70, 80, and 90% lower confidence bound on yield are provided. The \times % lower confidence bound is given in the table as a function of the failure rate ratio $\frac{\overline{\lambda}_D}{\lambda_O}$, and the product of failure rate of a defect $(\overline{\lambda}_D)$ and the time duration of the screen (T). Failure rate ratios $\frac{\overline{\lambda}_D}{\lambda_O}$ ranging from .1 to 1, in increments of .1 and from 2 to 60 or more, in increments of 1 and 10 are used in the tables.

| | TABLE 3.19 SU PERCENT LUWER CORFIDENCE BOUND ON YIELD $\frac{1}{\lambda_{D}}$ (.1-1.0) | | | | | | | | | | | | |
|------------------|--|--------------|--------------|---------|--------|--------------|-------------------|------|-----------|--------------|--|--|--|
| | | | | Failure | e Rate | Ratio X | 0 ^{/1} 0 | | | | | | |
| ٦ _۵ ۳ | 0.10 | 0.20 | 0.30 | 0.40 | 0.50 | 0.60 | 0.70 | 0.80 | 0.90 | 1.00 | | | |
| 0.1 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | | | |
| 0.2 | 0.25 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | | | |
| 0.4 | 1.00 | 0.54 | 0.14 | 0.07 | 0.05 | 0.04 | 0.03 | 0.03 | 0.02 | 0.02 | | | |
| 0.6 0.7 | 1.00 | 1.00 | 0.69 | 0.38 | 0.26 | 0.21 0.33 | 0.17 | 0.15 | 0.14 0.22 | 0.13 0.21 | | | |
| 0.8 0.9 | 1.00 | 1.00 | 1.00 | 0.78 | 0.56 | 0.45 0.58 | 0.39 | 0.35 | 0.32 | 0.29 | | | |
| 1.0 | 1.00 | 1.00 | 1.00 | 1.00 | 0.84 | 0.69 0.79 | 0.60 | 0.54 | 0.50 | 0.47 | | | |
| 1.2 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 0.88 0.95 | 0.78 | 0.71 | 0.66 | 0.62 | | | |
| 1.4 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 0.91 | 0.83 | 0.78 | 0.74 | | | |
| 1.6 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 0.93 | 0.88 | 0.84 | | | |
| 1.8 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 0.99 | 0.94 | 0.91 | | | |
| 1.9 2.0 | 1.00 | 1.00 1.00 | 1.00 1.00 | 1.00 | 1.00 | 1.00 1.00 | 1.00 | 1.00 | 0.99 | 0.95 | | | |

| TABLE 5.19 | 90 PERCENT | LOWER | CONFIDENCE | BOUND | ON | YIELD | $\frac{\lambda_0}{\lambda}$ (.1-1.0) |
|------------|------------|-------|------------|-------|----|-------|--------------------------------------|
|------------|------------|-------|------------|-------|----|-------|--------------------------------------|

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| TABLE 5.20 | 90 PERCENT LOWER CONFIDENCE BOUND ON YIELD | $\frac{\lambda_{D}}{\lambda_{O}}$ (i-60) |
|------------|--|--|
|------------|--|--|

| | Failure Rate Ratio $\frac{\lambda_D}{\lambda_O}$ | | | | | | | | | | | | |
|------------------|--|------|------|------|------|------|------|------|------|-------|-------|-------|------------------|
| ī _o ī | 1.00 | 2.00 | 3.00 | 4.00 | 5.00 | 6.00 | 7.00 | 8.00 | 9.00 | 10.00 | 20.00 | 40.00 | 60.00 or More |
| .0 | 0.47 | 0.35 | 0.32 | 0.30 | 0.29 | 0.29 | 0.28 | 0.28 | 0.28 | 0,28 | 0.27 | 0.27 | 0.26 |
| .1 | 0.55 | 0.42 | 0.38 | 0.36 | 0.35 | 0.35 | 0.34 | 0.34 | 0.34 | 0.33 | 0.33 | 0.32 | 0.32 |
| .2 | 0.62 | 0.48 | 0.44 | 0.42 | 0.41 | 0.40 | 0.40 | 0.40 | 0.39 | 0.39 | 0,38 | 0.38 | 0.37 |
| .3 | 0.69 | 0.54 | 0.50 | 0.48 | 0.47 | 0.46 | 0.45 | 0.45 | 0.45 | .0.44 | 0,43 | 0.43 | 0.43 |
| .4 | 0.74 | 0.59 | 0.55 | 0.53 | 0.52 | 0.51 | 0.50 | 0.50 | 0.50 | 0.49 | 0.48 | 0.48 | 0.47 |
| .5 | 0.79 | 0.64 | 0,60 | 0.57 | 0.56 | 0.55 | 0.55 | 0.54 | 0.54 | 0.54 | 0,53 | 0.52 | 0.52 |
| .6 | 0.84 | 0.68 | 0.64 | 0.62 | 0.61 | 0.50 | 0.59 | 0.59 | 0.58 | 0.58 | 0.57 | 0.56 | 0.56 |
| .1 | 0.87 | 0.72 | 0.68 | 0.66 | 0.64 | 0.64 | 0.63 | 0.63 | 0.62 | 0,62 | 0.61 | 0.60 | 0.60 |
| .8 | 0,91 | 0.76 | 0.71 | 0.69 | 0.68 | 0.67 | 0.67 | 0.66 | 0.66 | 0,66 | 0.65 | 0.64 | 0.64 |
| .9 | 0.93 | 0.79 | 0.75 | 0.73 | 0.71 | 0.71 | 0.70 | 0.70 | 0.69 | 0.69 | 0.68 | 0.67 | 0.67 |
| .0 | 0,95 | 0.82 | 0.77 | 0.75 | 0.74 | C.73 | 0.73 | 0.73 | 0.72 | 0.72 | 0.71 | 0.70 | 0.70 |
| .2 | 0.99 | 0.86 | 0.82 | 0.80 | 0.79 | 0.79 | 0.78 | 0.78 | 0.77 | 0,77 | 0,76 | 0.76 | 0.75 |
| .4 | 1.00 | 0.90 | 0.86 | 0.84 | 0,83 | 0.83 | 0.82 | 0.82 | 0.82 | 0,81 | 0.80 | 0.80 | 0.80 |
| . 6 | 1.00 | 0.92 | 0.89 | 0.88 | 0.87 | 0.86 | 0.86 | 0.85 | 0.85 | 0.85 | 0,84 | 0.84 | 0.83 |
| .8 | 1.00 | 0.94 | 0.92 | 0,90 | 0.89 | 0.88 | 0.88 | 0.88 | 0.88 | 0.87 | 0.87 | 0.87 | 0.86 |
| .0 | 1,00 | 0.96 | 0.93 | 0.92 | 0.91 | 0.91 | 0.91 | 0.90 | 0.90 | 0.90 | 0.89 | 0.89 | 0.89 |
| .5 | 1.00 | 0.98 | 0.97 | 0.96 | 0.95 | 0.95 | 0.95 | 0.94 | 0.94 | 0.94 | 0.94 | 0.93 | 0.93 |
| .0 | 1.00 | 0.99 | 0.98 | 0.98 | 0.97 | 0.97 | 0.97 | 0,97 | 0.97 | 0.97 | 0.96 | 0.96 | 0.96 |
| .0 | 1.00 | 1.00 | 1,00 | 0.99 | 0.99 | 0.99 | 0,99 | 0.99 | 0.99 | 0.99 | 0.99 | 0.99 | 0.99 |
| .0 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 0.99 | 0.99 |
| .0 | 1.00 | 1.00 | 1,00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 |

ž

0

0.60

0.00 0.04 0.15 0.30 0.48 0.65 0.80 0.93 1.00 1.00 1.00 1.00 1.00

0.70

0.00 0.03 0.12 0.25 0.40 0.55 0.68 0.80 0.90 0.98 1.00 1.00

1.00

0.50

0.00

0.06 0.19 0.39 0.61 0.99 1.00 1.00 1.00 1.00 1.00 1.00

<u>x</u> (.1-1.0)

0,90

 $\begin{array}{c} 0.00\\ 0.00\\ 0.03\\ 0.09\\ 0.20\\ 0.32\\ 0.44\\ 0.56\\ 0.56\\ 0.75\\ 0.82\\ 0.94\\ 0.98\\ 1.00\\ \end{array}$

0.80

0.00 0.03 0.10 0.22 0.35 0.48 0.61 0.72

0.81

0.95 1.00 1.00

0.00

0.02

0.09 0.18

0.29

0.41 0,52 0.62

0.78 0.84 0.89 0.93 0.97

| 19.129 C 19 |
|-------------|
| |
| |

















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TABLE 5.21 80 PERCENT LOWER CONFIDENCE BOUND ON YIELD

т₀г

0.1 0.2 0.3 0.4 0.5 0.6 0.7 0.8 0.9 1.0 1.1 1.2 1.3

1.4

1.5

0.10

0.00

 $\begin{array}{c} 1.00\\$

0.20

 $\begin{array}{c} 0.00\\ 0.06\\ 0.73\\ 1.00\\$

0.30

0.00

0.18

TABLE 5.22 80 PERCENT LOWER CONFIDENCE BOUND ON YIELD

Failure Rate Ratio,

0.40

0.00

0.09 0.29 0.57

| | | | | | Fai | lure R | ate Ra | t10 1 | <u>0</u> 0 | | | | |
|--|--|--|--|--|--|--|--|--|--|--|--|--|--|
| ٦ ₀ ۲ | 1.00 | 2.00 | 3.00 | 4.00 | 5.00 | 6.00 | 7.00 | 8.00 | 9.00 | 10.00 | 20.00 | 40.00 | 60.00 or More |
| 1.0 1.1 1.2 1.3 1.4 1.5 1.6 1.7 1.8 1.9 2.0 2.2 2.4 2.6 | 0.70 0.78 0.84 C.89 0.93 0.97 1.00 1.00 1.00 1.00 1.00 1.00 1.00 | 0.52 0.59 0.65 0.70 0.74 0.78 0.81 0.84 0.87 0.89 0.91 0.94 0.96 0.98 | 0.48 0.54 0.59 0.64 0.73 0.76 0.78 0.82 0.84 0.90 0.90 0.92 0.94 | 0.45 0.51 0.57 0.62 0.66 0.70 0.74 0.77 0.79 0.82 0.82 0.88 0.90 0.93 | 0.44 0.50 0.55 0.60 0.72 0.75 0.75 0.81 0.81 0.83 0.86 0.89 0.92 | 0.43 0.49 0.54 0.59 0.64 0.71 0.74 0.77 0.60 0.82 0.26 0.82 0.26 0.89 0.91 | 0.43 0.48 0.54 0.59 0.63 0.67 0.71 0.74 0.77 0.79 0.85 0.85 0.88 0.91 | 0.42 0.48 0.53 0.63 0.65 0.70 0.73 0.76 0.79 0.81 0.85 0.88 0.90 | 0.42 0.48 0.53 0.58 0.66 0.70 0.73 0.76 0.78 0.80 0.80 0.84 0.87 0.90 | 0.42 0.47 0.53 0.67 0.62 0.66 0.69 0.72 0.75 0.78 0.80 0.80 0.87 0.90 | 0.40 0.45 0.51 0.66 0.60 0.64 0.72 0.74 0.77 0.79 0.83 0.86 | 0.40 0.45 0.51 0.60 0.64 0.67 0.71 0.73 0.76 0.78 0.83 0.86 0.88 | 0.45 0.50 0.55 0.60 0.63 0.67 0.71 0.73 0.76 |
| 2.8 3.0 3.5 4.0 5.0 6.0 7.0 | 1.00 1.00 1.00 1.00 1.00 1.00 1.00 | 0.99 0.99 1.00 1.00 1.00 1.00 1.00 | 0.96 0.97 0.99 0.99 1.00 1.00 1.00 | 0.94 0.96 0.98 0.99 1.00 1.00 1.00 | 0.93 0.95 0.97 0.98 0.99 1.00 1.00 | 0.93 0.94 0.97 0.98 0.99 1.00 1.00 | 0.92 0.94 0.97 0.98 0.99 1.00 1.00 | 0.92 0.94 0.96 0.98 0.99 1.00 1.00 | 0.92 0.94 0.96 0.38 0.99 1.00 1.00 | 0.92 0.93 0.96 0.98 0.99 1.00 1.00 | 0.91 0.93 0.96 0.97 0.99 1.00 1.00 | 0.91 0.92 0.95 0.97 0.99 1.00 1.00 | 0.90 0.92 0.95 0.97 0.99 1.00 1.00 |

| | | Failure Rate Ratio $\frac{\overline{\lambda}_D}{\lambda_D}$ | | | | | | | | | | | |
|---|--|--|---|--|--|--|--|--|--|--|--|--|--|
| τ _ο τ | 0.10 | 0.20 | 0.30 | 0.40 | 0.50 | 0.60 | 0.70 | 0.80 | 0.90 | 1.00 | | | |
| 0.1 0.2 0.3 0.4 0.5 0.6 0.7 0.8 0.9 1.0 1.1 | 0.14 1.00 1.00 1.00 1.00 1.00 1.00 1.00 | 0.00 0.40 1.00 1.00 1.00 1.00 1.00 1.00 | 0.00 0.09 0.56 1.00 1.00 1.00 1.00 1.00 1.00 1.00 1.0 | 0.00 0.04 0.27 0.66 1.00 1.00 1.00 1.00 1.00 1.00 1.00 | 0.00 0.03 0.18 0.44 0.73 1.00 1.00 1.00 1.00 1.00 1.00 | 0.00 0.02 0.13 0.34 0.56 0.78 0.96 1.00 1.00 1.00 | 0.00 0.02 0.11 0.28 0.47 0.66 0.82 0.95 1.00 1.00 1.00 | 0.00 0.01 0.09 0.24 0.41 0.58 0.72 0.85 0.95 1.00 1.00 | 0.00 0.01 0.08 0.21 0.37 0.52 0.66 0.77 0.87 0.95 1.00 | 0.00 0.01 0.08 0.20 0.34 0.48 0.61 0.72 0.81 0.89 0.95 | | | |

| TABLE 5.23 | 70 PERCENT LOWER | CONFIDENCE SOUND | ON YIELD $\frac{\lambda_D}{\lambda_O}$ (.1-1.0) |
|------------|------------------|------------------|---|
| | | | |

| | | Failure Rate Ratic $\frac{\overline{\lambda_0}}{\lambda_0}$ | | | | | | | | | | | | |
|---|---|--|--|--|--|--|--|--|--|--|--|--|--|--|
| τ _ο τ | 1.00 | 2.00 | 3.00 | 4.00 | 5.00 | 6.00 | 7.00 | 8.00 | 9.00 | 10.00 | 20 .00 | 40.00 | 60.00 or More | |
| 1.0 1.1 1.2 1.3 1.4 1.5 1.6 1.7 1.8 | 0.89 0.95 1.00 1.00 1.00 1.00 1.00 1.00 1.00 1.0 | 0.66 0.72 0.77 0.81 0.85 0.98 0.90 0.92 0.94 0.96 0.97 0.99 1.00 1.00 1.00 1.00 1.00 1.00 | 0.60 0.66 0.71 0.75 0.82 0.84 0.87 0.90 0.92 0.94 0.96 0.97 0.98 0.99 1.00 1.00 1.00 | 0.57 0.63 0.72 0.76 0.79 0.82 0.84 0.86 0.90 0.92 0.94 0.96 0.97 0.98 0.99 1.00 1.00 | 0.56 0.61 0.66 0.70 0.77 0.80 0.82 0.85 0.87 0.88 0.91 0.93 0.95 0.96 0.97 0.98 0.99 1.00 | 0.55 0.60 0.65 0.79 0.71 0.81 0.84 0.86 0.87 0.90 0.92 0.94 0.95 0.96 0.98 0.99 1.00 1.00 | 0.54 0.59 0.64 0.72 0.75 0.75 0.81 0.83 0.85 0.87 0.90 0.92 0.94 0.95 0.96 0.98 0.99 1.00 | 0.53 0.59 0.63 0.68 0.71 0.75 0.78 0.80 0.82 0.84 0.86 0.90 0.91 0.93 0.95 0.96 0.98 0.99 1.00 | 0.53 0.58 0.63 0.71 0.74 0.77 0.80 0.82 0.84 0.86 0.89 0.91 0.93 0.94 0.96 0.97 0.99 1.00 1.00 | 0.53 0.68 0.67 0.71 0.74 0.77 0.79 0.82 0.84 0.85 0.89 0.91 0.93 0.94 0.95 0.97 0.97 0.90 1.00 | 0.51 0.66 0.61 0.65 0.72 0.75 0.78 0.80 0.82 0.88 0.90 0.92 0.93 0.95 0.97 0.98 0.99 1.00 1.00 | 0.50 0.56 0.64 0.68 0.72 0.74 0.77 0.79 0.82 0.87 0.89 0.91 0.93 0.94 0.97 0.98 0.99 1.00 1.00 | 0.50 0.55 0.60 0.64 0.68 0.71 0.74 0.77 0.79 0.81 0.83 0.86 0.89 0.91 0.93 0.94 0.96 0.99 1.00 1.00 | |

TABLE 5.24 70 PERCENT LOWER CONFIDENCE BOUND ON YIELD $\frac{X}{X_0}$ (1-60)

| | | Failure Rate Ratio $\frac{\lambda_{D}}{\lambda_{O}}$ | | | | | | | | | | | |
|-----|------|--|------|------|------|-------|------|------|------|------|--|--|--|
| DT | 0.10 | 0.20 | 0.30 | 0.40 | 0.50 | 0.60 | 0.70 | 0.80 | 0.90 | 1.00 | | | |
| 0.1 | 1.00 | 0.02 | 0.00 | 0.00 | 0.00 | 0.00. | 0.00 | 0.00 | 0.00 | 0.00 | | | |
|),2 | 1.00 | 1.00 | 0.32 | 0.15 | 0.10 | 0.07 | 0.06 | 0.05 | 0.04 | 0.04 | | | |
|).3 | 1.00 | 1.00 | 1.00 | 0.62 | 0.40 | 0.30 | 0.25 | 0.21 | 0.19 | 0.17 | | | |
| 5.4 | 1.00 | 1.00 | 1.00 | 1.00 | 0.79 | 0.60 | 0.50 | 0.43 | 0.38 | 0.35 | | | |
|),5 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 0.88 | 0.73 | 0.64 | 0.57 | 0.53 | | | |
|).6 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 0.93 | 0.82 | 0.74 | 0.68 | | | |
|).7 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 0.96 | 0.87 | 0.81 | | | |
| .8 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 0.98 | 0.91 | | | |
| .9 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 1.00 | 0.99 | | | |

| TABLE 5.25 | 60 PERCENT LOWER | R CONFIDENCE BOUND ON YIELD | $\frac{\lambda_{\rm D}}{\lambda_{\rm o}}$ (.1-1.0) |
|------------|------------------|-----------------------------|--|
|------------|------------------|-----------------------------|--|

TABLE 5.26 60 PERCENT LOWER CONFIDENCE BOUND ON YIELD $\frac{\overline{\lambda}_{D}}{\overline{\lambda}_{O}}$ (1-60)

| \int | | | | Fai | lure Re | ite Rat | io | | | | | | |
|--|--|--|--|--|--|--|--|--|--|--|--|--|--|
| Tot | 1.00 | 2.00 | 3.00 | 4.00 | 5.00 | 6.00 | 7.00 | 8.00 | 9.00 | 10.00 | 20.00 | 40.00 | 60.00 or More |
| 1.0 1.1 1.2 1.3 1.4 1.5 1.6 1.7 1.8 1.9 2.0 2.2 2.4 2.6 2.8 3.0 3.5 4.0 5.0 6.0 | $\begin{array}{c} 1.00\\$ | 0.78 0.83 0.97 0.91 0.95 0.97 0.99 1.00 1.00 1.00 1.00 1.00 1.00 1.00 | 0.71 0.76 0.80 0.83 0.86 0.99 0.91 0.92 0.94 0.95 0.96 0.99 1.00 1.00 1.00 1.00 1.00 | 0.68 0.73 0.77 0.80 0.83 0.86 0.99 0.99 0.99 0.99 0.99 1.00 1.00 1.00 | 0.66 0.71 0.75 0.78 0.81 0.84 0.86 0.90 0.91 0.92 0.94 0.96 0.97 0.98 0.99 1.00 1.00 1.00 | 0.65 0.69 0.73 0.77 0.80 0.83 0.85 0.87 0.89 0.90 0.91 0.93 0.95 0.96 0.97 0.98 0.99 1.00 1.00 | 0.64 0.68 0.73 0.76 0.82 0.84 0.86 0.88 0.89 0.91 0.93 0.94 0.97 0.97 0.97 0.97 0.99 1.00 1.00 | 0.63 0.68 0.72 0.75 0.78 0.81 0.83 0.85 0.87 0.89 0.90 0.93 0.94 0.95 0.96 0.97 0.99 0.99 1.00 1.00 | 0.63 0.67 0.71 0.75 0.81 0.83 0.85 0.87 0.88 0.90 0.92 0.94 0.95 0.96 0.97 0.98 0.99 1.00 1.00 | 0.62 0.67 0.71 0.74 0.80 0.83 0.85 0.86 0.88 0.88 0.89 0.92 0.93 0.95 0.96 0.97 0.98 0.99 1.00 1.00 | 0.90 0.92 0.94 0.95 0.96 0.98 0.99 1.00 1.00 | 0.92 0.93 0.95 0.96 0.97 0.98 0.99 1.00 | 0.92 0.93 0.95 0.96 0.97 0.98 0.99 1.00 |

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| | | Failure Rate Ratio $\frac{\lambda_{D}}{\lambda_{O}}$ | | | | | | | | | | |
|--|--|--|--|--|--|--|--|--|--|--|--|--|
| λ _D T. | 0.10 | 0.20 | 0.30 | 0.40 | 0.50 | 0.60 | 0.70 | 0.80 | 0.90 | 1.00 | | |
| 0.1 C.2 0.3 0.4 0.5 0.6 | 1.00 1.00 1.00 1.00 1.00 1.00 | 0.16 1.00 1.00 1.00 1.00 1.00 | 0.03 0.89 1.00 1.00 1.00 1.00 | 0.01 0.42 1.00 1.00 1.00 1.00 | 0.01 0.27 0.77 1.00 1.00 1.00 | 0.01 0.20 0.58 0.95 1.00 1.00 | 0.01 0.16 0.47 0.78 1.00 1.00 | 0.00 0.14 0.40 0.68 0.90 1.00 | 0.00 0.12 0.36 0.60 0.81 0.97 | 0.00 0.11 0.33 0.55 0.74 0.89 | | |

| TABLE 5.27 | 50 PERCENT | LOWER | CONFIDENCE | BOUND | ON | YIELD | ^x | (.1-1.0) |
|------------|------------|-------|------------|-------|----|-------|--------------|----------|
| | | | | | | | λ | |

| | | | λn |
|------------|---------------------|-------------------|------------------|
| TABLE 5.28 | 50 PERCENT LOWER CO | ONFIDENCE BOUND O | N YIELD 😴 (1-60) |

| | | | | | | | | _ | | 0 | | | |
|--|--|---|--|--|--|--|--|--|--|--|-------|--|----------------------|
| | | | | Fai | lure R | ate Ra | tio | $\frac{\overline{\lambda}_{0}}{\lambda_{0}}$ | | | | | |
| λ _ο τ | 1.00 | 2.00 | 3.00 | 4,00 | 5.00 | 6.00 | 7,00 | 8.00 | 9.00 | 10. 00 | 20.00 | 40.00 | 60.00 or More |
| $\begin{array}{c} 1.0\\ 1.1\\ 1.2\\ 1.3\\ 1.4\\ 1.5\\ 1.6\\ 1.7\\ 1.8\\ 1.9\\ 2.0\\ 2.4\\ 2.6\\ 3.0\\ 3.5\\ 4.0\\ 5.0\\ 6.0\\ \end{array}$ | 1.00 1.00 1.00 1.00 1.00 1.00 1.00 1.00 | 0.89 0.93 0.96 0.98 1.00 1.00 1.00 1.00 1.00 1.00 1.00 1.0 | 0.81 0.85 0.88 0.91 0.93 0.95 0.96 0.97 0.98 0.99 1.00 1.00 1.00 1.00 1.00 1.00 1.00 | 0.77 0.81 0.84 0.97 0.93 0.94 0.95 0.96 0.97 0.98 0.99 1.00 1.00 1.00 1.00 1.00 | 0.75 0.79 0.82 0.85 0.87 0.91 0.92 0.94 0.95 0.96 0.97 0.98 0.99 1.00 1.00 1.00 | 0.74 0.78 0.81 0.86 0.86 0.90 0.91 0.93 0.94 0.95 0.96 0.97 0.98 0.99 1.00 1.00 1.00 | 0.73 0.77 0.80 0.83 0.85 0.87 0.99 0.90 0.92 0.93 0.94 0.95 0.97 0.97 0.97 0.99 1.00 1.00 | 0.72 0.76 0.79 0.82 0.84 0.86 0.90 0.91 0.92 0.93 0.95 0.96 0.97 0.98 0.99 1.00 1.00 | 0.71 0.75 0.79 0.81 0.84 0.86 0.88 0.91 0.92 0.93 0.95 0.96 0.97 0.98 0.99 1.000 1.000 | 0.94 0.96 0.97 0.97 0.98 0.99 0.99 1.00 | 0.99 | 0.89 0.90 0.92 0.94 0.95 0.96 0.97 0.98 0.99 1.00 1.00 | 0.99 1.00 1.00 |

5.5 Procedure D - Cost Effectiveness Analysis

5.5.1 Objective. The objective of this procedure is to perform cost analyses so as to identify the screen selection and placement sequences from among many possible alternatives, which provide a cost effective screening program.

5.5.2 Methodology. Both fixed and recurring screening costs are identified for each candidate screening sequence determined from Procedure B. Costs are determined for each level of assembly, including a failure-free acceptance test at the system level. The total costs of screening and the number of defects to be eliminated are used to determine the cost per defect eliminated by screening. Comparison of the cost per defect eliminated by screening against a cost threshold of \$1000 is the criterion used for judging cost effectiveness.

5.5.3 Procedure Steps. The worksheet shown in Figure 5.6 or a similar aid should be used. Instructions for completing the worksheet follow.





COST WORKSHEET

| System/Project | |
|---|---|
| Prepared by | Date |
| Variable Scree Expected Assem Average Cost p | COST \$ |
| Variable Screen Expected Unit 1 Average Cost p Screening Repart | Cost |
| Variable Screen Expected System Average Cost prime | ST Cost |
| Total Variable Total Screening Total Expected Total Number of Total Screening | TS \$ Cost \$ Repair Cost \$ Fallout \$ Systems to be Produced. \$ Cost (add lines 7,8 and 9). \$ Cost per System (divide 12 by 11). \$ |
| COST PER DEFECT EL THRESHOLD COST | MINATED (Jivide line 12 by 10) |

Figure 5.6 Cost Analyses Worksheet

<u>Step 1 Fixed Screening Costs</u>. (lines 1) Determine the fixed screening costs for each level of assembly. These costs are one time expenditures necessary to conduct screening at a particular assembly level and include:

- Cost of screening facilities
- Cost of test equipment and fixtures
- Cost of screening program planning and the preparation of procedures
- Cost of training

Note that the cost of screening facilities, test end ment and fixtures should be apportioned to the program for which the cost analysis is to be performed. Costs of facilities and equipment which will also be used on other programs should be allocated in accordance with the internal cost accounting procedures of the manufacturer.

<u>Step 2 Variable Screening Costs</u>. (Lines 2) Determine the variable screening costs for the total number of items to be screened at each assembly level. These costs are recurring costs which are different for each level of assembly and which depend upon the number of items to During early production when be screened/rescreened and tested. defect density would be expected to be higher, a large driver of screening cost would be repair and rework costs. During late production when defect density would be expected to be lower, the primary driver would be the cost of labor to conduct the screens and their associated tests. The latter situation would be expected when failure-free screens and tests are employed at the system level. The costs to conduct failure-free acceptance tests and associated screens would thus be heavily dependent on the labor costs for screening and testing. Recurring costs include:

- Cost of labor to conduct screens and tests
- Cost of labor for screening program management

- Cost of labor to conduct failure analysis (if got already accounted for by MIL-Q-9858A section 3.5)
- Cost to record and analyze screening program data (if not already accounted for by MIL-STD-785B Task 104 requirements for a closed-loop Failure Reporting, Analysis and Corrective Action System (FRACAS).

<u>Step 3 Expected Fallout</u>. (Lines 3) Determine the expected fallout for the total number of items to be screened at each level of assembly and for each candidate screening sequence using Procedure B (Figure 5.4).

<u>Step 4 Average Cost of Repair</u>. (Lines 4) Establish the average inhouse cost of labor and materials to repair a failed item. These cost estimates are dependent on the type of equipment being screened, the manufacturer's repair and rework facilities and the level of assembly where the defect is found. When estimates are not available, an approximate value is given in the worksheet.

<u>Step 5 Screening Repair Costs</u>. (Lines 5) Calculate as indicated on the worksheet. (Multiply the expected fallout and the average cost of repair).

<u>Step 6 Screening Costs</u>. (Lines 6) Calculate as indicated on the worksheet. (Add lines 1, 2 and 5).

<u>Step 7 Total Fixed Costs</u>. (Line 7) Add the fixed costs for screening for each level of assembly. (Add all line 1 costs).

<u>Step 8 Total Variable Costs</u>. (Line 8) Add the variable costs for screening for each level of assembly (Add all line 2 costs).

<u>Step 9 Total Screening Repair Costs</u>. (Line 9) Add the screening repair costs for each level of assembly. (Add all line 5 costs).

<u>Step 10 Total Expected Fallout</u>. (Line 10) Add the expected fallout for each level of assembly. (Add all line 3 entries). This estimate represents the total number of defects precipitated and detected by a candidate screening sequence.

<u>Step 11 Total Number of Systems to be Produced</u>. (Line 11) Enter the total number of systems to be produced and/or exposed to stress screening.

<u>Step 12 Total Screening Costs</u>. (Line 12) Calculate as indicated on the worksheet. (Add lines 7, 8 and 9 entries).

<u>Step 13 Average Screening Cost Per System</u>. (Line 13) This is the screening cost per system obtained by dividing Line 12 by Line 11 entries.

<u>Step 14 Average Cost Per Defect Eliminated</u>. Calculate the in-house average cost to eliminate a defect in the factory by dividing Line 12 by Line 10 entries.

<u>Step 15 Threshold Cost</u>. (C_T) A threshold cost of \$1000 is used in the procedure. The threshold cost is related to the field cost of repair and should be viewed as a not-to-exceed cost.

<u>Step 16</u>. After completion of the worksheet cost analyses, a comparison of the cost threshold (C_T) and the cost per defect eliminated by the screening process (C_D) should be made. If $C_D > C_T$ then the planned screening process should be re-evaluated to determine alternative screening methods to reduce costs, so that $C_D \leq C_T$.

5.6 Procedure E - Monitoring, Evaluation and Control

<u>5.6.1 Objective</u>. This procedure is used to monitor, evaluate and control the screening and manufacturing process so as to assure achievement of goals on remaining defect density (yield). Objectives are to:

- (a) Obtain estimates of the defect density, based upon the observed screen fallout data, and establish whether the observed defect density falls within or outside of predetermined control limits.
- (b) Compare the observed part fraction defective with planning estimates to prioritize the need for corrective actions.
- (c) Determine and implement corrective actions to improve manufacturing and screening process capability. Four complementary procedures are used to accomplish the objective.

5.6.2 Procedure El - Quality Control Charts for Defect Control -Methodology. Consider a batch of screened items with the following data available:

- (a) Estimates of $\rm D_{IN}$ and TS in accordance with Procedure A & B.
- (b) Number of items in the batch.
- (c) Observed number of defects as fallout (F) from the screen.

Figure 5.7 illustrates the screened items and the parameters of interest. The screened items can represent boards, assemblies or units.

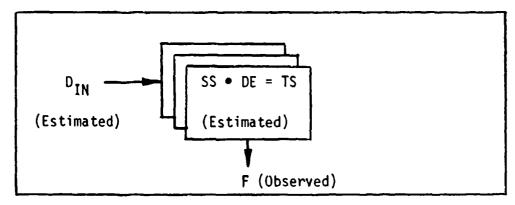


Figure 5.7 Screen Fallout Data Sample

The observed fallout can be above, below or within established control limits depending upon the degree to which the actual or "true" values of D_{IN} and TS differ from the planning estimates. The worst case situation, in terms of the effect on remaining defect density goals, is where ${\rm D}_{\rm IN}$ is higher than the planning estimate and TS is lower. ${\rm D}_{\rm IN}$ is reduced only through corrective actions which reduce future incoming defect density and thereby improves process capability. TS is increased by changing the screen type, stress levels or duration of the screen and by increasing the thoroughness of tests which are performed in conjunction with the screen. Table 5.29 illustrates the various possible conditions that can exist when the "true" values of ${\rm D}_{\rm IN}$ and TS are compared against planning estimates. The conditions are ranked according to severity and grouped into four categories depending upon whether outgoing defect density or costs are effected. The corrective actions required for each category are also shown in the table. Note that regardless of the outcome of the comparisons, corrective actions should always be taken to reduce $\mathsf{D}_{\ensuremath{\mathsf{IN}}}$ when opportunities to do so are presented.

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| and Test Strength (TS) V |
|--------------------------|
| nd Test Strength |
| nd Test |
| Б |
| ē |
| (^{NI} 0) |
| Density (D ₁ |
| Defect |
| Planned |
| ٧S |
| i of Actual vs Plann |
| n of |
| Comparison c |
| Table 5.29 |

| | | COMPARI SON | ISON | EFFECT ON | | ACTIONS | REQUIRED |
|-----------|--------|-------------|------|---------------------------------------|-----------|---------------------------|-------------------------|
| | | Actual | l vs | Remaining | Future | D _{IN} | 15 |
| | | Planned | ed | Defect Density Goal (D _R) | Screening | Reduce D _{IN} By | Changes to |
| CONDITION | _ | | | | Costs | Corrective Actions | Screen/Test |
| | | NIU | 15 | | | | |
| | æ | IH | 10 | | | | Increase |
| - | ۵ | Ŧ | ð | Higher than Expected | Increase | Essential | Strength |
| | U | ð | ΓO | | - | 4 | - |
| | 7 | | 1 | | | | |
| Ξ | • | = | | _ | | | |
| | e ا | 9 | 2 | lf Lower | | | |
| | 4- | 0K | IH | | • | • | Reduce |
| 111 | 5 | LO L | ŏ | Lower than Expected | Reduce | By Opportun!ty* | Strength |
| | 2 | ٢٥ | IH | | | | |
| 2 | • | ð | Ň | Likely to be Achieved | Reduce | By Opportunity* | No Change |
| | | | | | | | or Eventually Reduce |
| | | | | | | | |

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Corrective actions should always be taken when the opportunity presents itself and the costs to take actions are reasonable

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5.6.3 Procedure Steps.

<u>Step 1</u>. Using the observed number of defects and the quantities of parts and interconnection by type (from Procedure A) for the batch of screened items, calculate an observed part fraction defective for each part and interconnection type in the screened items.

<u>Step 2</u>. Calculate also the observed defect density using the relations:

$$P_{0} = \frac{n_{1} p_{1} + n_{2} p_{2} + \cdots + n_{k} p_{k}}{\sum_{i=1}^{k} n_{i}}$$

and:
$$\sum_{i=1}^{k} n_i = N$$
 $D_0 = NP_0$

Where: P = observed average part fraction defective per item

N = the total number of parts and interconnections
 per item

- n; = number of parts/interconnections of type i per item (i.e., diodes, transitors, hand soldered connection etc.)
- p = observed part fraction defective calculated
 from step 1 for each part type

 $D_0 = observed$ defect density per item.

<u>Step 3</u>. Rank the observed part fraction defective for each part and interconnection type i from the largest to the smallest.

<u>Step 4</u>. Determine D_{IN} , the expected incoming defect density for the batch of items subjected to the screen (from Procedure A).

<u>Step 5</u>. Compare the part fraction defective based upon observed fallout, calculated from step 1 against the planning estimates obtained from Procedure A, to determine those part types which show the largest (statistically significant) difference from planning estimates. The comparison should be based on a statistical test of significance which takes into account the sample size. Values of observed part fraction defective which exceed 1000 PPM should be specifically cited for corrective action.

<u>Step 6</u>. Determine the cause(s) for those part types showing the greatest differences and the corrective actions necessary to eliminate the sources of the defects from the product or process. Corrective actions might include, rejection of a suspect lot of parts, changing vendors, rescreening of the parts at incoming or changes to the manufacturing process.

<u>Step 7</u>. Prepare three control charts with the following trial values for the mean and standard deviation, $D_{IN} \stackrel{+}{=} 3 D_{IN}$, $D_{IN} \cdot TS \stackrel{+}{=} 3 D_{IN} \cdot TS$ and D_{IN} (1-TS) $\stackrel{+}{=} 3 D_{IN}$ (1-TS). Illustrations of the charts are shown in Figure 5.8 through Figure 5.10 with values of $D_{IN} = 3$ and TS = .7 as an example. The charts are shown separately, but note that the control limits overlap.

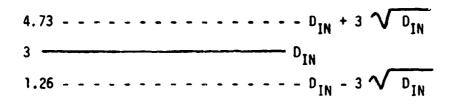
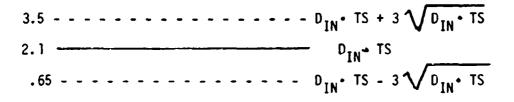
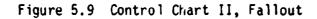


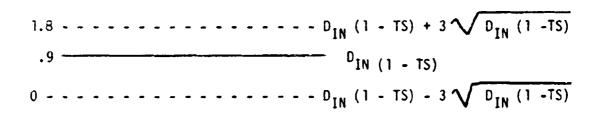
Figure 5.8 Control Chart I, Incoming Defect Density

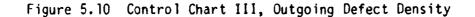




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<u>Step 8</u>. Plot and compare the observed value of observed defect density for the first and all subsequent batches of screened items as follows:

- (a) Values of D_0 which exceed the upper control limit (e.g. 4.73) on Chart I, <u>clearly</u> indicates that the planning estimates for incoming defect density (D_{IN}) were too low, without having to consider TS. Note that the trial values for the mean and standard deviation, for Chart I, presume perfect screens and tests (i.e., TS = 1). When the condition is evident from data, corrective actions must be taken to reduce D_{IN} , go to steps 5 and 6.
- (b) Values of the observed defect density which fall above the upper control limit of Chart II indicate the following possible conditions from Table 5.29:

| Conditio | n ^D IN | TS |
|----------|-------------------|---|
| I b | HI | OK |
| II d | HI | ні |
| III f | ОК | HI |
| Ιa | HI | LO L Hi values would be much |
| IV h | LO | OK HI HI LO Hi values would be much HI higher than expected |

Regardless of which of the conditions actually exist, the question of upmost concern is: Is the incoming defect density (D_{IN}) higher than planned or expected? Procedures E2, E3, or E4 should be used to address the question.

(c) Values of the observed defect density which fall below the lower control limit of Chart II indicate the following possible conditions from Table 5.29:

| Condition | DIN | TS | |
|-----------|-----|-----|--------------------------|
| Ic | ТОК | LO | |
| II e | LO | LO | |
| III g | LO | 0K | |
| Ιa | HI | LOZ | LO values would be |
| III h | LO | нт | much lower than expected |

Regardless of the condition which actually might exist the question of upmost concern is: Is the test strength lower than expected? Procedures E2 or E3 should be used to address the question.

(d) Values of the observed defect density which fall within the control limits of Chart II indicate the following possible conditions from Table 5.29:

| Condition | D _{IN} | TS |
|-----------|-----------------|----|
| IV 1 | OK | 0K |
| Ia | HI | LO |
| III h | LO | HI |

Regardless of the condition which actually might exist the questions of upmost concern are: Is the incoming defect density higher than expected and is the test strength lower than expected? Procedures E2 or E3 should be used to address the question.

<u>Step 9</u>. Evaluate which of the possible conditions which might exist in Step 8 by estimating the parameters of CDE model as outlined in Procedure E2 and E3.

<u>Step 10</u>. Control Chart III represents the outgoing defect density as a function of the planning estimates, D_{IN} and TS, at a given assembly level. Chart III also represents the incoming defect density at the next assembly level of screening, disregarding those defects which are newly introduced into the product at the next assembly level. Steps 1 through 9 are repeated at the next assembly level of screening, but with planning estimates of D_{IN} and TS. A repeat of Step 8a of the procedure at the next assembly level will provide some verification that estimates of D_{IN} and TS obtained from previous screening at the previous lower assembly level were correct.

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5.6.4 Procedure E2 - Use of CDE Model to Estimate D_{IN} and TS - Methodology. Obtain estimates of D_{IN} and TS from the screening data and compare them with the results of control chart methods, (Procedure E1). Determine the appropriate corrective action. The corrective actions might include; increasing or reducing TS or D_{IN} , depending upon the outcome of the comparison as indicated by Table 5.29 and the results of Procedure E1.

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<u>5.6.5 Procedure Steps</u>. The results from several batches of screened items will be needed to perform the following:

<u>Step 1</u>. Based upon time-to-failure or cycles to failure data, obtained from the screening fallout over time, estimate the parameters of the Chance Defective Exponential Model. Care should be exercised in using only part or interconnection (workmanship) fallout data for the analyses.

<u>Step 2</u>. Compare the estimates obtained from Step 1 with the planning estimates of D_{IN} and TS and the results of the Control Chart calculations of Procedure El. Establish which of the conditions of Table 1 exist and take the necessary actions to assure that the remaining defect density goals are achieved or to make the screening program more cost effective.

<u>Step 3</u>. When repeated estimates of the model parameters are made for several batches of screened items, and the estimates indicate significant differences from initial planning estimates, a re-evaluation of the screening program should be made.

<u>Step 4</u>. Change the screening regimen, as appropriate, to reflect the new estimates of the screening process variables so as to assure achievement of program objectives.

<u>Step 5</u>. Repeat Steps 1 through 4 interatively, on subsequent batches of screened items, at each level of assembly.

<u>5.6.6 Procedure E3 - Graphical Plotting - Methodology</u>. A graphical technique for estimating the parameters D_{IN} and $\overline{\lambda}_D$ of the CDE model can also be used. The technique uses the failure rate function of the CDE model.

$$\lambda_{s}(t) = \lambda_{0} + D \overline{\lambda}_{D} e^{-\overline{\lambda}_{D}t}$$

At lower assembly levels, the MIL-HDBK-217 predicted failure rates, λ_0 for the assembly will be very small because of the relatively small number of parts. In the interest of obtaining rough estimates of D_{IN} and λ_D , λ_0 can be assumed to be zero.

$$\lambda_{s}(t) \simeq D\bar{\lambda}_{D} e^{-\bar{\lambda}_{D}t}$$

Taking logarithms we have:

$$\ln \lambda_{s}(t) \simeq \ln (D \bar{\lambda}_{D}) - \bar{\lambda}_{D}t$$

By plotting observed values of $\lambda(t)$ on semi-log graph paper, estimates of D_{IN} and $\overline{\lambda}_D$ can be obtained as illustrated in Figure 5.11.

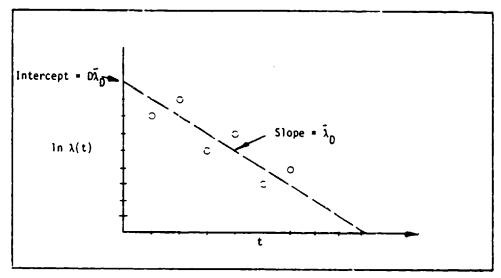


Figure 5.11 Failure Rate vs Time

5.6.7 Procedure Steps. Screening results for a batch of screened items and the time-to-failure for each defect must be available to perform the following:

<u>Step 1</u>. Group the fallout data into discrete intervals of time $(0_1, t_1)$, (t_1, t_2) . For temperature cycling screens a convenient grouping would be by each temperature cycle. Groupings for vibration screens might be in two minute time intervals.

<u>Step 2</u>. Estimate the instantaneous failure rate for each time interval using:

$$\lambda_{j} = \frac{F_{j}}{M_{j}(t_{j+1} - T_{j})}$$

Where: $F_i = fallout during the jth interval$

 λ_j = estimate of the instantaneous failure rate during j^{th} interval.

$$M_j = number of items which survived the (j-1) interval.$$

<u>Step 3</u>. Plot the estimates of λ_j versus time on a log-linear scale and fit a straight line to the data points.

<u>Step 4</u>. Determine the slope of the fitted line. The slope of this line provides an estimate of $\overline{\lambda}_{D}$.

<u>Step 5</u>. Determine the y intercept point for the fitted line. The y intercept provides and estimate of $D\overline{\lambda}_n$.

Step 6. Divide the y intercept $D\bar{\lambda}_D$ by the slope $\bar{\lambda}_D$ to obtain an estimate of $D_{1N}.$

<u>Step 7</u>. Step 6 provides a conservative estimate of D_{IN} . The estimate should be divided by the detection efficiency of the test which was used in conjunction with the screen to obtain D_{IN} .

<u>Step 8</u>. Calculate the screening strength of the screen by substituting the estimated λ_D and the total time duration of the screen T into SS(T)

= 1 -e D^{1} . Note that screening strength and λ_{D} are independent of the test detection efficiency.

<u>Step 9</u>. Multiply the screening strength by the test detection efficiency to determine the test strength.

<u>Step 10</u>. Compare the results obtained against the planned values to determine which condition of Table 5.29 exists and the required corrective action.

5.6.8 Procedure E4 - 90% Control Intervals on Expected Fallout. The objective of this procedure is to determine if the expected number of defects in a batch of screened items, based upon planning estimates of D_{IN} , is consistent with the actual fallout.

5.6.9 Methodology. This method should be applied when the user, either through prior experience or use of experiments, has a high degree of confidence that test strength (TS) values are correct. The 90% control limits are based upon the Binomial distribution. The model assumes that the defects fallout from the screen with the same probability (i.e. test strength) and are independent of one another. Under these assumptions, the defect fallout from the screen has a Binomial distribution:

P (defect fallout = k) =
$$\binom{M}{k}$$
 TS^k (1-TS)^{M-k}

Where M = postulated or expected number of defects entering the
 screen
 TS = test strength
 k = 0, 1, 2, ..., M.

The upper 90% control interval limit (denoted by UL) and the lower 90% control interval limit (denoted by LL) are obtained by solving the following equations for UL and LL.

UL is the smallest integer such that:

$$\begin{array}{c} M \\ \Sigma \\ k=UL+1 \end{array} \begin{pmatrix} M \\ k \end{array} \end{pmatrix} TS^{k} (1-TS)^{M-K} < .05$$

LL is the largest integer such that:

$$\begin{array}{c} LL-1 \\ \Sigma \\ k=0 \end{array} \begin{pmatrix} M \\ k \end{pmatrix} TS^{k} (1-TS)^{M-k} < .05$$

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The 90% control interval is given by [LL, UL]. Values of [LL, UL] are provided in Table 5.30 as a function of the test strength (TS) and the expected number of defects (M). The expected number of defects entering the screen is accepted as long as the fallout lies between LL and UL.

| | | | | | Test | Strengt | h | | | |
|--------------------|---------|---------|---------|---------|---------|----------|----------|----------|----------|----------|
| Expected No. of | · · | | | . • | • | • • | | | | |
| Defects | 0.50 | 0.55 | 0.60 | 0.65 | 0.70 | 0.75 | 0.80 | 0.85 | 0.90 | 0.95 |
| 5 | 4 | 5 1 | 5 1 | 5 1 | 5 | 5 2 | 5 2 | 5 3 | 5 3 | 5 4 |
| 6 | 5 1 | 5 1 | 5 | 6 | 6 2 | 6 3 | 6 3 | 6 | 6 4 | 6 5 |
| ר | 6 1 | 6 2 | 6 2 | 6 2 | 7 3 | 7 3 | 7 4 | 7 | 75 | 7 6 |
| 8 | 6 | 7 | 7 | 7 | 8 | 8 | 8 | 8 | 8 | 8 |
| | 2 | 2 | 3 | 3 | 3 | 4 | 4 | 5 | 6 | 6 |
| 9 | 7 | 7 3 | 8 3 | 8 3 | 8 4 | 9 5 | 9 5 | 9 6 | 9 6 | 9 7 |
| 10 | 8 | 8 | 8 | 9 | 9 | 10 | 10 | 10 | 10 | 10 |
| | 2 | 3 | 3 | 4 | 5 | 5 | 6 | 7 | 7 | 8 |
| 11 | 8 | 9 | 9 | 10 | 10 | 10 | 11 | וו | וו | וו |
| | 3 | 3 | 4 | 4 | 5 | 6 | 6 | 7 | 8 | 9 |
| 12 | 9 | 9 | 10 | 10 | 11 | 11 | 12 | 12 | 12 | 12 |
| | 3 | 4 | 4 | 5 | 6 | 6 | 7 | 8 | 9 | 10 |
| 13 | 9 | 10 | ון | 11 | 12 | 12 | 13 | 13 | 13 | 13 |
| | 4 | 4 | 5 | 6 | 6 | 7 | 8 | 9 | 10 | 11 |
| 14 | 10 | 11 | 11 | 12 | 12 | 13 | 13 | 14 | 14 | 14 |
| | 4 | 5 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 |
| 15 | 11 | 11 | 12 | 13 | 13 | 14 | 14 | 15 | 15 | 15 |
| | 4 | 5 | 6 | 7 | 7 | 8 | 9 | 10 | 11 | 13 |
| 16 | 11 | 12 | 13 | 13 | 14 | 15 | 15 | 16 | 16 | 16 |
| | 5 | 6 | 6 | 7 | 8 | 9 | 10 | 11 | 12 | 14 |
| 17 | 12 5 | 13 6 | 13 | 14 8 | 15 9 | 16 10 | 16 11 | 17 12 | 17 13 | 17 14 |
| 18 | 12 | 13 | 14 | 15 | 16 | 16 | דר | 18 | 18 | 18 |
| | 6 | 6 | 7 | 8 | 9 | 10 | וו | 13 | 14 | 15 |
| 19 | 13 | 14 | 15 | 16 | 16 | 17 | 18 | 18 | 19 | 19 |
| | 6 | 7 | 8 | 9 | 10 | 11 | 12 | 13 | 15 | 16 |
| 20 | 14 6 | 15 7 | 16 8 | 16 9 | 17 | 10 12 | 19 13 | 19 14 | 20 16 | 20 17 |

TABLE 5.30 90 PERCENT CONTROL PROBABILITY INTERVALS

\$26259372677277727727709003

5.6.10 Procedure Steps. Screening results for a batch of screened items must be available to perform the following:

<u>Step 1</u>. Multiply the planning estimate for defect density for the item by the number of screened items in the batch to obtain the expected number of the defects.

<u>Step 2</u>. Determine the test strength for the screen/test which was used for the batch of items.

<u>Step 3</u>. Find the value of test strength in the upper row and the expected number of defects (from Step 1) in the left most column of Table 5.30. Find the 90% control limits in the Table corresponding to the two values.

<u>Step 4</u>. If the actual number of defects observed for the batch of screened items falls within the 90% control limits, then the planning estimate of incoming defect density is accepted as being reasonably correct.

<u>Step 5</u>. If the actual number of defects observed for the batch of screened items falls above the upper control limits, then corrective actions to reduce D_{IN} and/or to increase the test strength should be determined.

<u>Step 6</u>. If the actual number of defects observed for the batch of screened items falls below the lower control limit, then corrective actions to reduce the screening regimen should be determined.

Appendix A Stress Screening Mathematical Models

10. General. The fundamental objective of a stress screening program is to reduce the number of defects in a production lot of equipment to an acceptable level by use of cost effective screening regimens. As basic principles, one would like to be able to use strong screens and efficient tests, within prescribed cost constraints, which have a high probability of precipitating and detecting defects so as to achieve stated objectives. In order to transform these principles into quantitative procedures, it is necessary to define various measures and their relationships to the screening process. Mathematical models for predicting important screening variables and for relating them to field reliability goals are also needed. This Appendix discusses the mathematical definitions and relationships between quantities such as defect density, screening strength and test detection efficiency.

<u>10.1 Defect Density</u>. Under reasonable assumptions that the number of defects in a product is independently and identically distributed, the number of defectives in an equipment can be described by the Binomial Probability distribution, with parameters N and \overline{P} .

Where N = total number of parts in the equipment \overline{P} = average part fraction defective over all part types

A part, as defined herein, is any identifiable item within the product which can be removed or repaired, (e.g., discrete semiconductor, resistor, integrated circuit, solder joint, connector). For large N and small \overline{P} the Binomial can be approximated by the Poisson distribution with the parameter D = N \overline{P}

Where D = Defect Density (average number of defects per item)

The defect density $D = N\overline{P}$ can also be represented as:

$$D = N\overline{P} = \sum_{i=1}^{k} n_{i} p_{i} \qquad (A-1)$$

where: $n_{i} = quantity of each part type i$
 $p_{i} = fraction defective for each part type i$
 $k = number of different part types$

The procedures contained in Procedure A of Section 5, for obtaining planning estimates of defect density, are based upon the mathematical relations just described.

<u>10.2 Defect Density and Yield</u>. Given prior estimates of p_i , equation A-1 can be used to estimate D_{IN} , the incoming defect density before assembly screening, since N and n_i are known for the assemblies and equipment to be screened. The outgoing or remaining defect density D_R can be described in a similar manner, except that the p_i , of equation 1, would be interpreted as the remaining part fraction defective. In terms of a production lot of equipment D_{IN} and D_R can also be expressed as:

total # of defects introduced

D_{IN} =

total # of equipments in the Lot

total # of defects remaining

D_R

total # of equipments in the Lot

Without an ESS program, a production lot of equipments will contain defects which are introduced into the equipments as escapes from previous part level screens and by poor workmanship or manufacturing processes. The defects introduced are expressed quantitatively as the average number of defects per equipment D_{IN} or defect density. Using the Poisson probability distribution, the probability that an equipment is defective P(D) (i.e., contains one or more defects) is given by:

$$P(D) = 1 - e^{-D} IN$$
 (A-2)

The objective of an ESS program is to reduce D_{IN} to an acceptable level, say D_R , where D_R is defined as the average number of defects remaining per equipment at delivery to the customer. Reducing D_{IN} to D_R also reduces P(D) so that:

$$P(D) = 1 - e^{-D_R}$$
 (A-3)

The probability that an equipment is defect-free when offered for acceptance is called Yield and using A-3.

Yield = 1 -
$$P(D) = e^{-D_R}$$
 (A-4)

If the Yield is specified as a goal, then D_R can be determined by:

$$D_{p} = -\ln (Yield) \tag{A-5}$$

and used as an objective for which an ESS program can be planned, implemented and subsequently monitored and controlled. Both D_R and Yield are used in the guidebook Procedures B and E, as the quantitative goal of the ESS program.

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<u>10.3 Screening Strength</u>. The screening strength (SS) of a screen is expressed as the probability that the screen will precipitate a defect to a detectable state given that a defect susceptible to the screen is present. Expressed as a function of time, the screening strength is:

$$SS(t) = 1 - e^{-\overline{\lambda}_{D}t}$$
 (A-6)

- where: SS(t) = Screening strength associated with a given screen type for stress duration of time t
 - $\bar{\lambda}_{D}$ = average failure rate of a defect under a given set of stress conditions. Note that under the exponential assumption $\bar{\lambda}_{D}$ is constant. However, $\bar{\lambda}_{D}$ corresponds to a specific set of stress conditions, i.e., larger stresses correlate to larger but constant $\bar{\lambda}_{D}$'s.

Screening strength and defect failure rates for various screen types are given in Tables 5.14 through 5.18.

<u>10.4 Screening Strength and Yield</u>. Screens are used to precipitate defects to failure. Using the relationships described previously, D_{IN} the incoming defect density is acted upon by the screening strength SS(t) to produce fallout from the screen.

$$D_{IN} \cdot SS(t) = F (fallout)$$
 (A-7)

The remaining defect density after the last screen is applied is $\ensuremath{\mathbb{D}_R}$ and is given by:

$$D_{R} = D_{IN} \left(1 - SS(t) \right)$$
 (A-8)

A4

Substituting equation A-8 into equation A-4 gives the following expression for yield:

Yield = exp
$$\left[-D_{IN} (1 - SS(t)) \right] = e^{-D_R}$$
 (A-9)
Yield = exp $\left[-D_{IN} e^{-\lambda} D^t \right] = e^{-D_R}$

or

Equation A-9 provides one of the relations for determining failure-free acceptance test requirements which is discussed further in Appendix C.

<u>10.5 Test Detection Efficiency</u>. The Test Detection Efficiency (DE) is a characteristic of a test which is measured by the ratio of the number of possible failure modes detectable to the total number of failure modes. Detection efficiency is characterized as the probability of detection. Test detection efficiency is a measure used in Procedure B. Guidance for determining test detection efficiency are discussed in Section 4.10.3.4 of the guidebook.

<u>10.6 Test Strength</u>. The Test Strength (TS) is defined as the joint probability that a screen will precipitate a defect to a detectable state and that a test will detect the defect and is given by:

 $TS = SS \cdot DE$

10.7 Relationships Between D_R and D_{IN} . Fig A-1 provides a model of the production screening process flow which incorporates the previously defined quantities (D_{IN} , SS, DE, TS and D_R). Average rework costs to repair or replace defectives at each assembly level are also shown in the figure.

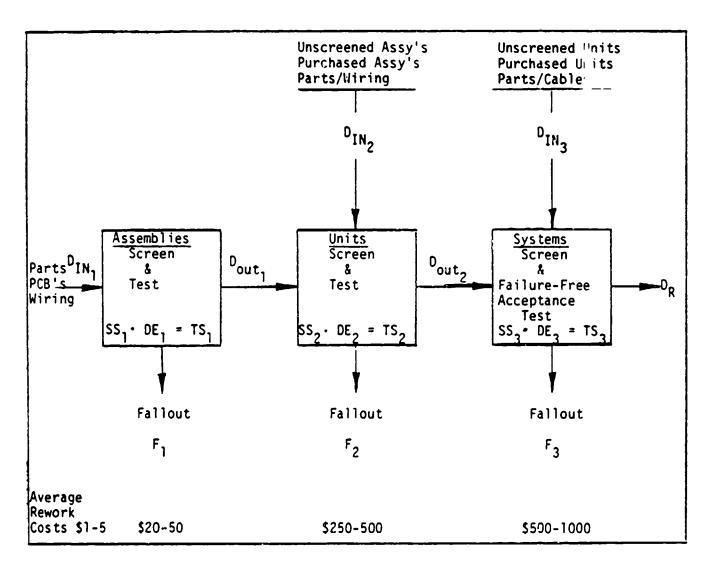


Figure A-1 Production Screening Process Flow Model

For a single screen i D_{out_i} is related to D_{IN_i} by: $D_{out_i} = D_{IN_i}$ (1-TS_i)

and the fallout F is given by:

 $F_i = D_{IN_i} (TS)_i$

For multiple screens at different assembly levels and assuming that screening is applied on a 100% basis at three assembly levels, D_R is related to D_{TN} by:

$$D_{R} = D_{IN_{1}} \frac{3}{i=1} (1-TS_{i}) + D_{IN_{2}} \frac{3}{i=2} (1-TS_{i}) + D_{IN_{3}} (1-TS_{3}) (A-10)$$

$$\frac{3}{D_{IN}} \frac{3}{i=1} D_{IN_{i}} \frac{1}{i=1} = \text{assembly stages}$$

In planning an ESS program and depending upon screen placement and the candidate screening sequences selected, variations of equation (A-10) are used to allocate Test Strengths to the various assembly levels, in an interative fashion, so as to achieve the required D_R . Procedure B in Section 5, "Screen Selection and Placement", is based upon use of the models and relationships just described. Screening and rework costs at each level of assembly must be taken into account as part of screen selection and placement process. Procedure D of Section 5, "Cost Effective Analysis" uses these costs in conjunction with the use of Procedure B.

<u>10.8 Cost Effectiveness of ESS Programs</u>. Without an ESS program, D_{IN} defects will remain in the equipment at delivery and eventually will fail early in field use due to the stresses naturally imposed by the operating environment. As the defects are weeded-out and assuming that no new defects are introduced during repair, and that no design problems exist, the reliability of the equipment can approach and perhaps exceed predicted (specified) values. The cost benefits to the government of finding and eliminating the defects in the factory versus the field depends in part, upon the cost per field repair. For example, if the average cost per field repair is \$5000 and the average cost to remove the defect in the factory is \$10,000 the screening program is clearly not cost effective. In planning an ESS program, a cost threshold C_T is compared against the cost of per defect removed in the factory C_D so that:

 $C_D < C_T$

AVA.

where:

Total Manufacturer's Screening Program Costs

с_D

Number of Defects Eliminated

And: C_{τ} = Average cost of a field repair (\$1000 is used)

For critical missions, where reliability is of overriding concern and cost is secondary, the cost threshold C_T is used as a baseline against which the cost of the screening program can be evaluated. The procuring activity must decide on how much the cost threshold should be exceeded in order to achieve high reliability requirements.

<u>10.9 Chance Defective Exponential Model (CDE)</u>. The CDE model is based upon the assumption that the population of parts within a lot of like equipments is comprised of two subpopulations, i.e., a main subpopulation of "good" parts and a much smaller subpopulation of defectives. The defectives contain major flaws which degrade with stress and time and are manifested as early-life failures. The failure rate of a defective part is several orders of magnitude greater than the failure rate of a "good" part. Therefore, relatively few defectives can dominate the reliability of the equipment during early product life.

Additional assumptions, terms and definitions which are used in the CDE model are:

(a) The number of defectives in an equipment is independent and identically distributed and the distribution is Binomial with parameters N and \bar{P} .

where: N = total number of parts in an equipment \overline{P} = average part fraction defective For large N and small \overline{P} the Binomial can be approximated by the Poisson distribution so that $D = N\overline{P}$ is the average number of defects per item (defect density).

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$$b = N\overline{P} = \sum_{i=1}^{k} n_i p_i$$

The defect density D is one of three parameters of the CDE model.

- (b) The failure distribution of the "good" or main subpopulation of parts in an equipment is exponential with parameter λ_0 and the reliability function is given by, $R_0(t) = e^{-\lambda_0 t}$. λ_0 is another parameter of the CDE model. The parameter λ_0 can also be expressed as $\lambda_0 = (N-D) \ \overline{\lambda}_G$, where $\overline{\lambda}_G$ is the average failure rate of a "good" part.
- (c) The failure distribution of a defective part is exponential with parameter $\bar{\lambda}_D$ and the reliability function is given by $R_D = e^{-\bar{\lambda}_D t}$. The parameter $\bar{\lambda}_D$ is defined as the average failure rate of a defective part under a particular stress environment. Note that $-\bar{\lambda}_D t$ when the CDE model is applied to a screen, $(1-R_D) = 1 e^{-\bar{\lambda}_D t} = SS(t)$, the screening strength. Note that the average failure rate of a defective part is much greater than the average failure rate of a "good" part. I.E. $\bar{\lambda}_D > \bar{\lambda}_G$ and with large defect densities the failure rate of the defective population can be greater than the population of "goods". I.E. $D\bar{\lambda}_D > (N-D) \bar{\lambda}_G$.

Given that a system contains n defective parts, the conditional reliability of the system R_s (t/n) is:

$$R_{s}(t/n) = R_{o}(t) \cdot R_{D}(t)$$
 n = 0,1,2..

Using the Binomial, the joint probability of survival and k defects present is:

$$R_{s}(t/k) \cdot P(k) = R_{o}(t) \sum_{k=0}^{N} \left[R_{D}(t) \right]^{k} {\binom{N}{p}}^{k} q^{N-k}$$

The number of defective parts can be approximated by the Poisson. with parameter $D = N\overline{P}$ so that the unconditional survival probability is given by:

$$R_{s}(t) = R_{0}(t) \Sigma R_{D}(t) \frac{(D) e}{k!}$$
(A-11)
k=0 k!

Performing the summation in A-11 gives the reliability function:

$$R_{s}(t) = R_{o}(t) e \qquad (A-12)$$

Using assumptions (b) $R_0(t) = e^{-\lambda_0 t}$ and assumption (c) $R_D(t) = -\overline{\lambda_0 t}$

$$R_{s}(t) = \exp\left[-\lambda_{o} t - D(1-e^{-\overline{\lambda}_{0}t})\right]$$
(A-13)

The failure rate for the system $\lambda_{c}(t)$ is given by:

$$\lambda_{s}(t) = -\frac{d}{dt} \ln R_{s}(t)$$
resulting in: $\lambda_{s}(t) = \lambda_{0} + D \overline{\lambda}_{D} e$
(A-14)

The probability density function for the system is given by:

$$f_{s}(t) = \lambda_{s}(t) \cdot R_{s}(t)$$

so that:

$$f_{s}(t) = \left[\lambda_{o} + D \overline{\lambda}_{D}e^{\overline{\lambda}_{D}t}\right] \exp\left[-\lambda_{o}t - D(1-e^{\overline{\lambda}_{D}t})\right] (A-15)$$

The expected number of failures for the system in time t is given by:

$$E_{s}(T) = \int_{0}^{T} t f_{s}(t) dt$$

which gives: $E_{s}(T) = \lambda_{0}T + D(1-e^{-\overline{\lambda}_{D}T})$ (A-16)

10.10 Relating D_R to Field Reliability and Failure Rate. Using the CDE model the reliability and failure rate of a system which has not had ESS exposure during manufacture is given by equations (A-13) and (A-14) as:

$$R_{s}(t) = exp\left[-\lambda_{o}t - D_{IN}(1-e^{-\overline{\lambda}_{o}t})\right]$$

 $\lambda_{s}(t) = \lambda_{0} + D_{IN} \overline{\lambda}_{D}^{e}$

 $\bar{\lambda}_{D}$ is viewed as the failure rate of a defective under the field stress conditions to which the system will be exposed and λ_{O} is the MIL-HDBK-217 predicted or specified failure rate for the system.

Given the same system which has been exposed to ESS during manufacture, then D_{IN} is reduced D_R and the other model parameters λ_0 and $\overline{\lambda}_D$ have the same interpretation as before. The failure rate function (equation A-14) both with and without an ESS program is illustrated in Fig A-2.

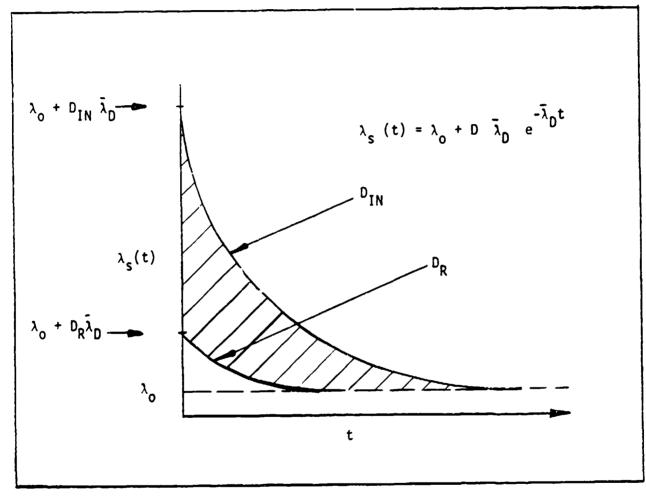


Figure A-2 Field Failure Rate vs Defect Density

The shaded area represents the defects removed from the product as a result of the ESS program conducted during manufacture.

Appendix B

Establishing Goals for Remaining Defect Density (Yield)

In establishing goals for remaining defect density, it is 10. General. necessary to relate measures which normally fall within the realm of statistical quality control (a manufacturing function) to measures which fall within the realm of reliability (a design function). The primary distinction between statistical quality control and reliability measures is that in the former, the measures are related to static populations and their sample statistics, whereas in the latter the measures are dynamically related to product performance over time in the field mission environment. For example, defect density is a static quality control measure and the failure rate or MTBF is a reliability measure. These two measures are used to arrive at values for remaining defect density. Two methods are described below, one makes use of the failure rate function of the CDE model and the other uses a lower bound on the part fraction defective of 50 PPM. Both methods relate defect density and failure rates and lead to reasonably consistent estimates for remaining defect density.

<u>10.1 Failure Rate Function - CDE Model</u>. Using the failure rate function for the CDE model and assuming a remaining defect density of D_R the system failure rate in the field environment as a function of time is given by:

$$\lambda_{s}(t) = \lambda_{o} + D_{R} \quad \overline{\lambda}_{D} e^{-\overline{\lambda}_{D} t}$$
(B-1)

Where:

 $\lambda_{c}(t)$ - failure rate of the system at time t

- λ_{o} specified failure rate for the system (non-screenable defects)
- ${\rm D}_{\rm R}$ average number of defects remaining per system at delivery
- $\overline{\lambda}_{D}$ average failure rate of a defect in the field environment

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For t=0, at the start of equipment life in the field environment equation (B-1) becomes:

$$\lambda_{s}(0) = \lambda_{0} + D_{R}\overline{\lambda}_{D}$$
(B-2)

The system failure rate at t=0 is thus seen to be the sum of λ_0 , the contractually specified or predicted failure rate and $D_R \overline{\lambda}_D$, the product of the field failure rate of a defect and the remaining defect density. When the defect density D_R =0, the failure rate of the system at the start of equipment life is equal to λ_0 , the specified failure rate.

The definition of a latent defect is not precise, i.e. a latent defect is an inherent or induced weakness which results in premature or early failure of the product in its intended use environment. What is an early or premature failure? To be slightly more precise one might say that latent defects represent a subpopulation in the equipment, whose average failure rate differs significantly from the main population of "good" parts. The average failure rate of a "good" part is in the range of approximately 1 failure per $10^6 - 10^7$ operating hours (MIL-HDBK-217). It would therefore be reasonable to assume that the average failure rate of defective in the field must be greater than one failure per thousand hours in order to be considered a prematurely failing latent defective. Failure'rates for defects which are in the range of 10^{-4} to 10^{-5} , are indeed possible, but they would be indistinguishable from the main population as early To summarize then, and in order to provide a slightly more failures. precise, but sill arbitrary definition: a latent defect can be defined as an inherent or induced weakness which has a failure rate in the field environment which is greater than 10^{-3} failures per hour.

Stress screening is designed to accelerate failure mechanisms of latent defects so that the defects can be precipitated to failure earlier than they would have failed in the intended use environment. Stated another way, stress screens are used to accelerate the failure rate of defectives.

Returning to the CDE failure rate equation (B-1) and dividing both sides of the equation by λ_0 we have:

$$\frac{\lambda_{s}(o)}{\frac{1}{\lambda_{o}}} = 1 + \frac{D_{R} \overline{\lambda_{D}}}{\lambda_{o}}$$
(B-3)

In equation (B-3), Letting:

$$\frac{\lambda_{s}(o)}{\lambda_{o}} = \frac{Failure Rate of the system at t = o}{Specified failure rate of the system} = R_{1}$$
and: $\overline{\lambda}_{D} = \frac{Failure rate of a defect}{\lambda_{o}} = \frac{R_{2}}{Specified failure rate of the system}$

Equation (B-3) then becomes: $R_1 = 1 + D_R R_2$

and solving for D_R we have

$$D_{R} = \frac{R_{1} - 1}{R_{2}}$$
 (B-4)

Given that a failure rate for the system, which is 10% larger than the specified failure rate at the start of product life is considered acceptable, then $R_1 = 1.1$. In addition, as was previously discussed, a reasonable range for the failure rate of a latent defect in the field environment is > 10⁻³ failures per hour. Selecting the upper value of 10⁻³ then the ratio R_2 can be calculated as a function of the specified failure rate, i.e., $R_2 = \frac{10^{-3}}{\lambda_0}$ Solving equation (B-4) for D_R , using $R_1 = 1.1$ and R_2

= $10^{-3}/\lambda_0$, results in the following table for remaining defect density as a function of the specified failure rate λ_0 .

Table B-1 Remaining Defect Density D_R vs Specified Failure Part

| $\lambda_0(\text{specified})$ | MTBF | D _R | |
|-------------------------------|--------|----------------|--|
| .1 | 10 | 10 | |
| .01 | 100 | 1 | |
| .005 | 200 | .5 | |
| .002 | 500 | .2 | |
| .001 | 1000 | .1 | |
| .0005 | 2000 | .05 | |
| .0002 | 5000 | .02 | |
| .0001 | 10000 | .01 | |
| .00001 | 100000 | .0001 | |
| | | | |

 $R_1 = 1.1$ $\overline{\lambda}_D = 10^{-3}$

A simple relation for determining remaining defect density as a function of λ_0 , can be noted from the data in Table B-1 as:

$$100 \lambda_0 = D_R \tag{B-5}$$

Table B-2 provides values for remaining defect density for values of $\overline{\lambda}_D$ which range from 10^{-1} to 10^{-4} . As will be shown in the next section, $\overline{\lambda}_D = 10^{-3}$, provides values of D_R which are consistent with goals on part fraction defective of 50 PPM.

Table B-2 Remaining Defect Density (D_R) vs Specified Failure Rate

| Specified Failure Rate | Defect Fai <u>l</u> ure Rate ^{\lambda_D} | | | | |
|------------------------------|--|------------|------------------|------------------|--|
| λ _o | 10 ⁻¹ | 10-2 | 10 ⁻³ | 10 ⁻⁴ | |
| .1 | .1 | 1 | 10 | 100 | |
| .01 .005 | .01 .005 | .1 .05 | .5 | 10 5 | |
| .002 | .002 | .02 .01 | .2 .1 | 2 | |
| .0005 | .0005 | .005 | .05 | .5 | |
| .0002 | .0002 | .002 | .02 | .2 | |
| .0001 | .0001 | .001 | .01 | .1 | |
| .00001 | .00001 | .00001 | .0001 | .01 | |

 $R_1 = 1.1$ and $\overline{\lambda}_D = 10^{-1}$, 10^{-2} , 10^{-3} , 10^{-4}

<u>10.2 Remaining Part Fraction Defective Goals in PPM</u>. Determining goals on remaining defect density for an equipment can also be approached by using yoals on the remaining fraction defective for the parts contained in the equipment. Recall from the discussions on the relationships between part and assembly fraction defective, in Section 4.6 that, if reasonable yields are to be achieved, part fraction defective levels must be < 100 PPM. The method described below uses a goal, for the part fraction defective, of 50 PPM to obtain estimates of remaining defect density. The calculations, shown below, are presented only to illustrate consistency with the results obtained in the previous section. The method should not be used, in practice, to determine remaining defect density goals. The CDE failure rate model of Section 10.1 should be used.

Using a series model, the failure rate of an equipment can be expressed as the sum of the failure rate of the electronic parts (diodes, transistors, etc.) and the interconnections (wire wrap, hand solder, etc.) which comprise the system. Assuming average failure rates for the parts and interconnections in the equipment, the system failure rate (λ_s) is given by:

| λ _s = | $N_p \overline{\lambda}_p + N_c \overline{\lambda}_c$ | (8-6) |
|------------------|---|-------|
|------------------|---|-------|

| wne | re | |
|-----|----|--|
| | | |

Number of electronic parts
 Number of interconnections
 Average failure rate of the parts
 Average failure rate of the interconnections

Reasonable values for $\overline{\lambda}_{p}$ and $\overline{\lambda}_{c}$ are, respectively, .5 x 10⁻⁶ and .0003 x 10⁻⁶ failures per hour (per MIL-HDBK-217). A review of prediction data for various equipment has shown that the average number of interconnections per part is about 3. Substituting in equation (B-6) we have:

 $\lambda_{\rm s} = .5N_{\rm p} + .0003 (3N_{\rm p}) \times 10^{-6}$

B6

$$\lambda_{s} = N_{p} (.5 + .0009) \times 10-6$$

 $\lambda_{s} = .50009 N_{p} \times 10^{-6}$

The contribution to the system failure rate of the interconnections can be seen to be negligible. The calculations in Table B-3, therefore, use $\lambda_s = .5 N_p \times 10^{-6}$ to estimate the system failure rate as a function of the number of parts used in the system. D_R is estimated using the relation $D_R = N_p$ (50PPM).

| N p | $\lambda_{\rm s} = N_{\rm p}(.5 \times 10^{-6})$ | MTBF | $D_{R} = N_{p} \cdot 50PPM$ |
|-------------|--|-------|-----------------------------|
| 20K | .01 | 100 | 1 |
| 10к | .005 | 200 | .5 |
| 5K | .0025 | 400 | .25 |
| 2K | .001 | 1000 | .1 |
| ТК | .0005 | 2000 | .05 |
| .2K | .0001 | 10000 | .01 |
| . 1K | .00005 | 20000 | .005 |

| Tab | le l | B-3 | Remaining | Defect | Density vs | Failure | Rate - | PPM Method |
|-----|------|-----|-----------|--------|------------|---------|--------|------------|
| | | | | | | | | |

The remaining defect density D_R calculated by the above method is consistent with the CDE model calculations in the previous Section 10.1, Equation B-5, therefore, provides a reasonable method for establishing goals on remaining defect density.

Appendix C

Failure Free Acceptance Test Derivation

<u>10.1 General</u>. A failure-free acceptance test applied at the system level provides a means of formally verifying that goals on remaining defect density (Yield) have been achieved. Yield can be verified by conducting a failure-free acceptance test of predetermined length T. The verified yield is defined as the conditional probability of having no defects given that the equipment survives a failure-free period of length T without failure.

10.2 Derivation. Using equation A-9 of Appendix A, the yield is given by:

Yield = $exp\left[-D exp\left(-\overline{\lambda}_{D}t\right)\right]$ (C-1)

where D = defect density at the start of failure-free period

$\overline{\lambda}_D$ = average failure rate of a defect under the stress conditions of the failure-free test.

A lower confidence bound on yield, based upon survival of a failure-free period of length T, can be computed by calculating an upper confidence bound on D. Following Brownlee (Ref 2.2.10) an upper confidence bound D^* is obtained by using the CDE model reliability function, equation B-13 of Appendix A, and solving:

$$\exp\left[-\lambda_{0}T - D^{*}(1 - \exp(-\overline{\lambda}_{D}T))\right] = 1 - CONF \qquad (C-2)$$



for D^* . The left side of equation (C-2) is the probability of surviving T according to the CDE model where λ_0 is the predicted or specified failure rate for the equipment and the other variables are as previously defined. CONF is the desired confidence level. The value of D^* is thus:

$$D^{*} = \frac{\ln \left[\frac{1}{(1 - CONF)} - \overline{\lambda}_{D}T/(\overline{\lambda}_{D}/\lambda_{0}) \right]}{1 - \exp(-\overline{\lambda}_{D}t)}$$
(C-3)

The upper confidence bound on D is then:

$$\overline{D} = Max (0, D^*)$$

and the lower confidence bound on yield is given by:

$$\exp\left[-\bar{D} \exp\left(\bar{\lambda}_{D} t\right)\right]. \qquad (C-4)$$

Tables 5.19-5.28, contained in Procedure D of Section 5, provide the x % lower confidence bounds on yield as a function of $\overline{\lambda}_D T$ and $\overline{\lambda}_D / \lambda_0$. The values in the tables were obtained by use of equations (C-3) and (C-4).

Successful application of a failure-free acceptance test is strongly dependent on accurate knowledge of the defect density at the start of the failure-free test and the screening strength of the screen. These values should be obtained from actual screening process results, using the monitoring and control methods outlined in Procedure E.

