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Volume 2, Chapters 5 and 6



DAMAGE TOLERANT DESIGN HANDBOOK

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Foreword

This report summarizes the results of a damage tolerant, material property data collection and reporting program conducted under USAF Contract F33615-91-C-5610. The work was sponsored by the Materials Directorate of Wright Laboratory with Mr. Jack Coate of the Systems Support Division serving as the project monitor. The technical effort was conducted between June 1991 and January 1994. The work was performed by the University of Dayton Research Institute under the general supervision of Dr. Joseph P. Gallagher with Dr. Alan P. Berens serving as Principal Investigator.

This final report comprises eight chapters which are presented in five volumes as follows:

<u>VOLUME</u>	<u>CHAPTER</u>	<u>DESCRIPTION</u>
1	1	Handbook organization and content
	2	Methods of calculation
	3	Alloy Steels
	4	Stainless Steels
2	5	Nickel Based Super Alloys
	6	Titanium Alloys
3	7	Aluminum 2000/6000 Series Alloys
4 & 5	8	Aluminum 7000/8000 Series Alloys

A detailed listing of the materials represented in the Handbook is contained in the preceding Table of Contents. In the body of the Handbook, the pages are numbered within chapters and the relevant portion of the table of contents is repeated at the beginning of each chapter.

CHAPTER 5

NICKEL BASED SUPER ALLOYS SECTIONS

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TABLE 5.0.1

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AVAILABLE DATA FOR NICKEL ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K_{Ic}	K_{Icc}	R Curve	da/dN	da/dt	K_{Iecc}	
ASTROLOY 901	Unspecified	Unspecified				15			
ASTROLOY P/M-H	2025F 3HR AC 1600F 8HR AC	Disk				8			
ASTROLOY P/M-W	2025F 4HRS AC 1600F 8HRS AC	Disk				8			
IN100	Unspecified	Unspecified				16			
		Forging				32			
		Forging				1			
IN100 P/M-G	2050F 2HRS OQ 1600F 0.67HR AC	Disk				6			
INCOLOY 901	Unspecified	Unspecified				1			
INCONEL 600	1585F .75HR AC	Plate				19			
INCONEL 625	MA	Plate				11			
INCONEL 718	Unspecified	Unspecified				4			
		1325F 8 HR FC TO 1150F HOLD 18 HR	Forging	1					
		1325F 8HRS FC TO 1150F	Forging				3		
		1325F 9 HR FC TO 1150F AT 100F/HR HOLD AT 1150F 8 HR AC	Forging	2					
		1750F 1HR AC 1325F	Plate				2		
INCONEL 718	1750F AC	Forged Bar				5			
		Sheet				1			
		Sheet				3			
		Plate				23			
		Forging				5			
	1750F AC 1325F	Forged Bar				5			

TABLE 5.0.1 (CONTINUED)
AVAILABLE DATA FOR NICKEL ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K_{Ic}	K_c	R Curve	da/dN	da/dt	K_{Isc}
INCONEL 718 (Cont)	1750F 1HR Q 1 325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR	Disk				15		
	1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR	Disk				30		
	1760F 1HR WQ 1325F 8HRS	Disk				5		
	1800F 1HR Q 1325F 8HRS FC TO 1150F HOLD 8HRS AC	Forging				4		
	1850F 1.5HR OQ 1360F 9HRS FC TO 1175F	Forged Bar						8
	1880F 1HR AC 1520F 8HR FC 1200F 16HR AC	Sheet				4		5
	1950F AC 1325F	Plate						
	ANNEALED 1800F 1 HR, AC, 1325F 8 HR, COOLED AT 100F PER HOUR TO 1150F, 1150F 8HR, AC	Plate				13		
	COLD ROLLED 30PERCENT AND AGED	Sheet		59				
	ST 1850F 1360 F 9HRS F/C 1175F	Forged Bar				11		
NASA IIB-7 P/M	ST-CW-A	Round Bar	3					
	STA	Forging	7			20		
		Round Bar	4					
	1650F 16HRS TO 2000F 1HR OQ	Disk				7		
P/M RENE 95	2080F 1HR AC 1600F 1HR AC	Disk				8		
	2100F 1HR SQ AT 1000F	Diak				4		
RENE 95 (H&F)	2000F 1HR SQ AT 1000F	Forging				4		
WASPALLOY	Unspecified	Unspecified				2		
		Forging				86		

TABLE 5.0.1 (CONCLUDED)

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AVAILABLE DATA FOR NICKEL ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K _{1c}	K _c	R Curve	da/dN	da/dt	K _{I_{secc}}
WASPALLOY (Cont)	1860F 2HRS 1350F 6HRS	Billet				1		
	1850F 2HRS 1600F 24HRS (FINE GS)	Billet				1		
	1875F 4HRS Oq 1550F 4HRS AC	Disk				9		
	2010F 2HRS 1350F 6HRS	Billet				1		
	2010F 2HRS 1600F 24HRS	Billet				1		

TABLE 5.0.2

PLANE STRAIN FRACTURE TOUGHNESS VALUES OF NICKEL BASE SUPER ALLOYS
AT ROOM TEMPERATURE

Alloy	Condition/ Heat Treatment	Product Form	Range of Product Thickness (in.)	K_{Ic} (Ksi \sqrt{in})											
				Specimen Orientation					S-L						
				L-T		T-L			L-T		T-L				
INCONEL 718	STA	Forging	3.00-3.30	Min Spec Thk	n	Mean	Std Dev	Min Spec Thk	n	Mean	Std Dev	Min Spec Thk	n	Mean	Std Dev
				1.50	2	1185	0.7	1.50	2	107.9	14.0

TABLE 5.0.4.1

1 of 1

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR NICKEL BASED ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: Unspecified STRESS RATIO: 0.05 - 0.1 FREQUENCY: 0.17 - 10. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
					2.5	5.0	10.0	20.0	50.0	100.0
					AK Level (Ksi/in)					
ASTROLOY 901	99	UNSPECIFIED	0.1	0.17				1.11	18.31	
INCONEL 625	MA	PLATE	0.05	10				2.07		
INCONEL 718	1750F 1HR AC 1325F	PLATE	0.05	10				1.65		
	1750F AC 1325F	SHEET	0.05	5				1.95		
WASPALLOY	1850F 2HRS 1350F 6HRS	FORGING	0.05	10				1.93	40.69	
	2010F 2HRS 1350F 6HRS	BILLET	0.1	10					10.46	
	2010F 2HRS 1600F 24HRS	BILLET	0.1	10					2.8	
		BILLET	0.1	10					9.56	

TABLE 5.0.4.2

1 of 1

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR NICKEL BASED ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: L-T STRESS RATIO: -0.2 - 0.8 FREQUENCY: 4. - 30. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-4} in/cycle)					
					ΔK Level (Kav/in)					
					2.5	5.0	10.0	20.0	50.0	100.0
INCONEL 718	1750F AC 1325F	PLATE	0.05	8.33				1.28	30.57	
	1950F AC 1325F	PLATE	0.05	8.33				1.05		
	ANNEALED 1800F 1 HR, AC, 1325F 8 HR, COOLED AT 100F PER HOUR TO 1150F, 1150F 8HR, AC	PLATE	-0.2	10				1.14		
			0.02	4				0.92	16.4	
			0.02	20				1.75		
	STA	FORGING	0.5	10				2.97		
			0.1	15-20				2.45	57.53	
			0.4	20				3.42		
	UNSPECIFIED	UNSPECIFIED	0.8	20-30			0.47			
			0.1	10			0.13	2.46	56.78	

TABLE 5.0.4.3

1 of 1

FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
 AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 FOR NICKEL BASED ALLOYS IN LAB AIR AT ROOM TEMPERATURE

ORIENTATION: T-L STRESS RATIO: -1. - 0.8 FREQUENCY: 5. - 30. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
					2.5	5.0	10.0	20.0	50.0	100.0
INCONEL 718	1750F 1HR AC 1325F	FORGED BAR	0.05	8.33-10				3.08	116.49	
			-1	5			0.69	29.84		
	STA	FORGING	0.1	15-20				2.19	83.07	
			0.4	20				3.16		
			0.8	20-30				0.47		

TABLE 5.0.4.4

1 of 1

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR NICKEL BASED ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: C-R STRESS RATIO: 0.0 - 0.05 FREQUENCY: 0.33 - 10. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-4} in/cycle)					
					ΔK Level (Ksi/in)	2.5	5.0	10.0	20.0	50.0
INCONEL 718	1750F AC 1325F	FORGED BAR	0.05	10				2.56	43.72	
	1760F 1HR WQ 1325F 8HRS	DISK	0.	0.33				3.39		
P/M RENE 95	2080F 1HR AC 1600F 1HR AC	DISK	0.	0.33				0.4	31.02	

TABLE 5.0.5

1 of 1

STRESS CORROSION CRACKING THRESHOLD DATA FOR NICKEL BASE ALLOYS AT ROOM TEMPERATURE								
Alloy	Condition/ Heat Treatment	Product Form	Specimen Orientation	K_{Icc} Ksi/\sqrt{in}				
				Environment				
				Shop Cleaning Solvent	Sump Tank Water	Martin- Marietta Refined Grade Hydrazine	Matheson- Coleman-Bell 97% Grade Hydrazine	Propellant Grade Hydrazine
INCONEL 718	1850F 1.5HR OQ 1360F 9HRS FC TO 1175F	Forged Bar	L-T	126(2)	133(2)			
			T-L		105(2)			
			S-L		93(2)			
		1880F 1HR AC 1520F 8HR FC 1200F 16HR AC	Sheet	---			79(2)	25.8

TABLE 5.1.1.1.2

1 of 1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 ASTROLOY 901 AT ROOM TEMPERATURE

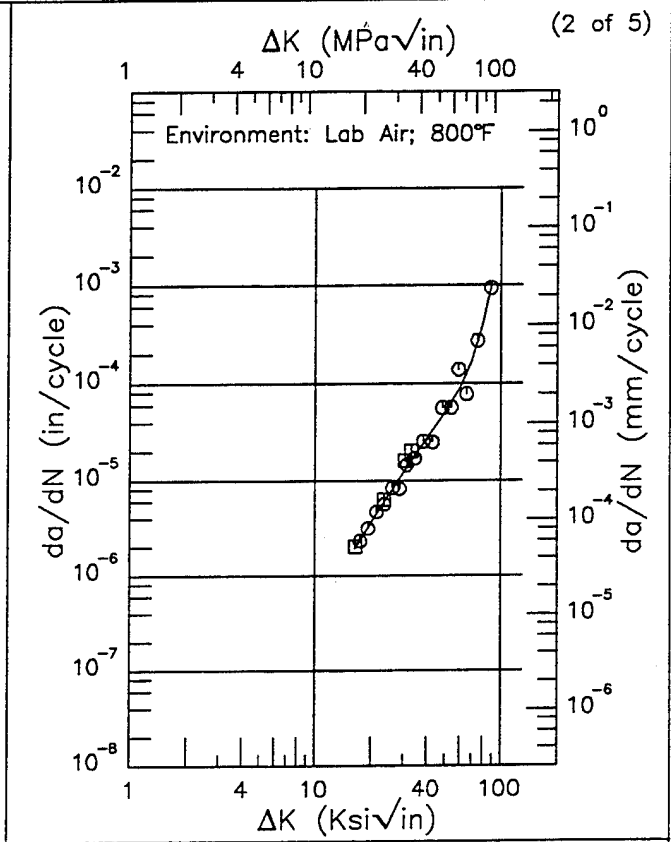
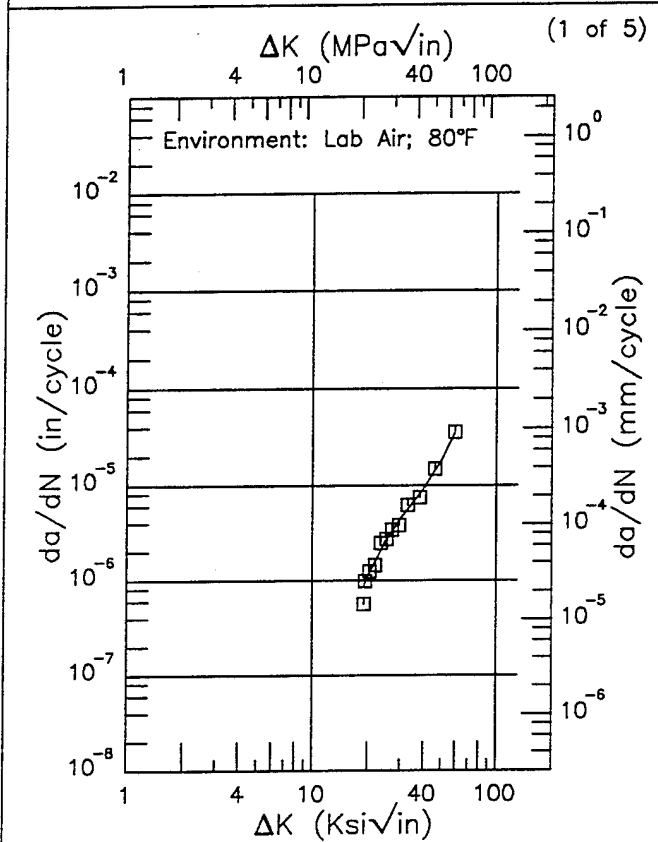
ORIENTATION: Unspecified ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi/in)					
UNSPECIFIED	UNSPECIFIED	0.1	0.17	2.5	5.0	10.0	20.0	50.0	100.0

ASTROLOY 901

Condition/Ht:
 Form:
 Specimen Type: SENT
 Orientation:
 Stress Ratio: 0.1

Yield Strength:
 Ult. Strength:
 Specimen Thk: 0.121 - 0.126 in.
 Specimen Width: 2.5 in.
 Ref: PW003



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
18.88 (min)	0.841
20.	1.11
25.	2.62
30.	4.44
35.	6.59
40.	9.30
50.	18.3
59.15 (max)	35.9

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
16.50 (min)	1.94
20.	3.60
25.	7.09
30.	12.1
35.	18.8
40.	27.4
50.	51.3
60.	88.1
70.	167.
80.	393.
88.23 (max)	948.

RMS $\%$ Error	Life Prediction Ratio Summary
12.50	

RMS $\%$ Error	Life Prediction Ratio Summary
21.01	

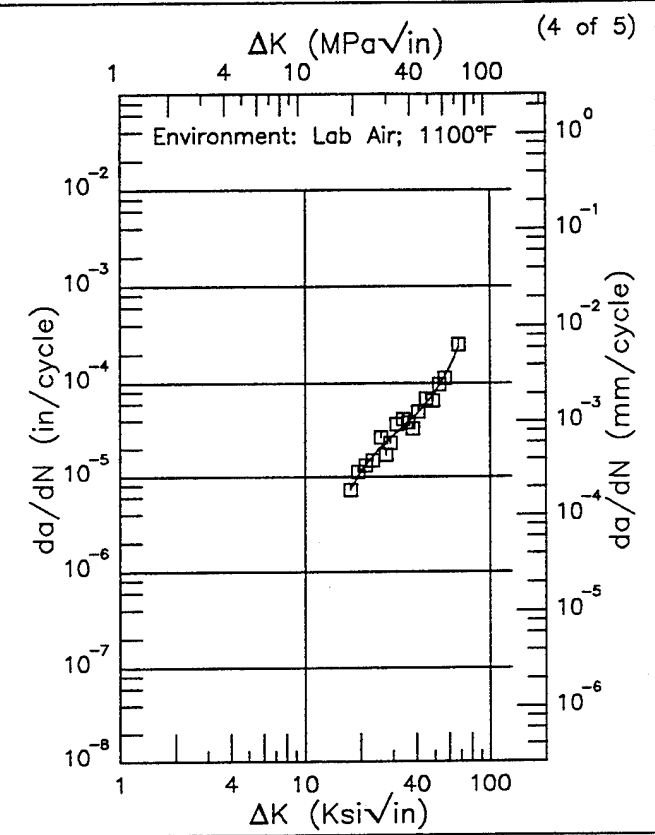
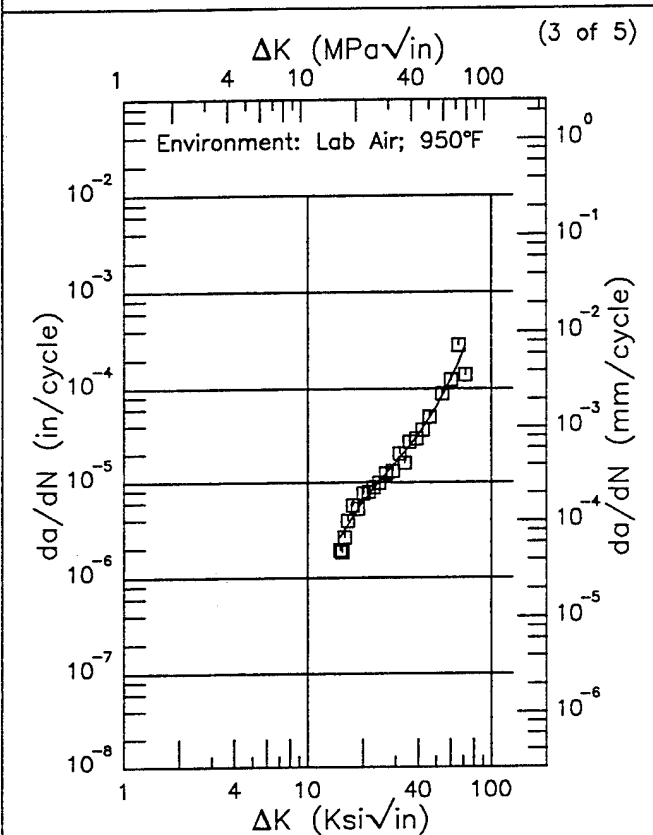
Figure 5.1.3.1.1

ASTROLOY 901

EF

Condition/Ht:
 Form:
 Specimen Type: SENT
 Orientation:
 Stress Ratio: 0.1

Yield Strength:
 Ult. Strength:
 Specimen Thk: 0.121 - 0.126 in.
 Specimen Width: 2.5 in.
 Ref: PW003



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
14.98 (min)	2.63
16.	3.28
20.	6.29
25.	10.8
30.	16.4
35.	23.4
40.	32.8
50.	63.5
60.	124.
70.	249.
71.72 (max)	281.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
17.59 (min)	7.56
20.	11.5
25.	19.7
30.	27.6
35.	36.2
40.	46.5
50.	79.0
60.	145.
67.44 (max)	241.

RMS % Error 20.98	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

RMS % Error 14.54	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

Figure 5.1.3.1.1 (Continued)

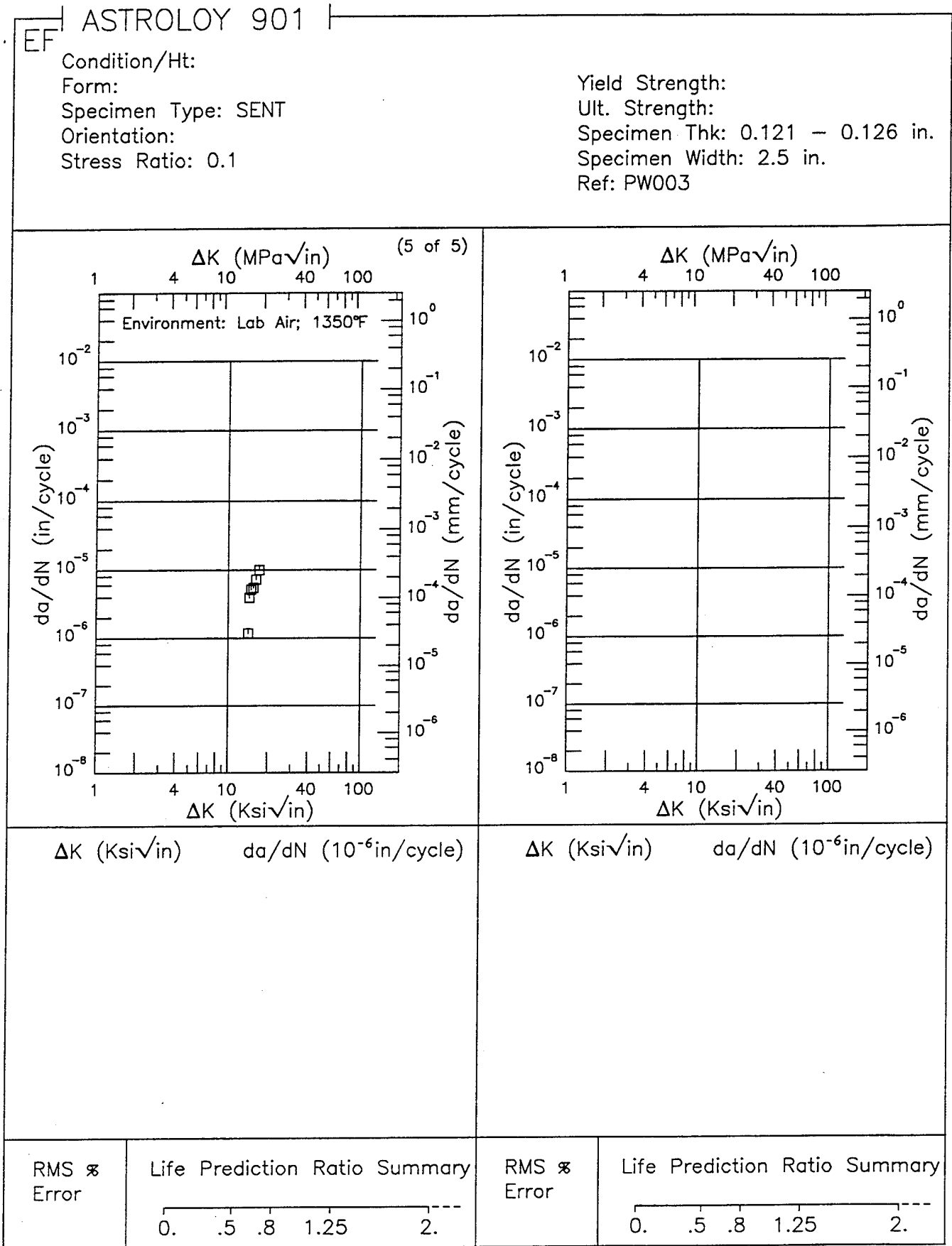


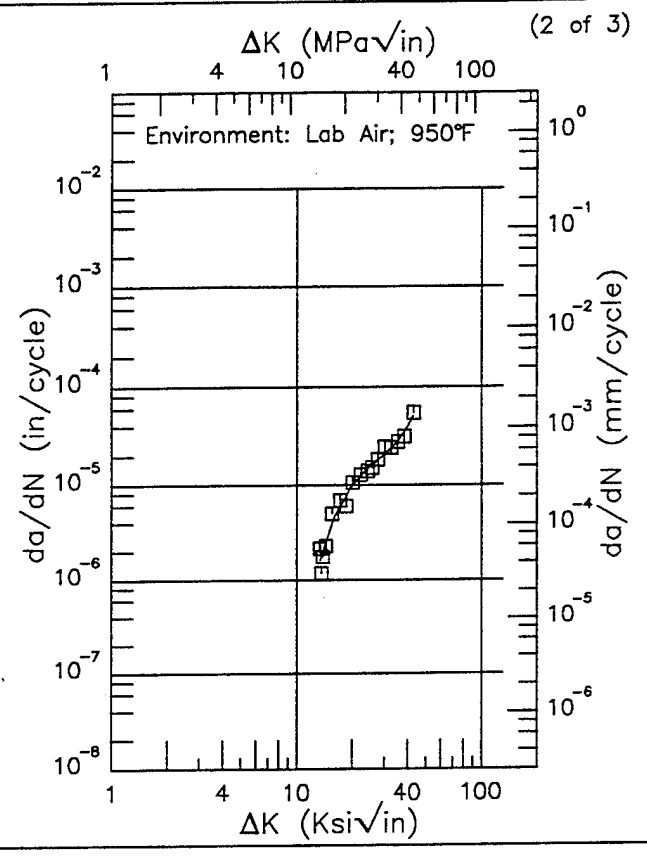
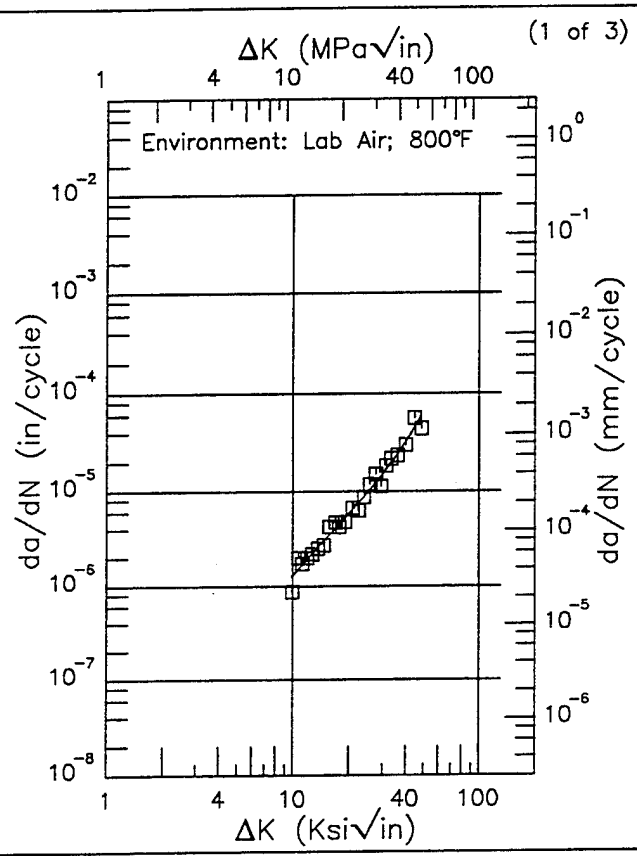
Figure 5.1.3.1.1 (Concluded)

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EF ASTROLOY 901

Condition/Ht:
 Form:
 Specimen Type: SENT
 Orientation:
 Stress Ratio: 0.5

Yield Strength:
 Ult. Strength:
 Specimen Thk: 0.12 - 0.124 in.
 Specimen Width: 2.5 in.
 Ref: PW003



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
9.87 (min)	1.27
10.	1.31
13.	2.34
16.	3.65
20.	5.91
25.	9.79
30.	15.2
35.	22.7
40.	32.9
48.57 (max)	59.3

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
13.38 (min)	1.60
16.	4.57
20.	10.0
25.	15.6
30.	20.7
35.	27.7
40.	39.7
42.89 (max)	50.9

RMS % Error 15.77	Life Prediction Ratio Summary
----------------------	-----------------------------------

RMS % Error 16.54	Life Prediction Ratio Summary
----------------------	-----------------------------------

Figure 5.1.3.1.2

ASTROLOY 901

EF

Condition/Ht:

Form:

Specimen Type: SENT

Orientation:

Stress Ratio: 0.5

Yield Strength:

Ult. Strength:

Specimen Thk: 0.12 - 0.124 in.

Specimen Width: 2.5 in.

Ref: PW003

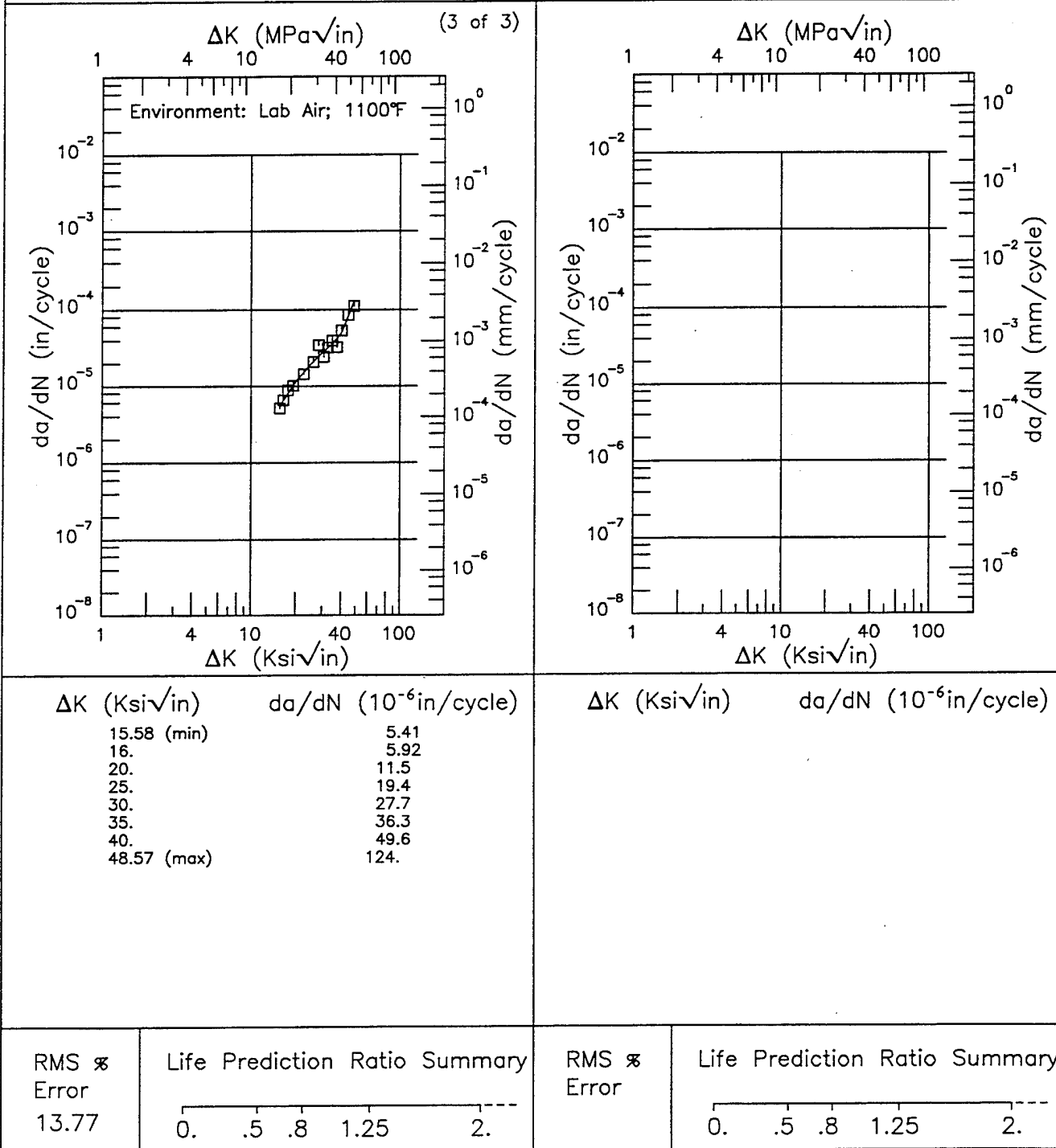
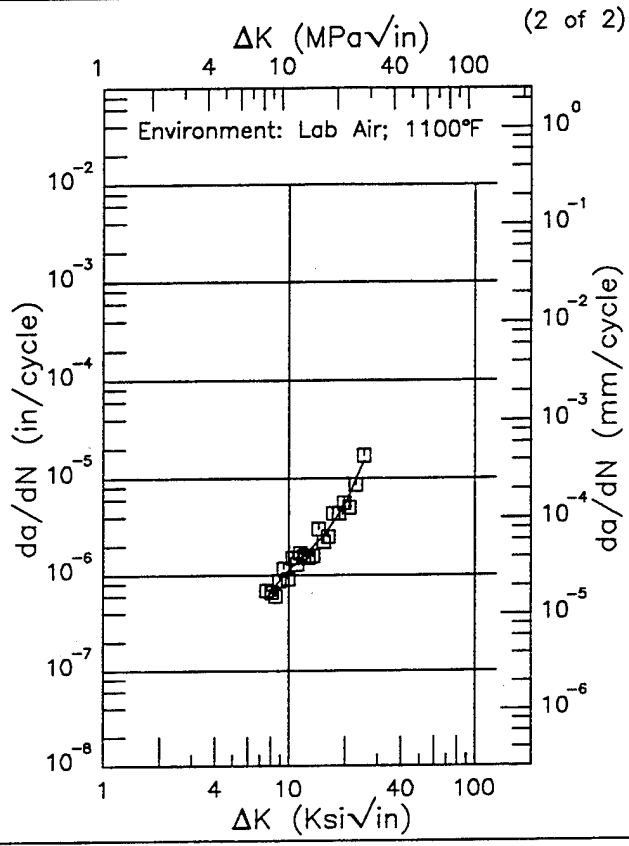
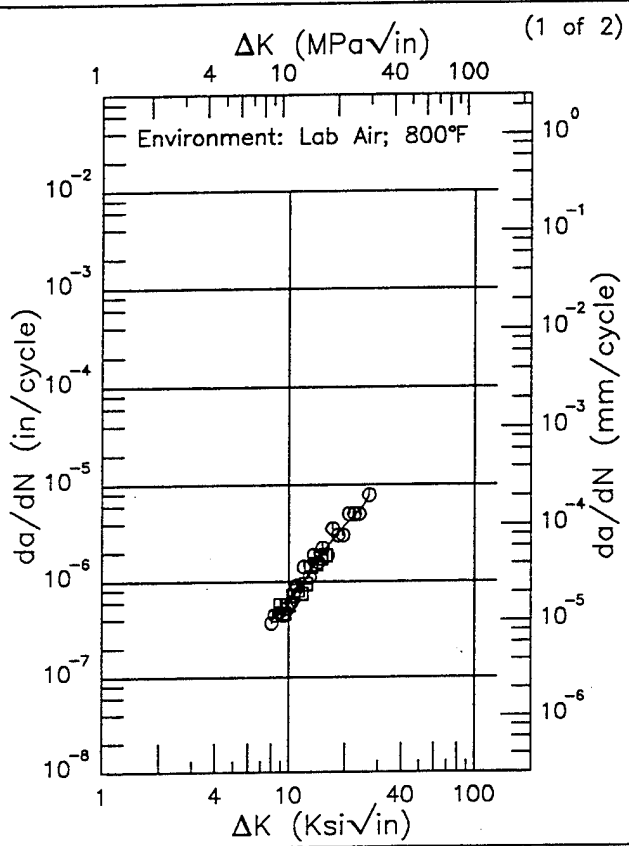


Figure 5.1.3.1.2 (Concluded)

EF ASTROLOY 901

Condition/Ht:
Form:
Specimen Type: SENT
Orientation:
Stress Ratio: 0.7

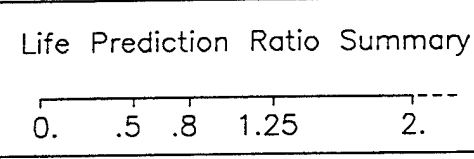
Yield Strength:
Ult. Strength:
Specimen Thk: 0.119 - 0.123 in.
Specimen Width: 2.5 in.
Ref: PW003



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
8.01 (min)	0.399
9.	0.472
10.	0.594
13.	1.26
16.	2.30
20.	3.86
25.	6.11
26.83 (max)	7.65

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
7.57 (min)	0.560
8.	0.661
9.	0.894
10.	1.12
13.	1.82
16.	2.78
20.	5.32
25.	14.3
25.23 (max)	15.0

RMS % Error
14.87



RMS % Error
16.94

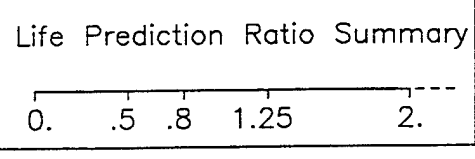


Figure 5.1.3.1.3

ASTROLOY 901

EF

Condition/Ht:

Form:

Specimen Type: SENT

Orientation:

Stress Ratio: 0.8

Yield Strength:

Ult. Strength:

Specimen Thk: 0.117 in.

Specimen Width: 2.5 in.

Ref: PW003

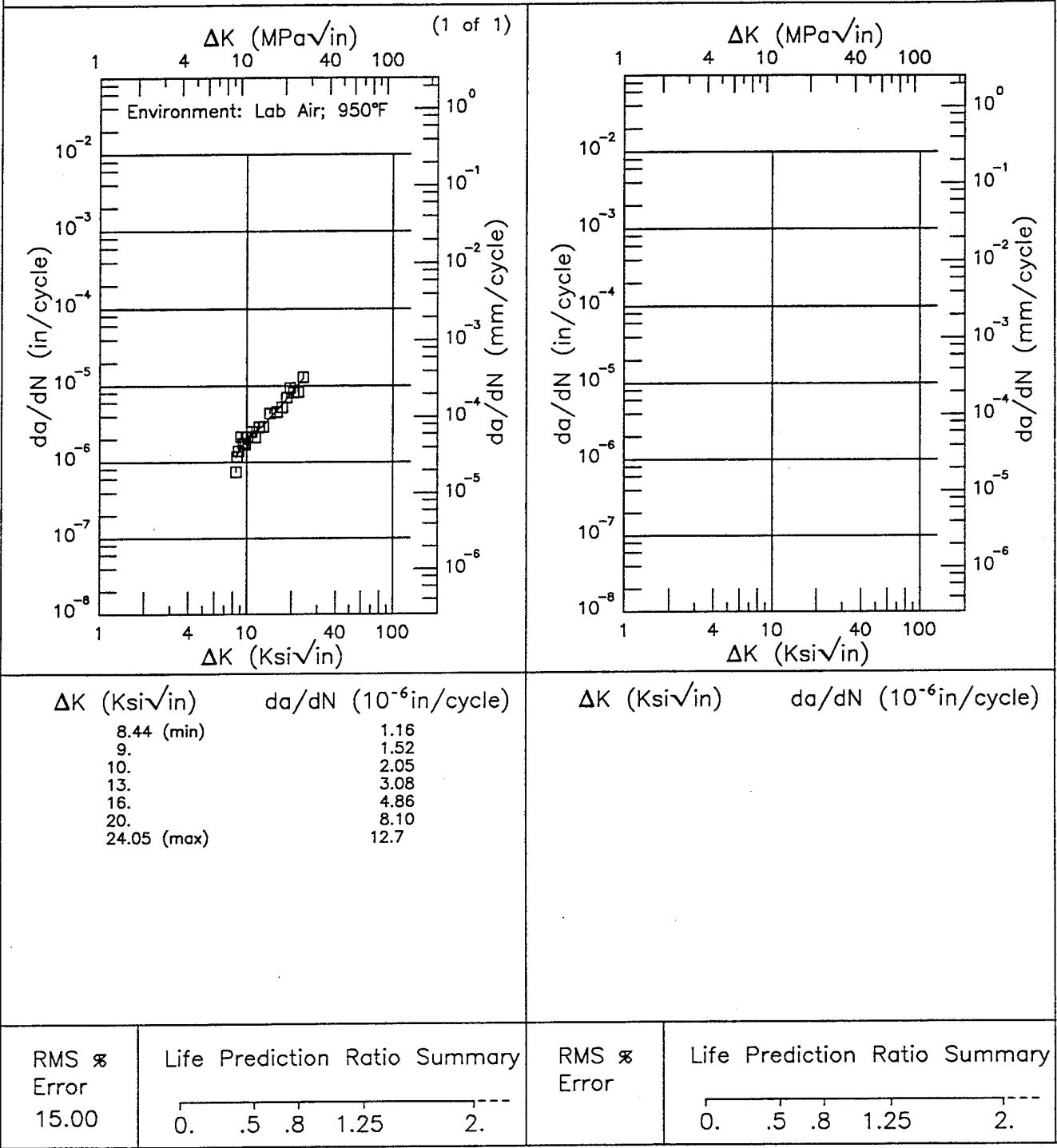
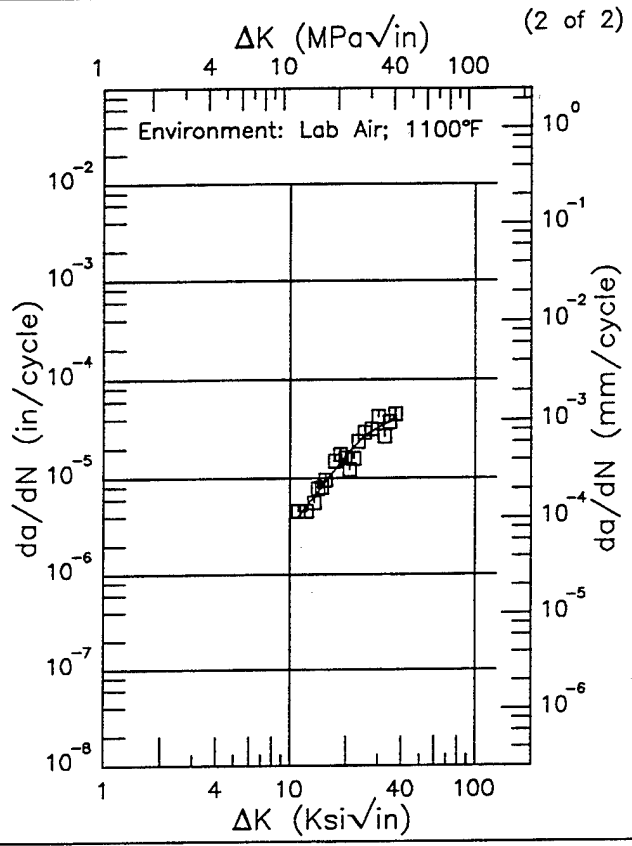
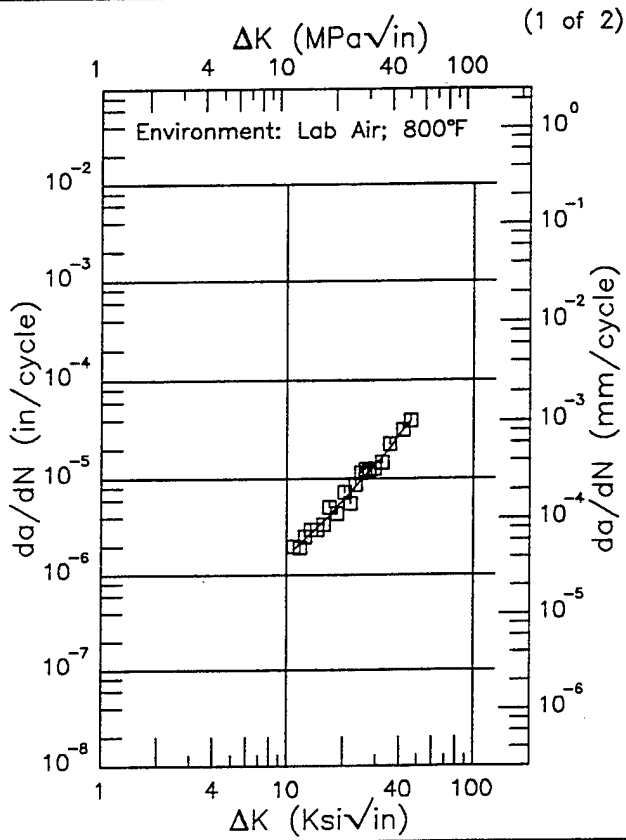


Figure 5.1.3.1.4

EF ASTROLOY 901

Condition/Ht:
 Form:
 Specimen Type: SENT
 Orientation: C-R
 Stress Ratio: 0.5

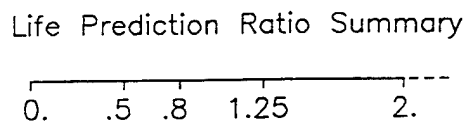
Yield Strength:
 Ult. Strength:
 Specimen Thk: 0.122 in.
 Specimen Width: 2.5 in.
 Ref: PW003



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
10.86 (min)	1.84
13.	2.55
16.	3.77
20.	5.94
25.	9.75
30.	14.3
35.	19.8
40.	28.5
45.80 (max)	38.4

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
11.04 (min)	3.90
13.	5.89
16.	9.84
20.	16.7
25.	26.5
30.	32.7
35.	37.9
36.86 (max)	42.0

RMS $\%$
 Error
 11.88



RMS $\%$
 Error
 16.45

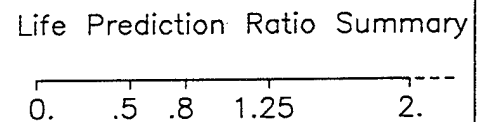


Figure 5.1.3.1.5

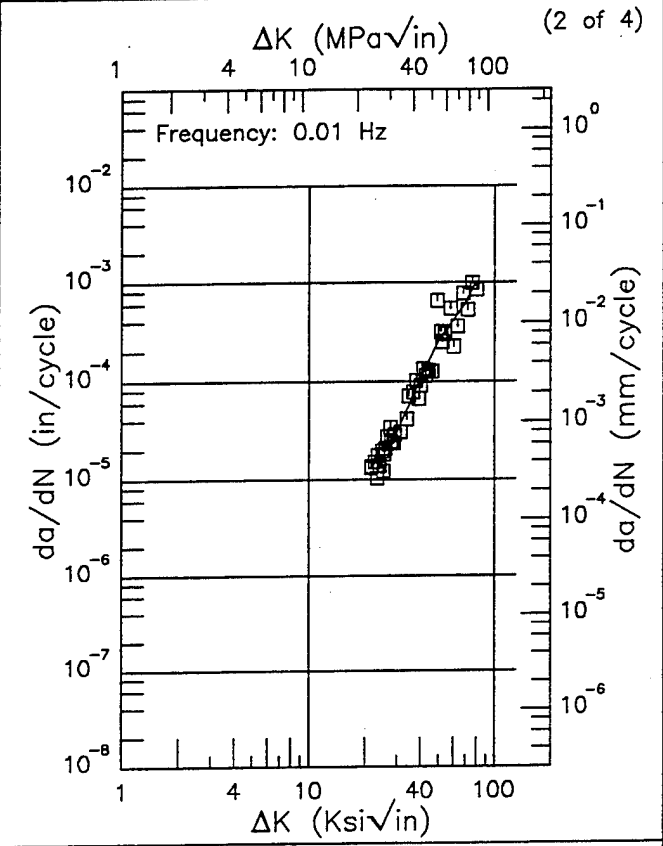
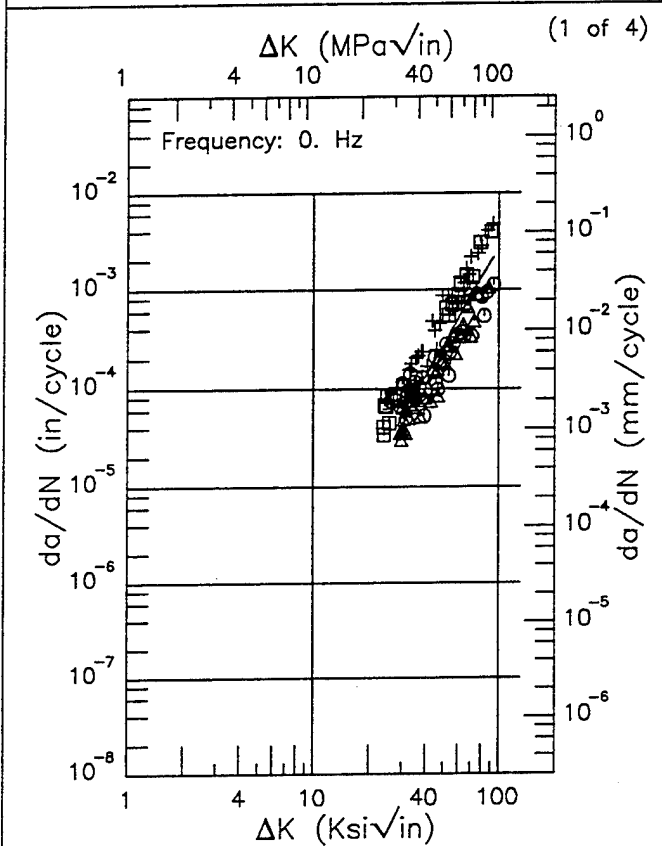
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ASTROLOY P/M-H

F

Condition/Ht: 2025F 3HR AC 1600F 8HR AC
 Form: Disk
 Specimen Type: WOL
 Orientation: C-R
 Stress Ratio: 0.05
 Environment: LAB AIR;1200°F

Yield Strength: 135.8 ksi
 Ult. Strength: 202 ksi
 Specimen Thk: 0.5 in.
 Specimen Width: 2.5 in.
 Ref: PW004



ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
23.79 (min)	61.8
25.	61.8
30.	71.6
35.	94.6
40.	132.
50.	262.
60.	492.
70.	845.
80.	1326.
90.	1913.
93.51 (max)	2136.

ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
21.63 (min)	14.1
25.	17.2
30.	30.7
35.	59.1
40.	108.
50.	258.
60.	411.
70.	607.
80.	1019.
80.05 (max)	1022.

RMS % Error 66.39	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

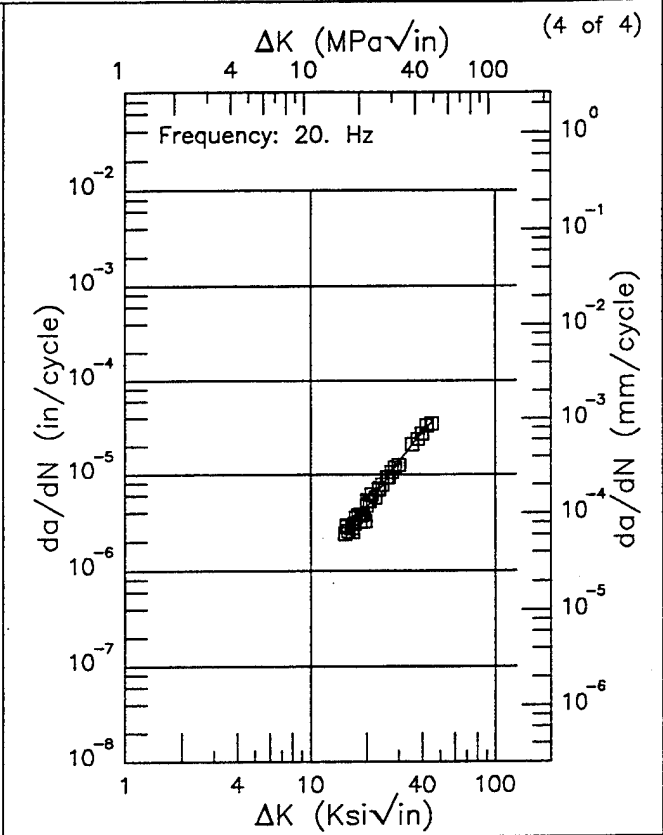
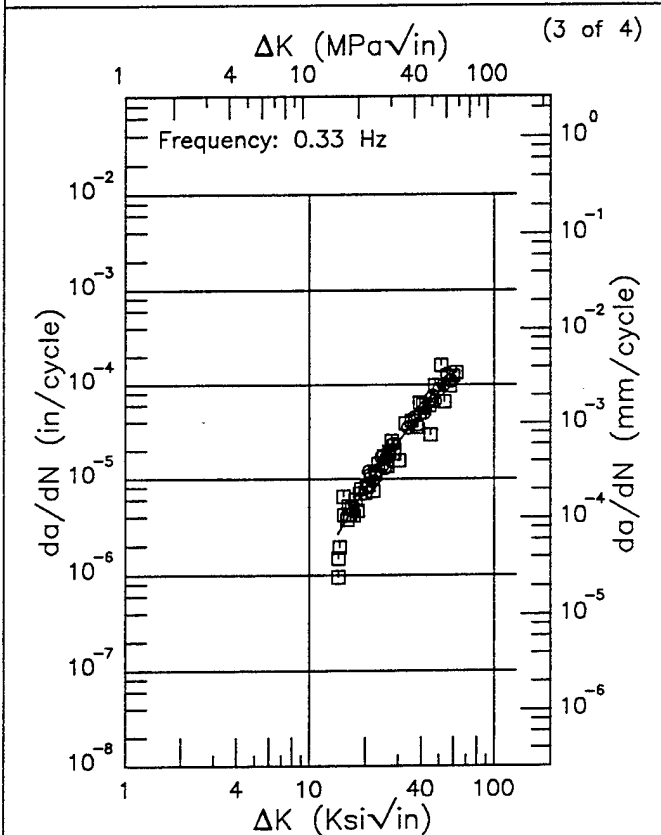
RMS % Error 35.81	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

Figure 5.2.3.1

ASTROLOY P/M-H F

Condition/Ht: 2025F 3HR AC 1600F 8HR AC
 Form: Disk
 Specimen Type: WOL
 Orientation: C-R
 Stress Ratio: 0.05
 Environment: LAB AIR;1200°F

Yield Strength: 135.8 ksi
 Ult. Strength: 202 ksi
 Specimen Thk: 0.5 in.
 Specimen Width: 2.5 in.
 Ref: PW004



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
14.25 (min)	2.67
16.	3.95
20.	7.93
25.	14.9
30.	24.2
35.	35.7
40.	49.4
50.	84.4
60.	130.
62.12 (max)	142.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
15.27 (min)	2.38
16.	2.63
20.	4.53
25.	8.25
30.	13.5
35.	20.2
40.	27.9
44.64 (max)	35.4

RMS % Error 25.09	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

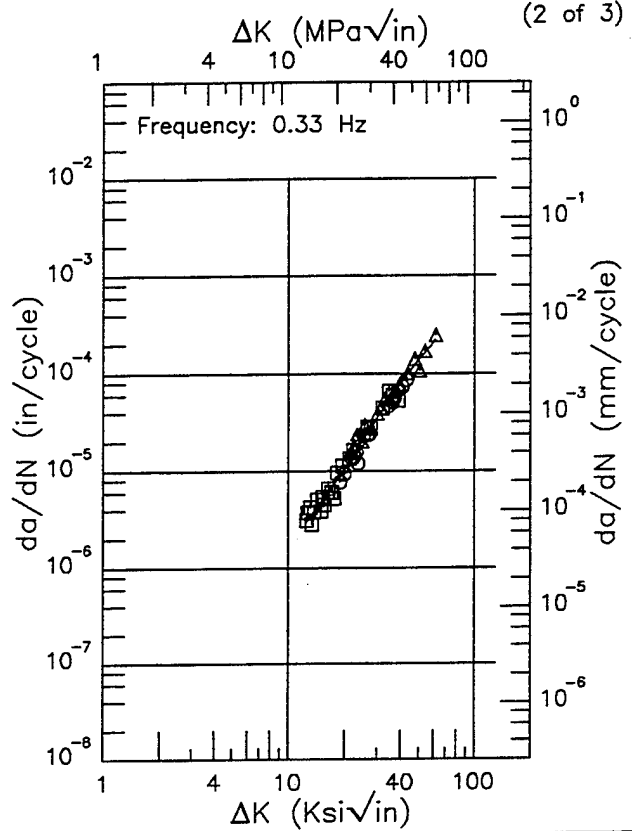
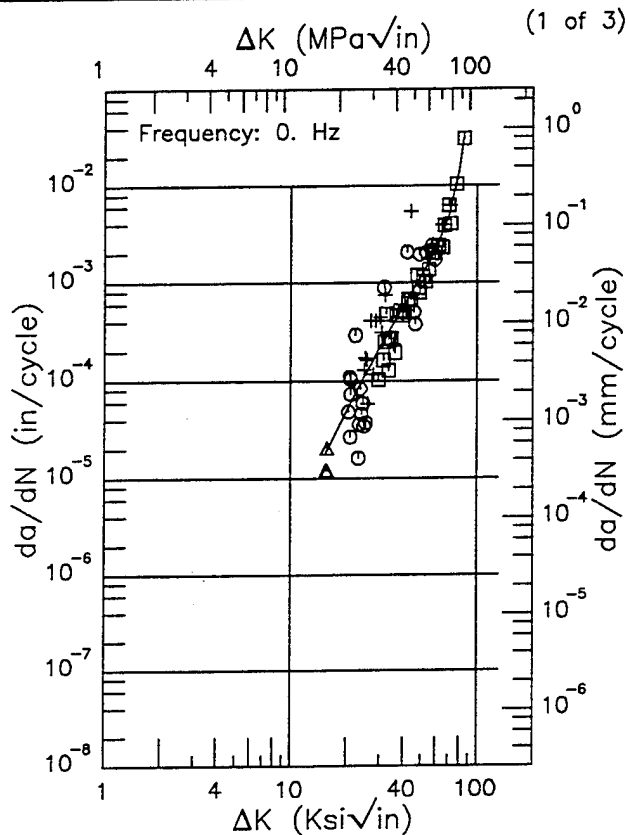
RMS % Error 8.69	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
---------------------	---

Figure 5.2.3.1 (Concluded)

ASTROLOY P/M-W

Condition/Ht: 2025F 4HRS AC 1600F 8HRS AC
 Form: Disk
 Specimen Type: WOL
 Orientation: C-R
 Stress Ratio: 0.05
 Environment: LAB AIR;1200°F

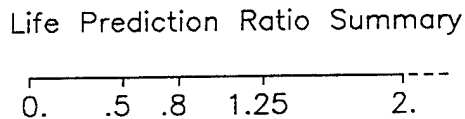
Yield Strength: 153 ksi
 Ult. Strength: 220 ksi
 Specimen Thk: 0.5 in.
 Specimen Width: 2.5 in.
 Ref: PW004



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
15.46 (min)	18.4
16.	21.0
20.	49.5
25.	114.
30.	220.
35.	371.
40.	571.
50.	1162.
60.	2199.
70.	4749.
80.	14463.
84.92 (max)	30958.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
12.48 (min)	2.98
13.	3.29
16.	5.65
20.	10.6
25.	20.6
30.	35.2
35.	54.9
40.	79.4
50.	140.
60.	208.
61.82 (max)	221.

RMS % Error
 >100.0



RMS % Error
 16.04

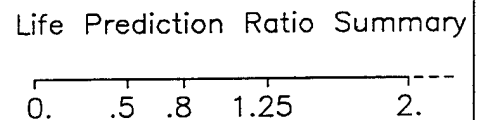


Figure 5.3.3.1

ASTROLOY P/M-W

F

Condition/Ht: 2025F 4HRS AC 1600F 8HRS AC
 Form: Disk
 Specimen Type: WOL
 Orientation: C-R
 Stress Ratio: 0.05
 Environment: LAB AIR;1200°F

Yield Strength: 153 ksi
 Ult. Strength: 220 ksi
 Specimen Thk: 0.5 in.
 Specimen Width: 2.5 in.
 Ref: PW004

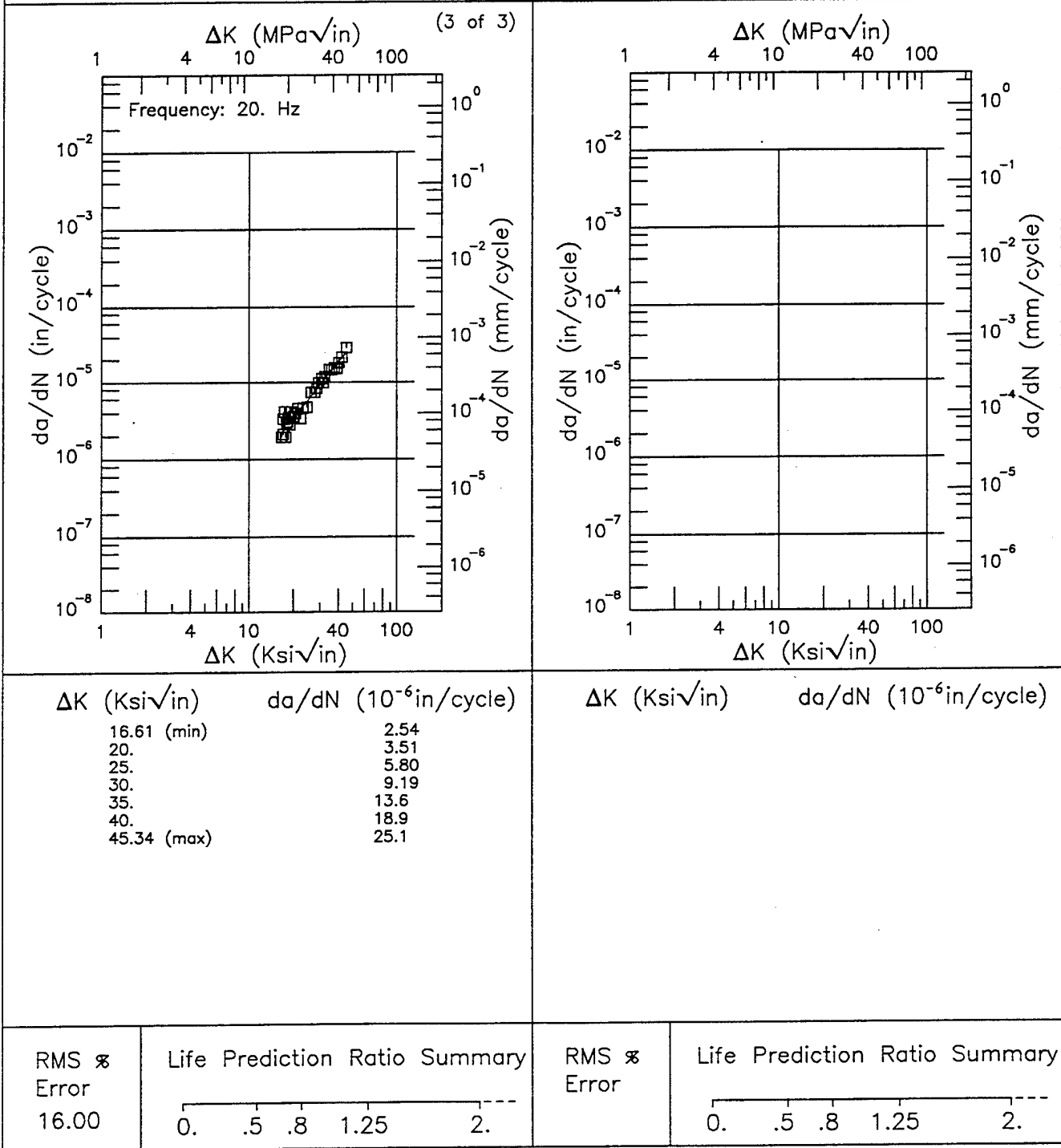


Figure 5.3.3.1 (Concluded)

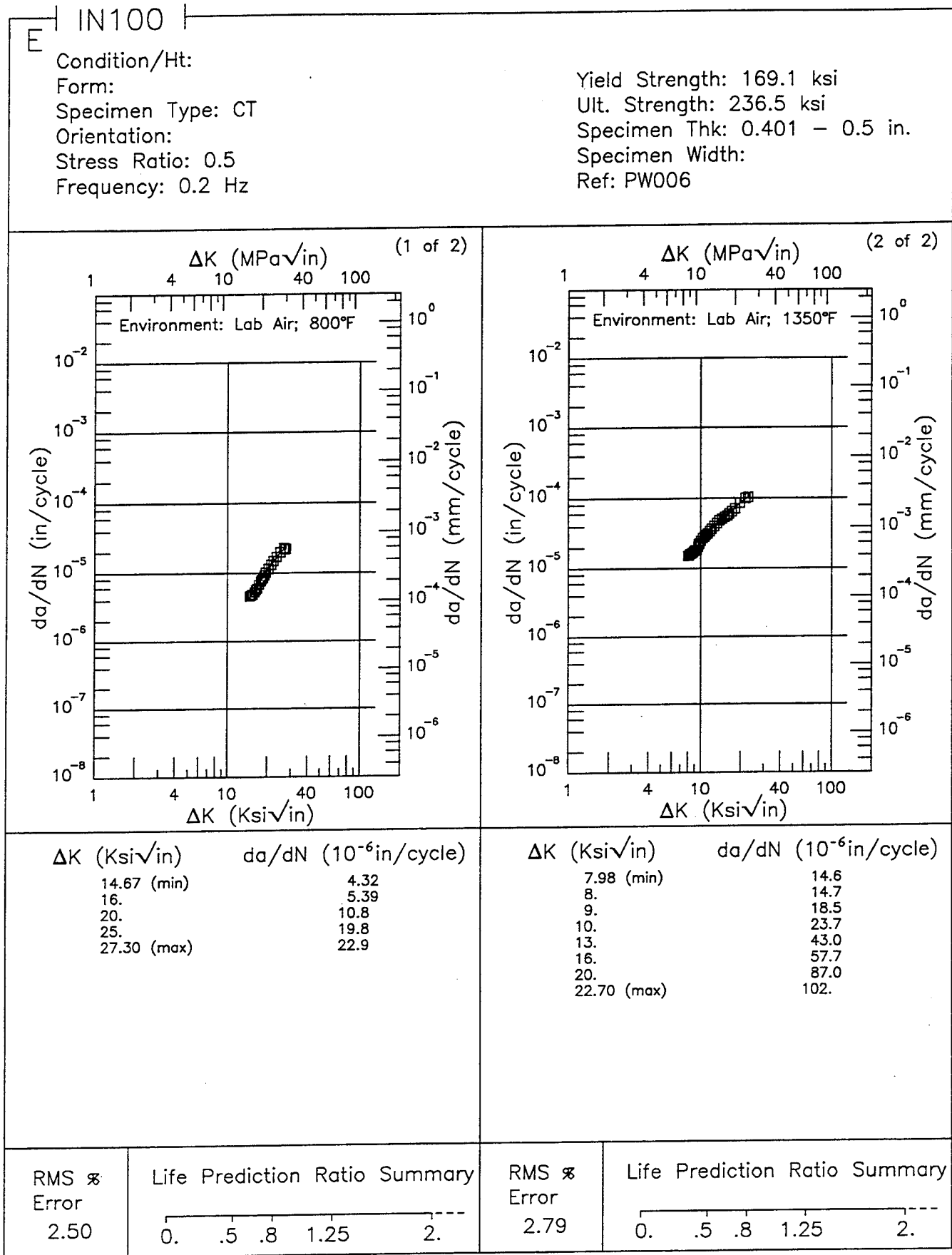
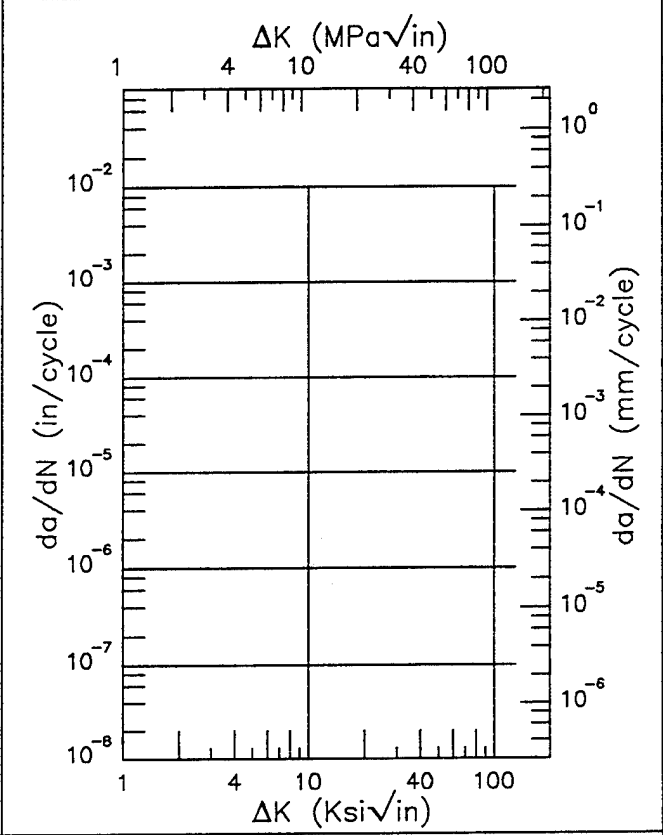
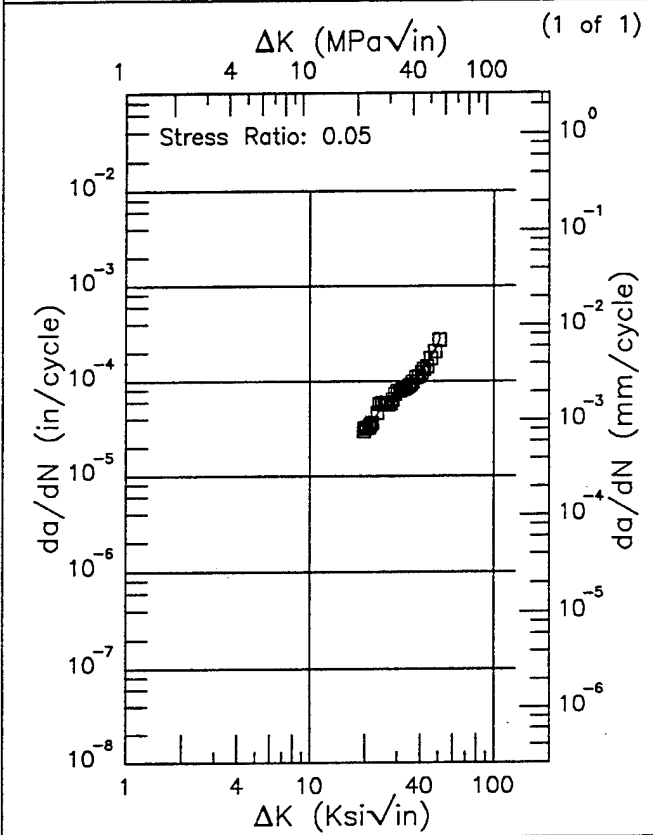


Figure 5.4.3.1.1

IN100 R

Condition/Ht:
 Form:
 Specimen Type: CT
 Orientation: C-R
 Frequency: 0.3 Hz
 Environment: LAB AIR;1200°F

Yield Strength: 169.1 ksi
 Ult. Strength: 236.5 ksi
 Specimen Thk: 0.5 in.
 Specimen Width:
 Ref: PW006



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
19.64 (min)	27.2
20.	29.2
25.	54.2
30.	72.8
35.	91.2
40.	118.
50.	244.
50.85 (max)	263.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
19.64 (min)	27.2
20.	29.2
25.	54.2
30.	72.8
35.	91.2
40.	118.
50.	244.
50.85 (max)	263.

RMS % Error	Life Prediction Ratio Summary
5.90	0. .5 .8 1.25 2.

RMS % Error	Life Prediction Ratio Summary
5.90	0. .5 .8 1.25 2.

Figure 5.4.3.1.2

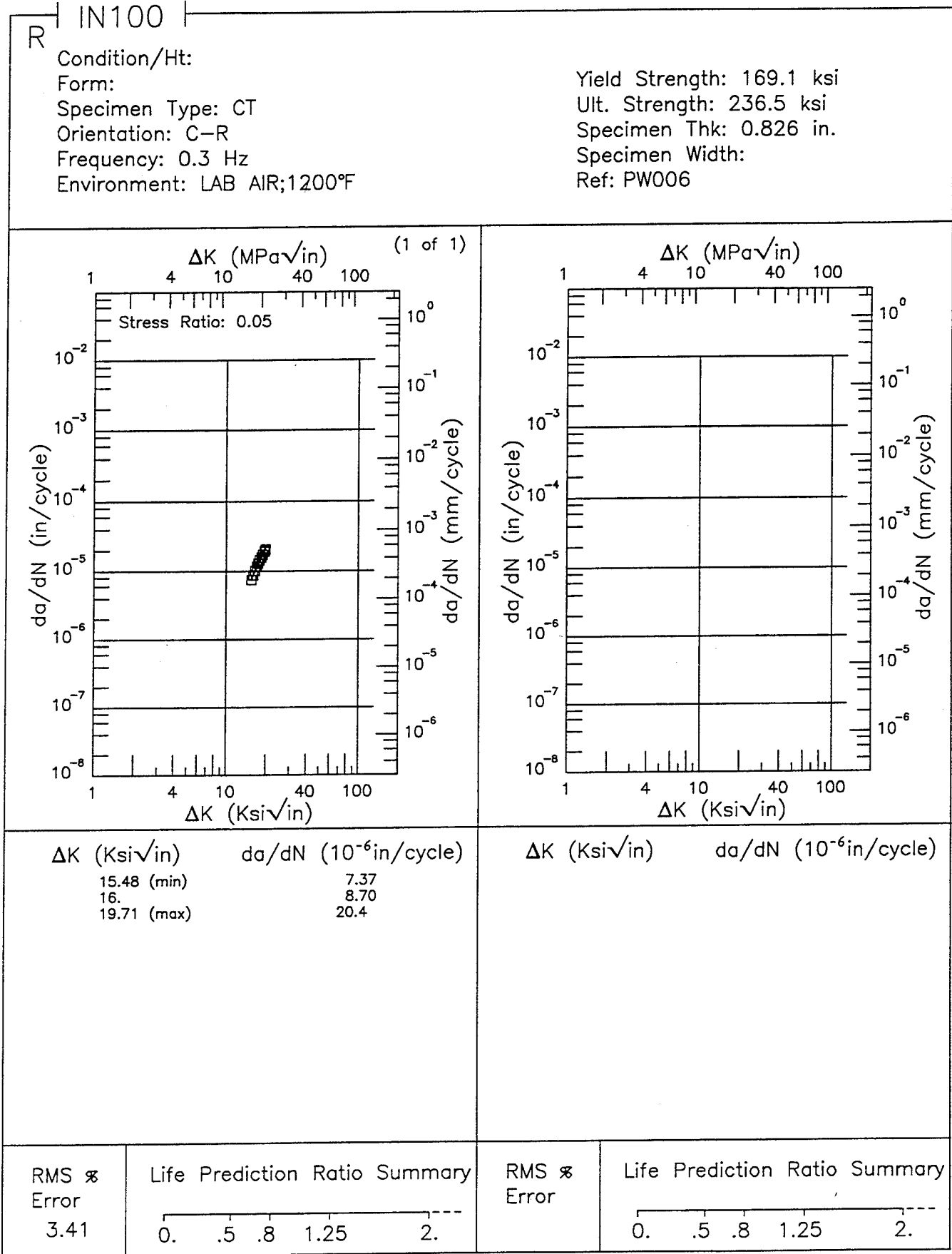


Figure 5.4.3.1.3

IN100 R

Condition/Ht:
 Form:
 Specimen Type: CT
 Orientation: C-R
 Frequency: 20 Hz
 Environment: LAB AIR;800°F

Yield Strength: 169.1 ksi
 Ult. Strength: 236.5 ksi
 Specimen Thk: 0.5 in.
 Specimen Width:
 Ref: PW006

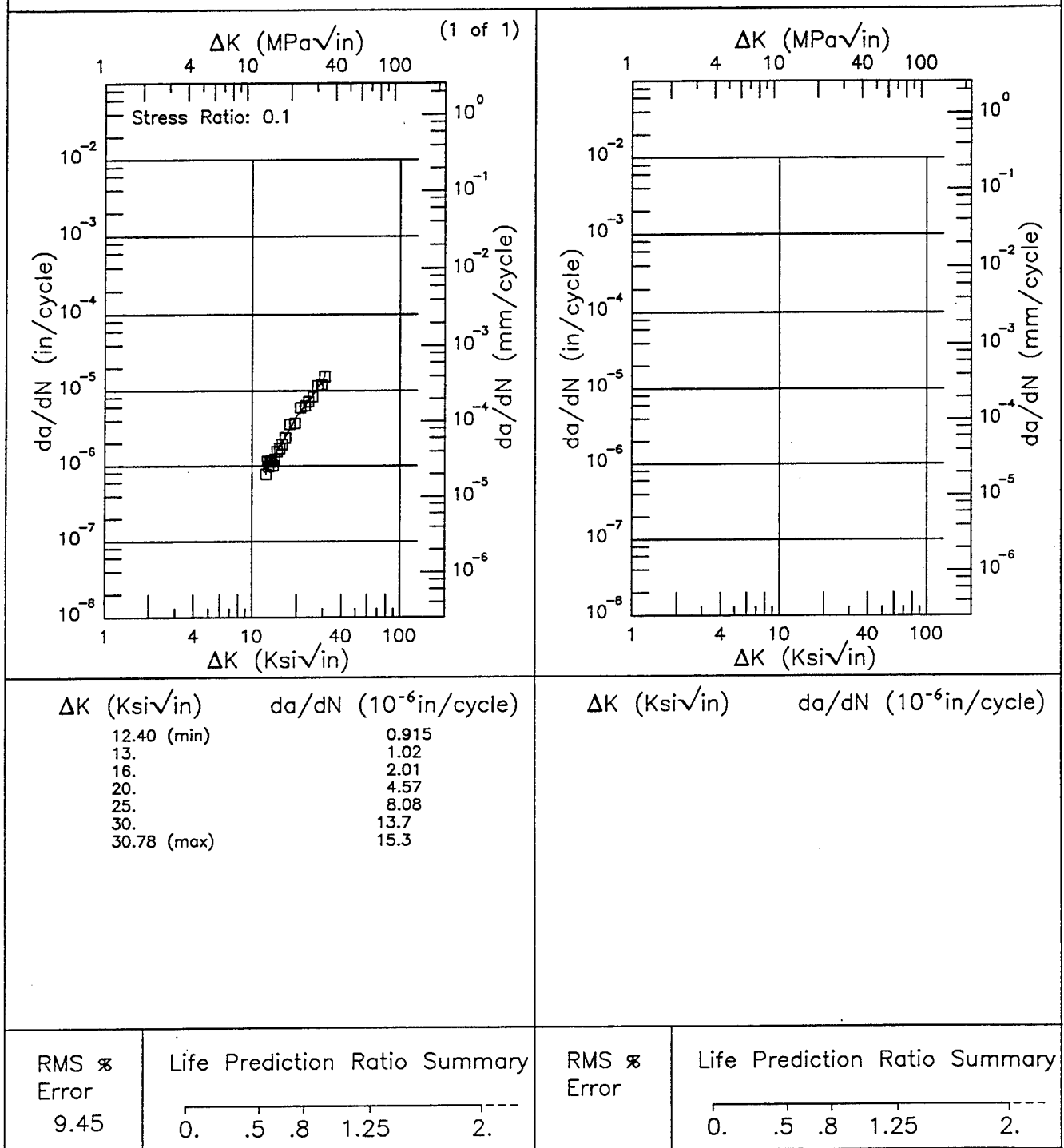


Figure 5.4.3.1.4

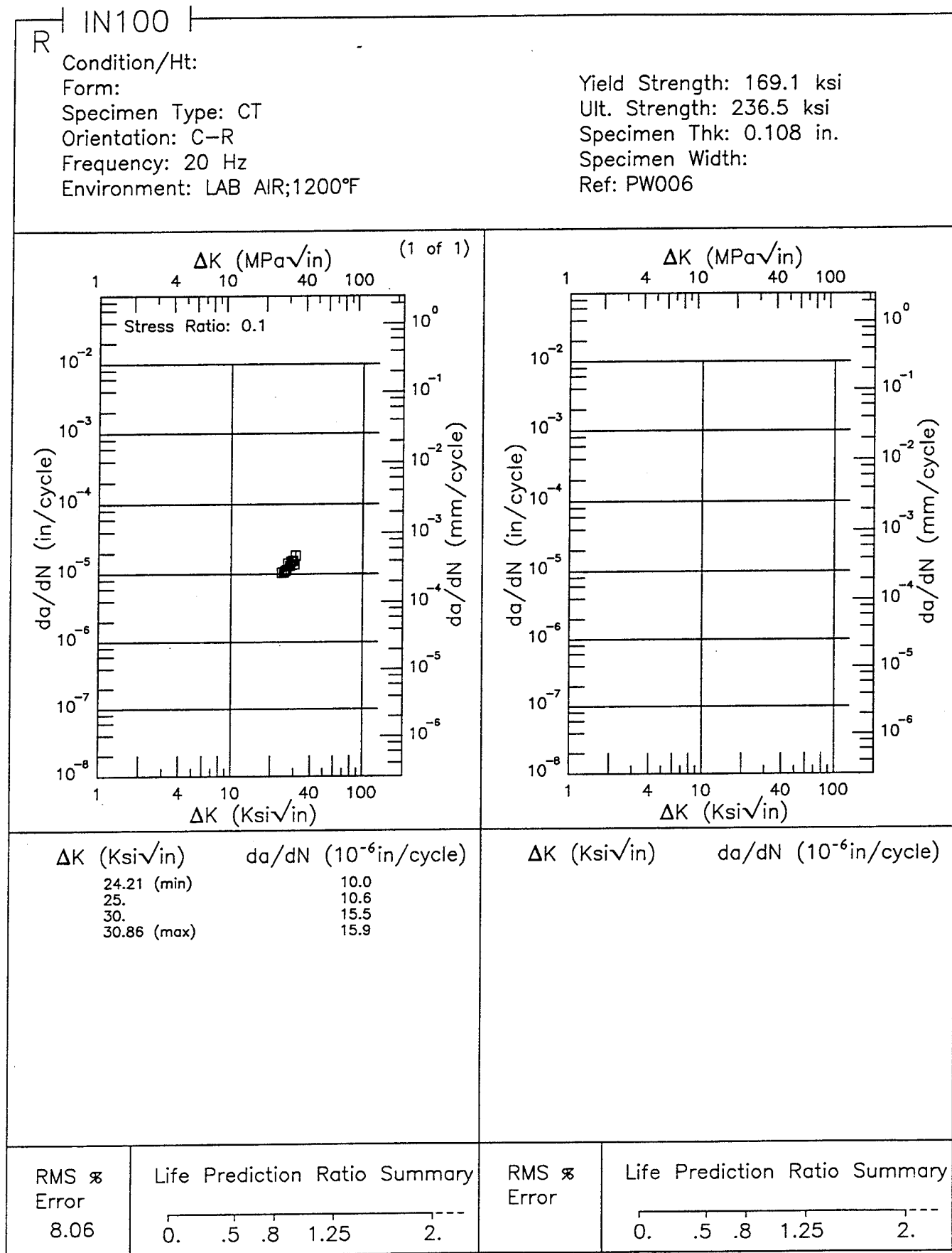
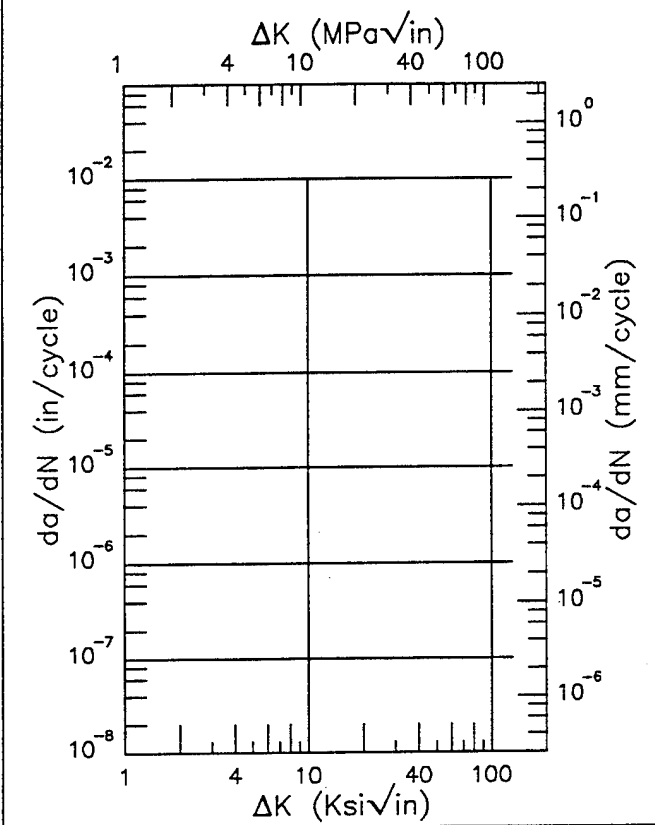
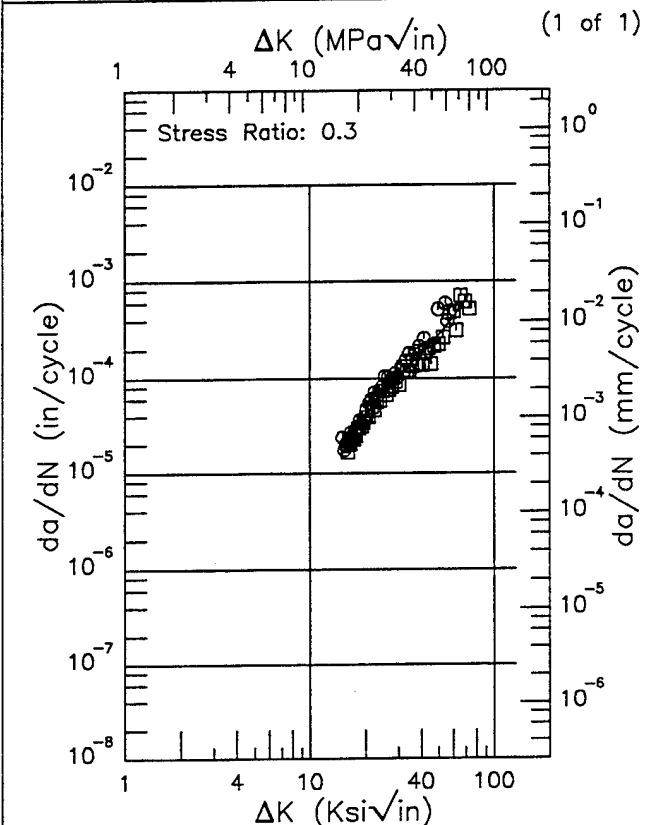


Figure 5.4.3.1.5

IN100 R

Condition/Ht:
 Form:
 Specimen Type: CT
 Orientation: C-R
 Frequency: 0. Hz
 Environment: LAB AIR;1100°F

Yield Strength: 169.1 ksi
 Ult. Strength: 236.5 ksi
 Specimen Thk: 0.251 in.
 Specimen Width:
 Ref: PW006



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
14.98 (min)	15.9
16.	20.4
20.	41.9
25.	73.4
30.	107.
35.	143.
40.	183.
50.	283.
60.	429.
70.	650.
72.77 (max)	731.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
14.98 (min)	15.9
16.	20.4
20.	41.9
25.	73.4
30.	107.
35.	143.
40.	183.
50.	283.
60.	429.
70.	650.
72.77 (max)	731.

RMS $\%$ Error 21.34	Life Prediction Ratio Summary
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RMS $\%$ Error	Life Prediction Ratio Summary
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Figure 5.4.3.1.6

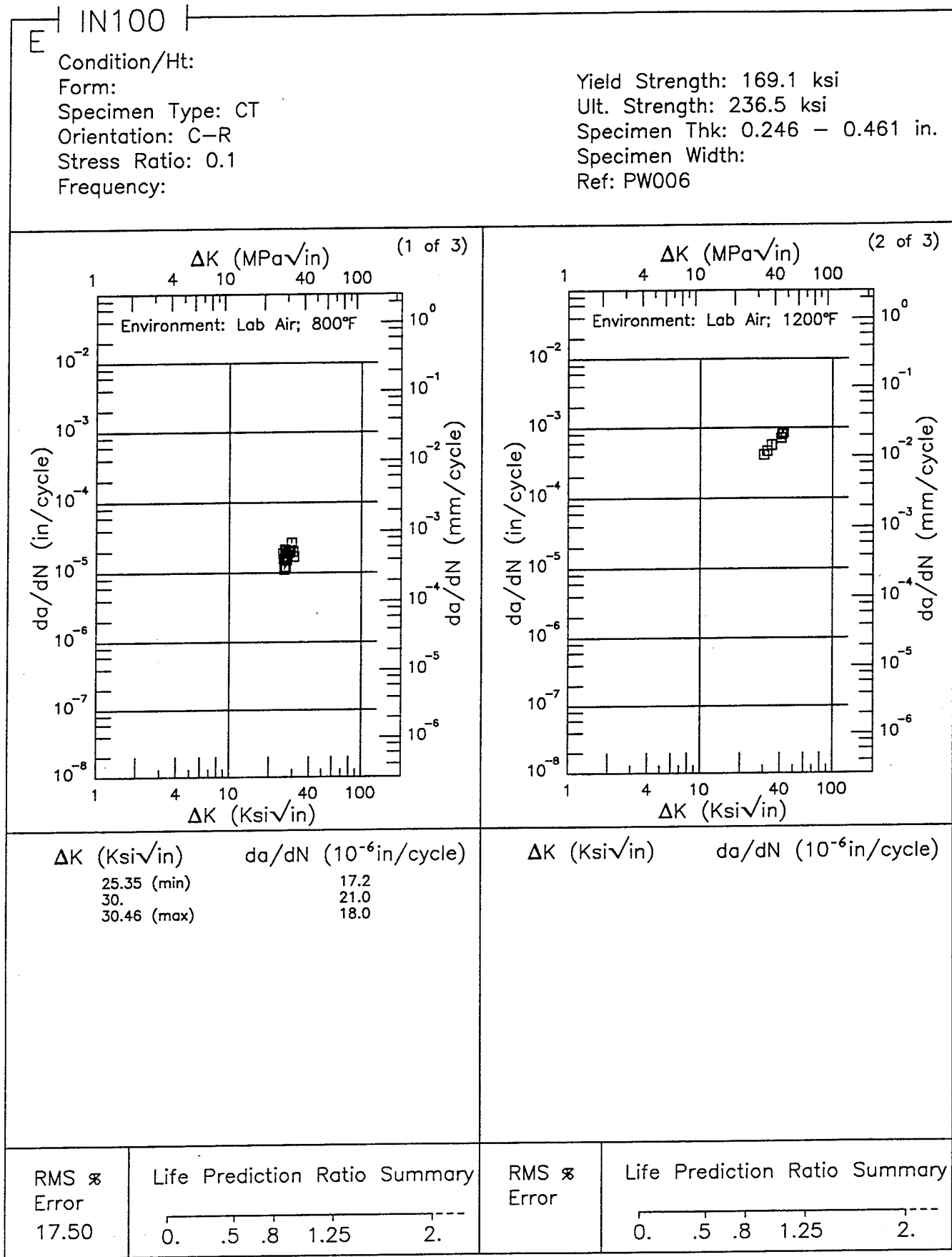


Figure 5.4.3.1.7

IN100 E

Condition/Ht:

Form:

Specimen Type: CT

Orientation: C-R

Stress Ratio: 0.1

Frequency:

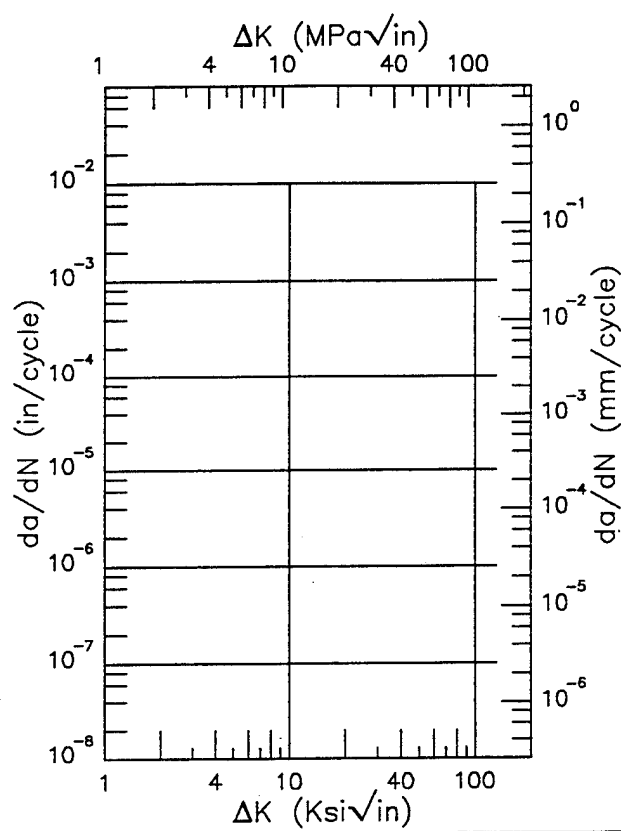
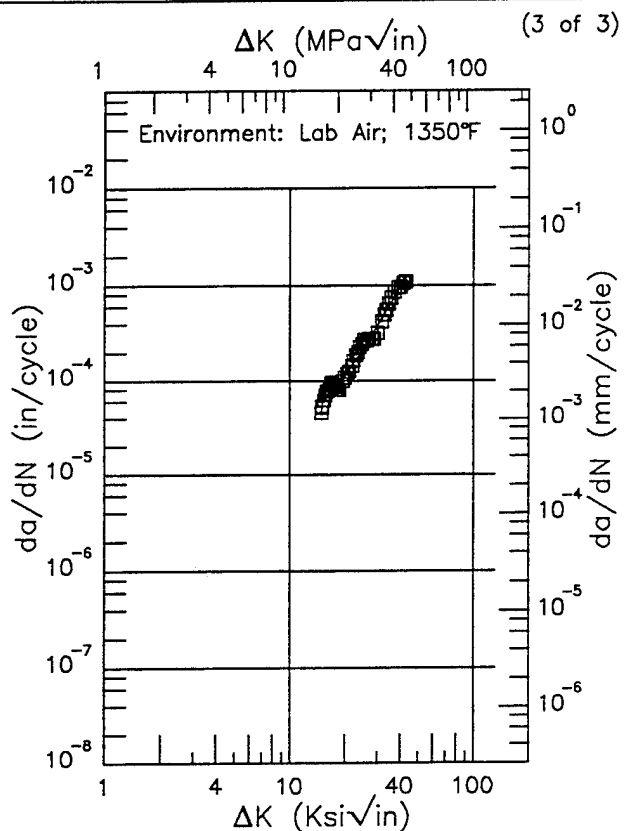
Yield Strength: 169.1 ksi

Ult. Strength: 236.5 ksi

Specimen Thk: 0.246 - 0.461 in.

Specimen Width:

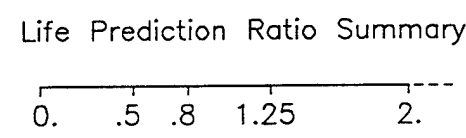
Ref: PW006



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
14.82 (min)	60.0
16.	69.9
20.	117.
25.	217.
30.	378.
35.	623.
40.	977.
42.68 (max)	1221.

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS %
Error
11.70



RMS %
Error

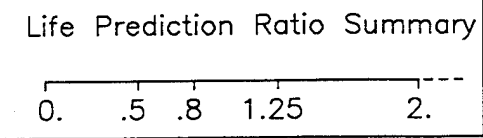


Figure 5.4.3.1.7 (Concluded)

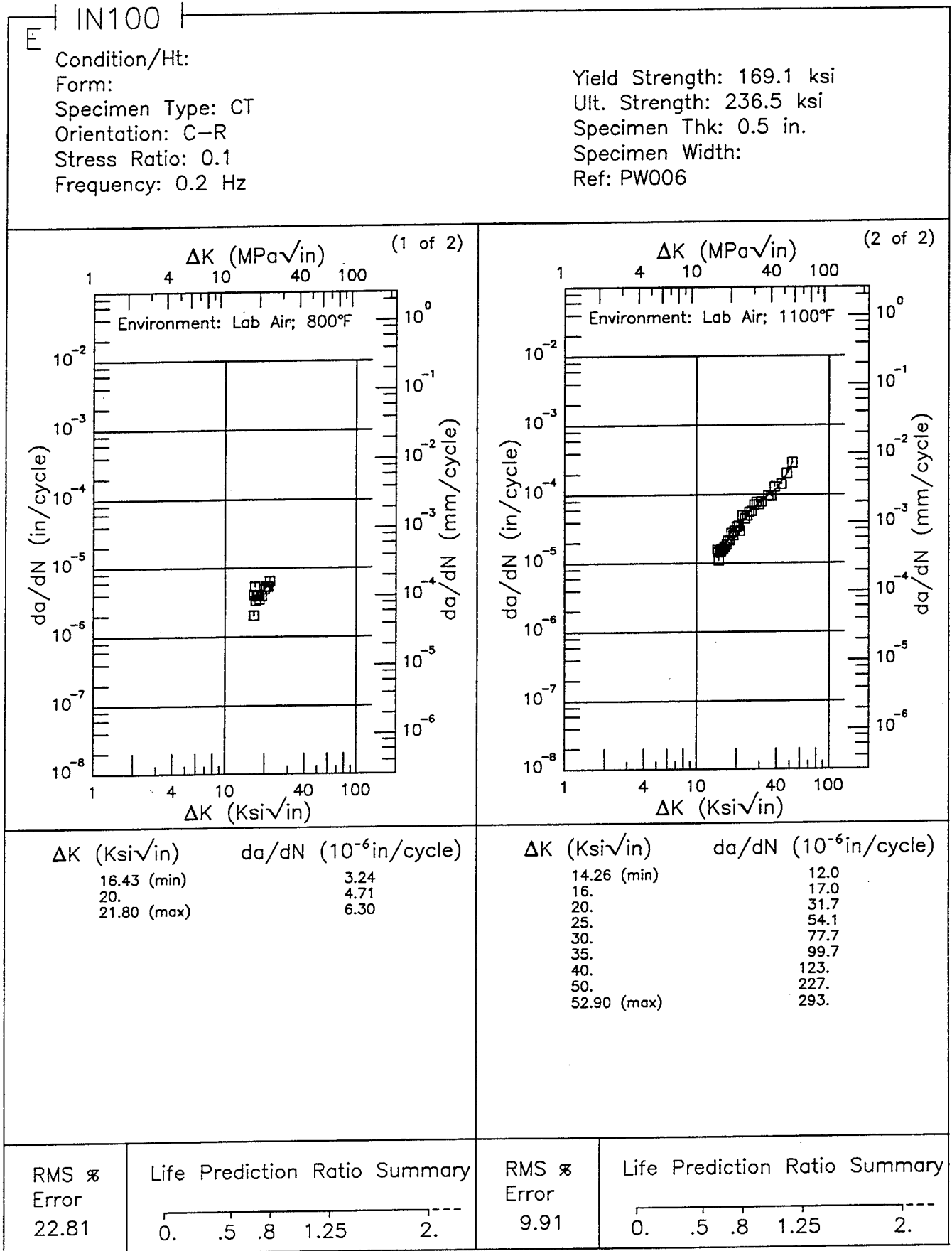
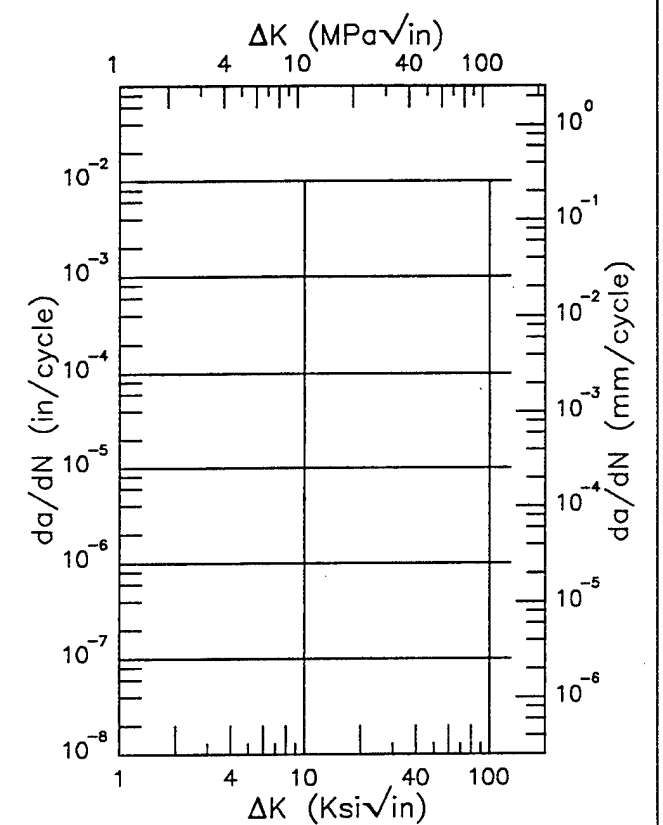
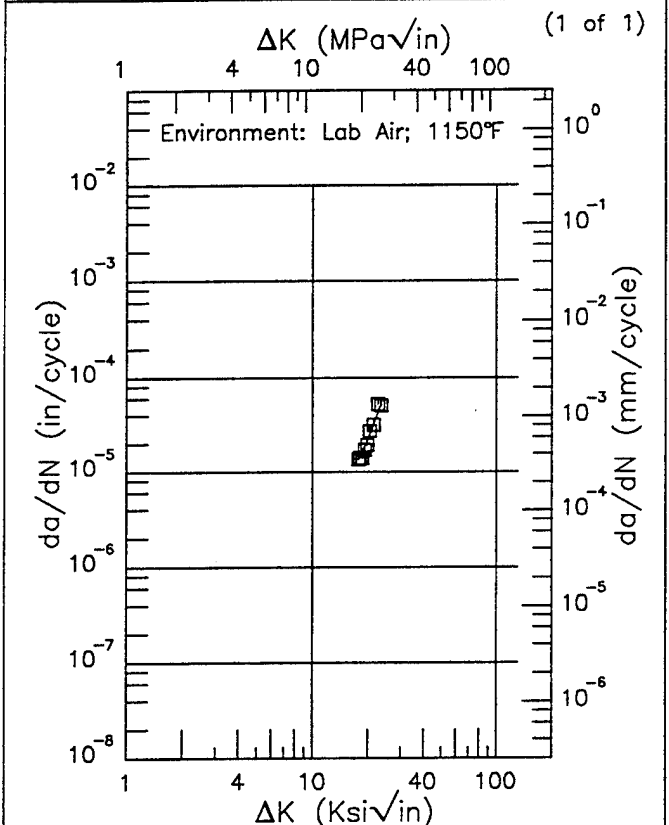


Figure 5.4.3.1.8

IN100 E

Condition/Ht:
 Form:
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.1
 Frequency: 0.2 Hz

Yield Strength: 169.1 ksi
 Ult. Strength: 236.5 ksi
 Specimen Thk: 0.25 in.
 Specimen Width:
 Ref: PW006



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
17.57 (min)	13.7
20.	22.2
23.45 (max)	53.8

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
17.57 (min)	13.7
20.	22.2
23.45 (max)	53.8

RMS % Error 10.89	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

RMS % Error 10.89	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

Figure 5.4.3.1.9

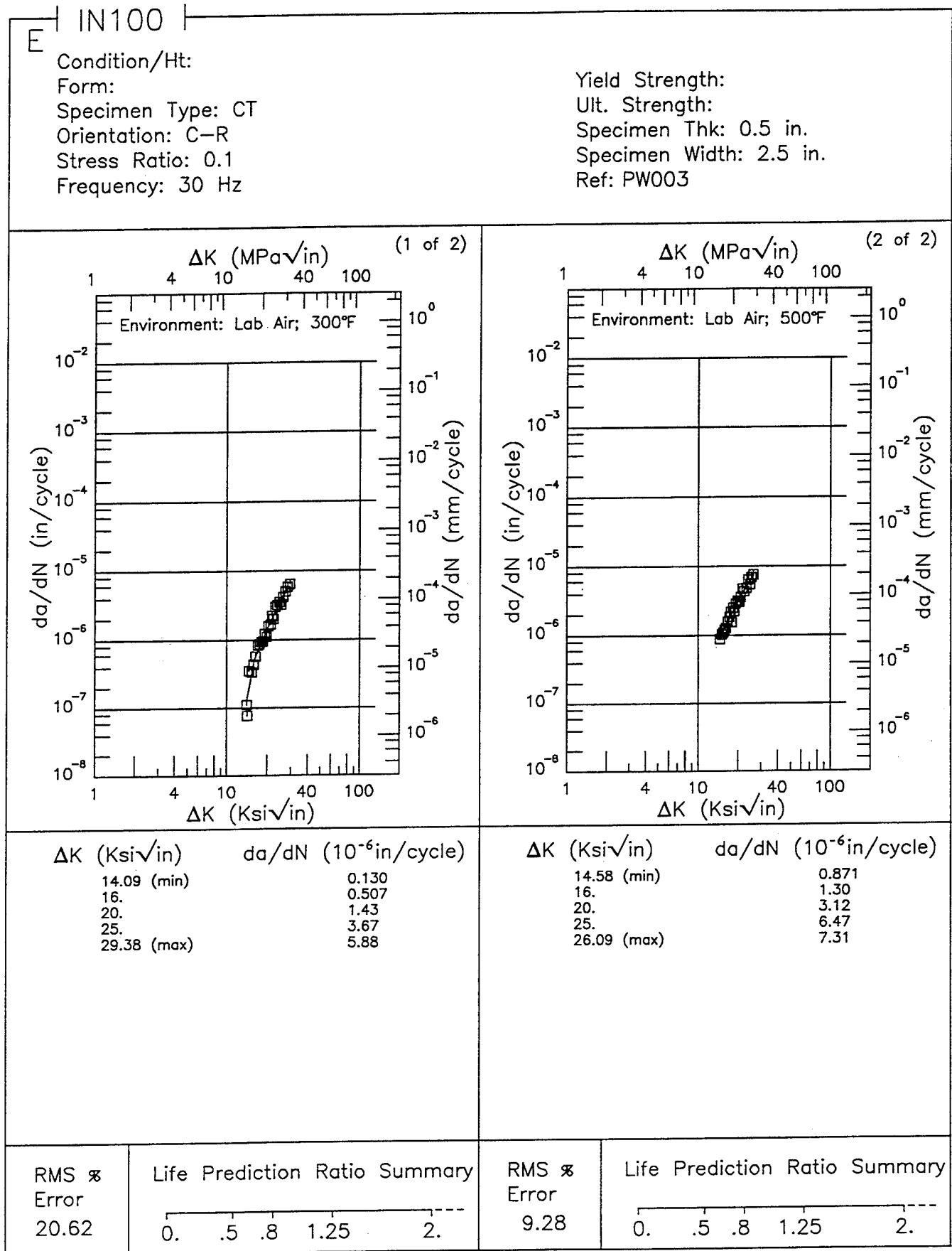
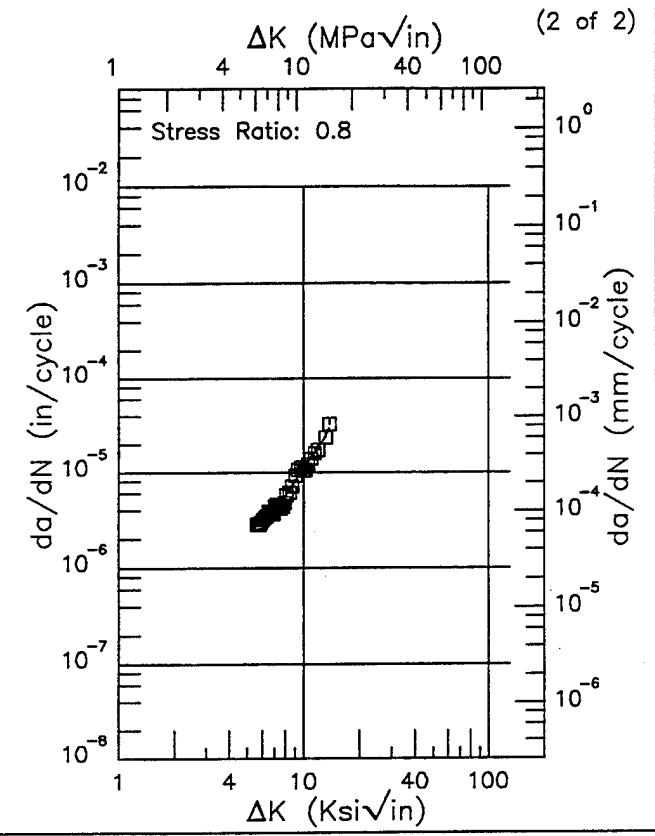
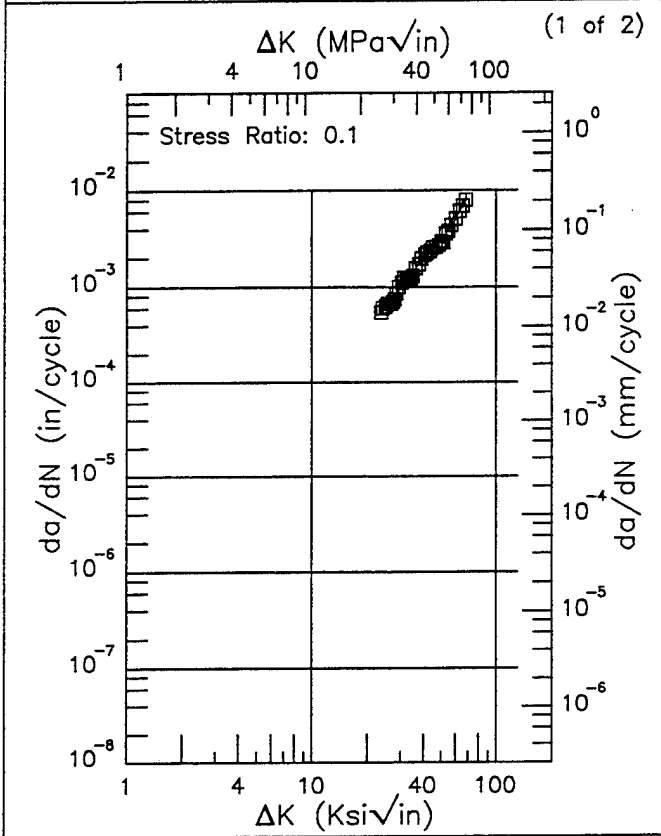


Figure 5.4.3.1.10

IN100 R

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation:
 Frequency: 0.2 Hz
 Environment: LAB AIR;1200°F

Yield Strength: 169.1 ksi
 Ult. Strength: 236.5 ksi
 Specimen Thk: 0.252 - 0.302 in.
 Specimen Width: 2.503 - 2.504 in.
 Ref: PW002



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
23.52 (min)	552.
25.	624.
30.	968.
35.	1434.
40.	1957.
50.	2985.
60.	5080.
67.89 (max)	7741.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
5.60 (min)	3.00
6.	3.04
7.	3.91
8.	5.68
9.	8.18
10.	11.0
13.	23.3
13.74 (max)	29.9

RMS $\%$ Error 5.85	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.
------------------------	--

RMS $\%$ Error 8.29	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.
------------------------	--

Figure 5.4.3.1.11

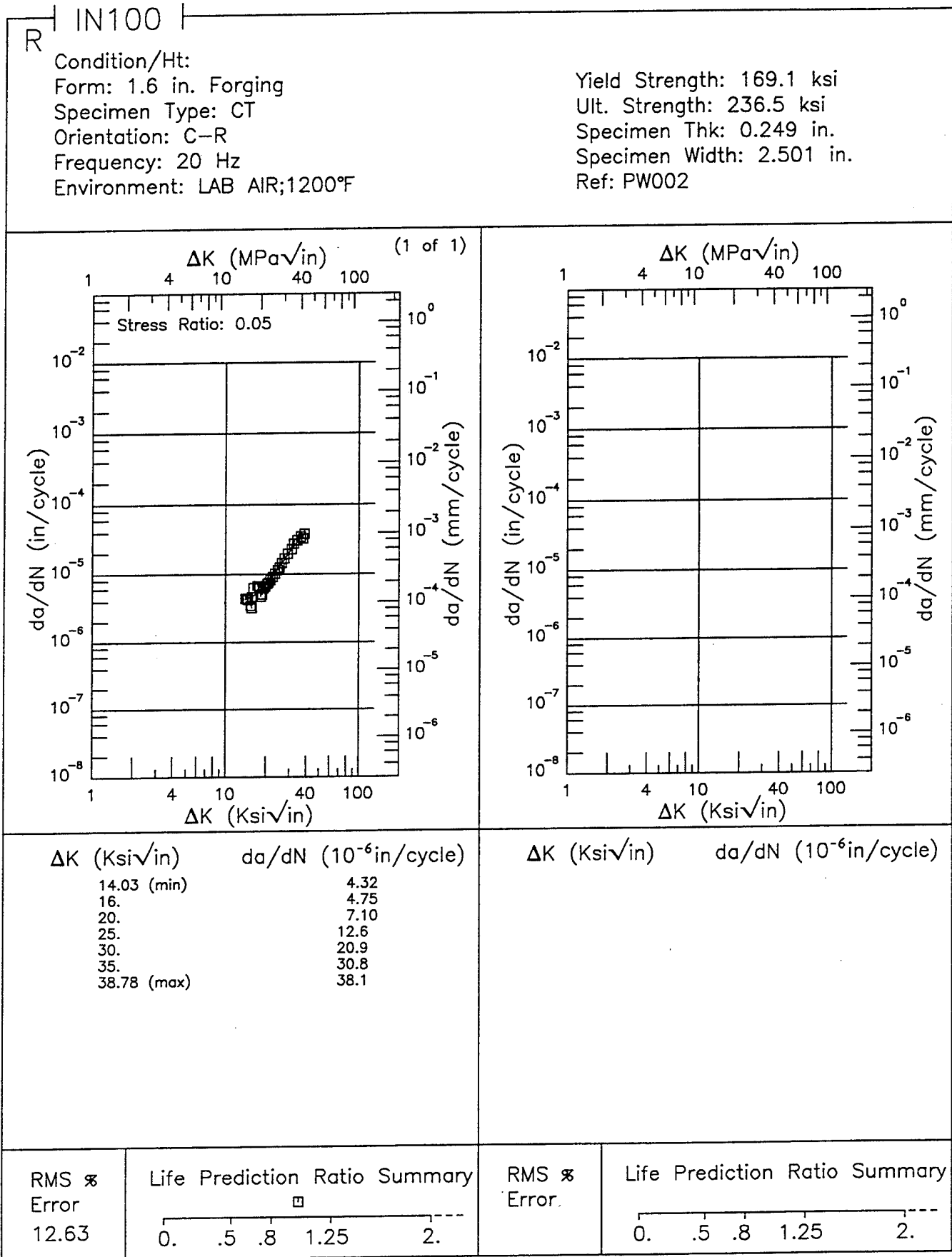
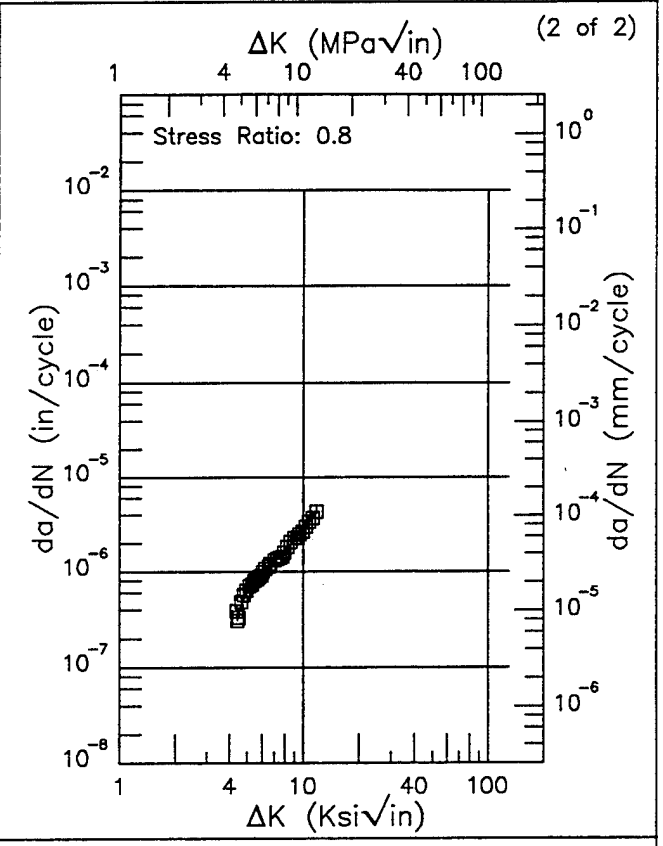
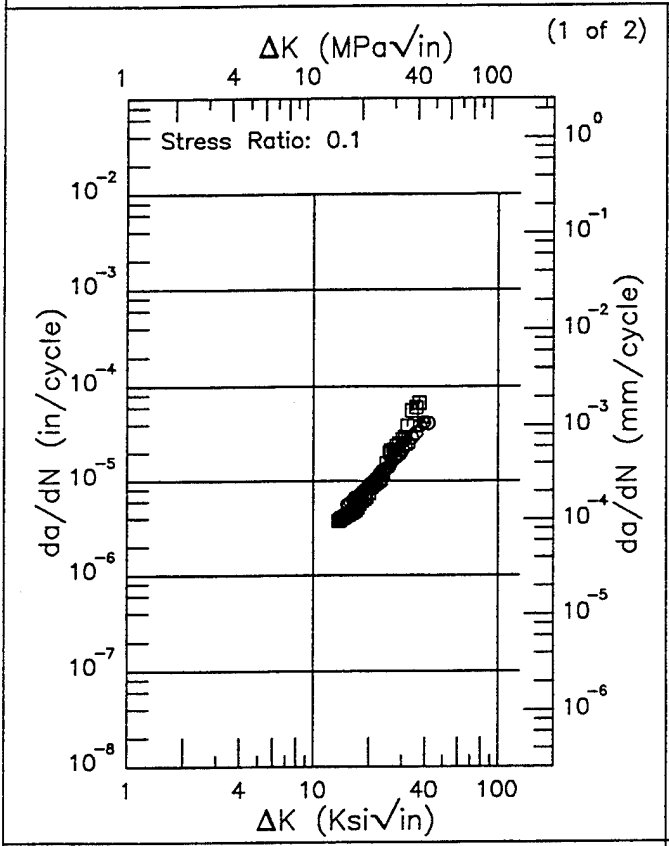


Figure 5.4.3.1.12

IN100 R

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation: C-R
 Frequency: 20 Hz
 Environment: LAB AIR;1200°F

Yield Strength: 169.1 ksi
 Ult. Strength: 236.5 ksi
 Specimen Thk: 0.206 - 0.442 in.
 Specimen Width: 2.494 - 2.508 in.
 Ref: PW002



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.55 (min)	3.94
16.	4.92
20.	8.16
25.	15.2
30.	25.3
35.	36.6
40.	46.6
41.45 (max)	49.0

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
4.39 (min)	0.346
5.	0.613
6.	0.990
7.	1.29
8.	1.63
9.	2.14
10.	2.81
11.78 (max)	4.09

RMS % Error 17.40	Life Prediction Ratio Summary □ ○ 0. .5 .8 1.25 2.
----------------------	--

RMS % Error 6.08	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.
---------------------	--

Figure 5.4.3.1.13

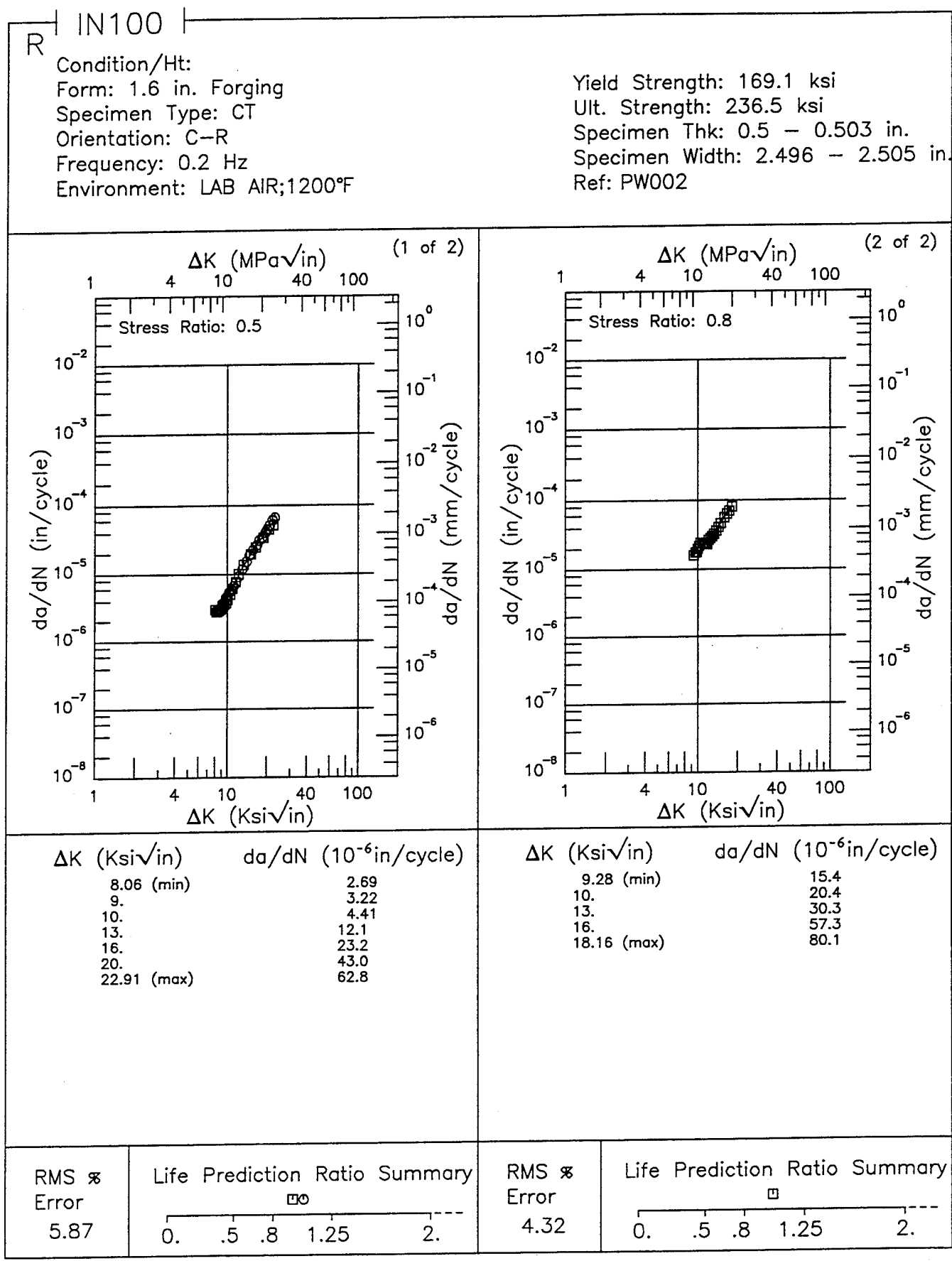


Figure 5.4.3.1.14

IN100 E

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.1
 Frequency:

Yield Strength: 169.1 ksi
 Ult. Strength: 236.5 ksi
 Specimen Thk: 0.25 - 0.856 in.
 Specimen Width: 2.436 - 2.51 in.
 Ref: PW002

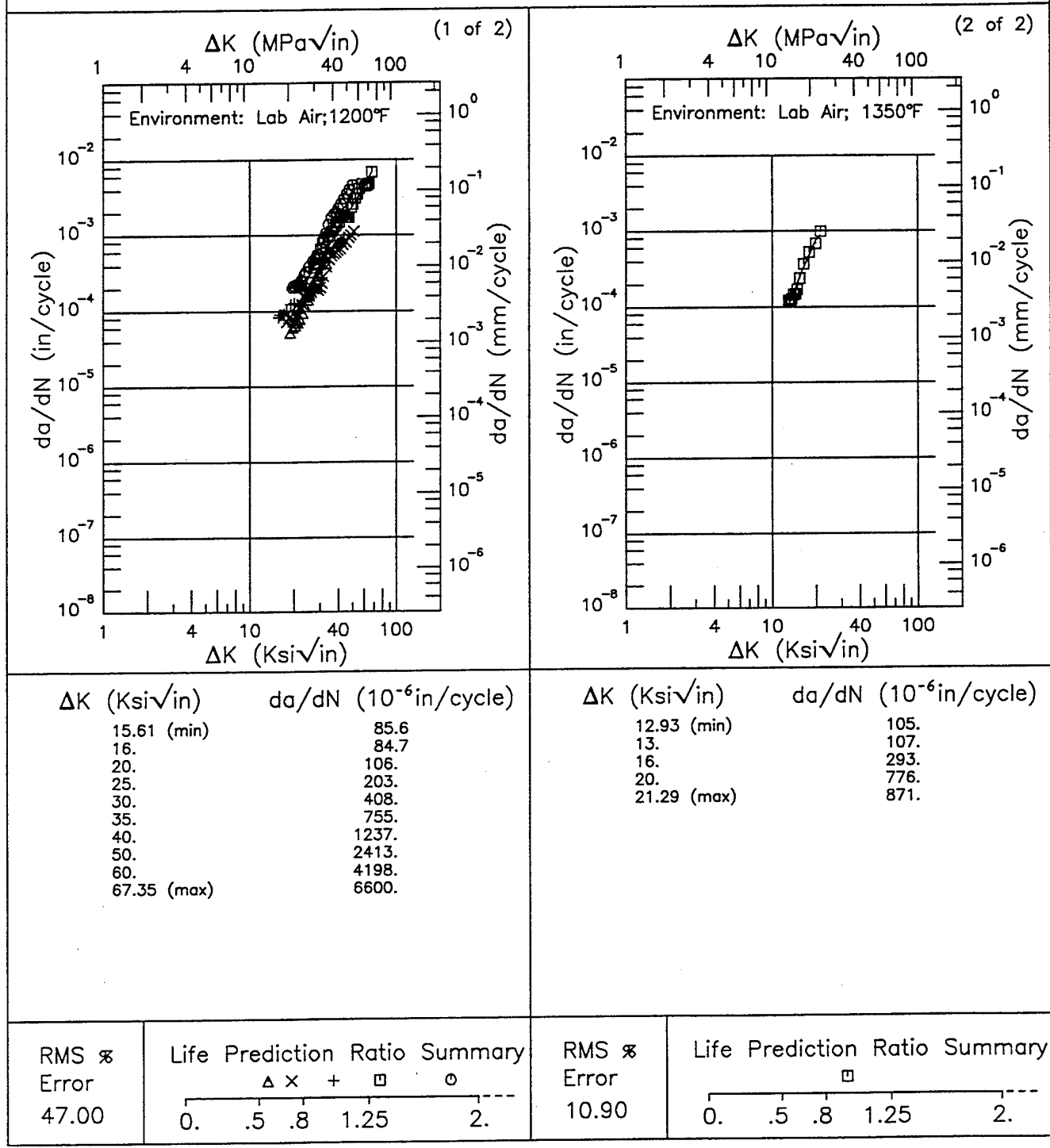


Figure 5.4.3.1.15

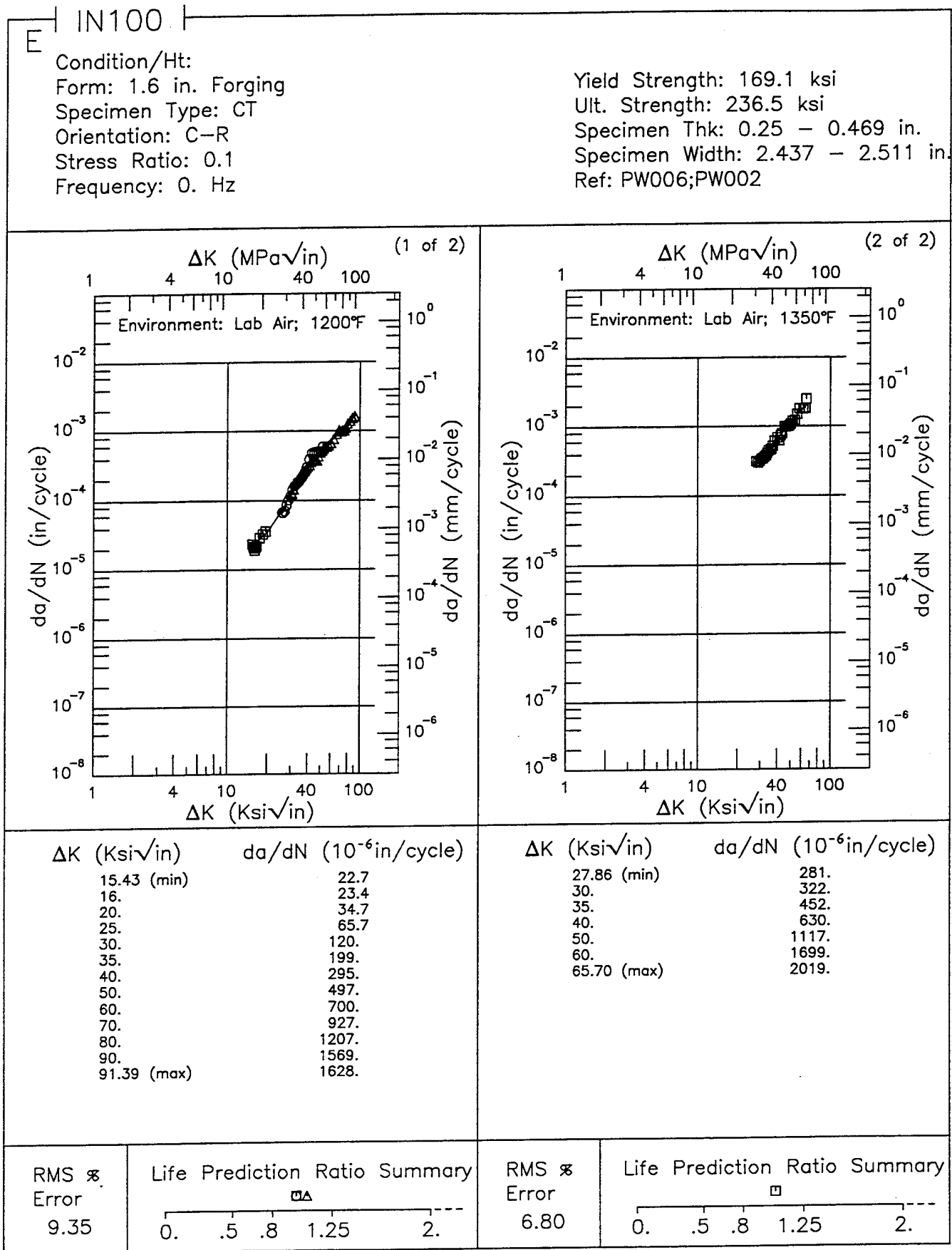


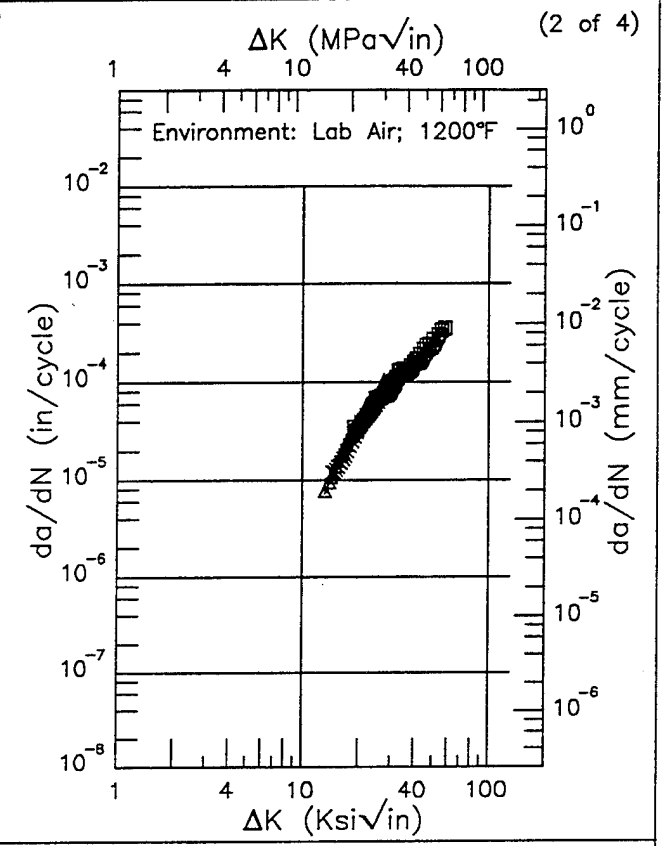
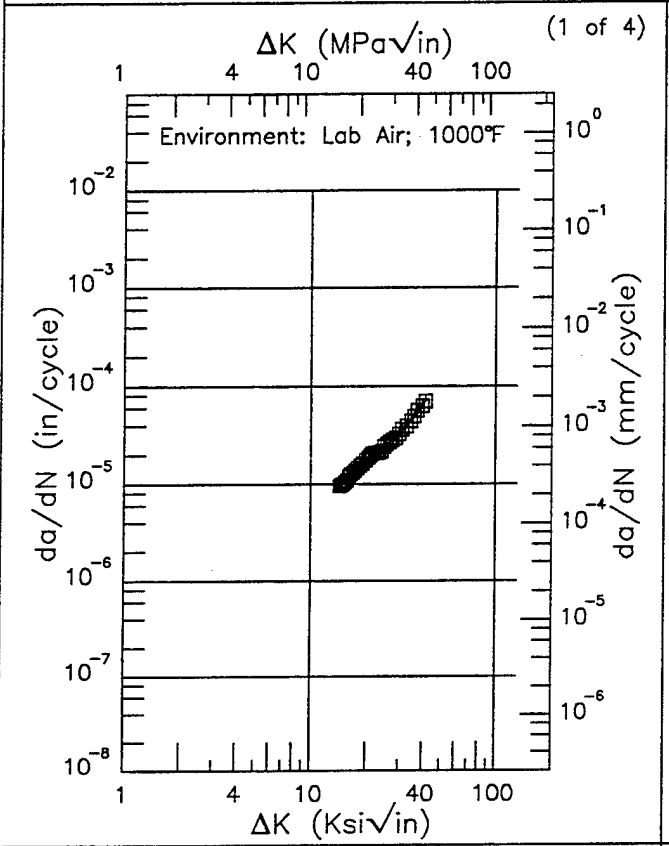
Figure 5.4.3.1.16

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E IN100

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.1
 Frequency: 0.2 Hz

Yield Strength: 169.1 ksi
 Ult. Strength: 236.5 ksi
 Specimen Thk: 0.25 - 0.87 in.
 Specimen Width: 2.493 - 2.511 in.
 Ref: PW002;PW006



ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
14.22 (min)	9.13
16.	11.9
20.	17.6
25.	24.6
30.	33.2
35.	45.8
40.	65.2
41.33 (max)	72.1

ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
13.17 (min)	7.25
16.	16.9
20.	35.3
25.	61.9
30.	89.9
35.	120.
40.	153.
50.	239.
57.86 (max)	338.

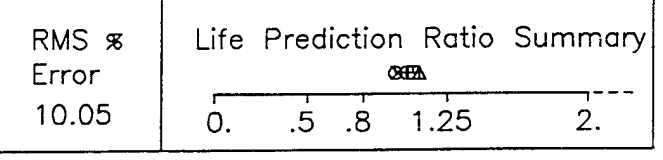
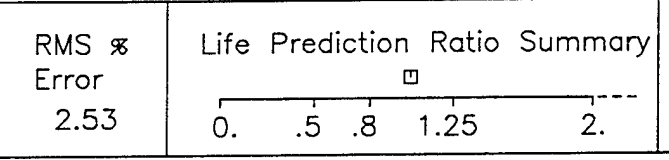
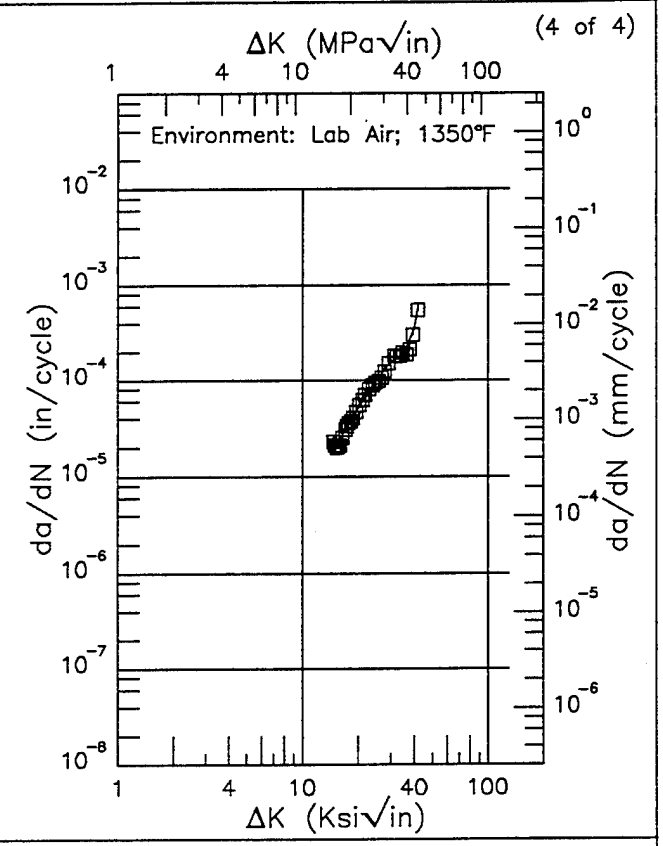
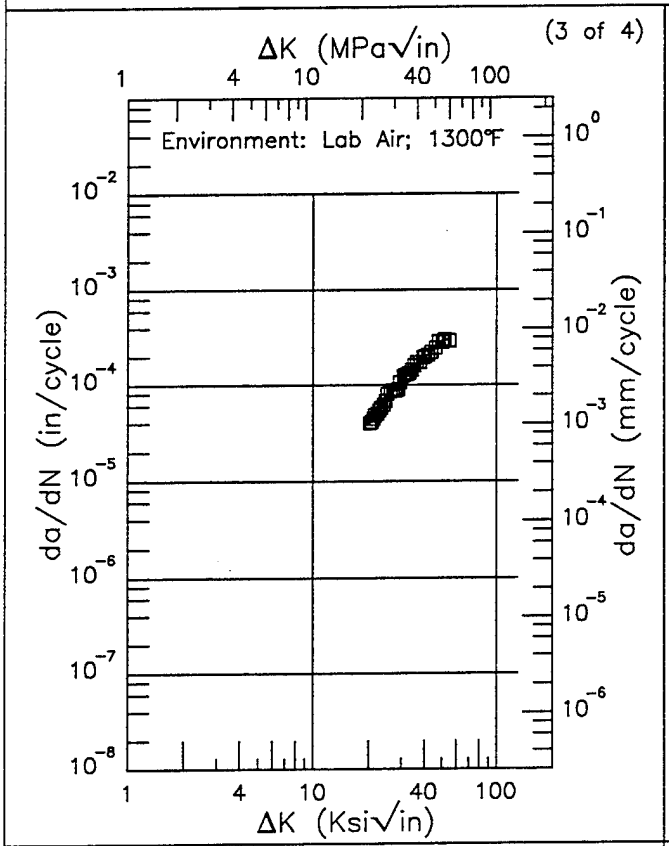


Figure 5.4.3.1.17

IN100 E

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.1
 Frequency: 0.2 Hz

Yield Strength: 169.1 ksi
 Ult. Strength: 236.5 ksi
 Specimen Thk: 0.25 - 0.87 in.
 Specimen Width: 2.493 - 2.511 in.
 Ref: PW002;PW006



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
20.20 (min)	39.0
25.	70.7
30.	110.
35.	153.
40.	197.
50.	280.
54.45 (max)	313.

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
14.68 (min)	17.7
16.	24.3
20.	50.8
25.	99.7
30.	157.
35.	186.
40.	355.
41.46 (max)	546.

RMS % Error 4.18	Life Prediction Ratio Summary
---------------------	-----------------------------------

RMS % Error 8.64	Life Prediction Ratio Summary
---------------------	-----------------------------------

Figure 5.4.3.1.17 (Concluded)

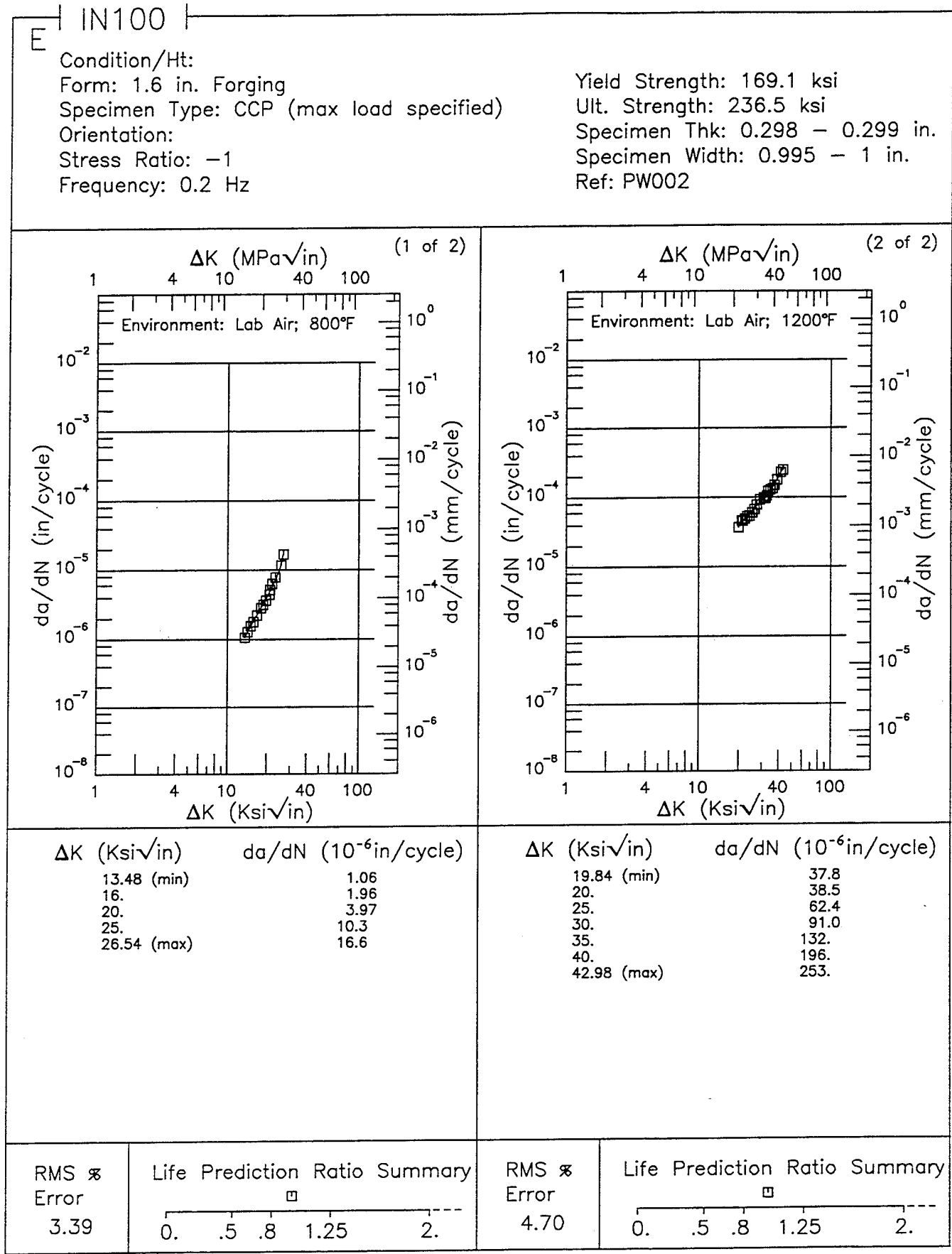
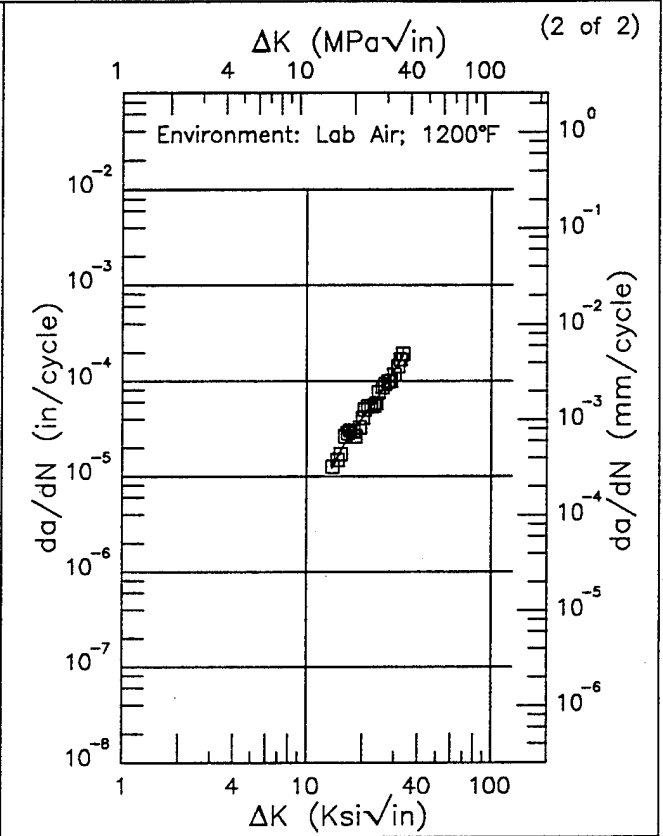
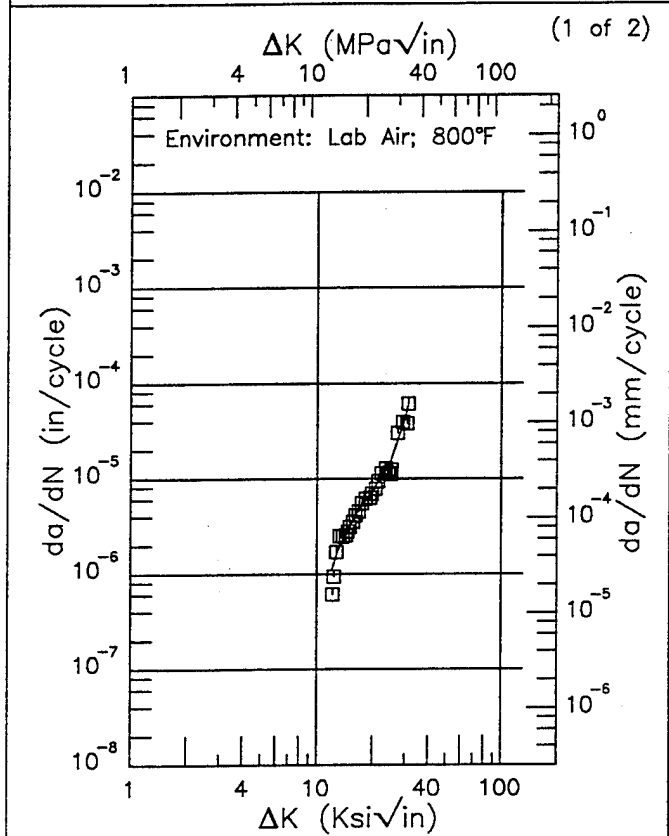


Figure 5.4.3.1.18

IN100 E

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation:
 Stress Ratio: -0.5
 Frequency: 0.2 Hz

Yield Strength: 169.1 ksi
 Ult. Strength: 236.5 ksi
 Specimen Thk: 0.228 - 0.309 in.
 Specimen Width: 0.988 - 1.038 in.
 Ref: PW002



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
12.10 (min)	1.13
13.	1.75
16.	4.11
20.	7.40
25.	15.4
30.	44.4
31.05 (max)	58.2

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
13.75 (min)	12.2
16.	21.5
20.	41.1
25.	73.7
30.	128.
33.24 (max)	186.

RMS $\%$ Error 16.76	Life Prediction Ratio Summary
----------------------------	-----------------------------------

RMS $\%$ Error 9.05	Life Prediction Ratio Summary
---------------------------	-----------------------------------

Figure 5.4.3.1.19

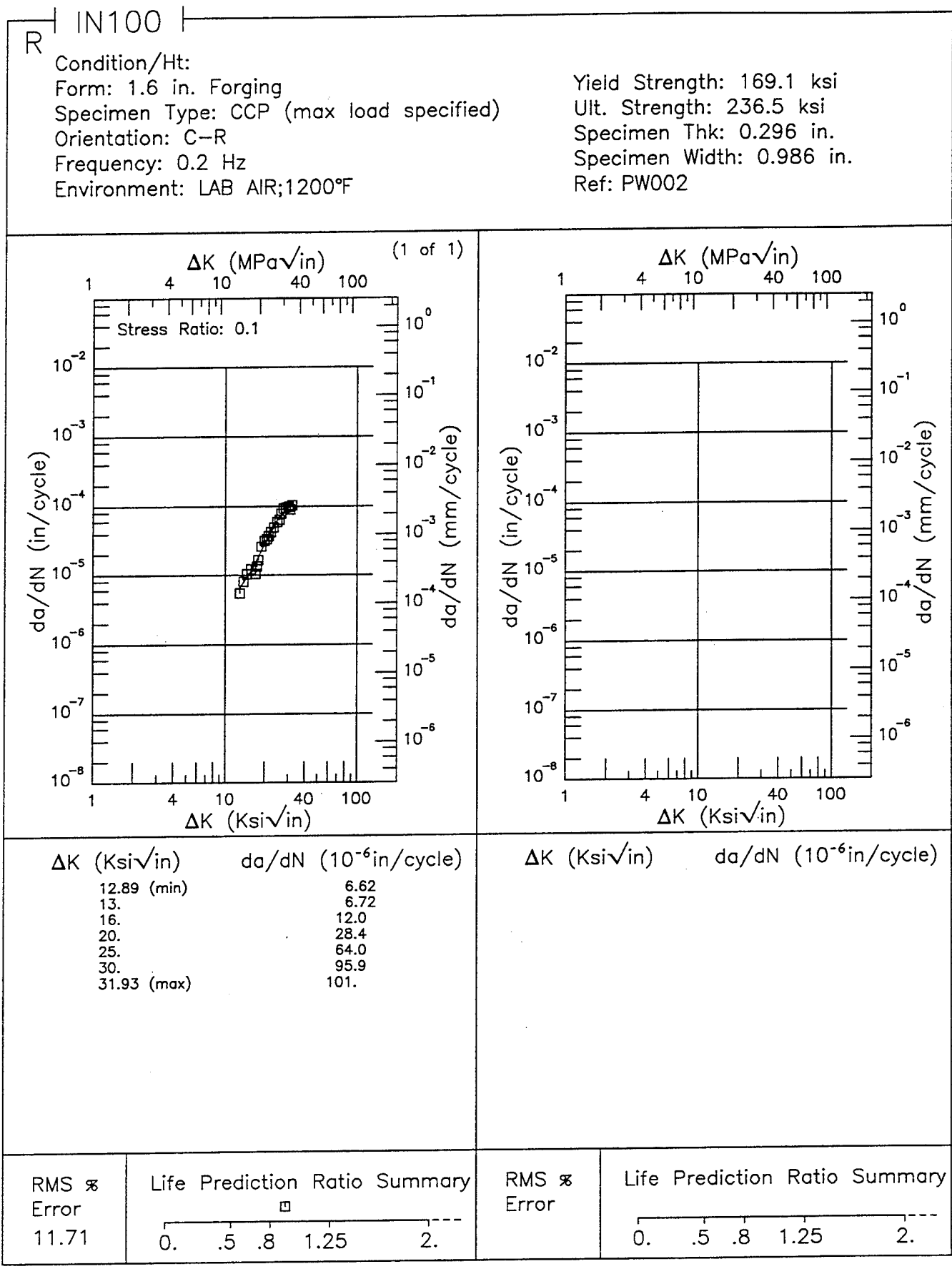


Figure 5.4.3.1.20

IN100 R

Condition/Ht: PRESTRAIN
 Form: 1.6 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation:
 Frequency: 0.2 Hz
 Environment: LAB AIR;1200°F

Yield Strength: 169.1 ksi
 Ult. Strength: 236.5 ksi
 Specimen Thk: 0.288 in.
 Specimen Width: 0.722 in.
 Ref: PW002

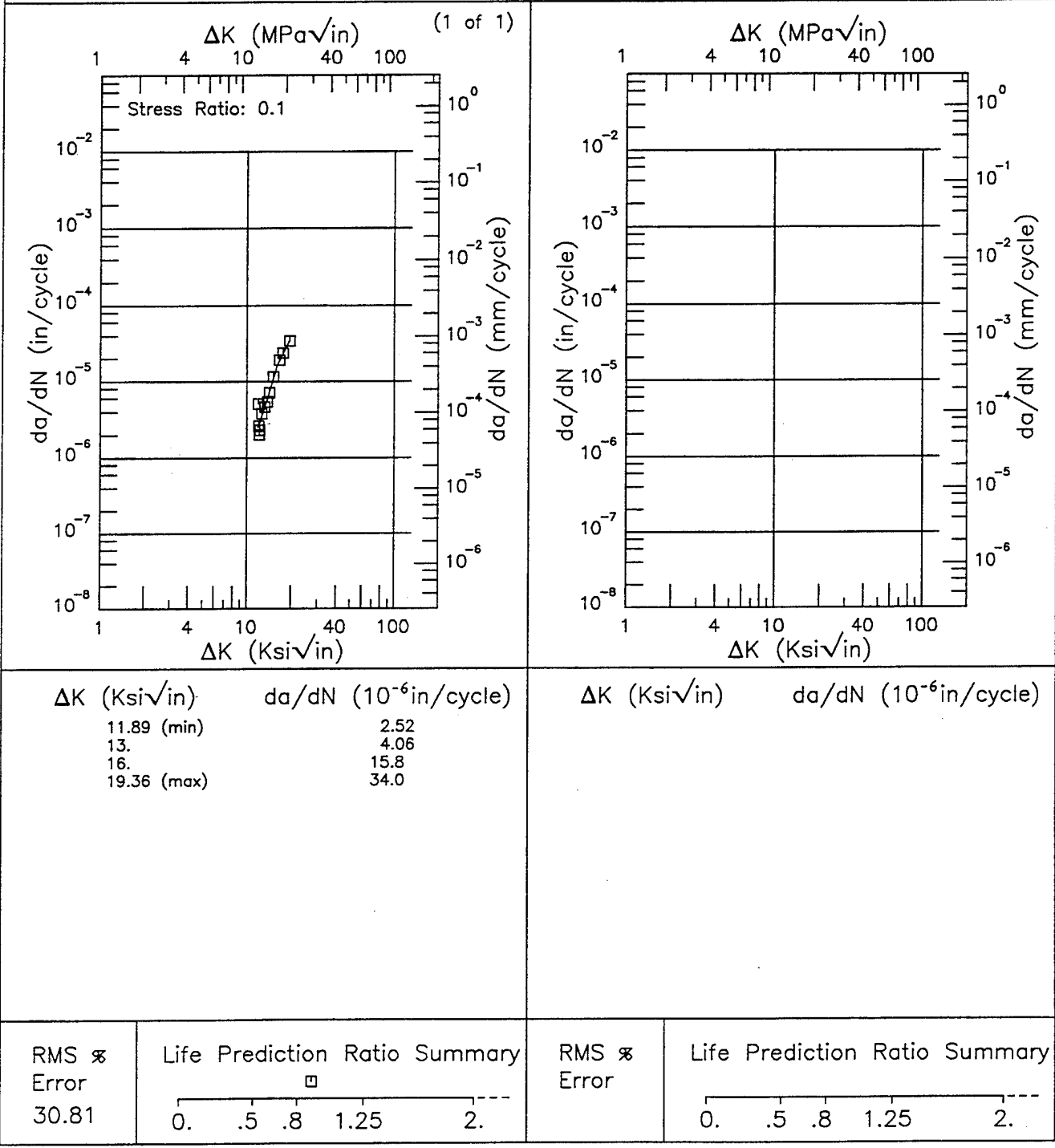


Figure 5.4.3.1.21

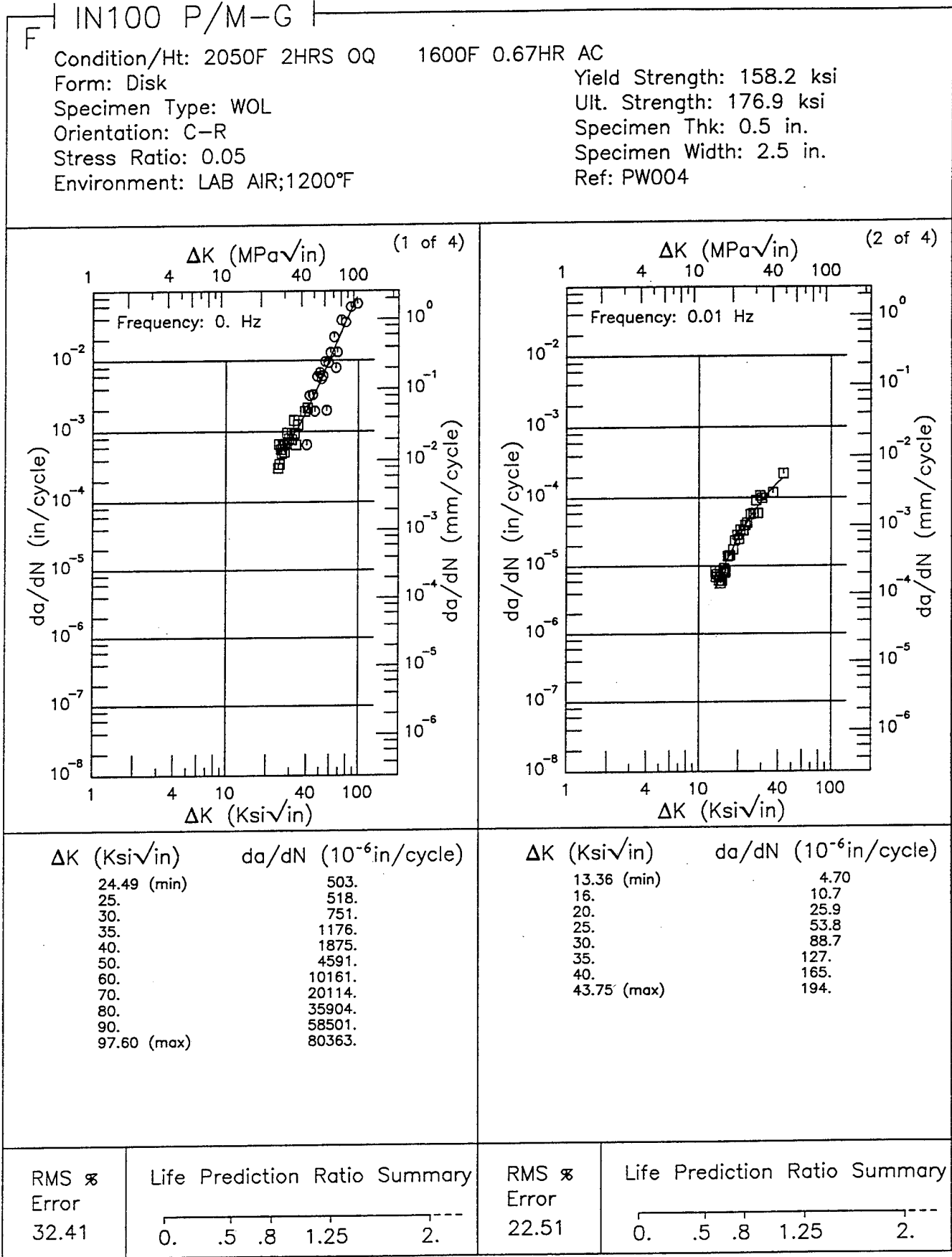
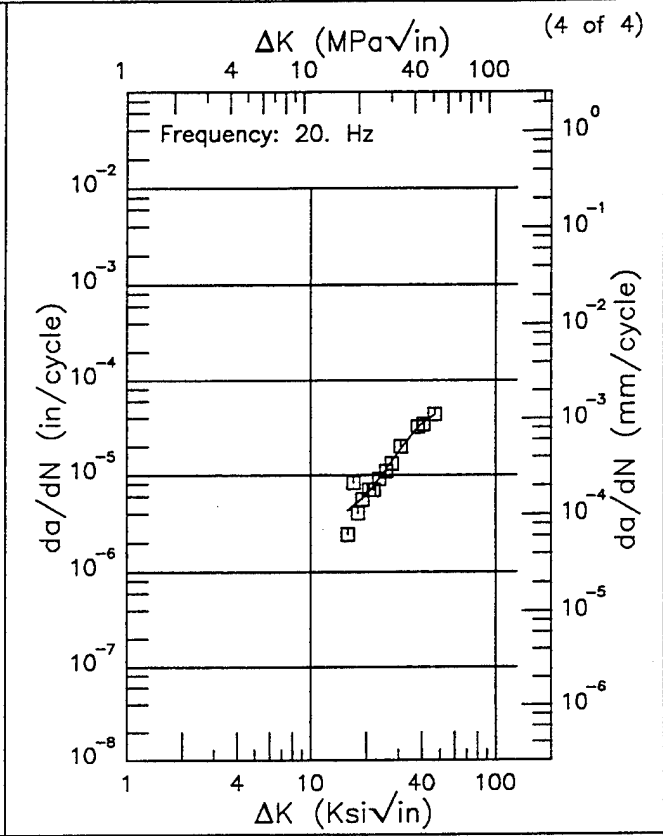
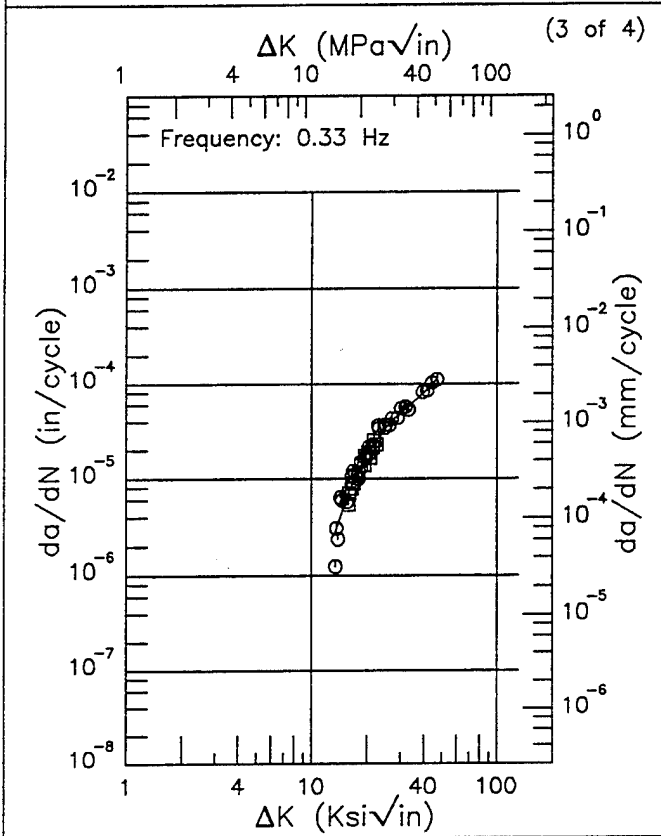


Figure 5.5.3.1

IN100 P/M-G F

Condition/Ht: 2050F 2HRS OQ 1600F 0.67HR AC
 Form: Disk
 Specimen Type: WOL
 Orientation: C-R
 Stress Ratio: 0.05
 Environment: LAB AIR;1200°F

Yield Strength: 158.2 ksi
 Ult. Strength: 176.9 ksi
 Specimen Thk: 0.5 in.
 Specimen Width: 2.5 in.
 Ref: PW004



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.44 (min)	2.78
16.	7.49
20.	18.5
25.	34.1
30.	49.2
35.	64.3
40.	81.4
46.90 (max)	112.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
15.92 (min)	4.26
16.	4.29
20.	6.15
25.	10.7
30.	17.5
35.	25.9
40.	34.5
46.70 (max)	43.3

RMS % Error 19.36	Life Prediction Ratio Summary 0. 0.5 0.8 1.25 2.
----------------------	---

RMS % Error 25.22	Life Prediction Ratio Summary 0. 0.5 0.8 1.25 2.
----------------------	---

Figure 5.5.3.1 (Concluded)

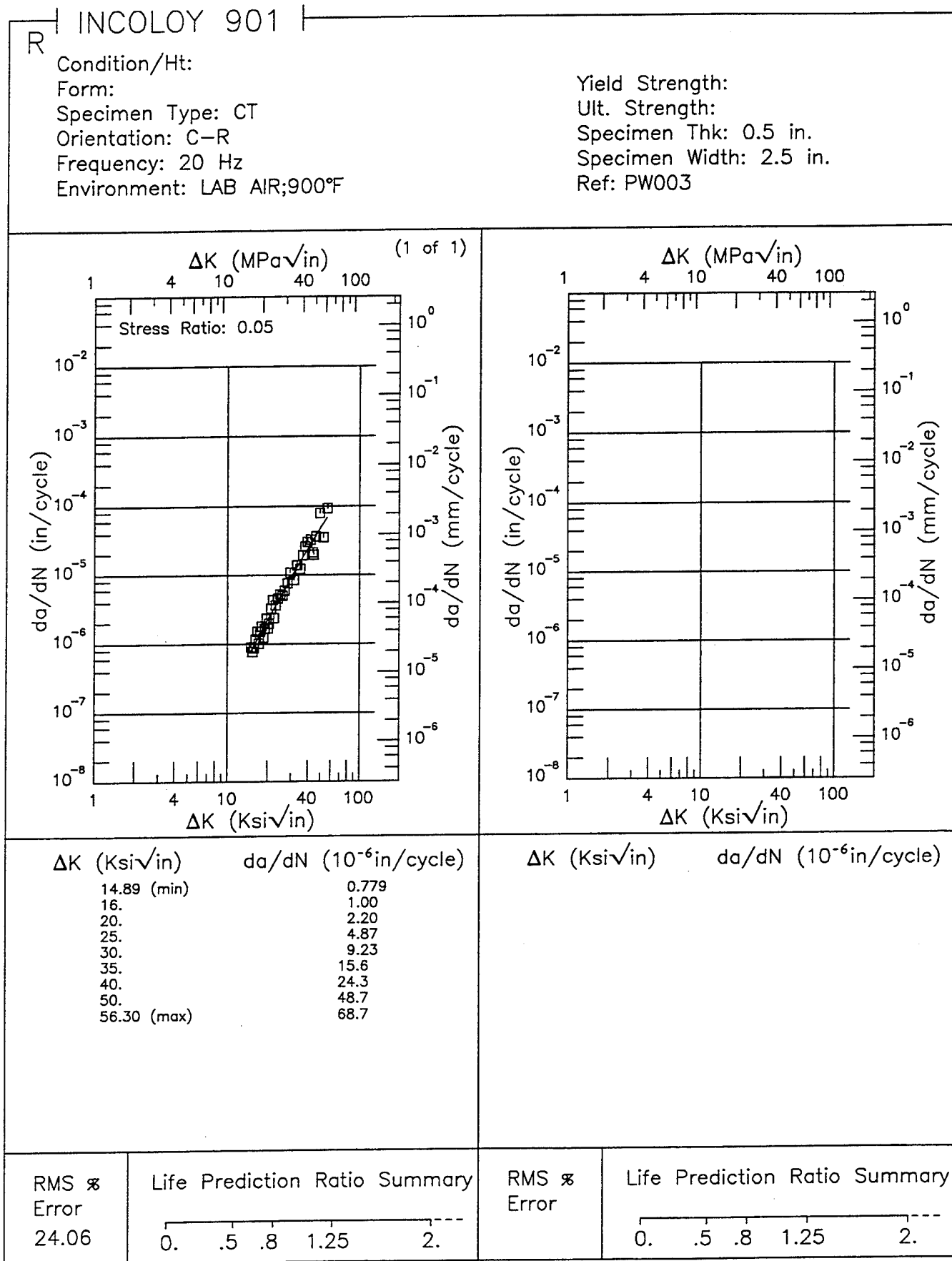


Figure 5.6.3.1

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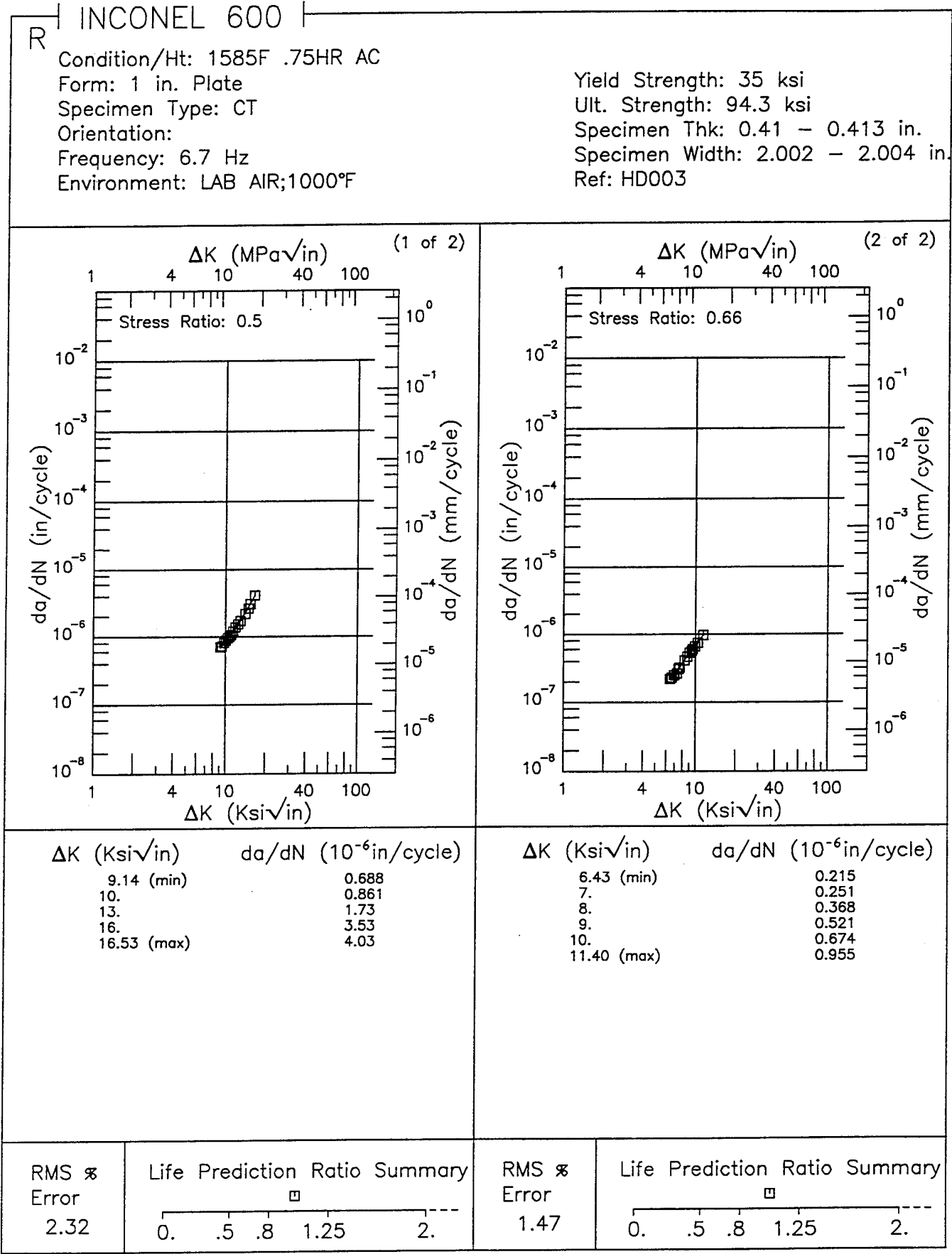


Figure 5.7.3.1.1

INCONEL 600

R

Condition/Ht: 1585F .75HR AC
 Form: 1 in. Plate
 Specimen Type: CT
 Orientation:
 Frequency: 6.7 Hz
 Environment: LAB AIR;800°F

Yield Strength: 35 ksi
 Ult. Strength: 94.3 ksi
 Specimen Thk: 0.413 in.
 Specimen Width: 2.002 in.
 Ref: HD003

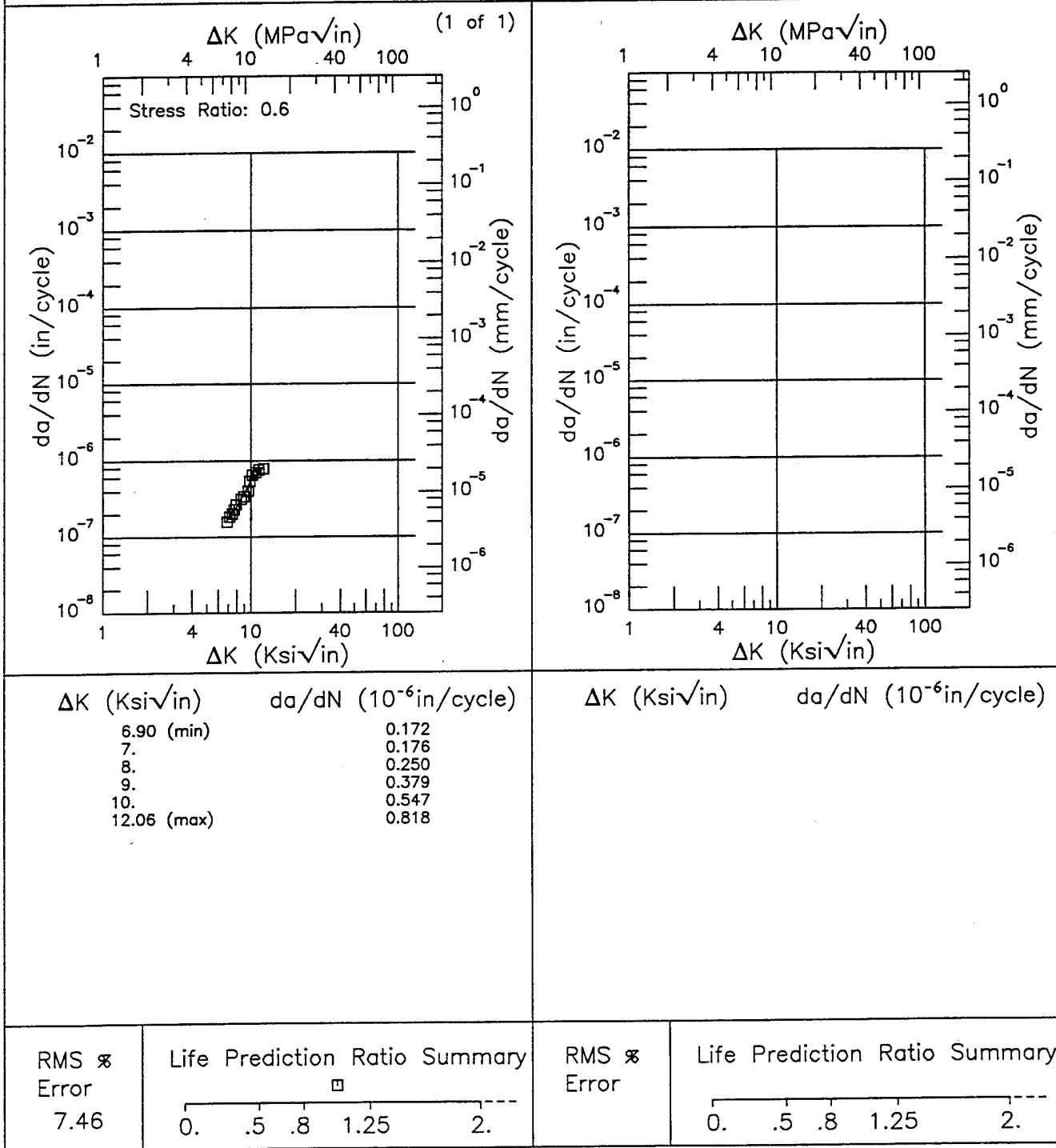


Figure 5.7.3.1.2

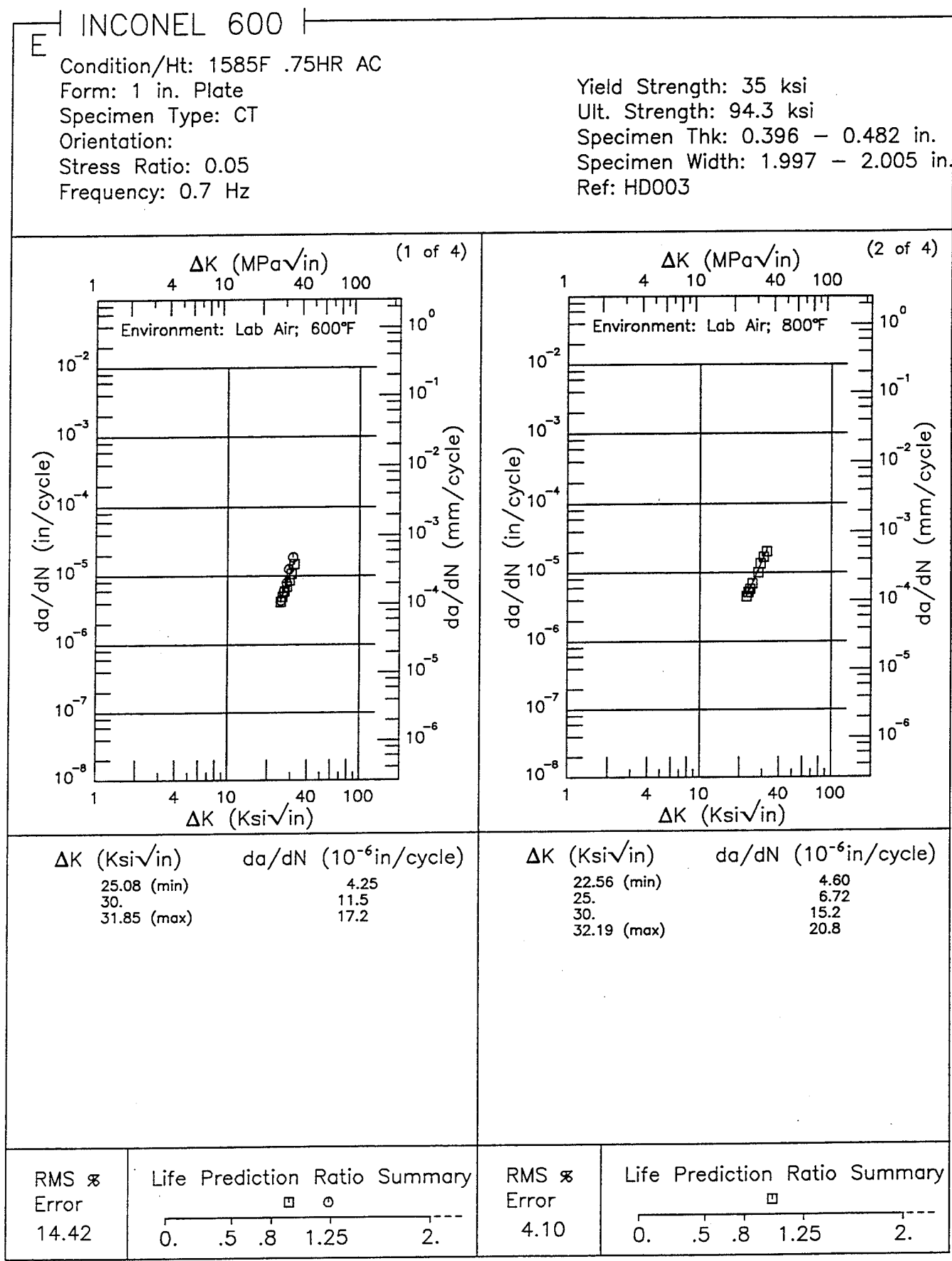


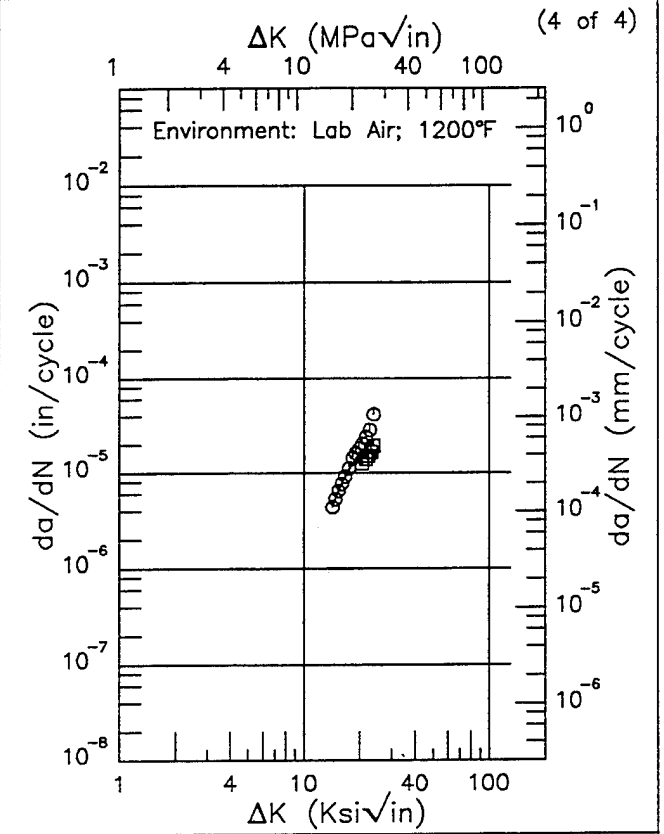
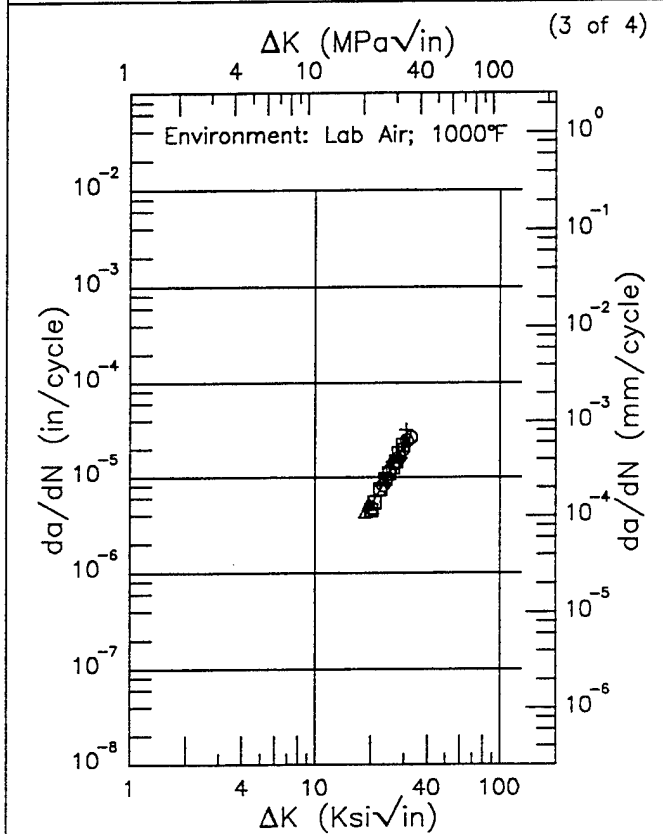
Figure 5.7.3.1.3

INCONEL 600

E

Condition/Ht: 1585F .75HR AC
 Form: 1 in. Plate
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

Yield Strength: 35 ksi
 Ult. Strength: 94.3 ksi
 Specimen Thk: 0.396 - 0.482 in.
 Specimen Width: 1.997 - 2.005 in.
 Ref: HD003



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
18.57 (min)	4.34
20.	5.04
25.	11.0
30.	22.0
32.64 (max)	28.2

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
14.24 (min)	4.53
16.	7.58
20.	16.7
23.73 (max)	21.2

RMS % Error 8.71	Life Prediction Ratio Summary
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RMS % Error 28.60	Life Prediction Ratio Summary
----------------------	-----------------------------------

Figure 5.7.3.1.3 (Concluded)

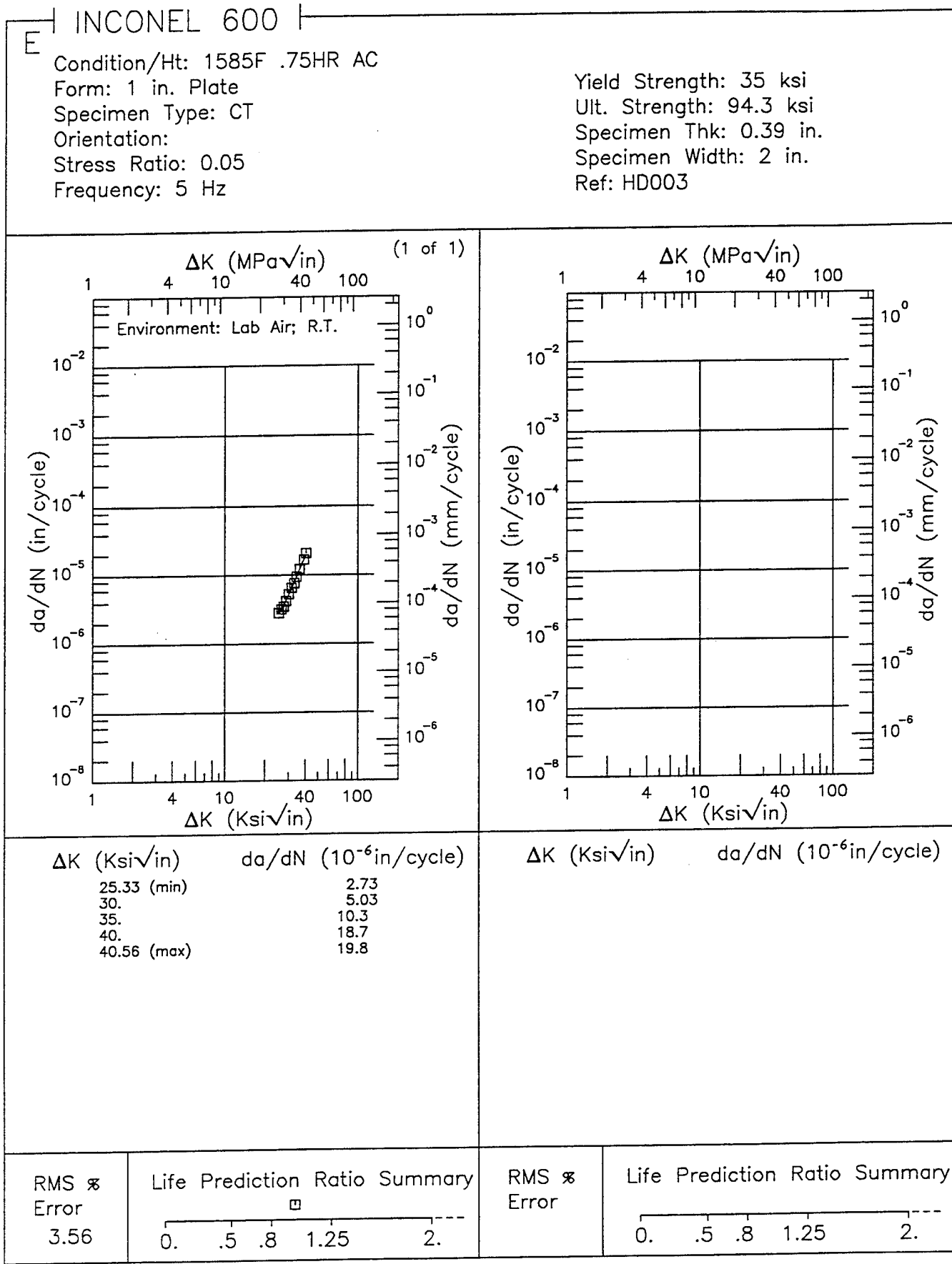


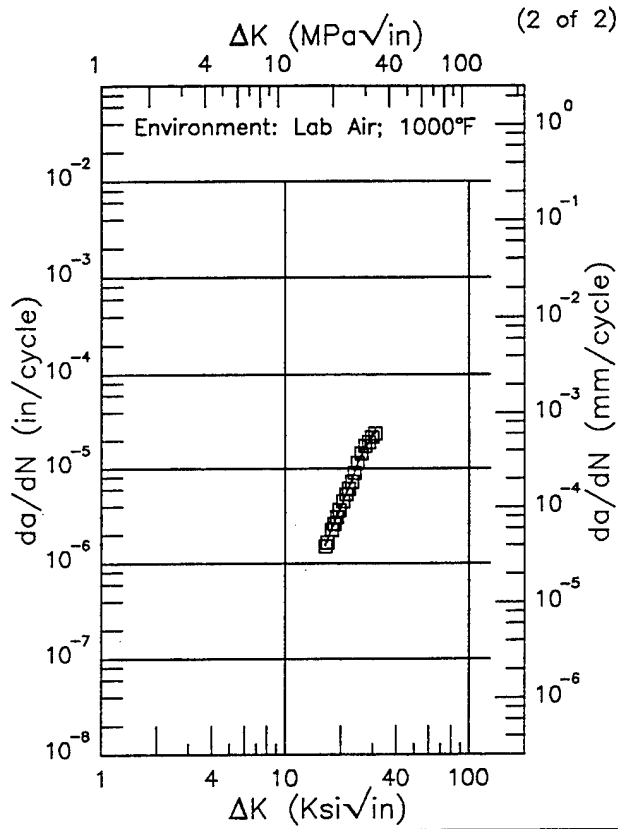
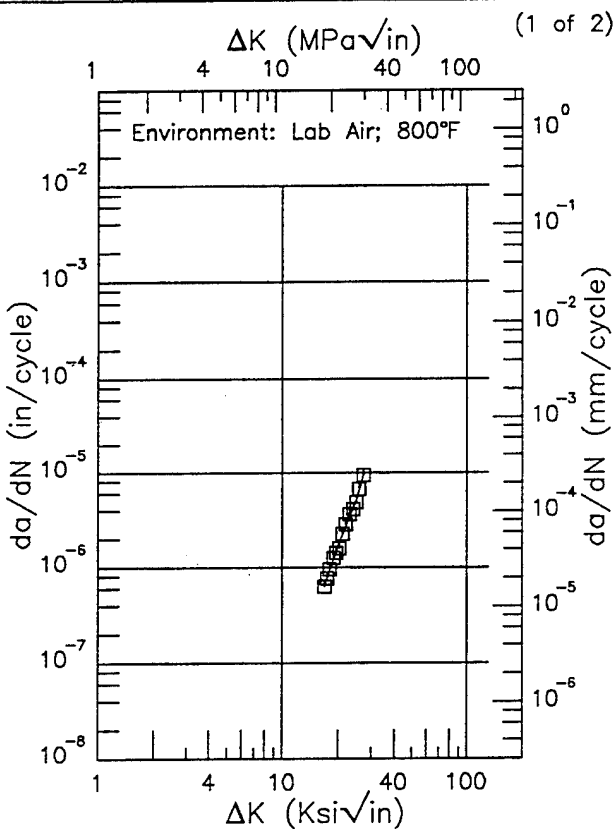
Figure 5.7.3.1.4

INCONEL 600

E

Condition/Ht: 1585F .75HR AC
 Form: 1 in. Plate
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05
 Frequency: 6.7 Hz

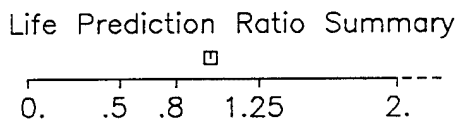
Yield Strength: 35 ksi
 Ult. Strength: 94.3 ksi
 Specimen Thk: 0.412 - 0.414 in.
 Specimen Width: 2.002 in.
 Ref: HD003



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
16.89 (min)	0.660
20.	1.71
25.	5.27
27.20 (max)	9.39

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
16.50 (min)	1.55
20.	3.85
25.	11.8
30.	22.9
30.56 (max)	23.9

RMS $\%$
 Error
 5.32



RMS $\%$
 Error
 3.90

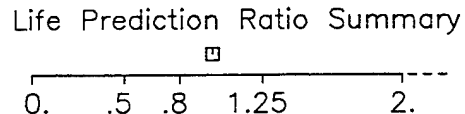


Figure 5.7.3.1.5

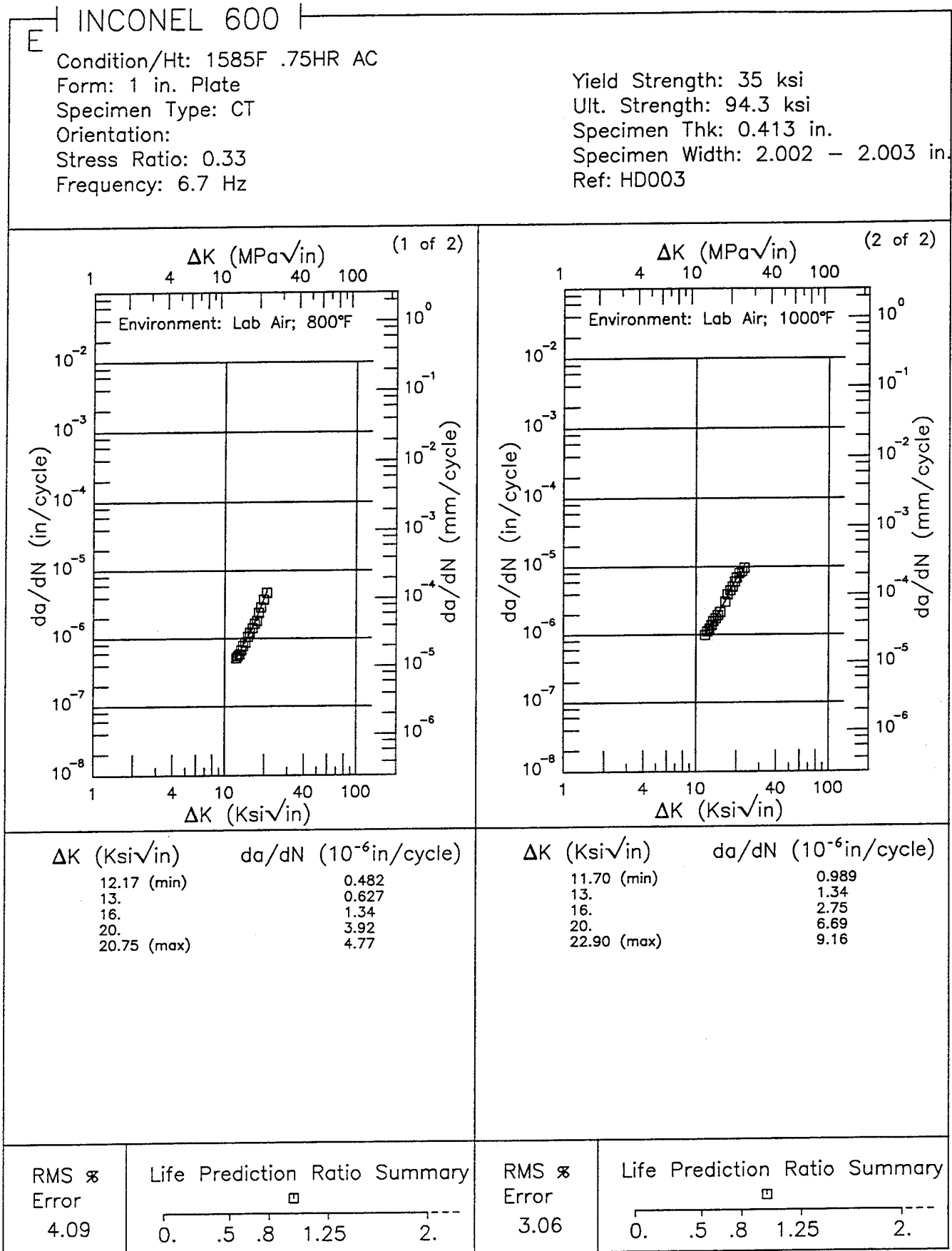


Figure 5.7.3.1.6

TABLE 5.8.1.2

1 of 1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 INCONEL 625 AT ROOM TEMPERATURE

ORIENTATION: Unspecified ENVIRONMENT: Lab Air

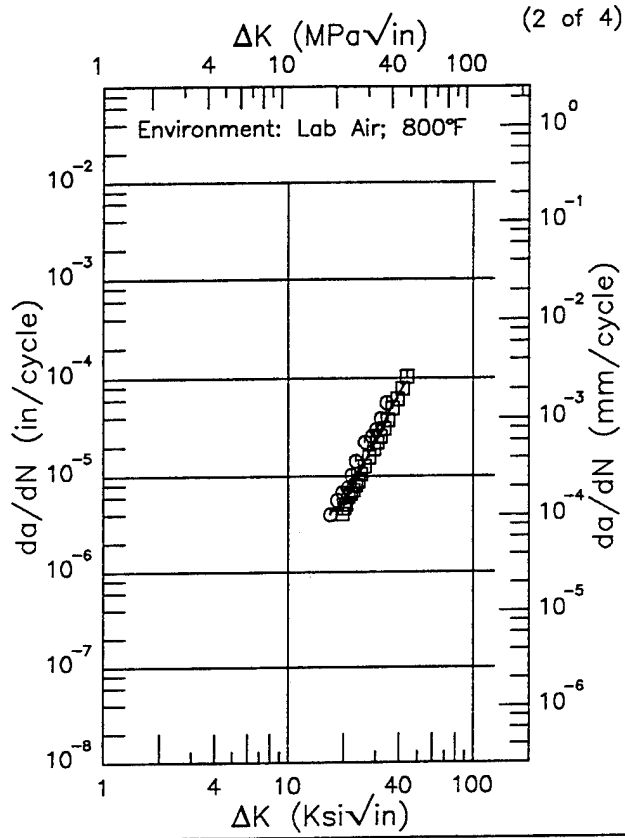
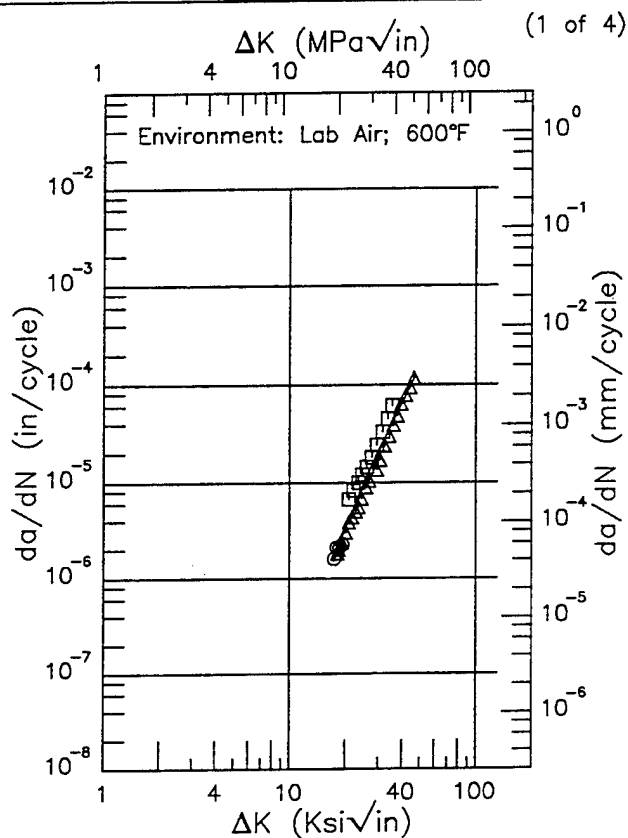
CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-8} in/cycle)						
				ΔK Level (K $\sqrt{\text{in}}$)						
MA	PLATE	0.05	10	2.5	5.0	10.0	20.0	50.0	100.0	

INCONEL 625

E

Condition/Ht: MA
 Form: 1 in. Plate
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

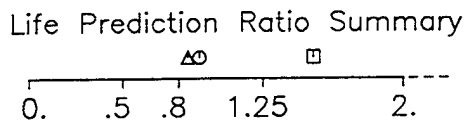
Yield Strength: 65.9 ksi
 Ult. Strength: 132.9 ksi
 Specimen Thk: 0.299 - 0.445 in.
 Specimen Width: 1.153 - 2.002 in.
 Ref: HD005



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
17.31 (min)	1.51
20.	3.33
25.	9.73
30.	21.2
35.	39.0
40.	65.0
46.44 (max)	115.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
16.88 (min)	4.04
20.	5.70
25.	11.7
30.	23.7
35.	43.2
40.	69.1
44.02 (max)	92.1

RMS \times
 Error
 25.35



RMS \times
 Error
 22.59

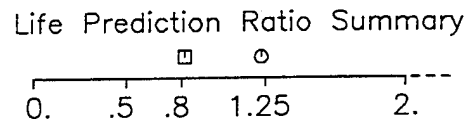


Figure 5.8.3.1.1

INCONEL 625

E

Condition/Ht: MA
 Form: 1 in. Plate
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

Yield Strength: 65.9 ksi
 Ult. Strength: 132.9 ksi
 Specimen Thk: 0.299 - 0.445 in.
 Specimen Width: 1.153 - 2.002 in.
 Ref: HD005

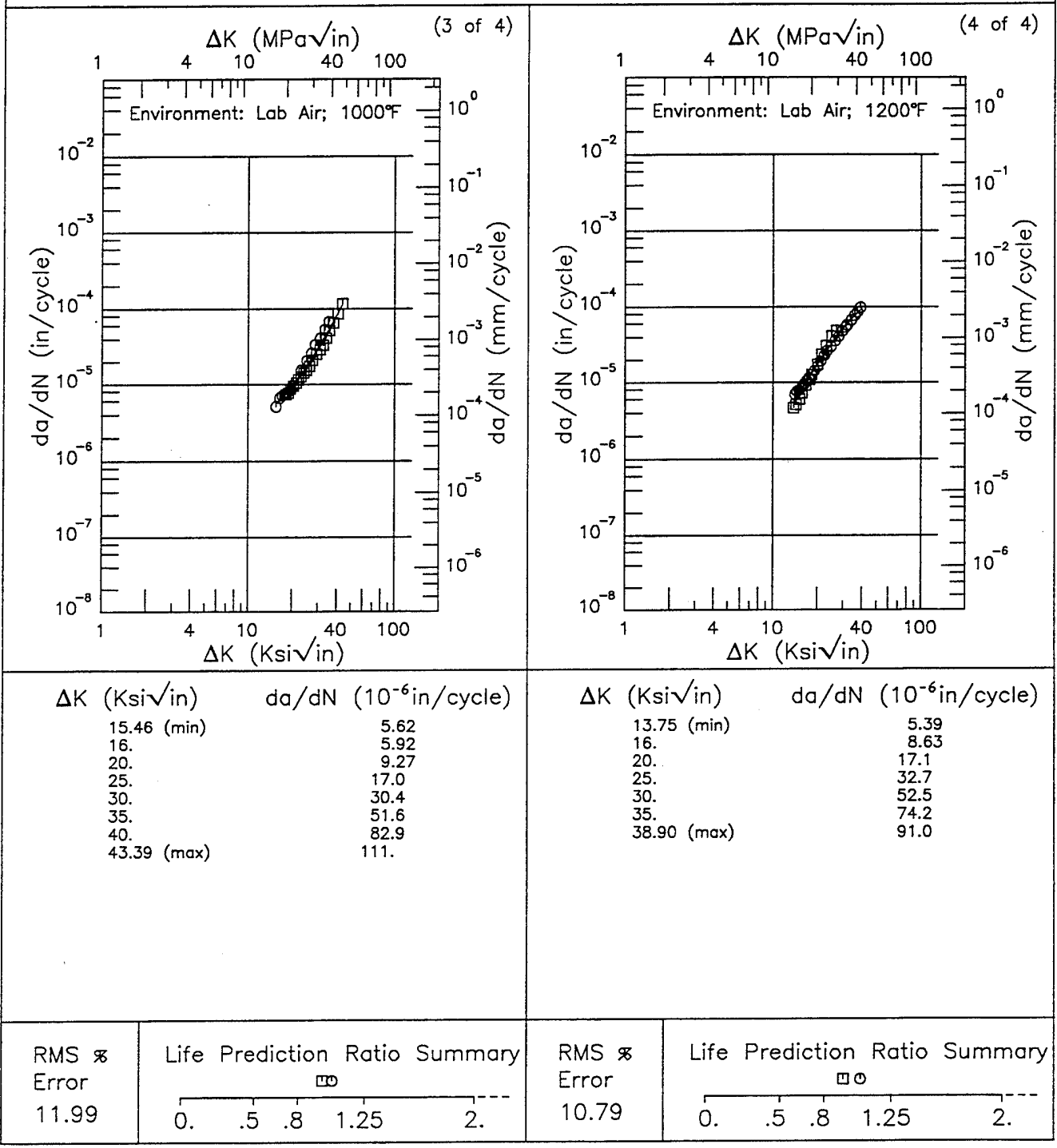


Figure 5.8.3.1.1 (Concluded)

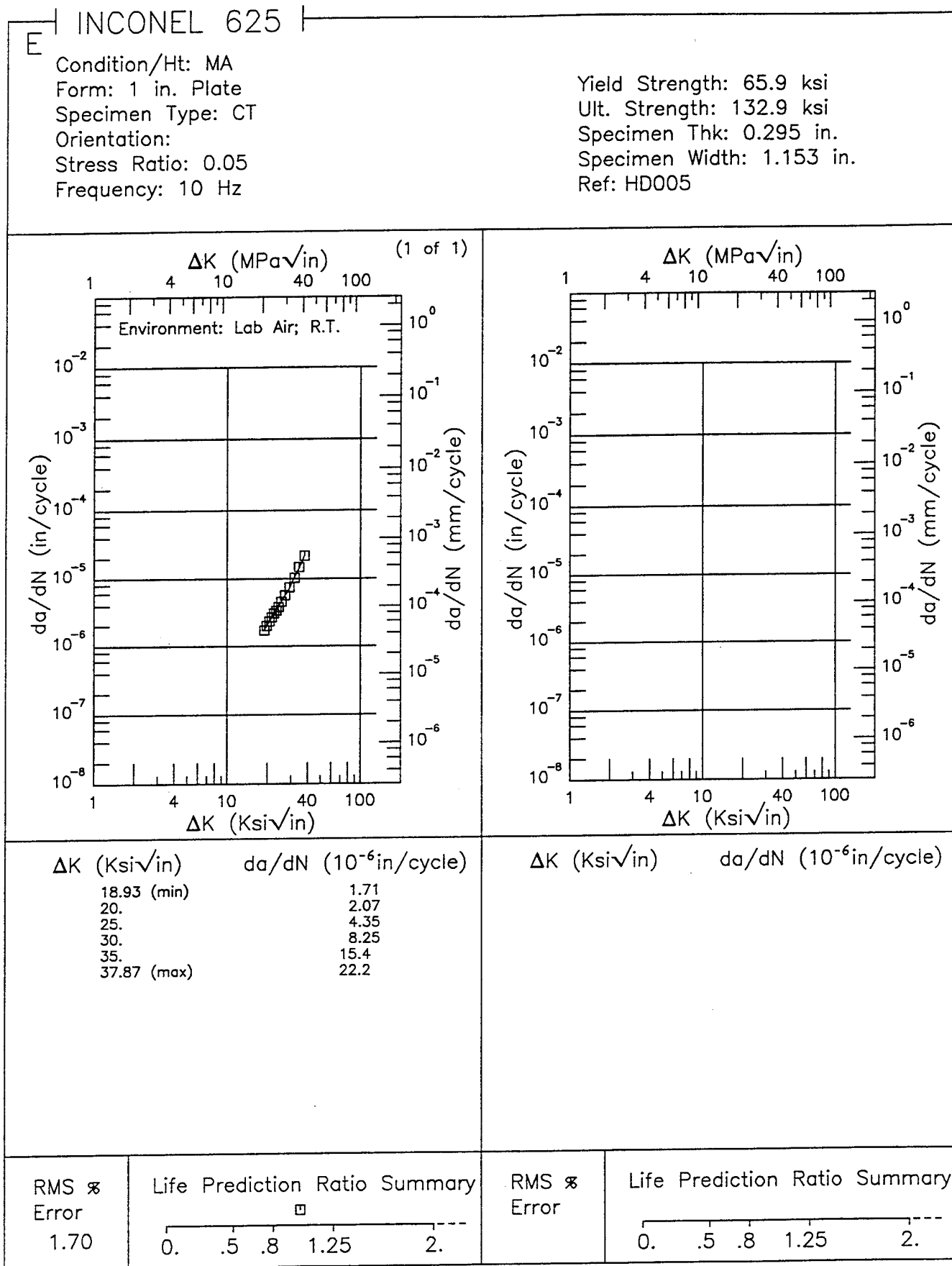


Figure 5.8.3.1.2

INCONEL 625 E

Condition/Ht: MA
 Form: 1 in. Plate
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05
 Frequency: 10 Hz

Yield Strength: 65.9 ksi
 Ult. Strength: 132.9 ksi
 Specimen Thk: 0.443 in.
 Specimen Width: 2.002 in.
 Ref: HD005

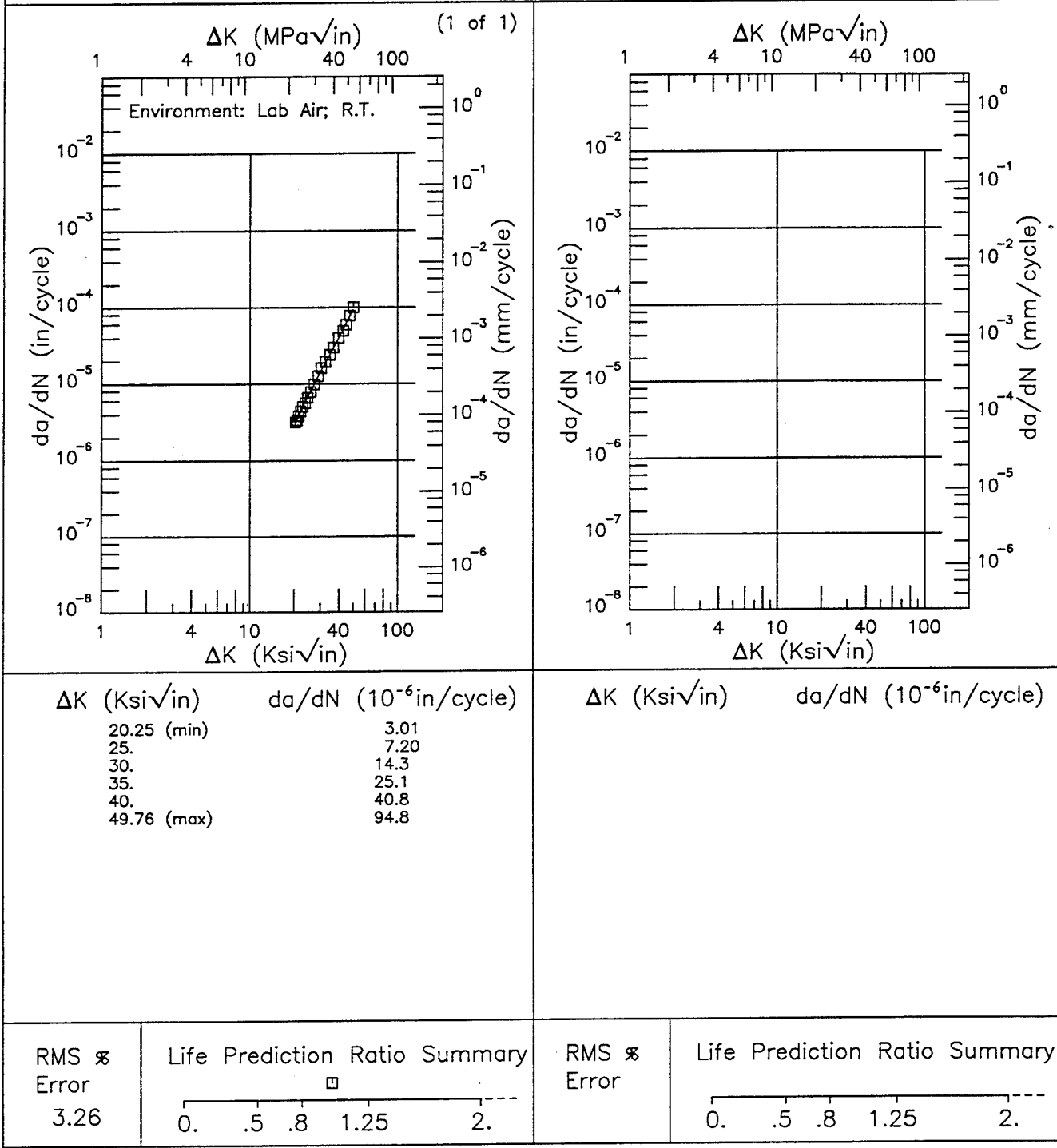


Figure 5.8.3.1.3

INCONEL 718

1 of 1

TABLE 5.9.1.1
MEAN PLANE STRAIN FRACTURE TOUGHNESS
FOR NICKEL BASE ALLOY INCONEL 718 AT ROOM TEMPERATURE

Product Form	Condition/Heat Treatment	K_{Ic} (ksi \sqrt{in})							
		Specimen Orientation				S-L			
		L-T		T-L		S-L		S-L	
Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	
Forging	STA	118.5	0.7	2	107.9	14.	2	---	---

TABLE 5.9.1.2.1

1 of 1

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE**

ORIENTATION: L-T ENVIRONMENT: Distilled Water

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-4} in/cycle)					
				ΔK Level (Kksi/in)					
				2.5	5.0	10.0	20.0	50.0	100.0
STA	FORGING	0.1	1				2	73.61	

INCONEL 718

1 of 1

TABLE 5.9.1.2.2

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: L.H.A.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi/in)						
				2.5	5.0	10.0	20.0	50.0	100.0	
1325F 8HRS FC TO 1150F	FORGING	0.08	6				1.13	19.35		
		0.5	6				2.17	33.96		
ST 1850F 1360 F 9HRS F/C 1175F	FORGED BAR	0.08	6					10.16		
		0.5	6				1.16	33.41		

TABLE 5.9.1.2.3

1 of 1

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE**

ORIENTATION: L-T**ENVIRONMENT: Lab Air**

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
1750F AC 1325F	PLATE	0.05	8.33				1.28	30.57		
1950F AC 1325F	PLATE	0.05	8.33				1.05			
ANNEALED 1800F 1 HR, AC, 1325F 8 HR, COOLED AT 100F PER HOUR TO 1150F, 1150F 8HR, AC	PLATE	-0.2	10				1.14			
		0.02	4				0.92	16.4		
		0.02	20				1.75			
STA	FORGING	0.5	10				2.97			
		0.1	15-20				2.45	57.53		
		0.4	20				3.42			
UNSPECIFIED	UNSPECIFIED	0.8	20-30			0.47				
		0.1	10			0.13	2.46	56.78		

INCONEL 718

1 of 1

TABLE 5.9.1.2.4

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: S.T.W.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Kksi/in)					
				2.5	5.0	10.0	20.0	50.0	100.0
ST 1850F 1360 F 9HRS F/C 1175F	FORGED BAR	0.08	1						11.87

TABLE 5.9.1.2.5

1 of 1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE

ORIENTATION: L-T		ENVIRONMENT: Vacuum							
CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCCR (10 ⁻⁶ in/cycle)					
				ΔK Level (Ksi/in)					
				2.5	5.0	10.0	20.0	50.0	100.0
ANNEALED 1800F 1 HR, AC, 1325F 8 HR, COOLED AT 100F PER HOUR TO 1150F, 1150F 8HR, AC	PLATE	0.1	10				1.02	25.74	

INCONEL 718

1 of 1

TABLE 5.9.1.2.6

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE

ORIENTATION: T-L		ENVIRONMENT: Distilled Water								
CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-8} in/cycle)						
				ΔK Level (K _{max} /in)						
STA	FORGING	0.1	1	2.5	5.0	10.0	20.0	50.0	100.0	
		0.8	1				2.3	127.2		
						0.88				

TABLE 5.9.1.2.7

1 of 1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: L.H.A.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi/in)						
				2.5	5.0	10.0	50.0	100.0		
1325F 8HRS FC TO 1150F	FORGING	0.08	6							
ST 1850F 1360 F 9HRS F/C 1175F	FORGED BAR	0.08	6						30.86	16.09

INCONEL 718

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TABLE 5.9.1.2.8

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE**

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	ENVIRONMENT: Lab Air						
				ORIENTATION: T-L						
				FCGR (10^{-6} in/cycle)						
1750F 1HR AC 1325F	FORGED BAR	0.05	8.33-10	2.5	5.0	10.0	20.0	50.0	100.0	
				AK Level (Kksi/in)						
							3.08	116.49		
							0.69	29.84		
							2.19	83.07		
STA	FORGING	0.4	20				3.16			
							0.47			

TABLE 5.9.1.2.9

1 of 1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE

ORIENTATION: T-L

ENVIRONMENT: S.C.S.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (K $\sqrt{\text{in}}$)					
				2.5	5.0	10.0	20.0	50.0	100.0
ST 1850F 1360 F 9HRS F/C 1175F	FORGED BAR	0.08	1						10.23

INCONEL 718

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TABLE 5.9.1.2.10

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: S.T.W.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Kksi/in)					
				2.5	5.0	10.0	20.0	50.0	100.0
ST 1850F 1360 F 9HRS F/C 1175F	FORGED BAR	0.08	1						8.23

TABLE 5.9.1.2.11

1 of 1

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE**

ORIENTATION: S-L ENVIRONMENT: L.H.A.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi/in)					
				2.5	5.0	10.0	20.0	50.0	100.0
ST 1850F 1360 F 9HRS F/C 1175F	FORGED BAR	0.08	6						26.94

INCONEL 718

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TABLE 5.9.1.2.12

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE**

CONDITION/ HEAT TREATMENT		PRODUCT FORM	R	FREQ (Hz)	ENVIRONMENT: Lab Air				
					ORIENTATION: C-R				
		$FCGR (10^{-6} \text{ in/cycle})$							
		$\Delta K \text{ Level (Kak/in)}$							
		2.5	5.0	10.0	20.0	50.0	100.0		
1750F AC 1325F	FORGED BAR	0.05		10				2.56	43.72
1760F 1HR WQ 1325F 8HRS	DISK	0.		0.33				3.39	

TABLE 5.9.1.2.13

1 of 1

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
INCONEL 718 AT ROOM TEMPERATURE**

CONDITION/ HEAT TREATMENT		PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
					ΔK Level (Ksi/in)					
					2.5	5.0	10.0	20.0	50.0	100.0
1750F 1HR AC 1325F	PLATE	0.05	10					1.95		
	SHEET	0.05	5					1.95		
1750F AC 1325F	FORGING	0.05	10					1.93	40.69	

ORIENTATION: Unspecified

ENVIRONMENT: Lab Air

INCONEL 718

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TABLE 5.9.2.1

INCONEL 718 K_{Ic}																													
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic} \cdot TYS)^*$	K_{Ic}			DATE	REFER														
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Ksi $\sqrt{in.}$)	K_{Ic} MEAN	STAN DEV																
1325F 8 HR FC TO 1150F HOLD 18 HR	Forging	3.00	R.T.	T-L	160.0	4.022	1.120	CT	1.987	0.63	74.00	---	---	1973	88187														
																1325F 9 HR FC TO 1150F AT 100F/HR HOLD AT 1150F 8 HR AC	Forging	3.30	R.T.	151.0	2.002	0.746	CT	1.949	0.68	78.80	---	1973	88187
ST-CW-A	Round Bar	1.39	0	210.0	1.202	0.600	NB	0.573	0.15	51.80	---	---	1991	NH006															
															STA	Forging	3.00	-65	168.6	3.006	1.498	CT	1.576	1.33	122.80	---	1987	DA007	
																													STA
STA	Forging	3.00	T-L	172.4	3.006	1.495	CT	1.656	1.03	110.90	---	---	1987	DA007															
															STA	Forging	3.00	L-T	161.2	3.007	1.489	CT	1.567	1.34	118.00	---	1987	DA007	
																													STA
STA	Forging	3.00	R.T.	155.4	3.006	1.500	CT	1.593	0.99	98.00	---	---	1987	DA007															
															STA	Forging	1.39	T-L	179.0	1.199	0.599	NB	0.737	0.47	77.30	---	1991	NH006	
																													STA
STA	Round Bar	1.39	0	179.0	1.199	0.599	NB	0.615	0.54	82.90	---	---	1991	NH006															
															STA	Round Bar	1.39	L-R	179.0	1.200	0.600	NB	0.619	0.56	84.90	---	---	1991	NH006
STA	Round Bar	1.39	0	179.0	1.199	0.599	NB	0.615	0.54	82.90	---	---	1991	NH006															
															STA	Round Bar	1.39	L-R	179.0	1.200	0.600	NB	0.619	0.56	84.90	---	---	1991	NH006

TABLE 5.9.2.2

CONDITIONS AT TREAT		PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (ksi)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K _{app}			K _G			DATE	REFER
FORM	THICK (in.)	WIDTH (in.)	THICK (in.)				INIT (in.)	FINAL (in.)	ONSET (ksi)	MAX (ksi)	K _{app} (ksi) √(in.)	K _{app} MEAN	STAN DEV	K _G (ksi) √(in.)	K _G MEAN	STAN DEV				
Cold Rolled 90 Percent and Aged	Sheet	0.03	0.03	-423	L-T	269.0	4.010	0.027	1.270	1.450	---	116.00	174.77	175.8	5.7	190.67	191.1	7.5	1964	60578
		0.03	0.03			269.0	4.010	0.027	1.240	1.370	---	117.00	173.64			185.14			1964	60578
		0.03	0.03			269.0	4.010	0.027	1.250	1.400	---	120.00	178.99			192.64			1964	60578
		0.03	0.03			269.0	4.010	0.027	1.270	1.400	---	119.00	179.29			191.03			1964	60578
		0.03	0.03			269.0	4.020	0.027	1.750	2.050	---	92.50	174.19			198.97			1964	60578
		0.03	0.03			269.0	4.000	0.028	1.310	1.340	---	116.00	178.34			180.98			1964	60578
		0.03	0.03			269.0	4.000	0.028	1.280	1.420	---	117.00	177.22			189.70			1964	60578
		0.03	0.03			269.0	4.000	0.028	1.370	1.480	---	112.00	177.30			186.79			1964	60578
		0.03	0.03			269.0	4.000	0.028	1.740	2.000	---	90.50	169.90			190.76			1964	60578
		0.03	0.03			269.0	4.010	0.028	1.240	1.500	---	119.00	176.61			200.22			1964	60578
		0.03	0.03			269.0	4.010	0.028	1.740	2.000	---	87.50	164.15			184.25			1964	60578
		0.03	0.03			269.0	4.010	0.028	0.750	0.890	---	168.00	186.38			204.90*			1964	60578
		0.03	0.03			269.0	4.010	0.028	1.250	1.400	---	118.00	176.01			189.43			1964	60578
		0.03	0.03			269.0	4.010	0.028	0.740	0.950	---	169.00	186.13			213.90*			1964	60578
		0.03	0.03			269.0	4.010	0.028	1.770	2.050	---	91.50	173.97			197.02			1964	60578
		0.03	0.03			269.0	4.010	0.028	0.750	1.020	---	158.00	175.29			208.37			1964	60578
0.03	0.03	269.0	4.010	0.028	1.730	1.920	---	89.00	166.22	180.87	1964	60578								

BUCKLING OF CRACK EDGES NOT RESTRAINED

* NOTE: NET SECTION STRESS EXCEEDS 80% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

INCONEL 718

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TABLE 5.9.2.2 (CONTINUED)

INCONEL 718 K _C																	
CONDITIONS AT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K _{app}		K _C		DATE	REFER
	FORM	THICK (in.)				WIDTH (in.)	THICK (in.)	B	INIT (in.)	FINAL (in.)	ONSET (KSI)	MAX (KSI)	K _{app} (KSI)	MEAN	STAN DEV		
Cold Rolled 30 Percent and Aged	Sheet	0.03	4.000	0.027	1.310	1.620	---	123.00	189.10	218.77	180.4	8.8	200.6	13.8	218.77	1964	60578
		0.03	4.010	0.027	1.250	1.400	---	126.00	187.94	202.27	180.4	8.8	200.6	13.8	202.27	1964	60578
		0.03	3.980	0.028	1.330	1.560	---	119.00	184.90	206.22	180.4	8.8	200.6	13.8	206.22	1964	60578
		0.03	4.000	0.028	0.730	0.930	---	169.00	184.78	211.35*	180.4	8.8	200.6	13.8	211.35*	1964	60578
		0.03	4.000	0.028	1.270	1.650	---	121.00	182.37	218.16	180.4	8.8	200.6	13.8	218.16	1964	60578
		0.03	4.000	0.028	0.730	0.830	---	171.00	186.97*	200.60*	180.4	8.8	200.6	13.8	200.60*	1964	60578
		0.03	4.000	0.028	1.290	1.650	---	118.00	179.63	212.75	180.4	8.8	200.6	13.8	212.75	1964	60578
		0.03	4.010	0.028	1.730	1.920	---	92.00	171.83	186.97	180.4	8.8	200.6	13.8	186.97	1964	60578
		0.03	4.010	0.028	1.300	1.750	---	125.00	191.16	235.55*	180.4	8.8	200.6	13.8	235.55*	1964	60578
		0.03	4.010	0.028	1.250	1.450	---	119.00	177.50	195.61	180.4	8.8	200.6	13.8	195.61	1964	60578
		0.03	4.010	0.028	1.730	1.790	---	89.10	166.41	170.92	180.4	8.8	200.6	13.8	170.92	1964	60578
		0.03	4.010	0.028	1.720	2.000	---	90.30	167.90	190.15	180.4	8.8	200.6	13.8	190.15	1964	60578
		0.03	4.010	0.028	1.730	2.090	---	81.90	171.64	201.46	180.4	8.8	200.6	13.8	201.46	1964	60578
		0.03	4.010	0.028	0.740	0.850	---	177.00	194.94*	224.02*	180.4	8.8	200.6	13.8	224.02*	1964	60578
		0.03	4.010	0.028	0.730	0.850	---	170.00	185.86*	202.06*	180.4	8.8	200.6	13.8	202.06*	1964	60578
		0.03	4.010	0.028	1.230	1.400	---	127.00	187.53	203.87	180.4	8.8	200.6	13.8	203.87	1964	60578
0.03	4.010	0.028	1.250	1.450	---	129.00	192.42	212.04	180.4	8.8	200.6	13.8	212.04	1964	60578		
0.03	4.010	0.028	1.720	1.950	---	91.90	170.88	189.27	180.4	8.8	200.6	13.8	189.27	1964	60578		

HUCKLING OF CRACK EDGES NOT RESTRAINED

* NOTE: NET SECTION STRESS EXCEEDS 80% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

TABLE 5.9.2.2 (CONTINUED)

INCONEL 718 K _C																			
CONDITIONE AT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K _{app}			K _C			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.)	THICK (in.)	INIT (in.)	FINAL (in.)	ONSET (KSI)	MAX (KSI)	K _{app} (KSI)	K _{app} MEAN	STAN DEV	K _C (KSI)	K _C MEAN	STAN DEV		
Cold Rolled 90 Percent and Aged	Sheet	0.03	R.T.	L-T	218.0	4.020	0.026	1.310	1.700	---	114.00	176.14	161.4	178.5	3.6	209.94*	1964	60578	
		0.03			218.0	4.010	0.027	1.280	1.500	---	118.00	178.68	161.4	178.5	3.6	198.64*	1964	60578	
		0.03			218.0	4.000	0.028	1.730	1.980	---	83.00	165.12	161.4	178.5	3.6	173.40	1964	60578	
		0.03			218.0	4.000	0.028	1.250	1.450	---	111.00	165.62	161.4	178.5	3.6	182.64	1964	60578	
		0.03			218.0	4.000	0.028	1.250	1.550	---	112.00	167.12	161.4	178.5	3.6	192.94*	1964	60578	
		0.03			218.0	4.000	0.028	1.720	2.100	---	81.80	162.20	161.4	178.5	3.6	180.32	1964	60578	
		0.03			218.0	4.000	0.028	0.730	1.050	---	138.00	150.89	161.4	178.5	3.6	185.16*	1964	60578	
		0.03			218.0	4.000	0.028	1.230	1.480	---	116.00	171.34	161.4	178.5	3.6	193.46*	1964	60578	
		0.03			218.0	4.010	0.028	1.720	2.150	---	85.50	168.98	161.4	178.5	3.6	192.66*	1964	60578	
		0.03			218.0	4.010	0.028	0.740	0.880	---	138.00	151.99	161.4	178.5	3.6	167.24*	1964	60578	
		0.03			218.0	4.010	0.028	0.720	0.940	---	145.00	157.34*	161.4	178.5	3.6	182.41*	1964	60578	
		0.03			218.0	4.010	0.028	1.240	1.460	---	110.00	163.25	161.4	178.5	3.6	181.66	1964	60578	
		0.03			218.0	4.010	0.028	0.730	1.000	---	148.00	161.80*	161.4	178.5	3.6	192.94*	1964	60578	
		0.03			218.0	4.010	0.028	1.720	1.950	---	85.80	169.53	161.4	178.5	3.6	176.70	1964	60578	
		0.03			218.0	4.010	0.028	0.720	0.890	---	143.00	155.17	161.4	178.5	3.6	174.41*	1964	60578	
		0.03			218.0	4.010	0.028	1.320	1.650	---	104.00	160.62	161.4	178.5	3.6	187.39*	1964	60578	
		0.03			218.0	4.010	0.028	1.240	1.550	---	112.00	166.22	161.4	178.5	3.6	192.84*	1964	60578	
		0.03			218.0	4.010	0.028	1.730	2.100	---	80.10	149.60	161.4	178.5	3.6	176.38	1964	60578	
0.03	218.0	4.020	0.028	1.270	1.730	---	108.00	162.67	161.4	178.5	3.6	201.57*	1964	60578					

BUCKLING OF CRACK EDGES NOT RESTRAINED

* NOTE: NET SECTION STRESS EXCEEDS 80% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

INCONEL 718

4 of 4

TABLE 5.9.2.2 (CONCLUDED)

INCONEL 718 K_{IC}																				
CONDITIONS AT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K_{app}			K_{IC}			DATE	REFER	
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	INIT (in.) $2a_i$	FINAL (in.) $2a_f$	ONSET (KSI) σ_o	MAX (KSI) σ_{max}	K_{app} (KSI $\sqrt{in.}$)	K_{app} MEAN	STAN DEV	K_{IC} (KSI $\sqrt{in.}$)	K_{IC} MEAN	STAN DEV			
Cold Rolled 30 Percent and Aged	Sheet	0.03	17.930	0.026	5.480	5.880	---	64.40	200.63	197.0	6.5	233.26	224.6	11.1	209.48	235.75	217.14	227.47	1964	60578
		0.03	17.930	0.026	5.480	5.880	---	64.30	200.31	197.0	6.5	233.26	224.6	11.1	209.48	235.75	217.14	227.47	1964	60578
		0.03	17.940	0.026	5.470	6.820	---	65.50	203.81	197.0	6.5	233.26	224.6	11.1	209.48	235.75	217.14	227.47	1964	60578
		0.03	18.080	0.026	5.480	6.680	---	61.30	190.77	197.0	6.5	233.26	224.6	11.1	209.48	235.75	217.14	227.47	1964	60578
		0.03	18.080	0.026	5.470	7.200	---	60.90	189.31	197.0	6.5	233.26	224.6	11.1	209.48	235.75	217.14	227.47	1964	60578

BUCKLING OF CRACK EDGES NOT RESTRAINED

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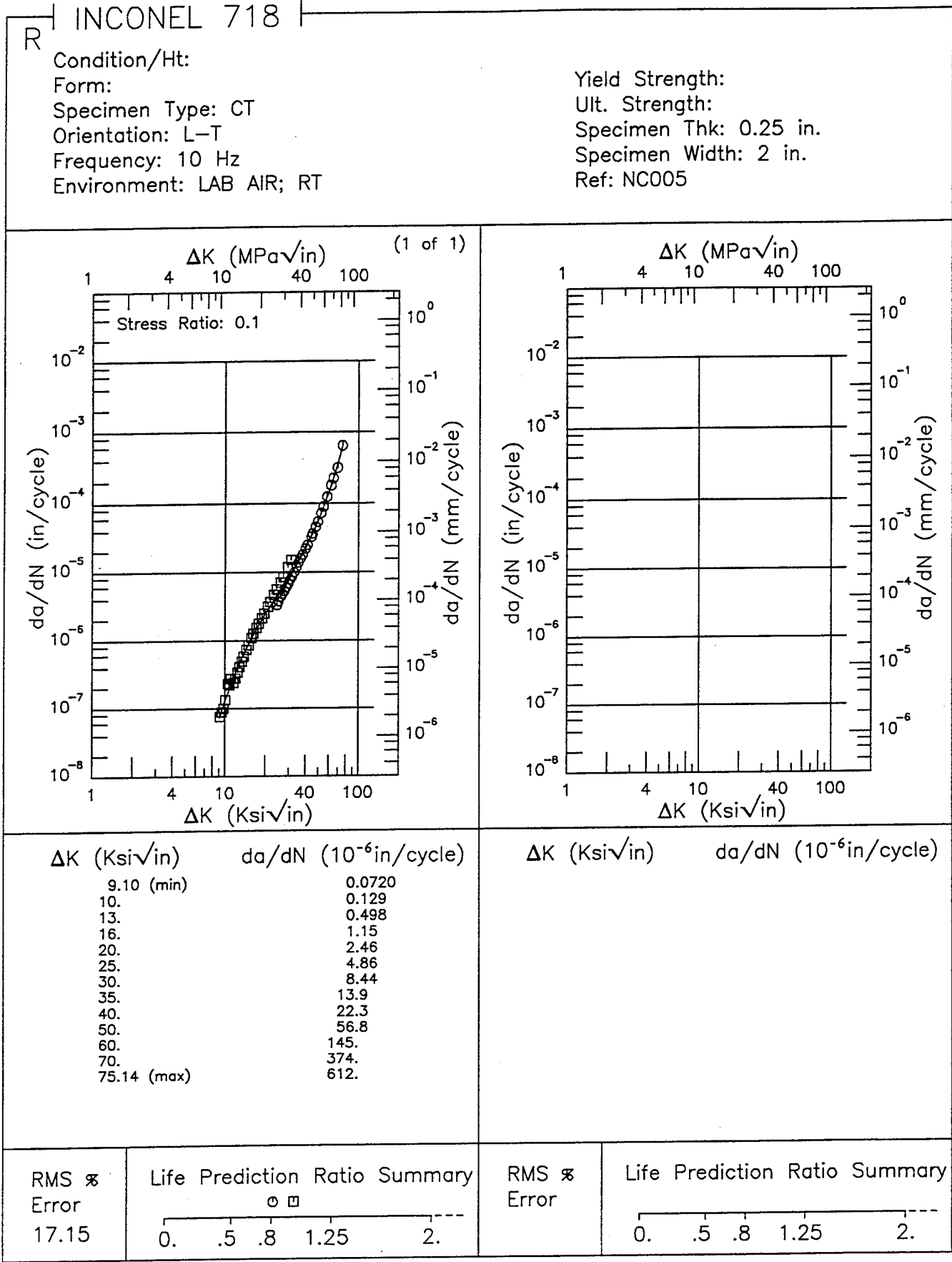


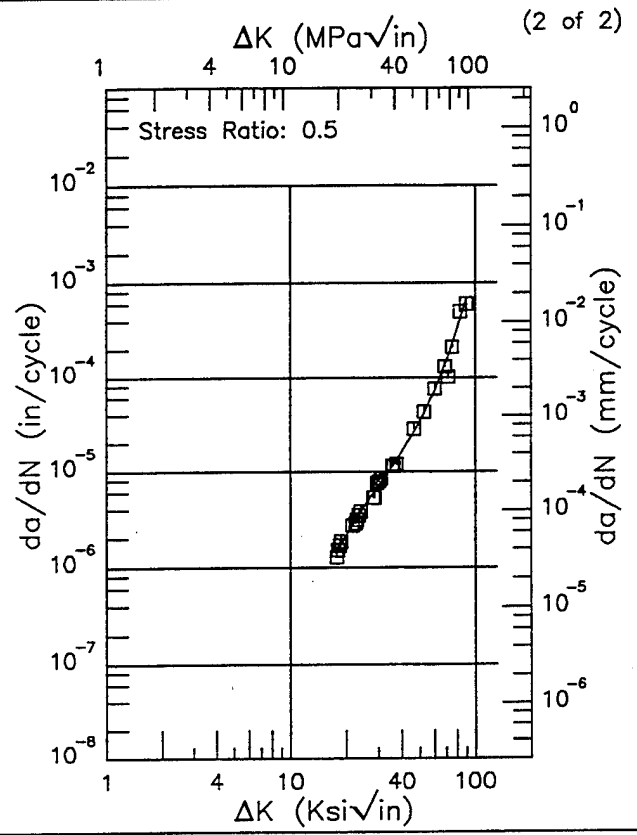
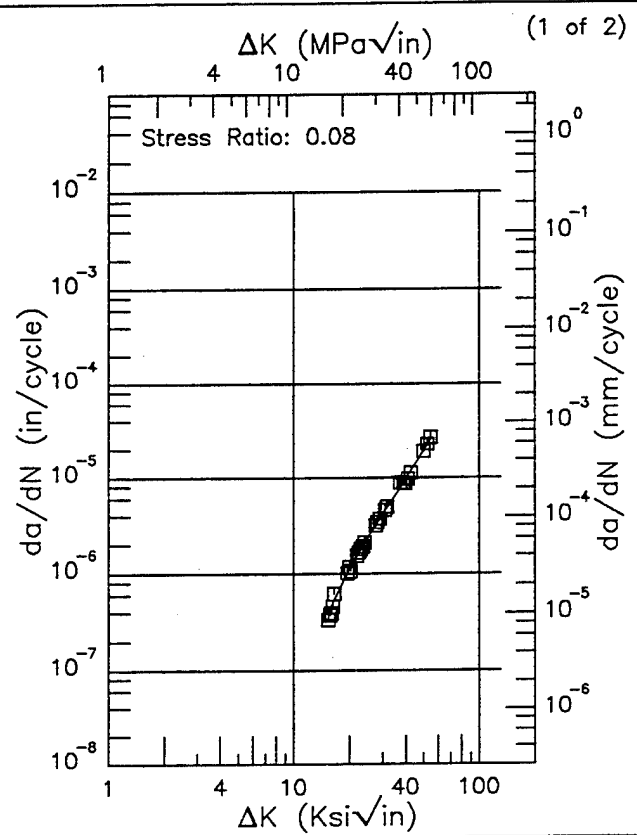
Figure 5.9.3.1.1

INCONEL 718

R

Condition/Ht: 1325F 8HRS FC TO 1150F
 Form: 4 in. Forging
 Specimen Type: CT
 Orientation: L-T
 Frequency: 6 Hz
 Environment: L.H.A.; RT

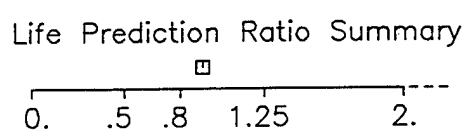
Yield Strength: 168 ksi
 Ult. Strength: 199 ksi
 Specimen Thk: 0.54 - 0.55 in.
 Specimen Width: 7.4 in.
 Ref: 88579



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
15.35 (min)	0.393
16.	0.470
20.	1.13
25.	2.38
30.	4.16
35.	6.54
40.	9.70
50.	19.3
54.42 (max)	25.6

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
17.63 (min)	1.40
20.	2.17
25.	4.27
30.	7.07
35.	10.9
40.	16.1
50.	34.0
60.	71.4
70.	152.
80.	329.
89.46 (max)	689.

RMS %
Error
7.76



RMS %
Error
11.73

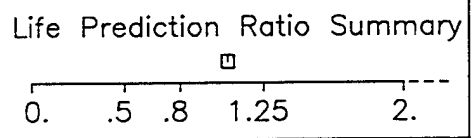
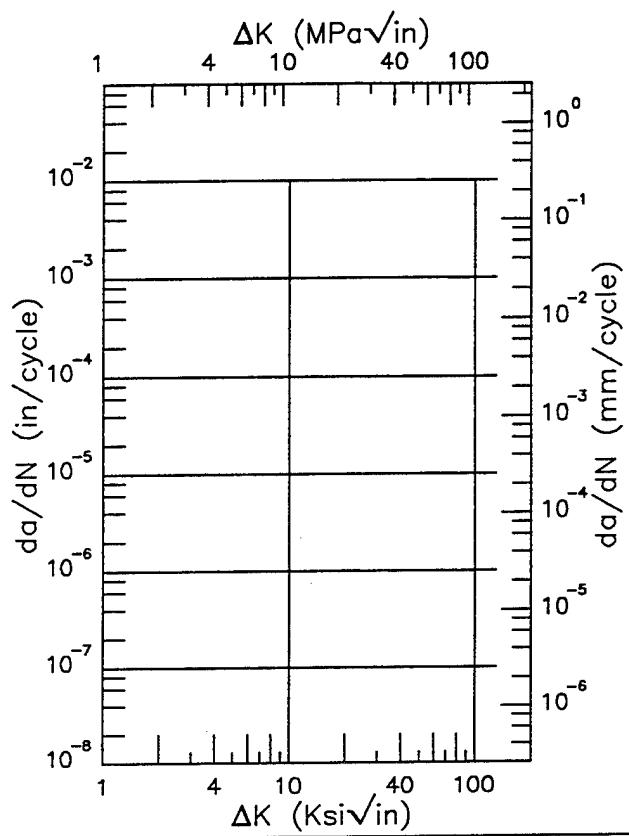
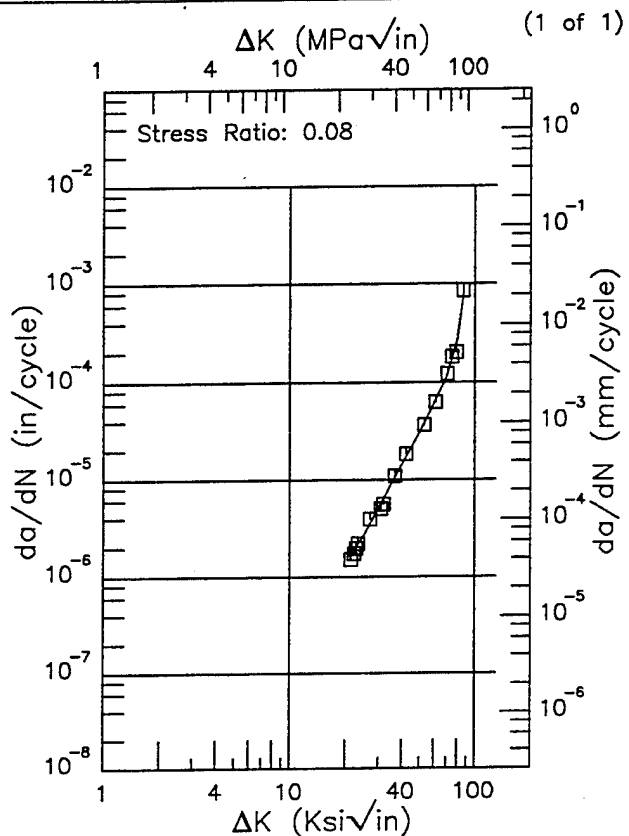


Figure 5.9.3.1.2

R INCONEL 718

Condition/Ht: 1325F 8HRS FC TO 1150F
 Form: 4 in. Forging
 Specimen Type: CT
 Orientation: T-L
 Frequency: 6 Hz
 Environment: L.H.A.; RT

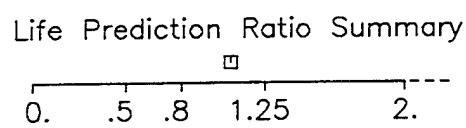
Yield Strength: 166 ksi
 Ult. Strength: 197 ksi
 Specimen Thk: 0.51 in.
 Specimen Width: 6.01 in.
 Ref: 88579



ΔK (Ksi \sqrt{in})	da/dN (10 ⁻⁶ in/cycle)
21.31 (min)	1.65
25.	2.68
30.	5.10
35.	8.98
40.	14.4
50.	30.9
60.	60.4
70.	113.
80.	265.
86.24 (max)	822.

ΔK (Ksi \sqrt{in})	da/dN (10 ⁻⁶ in/cycle)
-------------------------------	-------------------------------------

RMS \times
 Error
 8.08



RMS \times
 Error

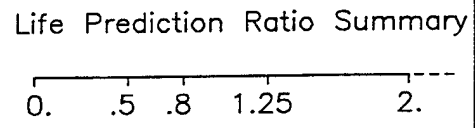
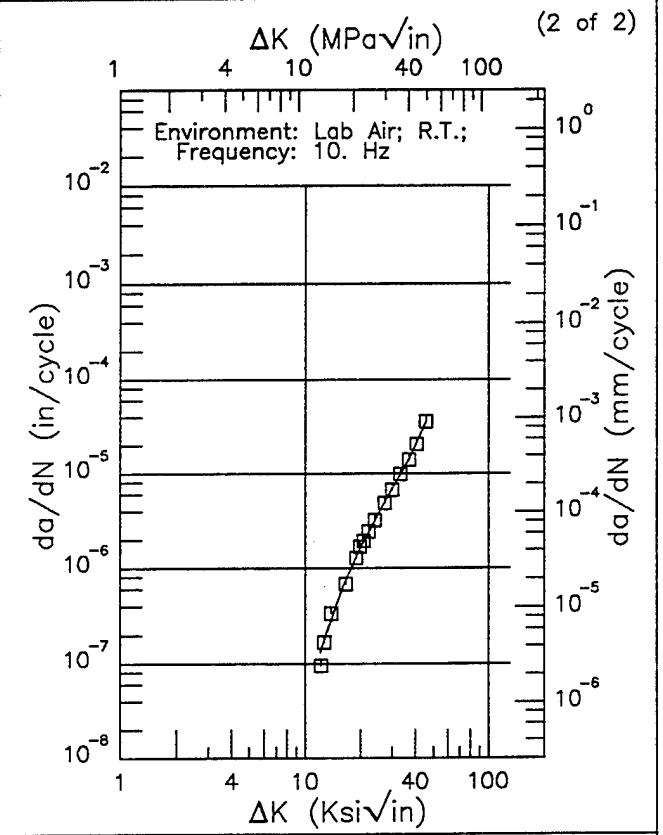
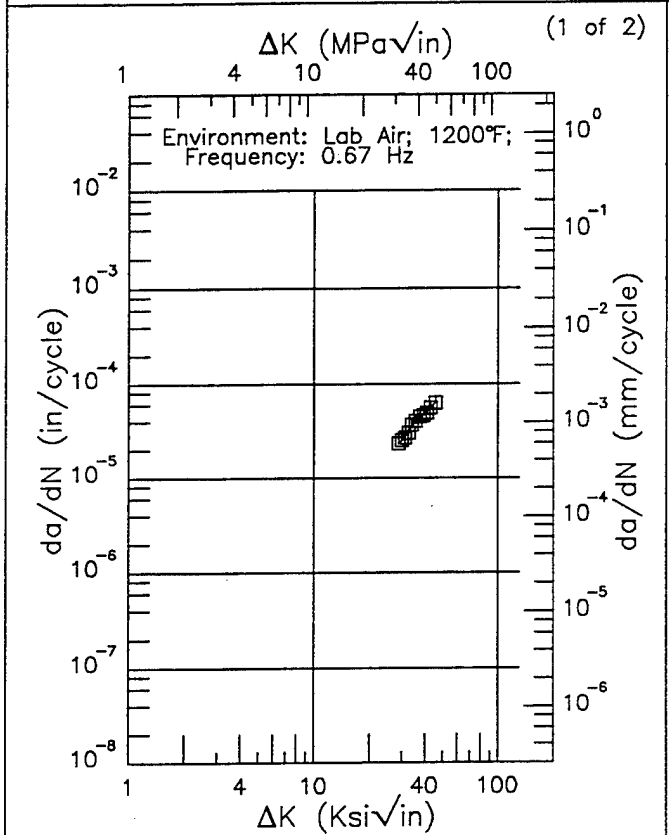


Figure 5.9.3.1.3

INCONEL 718 EF

Condition/Ht: 1750F 1HR AC 1325F
 Form: 0.5 in. Plate
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05

Yield Strength: 147.9 ksi
 Ult. Strength: 191.3 ksi
 Specimen Thk: 0.298 - 0.49 in.
 Specimen Width: 1.153 - 1.998 in.
 Ref: HD016



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
28.72 (min)	22.8
30.	24.7
35.	38.5
40.	48.4
45.46 (max)	63.3

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
12.11 (min)	0.132
13.	0.207
16.	0.633
20.	1.65
25.	3.69
30.	6.88
35.	11.9
40.	20.0
45.38 (max)	34.7

RMS % Error 2.74	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.
---------------------	--

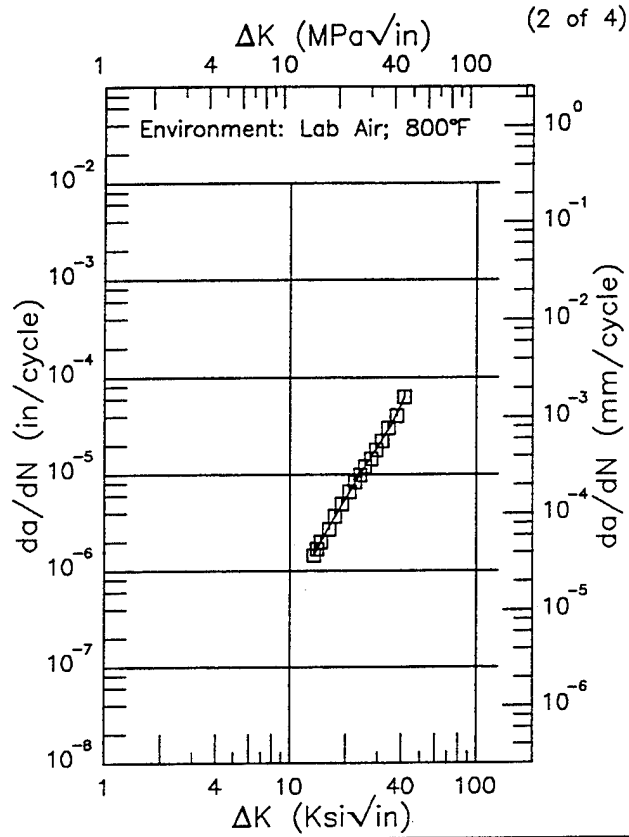
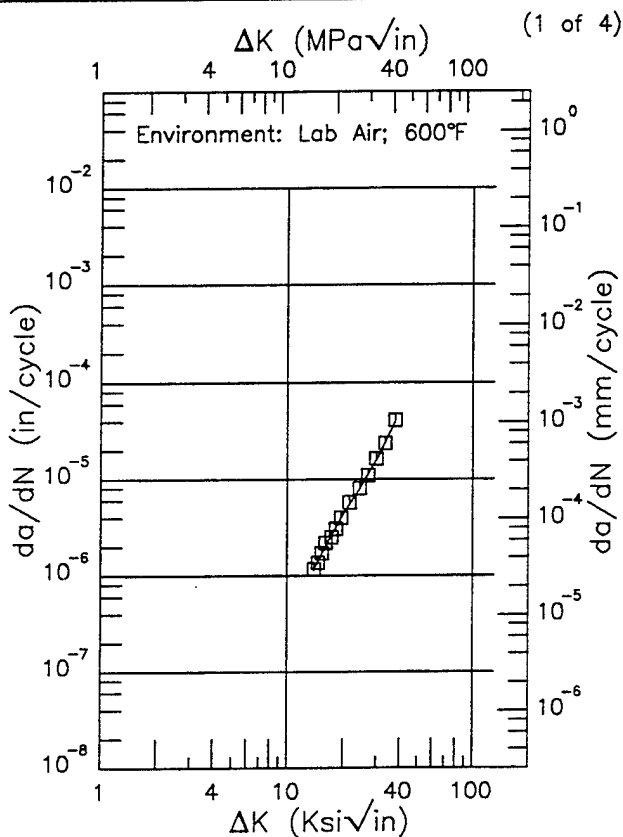
RMS % Error 9.80	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.
---------------------	--

Figure 5.9.3.1.4

E | INCONEL 718 |

Condition/Ht: 1750F 1HR AC 1325F
 Form: 0.63 in. Forged Bar
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

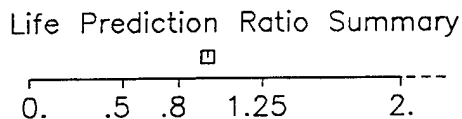
Yield Strength: 164.9 ksi
 Ult. Strength: 208.1 ksi
 Specimen Thk: 0.3 - 0.302 in.
 Specimen Width: 1.152 - 1.153 in.
 Ref: HD016



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.90 (min)	1.13
16.	2.00
20.	4.34
25.	8.76
30.	16.0
35.	28.9
37.80 (max)	40.8

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.42 (min)	1.42
16.	2.66
20.	5.67
25.	11.3
30.	19.2
35.	31.7
40.	53.7
41.21 (max)	61.6

RMS %
Error
3.21



RMS %
Error
2.22

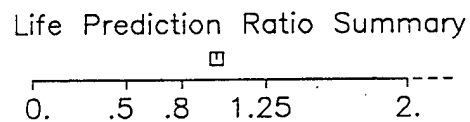


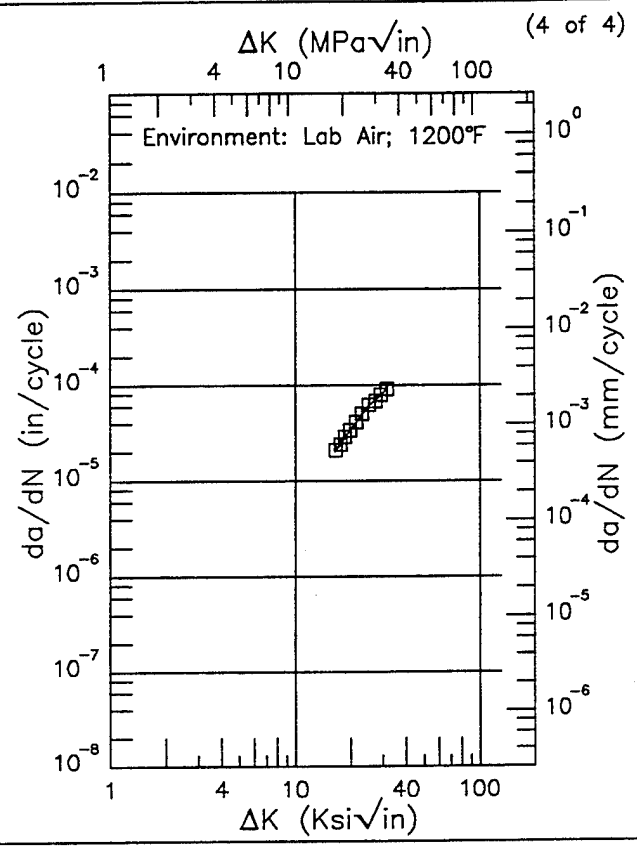
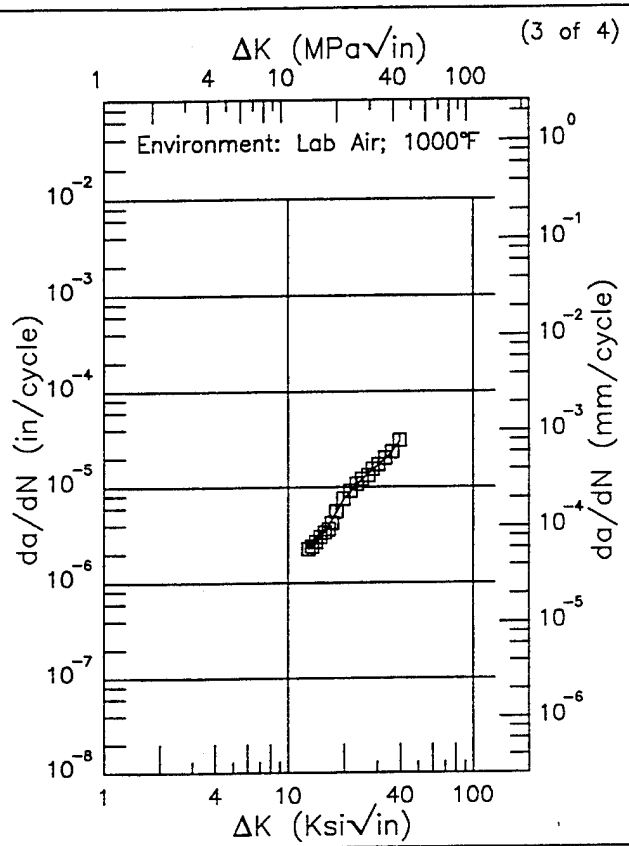
Figure 5.9.3.1.5

INCONEL 718

E

Condition/Ht: 1750F 1HR AC 1325F
 Form: 0.63 in. Forged Bar
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

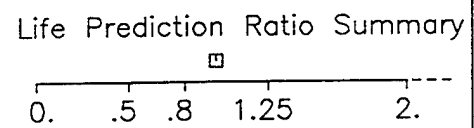
Yield Strength: 164.9 ksi
 Ult. Strength: 208.1 ksi
 Specimen Thk: 0.3 - 0.302 in.
 Specimen Width: 1.152 - 1.153 in.
 Ref: HD016



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
12.74 (min)	2.35
13.	2.34
16.	3.56
20.	7.57
25.	12.1
30.	16.5
35.	21.6
39.66 (max)	30.3

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.46 (min)	20.4
20.	34.4
25.	60.1
30.	82.4
31.33 (max)	91.2

RMS \times
 Error
 3.54



RMS \times
 Error
 1.74

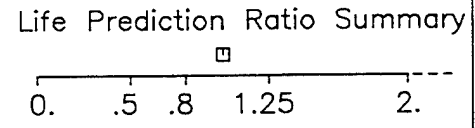


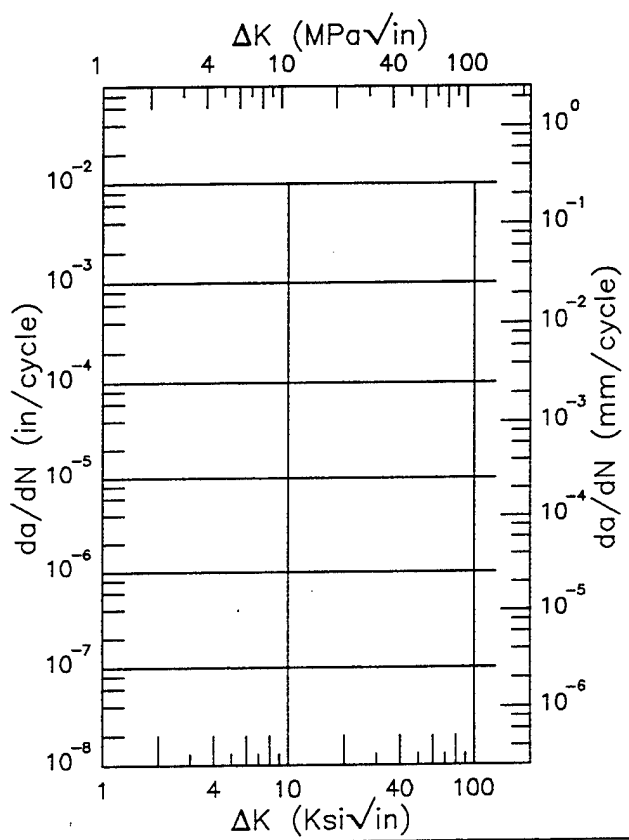
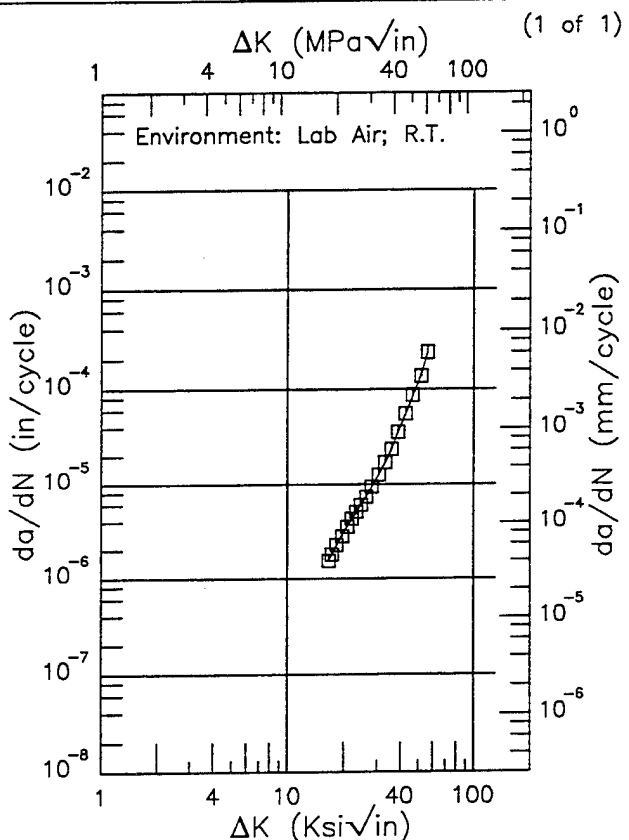
Figure 5.9.3.1.5 (Concluded)

INCONEL 718

E

Condition/Ht: 1750F 1HR AC 1325F
 Form: 0.63 in. Forged Bar
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.05
 Frequency: 8.3 - 10 Hz

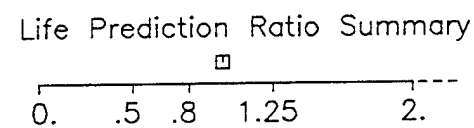
Yield Strength: 164.9 ksi
 Ult. Strength: 208.1 ksi
 Specimen Thk: 0.301 in.
 Specimen Width: 1.151 in.
 Ref: HD016



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
16.56 (min)	1.53
20.	3.08
25.	6.32
30.	11.6
35.	21.6
40.	41.8
50.	116.
56.03 (max)	235.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
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RMS % Error
2.00



RMS % Error

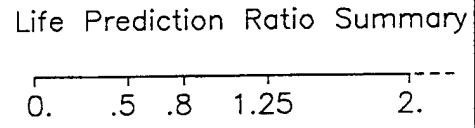


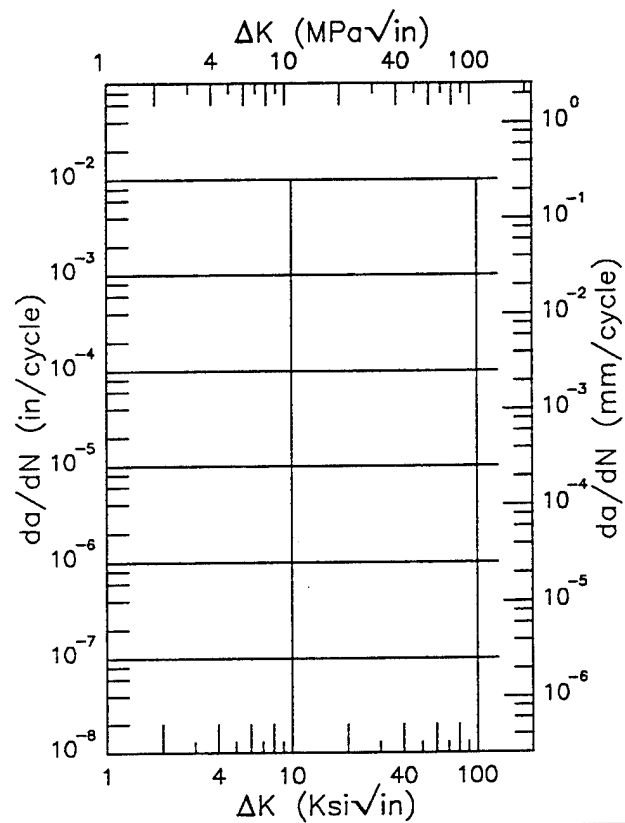
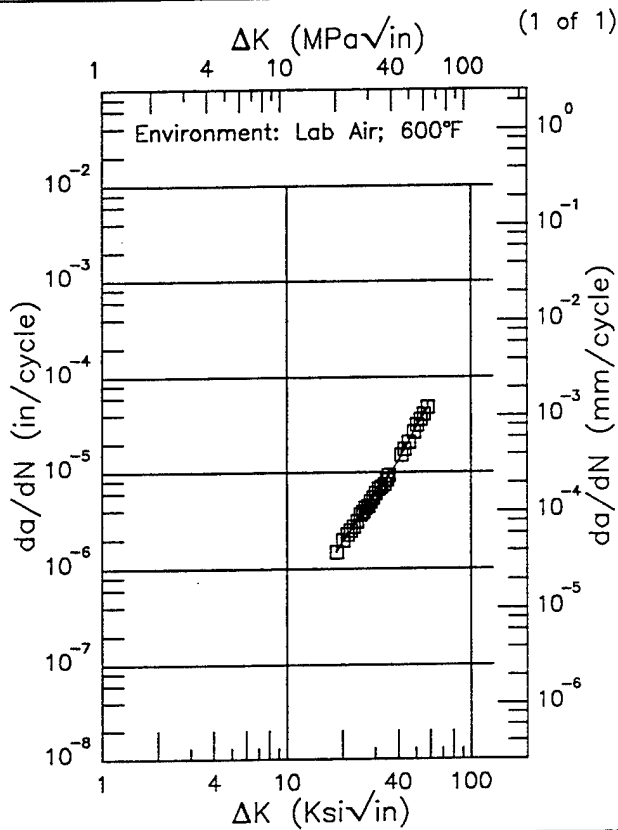
Figure 5.9.3.1.6

INCONEL 718

E

Condition/Ht: 1750F AC
 Form: 0.06 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation:
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

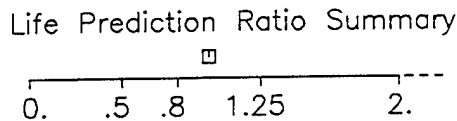
Yield Strength: 174 ksi
 Ult. Strength: 204.1 ksi
 Specimen Thk: 0.061 in.
 Specimen Width: 1.995 in.
 Ref: HD017



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
18.40 (min)	1.50
20.	1.90
25.	3.53
30.	5.73
35.	8.69
40.	13.1
50.	29.2
57.54 (max)	46.5

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
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RMS $\%$
 Error
 2.11



RMS $\%$
 Error

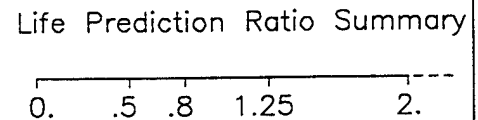
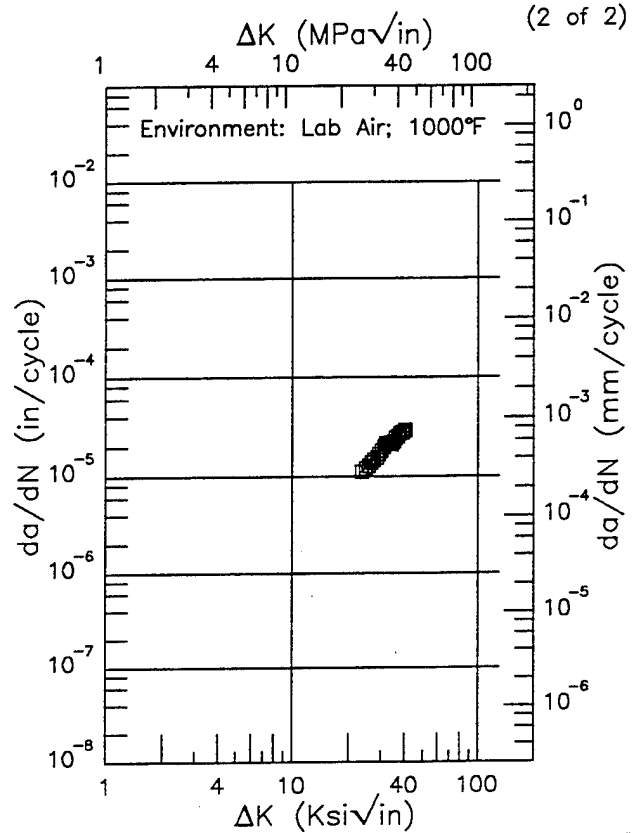
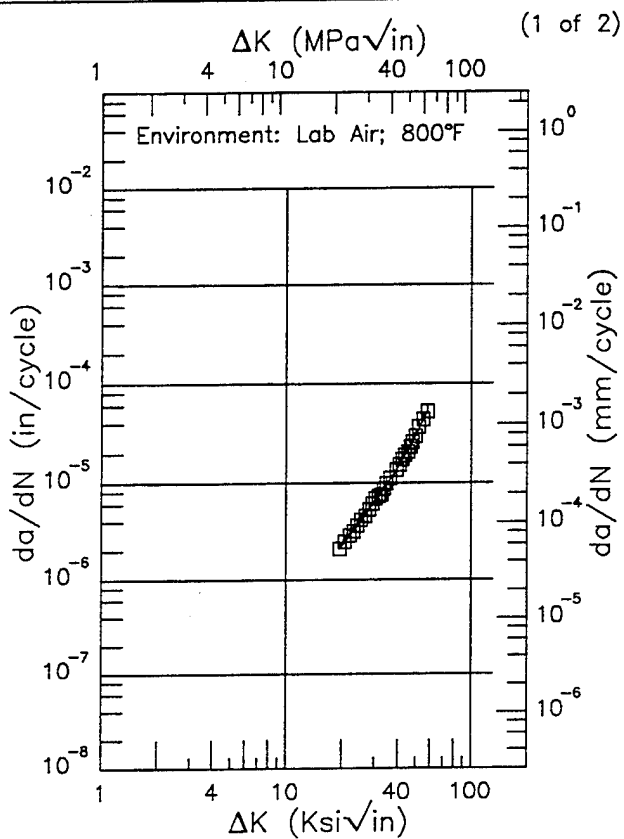


Figure 5.9.3.1.7

INCONEL 718

Condition/Ht: 1750F AC 1325F
 Form: 0.06 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation:
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

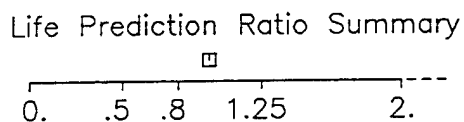
Yield Strength: 174 ksi
 Ult. Strength: 204.1 ksi
 Specimen Thk: 0.061 in.
 Specimen Width: 1.999 - 2 in.
 Ref: HD017



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
19.39 (min)	2.12
20.	2.26
25.	3.92
30.	6.43
35.	9.79
40.	14.3
50.	30.8
57.84 (max)	53.2

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
23.42 (min)	11.1
25.	12.1
30.	18.2
35.	23.3
40.	29.6
40.06 (max)	29.7

RMS $\%$
 Error
 2.79



RMS $\%$
 Error
 2.89

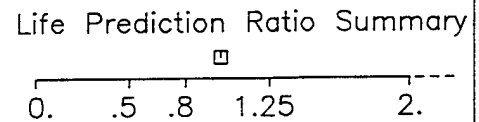


Figure 5.9.3.1.8

INCONEL 718

E

Condition/Ht: 1750F AC 1325F
 Form: 0.06 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation:
 Stress Ratio: 0.05
 Frequency: 5 Hz

Yield Strength: 174 ksi
 Ult. Strength: 204.1 ksi
 Specimen Thk: 0.061 in.
 Specimen Width: 1.994 in.
 Ref: HD017

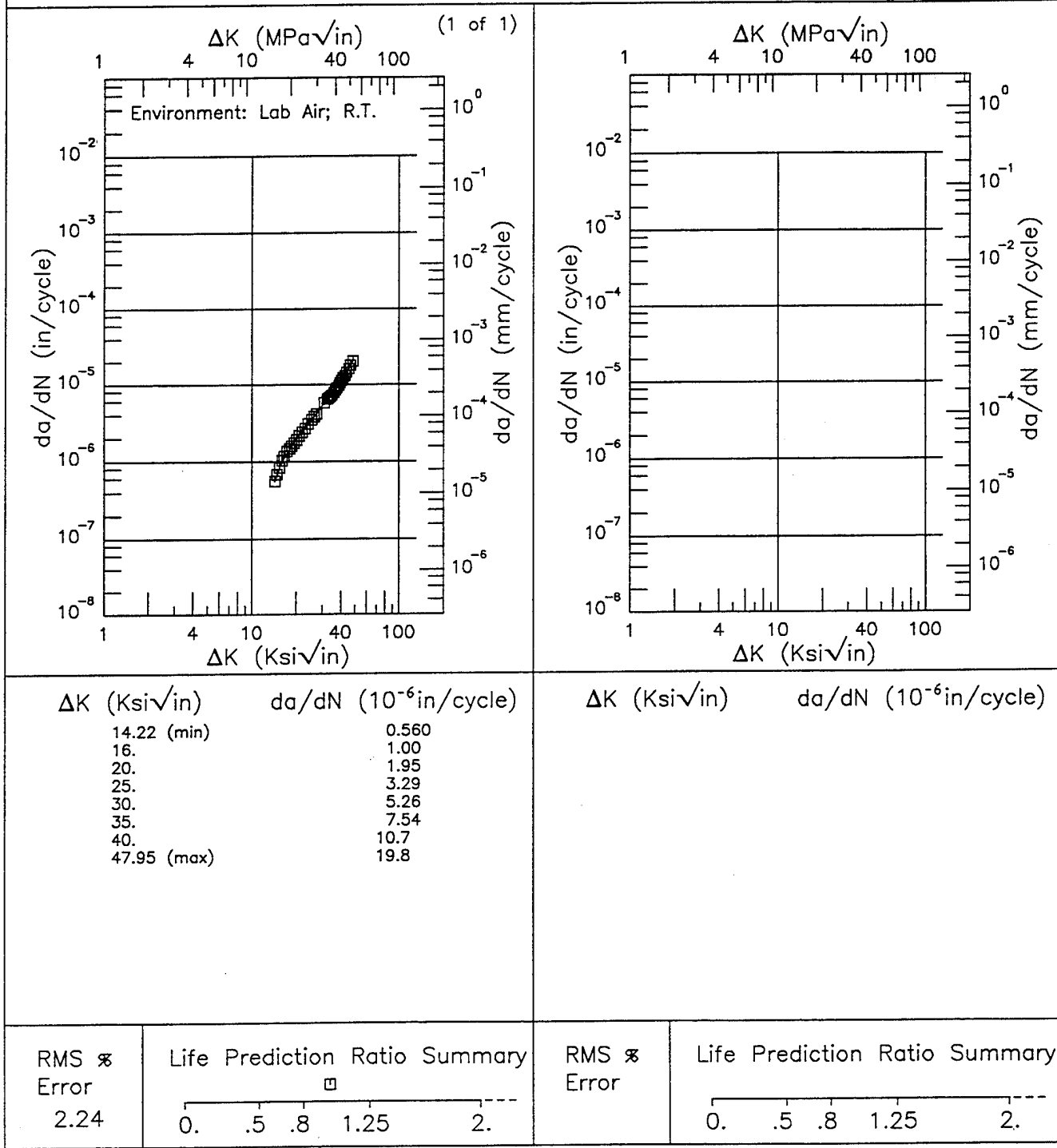


Figure 5.9.3.1.9

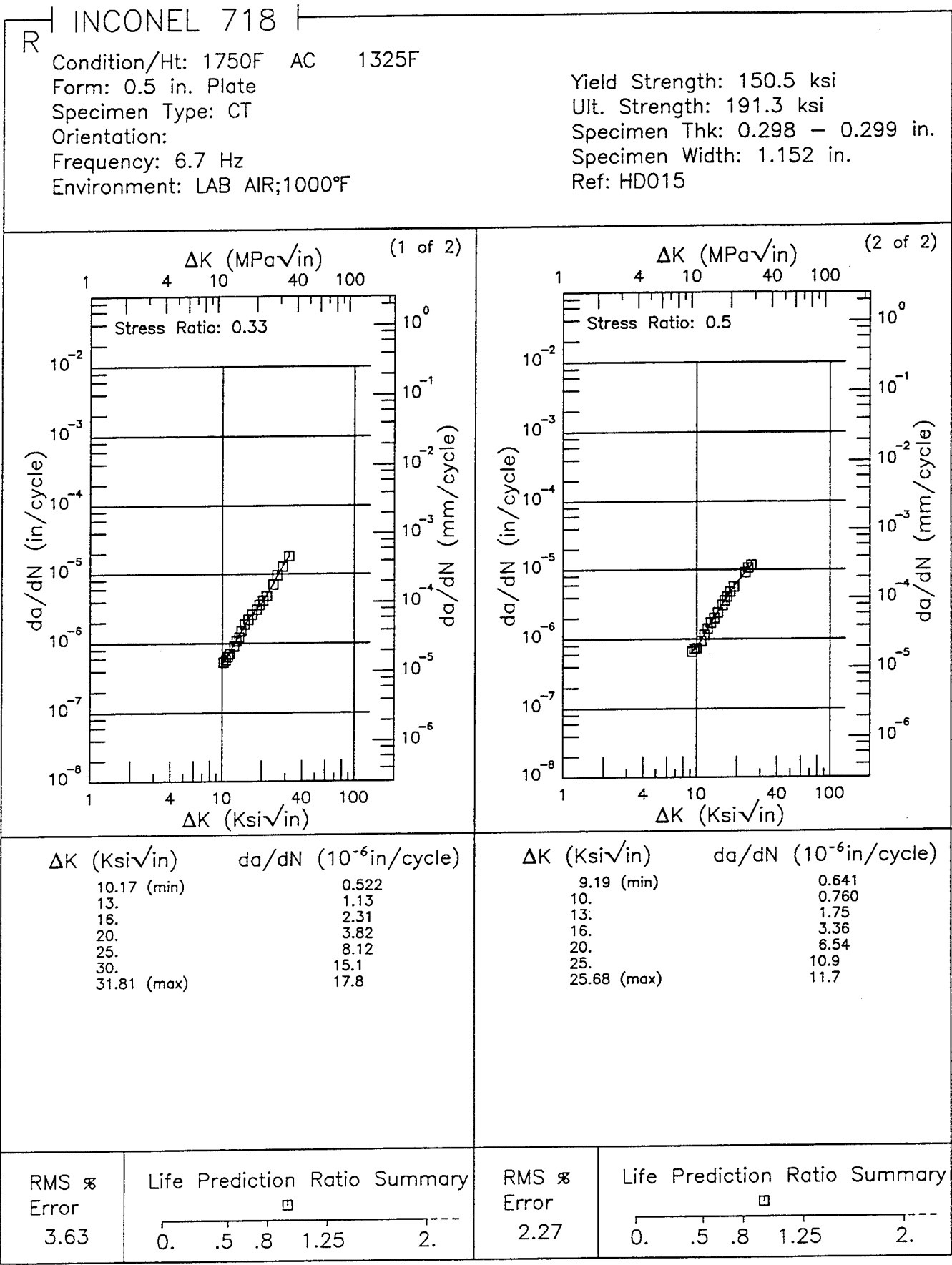


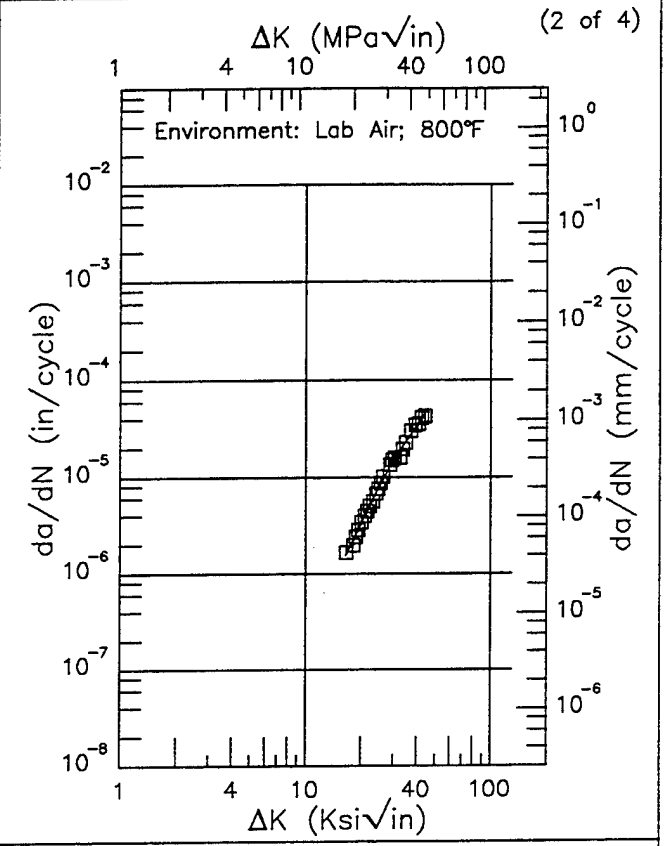
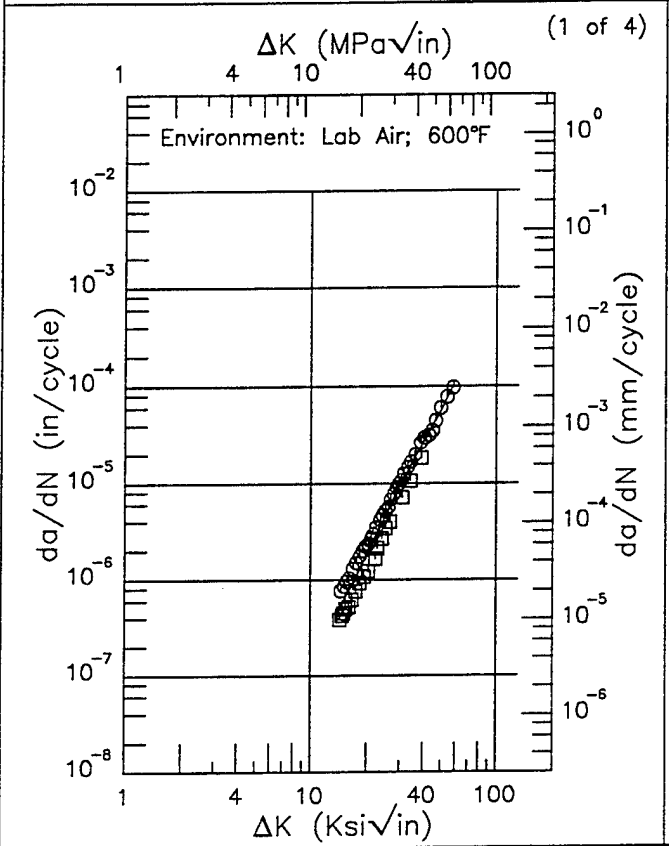
Figure 5.9.3.1.10

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E INCONEL 718

Condition/Ht: 1750F AC 1325F
 Form: 0.5 in. Plate
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

Yield Strength: 150.5 - 156.4 ksi
 Ult. Strength: 191.3 - 200.7 ksi
 Specimen Thk: 0.299 - 0.497 in.
 Specimen Width: 1.152 - 1.999 in.
 Ref: HD015;HD017



ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
14.29 (min)	0.513
16.	0.769
20.	1.82
25.	4.51
30.	9.32
35.	16.6
40.	26.5
50.	57.1
57.98 (max)	102.

ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
16.56 (min)	1.54
20.	3.24
25.	7.95
30.	15.2
35.	23.7
40.	34.1
44.06 (max)	46.0

RMS % Error 24.70	Life Prediction Ratio Summary
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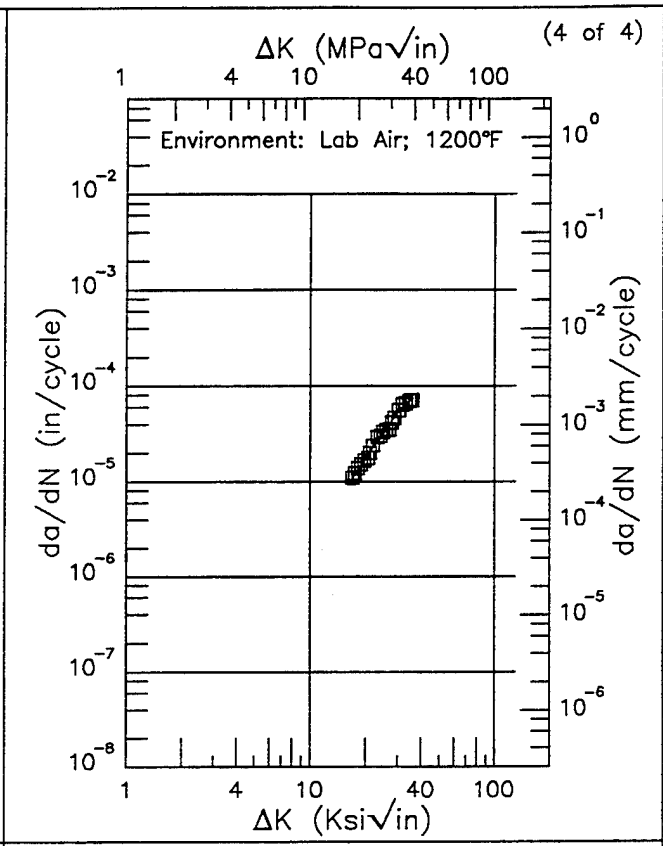
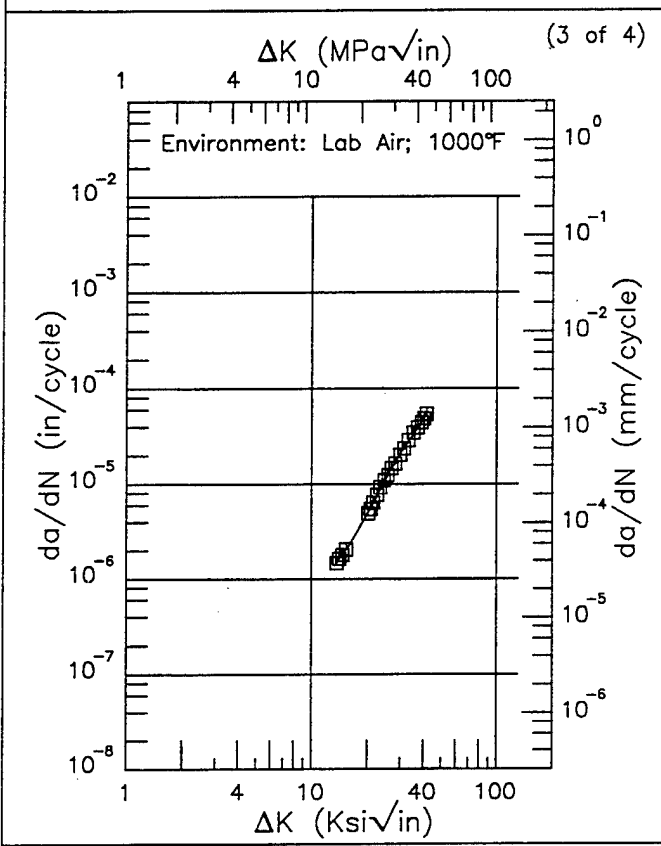
RMS % Error 6.13	Life Prediction Ratio Summary
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Figure 5.9.3.1.11

INCONEL 718 E

Condition/Ht: 1750F AC 1325F
 Form: 0.5 in. Plate
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

Yield Strength: 150.5 - 156.4 ksi
 Ult. Strength: 191.3 - 200.7 ksi
 Specimen Thk: 0.299 - 0.497 in.
 Specimen Width: 1.152 - 1.999 in.
 Ref: HD015;HD017



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.60 (min)	1.48
16.	2.30
20.	4.97
25.	11.4
30.	20.9
35.	32.8
40.	47.7
41.58 (max)	53.2

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.93 (min)	10.8
20.	17.8
25.	32.2
30.	53.3
35.	71.5
35.62 (max)	72.6

RMS \times Error 2.40	Life Prediction Ratio Summary
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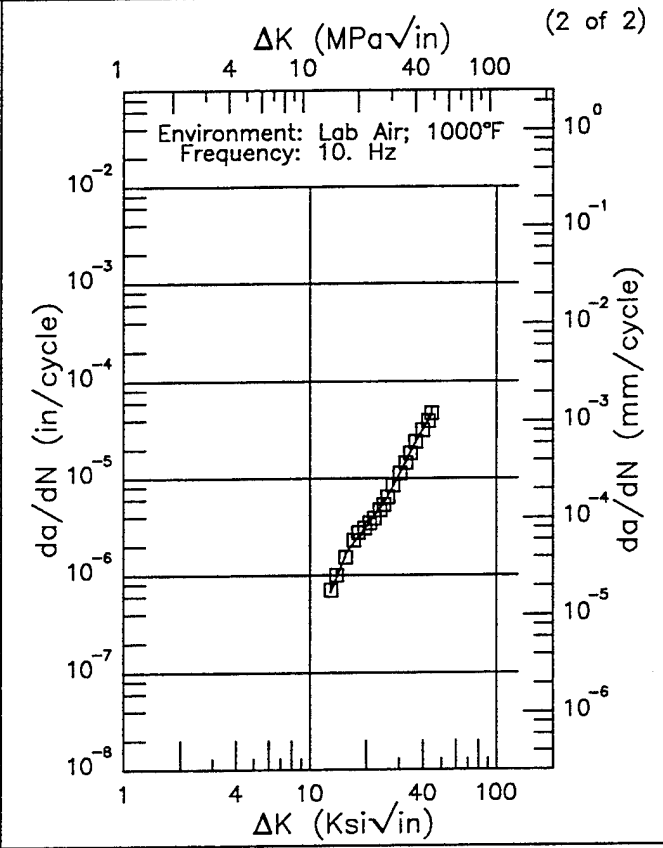
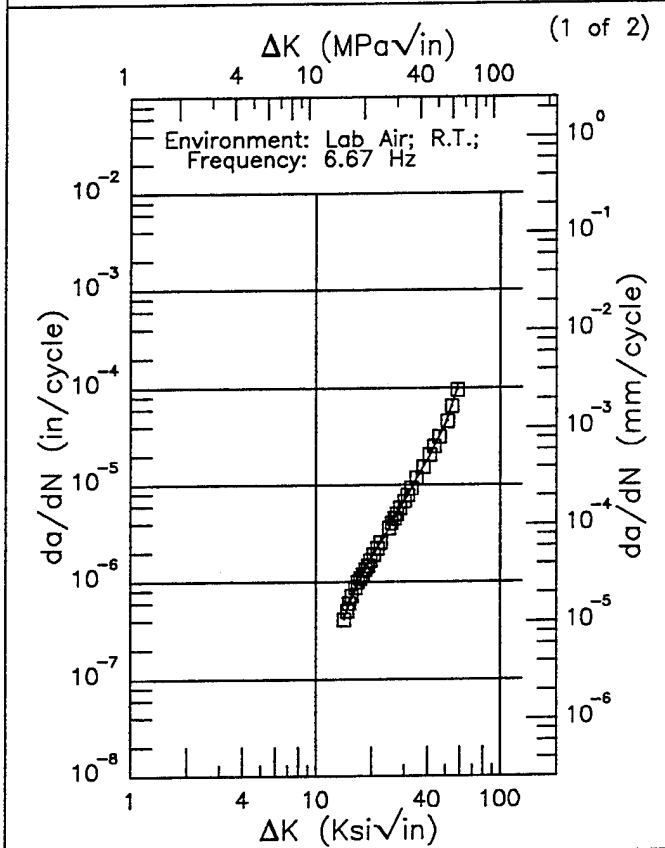
RMS \times Error 4.76	Life Prediction Ratio Summary
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Figure 5.9.3.1.11 (Concluded)

EF INCONEL 718

Condition/Ht: 1750F AC 1325F 8HRS FC TO 1150F HELD 18HRS AC
 Form: 0.5 in. Plate
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05

Yield Strength: 150.5 - 156.4 ksi
 Ult. Strength: 191.3 - 200.7 ksi
 Specimen Thk: 0.298 - 0.495 in.
 Specimen Width: 1.151 - 1.998 in.
 Ref: HD017;HD015



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
14.19 (min)	0.426
16.	0.764
20.	1.77
25.	3.63
30.	6.76
35.	11.6
40.	18.2
50.	42.5
58.21 (max)	96.7

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
12.91 (min)	0.658
13.	0.688
16.	1.80
20.	3.16
25.	5.60
30.	10.7
35.	20.1
40.	31.9
44.45 (max)	46.0

RMS % Error 2.68	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.
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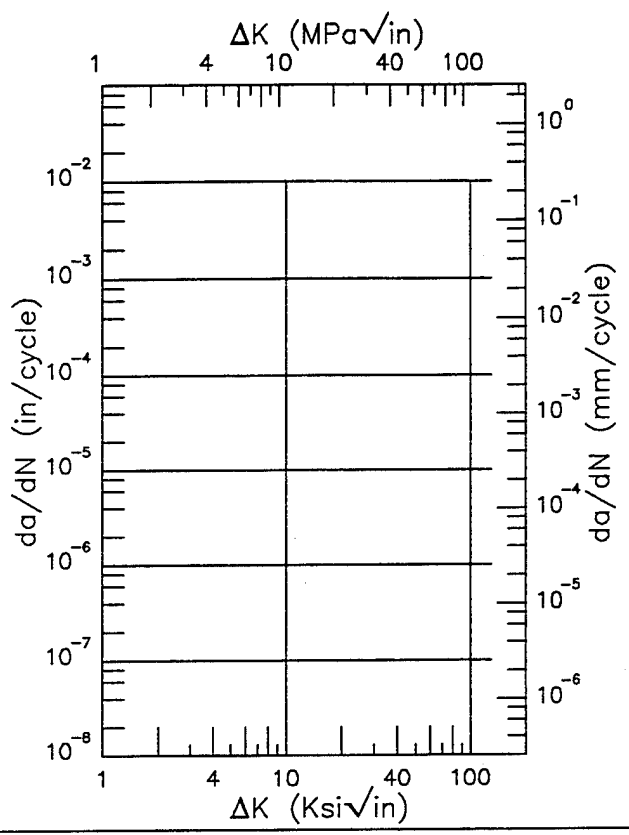
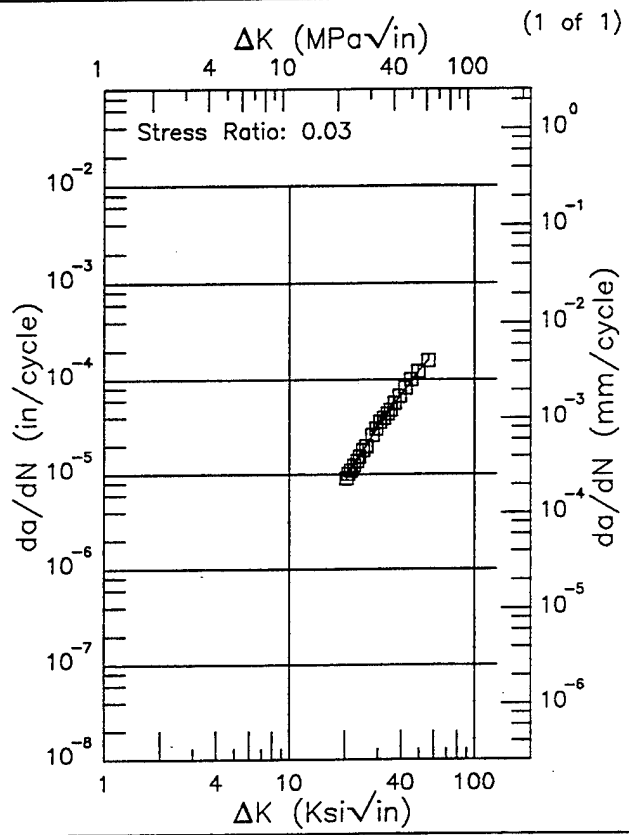
RMS % Error 4.61	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.
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Figure 5.9.3.1.12

INCONEL 718 R

Condition/Ht: 1750F AC 1325F
 Form: 0.5 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Frequency: 0.7 Hz
 Environment: LAB AIR;1200°F

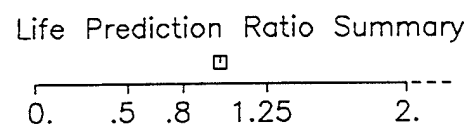
Yield Strength: 150.5 ksi
 Ult. Strength: 191.3 ksi
 Specimen Thk: 0.477 in.
 Specimen Width: 1.997 in.
 Ref: HD015



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
20.23 (min)	9.30
25.	17.7
30.	31.8
35.	47.6
40.	69.7
50.	125.
55.76 (max)	158.

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS %
 Error
 2.05



RMS %
 Error

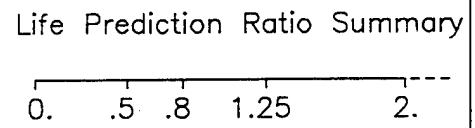


Figure 5.9.3.1.13

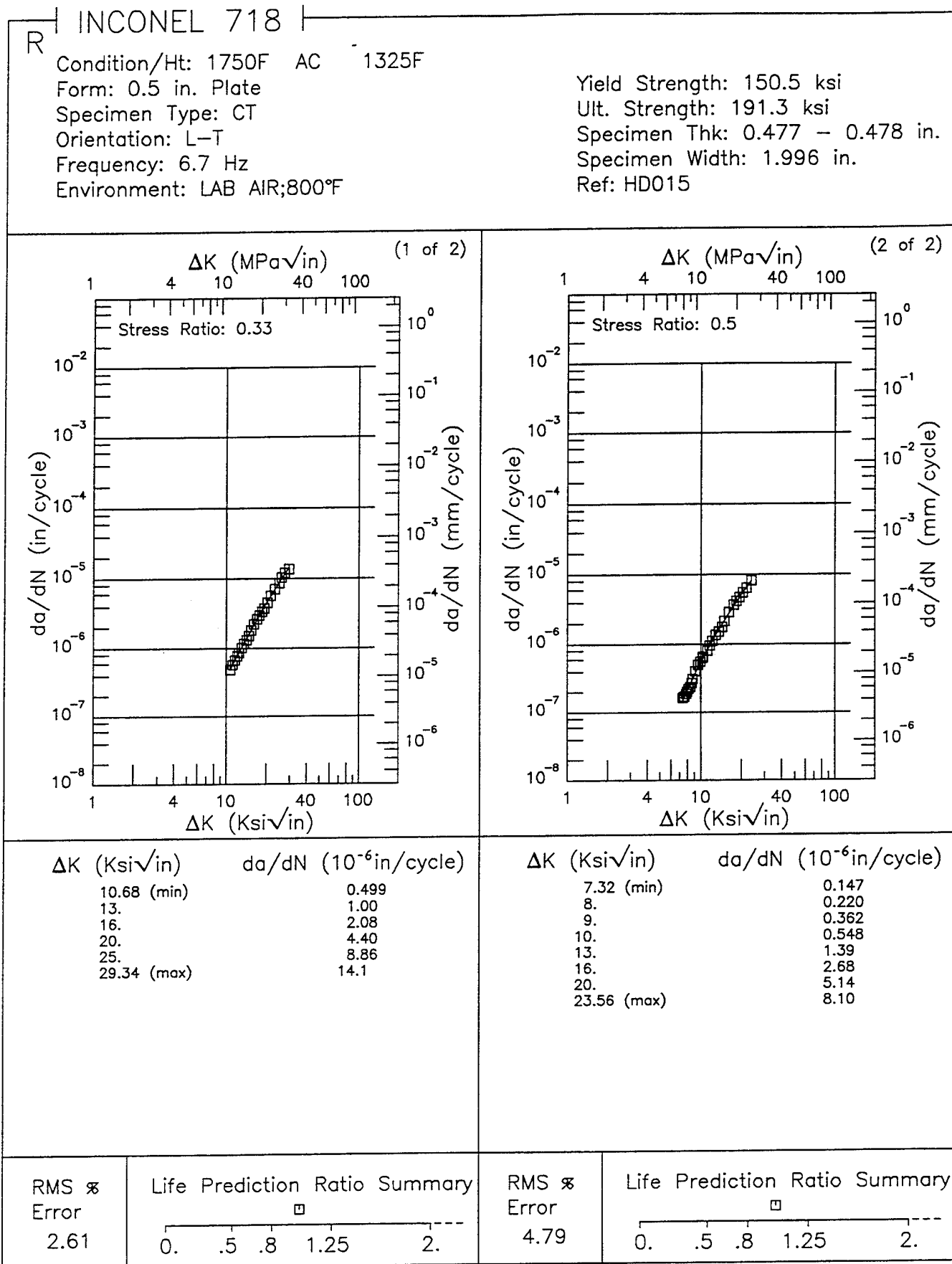
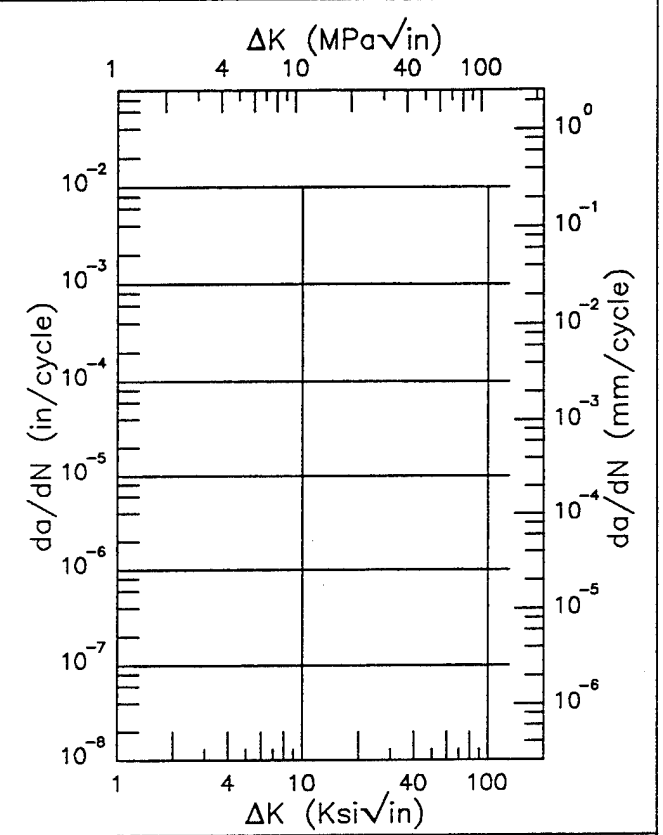
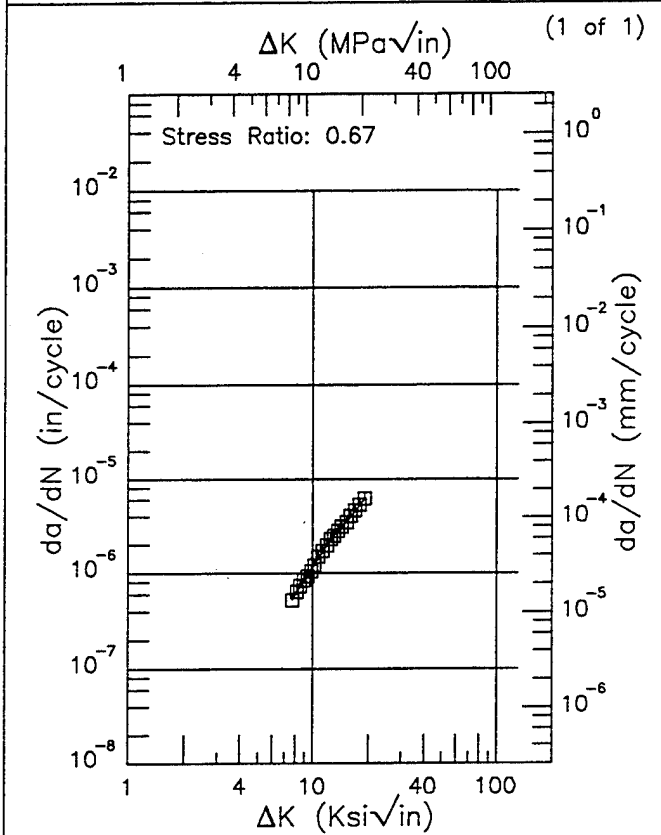


Figure 5.9.3.1.14

INCONEL 718 R

Condition/Ht: 1750F AC 1325F
 Form: 0.5 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Frequency: 6.7 Hz
 Environment: LAB AIR;1000°F

Yield Strength: 150.5 ksi
 Ult. Strength: 191.3 ksi
 Specimen Thk: 0.479 in.
 Specimen Width: 1.993 in.
 Ref: HD015



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.74 (min)	0.531
8.	0.591
9.	0.848
10.	1.16
13.	2.50
16.	3.95
18.84 (max)	6.13

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.74 (min)	0.531
8.	0.591
9.	0.848
10.	1.16
13.	2.50
16.	3.95
18.84 (max)	6.13

RMS % Error 2.29	Life Prediction Ratio Summary
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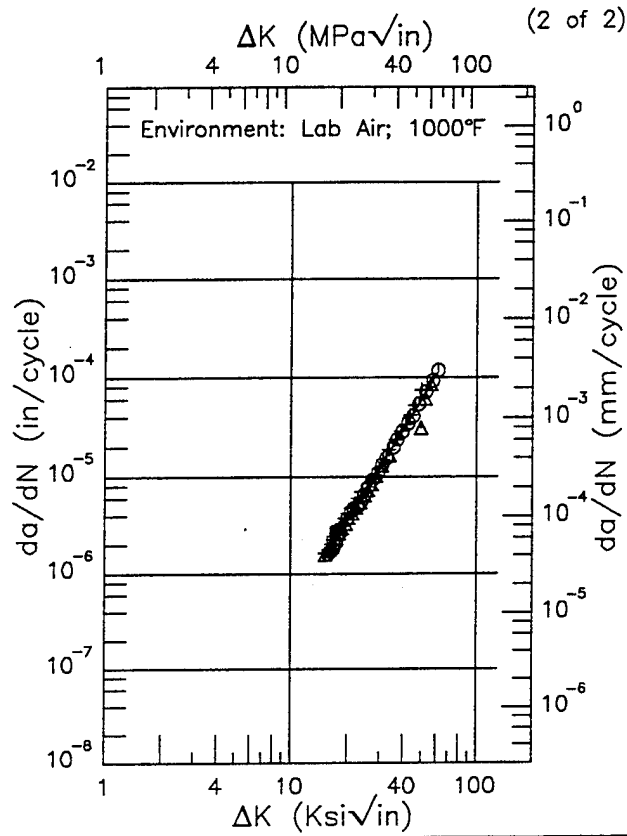
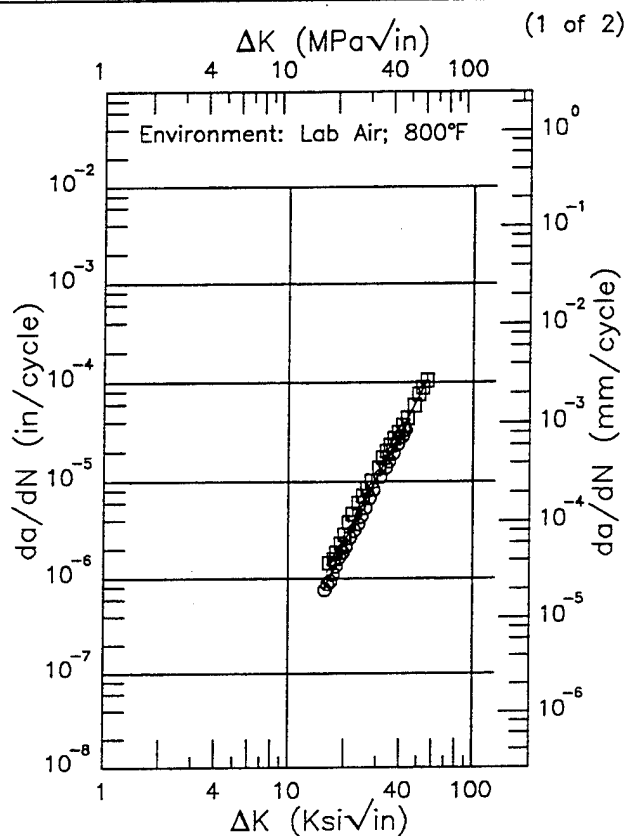
RMS % Error	Life Prediction Ratio Summary
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Figure 5.9.3.1.15

INCONEL 718

Condition/Ht: 1750F AC 1325F
 Form: 0.5 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.05
 Frequency: 0.7 - 6.7 Hz

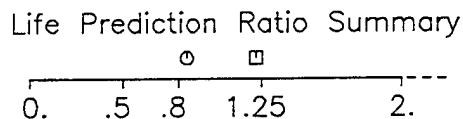
Yield Strength: 150.5 ksi
 Ult. Strength: 191.3 ksi
 Specimen Thk: 0.476 - 0.478 in.
 Specimen Width: 1.996 - 1.998 in.
 Ref: HD015



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
15.64 (min)	0.781
16.	0.871
20.	2.34
25.	5.62
30.	10.9
35.	18.8
40.	30.1
50.	68.5
56.07 (max)	107.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
15.20 (min)	1.52
16.	1.79
20.	3.62
25.	7.18
30.	12.5
35.	19.9
40.	29.8
50.	59.2
60.	106.
61.27 (max)	113.

RMS $\%$
 Error
 17.42



RMS $\%$
 Error
 9.25

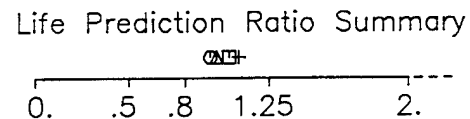


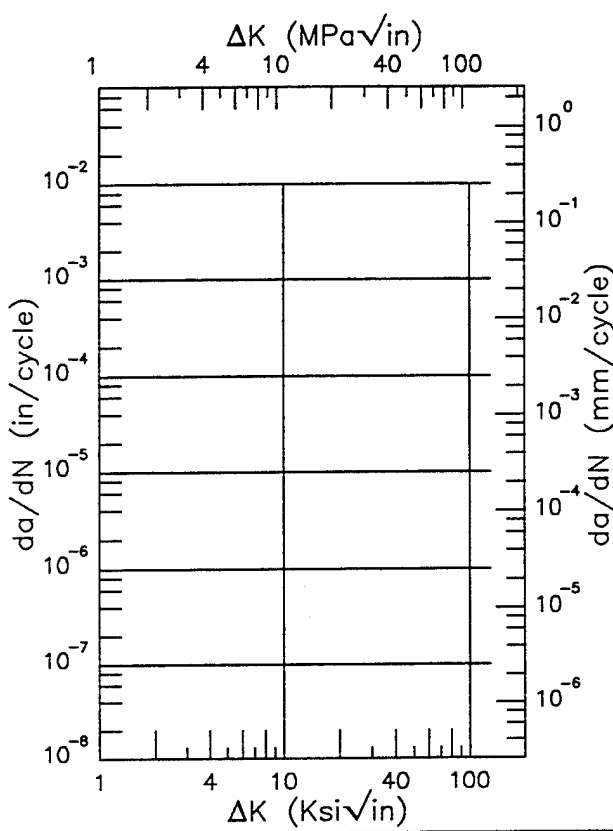
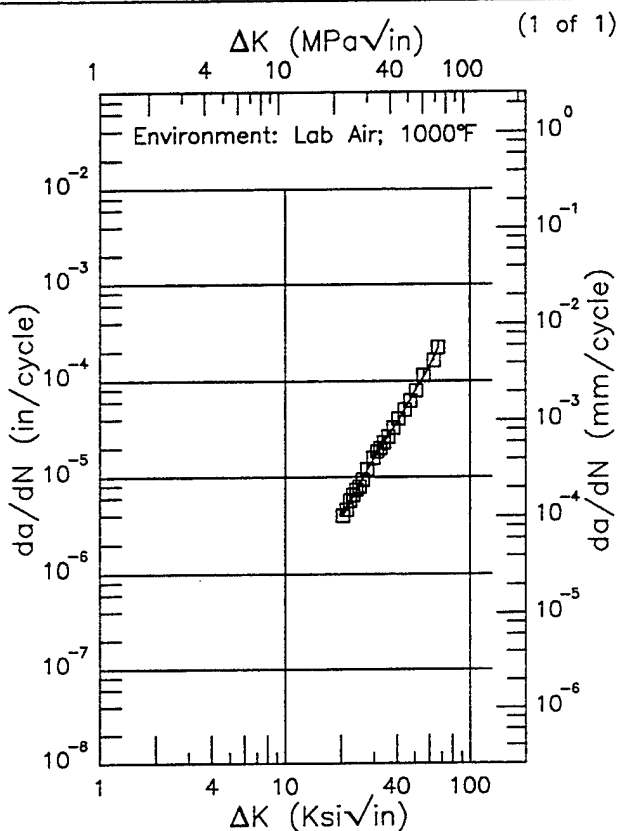
Figure 5.9.3.1.16

INCONEL 718

E

Condition/Ht: 1750F AC 1325F
 Form: 0.5 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

Yield Strength: 129 ksi
 Ult. Strength: 160.1 ksi
 Specimen Thk: 0.476 in.
 Specimen Width: 1.998 in.
 Ref: HD015



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
20.36 (min)	3.96
25.	8.47
30.	15.9
35.	25.3
40.	37.2
50.	78.2
60.	139.
66.35 (max)	215.

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS % Error 2.80	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.
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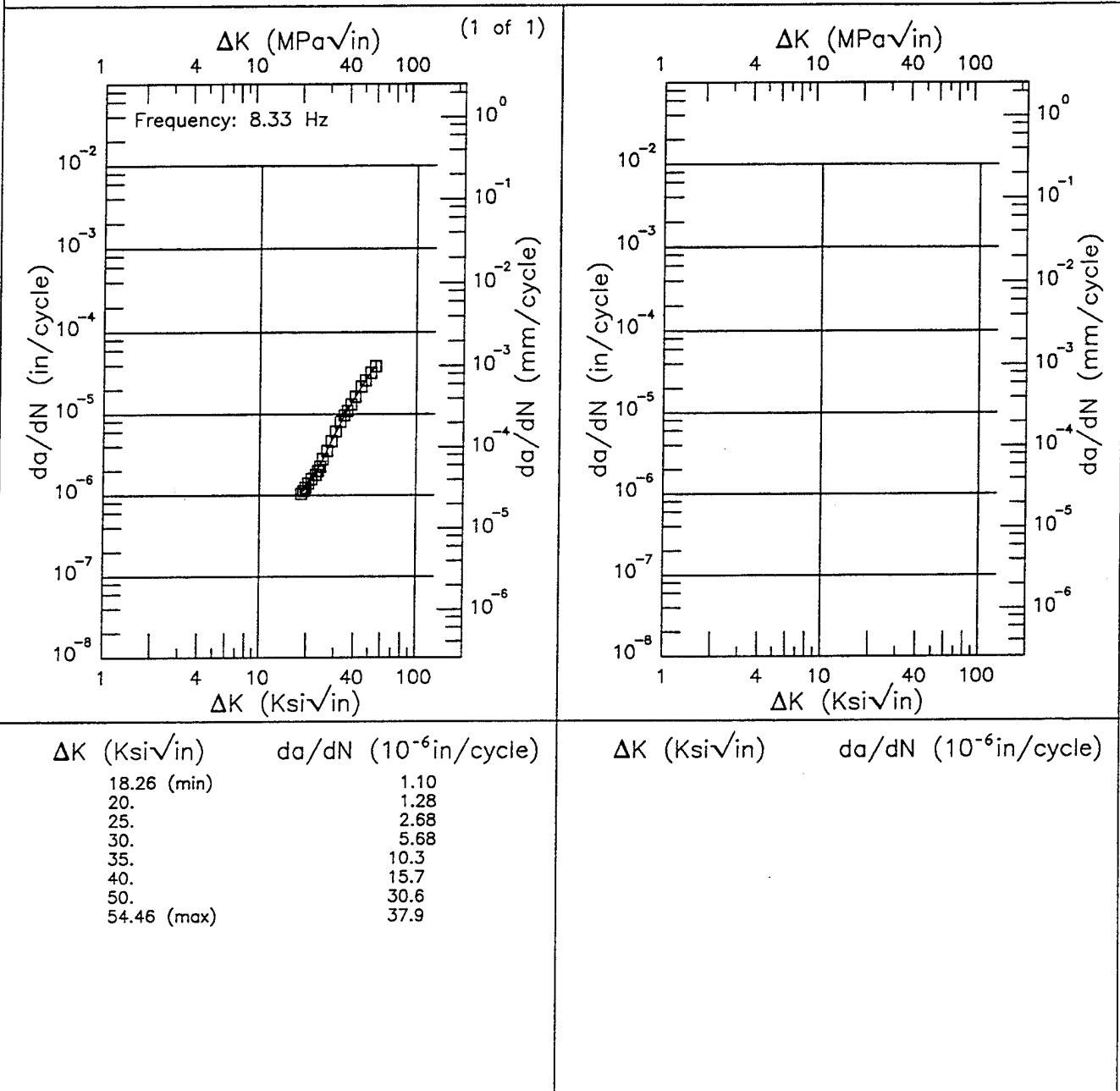
RMS % Error	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
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Figure 5.9.3.1.17

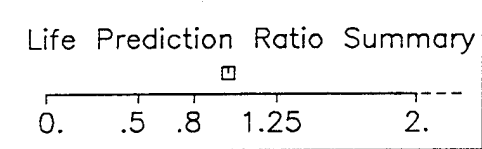
F | INCONEL 718 |

Condition/Ht: 1750F AC 1325F
 Form: 0.5 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.05
 Environment: LAB AIR; RT

Yield Strength: 150.5 ksi
 Ult. Strength: 191.3 ksi
 Specimen Thk: 0.477 in.
 Specimen Width: 1.998 in.
 Ref: HD015



RMS \times
 Error
 3.72



RMS \times
 Error

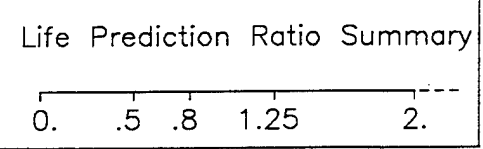


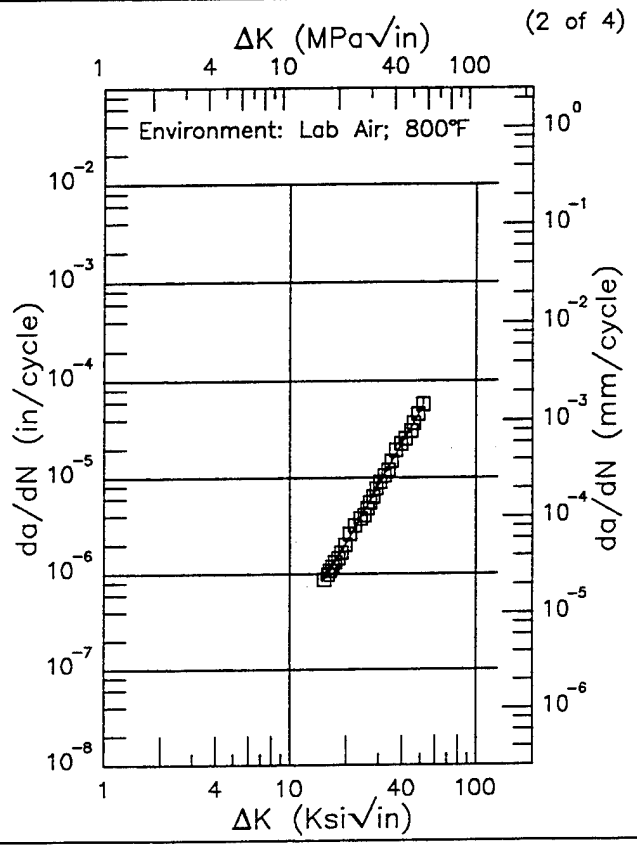
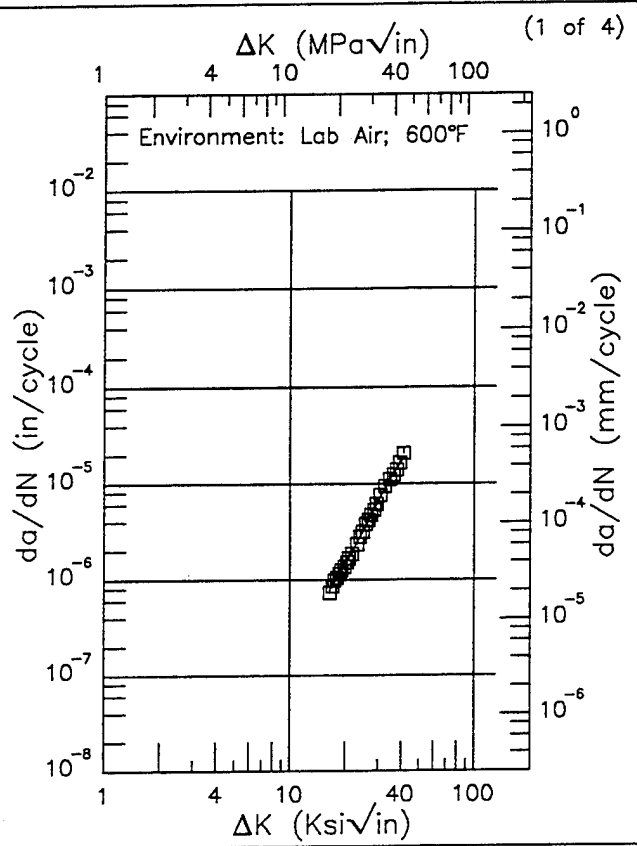
Figure 5.9.3.1.18

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INCONEL 718

Condition/Ht: 1750F AC 1325F
 Form: Forging
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

Yield Strength: 168.4 ksi
 Ult. Strength: 196.7 ksi
 Specimen Thk: 0.402 - 0.403 in.
 Specimen Width: 1.996 - 1.998 in.
 Ref: HD017



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.43 (min)	0.786
20.	1.44
25.	3.28
30.	6.61
35.	11.5
40.	17.4
41.28 (max)	19.0

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
15.22 (min)	0.849
16.	0.993
20.	2.05
25.	4.40
30.	8.32
35.	14.3
40.	22.7
50.	48.6
51.90 (max)	54.9

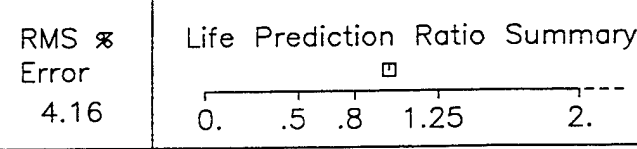
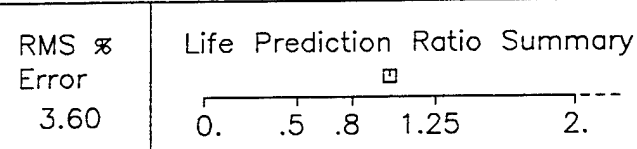


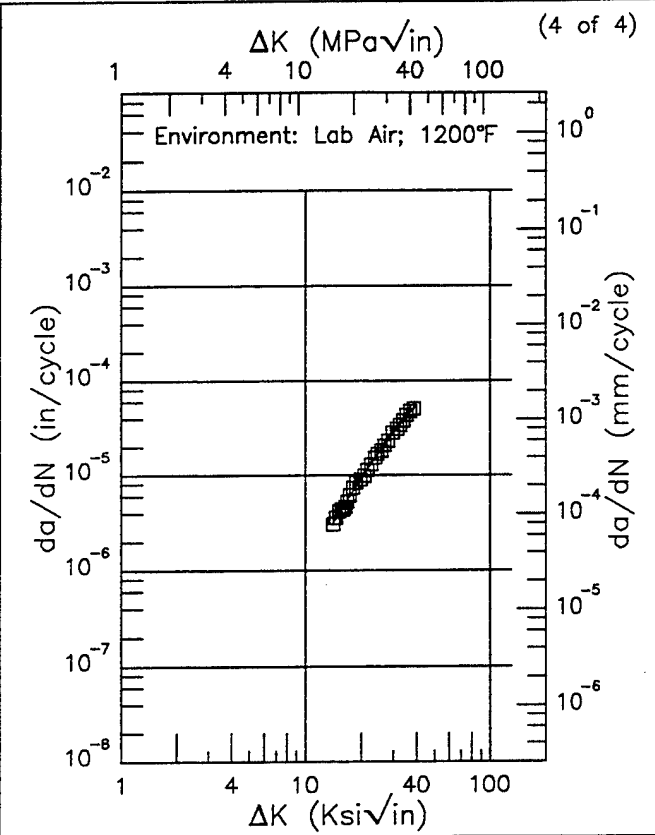
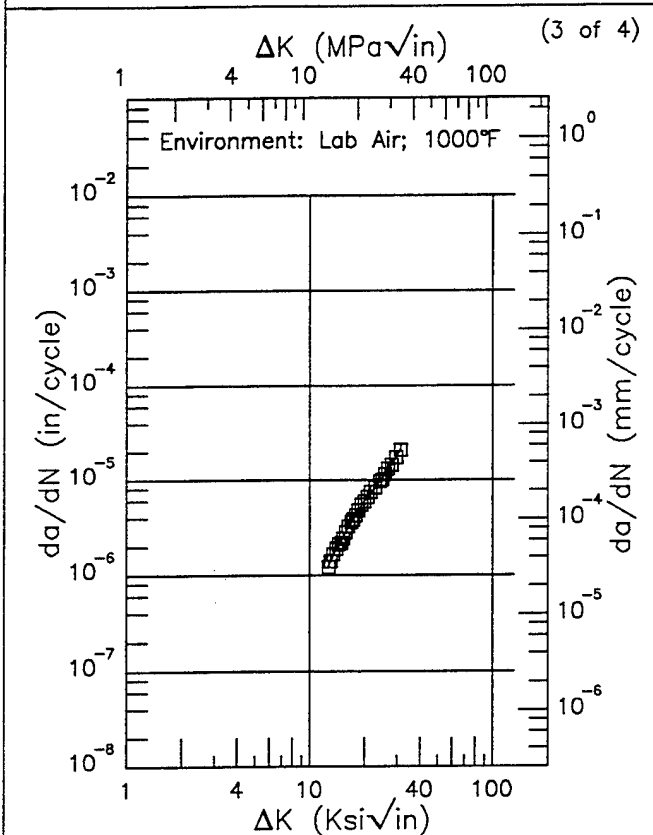
Figure 5.9.3.1.19

INCONEL 718

E

Condition/Ht: 1750F AC 1325F
 Form: Forging
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

Yield Strength: 168.4 ksi
 Ult. Strength: 196.7 ksi
 Specimen Thk: 0.402 - 0.403 in.
 Specimen Width: 1.996 - 1.998 in.
 Ref: HD017



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
12.64 (min)	1.24
13.	1.40
16.	3.02
20.	6.04
25.	10.5
30.	18.2
31.19 (max)	20.8

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
14.06 (min)	3.10
16.	4.69
20.	9.37
25.	17.3
30.	28.2
35.	42.3
38.42 (max)	50.8

RMS % Error 2.43	Life Prediction Ratio Summary
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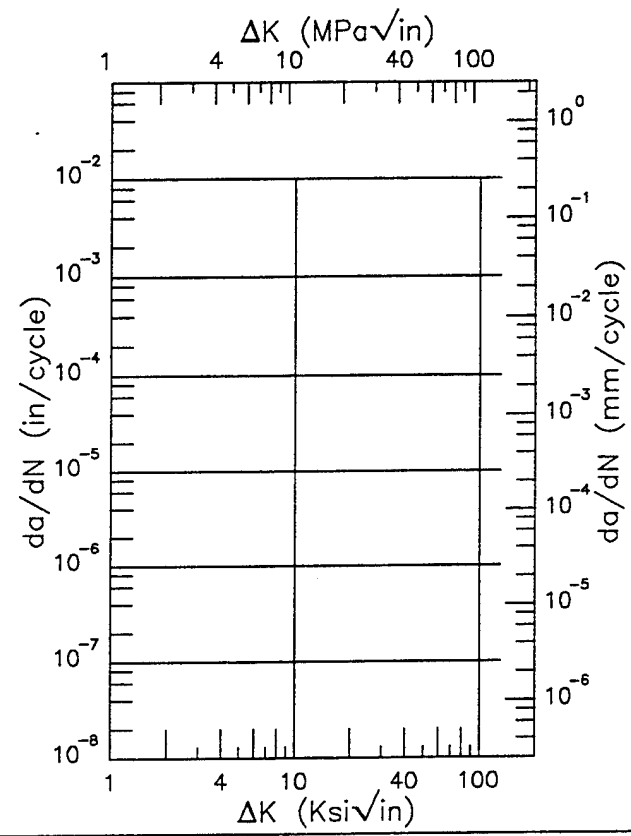
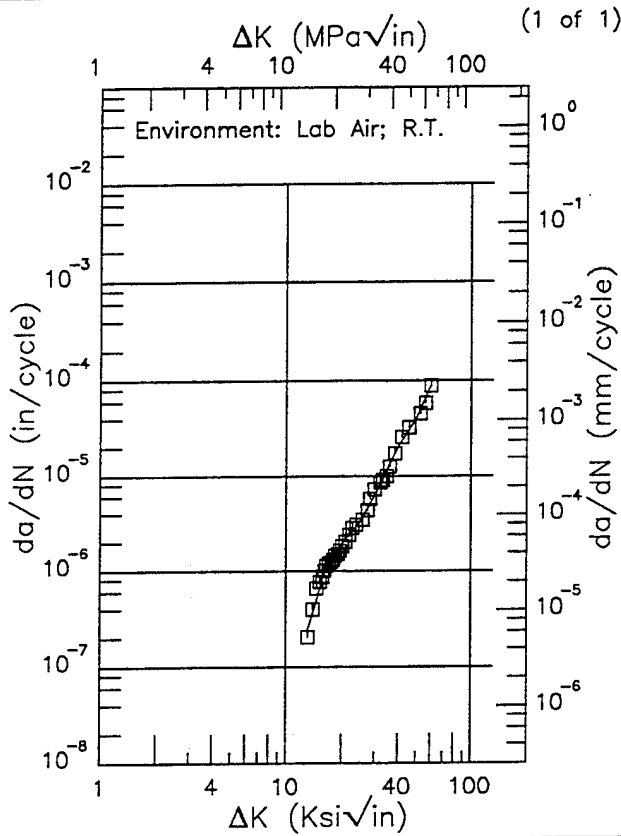
RMS % Error 4.00	Life Prediction Ratio Summary
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Figure 5.9.3.1.19 (Concluded)

INCONEL 718

Condition/Ht: 1750F AC 1325F
 Form: Forging
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05
 Frequency: 10 Hz

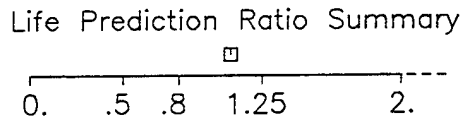
Yield Strength: 168.4 ksi
 Ult. Strength: 195.7 ksi
 Specimen Thk: 0.402 in.
 Specimen Width: 1.998 in.
 Ref: HD017



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
13.06 (min)	0.238
16.	0.925
20.	1.93
25.	3.33
30.	6.33
35.	12.0
40.	20.6
50.	40.7
59.94 (max)	88.8

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
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RMS $\%$
 Error
 9.56



RMS $\%$
 Error

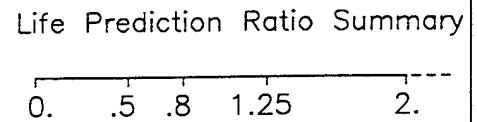


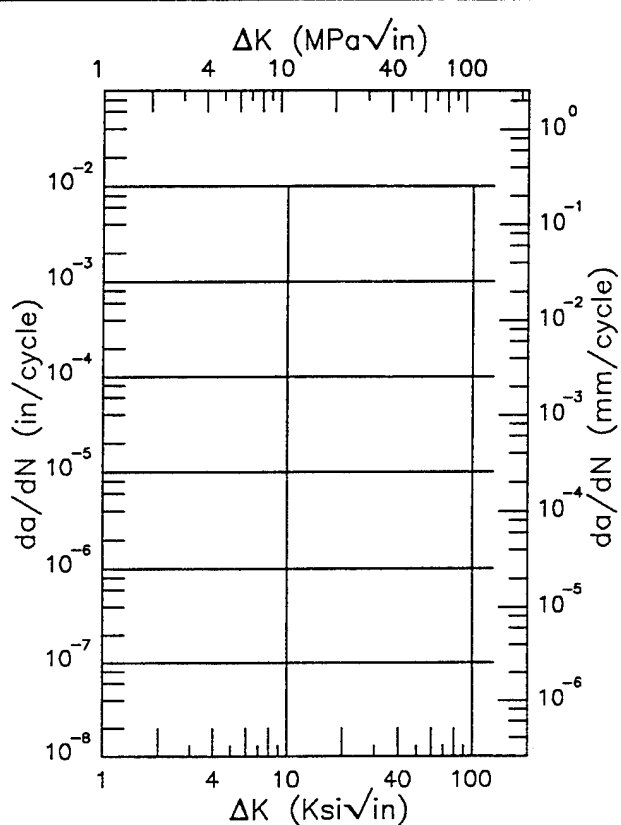
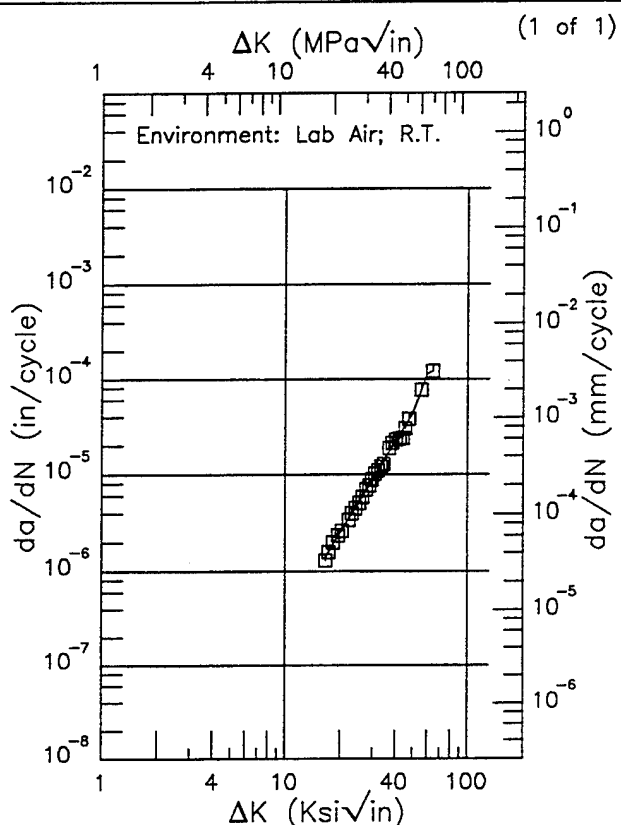
Figure 5.9.3.1.20

INCONEL 718

E

Condition/Ht: 1750F AC 1325F
 Form: 2 in. Forged Bar
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.05
 Frequency: 10 Hz

Yield Strength: 152.1 ksi
 Ult. Strength: 194.4 ksi
 Specimen Thk: 0.304 in.
 Specimen Width: 1.48 in.
 Ref: HD017



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.57 (min)	1.36
20.	2.56
25.	4.99
30.	8.81
35.	14.6
40.	21.8
50.	43.7
60.	114.
63.69 (max)	121.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
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RMS % Error	Life Prediction Ratio Summary
5.64	

RMS % Error	Life Prediction Ratio Summary

Figure 5.9.3.1.21

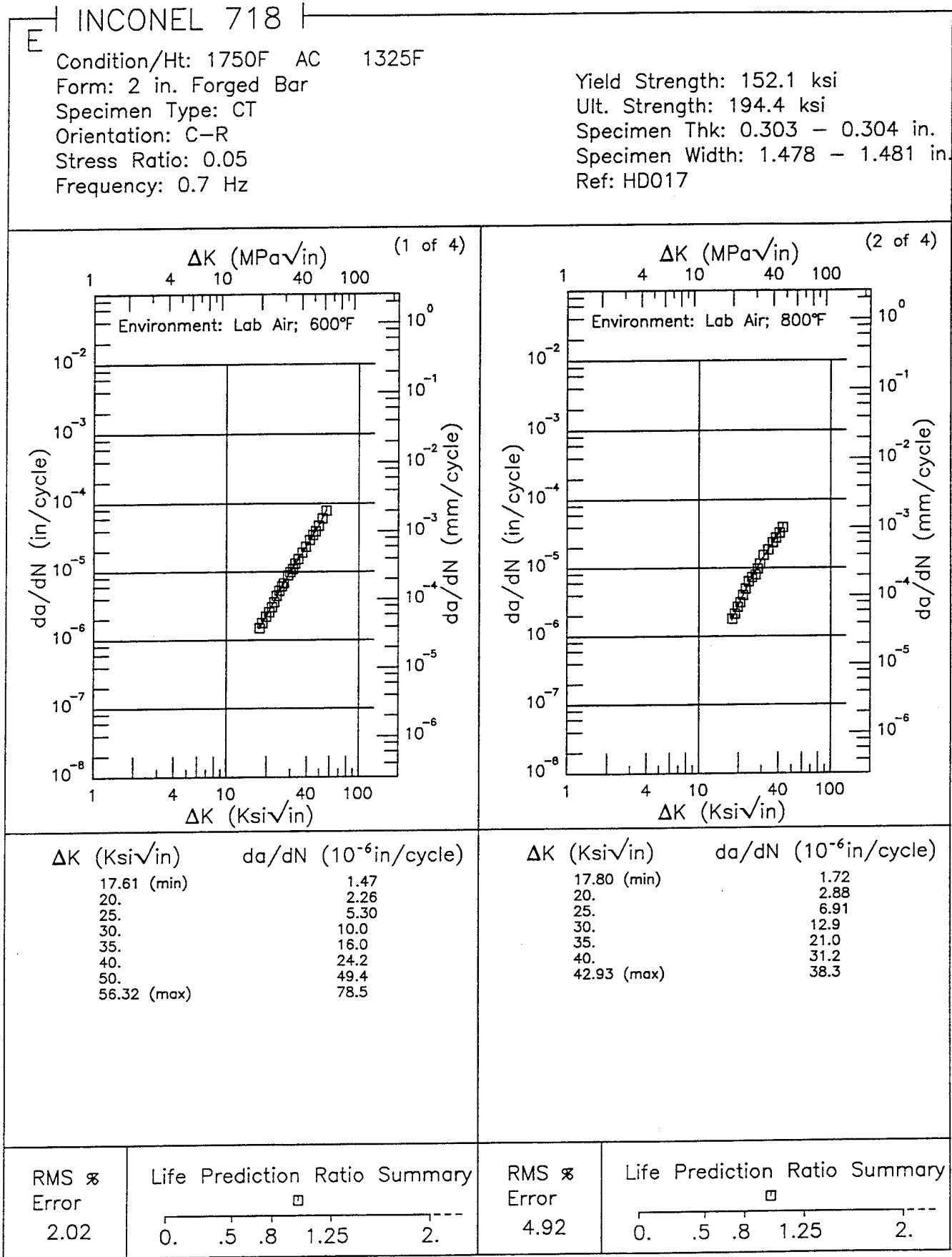
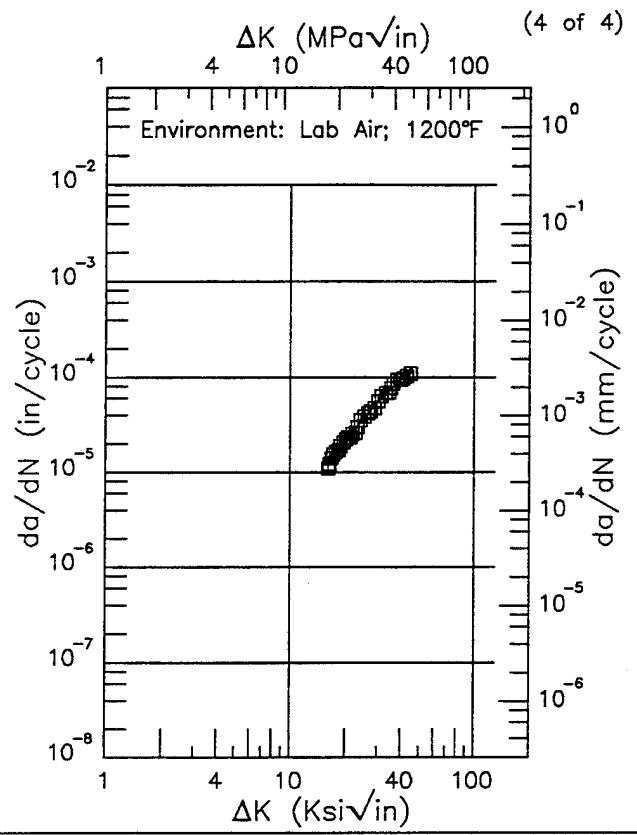
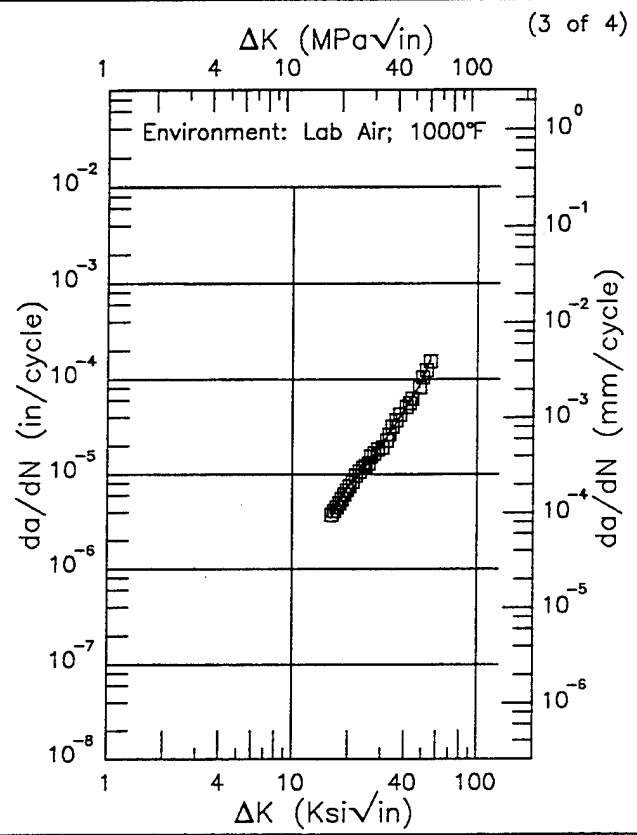


Figure 5.9.3.1.22

INCONEL 718 E

Condition/Ht: 1750F AC 1325F
 Form: 2 in. Forged Bar
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.05
 Frequency: 0.7 Hz

Yield Strength: 152.1 ksi
 Ult. Strength: 194.4 ksi
 Specimen Thk: 0.303 - 0.304 in.
 Specimen Width: 1.478 - 1.481 in.
 Ref: HD017



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.17 (min)	3.47
20.	7.35
25.	12.7
30.	19.9
35.	31.5
40.	47.6
50.	97.3
55.00 (max)	158.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.10 (min)	12.5
20.	21.0
25.	36.4
30.	55.6
35.	76.1
40.	95.3
44.50 (max)	110.

RMS $\%$ Error 4.05	Life Prediction Ratio Summary
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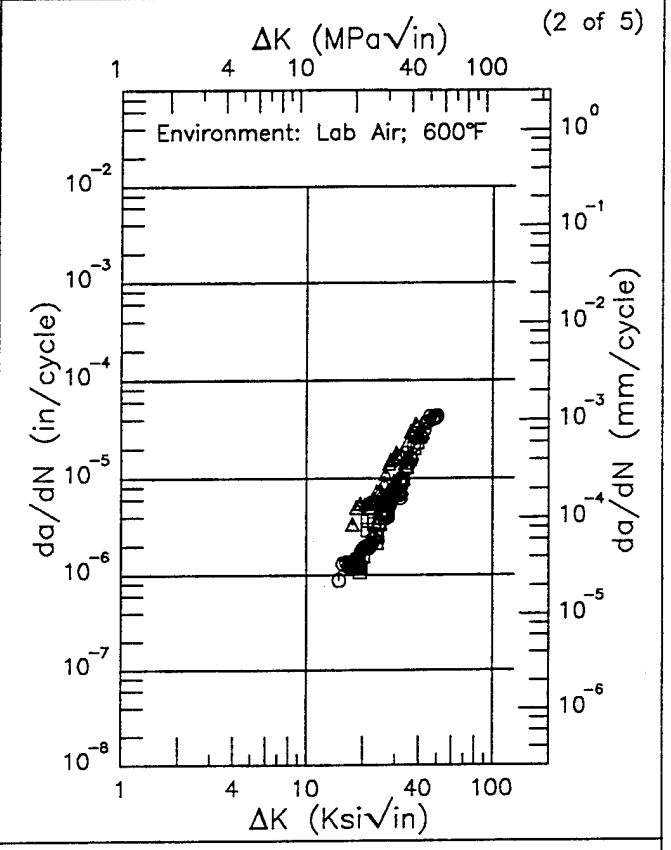
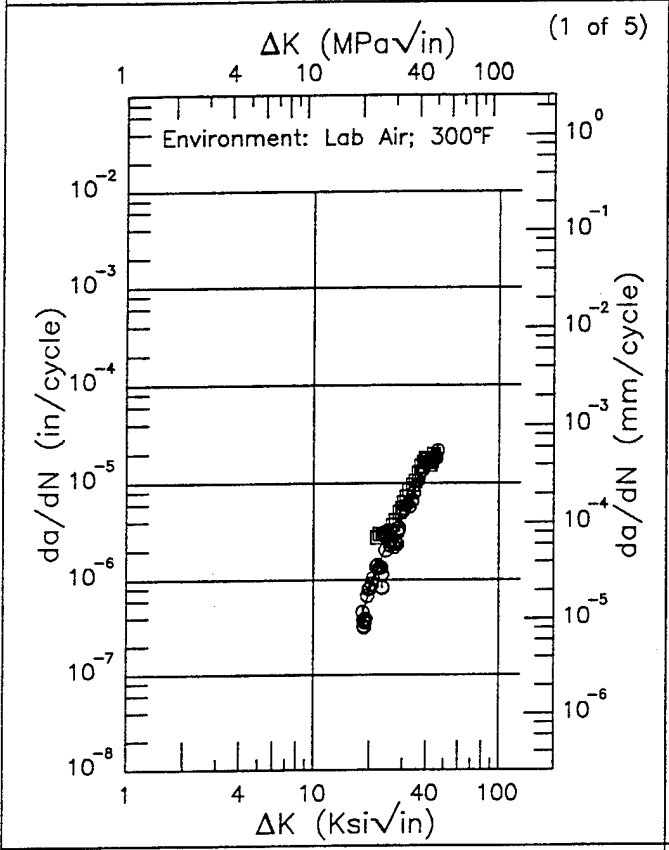
RMS $\%$ Error 4.63	Life Prediction Ratio Summary
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Figure 5.9.3.1.22 (Concluded)

INCONEL 718

Condition/Ht: 1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR
 Form: 1.13 - 1.3 in. Disk
 Specimen Type: KB Bar
 Orientation: C-R
 Stress Ratio: -0.33
 Frequency: 0.3 Hz

Yield Strength: 150 ksi
 Ult. Strength:
 Specimen Thk: 0.251 - 0.255 in.
 Specimen Width: 0.903 - 0.911 in.
 Ref: GE005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
18.27 (min)	0.428
20.	0.771
25.	2.54
30.	5.46
35.	9.46
40.	14.6
46.06 (max)	22.8

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
14.88 (min)	1.06
16.	1.19
20.	2.08
25.	4.42
30.	8.75
35.	15.6
40.	24.9
49.95 (max)	47.9

RMS % Error 29.13	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
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RMS % Error 50.94	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

Figure 5.9.3.1.23

INCONEL 718

E

Condition/Ht: 1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR
 Form: 1.13 - 1.3 in. Disk Yield Strength: 150 ksi
 Specimen Type: KB Bar Ult. Strength:
 Orientation: C-R Specimen Thk: 0.251 - 0.255 in.
 Stress Ratio: -0.33 Specimen Width: 0.903 - 0.911 in.
 Frequency: 0.3 Hz Ref: GE005

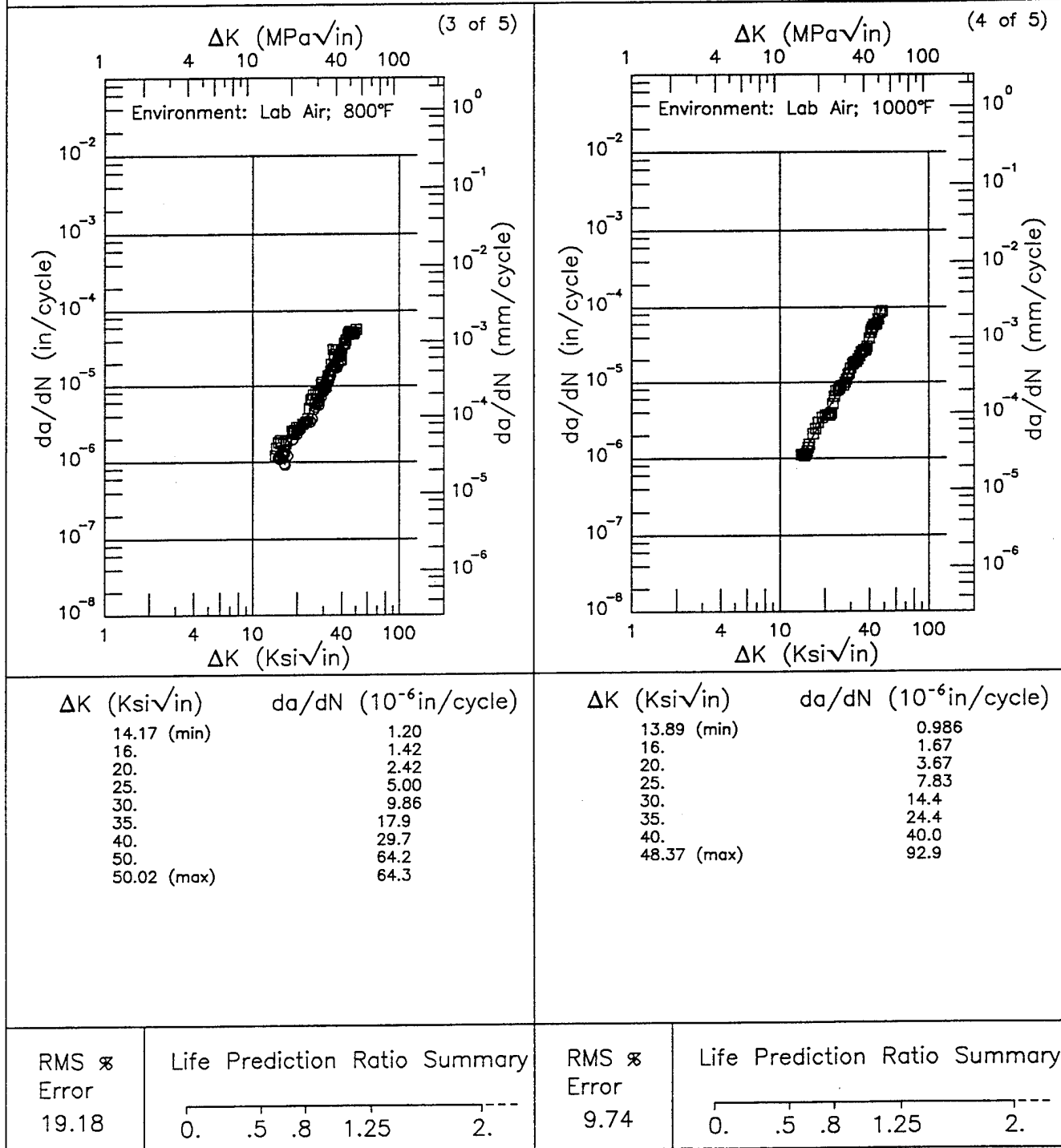
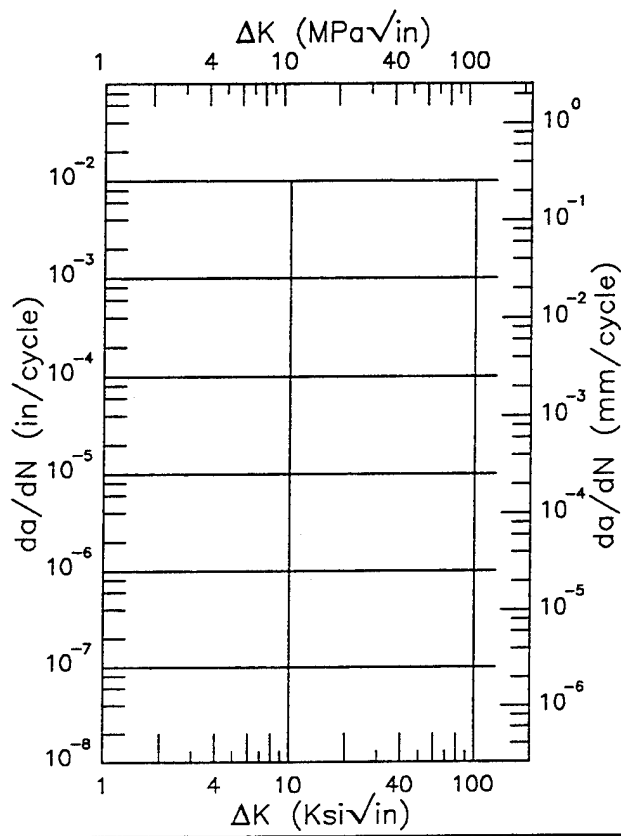
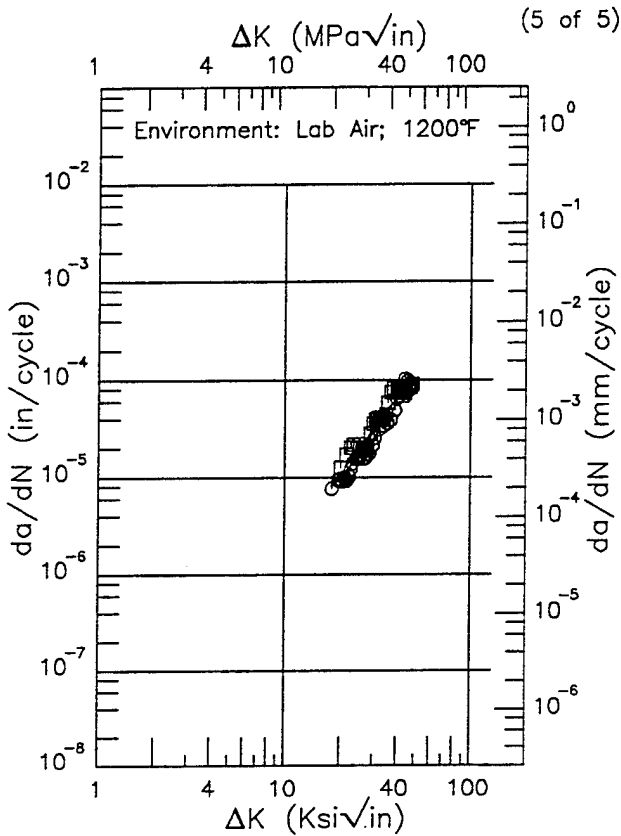


Figure 5.9.3.1.23 (Continued)

INCONEL 718

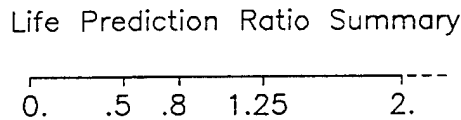
Condition/Ht: 1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR
 Form: 1.13 - 1.3 in. Disk Yield Strength: 150 ksi
 Specimen Type: KB Bar Ult. Strength:
 Orientation: C-R Specimen Thk: 0.251 - 0.255 in.
 Stress Ratio: -0.33 Specimen Width: 0.903 - 0.911 in.
 Frequency: 0.3 Hz Ref: GE005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
17.86 (min)	8.13
20.	9.75
25.	16.8
30.	28.8
35.	45.6
40.	65.4
48.27 (max)	95.9

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
---------------------	-------------------------------

RMS \times
Error
19.95



RMS \times
Error

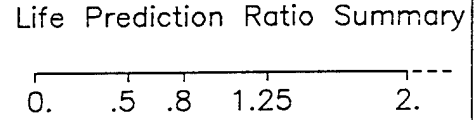


Figure 5.9.3.1.23 (Concluded)

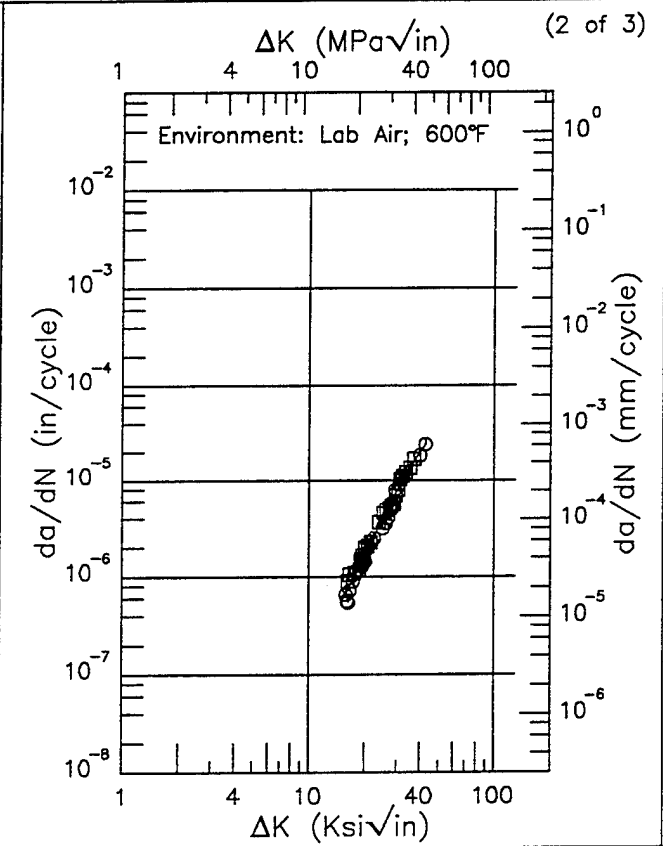
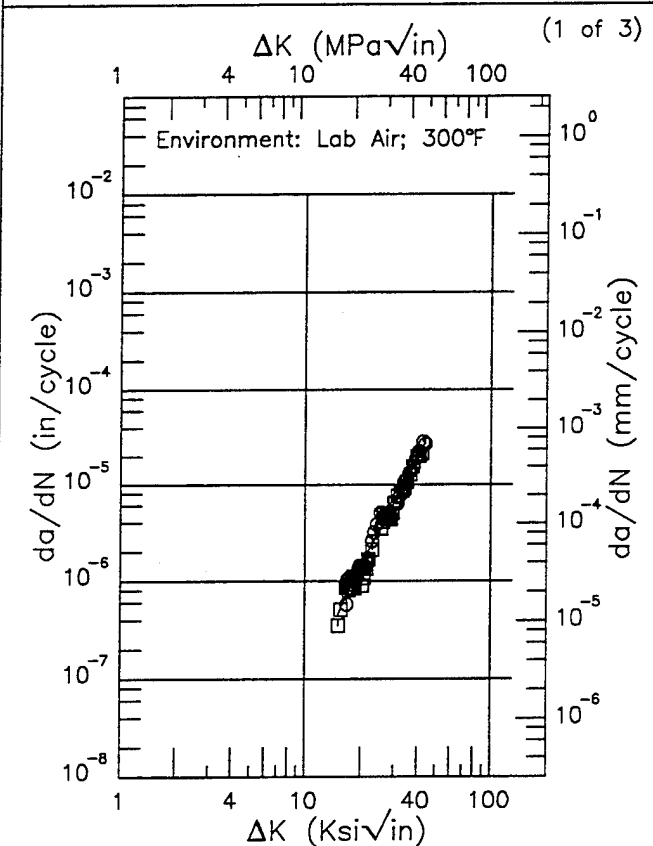
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INCONEL 718

E

Condition/Ht: 1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR
 Form: 1.13 in. Disk
 Specimen Type: KB Bar
 Orientation: C-R
 Stress Ratio: 0.
 Frequency: 0.3 Hz

Yield Strength: 150 ksi
 Ult. Strength:
 Specimen Thk: 0.25 - 0.252 in.
 Specimen Width: 0.9 - 0.906 in.
 Ref: GE005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
15.09 (min)	0.456
16.	0.592
20.	1.46
25.	3.28
30.	6.27
35.	11.1
40.	18.8
44.05 (max)	28.5

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
15.82 (min)	0.655
16.	0.682
20.	1.59
25.	3.84
30.	7.72
35.	13.2
40.	20.0
42.52 (max)	23.6

RMS $\%$ Error	Life Prediction Ratio Summary
16.69	

RMS $\%$ Error	Life Prediction Ratio Summary
13.10	

Figure 5.9.3.1.24

INCONEL 718

Condition/Ht: 1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR
 Form: 1.13 in. Disk
 Specimen Type: KB Bar
 Orientation: C-R
 Stress Ratio: 0.
 Frequency: 0.3 Hz

Yield Strength: 150 ksi
 Ult. Strength:
 Specimen Thk: 0.25 - 0.252 in.
 Specimen Width: 0.9 - 0.906 in.
 Ref: GE005

E

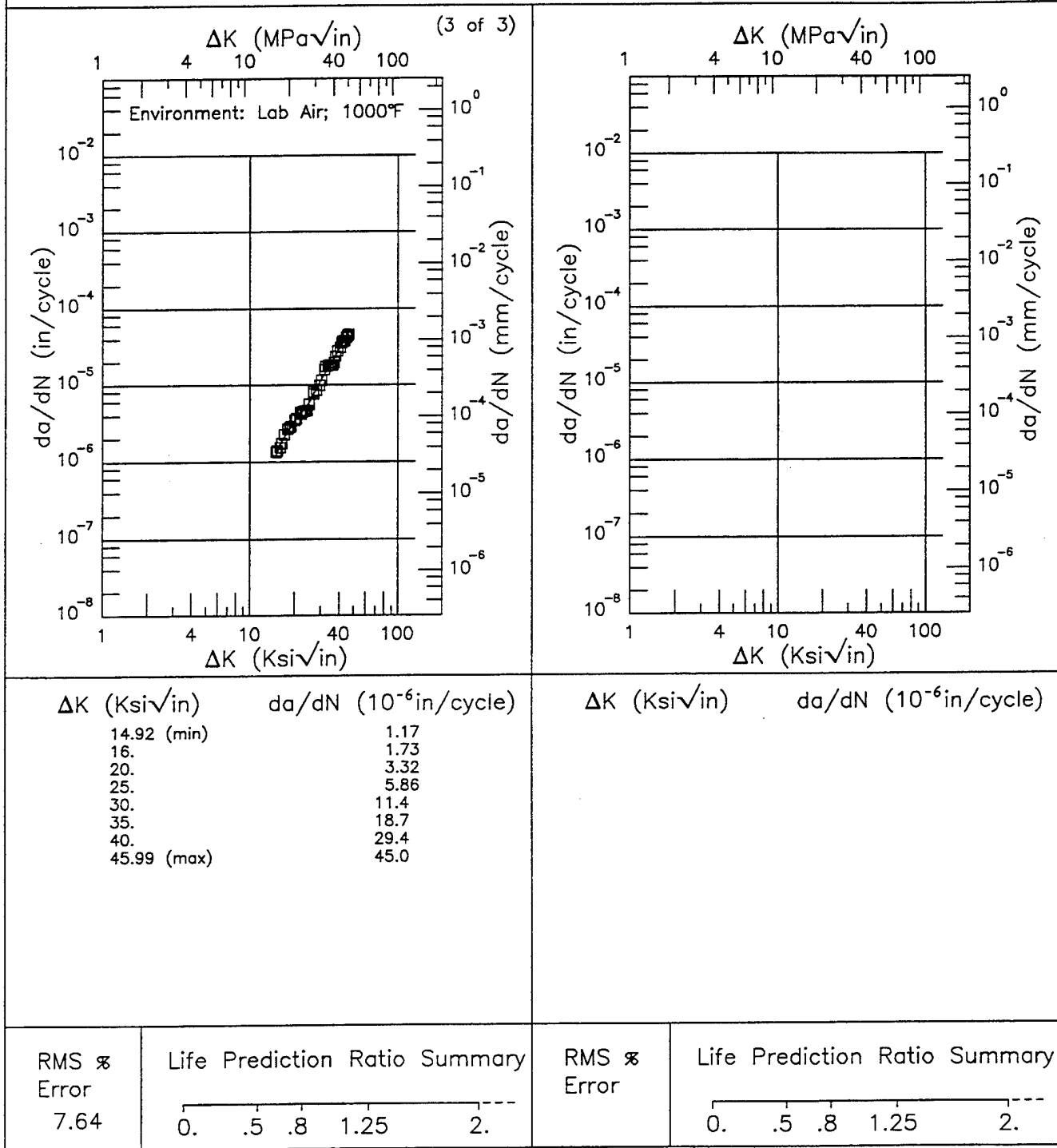
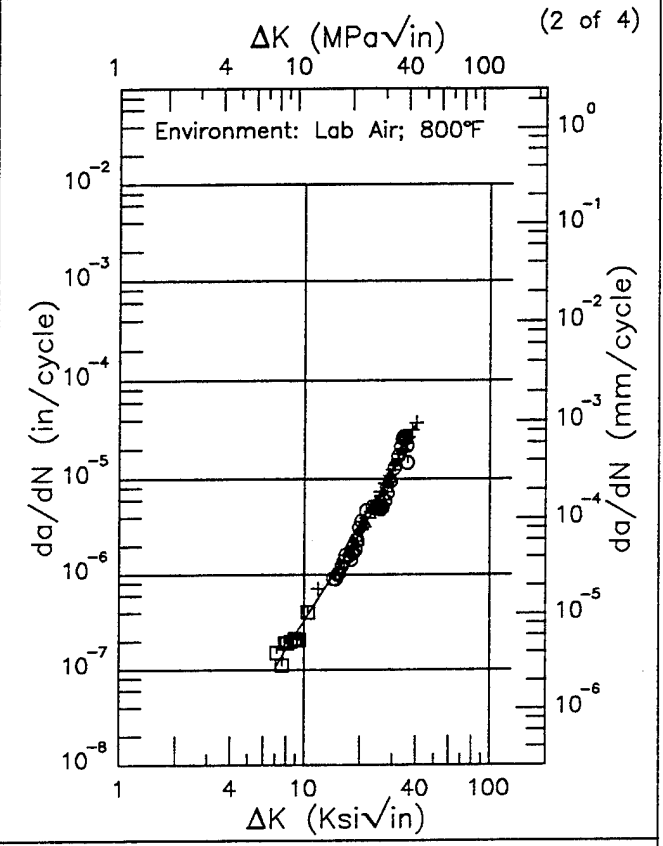
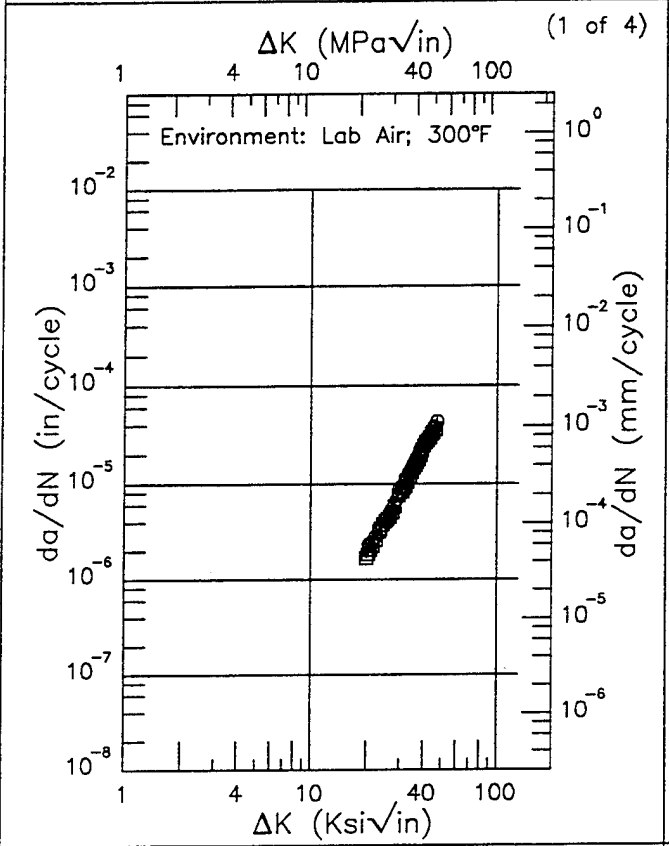


Figure 5.9.3.1.24 (Concluded)

E | INCONEL 718 |

Condition/Ht: 1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR
 Form: 1.13 - 1.3 in. Disk
 Specimen Type: KB Bar
 Orientation: C-R
 Stress Ratio: 0.03
 Frequency: 0.3 Hz

Yield Strength: 150 ksi
 Ult. Strength:
 Specimen Thk: 0.25 - 0.254 in.
 Specimen Width: 0.9 - 0.91 in.
 Ref: GE005



ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
19.86 (min)	1.82
20.	1.86
25.	3.98
30.	7.94
35.	14.3
40.	23.4
47.23 (max)	40.8

ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
7.10 (min)	0.111
8.	0.165
9.	0.240
10.	0.332
13.	0.731
16.	1.36
20.	2.72
25.	5.85
30.	12.3
35.	23.9
40.	35.9
40.11 (max)	36.1

RMS % Error	Life Prediction Ratio Summary
5.59	

RMS % Error	Life Prediction Ratio Summary
15.90	

Figure 5.9.3.1.25

INCONEL 718

E

Condition/Ht: 1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR
 Form: 1.13 - 1.3 in. Disk
 Specimen Type: KB Bar
 Orientation: C-R
 Stress Ratio: 0.03
 Frequency: 0.3 Hz
 Yield Strength: 150 ksi
 Ult. Strength:
 Specimen Thk: 0.25 - 0.254 in.
 Specimen Width: 0.9 - 0.91 in.
 Ref: GE005

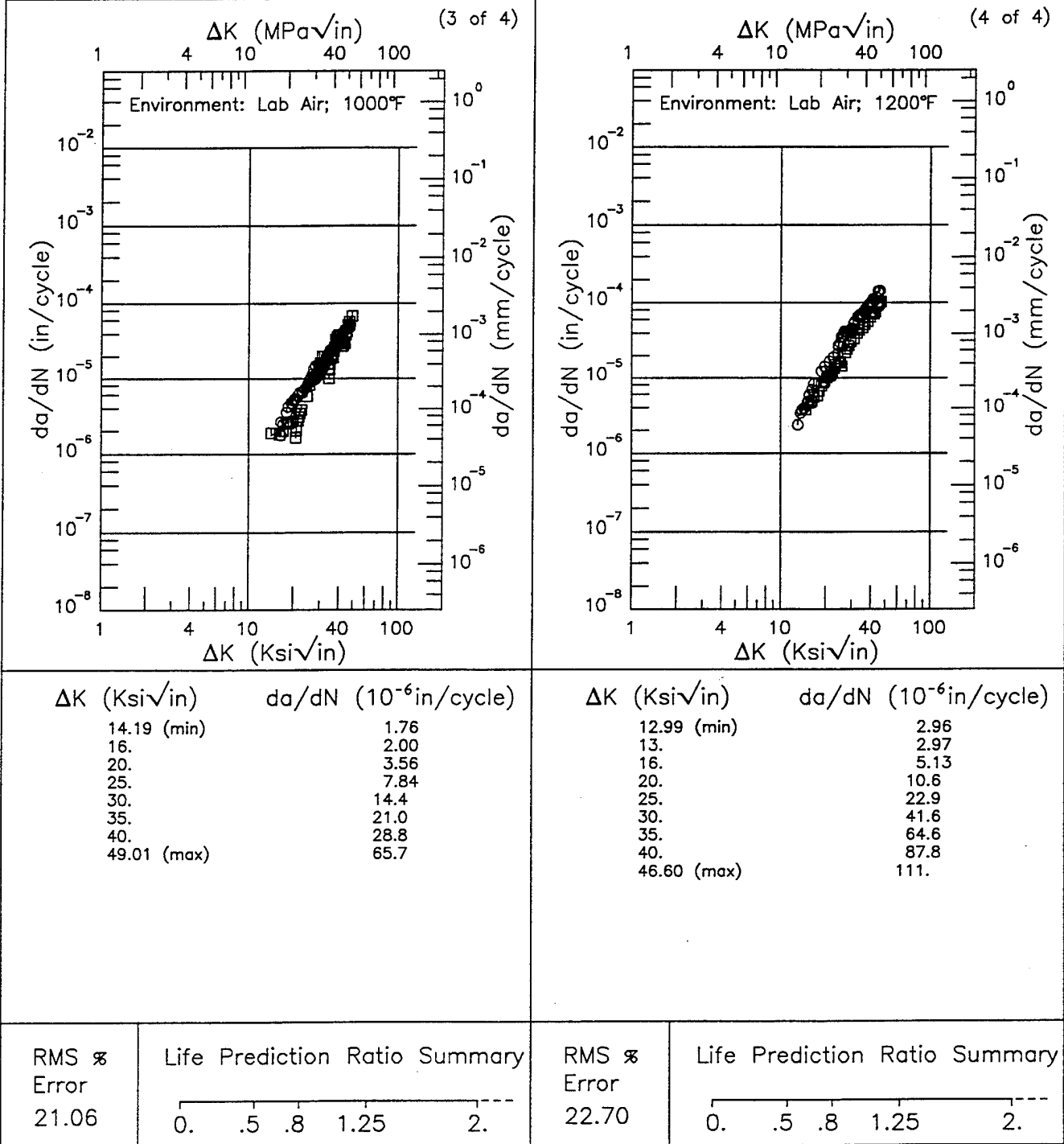
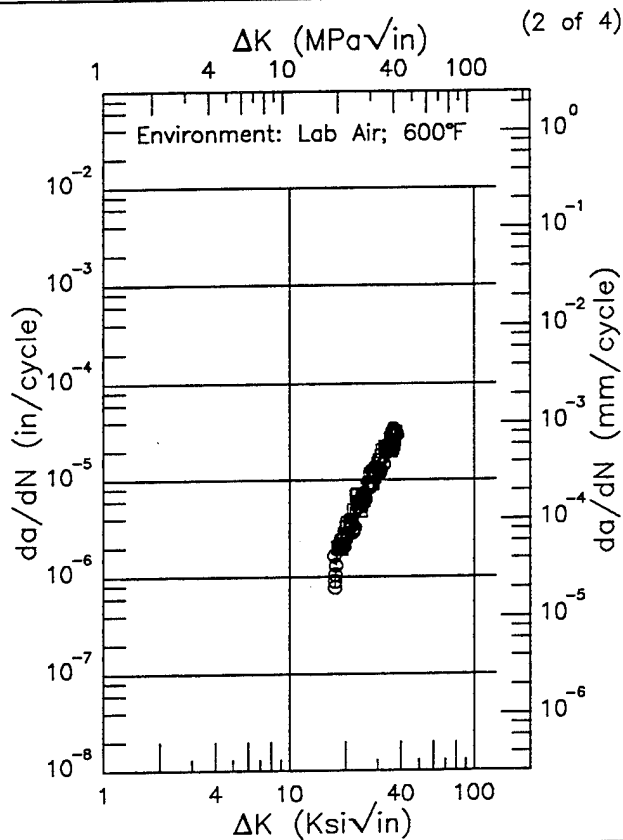
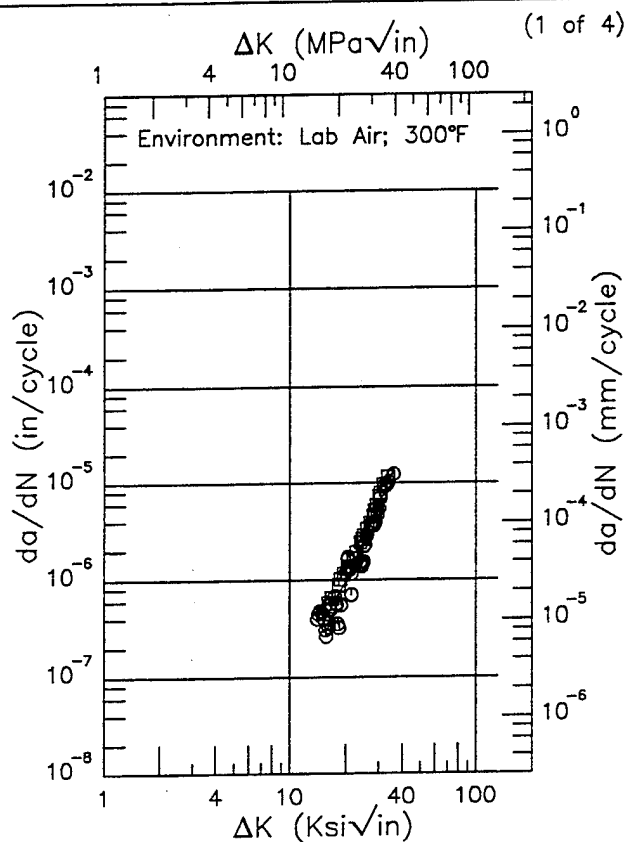


Figure 5.9.3.1.25 (Concluded)

INCONEL 718

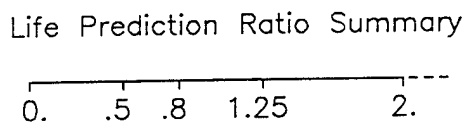
Condition/Ht: 1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR
 Form: 1.13 - 1.3 in. Disk Yield Strength: 150 ksi
 Specimen Type: KB Bar Ult. Strength:
 Orientation: C-R Specimen Thk: 0.25 - 0.253 in.
 Stress Ratio: 0.25 Specimen Width: 0.9 - 0.909 in.
 Frequency: 0.3 Hz Ref: GE005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.99 (min)	0.373
16.	0.465
20.	0.949
25.	2.53
30.	6.03
35.	12.1
35.84 (max)	13.4

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
17.33 (min)	1.36
20.	2.67
25.	6.83
30.	13.8
35.	24.3
38.01 (max)	32.8

RMS $\%$
Error
23.83



RMS $\%$
Error
16.68

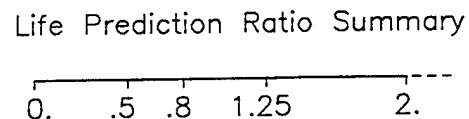
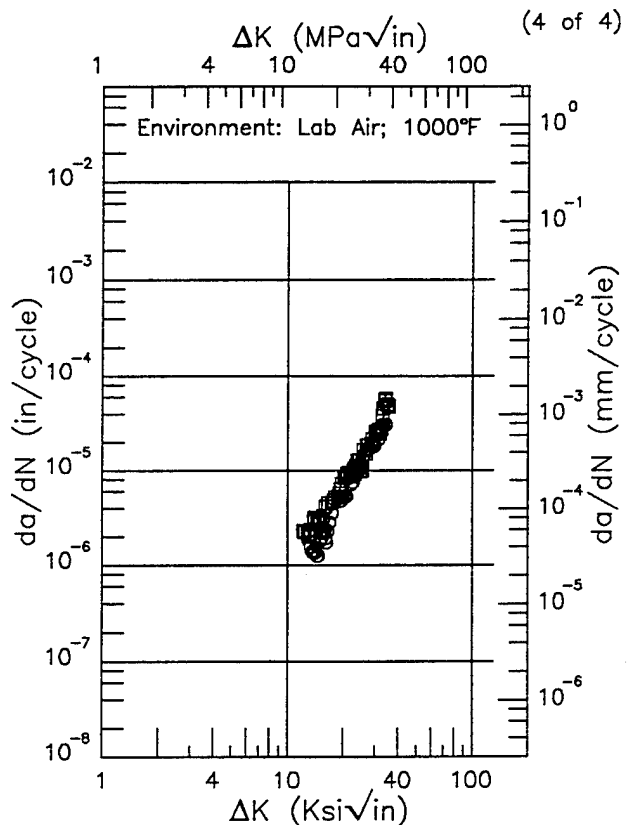
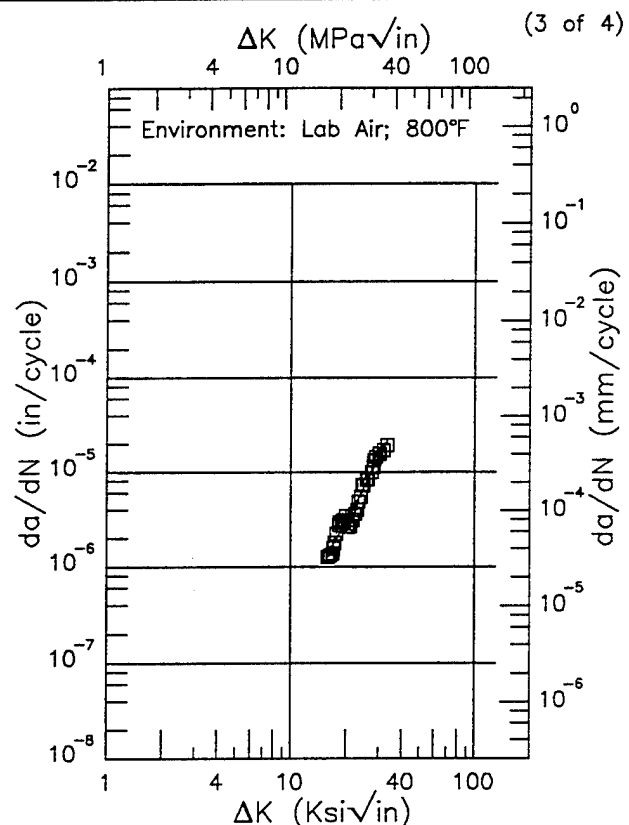


Figure 5.9.3.1.26

INCONEL 718

E

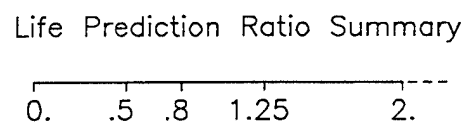
Condition/Ht: 1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR
 Form: 1.13 - 1.3 in. Disk Yield Strength: 150 ksi
 Specimen Type: KB Bar Ult. Strength:
 Orientation: C-R Specimen Thk: 0.25 - 0.253 in.
 Stress Ratio: 0.25 Specimen Width: 0.9 - 0.909 in.
 Frequency: 0.3 Hz Ref: GE005



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
15.78 (min)	1.36
16.	1.41
20.	2.91
25.	6.95
30.	14.4
33.30 (max)	21.3

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
12.17 (min)	1.63
13.	1.80
16.	2.89
20.	5.80
25.	13.0
30.	25.5
35.	43.8
35.35 (max)	45.3

RMS % Error
15.48



RMS % Error
22.77

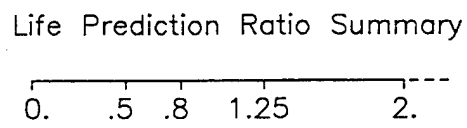
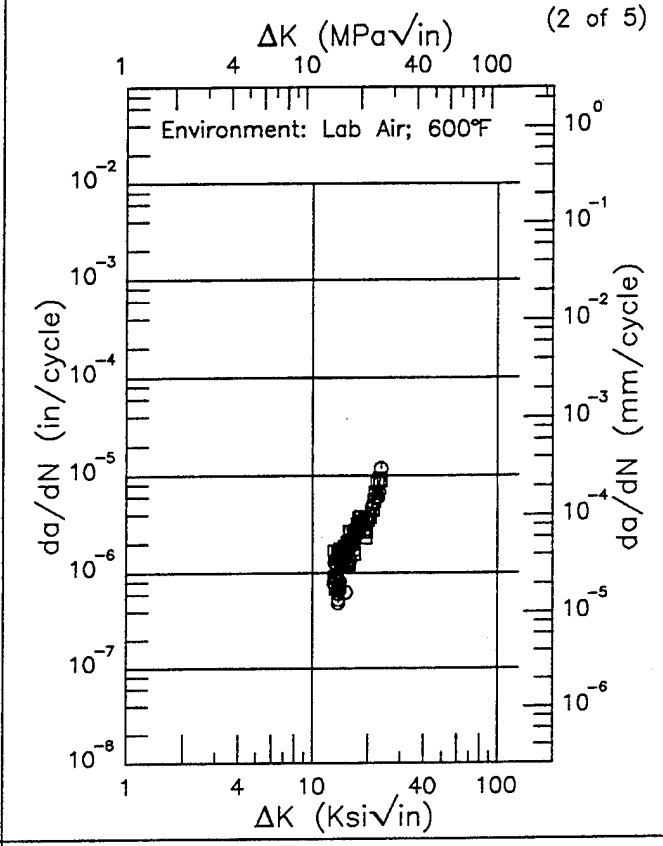
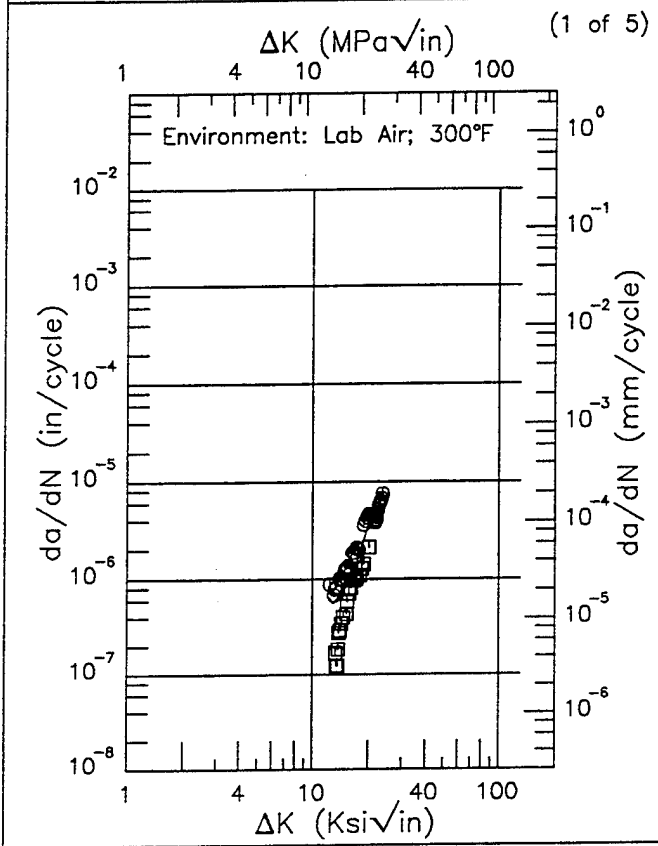


Figure 5.9.3.1.26 (Concluded)

INCONEL 718

Condition/Ht: 1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR
 Form: 1.13 - 1.3 in. Disk Yield Strength: 150 ksi
 Specimen Type: KB Bar Ult. Strength:
 Orientation: C-R Specimen Thk: 0.25 - 0.255 in.
 Stress Ratio: 0.54 Specimen Width: 0.902 - 0.922 in.
 Frequency: 0.3 Hz Ref: GE005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
12.37 (min)	0.643
13.	0.555
16.	0.934
20.	3.59
23.59 (max)	5.83

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
12.93 (min)	1.08
13.	1.07
16.	1.97
20.	3.63
23.39 (max)	9.65

RMS % Error	Life Prediction Ratio Summary
41.56	

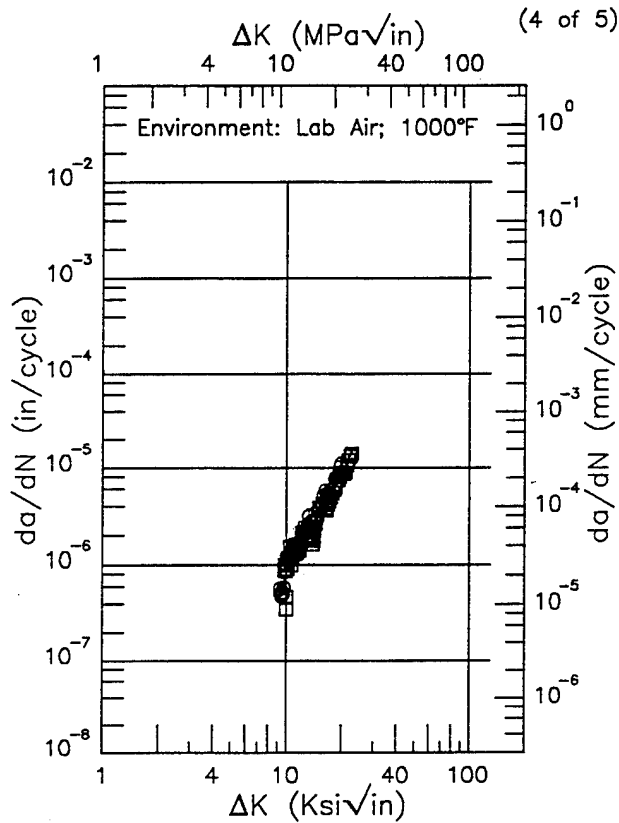
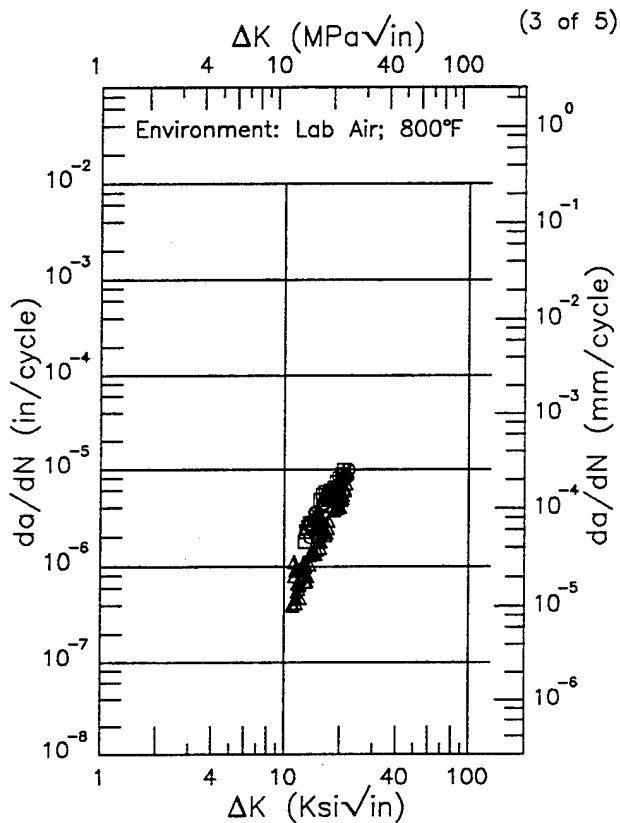
RMS % Error	Life Prediction Ratio Summary
23.62	

Figure 5.9.3.1.27

INCONEL 718

Condition/Ht: 1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR
 Form: 1.13 - 1.3 in. Disk Yield Strength: 150 ksi
 Specimen Type: KB Bar Ult. Strength:
 Orientation: C-R Specimen Thk: 0.25 - 0.255 in.
 Stress Ratio: 0.54 Specimen Width: 0.902 - 0.922 in.
 Frequency: 0.3 Hz Ref: GE005

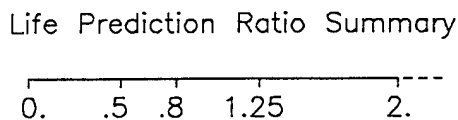
E



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
11.11 (min)	0.592
13.	1.14
16.	3.30
20.	5.78
21.92 (max)	9.00

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
9.26 (min)	0.485
10.	0.820
13.	2.22
16.	3.86
20.	8.98
22.64 (max)	12.6

RMS %
Error
35.53



RMS %
Error
19.95

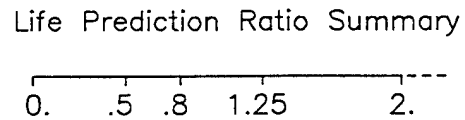
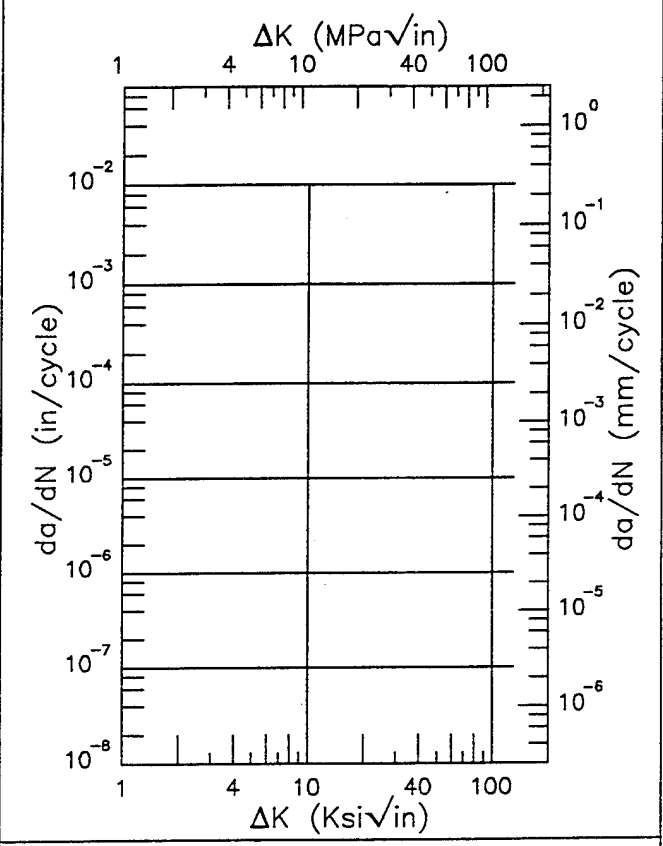
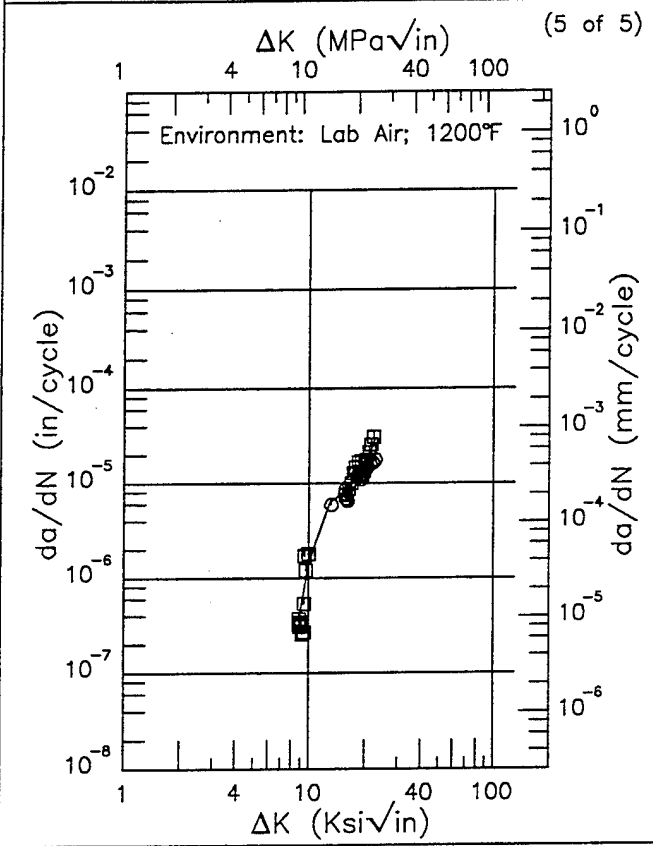


Figure 5.9.3.1.27 (Continued)

E | INCONEL 718 |

Condition/Ht: 1750F 1HR Q 1325F 8HR FC TO 1150F AT 100F/HR HOLD 8HR
 Form: 1.13 - 1.3 in. Disk Yield Strength: 150 ksi
 Specimen Type: KB Bar Ult. Strength:
 Orientation: C-R Specimen Thk: 0.25 - 0.255 in.
 Stress Ratio: 0.54 Specimen Width: 0.902 - 0.922 in.
 Frequency: 0.3 Hz Ref: GE005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.80 (min)	0.304
9.	0.424
10.	1.47
13.	6.28
16.	8.59
20.	14.4
22.63 (max)	29.7

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.80 (min)	0.304
9.	0.424
10.	1.47
13.	6.28
16.	8.59
20.	14.4
22.63 (max)	29.7

RMS % Error	Life Prediction Ratio Summary
30.20	0. .5 .8 1.25 2.

RMS % Error	Life Prediction Ratio Summary
	0. .5 .8 1.25 2.

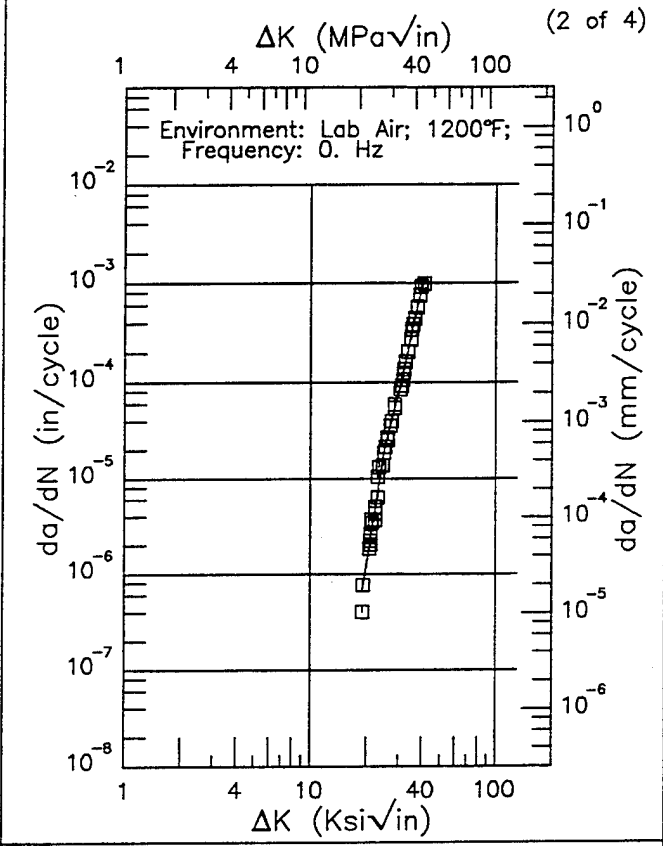
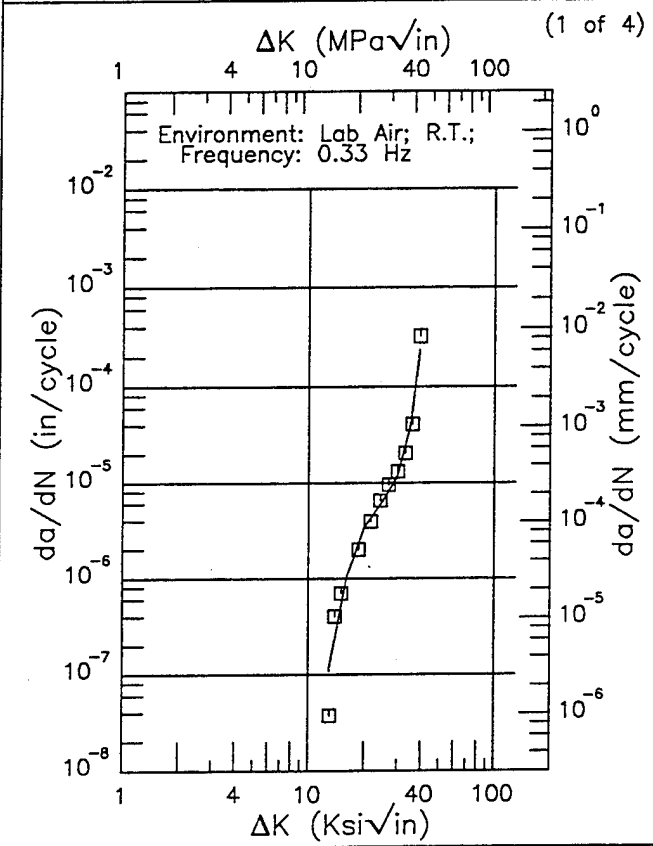
Figure 5.9.3.1.27 (Concluded)

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EF | INCONEL 718 |

Condition/Ht: 1760F 1HR WQ 1325F 8HRS
 Form: 3.6 in. Disk
 Specimen Type: CCP (max stress specified)
 Orientation: C-R
 Stress Ratio: 0.

Yield Strength: 171.6 ksi
 Ult. Strength: 201.7 ksi
 Specimen Thk: 0.08 in.
 Specimen Width: 2 in.
 Ref: GE008



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
12.93 (min)	0.108
13.	0.116
16.	1.01
20.	3.39
25.	6.49
30.	11.9
35.	41.2
39.18 (max)	236.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
19.03 (min)	0.685
20.	1.21
25.	20.1
30.	80.0
35.	311.
40.	916.
41.00 (max)	1009.

RMS % Error	Life Prediction Ratio Summary
38.40	0. .5 .8 1.25 2.

RMS % Error	Life Prediction Ratio Summary
18.18	0. .5 .8 1.25 2.

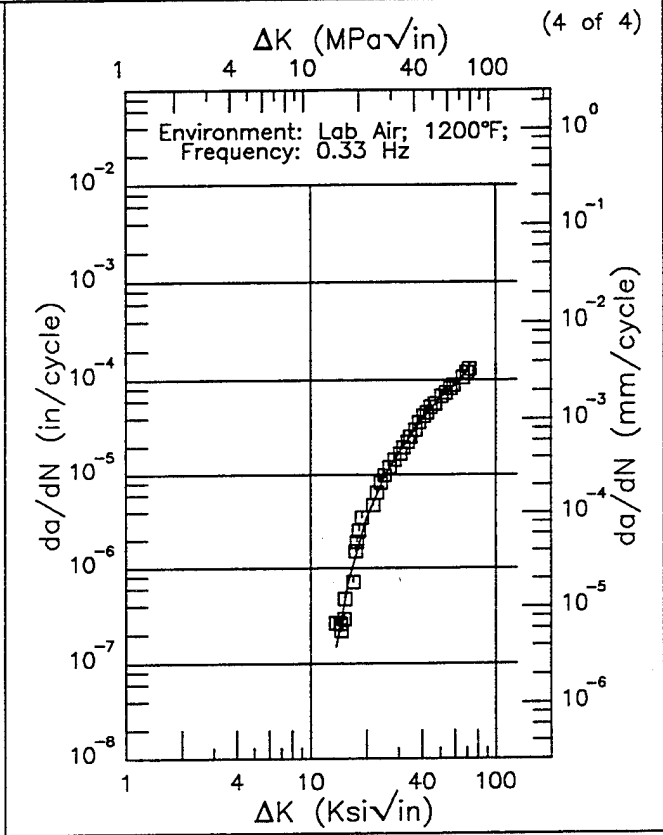
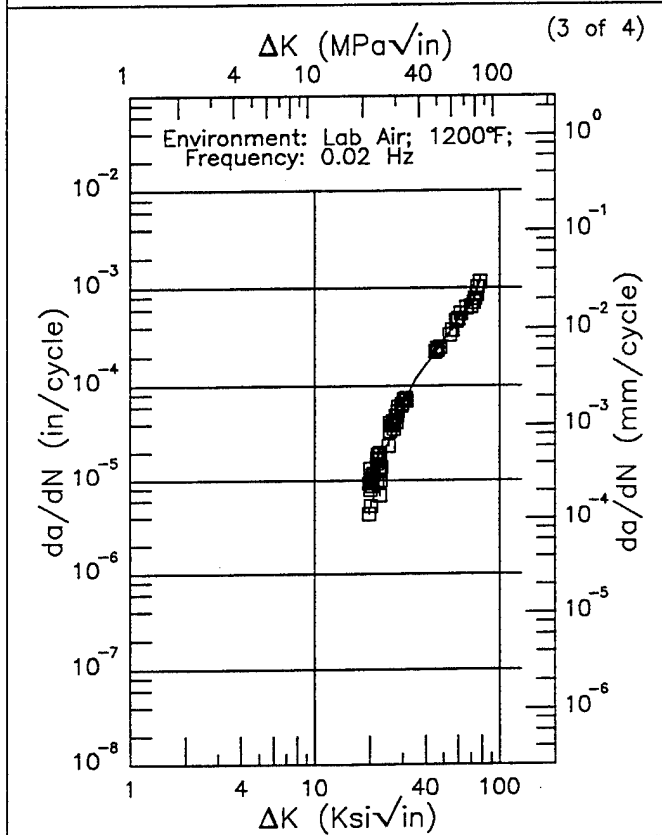
Figure 5.9.3.1.28

INCONEL 718

EF

Condition/Ht: 1760F 1HR WQ 1325F 8HRS
 Form: 3.6 in. Disk
 Specimen Type: CCP (max stress specified)
 Orientation: C-R
 Stress Ratio: 0.

Yield Strength: 171.6 ksi
 Ult. Strength: 201.7 ksi
 Specimen Thk: 0.08 in.
 Specimen Width: 2 in.
 Ref: GE008



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
19.60 (min)	8.25
20.	8.94
25.	25.6
30.	63.0
35.	116.
40.	167.
50.	287.
60.	489.
70.	679.
77.64 (max)	1132.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.68 (min)	0.148
16.	0.713
20.	3.42
25.	9.41
30.	17.0
35.	26.5
40.	38.6
50.	65.8
60.	89.3
70.	121.
71.33 (max)	127.

RMS % Error	Life Prediction Ratio Summary
20.58	0. .5 .8 1.25 2.

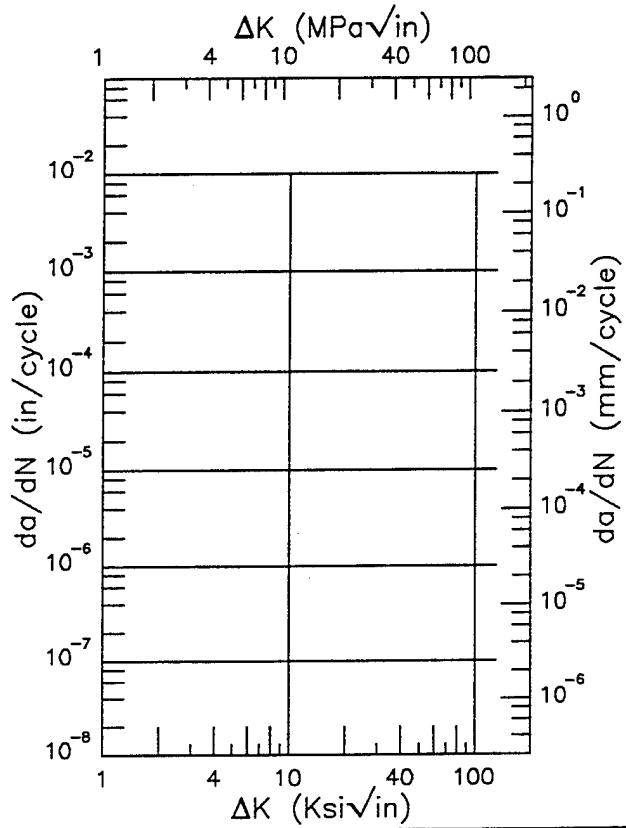
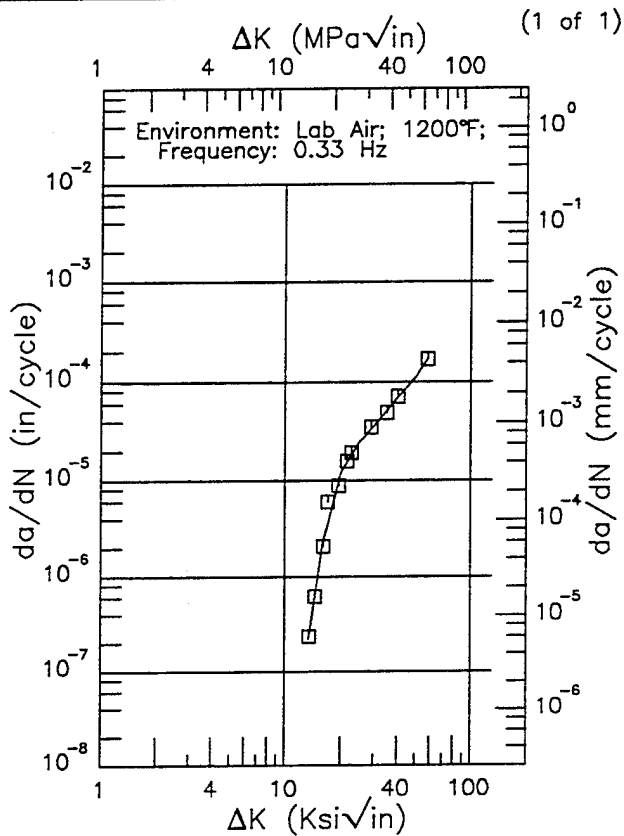
RMS % Error	Life Prediction Ratio Summary
20.50	0. .5 .8 1.25 2.

Figure 5.9.3.1.28 (Concluded)

EF INCONEL 718

Condition/Ht: 1760F 1HR WQ 1325F 8HRS
 Form: 3.6 in. Disk
 Specimen Type: CCP (max stress specified)
 Orientation: C-R
 Stress Ratio: 0.

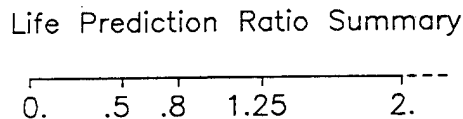
Yield Strength: 171.6 ksi
 Ult. Strength: 201.7 ksi
 Specimen Thk: 0.25 in.
 Specimen Width: 0.6 in.
 Ref: GE008



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.48 (min)	0.223
16.	2.30
20.	12.0
25.	24.7
30.	35.7
35.	50.6
40.	68.7
50.	109.
58.25 (max)	166.

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS %
 Error
 20.72



RMS %
 Error

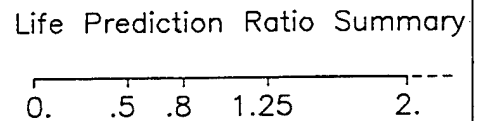


Figure 5.9.3.1.29

INCONEL 718

R

Condition/Ht: 1800F 1HR Q 1325F 8HRS FC TO 1150F HOLD 8HRS AC
 Form: 4.67 in. Forging
 Specimen Type: KB Bar
 Orientation: C-R
 Frequency: 0.3 Hz
 Environment: LAB AIR;1200°F

Yield Strength: 169.2 ksi
 Ult. Strength:
 Specimen Thk: 0.25 in.
 Specimen Width: 0.6 in.
 Ref: GE001

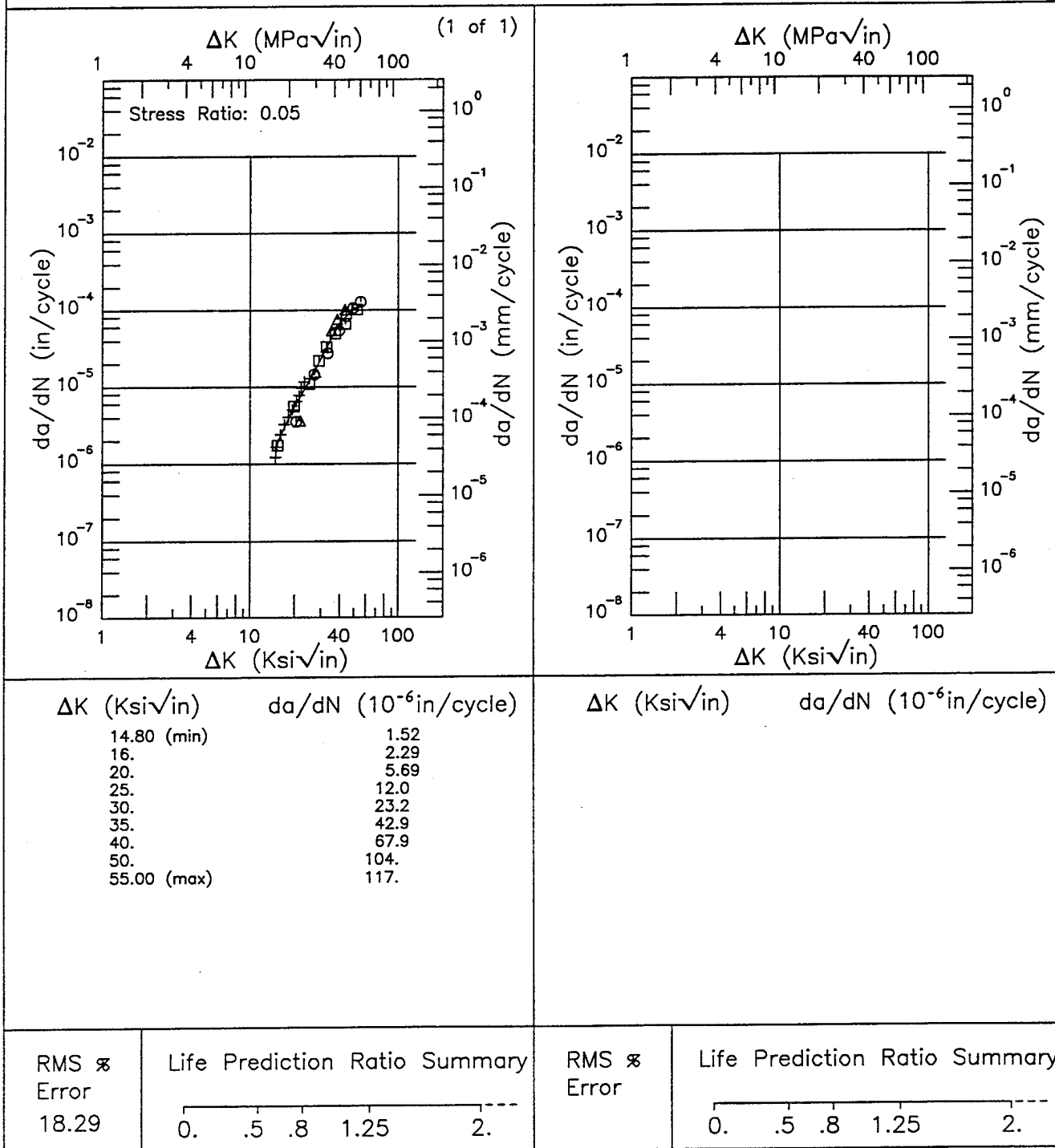
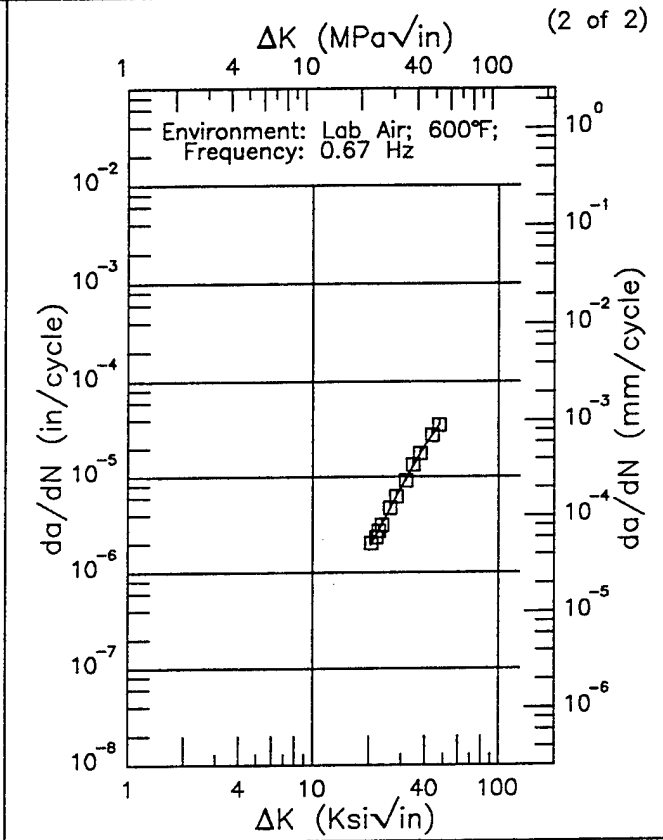
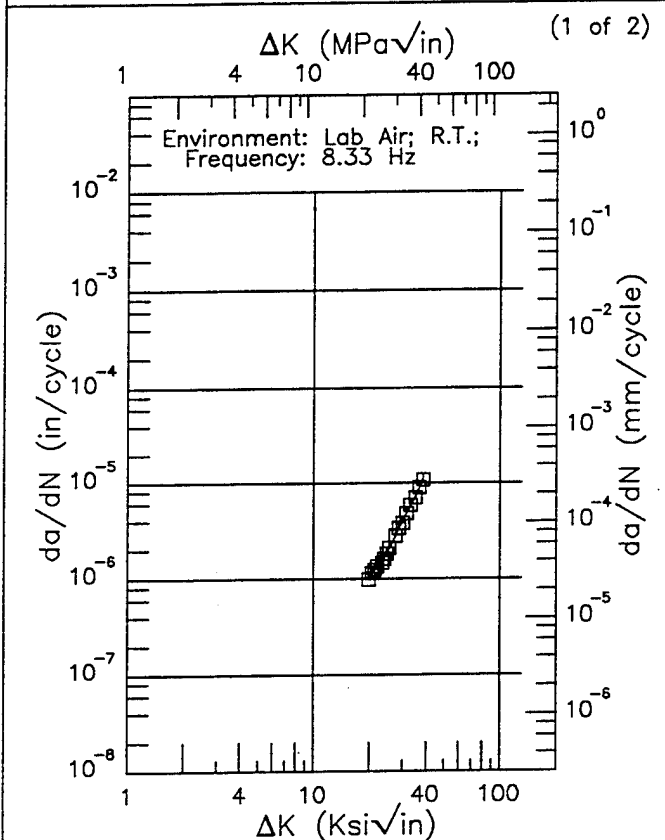


Figure 5.9.3.1.30

EF INCONEL 718

Condition/Ht: 1950F AC 1325F
 Form: 0.5 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.05

Yield Strength: 161.1 ksi
 Ult. Strength: 198.2 ksi
 Specimen Thk: 0.49 - 0.491 in.
 Specimen Width: 1.992 - 1.997 in.
 Ref: HD015



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
19.66 (min)	0.998
20.	1.05
25.	1.93
30.	4.02
35.	7.24
38.40 (max)	10.8

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
20.46 (min)	1.92
25.	3.93
30.	7.66
35.	13.5
40.	21.1
47.52 (max)	34.8

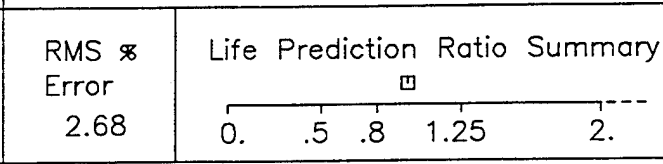
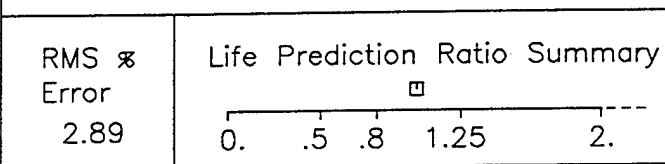


Figure 5.9.3.1.31

INCONEL 718

EF

Condition/Ht: 1950F AC 1325F
 Form: 0.5 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.05

Yield Strength: 161.1 ksi
 Ult. Strength: 198.2 ksi
 Specimen Thk: 0.49 in.
 Specimen Width: 1.996 - 1.998 in.
 Ref: HD015

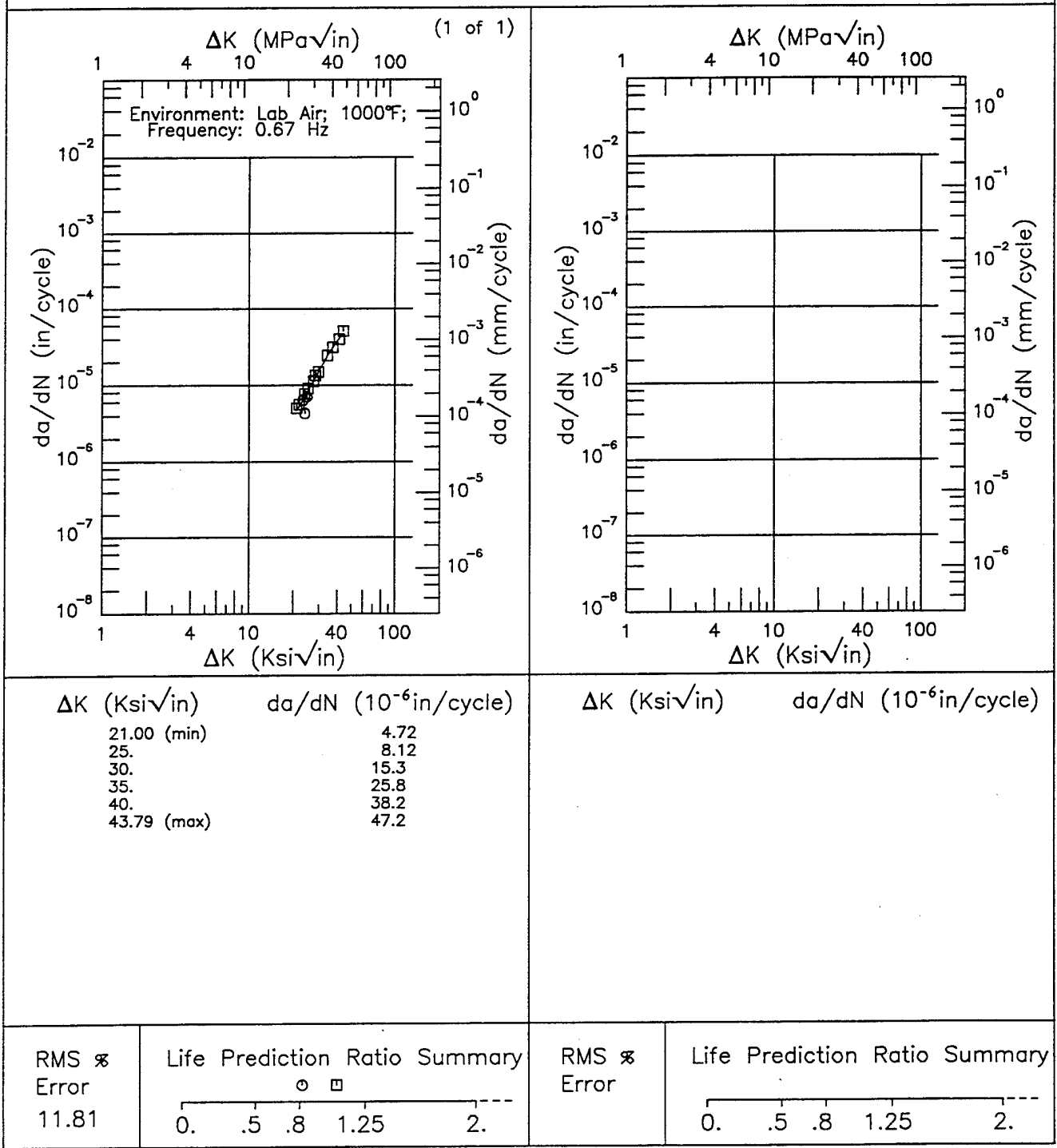


Figure 5.9.3.132

R | INCONEL 718 |

Condition/Ht: ANNEALED 1800F 1 HR, AC, 1325F 8 HR, COOLED AT 100F PER HR TO 1150F, 1150F 8HR, AC

Form: 0.75 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Frequency: 20 Hz
 Environment: LAB AIR; RT

Yield Strength: 180 ksi
 Ult. Strength: 218 ksi
 Specimen Thk: 0.187 in.
 Specimen Width: 1.5 in.
 Ref: MA016

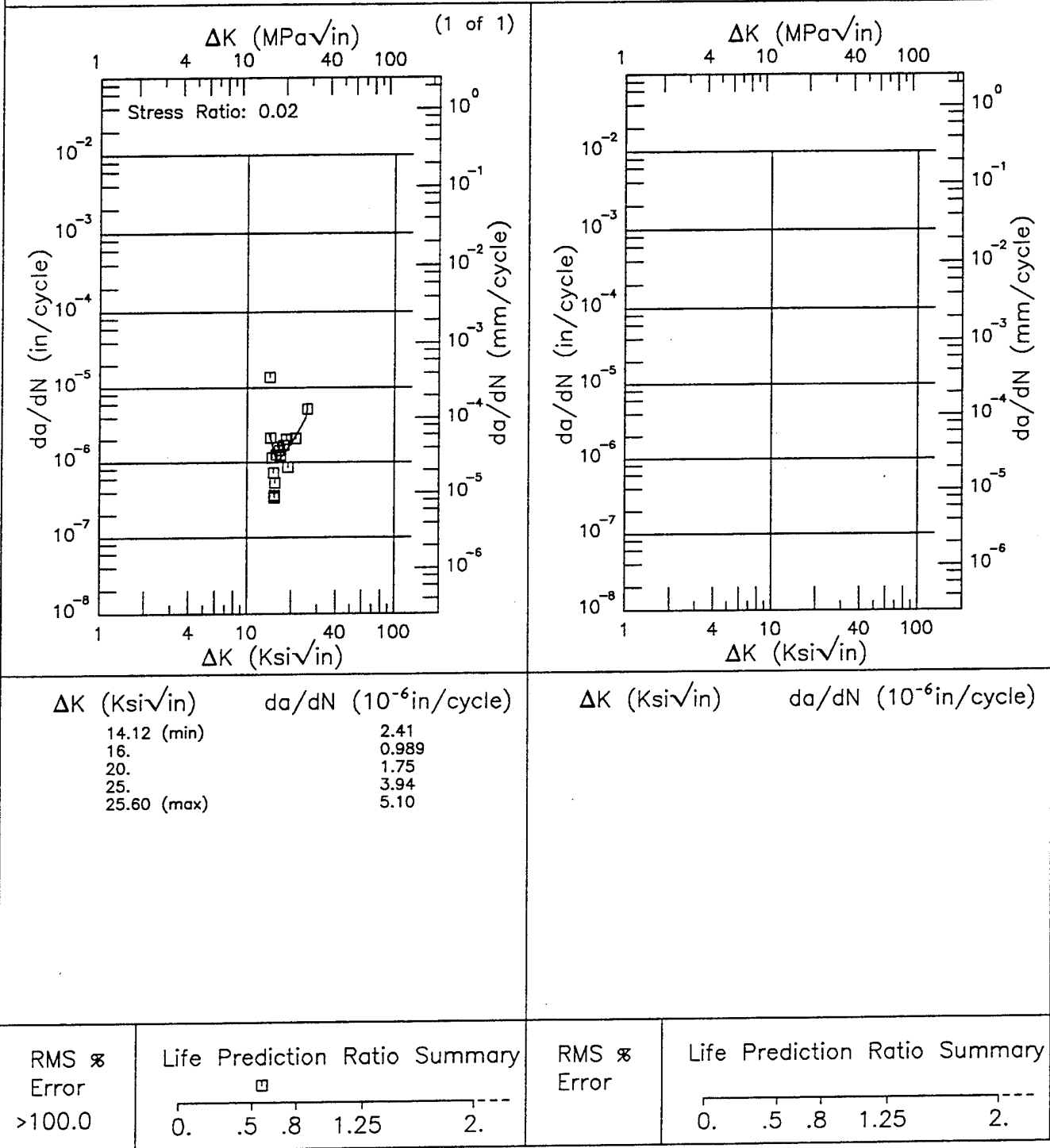


Figure 5.9.3.1.33

INCONEL 718

R

Condition/Ht: ANNEALED 1800F 1 HR, AC, 1325F 8 HR, COOLED AT 100F PER HR TO 1150F, 1150F 8HR, AC

Form: 0.75 in. Plate

Specimen Type: CT

Orientation: L-T

Frequency: 10 Hz

Environment: VACUUM; RT

Yield Strength: 180 ksi

Ult. Strength: 218 ksi

Specimen Thk: 0.187 in.

Specimen Width: 1.5 in.

Ref: MA016

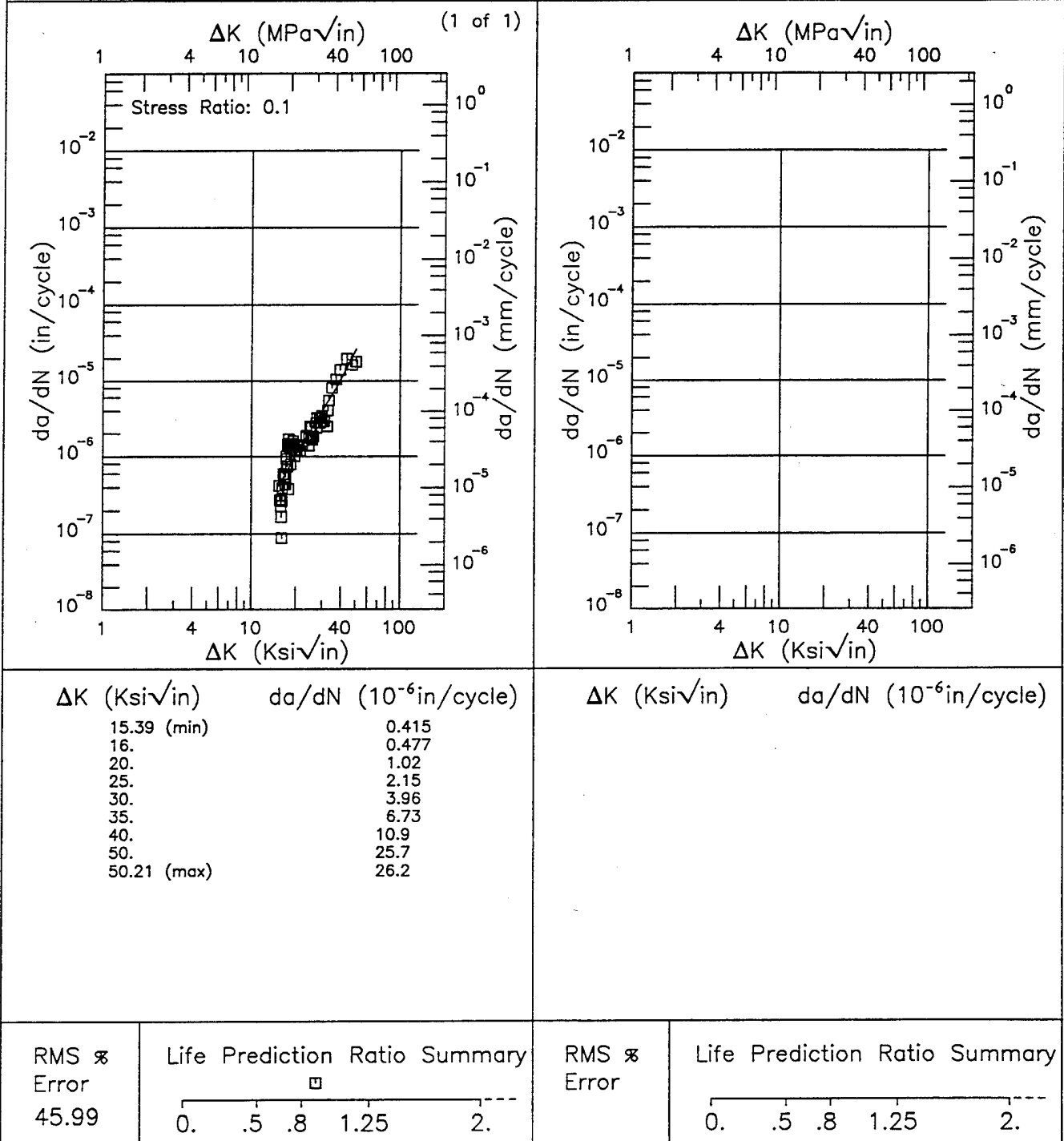


Figure 5.9.3.1.34

R INCONEL 718

Condition/Ht: ANNEALED 1800F 1 HR, AC, 1325F 8 HR, COOLED AT 100F PER HR TO 1150F, 1150F 8HR, AC

Form: 0.75 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 0. Hz
 Environment: LAB AIR;1200°F

Yield Strength: 155 ksi
 Ult. Strength: 178 ksi
 Specimen Thk: 0.187 in.
 Specimen Width: 3 in.
 Ref: MA016

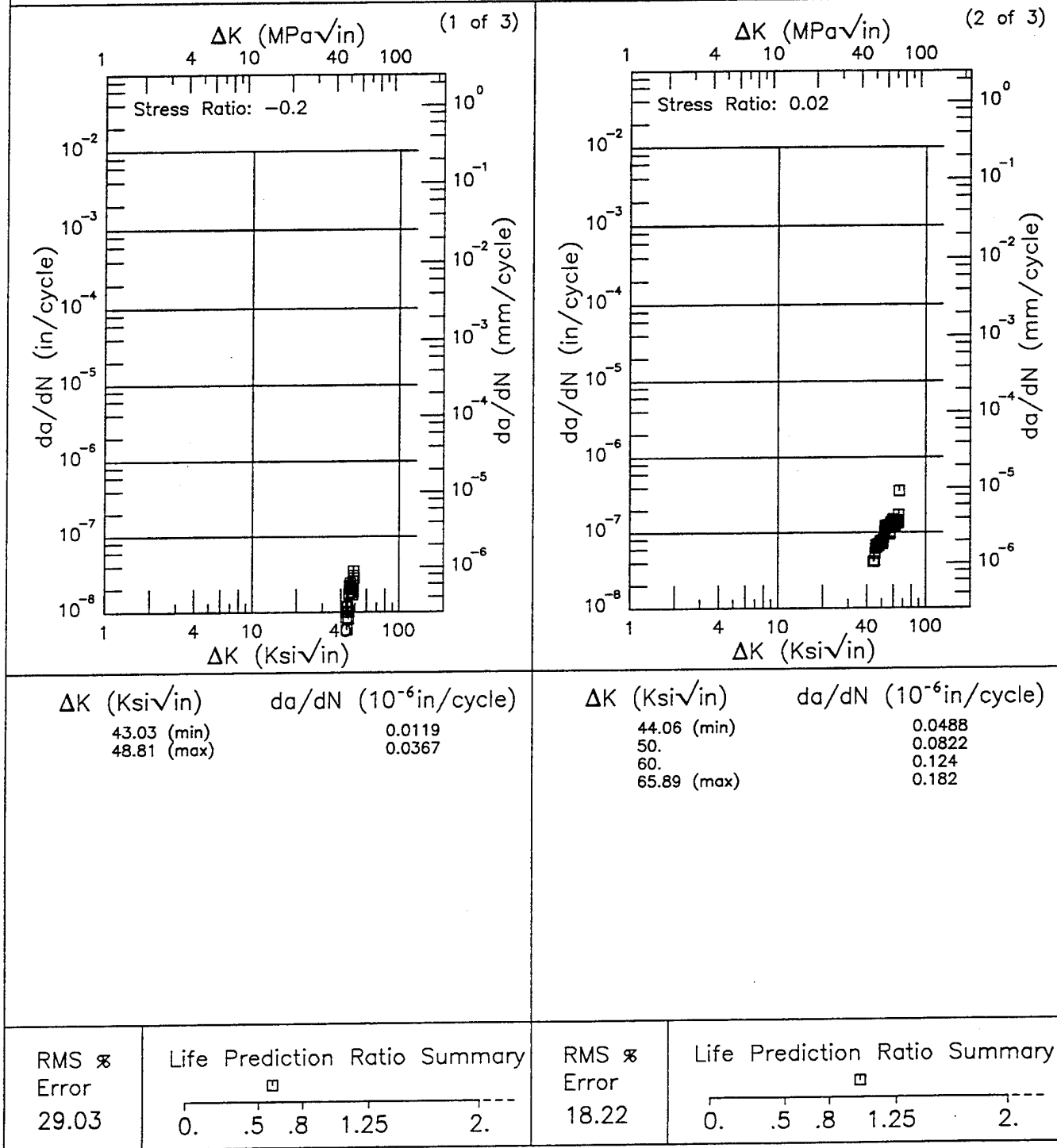


Figure 5.9.3.1.35

INCONEL 718

Condition/Ht: ANNEALED 1800F 1 HR, AC, 1325F 8 HR, COOLED AT 100F PER HR TO 1150F, 1150F 8HR, AC

Form: 0.75 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 0. Hz
 Environment: LAB AIR;1200°F

Yield Strength: 155 ksi
 Ult. Strength: 178 ksi
 Specimen Thk: 0.187 in.
 Specimen Width: 3 in.
 Ref: MA016

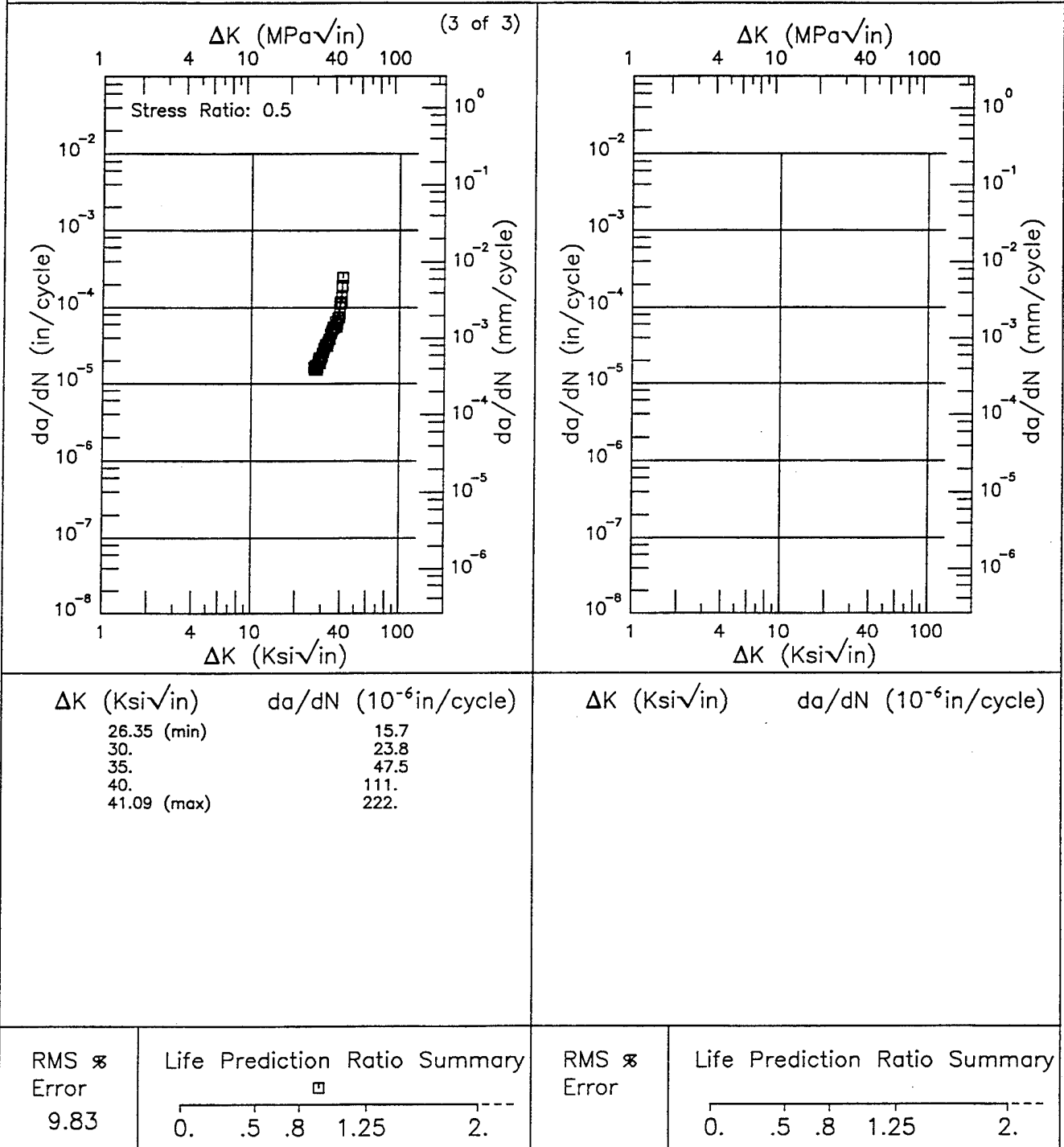


Figure 5.9.3.1.35 (Concluded)

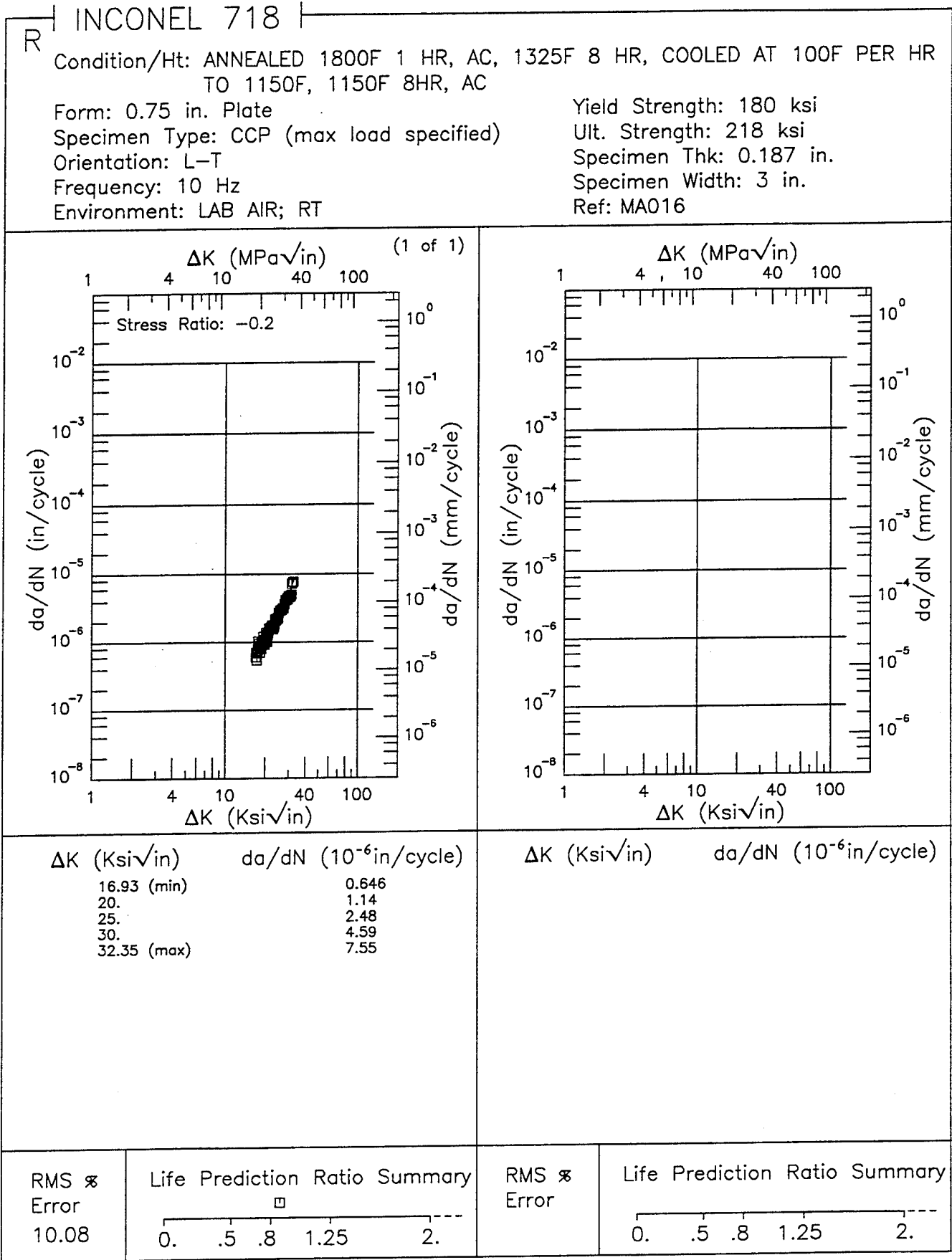


Figure 5.9.3.1.36

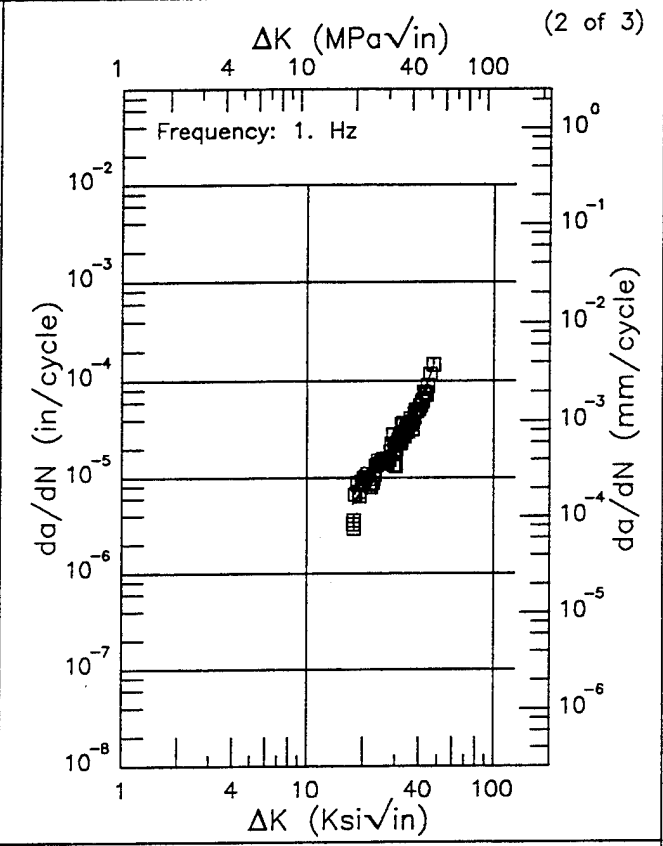
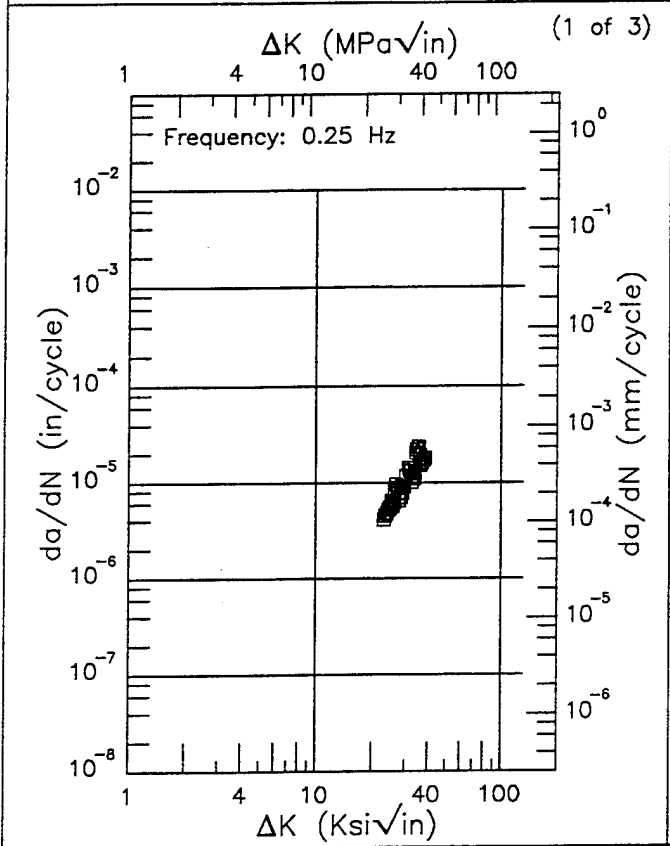
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F | INCONEL 718 |

Condition/Ht: ANNEALED 1800F 1 HR, AC, 1325F 8 HR, COOLED AT 100F PER HR TO 1150F, 1150F 8HR, AC

Form: 0.75 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.02
 Environment: LAB AIR;1200°F

Yield Strength: 155 ksi
 Ult. Strength: 178 ksi
 Specimen Thk: 0.187 in.
 Specimen Width: 3 in.
 Ref: MA016



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
23.08 (min)	3.97
25.	5.88
30.	9.69
35.	17.4
38.50 (max)	16.6

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
17.64 (min)	5.25
20.	8.09
25.	14.3
30.	21.7
35.	33.3
40.	54.9
47.54 (max)	140.

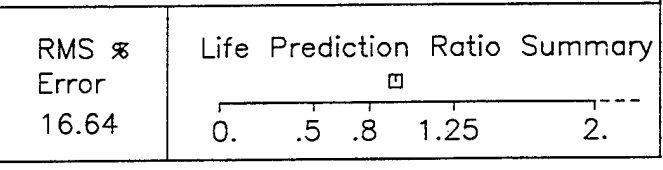
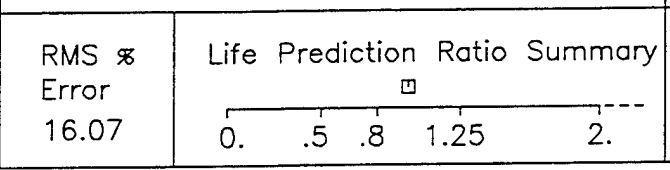


Figure 5.9.3.1.37

INCONEL 718

Condition/Ht: ANNEALED 1800F 1 HR, AC, 1325F 8 HR, COOLED AT 100F PER HR TO 1150F, 1150F 8HR, AC

Form: 0.75 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.02
 Environment: LAB AIR;1200°F

Yield Strength: 155 ksi
 Ult. Strength: 178 ksi
 Specimen Thk: 0.187 in.
 Specimen Width: 3 in.
 Ref: MA016

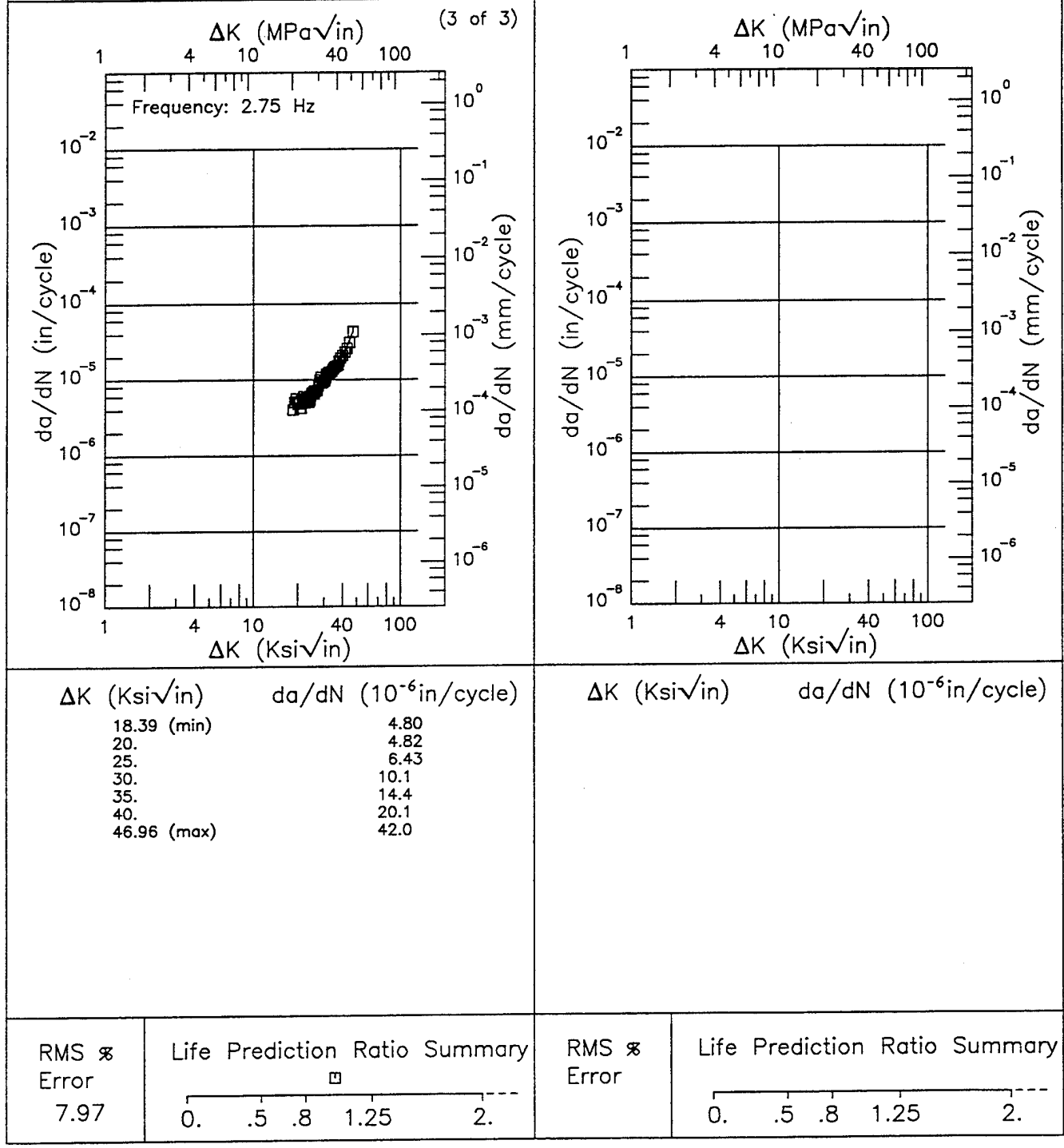


Figure 5.9.3.137 (Concluded)

EF INCONEL 718

Condition/Ht: ANNEALED 1800F 1 HR, AC, 1325F 8 HR, COOLED AT 100F PER HR TO 1150F, 1150F 8HR, AC

Form: 0.75 in. Plate

Specimen Type: CCP (max load specified)

Orientation: L-T

Stress Ratio: 0.02

Yield Strength: 167 - 180 ksi

Ult. Strength: 196 - 218 ksi

Specimen Thk: 0.187 in.

Specimen Width: 3 in.

Ref: MA016

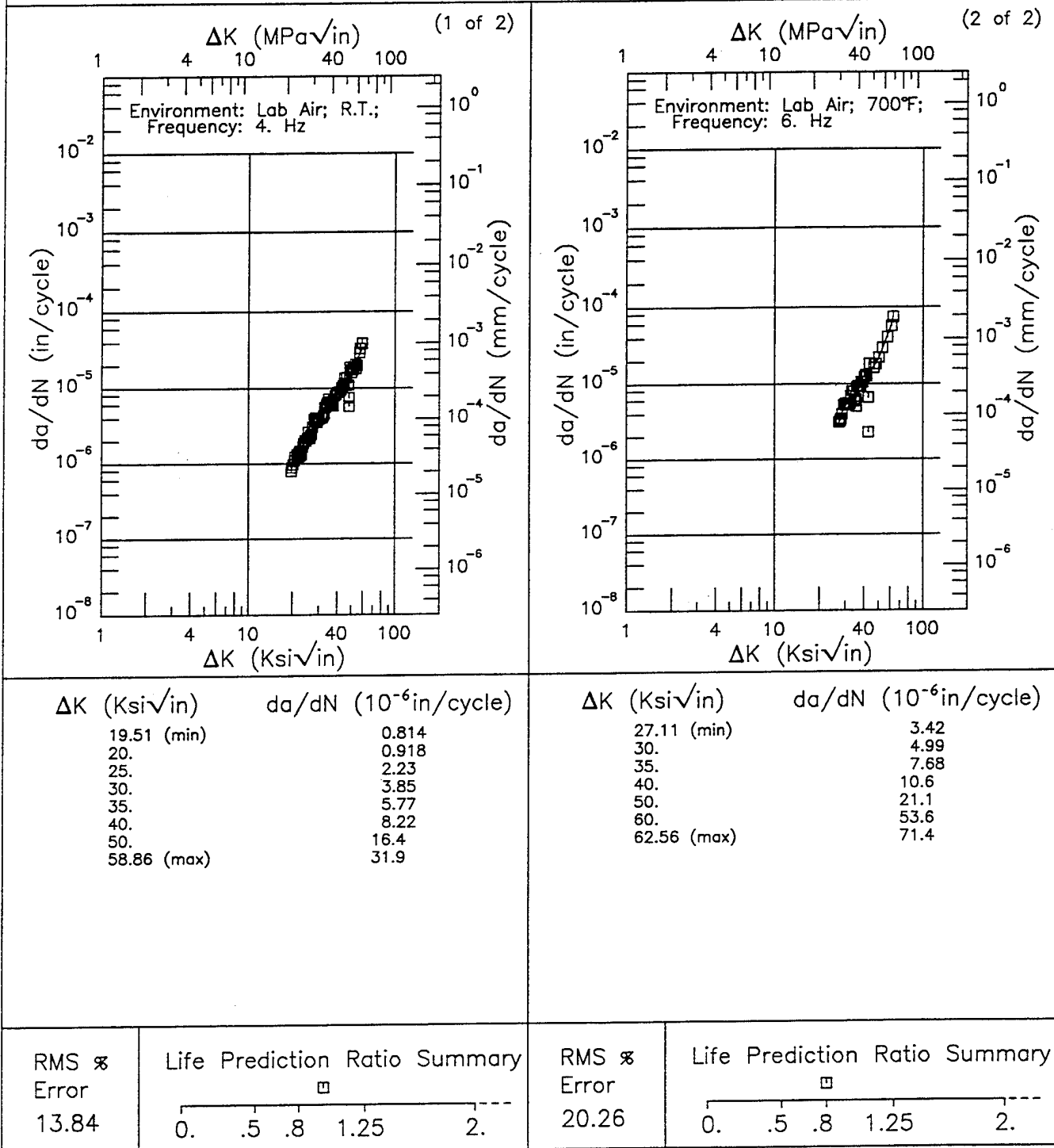


Figure 5.9.3.1.38

INCONEL 718

EF

Condition/Ht: ANNEALED 1800F 1 HR, AC, 1325F 8 HR, COOLED AT 100F PER HR TO 1150F, 1150F 8HR, AC

Form: 0.75 in. Plate

Specimen Type: CCP (max load specified)

Orientation: L-T

Stress Ratio: 0.5

Yield Strength: 155 - 180 ksi

Ult. Strength: 178 - 218 ksi

Specimen Thk: 0.187 in.

Specimen Width: 3 in.

Ref: MA016

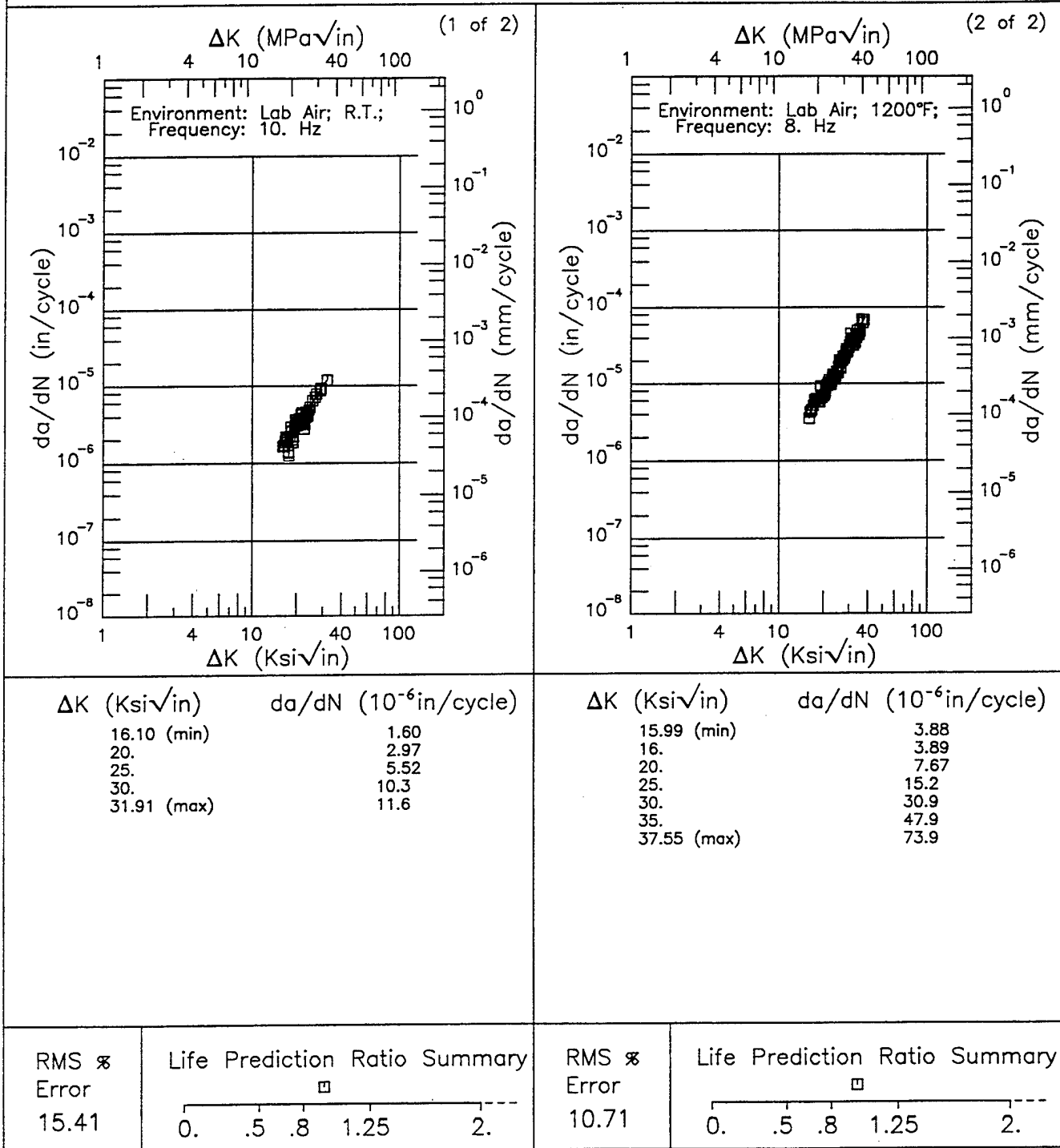


Figure 5.9.3.1.39

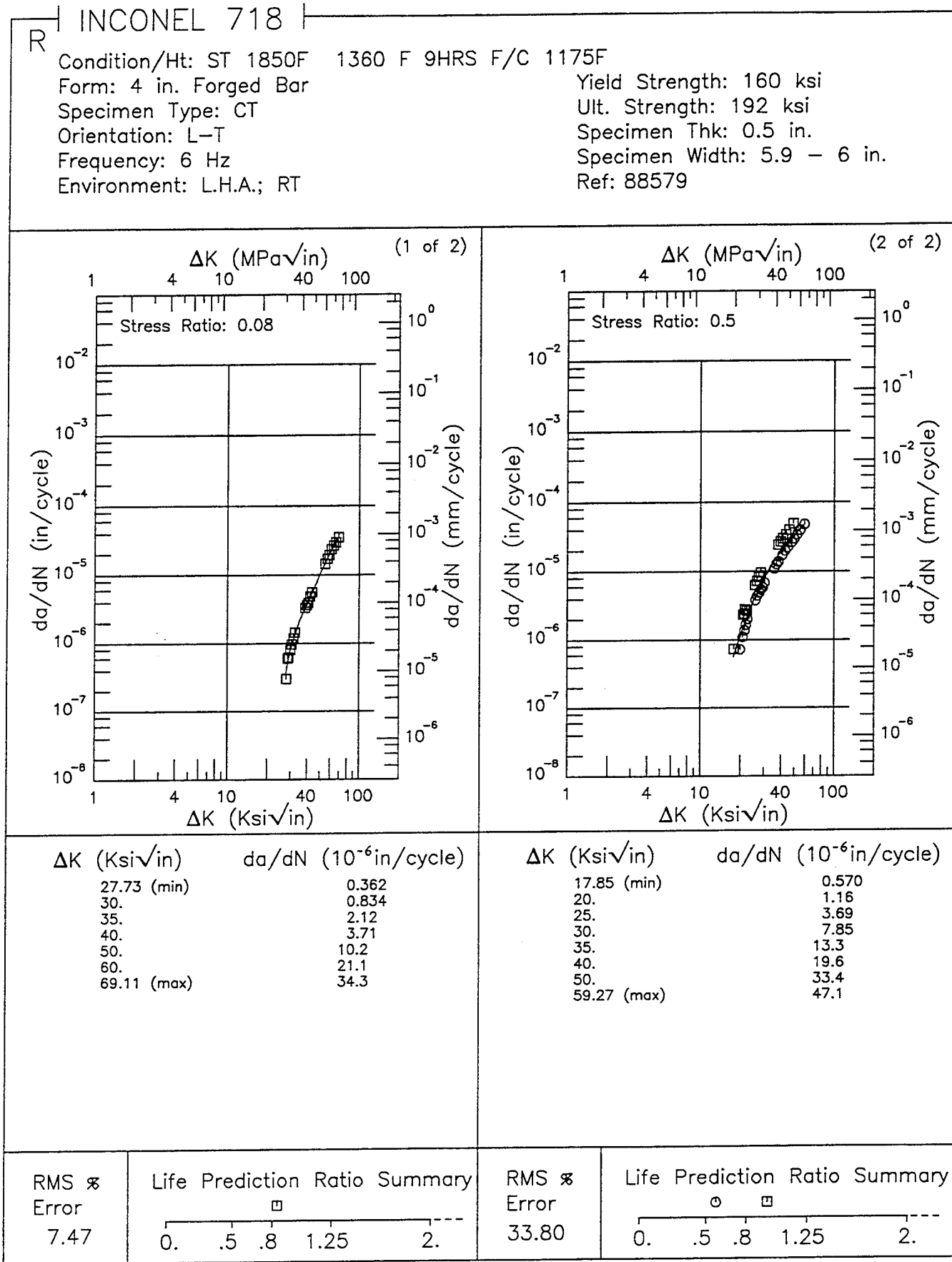


Figure 5.9.3.1.40

INCONEL 718

EF

Condition/Ht: ST 1850F 1360 F 9HRS F/C 1175F

Form: 4 in. Forged Bar

Specimen Type: CT

Orientation: L-T

Stress Ratio: 0.08

Yield Strength: 160 ksi

Ult. Strength: 192 ksi

Specimen Thk: 0.5 in.

Specimen Width: 5.99 - 6 in.

Ref: 88579

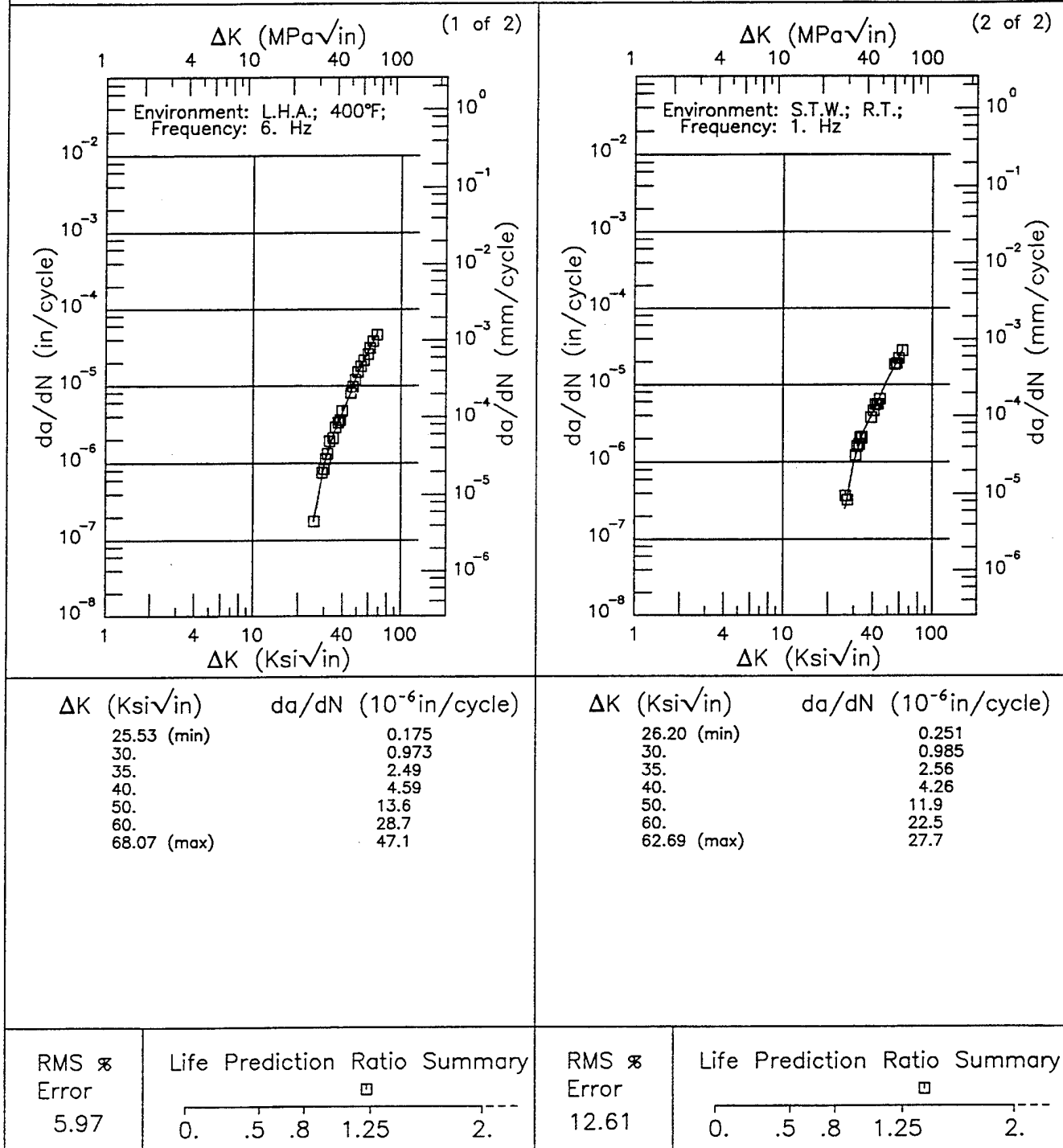


Figure 5.9.3.1.41

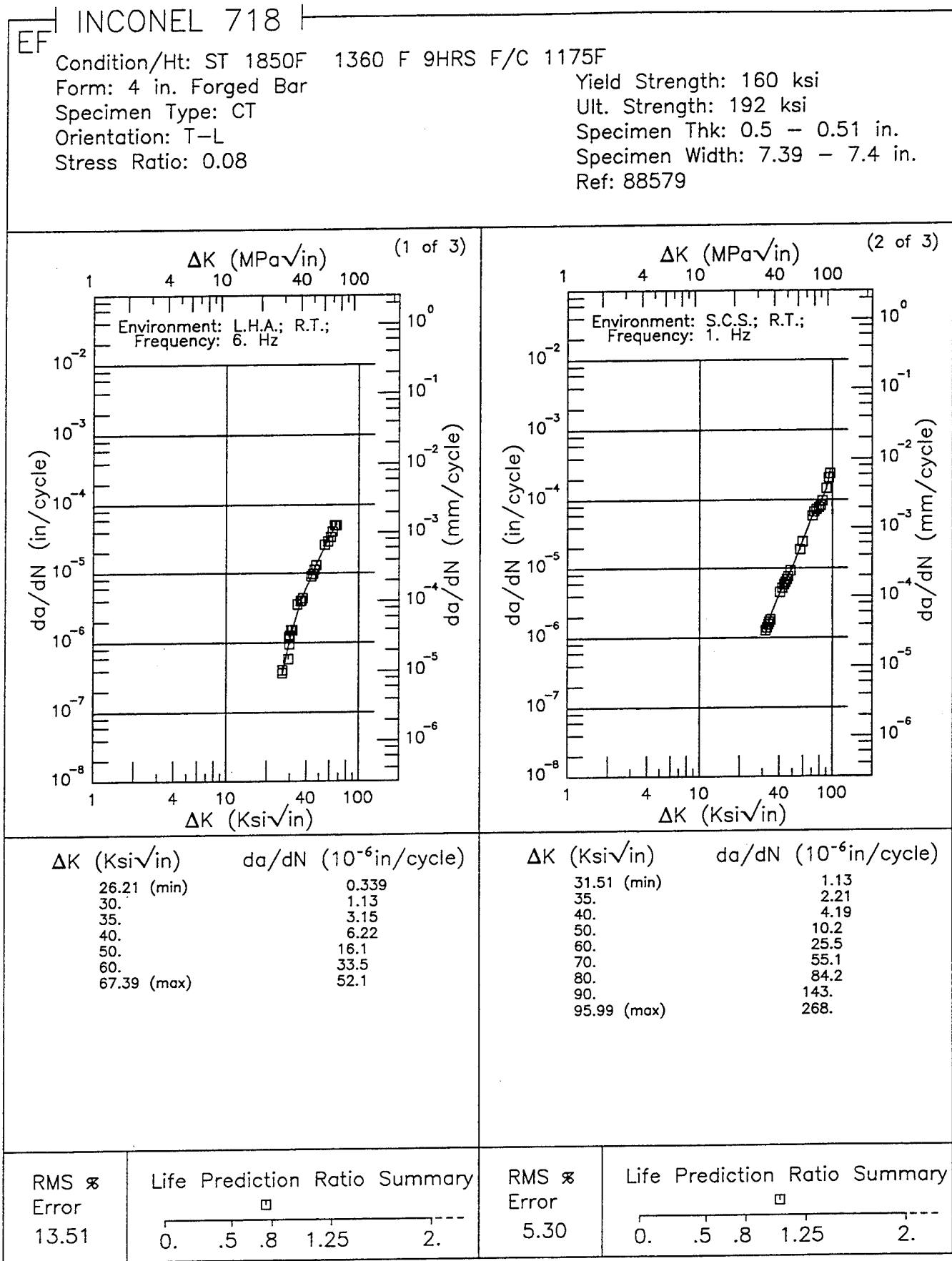


Figure 5.9.3.1.42

INCONEL 718

EF

Condition/Ht: ST 1850F 1360 F 9HRS F/C 1175F
 Form: 4 in. Forged Bar
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.08

Yield Strength: 160 ksi
 Ult. Strength: 192 ksi
 Specimen Thk: 0.5 - 0.51 in.
 Specimen Width: 7.39 - 7.4 in.
 Ref: 88579

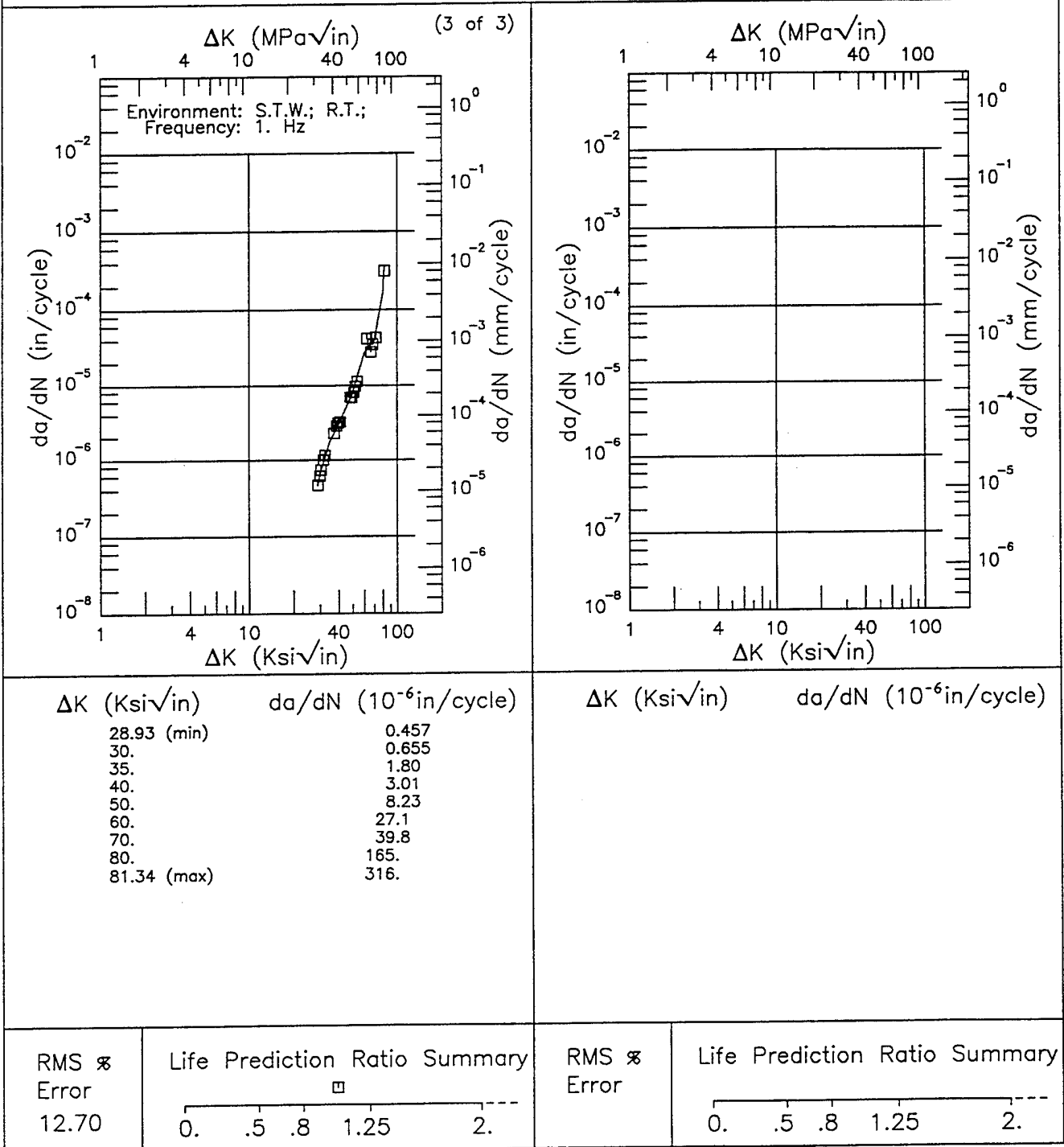
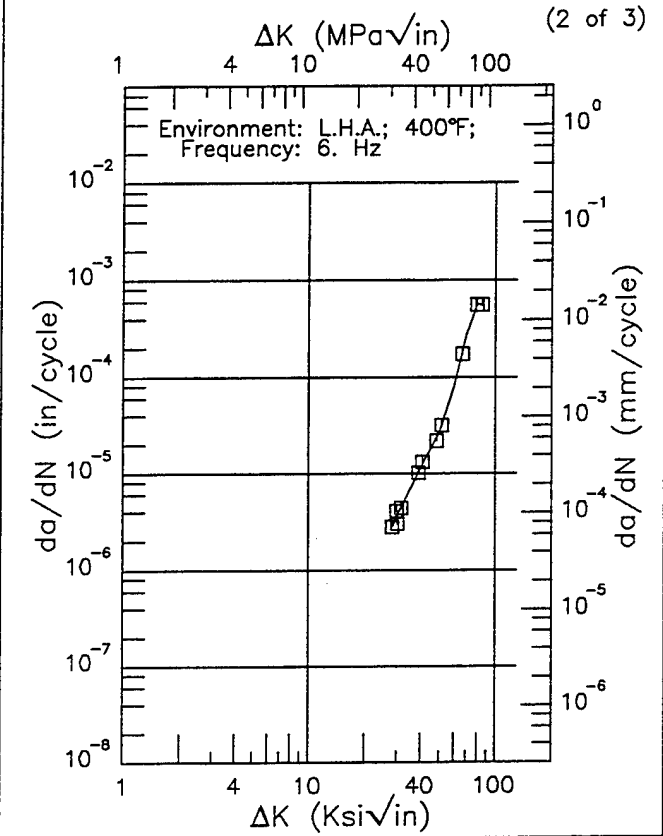
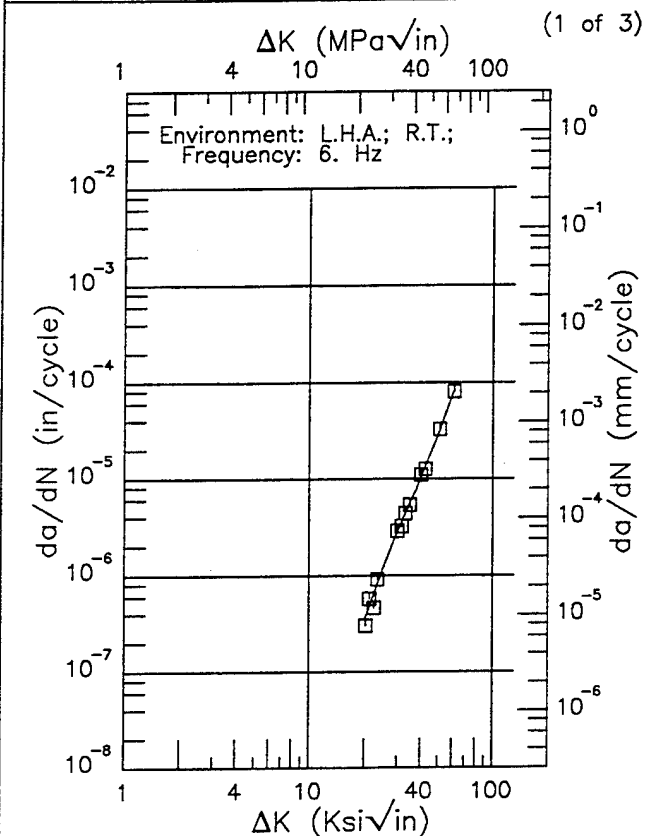


Figure 5.9.3.1.42 (Concluded)

EF | INCONEL 718 |

Condition/Ht: ST 1850F 1360 F 9HRS F/C 1175F
 Form: 4 in. Forged Bar
 Specimen Type: CT
 Orientation: S-L
 Stress Ratio: 0.08

Yield Strength: 160 ksi
 Ult. Strength: 192 ksi
 Specimen Thk: 1 in.
 Specimen Width: 3.77 in.
 Ref: 88579



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
20.07 (min)	0.356
25.	1.18
30.	2.82
35.	5.62
40.	10.1
50.	28.6
60.	74.5
61.11 (max)	82.7

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
28.21 (min)	2.81
30.	3.58
35.	6.71
40.	11.4
50.	26.9
60.	80.1
70.	273.
80.	565.
84.76 (max)	550.

RMS % Error	Life Prediction Ratio Summary
12.82	

RMS % Error	Life Prediction Ratio Summary
7.63	

Figure 5.9.3.1.43

INCONEL 718 EF

Condition/Ht: ST 1850F 1360 F 9HRS F/C 1175F
 Form: 4 in. Forged Bar
 Specimen Type: CT
 Orientation: S-L
 Stress Ratio: 0.08

Yield Strength: 160 ksi
 Ult. Strength: 192 ksi
 Specimen Thk: 1 in.
 Specimen Width: 3.77 in.
 Ref: 88579

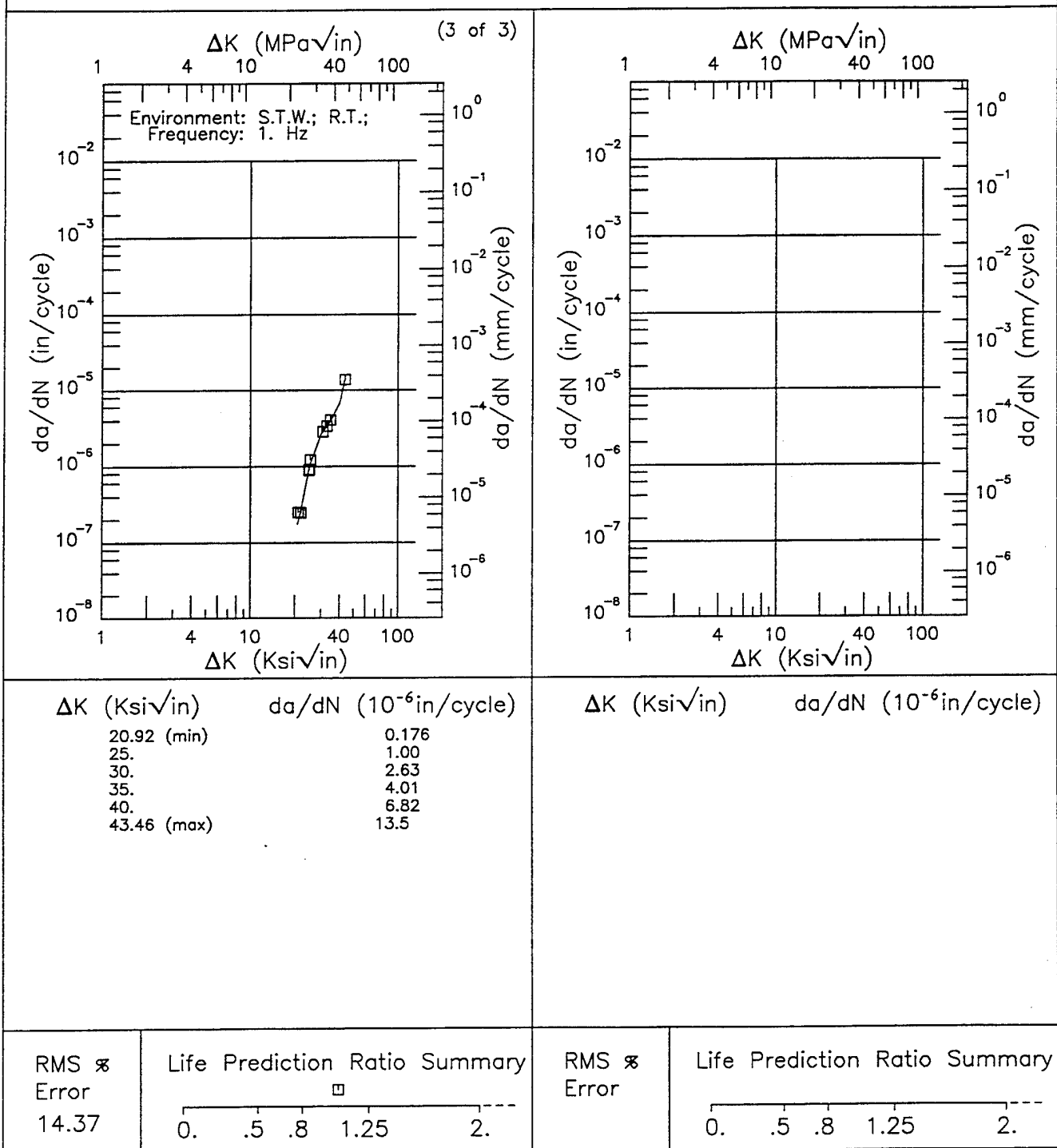
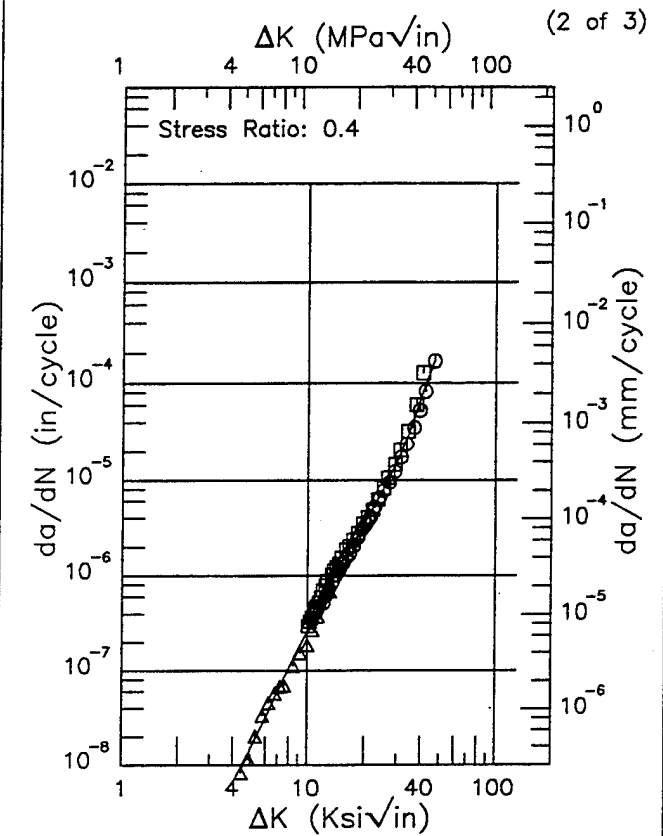
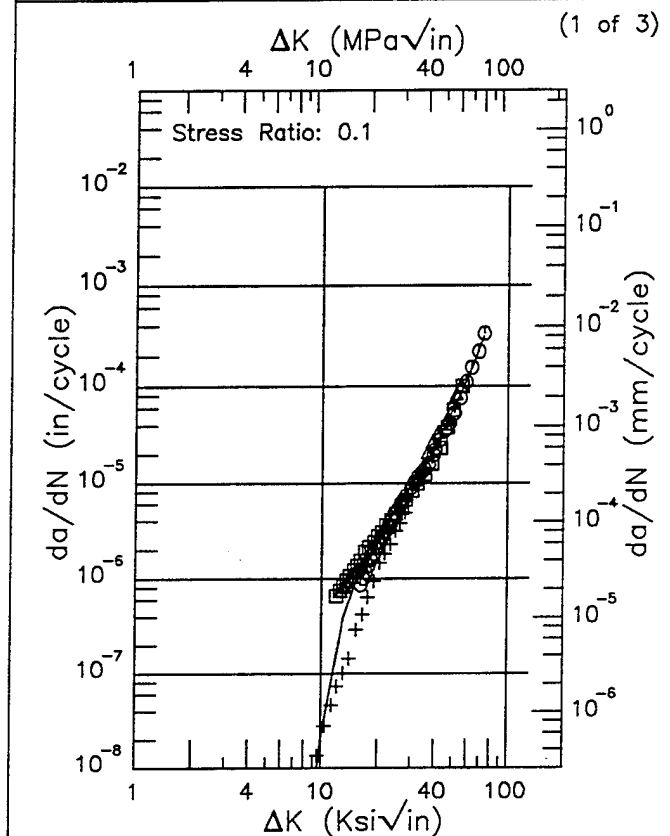


Figure 5.9.3.1.43 (Concluded)

R | INCONEL 718 |

Condition/Ht: STA
 Form: 3 in. Forging
 Specimen Type: CT
 Orientation: L-T
 Frequency: 15 - 30 Hz
 Environment: LAB AIR; RT

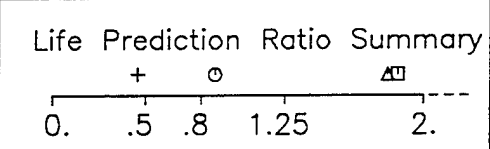
Yield Strength: 161.2 - 177.4 ksi
 Ult. Strength:
 Specimen Thk: 0.244 - 0.254 in.
 Specimen Width: 2.002 - 2.005 in.
 Ref: DA006;DA007



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
9.50 (min)	0.00991
10.	0.0218
13.	0.381
16.	1.24
20.	2.42
25.	4.11
30.	7.59
35.	14.2
40.	24.5
50.	56.4
60.	121.
70.	252.
73.51 (max)	313.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
4.45 (min)	0.00926
5.	0.0149
6.	0.0318
7.	0.0605
8.	0.105
9.	0.171
10.	0.263
13.	0.739
16.	1.60
20.	3.51
25.	7.75
30.	15.7
35.	32.5
40.	71.5
46.73 (max)	163.

RMS % Error
55.89



RMS % Error
14.60

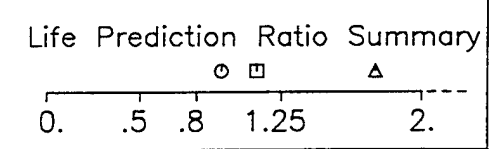


Figure 5.9.3.1.44

INCONEL 718 R

Condition/Ht: STA
 Form: 3 in. Forging
 Specimen Type: CT
 Orientation: L-T
 Frequency: 15 - 30 Hz
 Environment: LAB AIR; RT

Yield Strength: 161.2 - 177.4 ksi
 Ult. Strength:
 Specimen Thk: 0.244 - 0.254 in.
 Specimen Width: 2.002 - 2.005 in.
 Ref: DA006;DA007

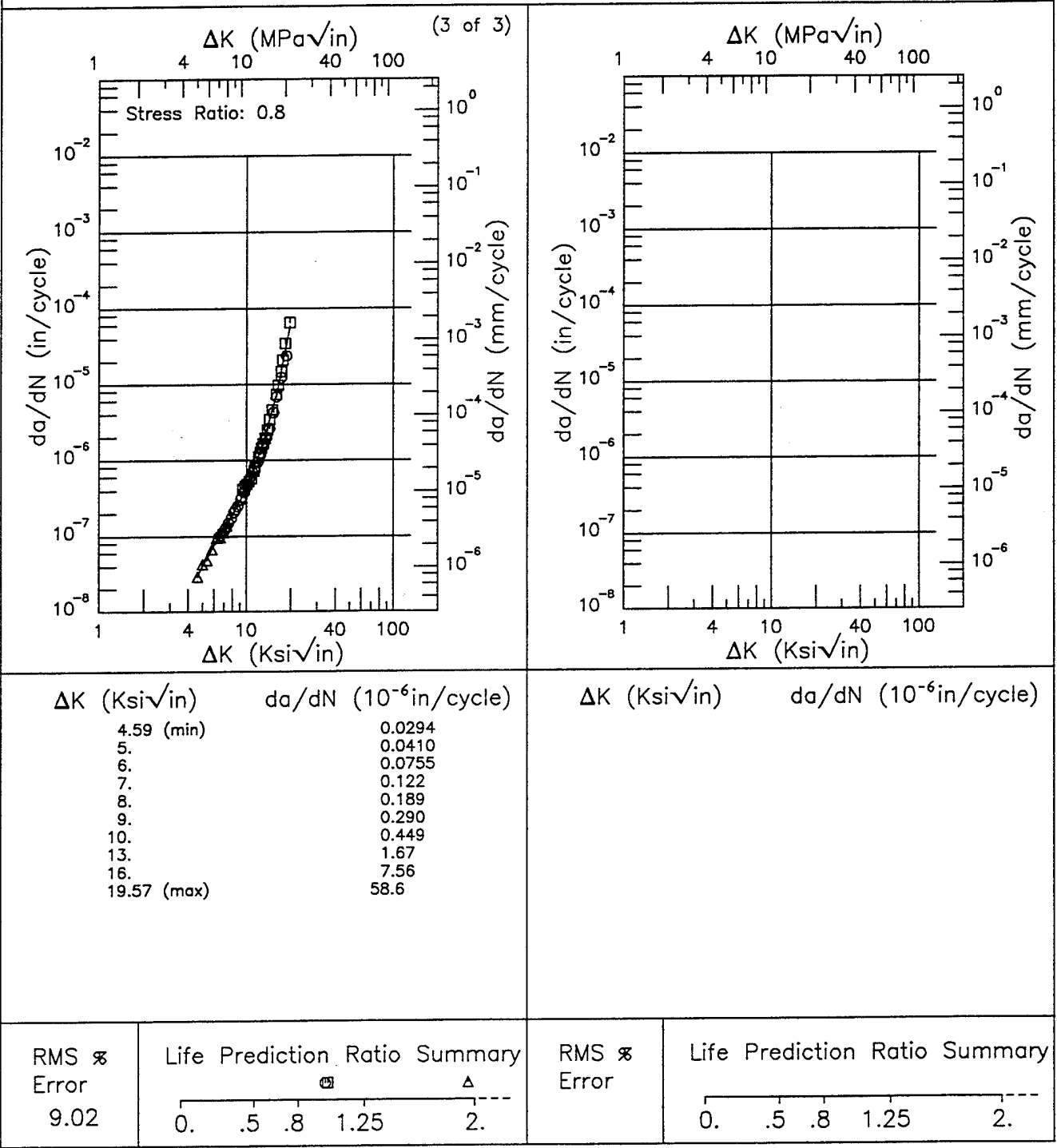


Figure 5.9.3.144 (Concluded)

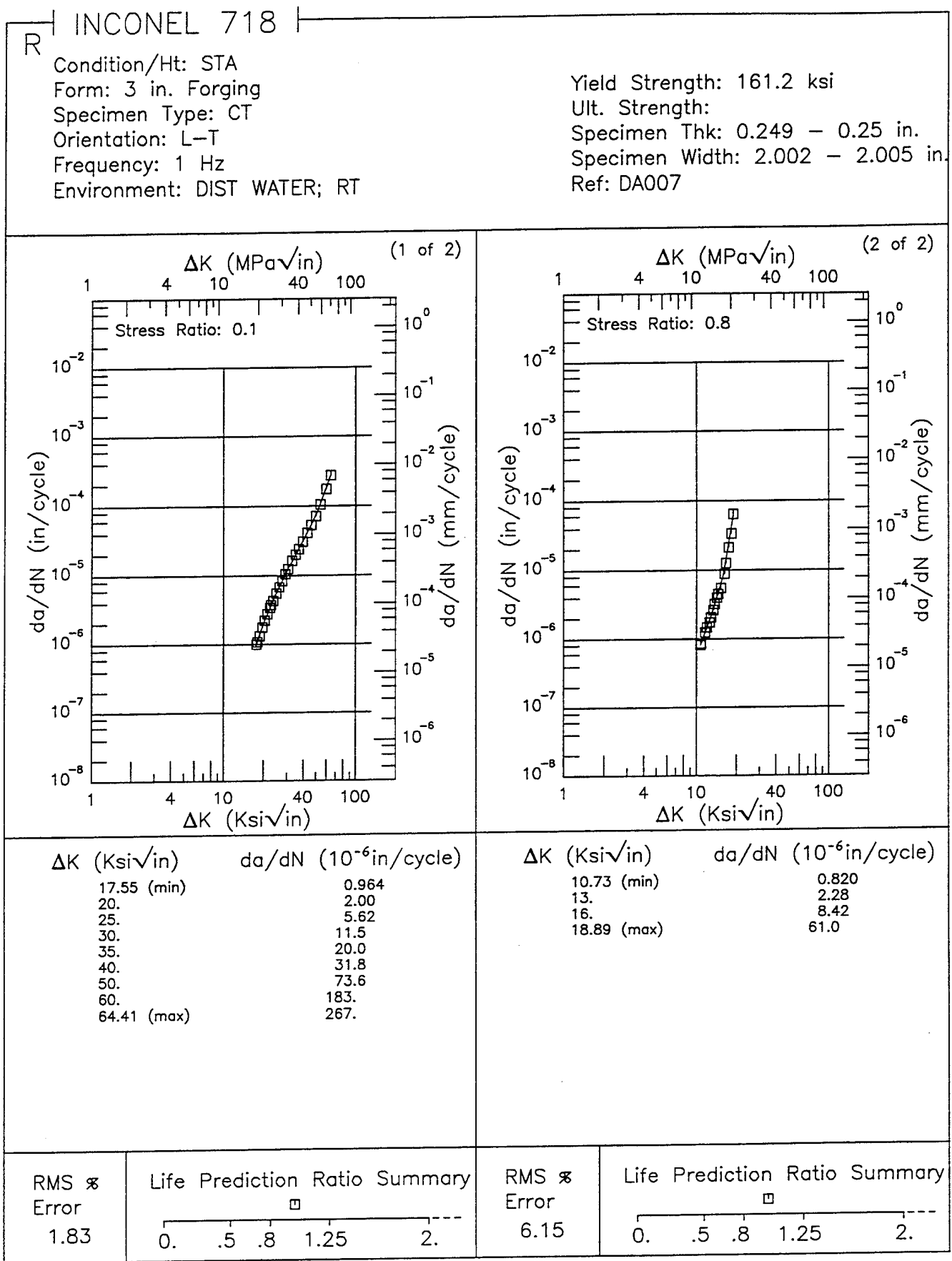


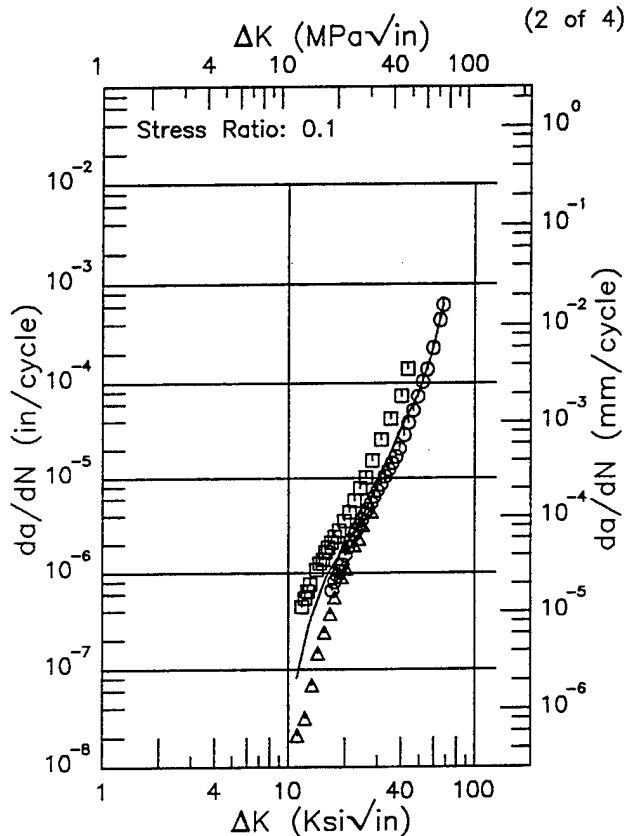
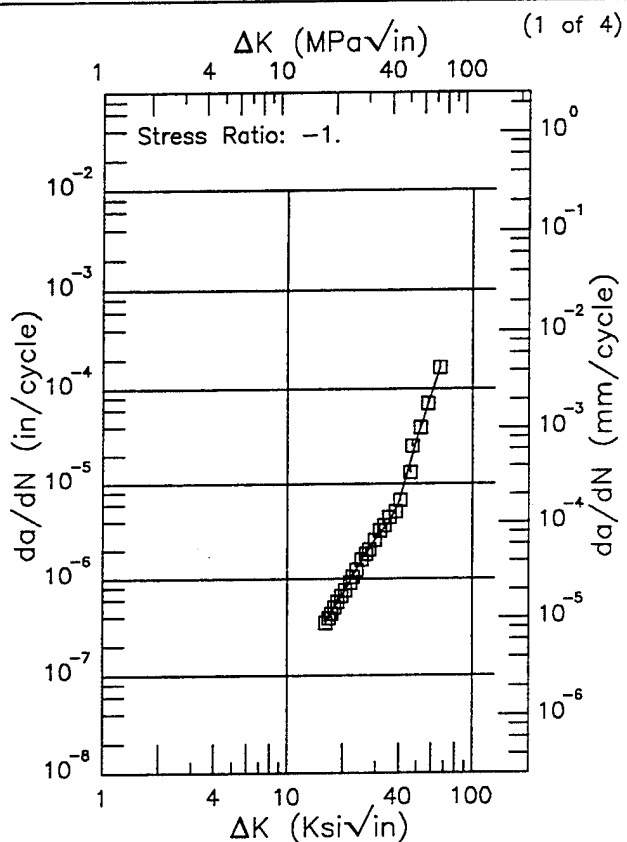
Figure 5.9.3.1.45

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R INCONEL 718

Condition/Ht: STA
 Form: 3 in. Forging
 Specimen Type: CT
 Orientation: T-L
 Frequency: 5 - 30 Hz
 Environment: LAB AIR; RT

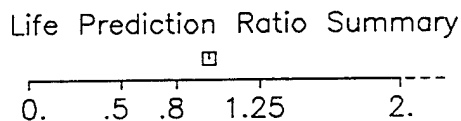
Yield Strength: 155.4 - 169. ksi
 Ult. Strength:
 Specimen Thk: 0.25 - 0.254 in.
 Specimen Width: 2.001 - 2.007 in.
 Ref: DA006;DA007



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
16.01 (min)	0.360
20.	0.686
25.	1.54
30.	2.64
35.	3.87
40.	6.47
50.	29.8
60.	86.9
66.61 (max)	165.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
11.12 (min)	0.0823
13.	0.304
16.	0.919
20.	1.94
25.	3.94
30.	8.68
35.	18.2
40.	32.9
50.	82.1
60.	241.
67.33 (max)	613.

RMS %
 Error
 7.49



RMS %
 Error
 81.12

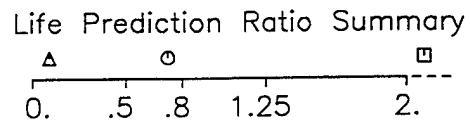


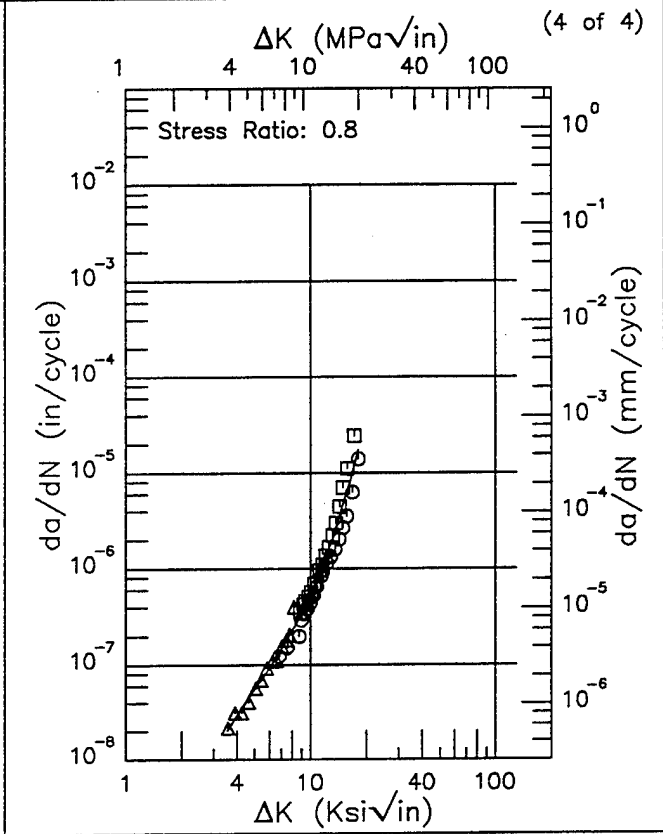
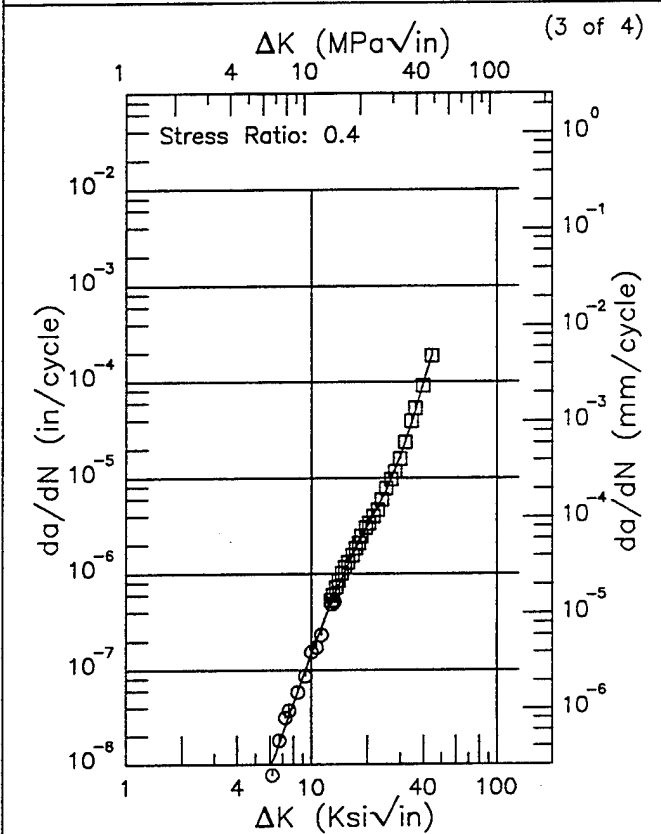
Figure 5.9.3.1.46

INCONEL 718

R

Condition/Ht: STA
 Form: 3 in. Forging
 Specimen Type: CT
 Orientation: T-L
 Frequency: 5 - 30 Hz
 Environment: LAB AIR; RT

Yield Strength: 155.4 - 169. ksi
 Ult. Strength:
 Specimen Thk: 0.25 - 0.254 in.
 Specimen Width: 2.001 - 2.007 in.
 Ref: DA006;DA007



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
6.15 (min)	0.0105
7.	0.0211
8.	0.0441
9.	0.0839
10.	0.148
13.	0.553
16.	1.37
20.	3.13
25.	7.28
30.	16.8
35.	40.9
40.	99.9
44.22 (max)	183.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
3.56 (min)	0.0201
4.	0.0293
5.	0.0560
6.	0.0930
7.	0.146
8.	0.224
9.	0.340
10.	0.518
13.	1.87
16.	7.01
18.01 (max)	17.3

RMS % Error	Life Prediction Ratio Summary
9.55	

RMS % Error	Life Prediction Ratio Summary
27.65	

Figure 5.9.3.1.46 (Concluded)

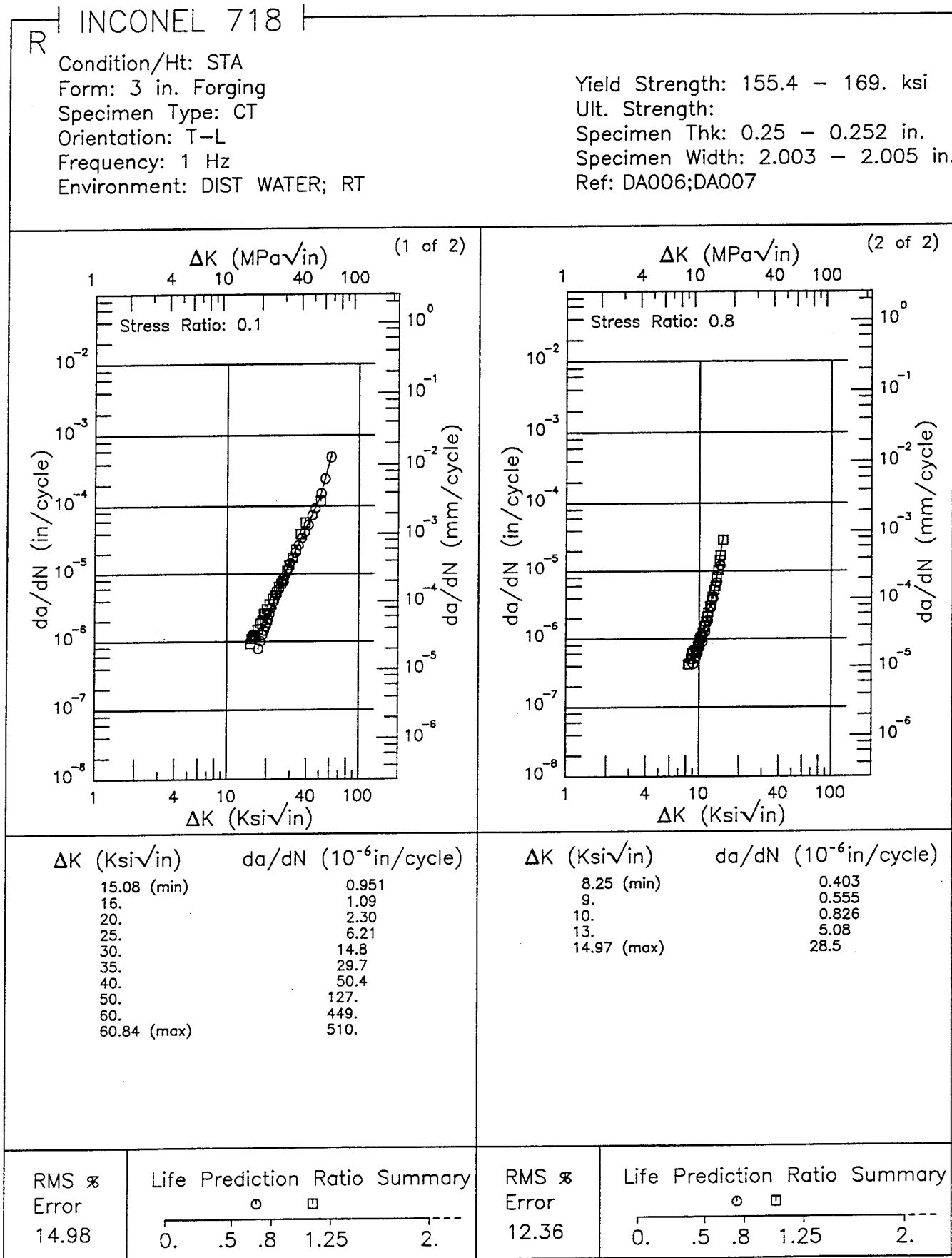


Figure 5.9.3.1.47

INCONEL 718

R

Condition/Ht: STA
 Form: 3 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 5 Hz
 Environment: LAB AIR; RT

Yield Strength: 177.2 ksi
 Ult. Strength:
 Specimen Thk: 0.204 in.
 Specimen Width: 4.008 in.
 Ref: DA006

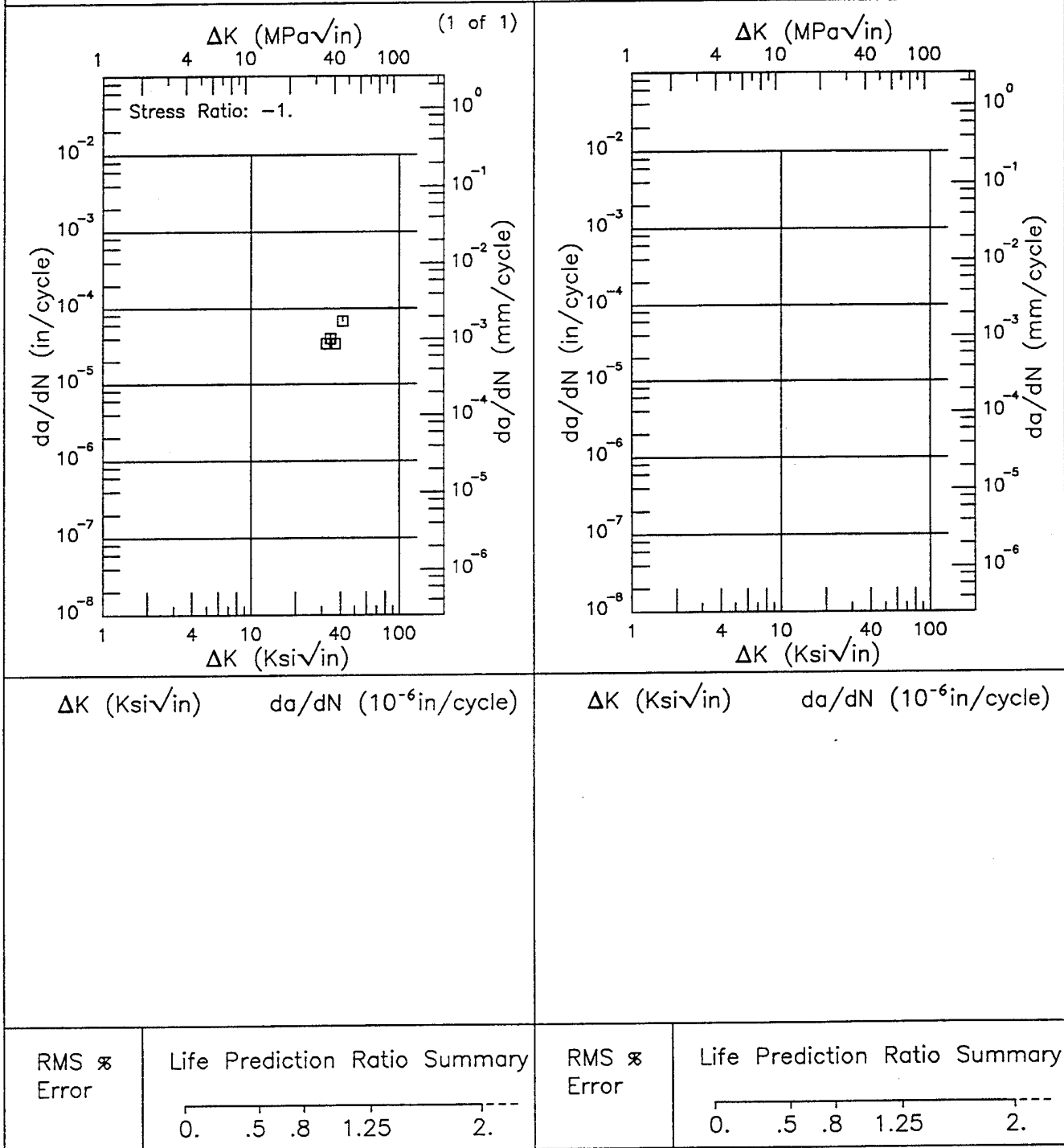


Figure 5.9.3.1.48

INCONEL 718

(1 of 2)

TABLE 5.9.3.3
K_{Iacc} SUMMARY FOR NICKEL-BASED SUPER ALLOY INCONEL 718

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K _Q (Ksi√in)	K _{Iacc} (Ksi√in)	Test Time (min)	Test Date	Refer
						Design	Width (in)	Thick (in)							
1850F 1.5HR OQ; 1360F 9HRS FC TO 1175F OQ; 1360F 9HRS FC TO 1175F			L-T	160	S.C.S.	DCB	2	1	4	---	193	>186*	60180	1976	RI006
							DCB	2	1	4	---	193	>86	60180	1976
						DCB	2	1	4	---	103	>86	60060	1976	RI006
							DCB	2	1	4	---	103	>180*	119100	1976
						DCB	2	1	4	---	104	>89	60060	1976	RI006
							DCB	2	1	4	---	104	121*	119100	1976
	DCB	2	1	4	160	S.T.W.	2	1	4	---	104	>87	60120	1976	RI006
		DCB	2	1	4	160	S.T.W.	2	1	4	---	104	>99	76380	1976

(2 of 2)

TABLE 5.9.3.3 (CONCLUDED)
K_{Isc} SUMMARY FOR NICKEL-BASED SUPER ALLOY INCONEL 718

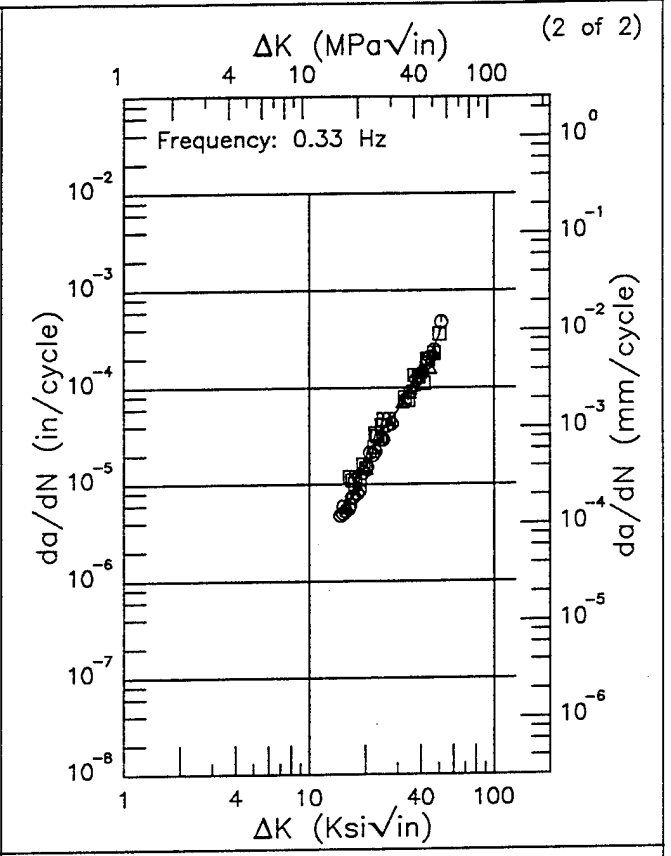
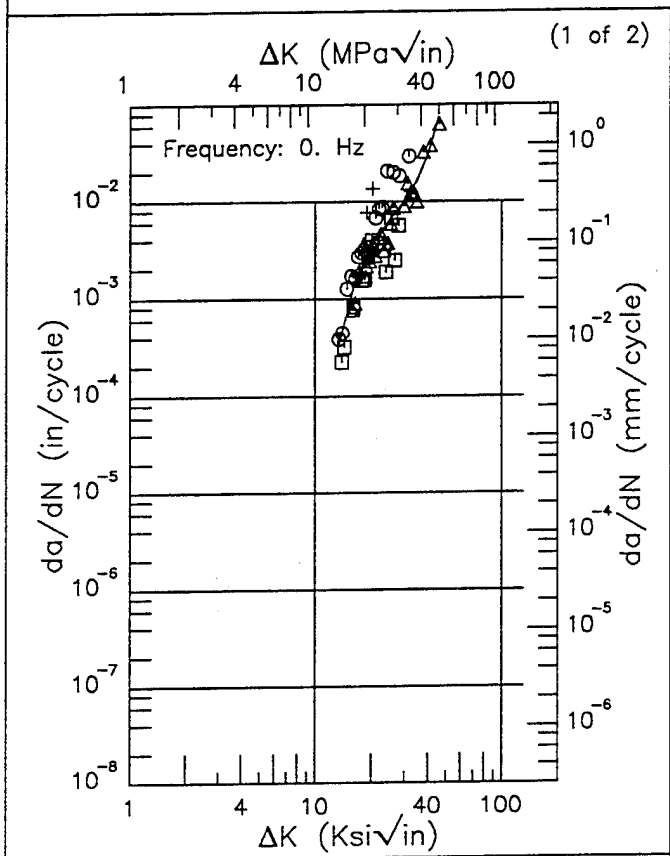
Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K _Q (Ksi√in)	K _{Isc} (Ksi√in)	Test Time (min)	Test Date	Refer			
						Design	Width (in)	Thick (in)										
1880F 1HR AC; 1520F 8HR FC; 1200F 16HR AC; 1520F 8HR FC; 1200F 16HR AC	s	R.T.	---	---	Aerofine 50	WOL	1.3	0.125	0.13	---	---	86*	---	1974	88700			
					Martin- Marietta refined grade hydrazine	WOL	1.3	0.125	0.13	---	---	79*	---	---	---	1974	88700	
					Martin- Marietta refined grade hydrazine - 2% oxygen	WOL	1.3	0.125	0.13	---	---	79*	---	---	---	---	1974	88700
					Matheson- Coleman-Bell 97% grade hydrazine	WOL	1.3	0.125	0.13	---	---	25.8*	---	---	---	---	1974	88700
					Propellant grade hydrazine	WOL	1.3	0.125	0.13	---	---	87.5*	---	---	---	---	1974	88700

* specimen thickness does not meet minimum requirements of $2.5 \left(\frac{K_{Isc}}{\sigma_{yp}} \right)^2$

F | NASA IIB-7 P/M |

Condition/Ht: 1650F 16HRS TO 2000F 1HR OQ
 Form: 1.75 in. Disk
 Specimen Type: WOL
 Orientation: C-R
 Stress Ratio: 0.05
 Environment: LAB AIR;1200°F

Yield Strength: 208.6 ksi
 Ult. Strength: 257.1 ksi
 Specimen Thk: 0.5 in.
 Specimen Width: 2.5 in.
 Ref: PW004



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.30 (min)	315.
16.	1303.
20.	3593.
25.	6526.
30.	10084.
35.	16495.
40.	30610.
45.79 (max)	74742.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
14.67 (min)	4.74
16.	6.09
20.	15.4
25.	36.7
30.	59.0
35.	91.8
40.	141.
50.	338.
51.24 (max)	432.

RMS % Error	Life Prediction Ratio Summary
73.31	

RMS % Error	Life Prediction Ratio Summary
16.78	

Figure 5.10.3.1

TABLE 5.11.1.2

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
P/M RENE 95 AT ROOM TEMPERATURE

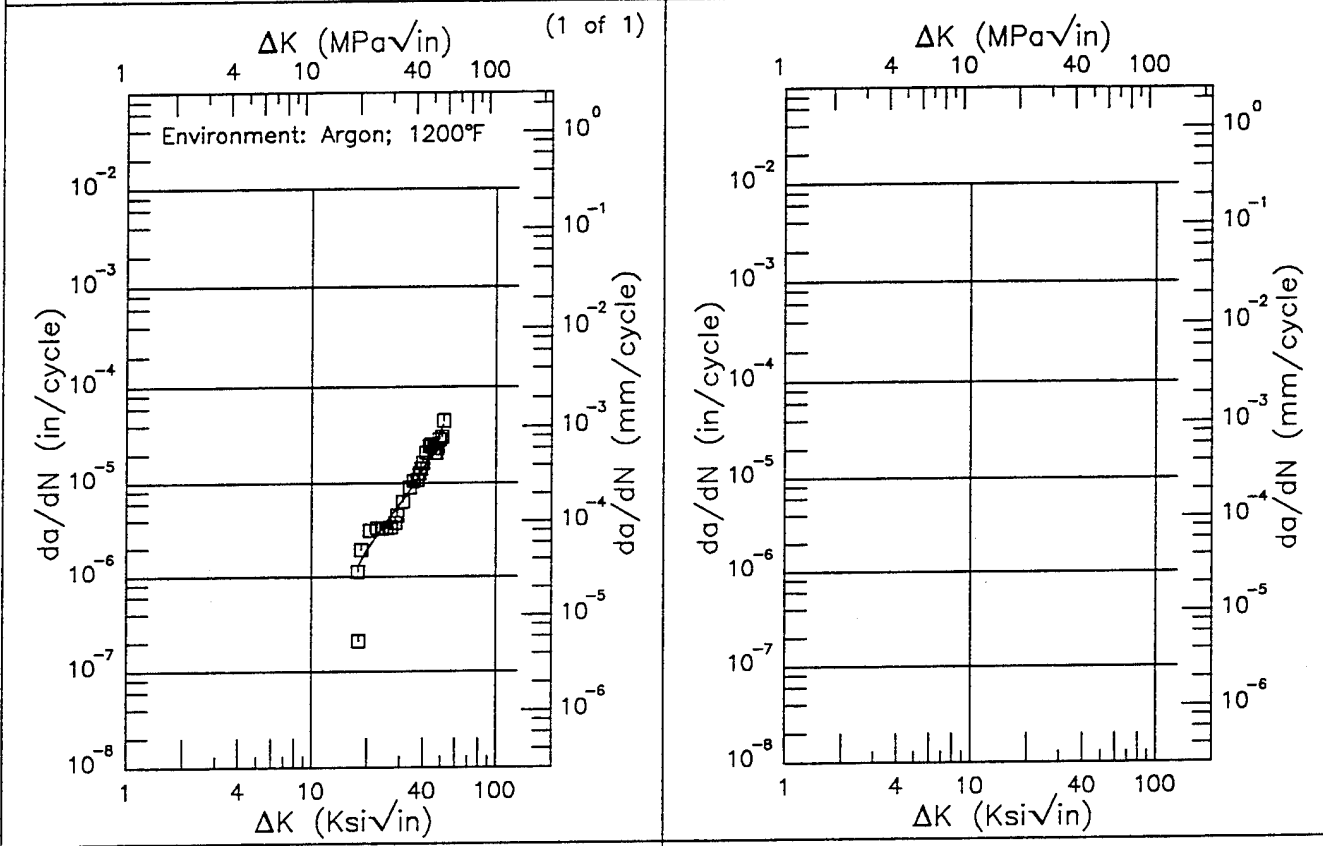
ORIENTATION: C-R ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi/in)					
				2.5	5.0	10.0	20.0	50.0	100.0
2080F 1HR AC 1600F 1HR AC	DISK	0.	0.33				0.4	31.02	

E | P/M RENE 95 |

Condition/Ht: 2080F 1HR AC 1600F 1HR AC
 Form: 2.5 in. Disk
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Stress Ratio: 0.
 Frequency: 0.3 Hz

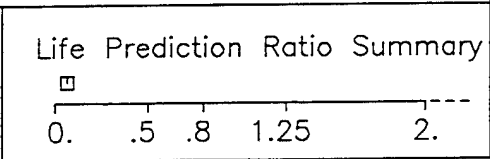
Yield Strength: 165.7 ksi
 Ult. Strength: 228.6 ksi
 Specimen Thk: 0.08 in.
 Specimen Width: 2 in.
 Ref: GE004



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
17.82 (min)	1.29
20.	1.94
25.	3.71
30.	5.95
35.	9.11
40.	14.0
50.	35.3
51.58 (max)	41.3

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS % Error
24.97



RMS % Error

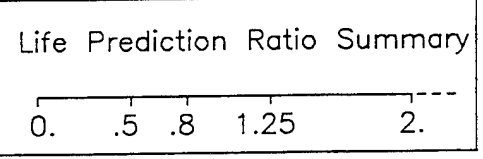


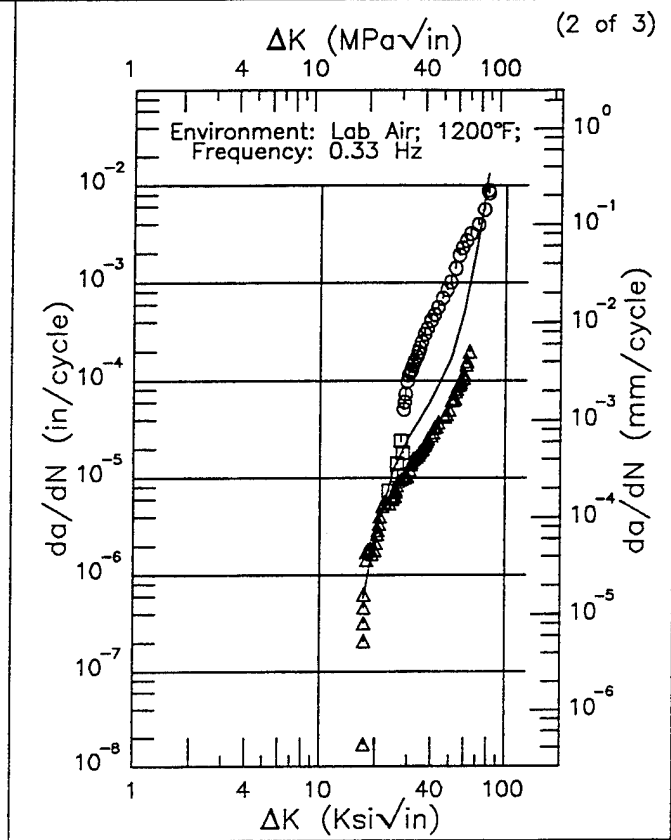
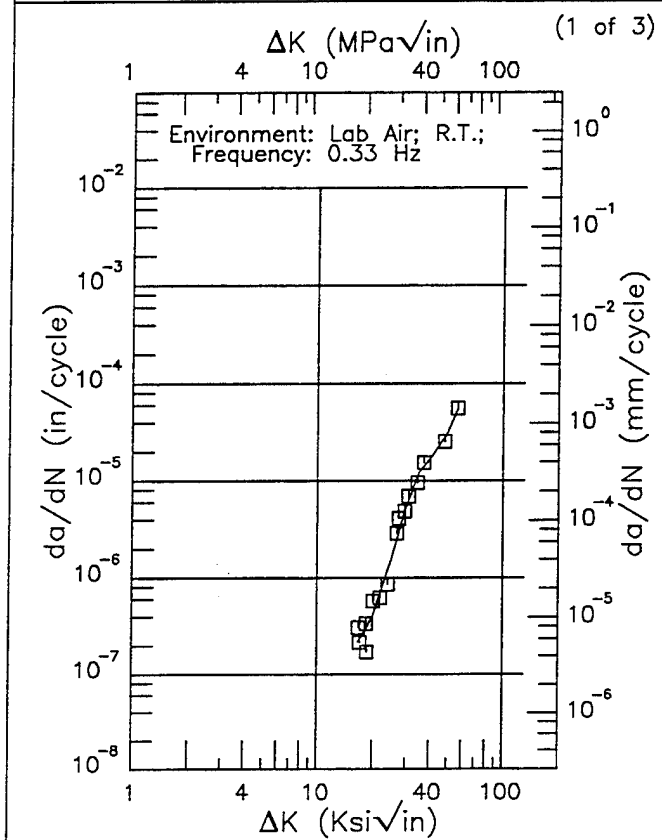
Figure 5.11.3.1.1

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EF | P/M RENE 95 |

Condition/Ht: 2080F 1HR AC 1600F 1HR AC
 Form: 2.5 in. Disk
 Specimen Type: CCP (max stress specified)
 Orientation: C-R
 Stress Ratio: 0.

Yield Strength: 165.7 ksi
 Ult. Strength: 228.6 ksi
 Specimen Thk: 0.08 in.
 Specimen Width: 2 in.
 Ref: GE008



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.68 (min)	0.220
20.	0.400
25.	1.56
30.	5.41
35.	12.2
40.	16.5
50.	31.0
56.41 (max)	55.2

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
17.22 (min)	0.577
20.	2.67
25.	12.0
30.	25.7
35.	42.0
40.	63.7
50.	159.
60.	517.
70.	2253.
80.	12711.
80.21 (max)	13214.

RMS % Error 24.27	Life Prediction Ratio Summary 0. .5 .8 1.25 2.-----
----------------------	--

RMS % Error >100.0	Life Prediction Ratio Summary 0. .5 .8 1.25 2.-----
-----------------------	--

Figure 5.11.3.1.2

P/M RENE 95

EF

Condition/Ht: 2080F 1HR AC 1600F 1HR AC
 Form: 2.5 in. Disk
 Specimen Type: CCP (max stress specified)
 Orientation: C-R
 Stress Ratio: 0.

Yield Strength: 165.7 ksi
 Ult. Strength: 228.6 ksi
 Specimen Thk: 0.08 in.
 Specimen Width: 2 in.
 Ref: GE008

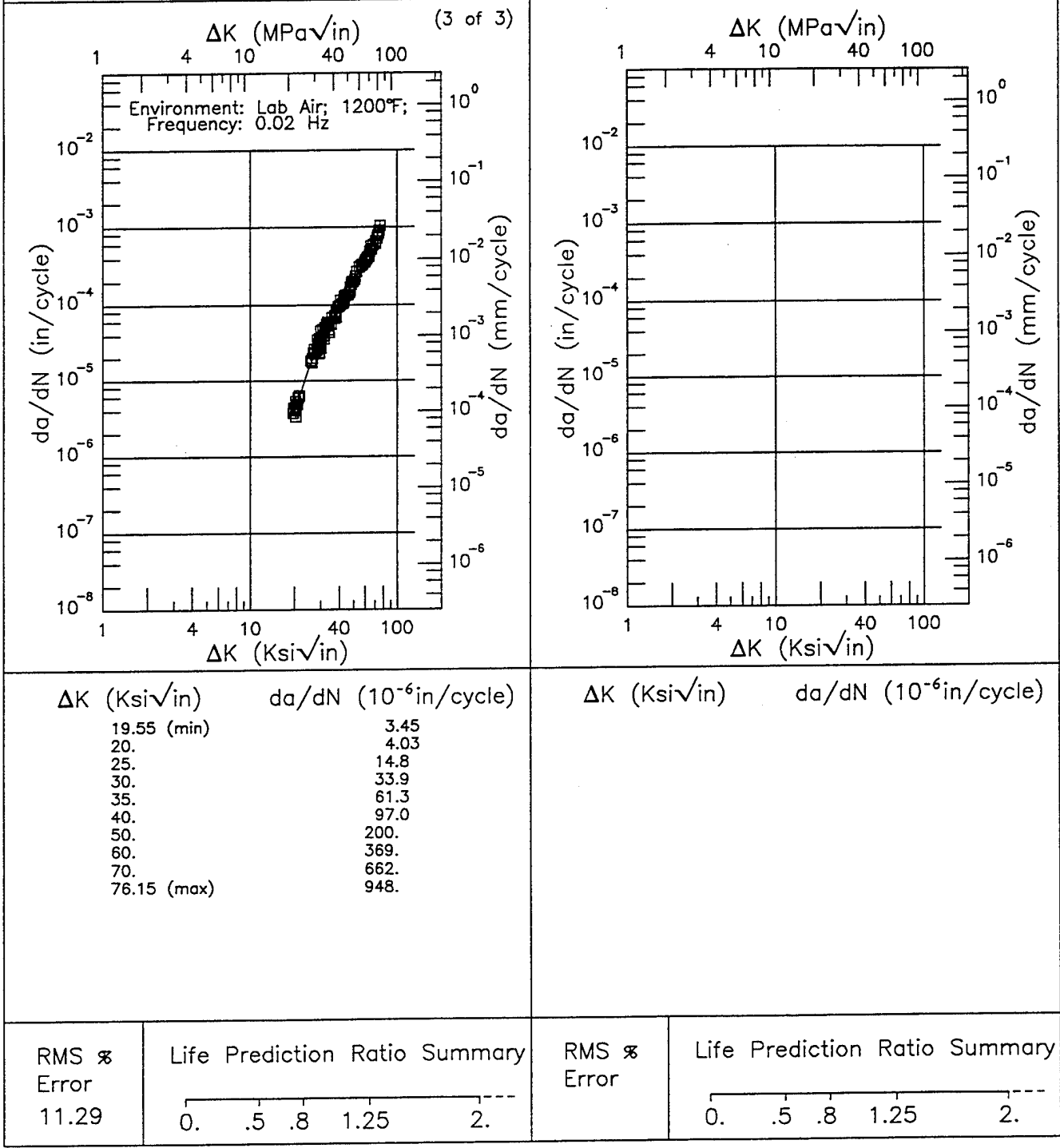
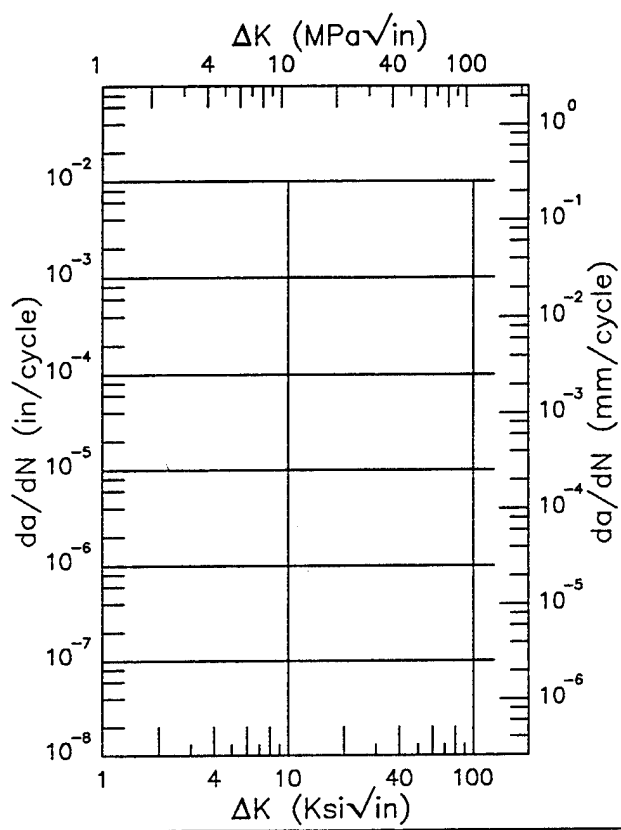
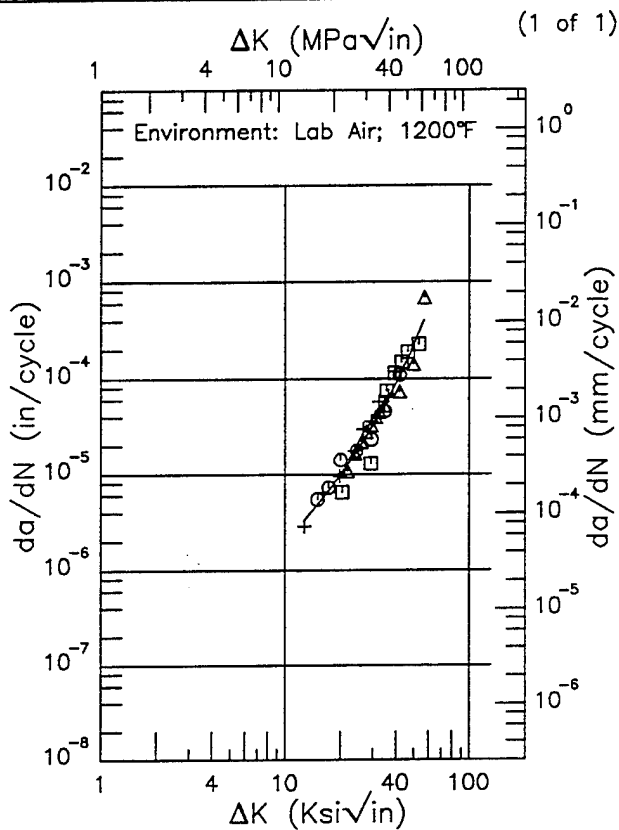


Figure 5.11.3.1.2 (Concluded)

P/M RENE 95 E

Condition/Ht: 2100F 1HR SQ AT 1000F
 Form: 1 in. Disk
 Specimen Type: KB Bar
 Orientation: C-R
 Stress Ratio: 0.05
 Frequency: 0.3 Hz

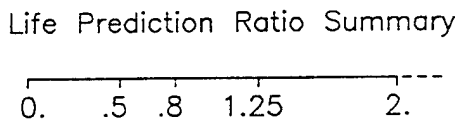
Yield Strength: 176.1 ksi
 Ult. Strength:
 Specimen Thk: 0.25 in.
 Specimen Width: 0.6 in.
 Ref: GE001



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
12.70 (min)	3.28
13.	3.46
16.	5.69
20.	10.0
25.	18.6
30.	32.7
35.	55.1
40.	90.4
50.	228.
56.60 (max)	405.

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS %
Error
25.51



RMS %
Error

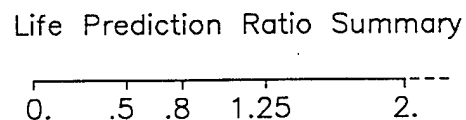


Figure 5.11.3.1.4

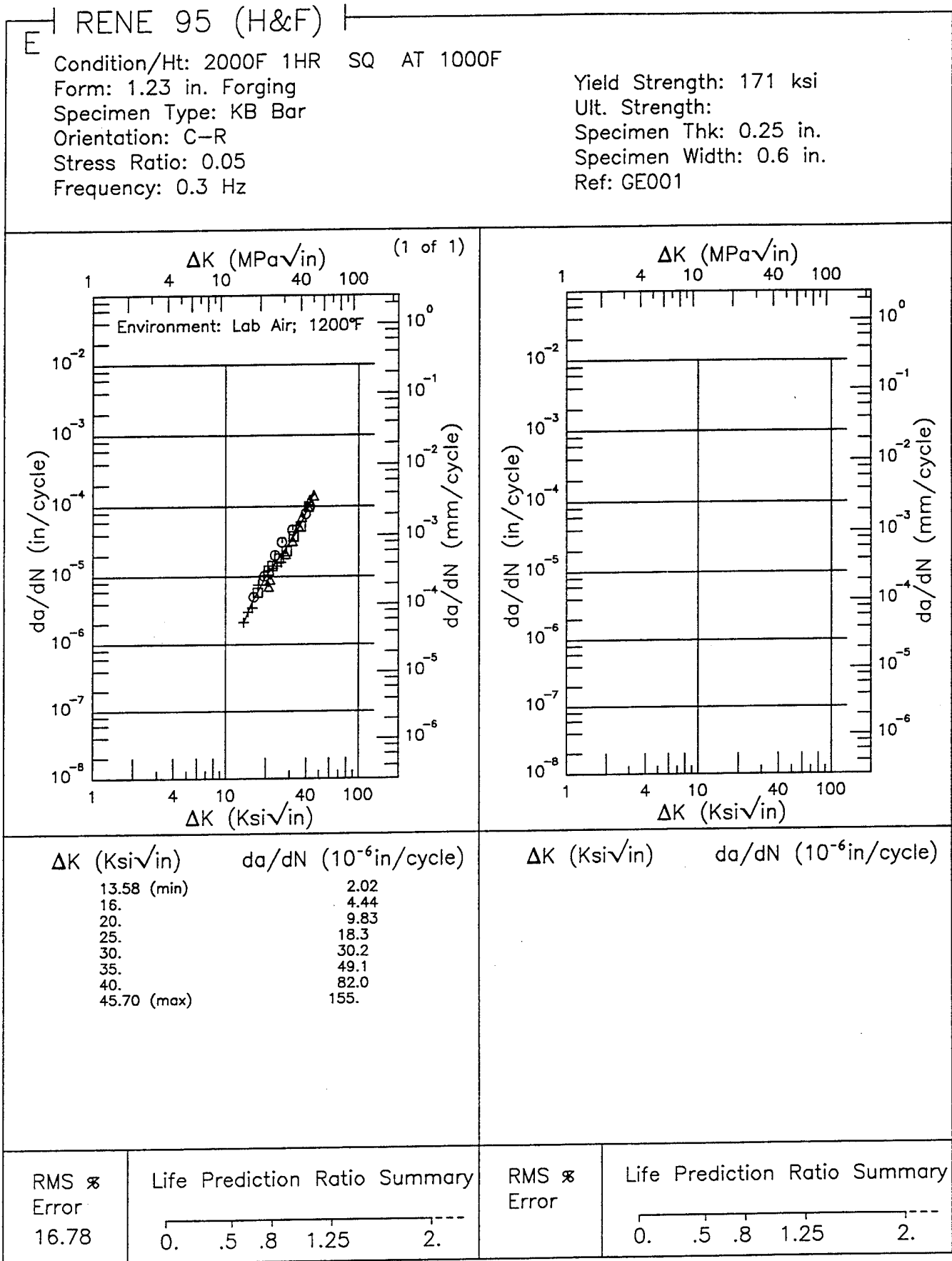


Figure 5.12.3.1

TABLE 5.13.1.2

1 of 1

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
WASPALLOY AT ROOM TEMPERATURE**

ORIENTATION: Unspecified		ENVIRONMENT: Lab Air								
CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
1850F 2HRS 1350F 6HRS	BILLET	0.1	10					10.46		
2010F 2HRS 1350F 6HRS	BILLET	0.1	10					2.8		
2010F 2HRS 1600F 24HRS	BILLET	0.1	10					9.56		

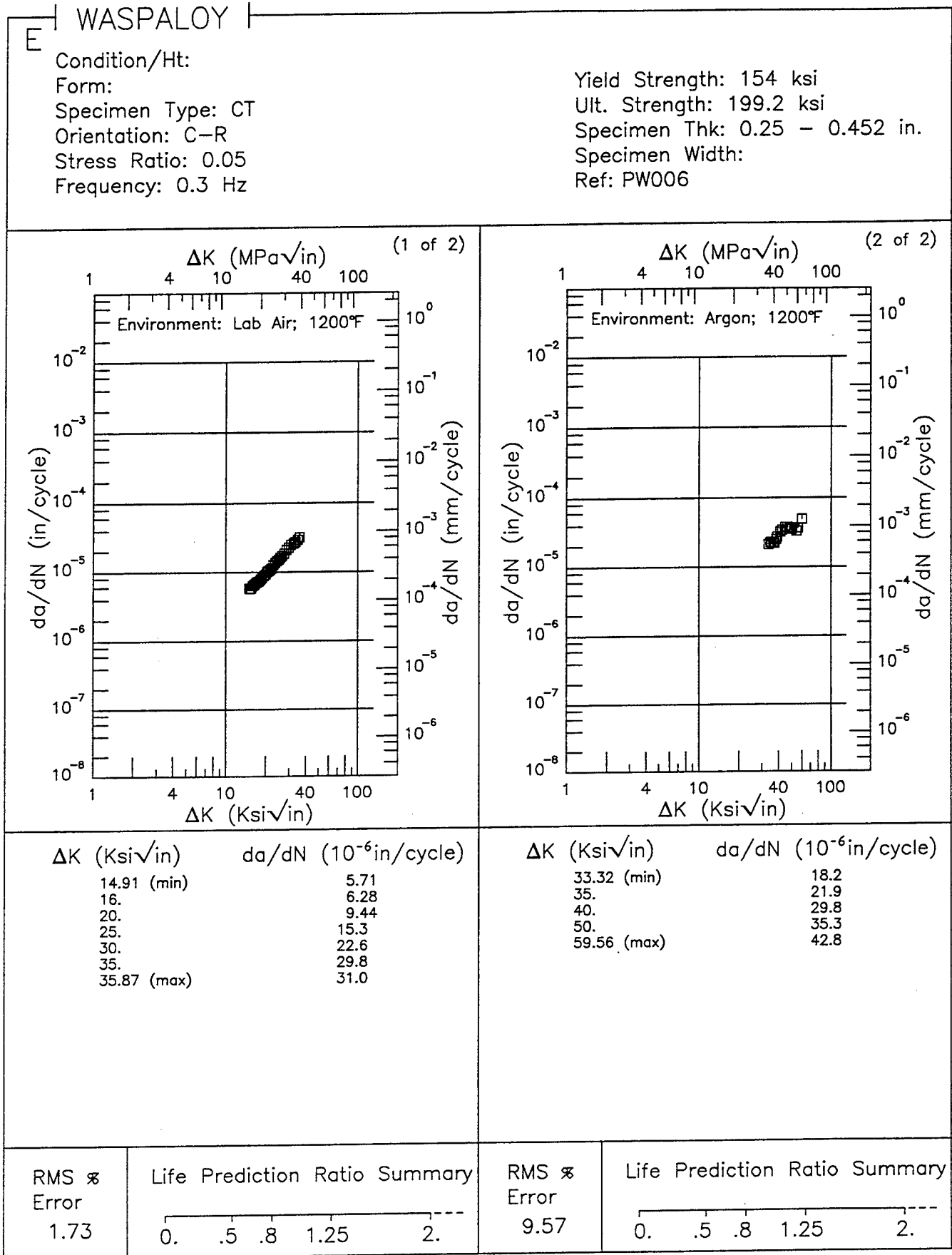


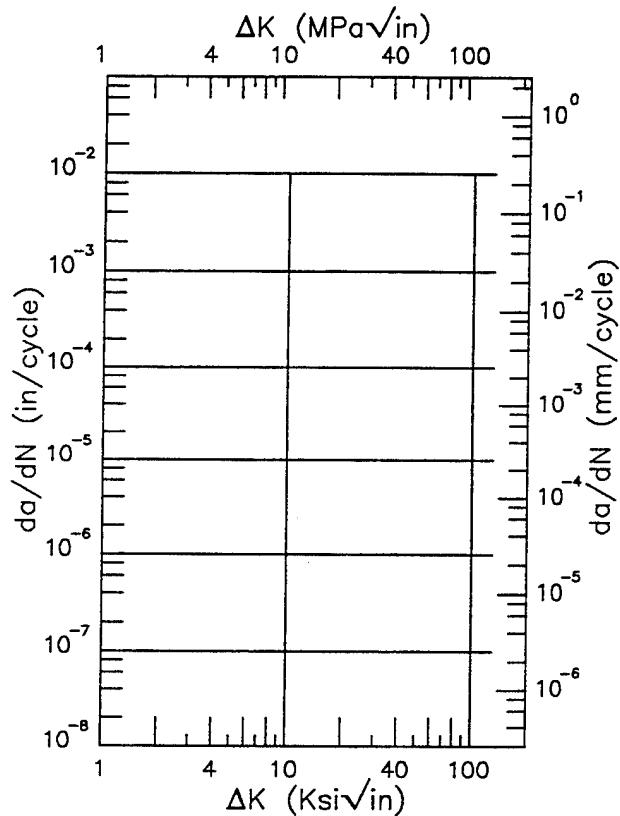
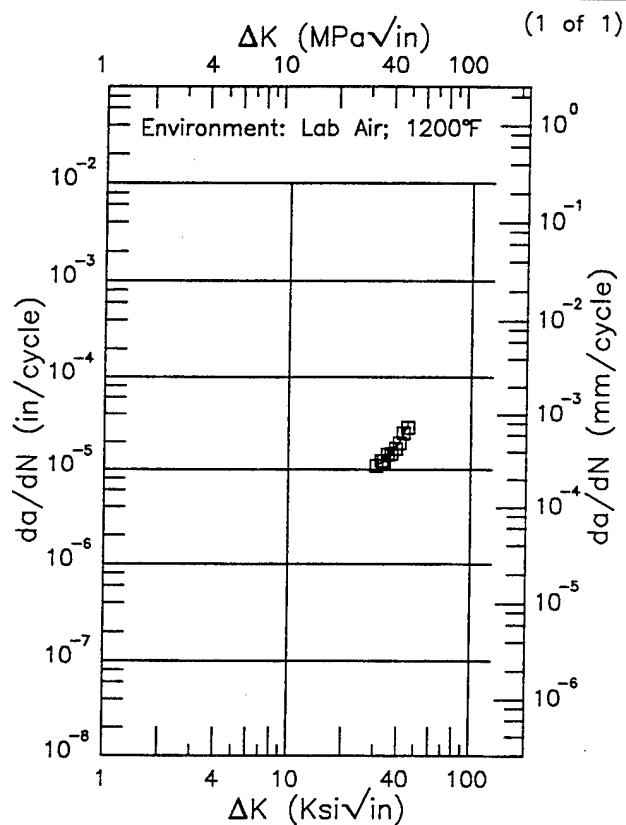
Figure 5.13.3.1.1

WASPALLOY

E

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.05
 Frequency: 20 Hz

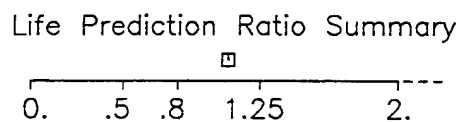
Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.753 in.
 Specimen Width: 2.505 in.
 Ref: PW001



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
30.41 (min)	11.0
35.	13.7
40.	18.8
44.85 (max)	29.4

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS %
 Error
 3.19



RMS %
 Error

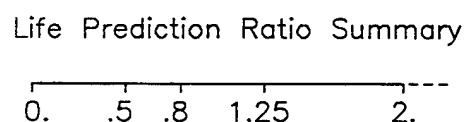
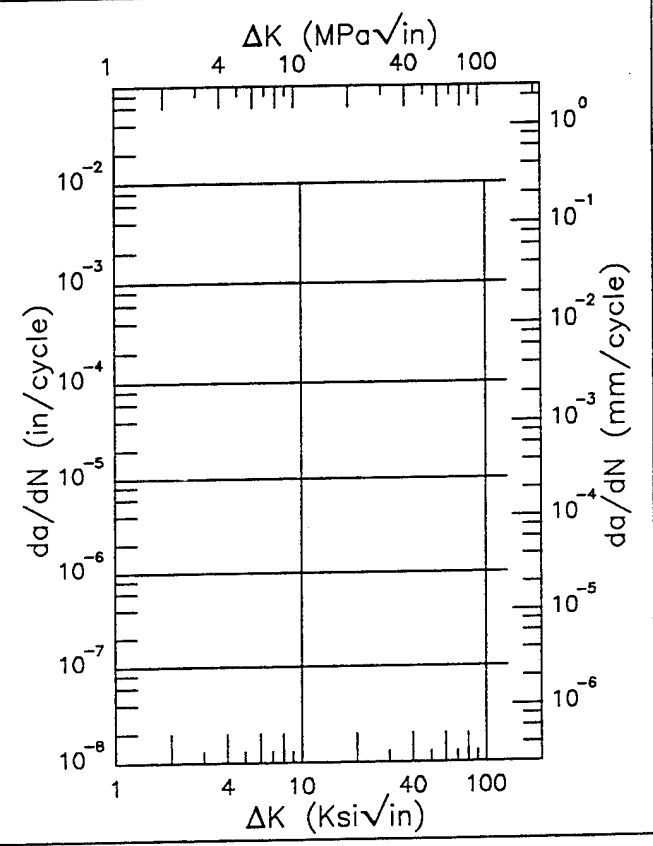
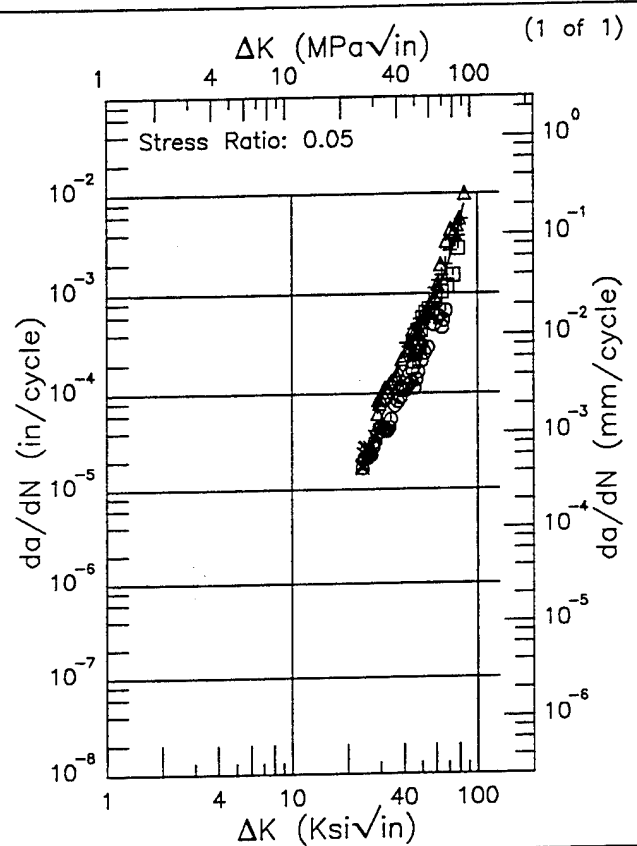


Figure 5.13.3.1.2

R | WASPALOY |

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation: C-R
 Frequency:
 Environment: LAB AIR;1200°F

Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.316 - 0.422 in.
 Specimen Width: 2.504 - 2.509 in.
 Ref: PW001



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
23.65 (min)	16.2
25.	22.3
30.	54.9
35.	103.
40.	170.
50.	394.
60.	885.
70.	2061.
80.	5074.
84.43 (max)	7701.

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS % Error	Life Prediction Ratio Summary
38.52	

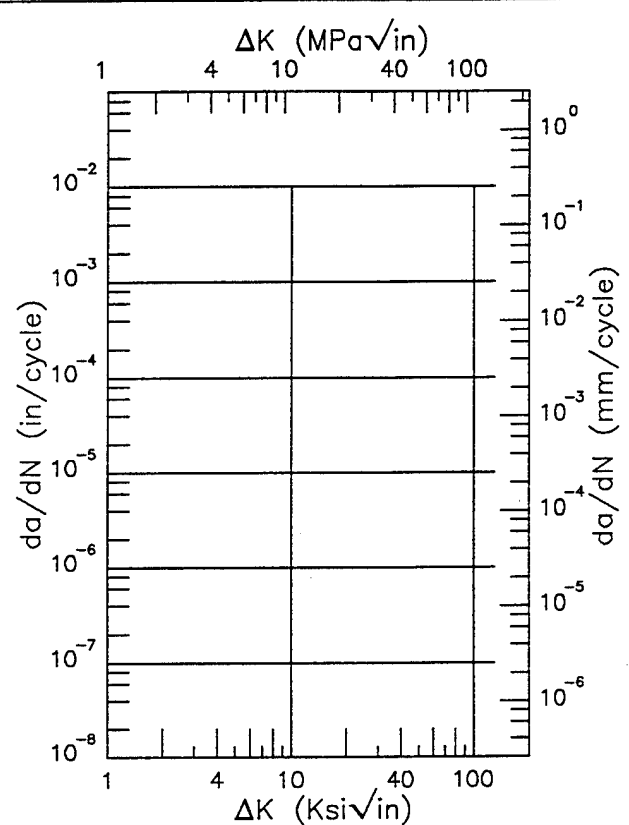
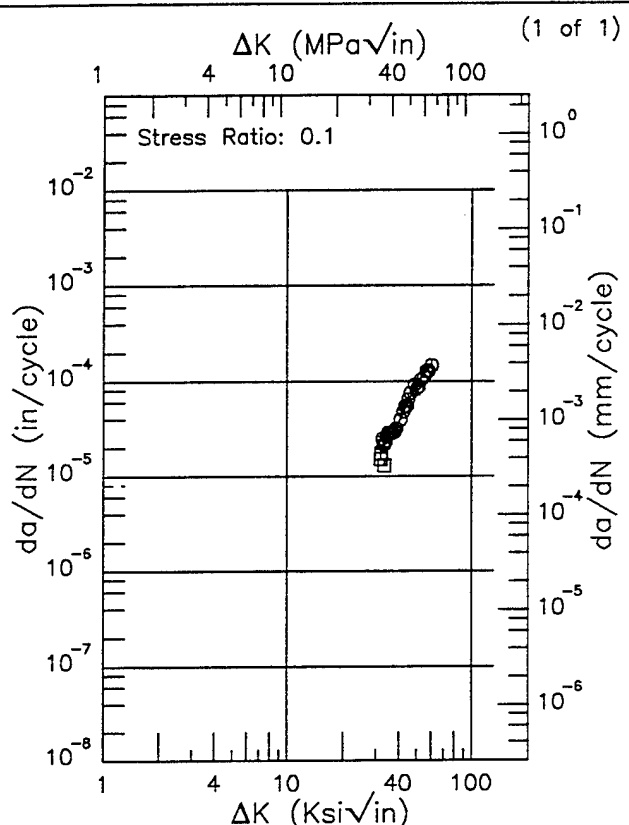
RMS % Error	Life Prediction Ratio Summary

Figure 5.13.3.1.3

WASPALLOY R

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation: C-R
 Frequency:
 Environment: LAB AIR;1350°F

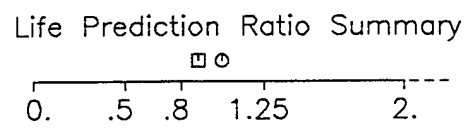
Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.299 - 0.3 in.
 Specimen Width: 2.504 - 2.511 in.
 Ref: PW001



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
31.69 (min)	17.6
35.	25.5
40.	37.6
50.	93.8
60.	150.
60.02 (max)	150.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
--------------------------------------	-------------------------------

RMS \times
 Error
 11.82



RMS \times
 Error

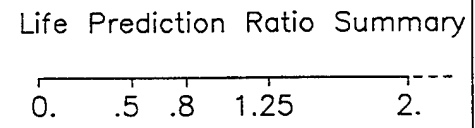
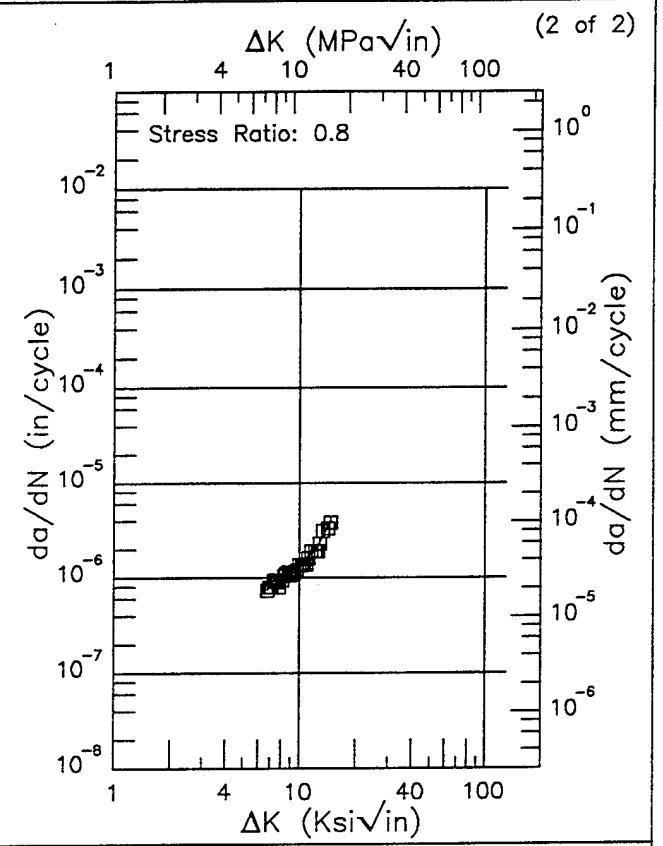
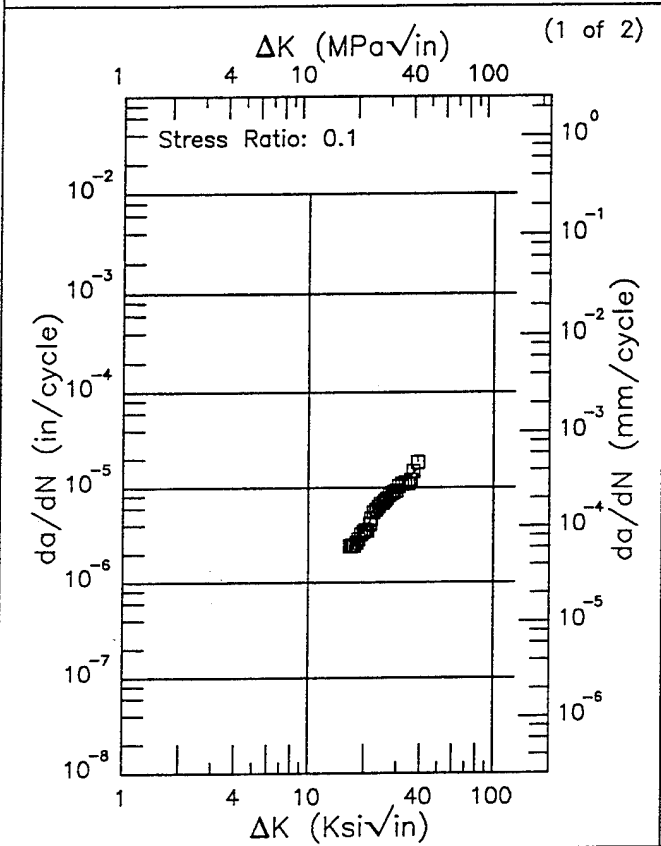


Figure 5.13.3.1.4

R | WASPALOY |

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation: C-R
 Frequency: 20 Hz
 Environment: LAB AIR;1200°F

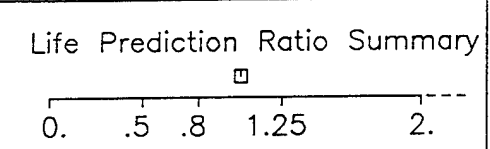
Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.3 - 0.302 in.
 Specimen Width: 2.503 - 2.518 in.
 Ref: PW001



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.66 (min)	2.04
20.	3.73
25.	6.59
30.	9.62
35.	13.0
38.72 (max)	15.8

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
6.75 (min)	0.761
7.	0.801
8.	0.947
9.	1.10
10.	1.28
13.	2.43
14.64 (max)	3.92

RMS \times
 Error
 7.34



RMS \times
 Error
 7.28

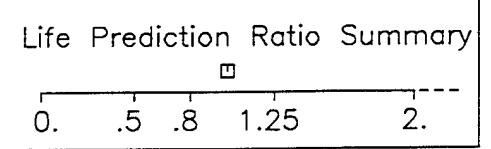


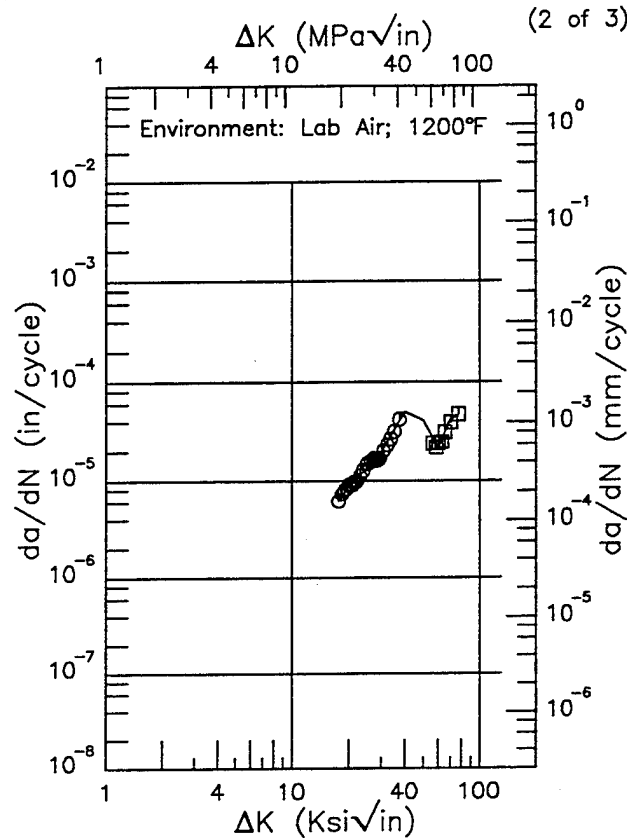
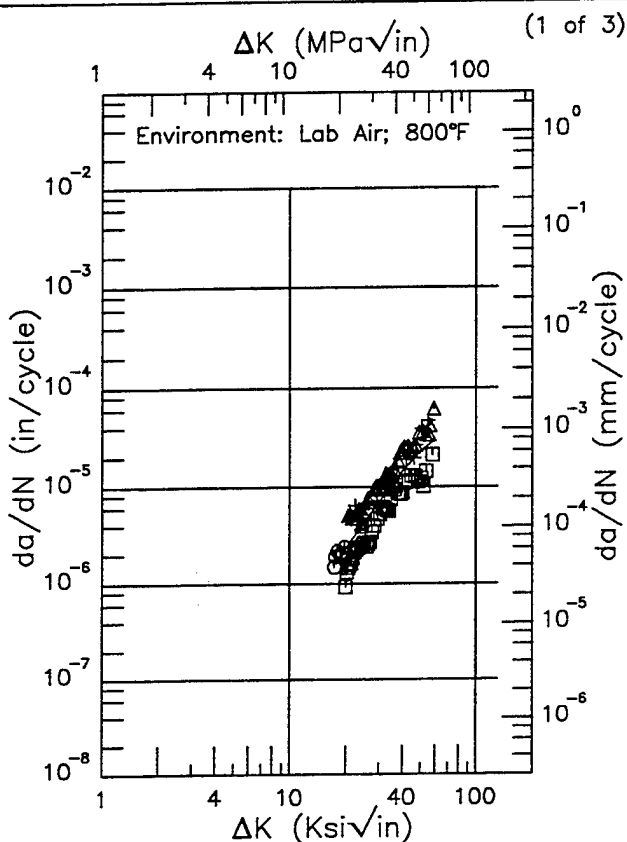
Figure 5.13.3.1.5

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WASPALLOY

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.05
 Frequency: 0.2 Hz

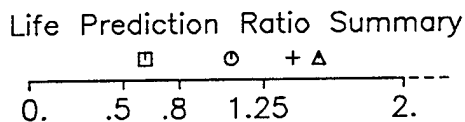
Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.102 - 0.501 in.
 Specimen Width: 2.486 - 2.519 in.
 Ref: PW001



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
17.25 (min)	1.57
20.	2.32
25.	4.28
30.	7.01
35.	10.4
40.	14.4
50.	22.8
58.47 (max)	29.4

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
17.60 (min)	6.24
20.	8.79
25.	13.3
30.	19.6
35.	32.5
40.	50.4
50.	41.3
60.	21.8
70.	42.1
76.85 (max)	45.9

RMS $\%$
 Error
 44.49



RMS $\%$
 Error
 5.85

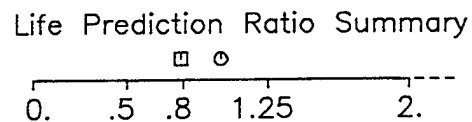
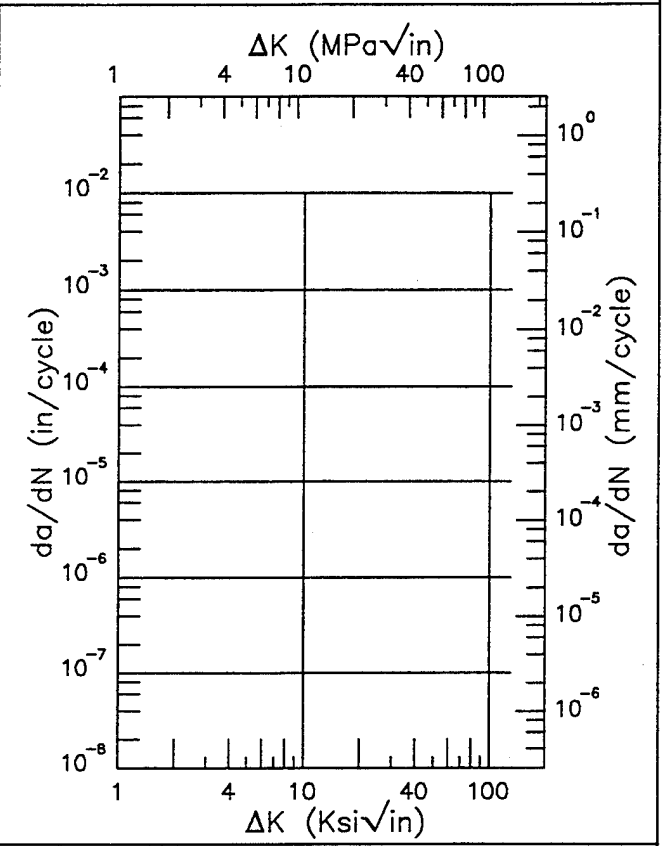
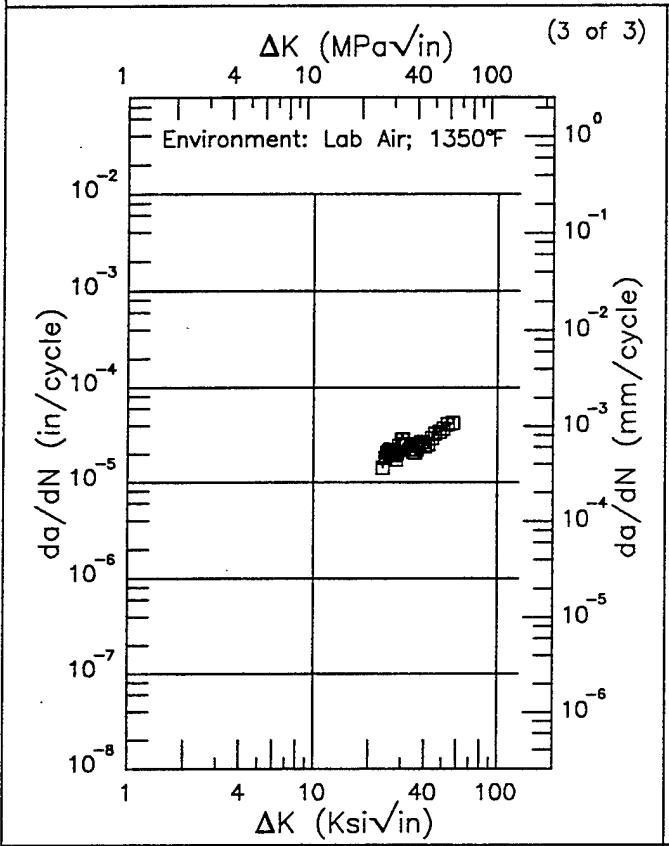


Figure 5.13.3.1.6

WASPALLOY E

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.05
 Frequency: 0.2 Hz

Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.102 - 0.501 in.
 Specimen Width: 2.486 - 2.519 in.
 Ref: PW001



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
23.76 (min)	15.2
25.	18.6
30.	23.4
35.	22.8
40.	25.0
50.	36.3
56.57 (max)	42.1

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)

RMS % Error	Life Prediction Ratio Summary
9.06	

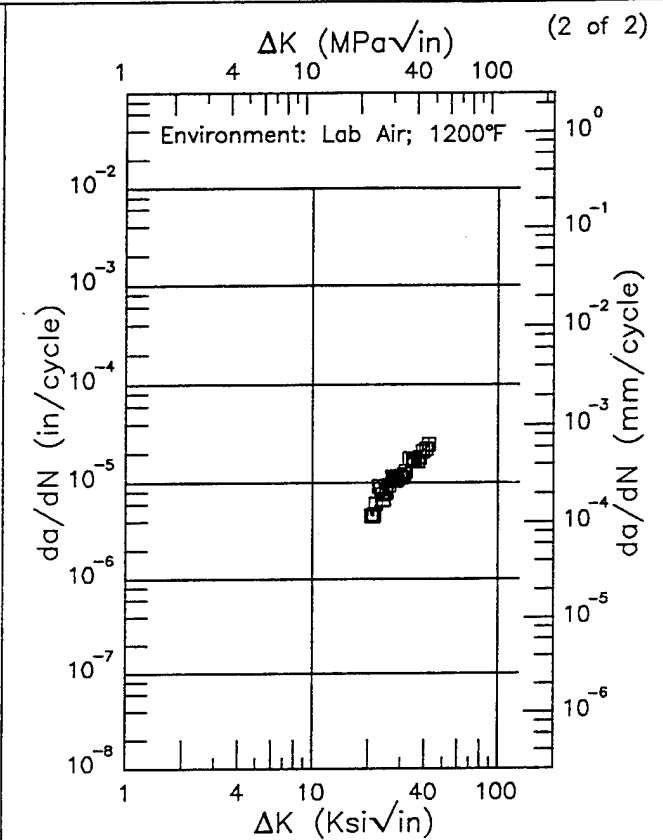
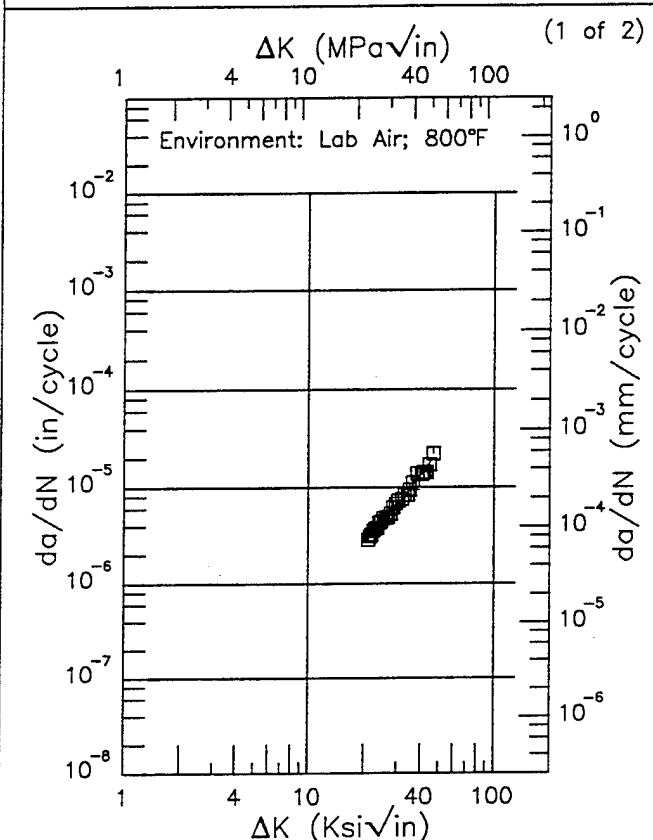
RMS % Error	Life Prediction Ratio Summary

Figure 5.13.3.1.6 (Concluded)

WASPALLOY

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.1
 Frequency: 20 Hz

Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.307 - 0.5 in.
 Specimen Width: 2.508 - 2.51 in.
 Ref: PW001



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
20.94 (min)	2.96
25.	4.45
30.	6.72
35.	9.60
40.	13.3
46.62 (max)	19.9

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
20.71 (min)	4.85
25.	8.72
30.	12.5
35.	16.2
40.	21.5
42.08 (max)	24.6

RMS \times Error 5.34	Life Prediction Ratio Summary
-------------------------------	-----------------------------------

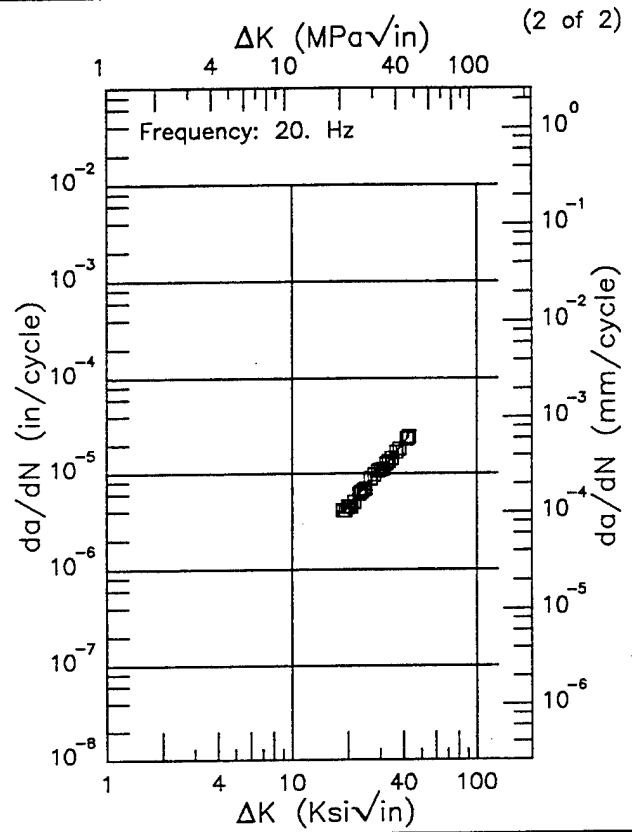
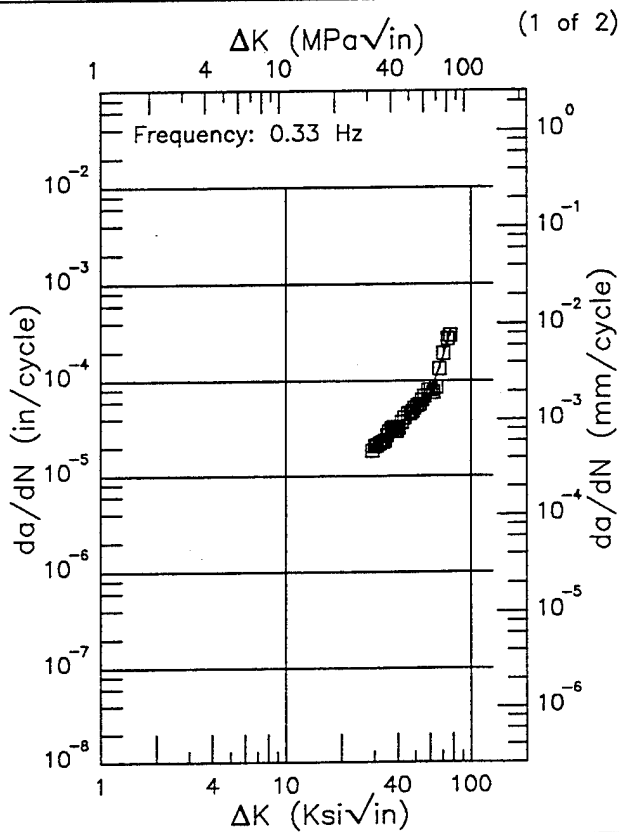
RMS \times Error 11.97	Life Prediction Ratio Summary
--------------------------------	-----------------------------------

Figure 5.13.3.1.7

WASPALOY F

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.05
 Environment: LAB AIR;1200°F

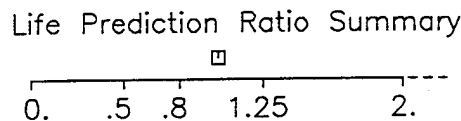
Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.432 - 0.447 in.
 Specimen Width: 2.502 - 2.503 in.
 Ref: PW001



ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
29.22 (min)	18.3
30.	19.6
35.	26.2
40.	32.8
50.	54.9
60.	74.3
70.	176.
76.26 (max)	309.

ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
18.78 (min)	4.06
20.	4.32
25.	7.45
30.	11.0
35.	14.8
40.	21.1
42.53 (max)	23.9

RMS %
 Error
 7.02



RMS %
 Error
 2.80

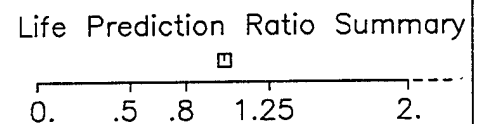
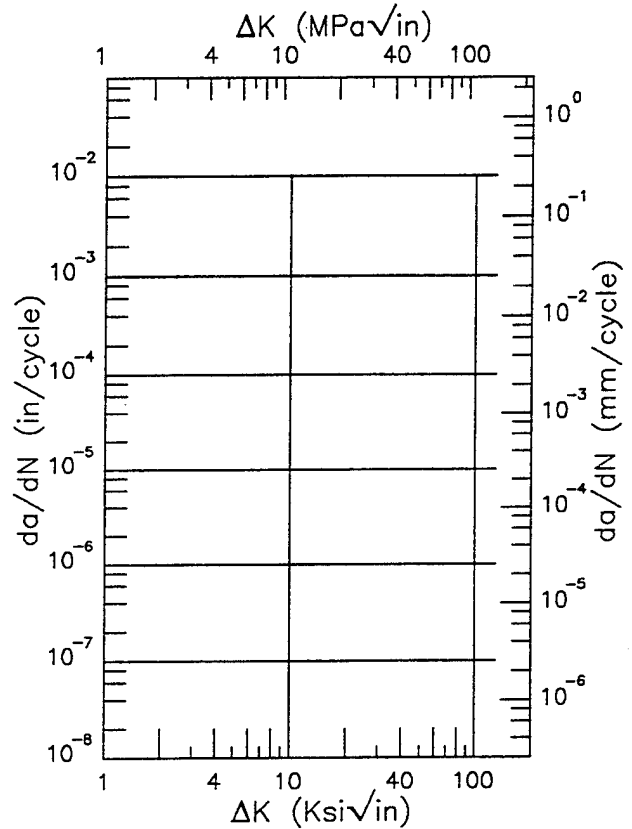
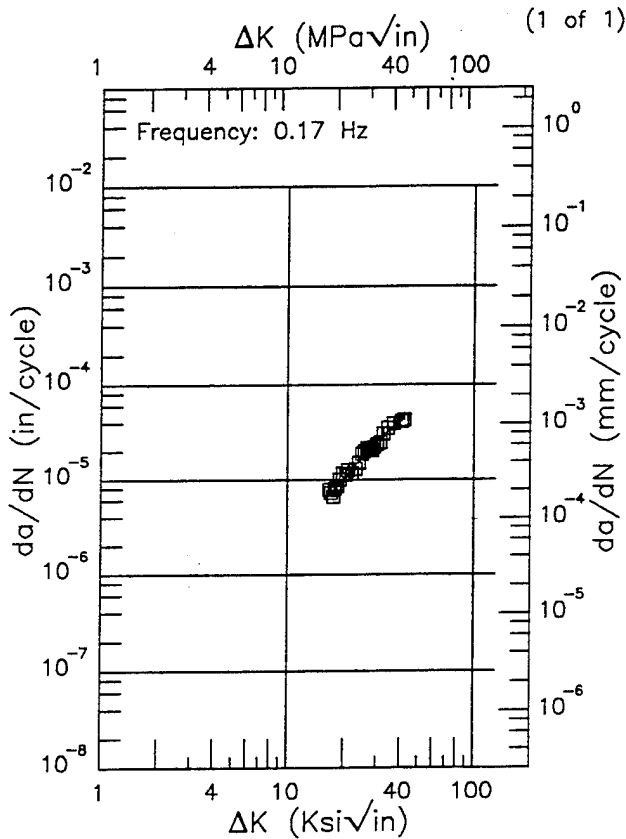


Figure 5.13.3.1.8

WASPALLOY

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.1
 Environment: LAB AIR;1200°F

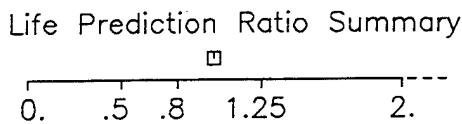
Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.318 in.
 Specimen Width: 2.484 in.
 Ref: PW001



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.71 (min)	7.03
20.	10.4
25.	17.1
30.	25.1
35.	33.6
40.	41.5
42.17 (max)	44.6

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS %
 Error
 8.10



RMS %
 Error

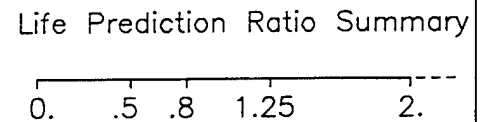


Figure 5.13.3.1.9

WASPALLOY R

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation:
 Frequency: 0.2 Hz
 Environment: LAB AIR;1200°F

Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.287 - 0.297 in.
 Specimen Width: 0.973 - 0.998 in.
 Ref: PW001

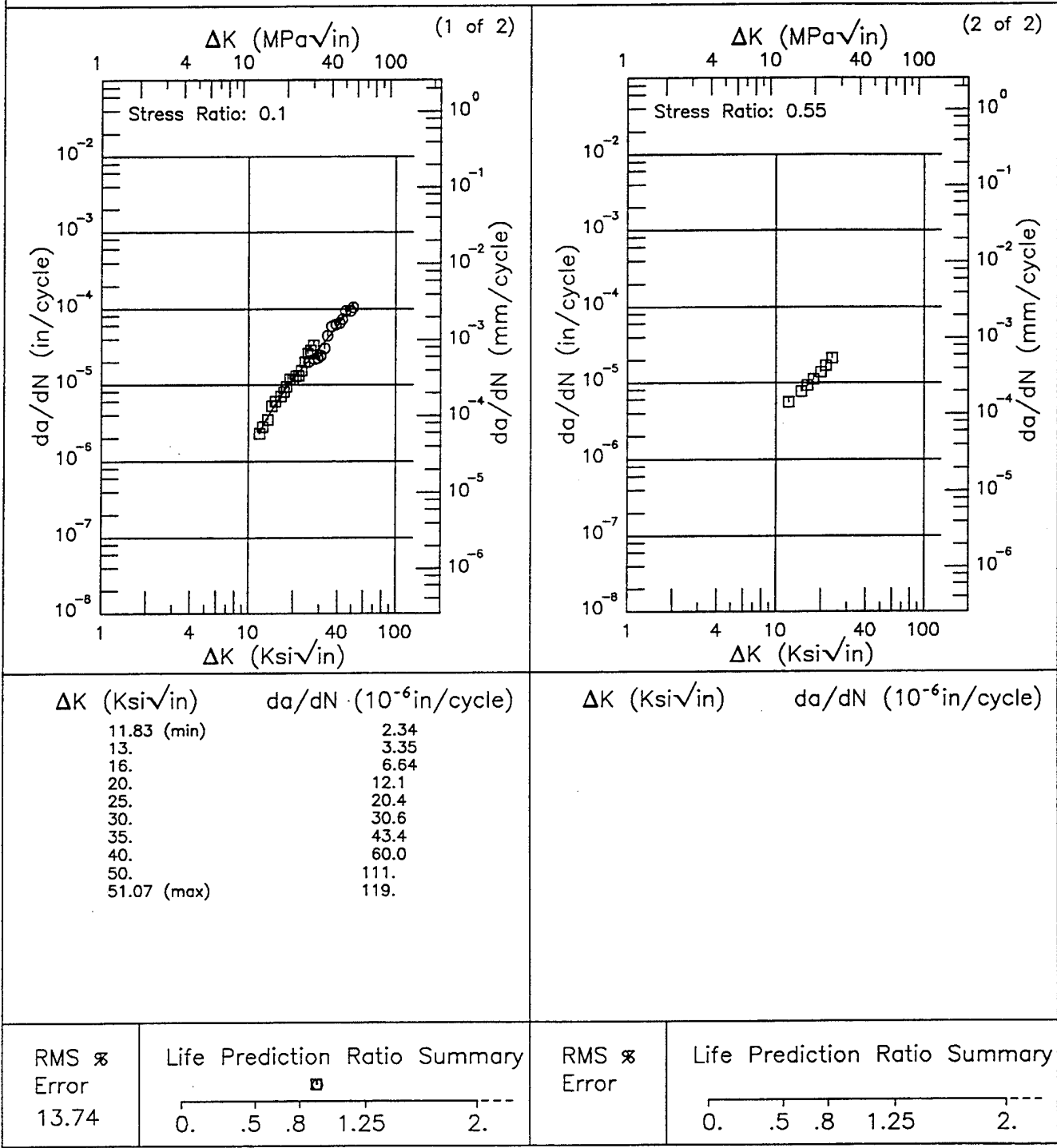
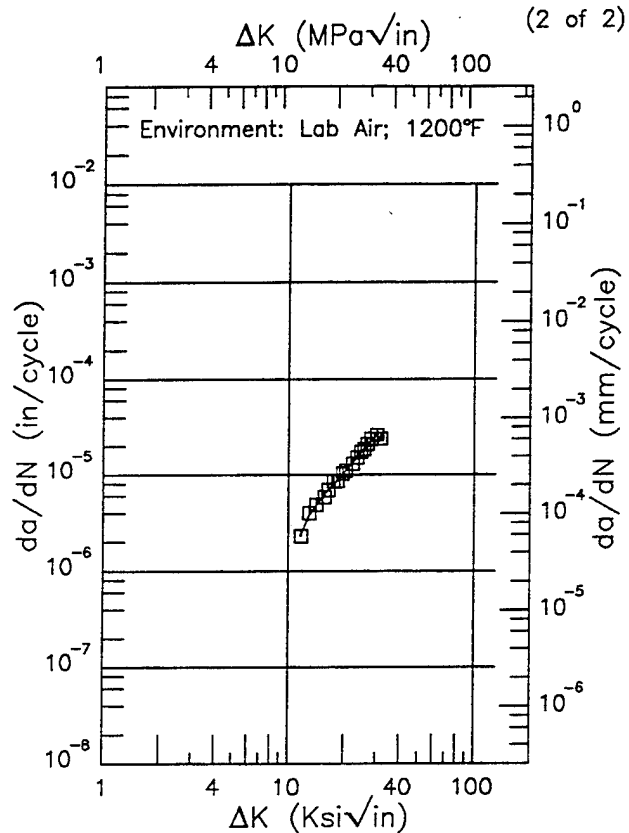
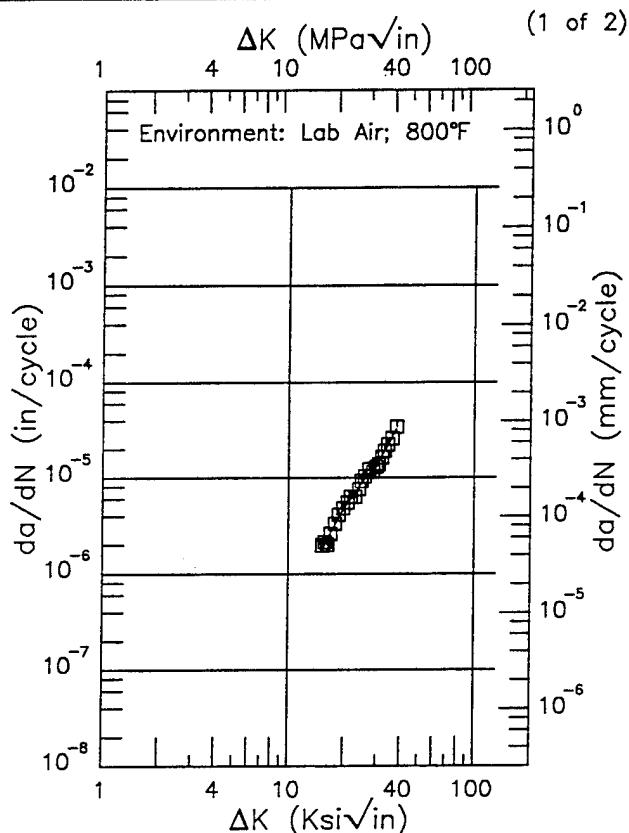


Figure 5.13.3.1.10

E | WASPALLOY |

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation:
 Stress Ratio: -1
 Frequency: 0.2 Hz

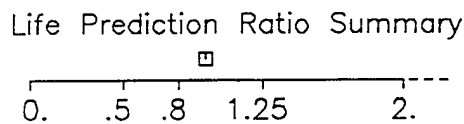
Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.297 - 0.3 in.
 Specimen Width: 0.997 in.
 Ref: PW001



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
15.16 (min)	1.74
16.	2.18
20.	4.80
25.	9.05
30.	13.7
35.	23.6
37.90 (max)	32.6

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
11.63 (min)	2.32
13.	3.84
16.	6.44
20.	10.5
25.	17.3
30.	25.8
31.47 (max)	23.4

RMS \times
 Error
 6.56



RMS \times
 Error
 3.61

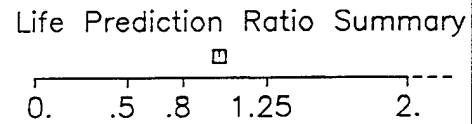


Figure 5.13.3.1.11

WASPALLOY

E

Condition/Ht:

Form: 1.6 in. Forging

Specimen Type: CCP (max load specified)

Orientation:

Stress Ratio: -0.5

Frequency: 0.2 Hz

Yield Strength: 154 ksi

Ult. Strength: 199.2 ksi

Specimen Thk: 0.298 in.

Specimen Width: 0.996 - 0.998 in.

Ref: PW001

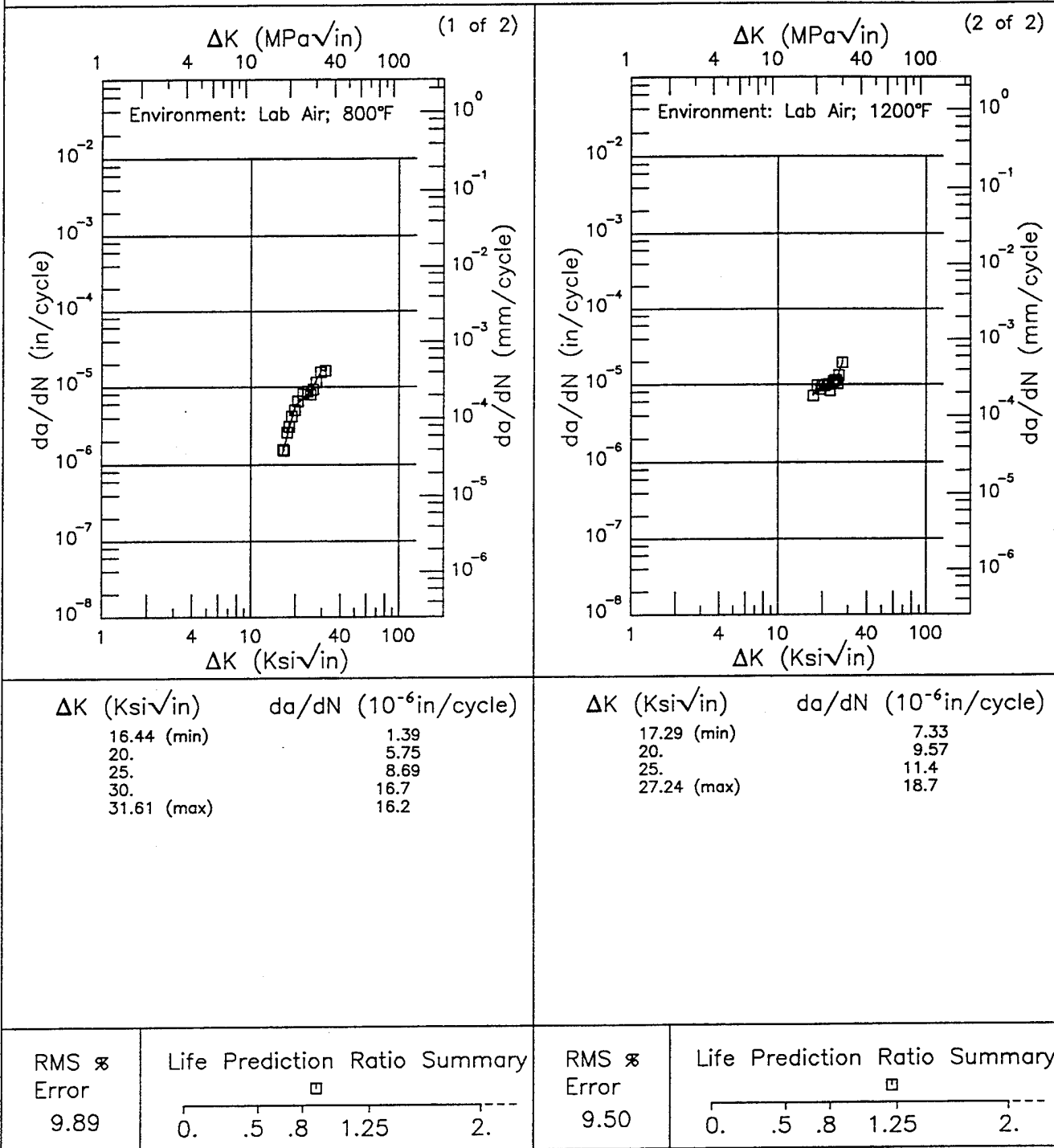


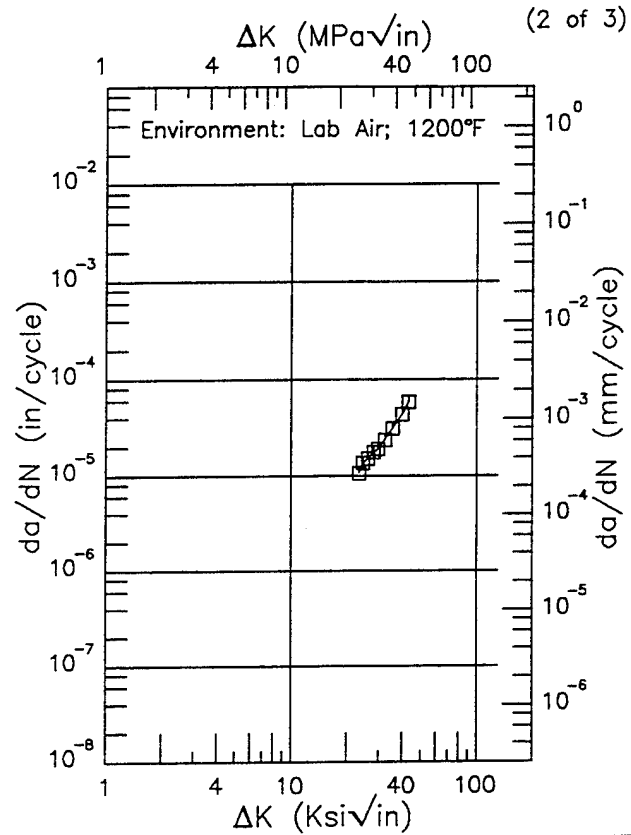
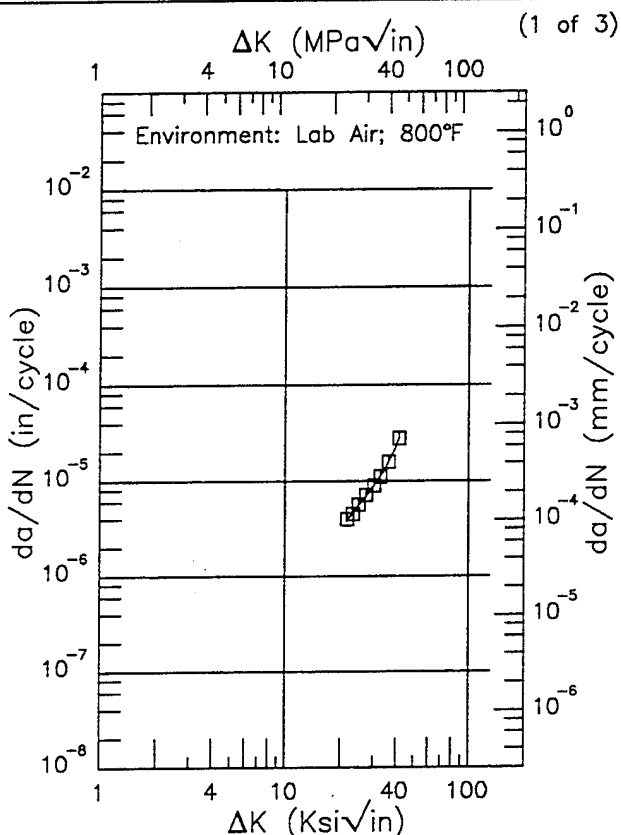
Figure 5.13.3.1.12

WASPALLOY

E

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation:
 Stress Ratio: 0.05
 Frequency: 0.2 Hz

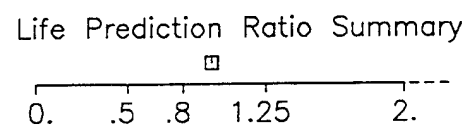
Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.301 - 0.303 in.
 Specimen Width: 0.996 - 0.998 in.
 Ref: PW001



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
21.55 (min)	3.81
25.	5.56
30.	8.66
35.	13.7
40.	23.2
41.48 (max)	27.5

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
23.34 (min)	10.8
25.	14.1
30.	19.8
35.	30.3
40.	44.7
43.07 (max)	58.1

RMS σ
 Error
 1.89



RMS σ
 Error
 1.81

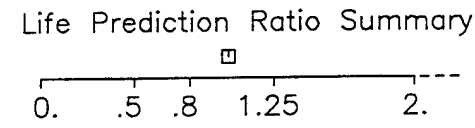


Figure 5.13.3.1.13

WASPALLOY E

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation:
 Stress Ratio: 0.05
 Frequency: 0.2 Hz

Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.301 - 0.303 in.
 Specimen Width: 0.996 - 0.998 in.
 Ref: PW001

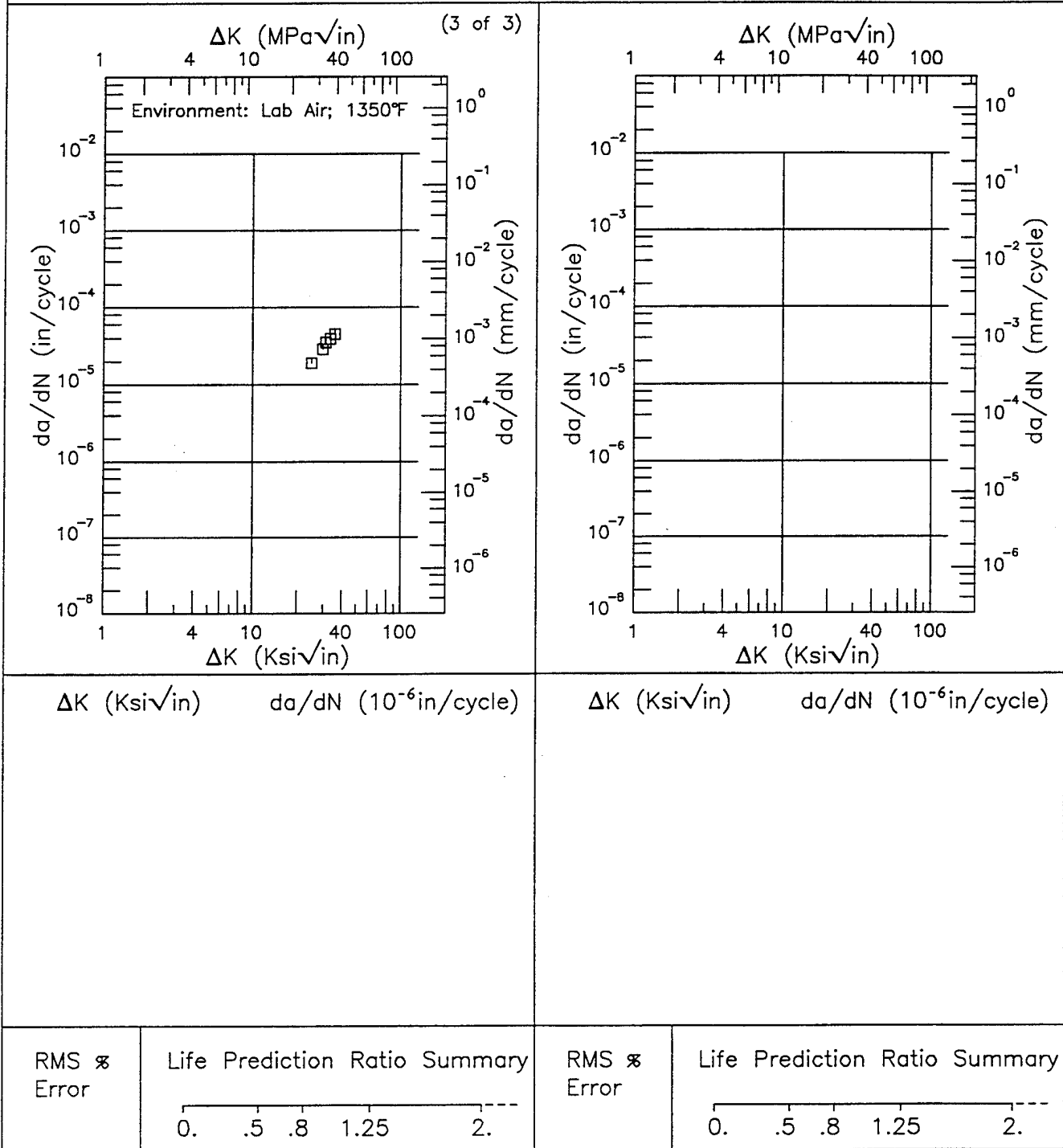
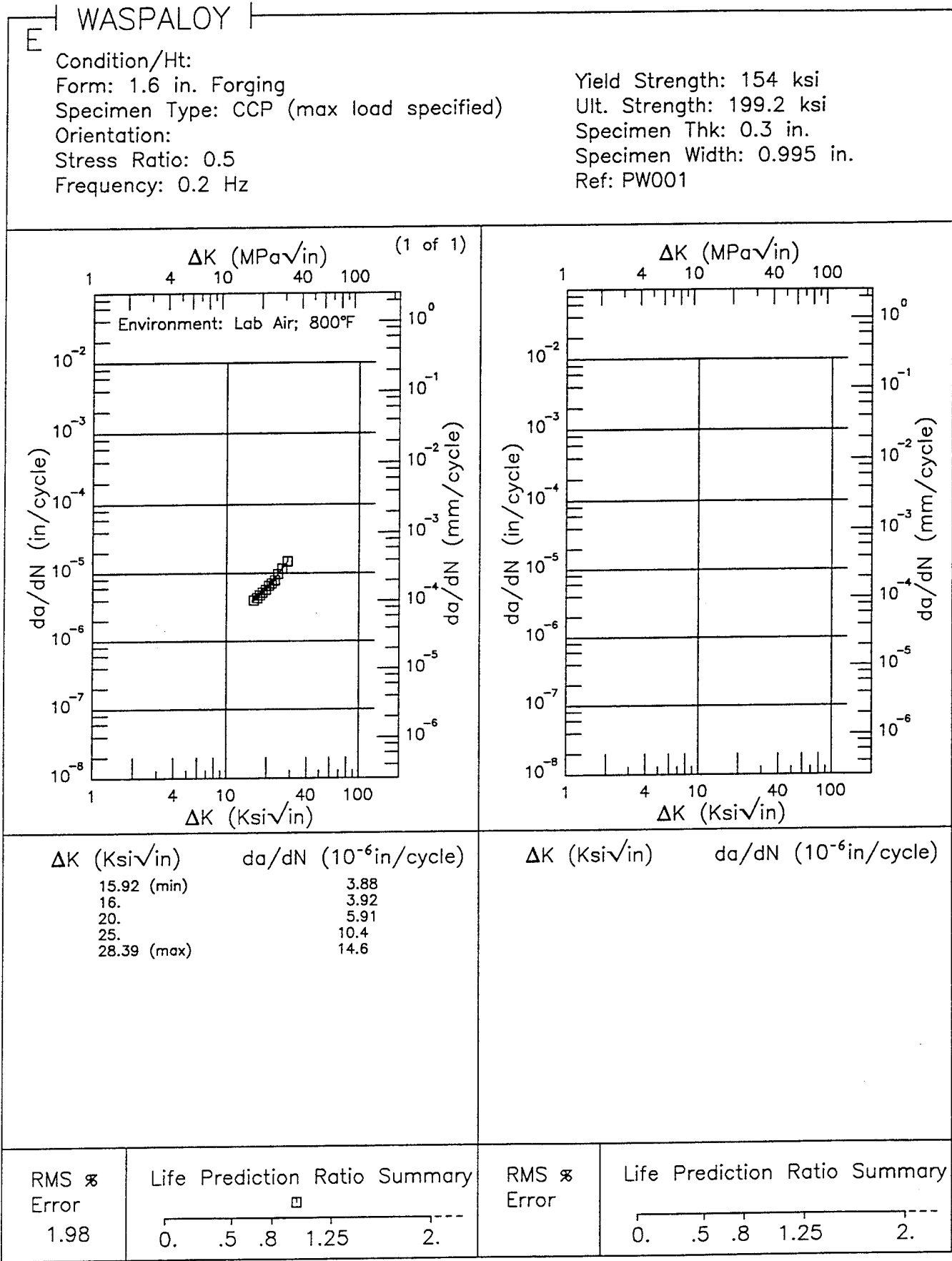


Figure 5.13.3.1.13 (Concluded)



WASPALLOY

E

Condition/Ht:
 Form: 1.6 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation:
 Stress Ratio: 0.8
 Frequency: 0.2 Hz

Yield Strength: 154 ksi
 Ult. Strength: 199.2 ksi
 Specimen Thk: 0.3 in.
 Specimen Width: 0.996 - 0.998 in.
 Ref: PW001

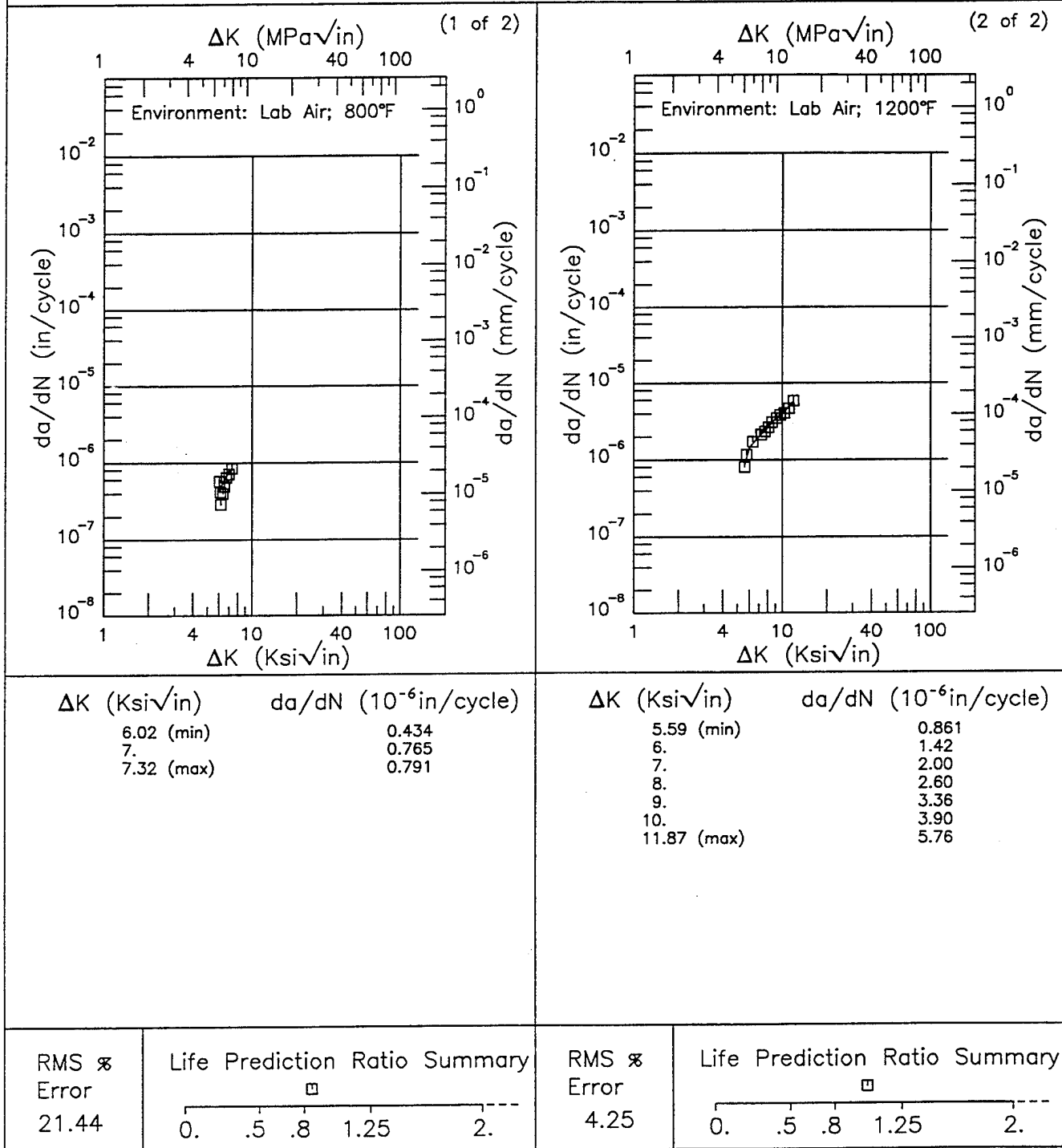


Figure 5.13.3.1.15

WASPALLOY

Condition/Ht:

Form: 1.6 in. Forging

Specimen Type: CCP (max load specified)

Orientation: C-R

Stress Ratio: 0.1

Frequency: 0. Hz

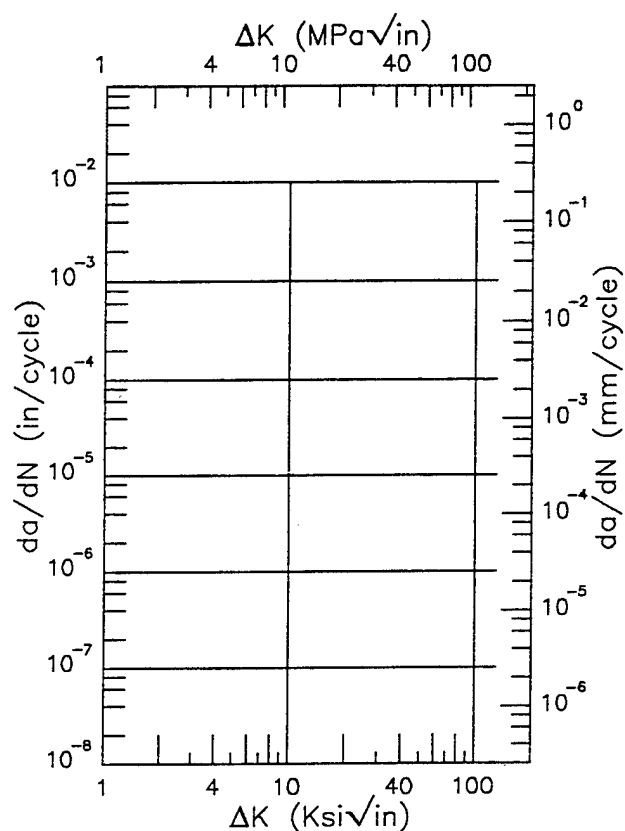
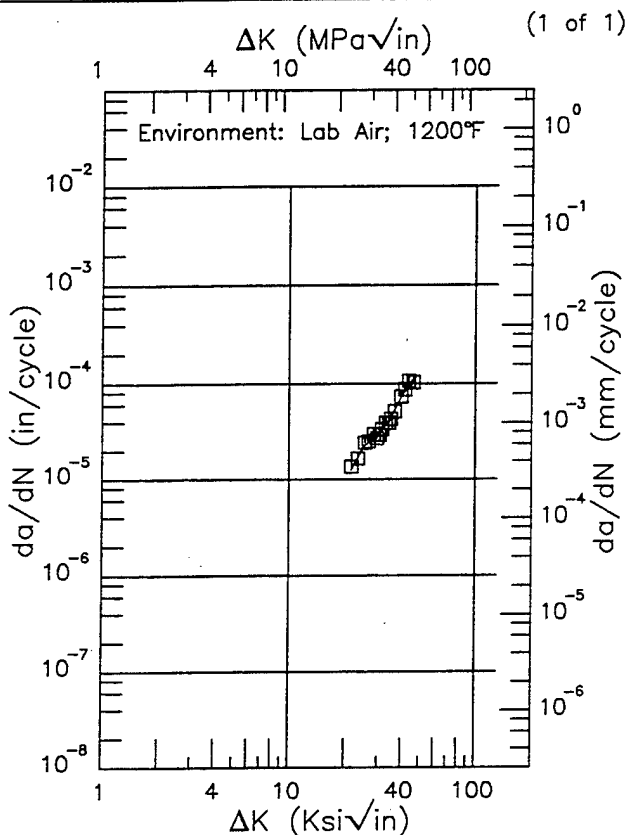
Yield Strength: 154 ksi

Ult. Strength: 199.2 ksi

Specimen Thk: 0.299 in.

Specimen Width: 0.998 in.

Ref: PW001

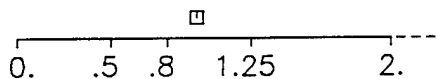


ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
21.41 (min)	12.5
25.	22.2
30.	30.0
35.	44.3
40.	75.4
45.95 (max)	109.

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS %
Error
5.56

Life Prediction Ratio Summary



RMS %
Error

Life Prediction Ratio Summary

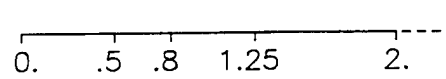


Figure 5.13.3.1.16

WASPALLOY

E

Condition/Ht: 1850F 2HRS 1350F 6HRS
 Form: 1.18 in. Billet
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.1
 Frequency: 10 Hz

Yield Strength: 145 ksi
 Ult. Strength: 249 ksi
 Specimen Thk: 0.125 in.
 Specimen Width: 1.25 in.
 Ref: UC001

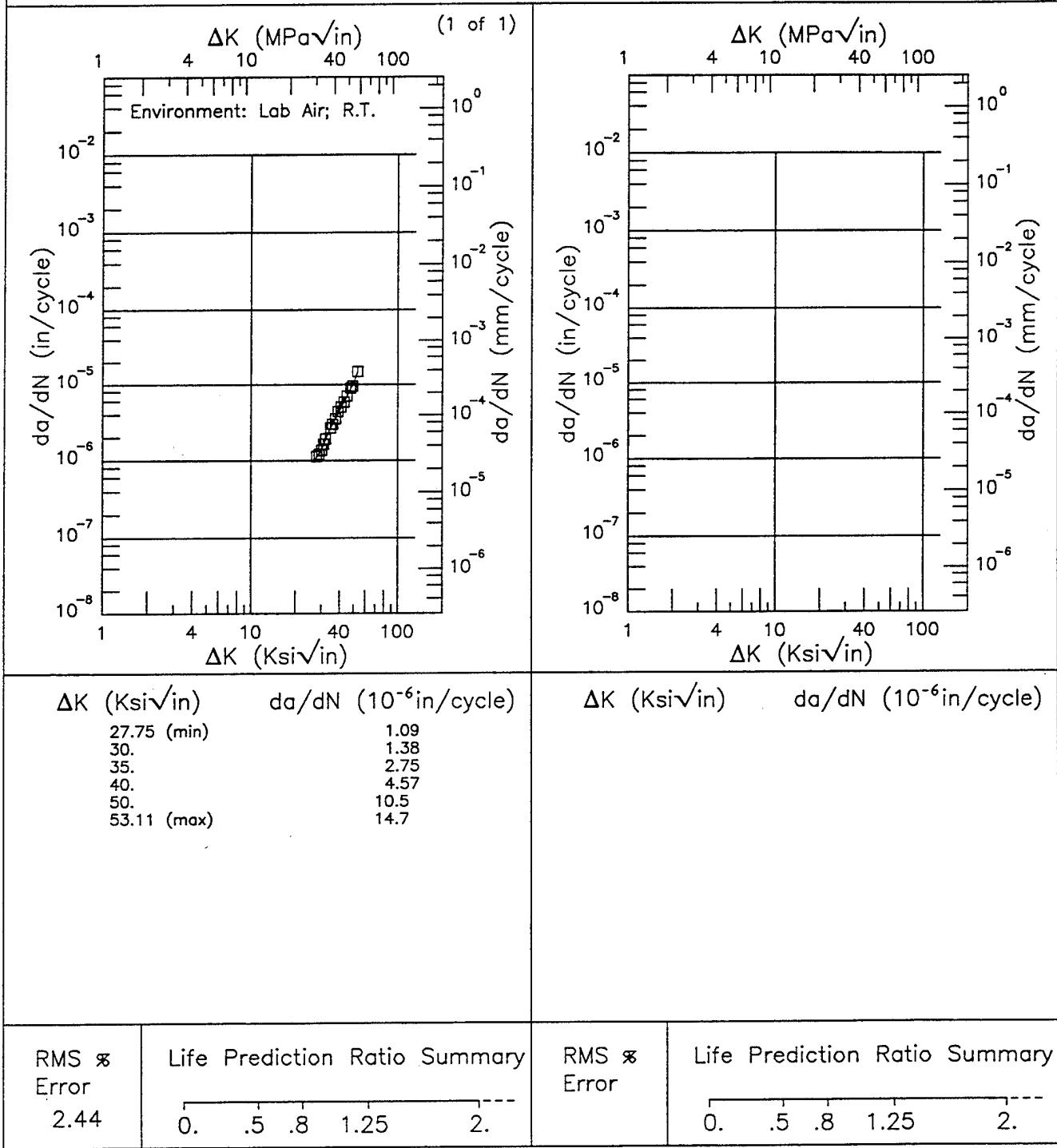
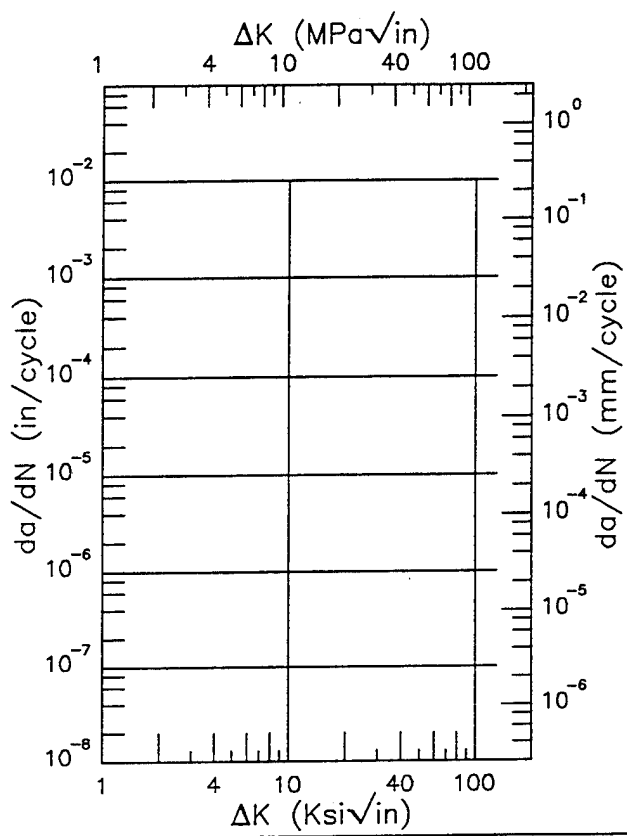
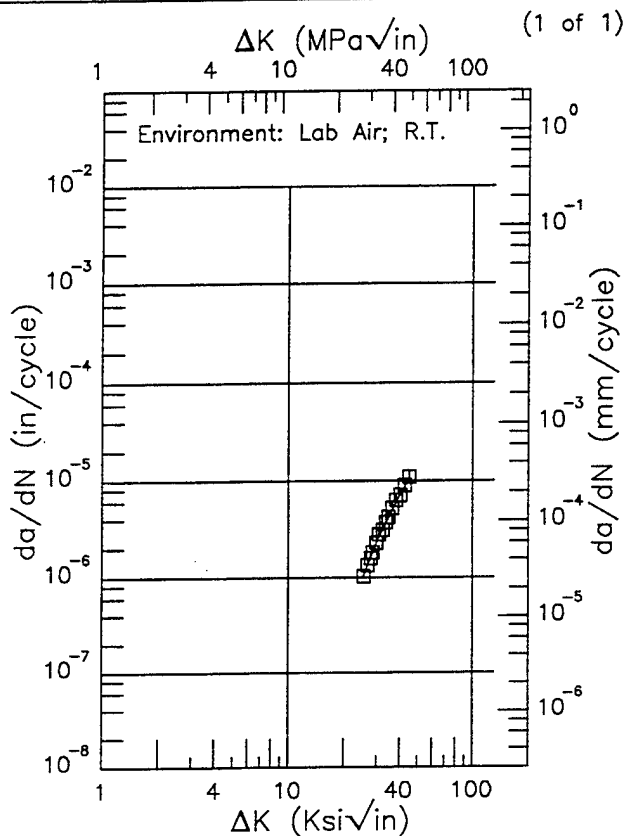


Figure 5.13.3.1.17

WASPALLOY

Condition/Ht: 1850F 2HRS 1600F 24HRS (FINE GS)
 Form: 1.18 in. Billet
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.1
 Frequency: 10 Hz

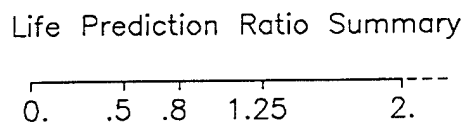
Yield Strength: 111.5 ksi
 Ult. Strength: 228 ksi
 Specimen Thk: 0.125 in.
 Specimen Width: 1.25 in.
 Ref: UC001



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
25.47 (min)	1.03
30.	2.33
35.	4.39
40.	7.13
45.09 (max)	10.9

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS \times
Error
2.60



RMS \times
Error

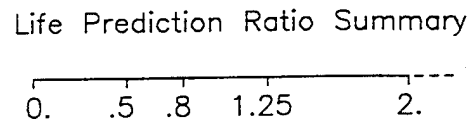


Figure 5.13.3.1.18

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F | WASPALOY |

Condition/Ht: 1875F 4HRS OQ 1550F 4HRS AC

Form: Disk

Specimen Type: WOL

Orientation: C-R

Stress Ratio: 0.05

Environment: LAB AIR;1200°F

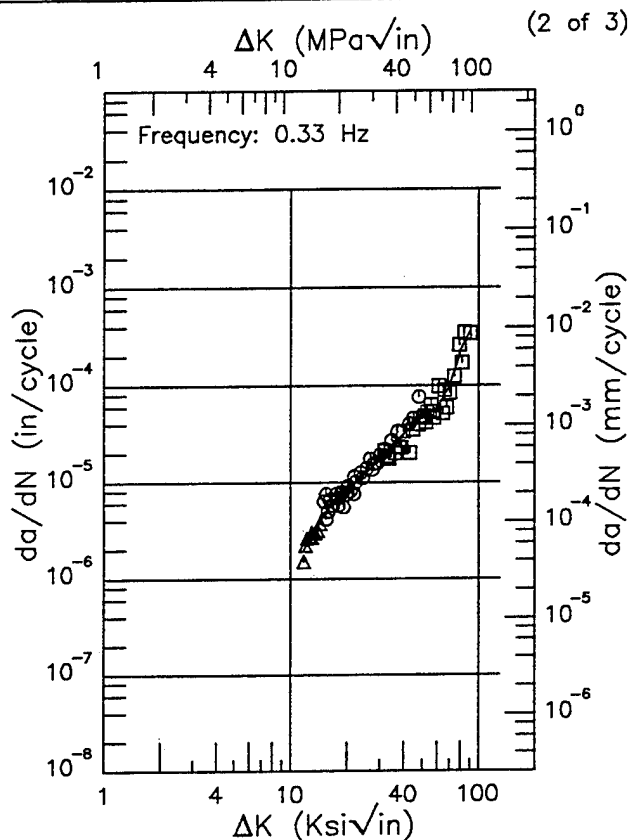
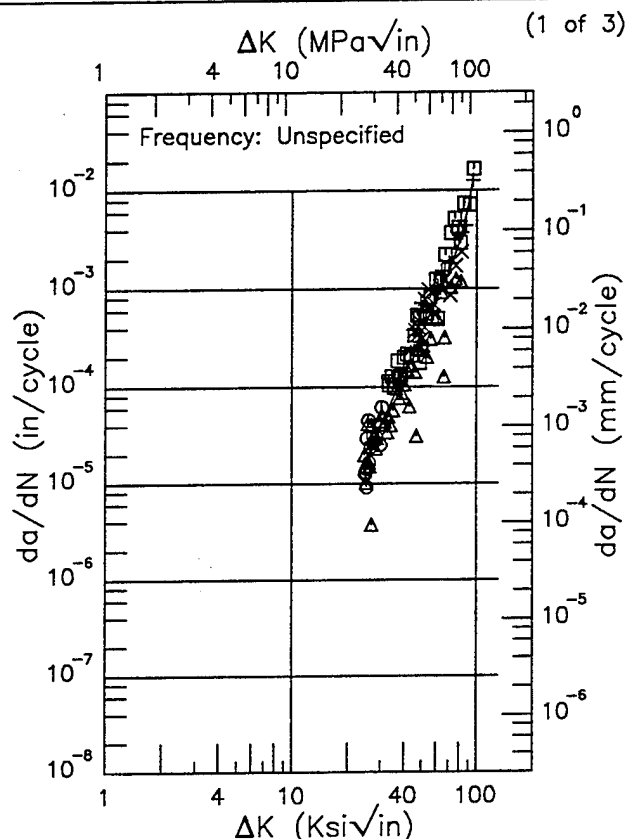
Yield Strength: 153.7 ksi

Ult. Strength: 199.7 ksi

Specimen Thk: 0.5 in.

Specimen Width: 2.5 in.

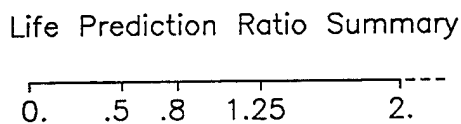
Ref: PW004



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
24.46 (min)	15.2
25.	16.7
30.	38.2
35.	77.1
40.	140.
50.	360.
60.	745.
70.	1443.
80.	2886.
90.	7179.
95.27 (max)	13261.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
11.78 (min)	1.94
13.	2.84
16.	5.27
20.	8.42
25.	12.3
30.	17.0
35.	23.5
40.	31.7
50.	47.3
60.	59.0
70.	97.1
80.	222.
89.65 (max)	360.

RMS $\%$
Error
49.61



RMS $\%$
Error
20.49

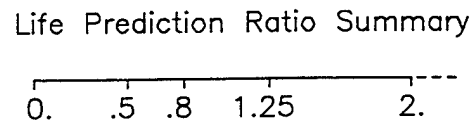


Figure 5.13.3.1.19

WASPALLOY

F

Condition/Ht: 1875F 4HRS OQ 1550F 4HRS AC
 Form: Disk
 Specimen Type: WOL
 Orientation: C-R
 Stress Ratio: 0.05
 Environment: LAB AIR;1200°F

Yield Strength: 153.7 ksi
 Ult. Strength: 199.7 ksi
 Specimen Thk: 0.5 in.
 Specimen Width: 2.5 in.
 Ref: PW004

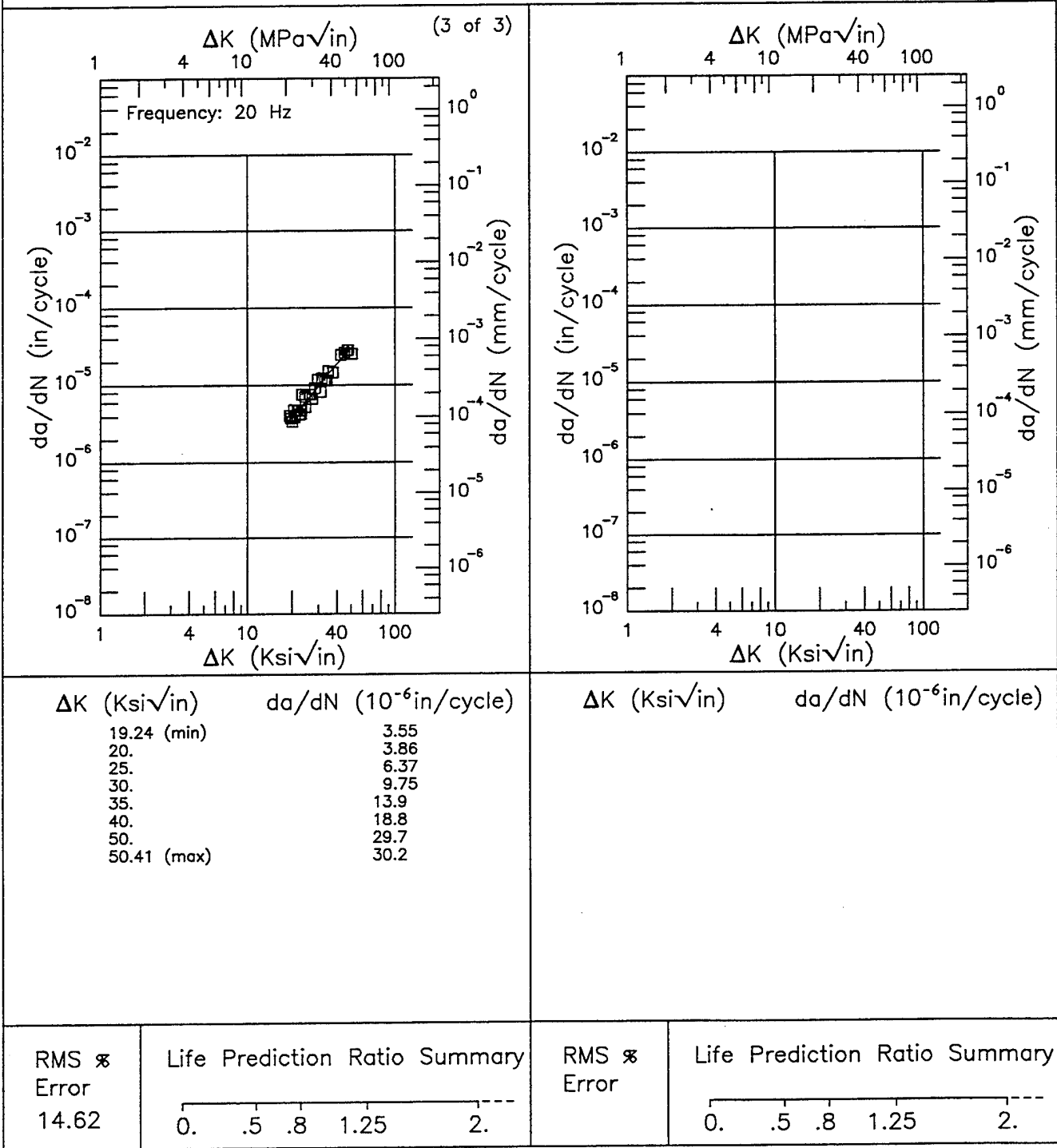
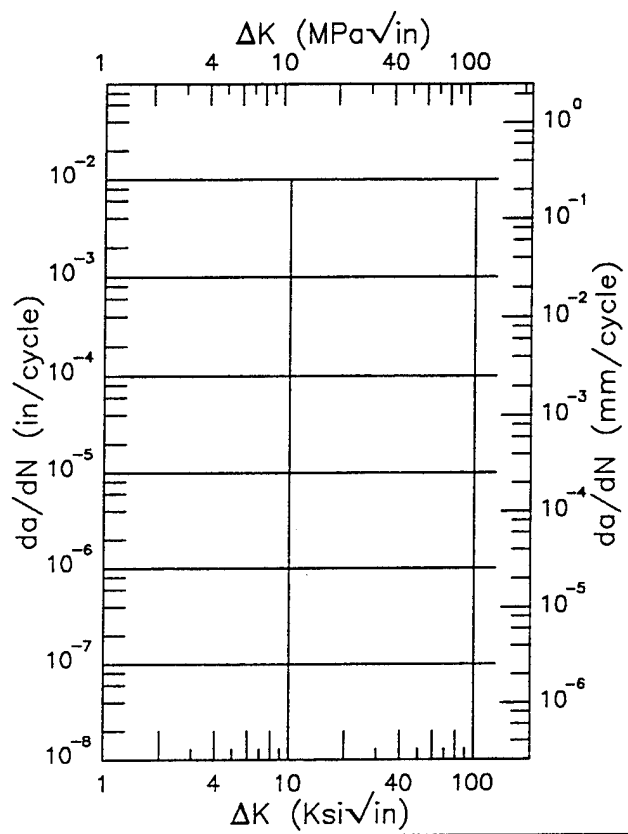
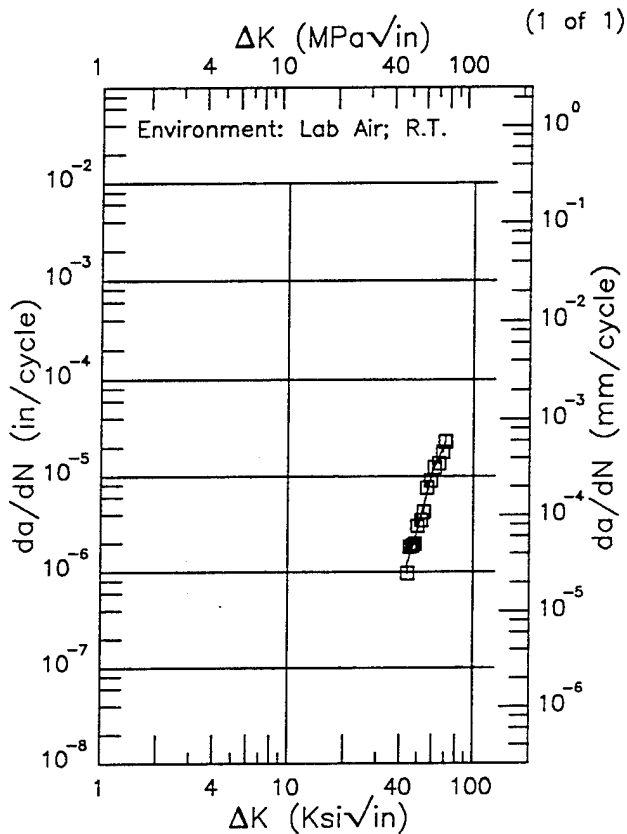


Figure 5.13.3.1.19 (Concluded)

WASPALLOY

Condition/Ht: 2010F 2HRS 1350F 6HRS
 Form: 1.18 in. Billet
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.1
 Frequency: 10 Hz

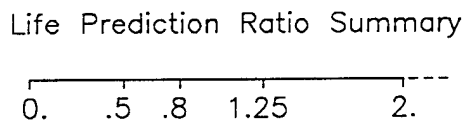
Yield Strength: 95 ksi
 Ult. Strength: 230.5 ksi
 Specimen Thk: 0.125 in.
 Specimen Width: 1.25 in.
 Ref: UC001



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
43.83 (min)	1.16
50.	2.80
60.	11.0
70.	22.0
70.45 (max)	24.0

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
---------------------	-------------------------------

RMS % Error
12.70



RMS % Error

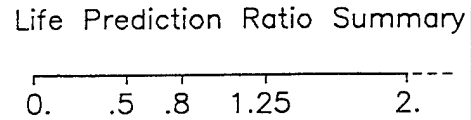


Figure 5.13.3.1.20

WASPALLOY

E

Condition/Ht: 2010F 2HRS 1600F 24HRS
 Form: 1.18 in. Billet
 Specimen Type: CT
 Orientation:
 Stress Ratio: 0.1
 Frequency: 10 Hz

Yield Strength: 96.5 ksi
 Ult. Strength: 227.5 ksi
 Specimen Thk: 0.125 in.
 Specimen Width: 1.25 in.
 Ref: UC001

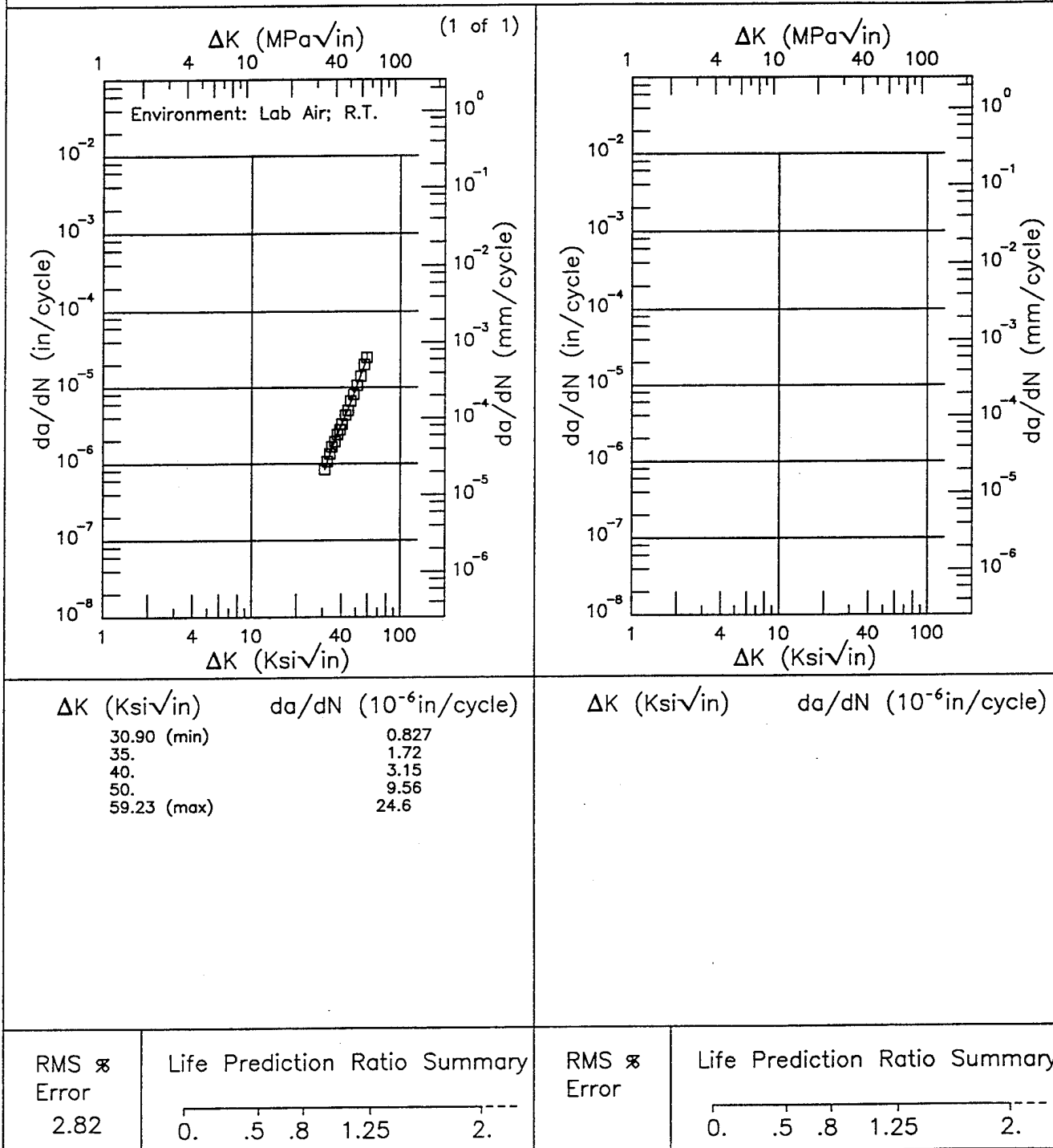


Figure 5.13.3.1.21

TABLE 5.14

REFERENCES FOR THE NICKEL BASE ALLOY DATA

- 60578 INCONEL 718 K_I
 Christian, J. L., Yang, C. T., and Witzell, W. E., "Physical and Mechanical Properties of Pressure Vessel Materials for Application in a Cryogenic Environment," ASD-TDR-62-258, Part III, General Dynamics/Astronautics (December 1964).
- 88187 INCONEL 718 K_{Ic}
 "Inconel 718 Test Data of September 19, 1973," Schultz Steel Company, South Gate, CA., September 1, 1973.
- 88579 INCONEL 718 a-vs-N; da/dN
 "B-1 Program da/dN Data for Aluminum Alloys," Rockwell International Corporation: Memorandum to H. D. Moran from E. W. Cawthorne, Battelle's Columbus Laboratories, April 3, 1974.
- 88700 INCONEL 718 K_{Isc}
 Gilbreath, W. P., and Adamson, M. J., "The Stress Corrosion Susceptibility of Several Alloys in Hydrazine Fuels," NASA Technical Note, Report NASA TN D-7604, Ames Research Center, Moffett Field, CA, February 1974.
- GE001 INCONEL 718 da/dN
 P/M RENE 95 da/dN
 RENE 95 (H&F) da/dN
 Shanini, V. and Popp, H. G., "Evaluation of Cyclic Behavior of Aircraft Turbine Disk Alloys," General Electric, Evendale, Ohio, Contract No. NAS3-20368, Report No. NASA-CR-159433, June 1978.
- GE004 P/M RENE 95 a-vs-N; da/dN
 "Argon Environment Testing," Thermal-Mechanical Crack Propagation Program; Data Sheets sent from M. S. Gilbert, General Electric Co., Evendale, Ohio, Contract No. F33615-77-C-5193, November 1980.
- GE005 INCONEL 718 da/dN
 "Fatigue Crack Growth Rate Data on Inconel 718 Using K_I Bar Specimens from the TF34 DTA Effort;" Data sent from M. S. Gilbert, General Electric Co., Evendale, Ohio, October 1982.

TABLE 5.14 (CONTINUED)

REFERENCES FOR THE NICKEL BASE ALLOY DATA

GE008	INCONEL 718 P/M RENE 95	da/dN da/dN
	Domas, P. A., "Crack Propagation Under Thermal Mechanical Cycling," General Electric Co., Aircraft Engine Group, Evendale, Ohio, Contract No. F33615-77-C-5193, November 1979.	
HD003	INCONEL 600	a-vs-N; da/dN
	James, L. A., "Fatigue Crack Propagation Behavior of Inconel 600," International Journal of Pressure Vessels and Piping, Vol. 5, 241-259. 1977.	
HD005	INCONEL 625	a-vs-N; da/dN
	James, L. A., "The Effect of Temperature upon the Fatigue Crack Propagation Behavior of Inconel 625," Report HEDL-TME 77-2, Westinghouse Hanford Co., Richland, WA., March 1977.	
HD015	INCONEL 718	a-vs-N; da/dN
	James, L. A., "Fatigue Crack Propagation Behavior of Inconel 718," Report HEDL-TME 75-80, Westinghouse Hanford Co., Richland, WA., September 1975.	
HD016	INCONEL 718	a-vs-N; da/dN
	Mills, W. J., and James, L. A., "Effect of Heat Treatment on Elevated Temperatures Fatigue-Crack Growth Behavior of Two Heats of Alloy 718," ASME Paper 78-WA/FVP-2, December 1978.	
HD017	INCONEL 718	a-vs-N; da/dN
	James, L. A., "The Effect of Product Form Upon the Fatigue-Crack Growth Behavior in Alloy 718," Journal of Engineering Materials and Technology, Vol. 103, 234-239, 1981.	
PW001	WASPALLOY	a-vs-N; da/dN
	Larsen, J. M., Schwartz, B. J., Annis, C. G. Jr., "Cumulative Damage Fracture Mechanics Under Engine Spectra," Pratt and Whitney Aircraft Group, Government Product Division, West Palm Beach, Fla., Report No. AFML-TR-77-4159, January 1980.	

TABLE 5.14 (CONCLUDED)

REFERENCES FOR THE NICKEL BASE ALLOY DATA

- PW002 IN100 a-vs-N; da/dN
- Beyer, J. R., Sims, D. L., and Wallace, R. M., "Titanium Damage Tolerant Design Data for Propulsion Systems," United Technologies Corp., Pratt and Whitney Aircraft Group, West Palm Beach, FL, Report AFML-TR-77-101, Contract No. F33615-75-C-5130, June 1977.
- PW003 ASTROLOY 901 da/dN
 IN100 da/dN
 INCOLOY 901 da/dN
- Fatigue Crack Growth Rate Data on Titanium and Nickel Base Alloys from B. S. Schwartz, Pratt and Whitney Aircraft Group, Government Product Division, West Palm Beach, FL., July 1982.
- PW004 ASTROLOY P/M-H da/dN
 ASTROLOY P/M-W da/dN
 IN100 P/M-G da/dN
 NASA IIB-7 P/M da/dN
 WASPALOY da/dN
- Cowles, B. A., Sims, D. L., Warren, J. R., "Evaluation of the Cyclic Behavior of Aircraft Turbine Disk Alloys," Pratt and Whitney Aircraft Group, United Technologies Corp., West Palm Beach, FL., Contract No. NAS3-20367, Report No. NASA CR-159409, October 1978.
- PW006 IN100 a-vs-N; da/dN
 WASPALOY da/dN
- Fatigue Crack Growth Rate Data on Nickel Base Alloys from Pratt and Whitney Aircraft Group, Government Product Division, West Palm Beach, FL., July 1982.
- RI006 INCONEL 718 K_{Isc}
- Ferguson, R. R., Berryman, R. C., "Fracture Mechanics Evaluation of B-1 Materials," Rockwell International B-1 Division, Los Angeles, CA., Contract No. F33657-70-C-0800, Report No. AFML-TR-76-137, October 1976.
- UC001 WASPALOY da/dN
- Lawless, B., et al., "The Effect of Microstructure on the FCP and Overload Behavior of Waspaloy at Room Temperature," Dept. of Materials Science and Metallurgical Engineering, University of Cincinnati, December 1980.

CHAPTER 6 TITANIUM ALLOYS SECTIONS

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6.14	Ti-6-2-4-2 ELI	6-141
6.15	Ti-6-2-4-6	6-154
6.16	Ti-6Al-4V	6-169
6.17	Ti-6Al-4V ELI	6-433
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6.20	Ti-6Al2Sn4Zr6Mo	6-516
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TABLE 6.0.1
AVAILABLE DATA FOR TITANIUM ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K _{Ic}	K _c	R Curve	da/dN	da/dt	K _{Iacc}
BETA	1745F WQ	Sheet						6
	1745F WQ +1095F 1000HR	Sheet						1
	1745F WQ +1095F 16HR	Sheet						2
	1745F WQ +1095F 250HR	Sheet						1
	1745F WQ +1095F 500HR	Sheet						1
	BETA STAB	Sheet						1
BETA-C	STA	Sheet				11		
		Plate	8					
	1325F .25HR WQ 925F 8HR	Plate	3					
	1350F 0.5HR WQ 950F 8HR AC	Extrusion	3					
	AGED 1000F 100HR	Unspecified					1	
	AGED 1250F 50HR	Unspecified					1	
BETA-III	AGED 900F 100HR	Unspecified					1	
	BETA STAB +AGED 900F 11HR	Sheet						7
	STA	Plate				4		
	STA - 1325F WQ 1045F 8HR	Plate	1					
	STA - 1325F WQ 1045F 8HR (ELECTRON BEAM WELD ZONE)	Plate	1					
	STA - 1325F WQ 1045F 8HR (HEAT AFFECTED ZONE)	Plate	1					
	STA 900F 100HR	Unspecified					1	
	STA 900F 40HR	Unspecified					1	

TABLE 6.0.1 (CONTINUED)

AVAILABLE DATA FOR TITANIUM ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K _{Ic}	K _c	R Curve	da/dN	da/dt	K _{Iacc}
BETA-III (Cont'd)	STA 900F 8HR	Unspecified					1	
	STA E.B. WELDMENT (HAZ)	Weldment				2		
	STA E.B. WELDMENT (WELD ZONE)	Weldment				1		
BETA-Ti	BETA STABILIZED	Sheet					1	
CORONA 5	ALPHA-BETA FORGED & LOW ANNEAL & AGE	Forging	1					
IMI-834	1864F 2HRS; OQ TO R.T.; AGE 1161F 2HRS	Disk				11		
Ti.*	1740F 1HR AC	Plate	3					
	STA - 1740F 1HR AC 1000F 8HR AC	Plate	3					
Ti-10-2-3	Unspecified	Unspecified				4		
Ti-4Al-3Mo-1V	MA	Plate						1
Ti-5-2.5 ELI	Unspecified	Disk				73		
Ti-5Al-2.5Sn	ANNEALED	Sheet		75		64		
Ti-6-2-2-2	ST	Plate				4		
	STA	Plate				4		
Ti-6-2-4-2	1790F 1HR AC 1100F 8HR AC	Forging				9		
Ti-6-2-4-2 ELI	ANNEAL 1450F 1HR AC	Plate				13		
	Unspecified	Extrusion				5		
Ti-6-2-4-6	1690F 2HRS AC 1550F 2HRS OQ 1100F 8HRS AC	Forging				14		
	Unspecified	Unspecified				13		1
Ti-6Al-4V	1000F 2HR	Forging						11
	1300F 1HR AC	Forging		4			2	
		Forging						

TABLE 6.0.1 (CONTINUED)

AVAILABLE DATA FOR TITANIUM ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K _{1c}	K _c	R Curve	da/dN	da/dt	K _{Isec}
Ti-6Al-4V (Cont'd)	1300F 2HR AC	Extrusion						2
	1450F 1HR AC	Plate	1					
	1550F 4HRS FC 1000F 4HRS ARGON COOLED	Forging				9		
	1700F 4HR FC TO 1400F AC DB THERMAL CYCLE	Plate						9
	1700F 6HR AC 1400F 6HR AC	Forging	12					
	1725F 1HR WQ 1000F 1HR AC (STA)	Extrusion						2
	1725F 1HR WQ 1250F 4HR AC (STOA)	Extrusion						2
	1750F 1.5HR WQ 1050F-1100F 8HR 950F 8HR	Forging						6
	1750F 1000F 2HR AC	Forging						2
	1750F 1HR FC TO 1100F	Plate	3					
	1750F 1HR FC TO RT	Plate	4					
	1750F 1HR WQ 1000F 4HR	Forging	15					
	1750F 2HR FC TO 900F AT 100F/HR AC	Forging	2					
	1750F 2HR WQ 1000F 2HR AC 1300F 2HR AC STA	Plate	2					
	1750F 4HRS ARGON COOLED	Forging				9		
	1750F WQ 1000F 8HR 1000F (ALPHA-BETA)	Forging						4
	1775F 1HR WQ 1675F 1HR WQ 1000F 4HR AC	Disk				2		
	1775F 1HR WQ 1675F 1HR WQ 1000F-1200F 2-9HR AC	Disk				17		
	1790F 1.5HR WQ 1160F 8HR + 1025F 8HR AC	Sheet						
	1900F 0.5HR AC 1350F 2HRS AC	Plate	1					2
1950F 4HRS WQ 1000F 4HRS ARGON COOLED	Forging				7			

TABLE 6.0.1 (CONTINUED)

AVAILABLE DATA FOR TITANIUM ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K _{Ic}	K _c	R Curve	da/dN	da/dt	K _{Isec}	
Ti-6Al-4V (Cont'd)	AB FORGED-MA ALPHA-BETA FORGED MA	Forging	12						
	AB FORGED-RA ALPHA-BETA FORGED RA 1700F 4HR FC	Forging	8						
	ALPHA-BETA FORGE-ANNEALED	Forging				3			
	ALPHA-BETA FORGED	Forging						3	
	ANNEALED	Sheet	Sheet		19				
		Forging	Forging	17			19		
		Extrusion	Extrusion	19			22		
		Billet	Billet	2			2		
		Billet	Billet	2					
		Forging	Forging	22					
	ANNEALED 1000F 2HR AC	Plate	2						
	ANNEALED 1300F 4HR AC	Plate	2						
	ANNEALED 1375F 3HR AC	Forging	5						
	ANNEALED 2200F 2HR	Plate				2			
	ANNEALED AT 1375F 3HRS AC	Forged Bar							
	AS RECEIVED	Plate							
	AS RECEIVED PROBABLY MA	Forged Bar							
	AS RECEIVED-AB (ALPHA-BETA FORGED)	Weldment				1			
	AS WELDED E.B. WELDMENT (HAZ)	Weldment				1			
	B FORGED BETA FORGED REHEATED TO 1950F DRAWN TO SIZE	Forged Bar		4					
B FORGED-MA BETA FORGED MA	Forging		2						
B FORGED-MA BETA FORGED MA 1300F 2HR AC	Forging		8						

TABLE 6.0.1 (CONTINUED)

AVAILABLE DATA FOR TITANIUM ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K _{Ic}	K _c	R Curve	da/dN	da/dt	K _{Isec}	
Ti-6Al-4V (Cont'd)	BA	Sheet				6			
		Plate				21			
		Forging				20			
	BB AB FIN-30MA BETA BLOCKED	Forging	2						
	BB AB FIN-MA BETA BLOCKED	Forging	4						
	BB AB FIN-RA BETA BLOCKED	Forging	4						
	BB AB FIN10STO BETA-BLOCKED	Forging	3						
	BB AB FIN36STO BETA-BLOCKED	Forging	3						
	BB B FIN-10MA BETA BLOCKED	Forging	3						
	BB B FIN10STOA BETA BLOCKED	Forging	3						
	BETA ANNEALED	Plate	6						
	BETA ANNEALED PLATE EB WELDED THEN BETA ANN	Plate	2						
	BETA FORGED	Forging						3	
	BETA PROCESSED - MA	Sheet					1		
		Plate	3				6		
		Plate	7				4		
	DB	Billet	7						
	DB + 2DHTC	Plate					2		
	DB + 4DHTC	Plate					1		
	DB + TR	Plate					1		
	DBA	Billet		22					

TABLE 6.0.1 (CONTINUED)
AVAILABLE DATA FOR TITANIUM ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K_{Ic}	K_c	R Curve	da/dN	da/dt	K_{Iacc}	
Ti-6Al-4V (Cont'd)	DBT + PC	Plate				1			
	DBTC	Plate				12			
	DBTC(RA)	Plate				1			
	EB WELD STRESS RELIEVED (HEAT AFFECTED ZONE)	Weldment					2		
	EB WELD STRESS RELIEVED (WELD ZONE)	Weldment					2		
	FINISH ROLLED 1440F	Plate						11	
	GTA WELD POSTWELD 1100F 2HR (HEAT AFFECTED ZONE)	Plate						5	
	GTA WELD POSTWELD 1100F 2HR (WELD ZONE)	Plate						1	
	GTA WELD POSTWELD 1200F 1HR (HEAT AFFECTED ZONE)	Plate						2	
	GTA WELD POSTWELD 1400F 1HR (HEAT AFFECTED ZONE)	Plate						2	
	HIP 1650F 15 KSI	Coating					3		
	MA		Unspecified				2		
			Sheet		29		9	16	13
			Plate	13	13		22	2	2
			Forging				10		
			Extrusion	11			25		
MA 10-20%ALPHA 10 TO 20%PRIMARY ALPHA MA 1300F 2HR AC	Forging		3						
MA 1300F 2HR AC	Forging		10						
	Billet		3						

TABLE 6.0.1 (CONTINUED)
AVAILABLE DATA FOR TITANIUM ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K _{1c}	K _c	R Curve	da/dN	da/dt	K _{Iacc}
Ti-6Al-4V (Cont'd)	MA 1300F 2HRS AC	Unspecified				7		
		Disk				5		
	MA 40-50%ALPHA 40 TO 50%PRIMARY ALPHA MA 1300F 2HR AC	Forging	3					
		Forging	5					
	MA COARSE GRAIN 1300F 2HR AC	Forging	6					
		Plate						1
	MA FINE GRAIN 1300F 2HR AC	Plate	46			97		21
		Forging	100			13		8
	MINUTEMAN CASING	Plate				1		
		Plate						
	RA(FAST COOLED)	Forging						
		Forging						
	SOL TREATED 1050F 4+4 HR	Plate	3					
		Forging	1					
	SOL TREATED 1050F 4HR WELDED 1050F 4HR	Plate						
Forging								
STA	Plate							
	Forging							
STOA	Plate				1			
	Forging							
STOA - 1750F 1HR WQ 1300F 2HR AC	Forging	1						
	Forging	3						
STRESS RELIEVED E.B. WELDMENT (HAZ)	Weldment				14			
STRESS RELIEVED E.B. WELDMENT (WELD ZONE)	Weldment				6			
WELDED & STRESS RELIEVED 1100F 2HRS (HAZ)	Weldment				2			

TABLE 6.0.1 (CONTINUED)

AVAILABLE DATA FOR TITANIUM ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K _{1c}	K _c	R Curve	da/dN	da/dt	K _{Isec}	
Ti-6Al-4V ELI	1800F 1HR HELIUM COOL	Plate						1	
	ANNEALED	Sheet		15					
		Forging		6		6			
	BA	Plate				36			
	RA	Plate	6			28			
	1300F 2HR	Forging					2		
	1650F 1HR WQ	Forging	8						
	1675F 2HR AC 1600F 1HR FC	Plate		2					
		Forging		2					
		Forged Bar		2					
Forging			6						
Ti-6Al-6V-2Sn	ANNEAL - COARSE GRAIN - 1350F 2HR AC	Forging	6						
	ANNEAL - FINE GRAIN - 1350F 2HR AC	Forging	6						
	ANNEALED 10-20 10-20% PRIMARY ALPHA ANNEALED 1350F 2HR AC	Forging	2						
	ANNEALED 40-60 40-50% PRIMARY ALPHA ANNEALED 1350F 2HR AC	Forging	2						
	BA	Plate				3			
	BB AB FIN-10 BETA BLOCKED ALPHA-BETA FINISHED 10% REDUCTION SOLUTION	Forging	2						
	BB AB FIN-10MA BETA BLOCKED ALPHA-BETA FINISHED 10% REDUCTION MA	Forging	3						
	BB AB FIN-30 BETA BLOCKED ALPHA-BETA FINISHED 30% REDUCTION SOLUTION	Forging	3						
	BB AB FIN-30MA BETA BLOCKED ALPHA-BETA FINISHED 30% REDUCTION MA	Forging	3						

TABLE 6.0.1 (CONTINUED)
AVAILABLE DATA FOR TITANIUM ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K _{1c}	K _c	R Curve	da/dN	da/dt	K _{Isec}	
Ti-6Al-6V-2Sn (Cont'd)	BB B FIN-10 BETA BLOCKED BETA FINISHED 10% REDUCTION SOLUTION TREATED & OVERAGED 1650F 1HR WQ 1300F 2HR AC	Forging	2						
	BETA ANNEAL 1810F 1HR ARGON COOL	Plate	3						
	BETA ANNEAL & STOA-1800F 0.5HR AC 1675F 0.5HR WQ 1050F 8HR AC	Plate	2						
	BETA ANNEALED	Plate	1						
	BF AB FOR-ANN BETA FLECTED ALPHA-BETA FORGED ANNEALED 1350F 2HR AC	Forging	3						
	BF B FOR-ANN BETA FLECTED BETA FORGED ANNEALED 1350F 2HR AC	Forging	3						
	BF LAB FOR-ANN BETA FLECTED LOW ALPHA-BETA FORGED (1600F) ANNEALED 1350F 2HR	Forging	3						
	BF LAB FOR-ANN BETA FLECTED LOW ALPHA-BETA FORGED (1600F) ANNEALED 1350F 2HR AC	Forging	3						
	DUPLEX ANNEAL	Plate	3						
	MA	Unspecified	Unspecified				1		
		Plate	Plate	6					
		Forging	Forging	8			10		
		Extrusion	Extrusion				2		
		Forged Bar	Forged Bar	1					
Billet		Billet	4						
MA 1000F 2HR AC	Billet	2							

TABLE 6.0.1 (CONTINUED)

AVAILABLE DATA FOR TITANIUM ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K _{1c}	K _c	R Curve	da/dN	da/dt	K _{Isec}
Ti-6Al-6V-2Sn (Cont'd)	RA	Unspecified				2		
		Plate	1					
	STA - 1600F 0.5HR WQ 1000F 6HR AC	Forging	9					
	STA - 1650F 0.5HR WQ 1050F 24HR AC	Forging	7					
	STA - 1675F 0.25HR WQ 1100F 4HR AC	Plate	3					
	STOA	Plate				14		
	STOA - 1600F 1.5HR WQ 1250F 6HR AC	Extrusion	3					
	STOA - 1650F 1HR WQ 1300F 2HR AC	Forging	2					
		Plate	7					
	STOA - 1700F 1HR WQ 1400F 1HR AC	Billet	8					
Ti-6Al-6V-2.5Sn	Unspecified	Plate						1
	1000F 2HR AC	Forging						1
	1300F 2HR AC	Forging						1
	1550F 1HR WQ 900F 4HR AC	Plate						1
	50% PRIMARY ALPHA	Forging	1					
	BETA PROCESSED	Forging	1					
Ti-6Al2Sn4Zr6Mo	BU B FIN-10MA BETA UPSET BETA FINISHED 10% PRIMARY ALPHA MA 1300F 1HR AC	Forging	3					
	BU B FIN-10STA BETA UPSET BETA FINISHED 10% PRIMARY ALPHA SOLUTION TREATED &	Forging	3					
	BU B FIN-10STO BETA UPSET BETA FINISHED 10% PRIMARY ALPHA SOLUTION TREATED & OVERAGED 1625F 1HR AC 1300F 1HR AC	Forging	2					

TABLE 6.0.1 (CONTINUED)

AVAILABLE DATA FOR TITANIUM ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K _{Ic}	K _c	R Curve	da/dN	da/dt	K _{Isec}	
Ti-6Al2Sn4Zr6Mo (Cont'd)	BU B FIN-50MA BETA UPSET BETA FINISHED 50% PRIMARY ALPHA MA 1300F 1HR AC	Forging	2						
	BU B FIN-50STA BETA UPSET BETA FINISHED 50% PRIMARY ALPHA SOLUTION TREATED & AGED 1625F 1HR AC 1100F 8HR AC	Forging	2						
	BU HABFIN10STA BETA UPSET HI ALPHA-BETA FINISHED 10% REDUCTION SOLUTION	Forging	3						
	BU HABFIN30STA BETA UPSET HI ALPHA-BETA FINISHED 30% REDUCTION SOLUTION TREATED	Forging	3						
	BU LABFIN10STA BETA UPSET LO ALPHA-BETA FINISHED 10% REDUCTION SOLUTION TREA	Forging	3						
	STA - 1625F 2HR AC 1100F 8HR AC	Forging	6						
	Unspecified		Unspecified					6	
			Sheet				2	1	1
	Ti-6Al-1Mo-1V	Unspecified	Plate						1
			Plate					12	53
1675F 1HR AC 1075F 8HR AC 1000F 2HR AC		Plate						1	
1700F 1HR AC 1200F 2HR WQ		Plate						1	
1725F FC 1200F 3HR WQ		Plate					8		
1775F 0.5HR FC TO 1200F 1200F 0.5HR AC 1200F 3HR ARGON QUENCH		Plate						7	
1825F 1HR AC		Plate						1	
1825F 1HR AC 1350F 2HRS AC		Unspecified				4			
1830F 1HR WQ 1100F 8HRS AC	Forging	1			3				
2000F 0.5HR AC	Plate						1		

TABLE 6.0.1 (CONCLUDED)

AVAILABLE DATA FOR TITANIUM ALLOYS

Alloy	Condition/ Heat Treatment	Product Form	K_{Ic}	K_c	R Curve	da/dN	da/dt	K_{Iacc}
Ti-9Al-1Mo-1V (Cont'd)	DA	Sheet		25		22		
	MA	Sheet				2	1	
		Plate					2	8
	MA 1435F 8HR FC	Sheet						2
Ti-6Al-2VFe3Al	VAC ANNEALED	Plate						8
	1475F 1.5HR WQ 1000F 8HR AC	Extrusion	3					
	STA REAGED AT 1100F 6HR	Plate	6					
	ANNEALED	Forging	13					
Ti6Al2.5Sn ELI	ANNEALED (ES)	Forging	6					
	ANNEALED (IS)	Forging	5					
Ti6Al6V2Sn ELI	1600F 1HR WQ 1050F 4HR AC	Plate	6					
	1650F 1HR WQ 1125F 4HR AC	Plate	6					

TABLE 6.0.2

**PLANE STRAIN FRACTURE TOUGHNESS VALUES OF TITANIUM ALLOYS
AT ROOM TEMPERATURE**

Alloy	Condition/ Heat Treatment	Product Form	Range of Product Thickness (in.)	K_{Ic} ($Ksi\sqrt{in}$)													
				L-T						T-L						S-L	
				Min Spec Thk	n	Mean	Std Dev	Min Spec Thk	n	Mean	Std Dev	Min Spec Thk	n	Mean	Std Dev	Min Spec Thk	n
BETA-C	STA	Plate	2.50	1.00	3	44.1	1.4	43.9	0.6	1.00	2	43.9	0.6
BETA-III	1325F .25HR WQ 925F 8HR	Plate	0.80	0.75	3	49.8	1.2
TI*	1740F 1HR AC	Plate	0.62-1.00	0.63	3	61.6	1.6
	STA - 1740F 1HR AC 1000F 8HR AC	Plate	0.62	0.62	3	55.3	1.5
T-6Al-4V	1700F 6HR AC 1400F 6HR AC	Forging	1.40	1.25	6	75.9	4.2	81.2	5.8	1.28	6	81.2	5.8
	1750F 1HR FC TO 1100F	Plate	1.50	91.5	2.1	1.50	2	91.5	2.1
	1750F 1HR FC TO RT	Plate	1.50	1.50	2	71.8	3.2	91.6	1.3	1.50	2	91.6	1.3
	1750F 1HR WQ 1000F 4HR	Forging	3.00	79.3	4.9	2.00	3	79.3	4.9
	1750F 2HR WQ 1000F 2HR AC 1300F 2HR AC STA	Plate	0.62	0.63	2	41.4	2.3
T-6Al-4V	AB FORGED-MA ALPHA-BETA FORGED MA	Forging	2.25	35.4	2.7	1.00	4	35.4	2.7
	ANNEALED	Forging	1.50-3.00	1.50	4	70.8	15.9	67.3	13.6	1.49	6	67.3	13.6
		Extrusion	1.50-4.00	1.50	5	82.6	5.3	85.2	6.5	1.49	6	85.2	6.5
	Billet	6.00	1.25	2	79.6	9.6
ANNEALED 1000F 2HR AC	Billet	2.30	1.25	2	50.9	0.6
	Forging	2.30	0.75	3	58.1	1.2	62.2	3.0	0.75	3	62.2	3.0	68.1	2	68.1	1.0	

TABLE 6.0.2 (CONTINUED)

**PLANE STRAIN FRACTURE TOUGHNESS VALUES OF TITANIUM ALLOYS
AT ROOM TEMPERATURE**

Alloy	Condition/ Heat Treatment	Product Form	Range of Product Thickness (in.)	K_{Ic} ($Ksi\sqrt{in}$)																					
				L-T						T-L						S-L									
				Min Spec Thk	n	Mean	Std Dev	Min Spec Thk	n	Mean	Std Dev	Min Spec Thk	n	Mean	Std Dev										
Ti-6Al-4V (Cont'd)	ANNEALED 1875F 3HR AC	Plate	2.75	1.25	2	60.4	5.5	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---		
	AS RECEIVED	Forged Bar	¾-1.00-3.50	0.58	14	57.1	10.4	0.50	54.9	10.8	---	---	---	---	---	---	---	---	---	---	---	---	---		
	B FORGED BETA FORGED REHEATED TO 1950F DRAWN TO SIZE	Forged Bar	2.25-3.50	---	---	---	---	---	1.00	4	42.6	4.3	---	---	---	---	---	---	---	---	---	---	---	---	
	B FORGED-MA BETA FORGED MA 1800F 2HR AC	Forging	2.00	1.00	3	70.6	4.9	1.00	71.0	0.4	1.00	0.4	---	---	---	---	---	---	---	---	---	---	---	---	
	BETA PROCESSED - MA	Plate	1.00-3.00	1.50	3	94.9	4.8	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---	
	DBA	Billet	0.62-3.50	0.98	9	68.2	9.7	0.68	64.2	11.8	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	MA	Plate	1.00-2.00	1.24	3	74.4	32.6	1.00	91.6	24.4	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---
		Extrusion	1.80-4.00	1.47	5	83.5	3.1	1.50	87.5	4.1	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	MA 1300F 2HR AC	Forging	2.00-4.50	1.00	4	60.9	6.9	1.00	49.5	3.9	1.00	3	43.6	5.8	---	---	---	---	---	---	---	---	---	---	---
		Billet	2.30	1.25	3	84.0	3.4	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---
Ti-6Al-4V ELI	RA	Plate	1.00-2.50	---	22	82.8	7.8	---	80.8	10.8	---	---	---	---	---	---	---	---	---	---	---	---	---	---	
	STA	Plate	0.62	---	---	---	---	0.63	3	42.6	2.0	---	---	---	---	---	---	---	---	---	---	---	---	---	
	ANNEALED	Forging	3.00	2.00	3	83.5	1.3	2.01	84.3	0.4	---	---	---	---	---	---	---	---	---	---	---	---	---	---	
	RA	Plate	3.00	2.00	3	78.1	4.0	2.00	76.8	0.7	---	---	---	---	---	---	---	---	---	---	---	---	---	---	

TABLE 6.0.2 (CONCLUDED)

PLANE STRAIN FRACTURE TOUGHNESS VALUES OF TITANIUM ALLOYS
AT ROOM TEMPERATURE

Alloy	Condition/ Heat Treatment	Product Form	Range of Product Thickness (in.)	K_{Ic} ($Ksi\sqrt{in}$)													
				L-T						T-L						S-L	
				Min Spec Thk	n	Mean	Std Dev	Min Spec Thk	Mean	Std Dev	n	Mean	Std Dev	Min Spec Thk	n	Mean	Std Dev
Ti-6Al-6V-2Sn	BETA ANNEAL 1810F 1HR ARGON COOL	Plate	0.50	0.45	3	54.3	2.0	
	BETA ANNEAL & STOA-1800F 0.5HR AC 1575F 0.5HR WQ 1050F 8HR AC	Plate	0.62	...	50.1	1.8	
	DUPLX ANNEAL	Plate	0.50-1.00	0.50	3	65.1	2.0	
		Plate	0.50-1.00	0.49	4	35.0	5.2	
	MA	Forging	3.80	1.00	58.6	2.7	
		Billet	2.20	1.24	52.3	6.4	
	MA 1000F 2HR AC	Billet	2.20	1.25	57.1	2.2	
	STA - 1600F 0.5HR WQ 1000F 6HR AC	Forging	3.80	1.01	30.8	0.7	
	STA - 1875F 0.25HR WQ 1100F 4HR AC	Plate	1.25	0.50	3	34.1	3.8	
	STOA - 1700F 1HR WQ 1400F 1HR AC	Plate	0.38	0.38	42.9	1.3	0.38	4	46.1	3.1	
Billet		%12.00	1.02	62.8	6.9	1.02	4	57.0	3.7		
STA REAGED AT 1100F 6HR	Plate	1.00	1.00	54.0	1.0	0.99	3	53.9	1.0		
Ti-Mo8V2Fe8Al	1600F 1HR WQ 1050F 4HR AC	Plate	1.00	0.25	29.8	0.5	
	1650F 1HR WQ 1125F 4HR AC	Plate	1.00	0.25	34.0	3.5	

TABLE 6.0.3.1

**PLANE STRESS AND TRANSITIONAL FRACTURE TOUGHNESS
OF TITANIUM ALLOYS (WITHOUT BUCKLING CONSTRAINTS)**

Alloy	Condition/ Heat Treatment	Test Temp (°F)	Specimen		Yield Strength (Ksi)	K_c (K_{SI}/\sqrt{in})												
						n - Sample size						Specimen Thickness (in.)						
						0.02		0.04		0.05		0.02		0.04		0.05		
						n	μ	σ	n	μ	σ	n	μ	σ	n	μ	σ	
Ti-6Al-2.5Sn	ANNEALED	-423.	Orient	Width (in.)	203.5	6	115.6	4.9	---	---	---	---	---	---	---	---	---	---
						L-T	9	109.4	6.6	---	---	---	---	---	---	---	---	
							14	107.0	9.0	---	---	---	---	---	---	---	---	
			T-L	2	97.1	9.6	---	---	---	---	---	---	---	---	---			
				2	107.7	16.0	2	147.6	28.9	---	---	---	---	---	---			
Ti-6Al-4V	MA	-320.	Orient	Width (in.)	171.2	2	141.8	2.7	---	---	---	---	---	---	---	---	---	---
						L-T	---	---	---	---	---	---	---	---	---	---	---	
							8.0	163.3-164.3	---	---	---	---	---	---	---	---	---	---
						T-L	---	---	---	---	---	---	---	---	---	---	---	---
24.0	133.8-136.7	---	---	---	---		---	---	---	---	---	---	---	---	---	---	---	
Ti-6Al-4V ELI	ANNEALED	R.T.	Orient	Width (in.)	136.0	5	161.6	6.5	---	---	---	---	---	---	---	---	---	---
						L-T	---	---	---	---	---	---	---	---	---	---	---	
							18.0	196.4	19.9	---	---	---	---	---	---	---	---	---
Ti-8Al-1Mo-1V	DA	R.T.	Orient	Width (in.)	135.5	3	111.7	15.0	---	---	---	---	---	---	---	---	---	---
						L-T	---	---	---	---	---	---	---	---	---	---	---	
20.0	133.6	---	---	---	---		---	---	---	---	---	---	---	---	---	---	---	---

TABLE 6.0.3.2

1 of 1

**PLANE STRESS AND TRANSITIONAL FRACTURE TOUGHNESS
OF TITANIUM ALLOY (WITH BUCKLING CONSTRAINTS)**

Alloy	Condition/ Heat Treatment	Test Temp (°F)	Specimen		Yield Strength (Ksi)	K_c (K_{Ish}/\sqrt{in})											
			Orient	Width (in.)		n - Sample size			Specimen Thickness (in.)								
						μ	σ	n	μ	σ	n						
TI-6Al-4V	1300F 1HR AC	R.T.	L-T	6.0	147.1	0.400											
						n	μ	σ	n	μ	σ	n	μ	σ	n	μ	σ
						2	144.3	8.7									

TABLE 6.0.4.1

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: Unspecified STRESS RATIO: 0.1 - 0.7 FREQUENCY: 0.1 - 30. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
					ΔK Level (Ksi/in)						
					2.5	5.0	10.0	20.0	50.0	100.0	
Ti-6Al-4V	BA	SHEET	0.1	10			4.66				
			0.1	10			1.84	14.44	122.21		
			0.3	10				14.91	210.93		
			0.7	10				3.87	26.35		
	HIP 1650F 15 KSI	CASTING	0.1	0.1-20			0.16	5.23			
			0.1	30			1.33				
	Unspecified	Unspecified									

TABLE 6.0.4.2

1 of 5

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: L-T STRESS RATIO: -1.0 - 0.8 FREQUENCY: 0.1 TO 55. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
					ΔK Level (Ksi \sqrt{in})						
					2.5	5.0	10.0	20.0	50.0	100.0	
Ti-10-2-3	Unspecified	Unspecified	0.05	10							
			0.1	10			1.16				
			0.33	10			0.65				
Ti-5Al-2.5Sn	ANNEALED	SHEET	0.1	30					11.56	123.8	
			0.1	50					11.69		
			0.67	55			0.15	2.17			
Ti-6-2-4-2 ELI	ANNEAL 1450F 1HR AC	PLATE	-0.3	9					21.3		
			0.1	9					13.49		
			0.5	9					43.6		
Ti-6-2-4-6	Unspecified	EXTRUSION	0.1	20				0.82	9.45		
			0.1	30				1.02			
Ti-6Al-4V	ALPHA-BETA FORGE-ANNEALED	FORGING	0.1	30			0.05				
			0.1	30			0.05				
			-1	5				0.87	10.33		
Ti-6Al-4V	ANNEALED	FORGING	0.1	5				0.29	9.53		
			0.1	15					11.12		

TABLE 6.0.4.2 (CONTINUED)

FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
 AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE

ORIENTATION: L-T STRESS RATIO: -1.0 - 0.8 FREQUENCY: 0.1 TO 55. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
					ΔK Level (Ksi/in)						
					2.5	5.0	10.0	20.0	50.0	100.0	
Ti-6Al-4V (Cont'd)	FORGING (Cont'd)		0.1	20			0.38				
						0.05	1.32	16.33			
						0.02	1.09	17.6			
						0.07					
								0.41	7.29		
								0.34	11.63		
	ANNEALED (Cont'd)	EXTRUSION		0.1	15				8.68	242.97	
									0.18		
									0.03	0.57	16.54
									0.13	1.68	
									0.1	1.37	
									0.08	1	
									0.27	11.05	
									0.25	11.03	
ANNEALED AT 1375F 3HRS AC	PLATE		0.02	10-20							
BA	FORGING		0.02	0.1-20				2.55	105.12		

TABLE 6.0.4.2 (CONTINUED)

3 of 5

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: L-T STRESS RATIO: -1.0 - 0.8 FREQUENCY: 0.1 TO 55. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-4} in/cycle)					
					ΔK Level (Ksn/in)					
					2.5	5.0	10.0	20.0	50.0	100.0
Ti-6Al-4V (Cont'd)	BETA PROCESSED - MA	PLATE	0.1	1				0.87		
			0.5	1		1.84				
			-1	10		1.14	12.91	322.86		
			0.02	0.1-20		0.32	10.8			
			0.02	1-27		0.11				
	0.02	5-30		0.02	0.4	15.79				
	0.02	0.1-30		0.15	6.77					
	0.04	20					93.19			
	0.05	20					6.34			
	0.3	20			0.99					
	0.5	10			8.51	97.18				
	0.02	1-30	FORGING				8.18	292.11		
	0.1	10	EXTRUSION			4.64				
	0.1	1-10					9.35	189.23		
	0.1	1-20					12.2	225.55		
0.3	10				0.39	8.82				
0.55	10	Unspecified			0.84					

TABLE 6.0.4.2 (CONTINUED)

4 of 5

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: L-T STRESS RATIO: 1.0 - 0.8 FREQUENCY: 0.1 TO 55. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-4} in/cycle)						
					ΔK Level (KSI/in)						
					2.5	5.0	10.0	20.0	50.0	100.0	
Ti-6Al-4V ELI	ANNEALED	FORGING	0.1	1-10				13.27	209.7		
			-1	8		3.1					
			-1	10				23.32			
			-0.66	8			2.97	22.01			
			-0.66	10				23.96			
			-0.33	8			2.5	21.42			
			-0.33	10				20.27			
			0.	8			1.61	21.49			
			0.	10			1.71	18.61			
			0.1	10				11.54			
Ti-6Al-6V-2Sn	MA	EXTRUSION	0.55	8	0.01	0.2	5.22				
			0.02	0.1-10		0.43	8				
			0.02	20		0.05	0.65				

TABLE 6.0.4.2 (CONCLUDED)

5 of 5

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: L-T STRESS RATIO: -1.0 - 0.8 FREQUENCY: 0.1 TO 55. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
					ΔK Level (Ksi \sqrt{in})					
					2.5	5.0	10.0	20.0	50.0	100.0
Ti-6Al-1Mo-1V	1825F 1HR AC 1350F 2HRS AC	Unspecified	0.	0.33				10.22		
						0.51				
	DA	SHEET	0.1	43				7.92	238.6	
									160.57	
							2.54			
	MA	SHEET	0.1	43				7.3		
							2.28	13.54	144.35	
	Unspecified	SHEET	0.02	0.1-12						

TABLE 6.0.4.3

1 of 2

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: T-L		STRESS RATIO: 0.02 - 0.8		FREQUENCY: 0.1 - 58.3 Hz								
ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-8} in/cycle)							
					ΔK Level (Ksi \sqrt{in})							
					2.5	5.0	10.0	20.0	50.0	100.0		
Ti-5Al-2.5Sn	ANNEALED	SHEET	0.1	30				11.18	140.04			
								11.76				
						0.16	3.08					
						0.1				13.11		
						0.1			0.65	12.14		
						0.4			0.1	2.41	18.6	
Ti-6Al-4V	ANNEALED	FORGING	0.4	20				1.41	17.34			
								0.05				
								0.23				
						0.8				0.25	6.32	
						0.8				0.34	7.95	
						0.1					5.84	161.87
		EXTRUSION	0.1	15								
								0.01	0.2			
								0.06	0.84	14.05		
						0.4				0.03	0.4	13.39
						0.4				0.08	0.87	
						0.8						

TABLE 6.0.4.3 (CONCLUDED)

2 of 2

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: T-L STRESS RATIO: 0.02 - 0.8 FREQUENCY: 0.1 - 58.3 Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-8} in/cycle)					
					ΔK Level (Ksi \sqrt{in})					
					2.5	5.0	10.0	20.0	50.0	100.0
Ti-6Al-4V (Cont'd)	AS WELDED E.B. WELDMENT (HAZ)	WELDMENT	0.1	10				6.52		
	AS WELDED E.B. WELDMENT (WELD ZONE)	WELDMENT	0.1	10				5.94		
	BA	FORGING	0.02	0.1-20				1.93	94.89	
	MA	EXTRUSION	0.1	5-20				13.75	276.81	
	RA	PLATE	0.1	10				23.53		
	STRESS RELIEVED E.B. WELDMENT (HAZ)	WELDMENT	0.1	10				11.35	508.19	
Ti-6Al-4V ELI	STRESS RELIEVED E.B. WELDMENT (WELD ZONE)	WELDMENT	0.1	10				10.93		
	ANNEALED	FORGING	0.1	1-10				12.21	249.29	
	RA	PLATE	0.1	5-20				8.34	158.75	
			0.1	1-10				7.61	227.75	

TABLE 6.0.4.4

1 of 1

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: R-C STRESS RATIO: 0.05 - 0.5 FREQUENCY: 0.1 - 10. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-8} in/cycle)						
					2.5	5.0	10.0	20.0	50.0	100.0	
Ti-5-2.5 ELI	Unspecified	DISK	0.05	0.1							
								11.41			
								11.62	142.07		
								2.35	16.79		
			0.5	10			2.76	17.77			

TABLE 6.0.4.5

1 of 2

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: C-R STRESS RATIO: -1.0 - 0.7 FREQUENCY: 0.1 - 30. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-8} in/cycle)					
					ΔK Level (Ksi $\sqrt{\text{in}}$)					
					2.5	5.0	10.0	20.0	50.0	100.0
IMI-834	1864F 2HRS; OQ TO R.T.; AGE 1161F 2HRS	DISK	0.1	1			0.22	7.72		
Ti-6-2.5 ELI	Unspecified	DISK	0.05	0.1				12.29		
							1.88	19.49	261.68	106.63
								2.79	15.32	
							2.81	24.16		
Ti-6-2-4-2	1790F 1HR AC 1100F 8HR AC	FORGING	0.1	0.16				10		
							2.3			
Ti-6-2-4-6	1690F 2HRS AC 1550F 2HRS OQ 1100F 8HRS AC	FORGING	0.1	30			0.1	1.01		
							0.06	0.55		
Ti-6Al-4V	1775F 1HR WQ 1675F 1HR WQ 1000F 4HR AC	DISK	0.05	0.33-10					162.45	
								0.82	11.45	
									9.93	
							0.25	0.33	22.1	
			0.25	0.5			17.73			

TABLE 6.0.4.5 (CONCLUDED)

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)							
					ΔK Level (Ksi \sqrt{in})							
					2.5	5.0	10.0	20.0	50.0	100.0		
Ti-6Al-4V (Cont'd)	Unspecified	Unspecified	-1	0.16								
			-0.5	0.16								
			0.1	20					10.35			
			0.3	20					16.26			
			0.5	20								
			0.7	20								
Ti-6Al-1Mo-1V	1830F 1HR WQ 1100F 8HRS AC	FORGING	0.1	30								

ORIENTATION: C-R

STRESS RATIO: -1.0 - 0.7

FREQUENCY: 0.1 - 30. Hz

TABLE 6.0.4.6

1 of 1

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: C-S STRESS RATIO: 0.05 - 0.5 FREQUENCY: 0.1 - 10. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
					ΔK Level (Ksi/in)					
					2.5	5.0	10.0	20.0	50.0	100.0
Ti-5-2.5 ELI	Unspecified	DISK	0.05	10						
			0.5	0.1			0.9	7.78		101.63
			0.5	10			0.46	8.65		

TABLE 6.0.4.7

1 of 1

FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
 AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE

ORIENTATION: S-C STRESS RATIO: 0.05 - 0.5 FREQUENCY: 0.1 - 10. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
					2.5	5.0	10.0	20.0	50.0	100.0
Ti-5-2.5 ELI	Unspecified	DISK	0.05	0.1						
			0.05	10						
			0.5	0.1			1.8	15.77		
			0.5	10			3	18.45		

TABLE 6.0.4.8

1 of 1

FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
 AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE

ORIENTATION: R-S STRESS RATIO: 0.05 - 0.5 FREQUENCY: 0.1 - 10. Hz

ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
					ΔK Level (Ksi/in)					
					2.5	5.0	10.0	20.0	50.0	100.0
Ti-5-2.5 ELI	Unspecified	DISK	0.05	0.1						
						0.44	3.78			
						0.08	3.48	71.92		
			0.5	10		0.93	13.57			

TABLE 6.0.4.9

1 of 1

**FATIGUE CRACK GROWTH RATE (FCGR) COMPARISON
AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
FOR TITANIUM ALLOYS IN LAB AIR AT ROOM TEMPERATURE**

ORIENTATION: S-R		STRESS RATIO: 0.05 - 0.5		FREQUENCY: 0.1 - 10. Hz							
ALLOY	CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
					ΔK Level (Ksi/in)						
					2.5	5.0	10.0	20.0	50.0	100.0	
Ti-6-2.5 ELI	Unspecified	DISK	0.05	0.1					8.45		
			0.05	10					7.15	194.49	
			0.5	0.1				1.9	14.85		
			0.5	10				1.89	20.85		

TABLE 6.0.5

		STRESS CORROSION CRACKING THRESHOLD DATA FOR TITANIUM ALLOYS AT ROOM TEMPERATURE				K_{Icc} Ksi \sqrt{in}			
						Environment			
Alloy	Condition/ Heat Treatment	Product Form	Specimen Orientation	3.5% NaCl	Sump Tank Water	Shop Cleaning Solvent	JP-4 Jet Fuel	Methanol	
BETA	BETA STAB	Sheet	L-S	68					
Ti-6Al-3Mo-1V	MA	Plate	L-S	105					
	1300F 2HR AC	Extrusion	L-S	65					
Ti-6Al-4V	1700F 4HR FC TO 1400F AC DB THERMAL CYCLE	Plate	L-T	73					
			L-T		66				
	1725F 1HR WQ 1000F 1HR AC (STA)	Extrusion	T-L		55.2(5)	69			
			L-S	48.5(2)					
	1725F 1HR WQ 1250F 4HR AC (STOA)	Extrusion	L-S	65(2)					
			L-T	31			43.3		
	ALPHA-BETA FORGED	Forging	T-L	27					
			T-S	105					
	AS RECEIVED PROBABLY MA	Plate	T-L	34					
			T-S	82.6(11)					
FINISH ROLLED 1440F	Plate								
GTA WELD POSTWELD 1100F 2HR (HEAT AFFECTED ZONE)	Plate		L-T		74.3(3)	64			
GTA WELD POSTWELD 1100F 2HR (WELD ZONE)	Plate		L-T		93				
GTA WELD POSTWELD 1200F 1HR (HEAT AFFECTED ZONE)	Plate		L-T		66.5(2)				

TABLE 6.0.5 (CONTINUED)

STRESS CORROSION CRACKING THRESHOLD DATA FOR TITANIUM ALLOYS AT ROOM TEMPERATURE										
Alloy	Condition/ Heat Treatment	Product Form	Specimen Orientation	K_{Isc} K_{SI}/in						
				Environment	3.5% NaCl	Sump Tank Water	Shop Cleaning Solvent	JP-4 Jet Fuel	Methanol	
Ti-6Al-4V (Cont'd)	GTA WELD POSTWELD 1400F 1HR (HEAT AFFECTED ZONE)	Plate	L-T			66(2)				
	MA	Plate	L-S		32					
			T-S		67					
			T-S		55					
	RA	Plate	L-T				61.7(12)			
			T-L				59.8(6)	69(2)		
			T-L				53(2)			
			S-L				58.7(6)			
	Ti-6Al-4V ELI	1800F 1HR HELIUM COOL	Plate	T-S		84				
	Ti-6Al-6V-2.5Sn	1000F 2HR AC	Forging	L-T						30.5
1300F 2HR AC		Forging	L-T		32.4					
1550F 1HR WQ 900F 4HR AC		Plate	T-S		21					
1520F 1HR WQ		Plate	T-L						13.8	
Ti-6Al-1Mo-1V	1675F 1HR AC 1075F 8HR AC 1000F 2HR AC	Plate	T-L		26.4					
	1700F 1HR AC 1200F 2HR WQ	Plate	T-S		28					
	1925F 1HR AC	Plate	T-S		23					
	2000F 0.5HR AC	Plate	T-L		47.3					

TABLE 6.0.5 (CONCLUDED)

3 of 3

STRESS CORROSION CRACKING THRESHOLD DATA FOR TITANIUM ALLOYS AT ROOM TEMPERATURE								
Alloy	Condition/ Heat Treatment	Product Form	Specimen Orientation	K_{Lec} K_{SI}/in				
				Environment				
				3.5% NaCl	Sump Tank Water	Shop Cleaning Solvent	JP-4 Jet Fuel	Methanol
Ti-8Al-1Mo-1V (Cont'd)	MA	Plate	L-S	20				
			T-S	43.4(7)				
	MA 1435F 8HR FC	Sheet	T-S	21.6(2)				
			T-L	24.3(3)				22.7(3)
VAC ANNEALED	Plate							

BETA

(1 of 1)

TABLE 6.1.3.3

K_{Iacc} SUMMARY FOR TITANIUM ALLOY BETA

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K _Q (Ksi√in)	K _{res} (Ksi√in)	Test Time (min)	Test Date	Reference
						Design	Width (in)	Thick (in)							
1745F WQ	S	R.T.	---	---	.6M KCL -1000MV	SENT	---	---	---	---	100	>55*	---	1970	82651
					.6M KCl -750MV	SENT	---	---	---	100	28*	---	1970	82651	
					.6M KCl -500MV	SENT	---	---	---	100	22*	---	1970	82651	
					.6M KCL 0 MV	SENT	---	---	---	100	32*	---	1970	82651	
					.6M KCL +500MV	SENT	---	---	---	100	34*	---	1970	82651	
					.6M KCL +1000MV	SENT	---	---	---	100	44*	---	1970	82651	
1745F WQ +1095F 16HR	S	R.T.	---	---	.6M KCl -500MV	SENT	---	---	---	60	26*	---	1970	82651	
					.6M KCl -500MV	SENT	---	---	---	32	22*	---	1970	82651	
1745F WQ +1095F 250HR	S	R.T.	---	---	.6M KCl -500MV	SENT	---	---	---	22	16*	---	1970	82651	
					.6M KCl -500MV	SENT	---	---	---	8	8*	---	1970	82651	
1745F WQ +1095F 500HR	S	R.T.	---	---	.6M KCl -500MV	SENT	---	---	---	8	8*	---	1970	82651	
					.6M KCl -500MV	SENT	---	---	---	8	8*	---	1970	82651	
BETA STAB	S	R.T.	L-S	136	3.5% NaCl	CNT	8	0.16	0.16	72	<68*	---	1969	77456	

* specimen thickness does not meet minimum requirements of $2.5 \left(\frac{K_{Iacc}^2}{\sigma_{ys}} \right)$

TABLE 6.2.1.1

MEAN PLANE STRAIN FRACTURE TOUGHNESS
FOR TITANIUM ALLOY BETA-C AT ROOM TEMPERATURE

Product Form	Condition/Heat Treatment	K_{Ic} (ksi \sqrt{in})									
		Specimen Orientation									
		L-T		T-L		S-L					
Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n
Plate	STA	44.1	1.4	3	43.9	0.6	2

BETA C

1 of 1

TABLE 6.2.1.2.1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 BETA C AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: Dry Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
STA	SHEET	0.1	6	2.5	5.0	10.0	20.0	50.0	100.0
						2.26	12.83		

TABLE 6.2.1.2.2

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 BETA C AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: S.T.W.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
STA	SHEET	0.1	0.1-1	2.5	5.0	10.0	20.0	50.0	100.0
						3.35	20.28		

BETA C

1 of 1

TABLE 6.2.2.1

TITANIUM BETA C K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Kgf)	SPECIMEN			CRACK LENGTH (in.) A	2.5° (K _{1c} TYS) ¹ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Kgf·√in.)	K_{Ic} MEAN	STAN DEV		
STA	Plate	2.50	-65	L-T	---	1.995	1.001	CT	1.107	0.07	31.40	31.1	0.5	1974	88575 (1)
		2.50				2.002	CT	1.076	0.06	30.50	1974			88575 (1)	
		2.50				1.988	CT	1.044	0.07	31.30	1974			88575 (1)	
STA	Plate	2.50	R.T.	L-T	180.0	1.993	0.999	CT	1.042	0.15	43.80	44.1	1.4	1974	88575
		2.50				1.994	CT	1.074	0.14	42.80	1974			88575	
		2.50				1.996	CT	1.090	0.16	45.60	1974			88575	
STA	Plate	2.50	R.T.	T-L	180.0	1.995	1.002	CT	1.047	0.14	43.40	43.9	0.6	1974	88575
		2.50				1.995	CT	1.041	0.15	44.30	1974			88575	

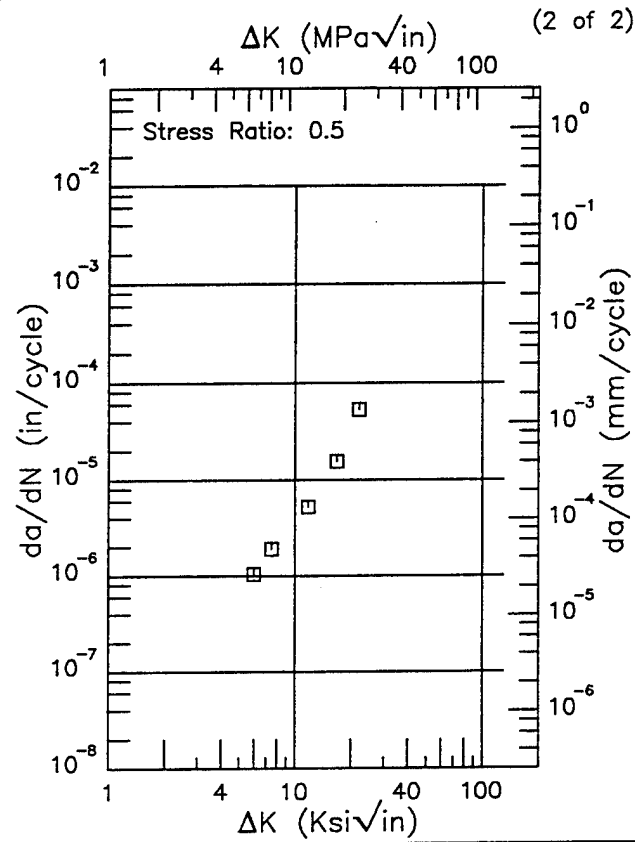
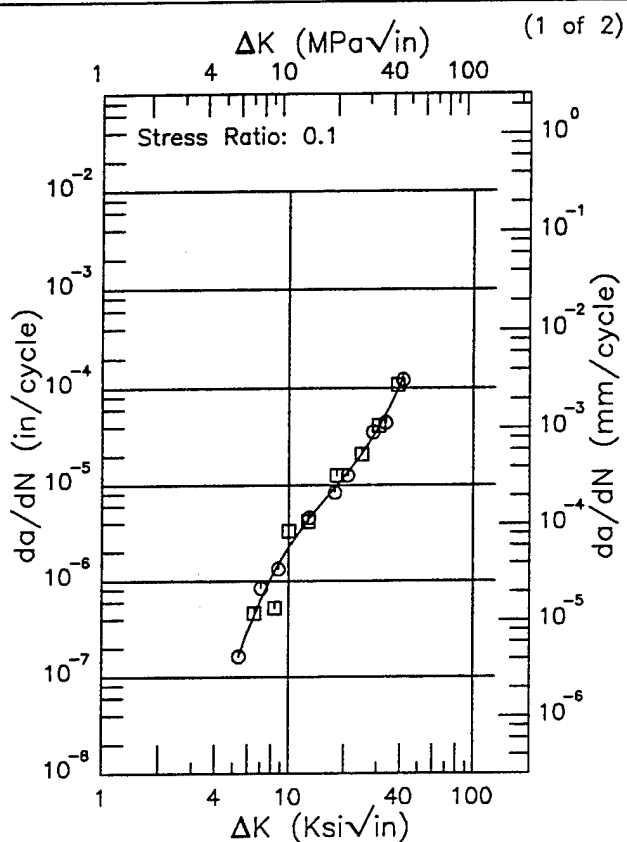
NOTES: (1) TYS APPROX. 190

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R | BETA C |

Condition/Ht: STA
 Form: 0.13 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 6 Hz
 Environment: DRY AIR; RT

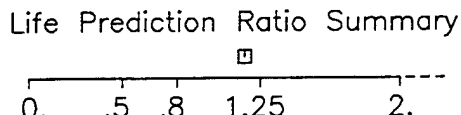
Yield Strength: 167.7 ksi
 Ult. Strength: 183.5 ksi
 Specimen Thk: 0.125 - 0.126 in.
 Specimen Width: 6.002 - 6.026 in.
 Ref: 88575



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
5.37 (min)	0.164
6.	0.301
7.	0.624
8.	1.07
9.	1.62
10.	2.26
13.	4.56
16.	7.40
20.	12.3
25.	21.6
30.	37.1
35.	63.5
40.	110.
41.22 (max)	125.

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
-------------------------------	-------------------------------

RMS $\%$
 Error
 22.15



RMS $\%$
 Error

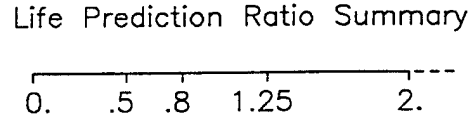
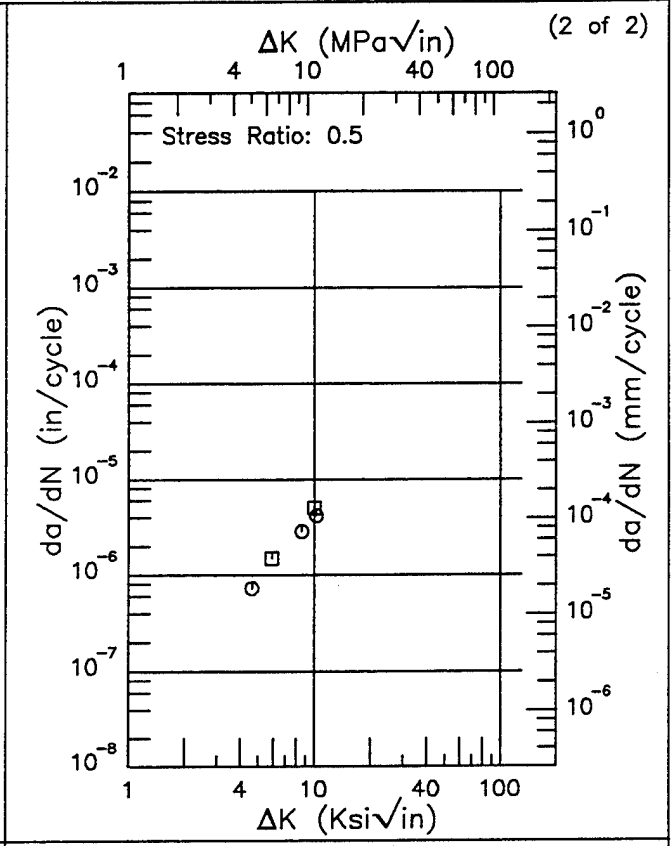
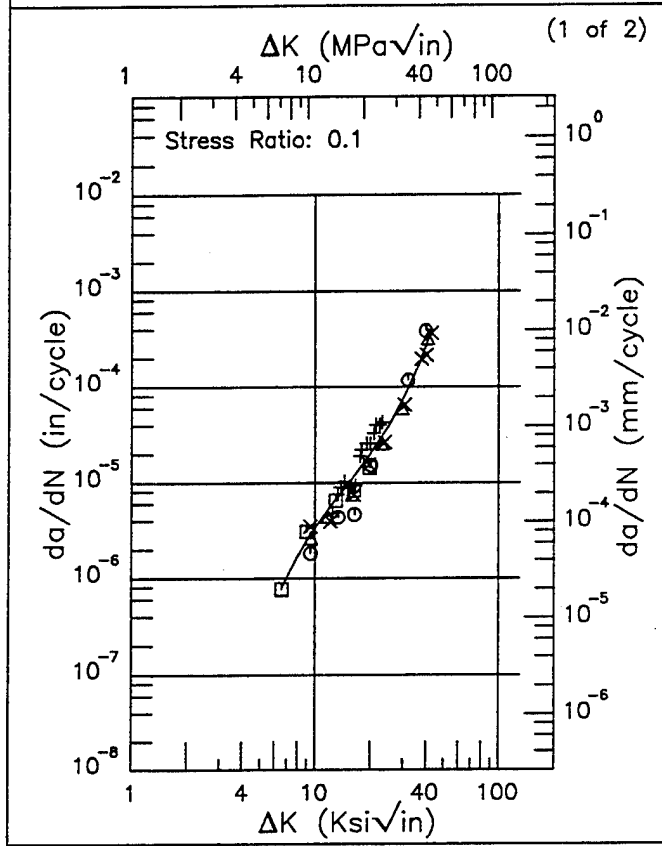


Figure 6.2.3.1.1

BETA C R

Condition/Ht: STA
 Form: 0.12 - 0.13 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 0.1 - 1 Hz
 Environment: S.T.W.; RT

Yield Strength: 167.7 ksi
 Ult. Strength: 183.5 ksi
 Specimen Thk: 0.116 - 0.127 in.
 Specimen Width: 6.002 - 6.023 in.
 Ref: 88575



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
6.58 (min)	0.805
7.	1.03
8.	1.67
9.	2.45
10.	3.35
13.	6.72
16.	11.3
20.	20.2
25.	39.4
30.	75.1
35.	143.
40.	271.
42.90 (max)	394.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
6.58 (min)	0.805
7.	1.03
8.	1.67
9.	2.45
10.	3.35
13.	6.72
16.	11.3
20.	20.2
25.	39.4
30.	75.1
35.	143.
40.	271.
42.90 (max)	394.

RMS % Error	Life Prediction Ratio Summary
29.82	

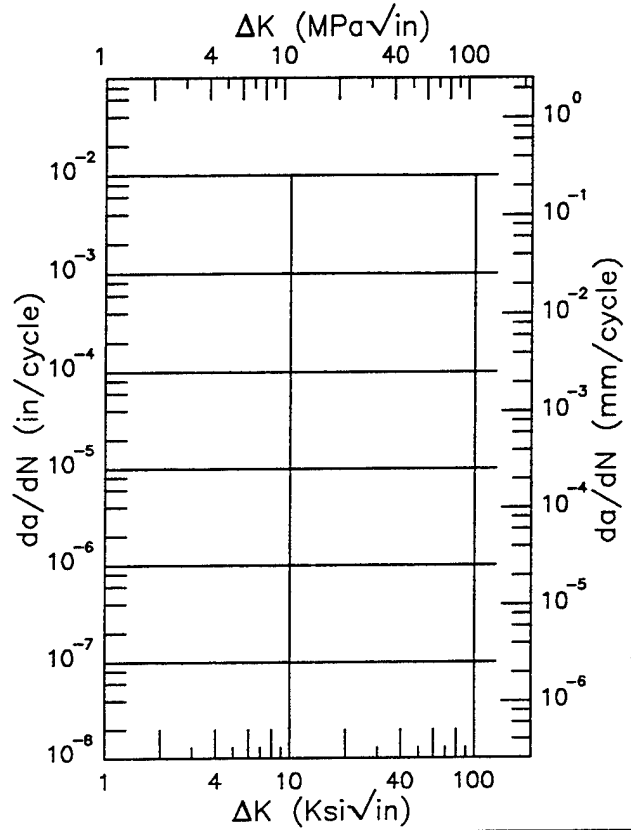
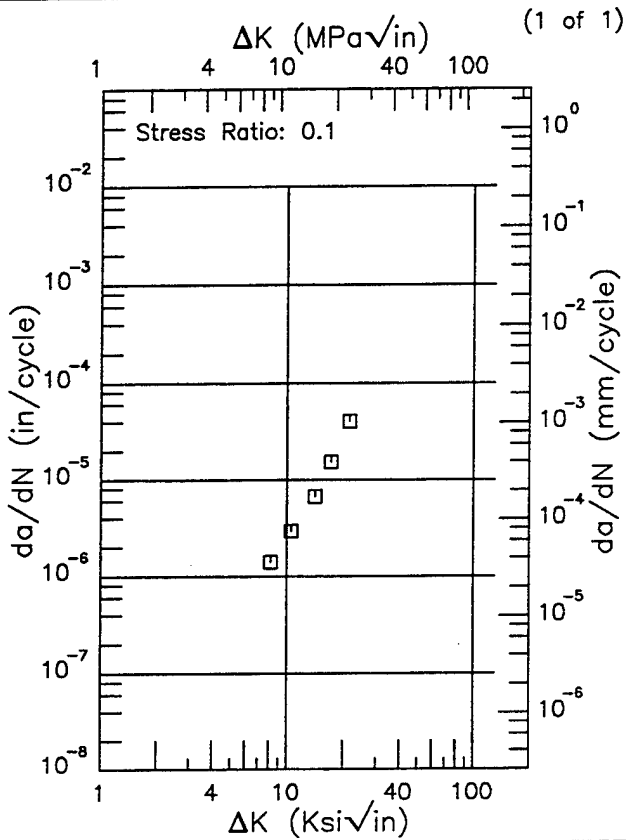
RMS % Error	Life Prediction Ratio Summary

Figure 6.2.3.1.2

R | BETA C |

Condition/Ht: STA
 Form: 0.13 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: T-L
 Frequency: 1 Hz
 Environment: S.T.W.; RT

Yield Strength: 167.7 ksi
 Ult. Strength: 183.5 ksi
 Specimen Thk: 0.127 in.
 Specimen Width: 6.009 in.
 Ref: 88575

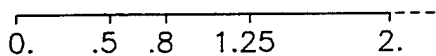


ΔK (Ksi√in) da/dN (10⁻⁶in/cycle)

ΔK (Ksi√in) da/dN (10⁻⁶in/cycle)

RMS %
Error

Life Prediction Ratio Summary



RMS %
Error

Life Prediction Ratio Summary

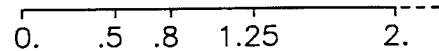


Figure 6.2.3.1.3

TABLE 6.3.1.1

MEAN PLANE STRAIN FRACTURE TOUGHNESS
FOR TITANIUM ALLOY BETA-III AT ROOM TEMPERATURE

Product Form	Condition/Heat Treatment	K_{Ic} (ksi \sqrt{in})									
		Specimen Orientation									
		L-T			T-L			S-L			
		Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	
Plate	1325F .25HR WQ 925F 8HR	49.8	1.2	3	

BETA III

1 of 1

TABLE 6.3.2.1

TITANIUM BETA III K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic}/TYS)^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Ksi $\cdot \sqrt{\text{in.}}$)	K_{Ic} MEAN	STAN DEV		
1325F .25HR WQ 825F 8HR	Plate	0.80	R.T.	L-T	186.0	1.503	0.750	CT	0.795	0.18	50.50	49.8	1.2	1974	91793
		0.80													
		0.80													
1350F 0.5 HR WQ 850F 8HR AC.	Extrusion	3.00	R.T.	C-R	178.0	1.489	0.750	CT	0.773	0.24	55.20	55.1	1.8	1973	87230 (1)
		3.00													
		3.00													
STA-1325F WQ 1045F 8 HR	Plate	1.00	R.T.	T-L	150.0	2.000	1.005	CT	0.977	0.67	85.70	--	--	1973	88144
STA-1325F WQ 1045F 8HR (ELECTION BEAM WELD ZONE)	Plate	1.00	R.T.	T-L	150.0	2.000	0.991	CT	0.917	0.17	42.20	--	--	1973	88144
STA-1325F WQ 1045F 8HR (HEAT AFFECTED ZONE)	Plate	1.00	R.T.	T-L	150.0	2.000	0.996	CT	0.930	0.55	76.30	--	--	1973	88144

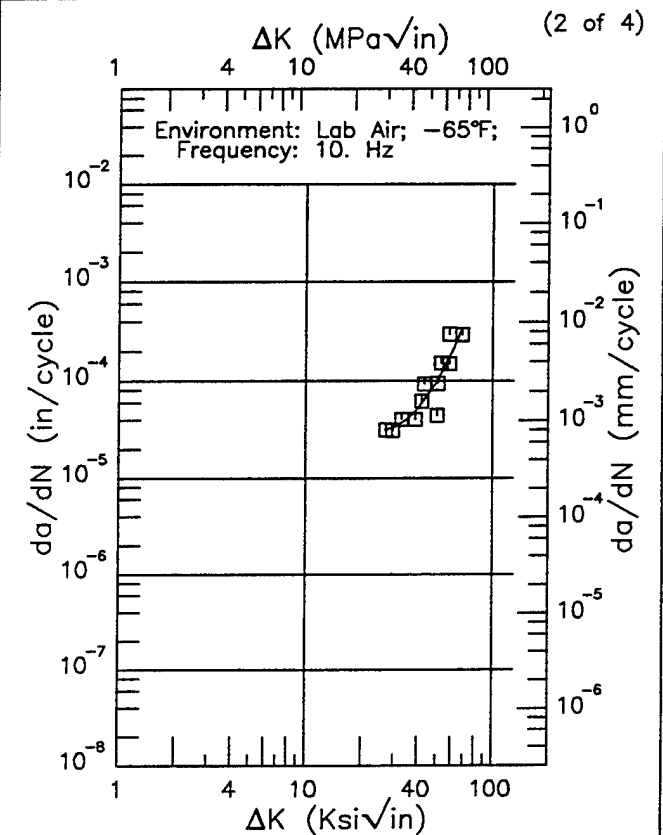
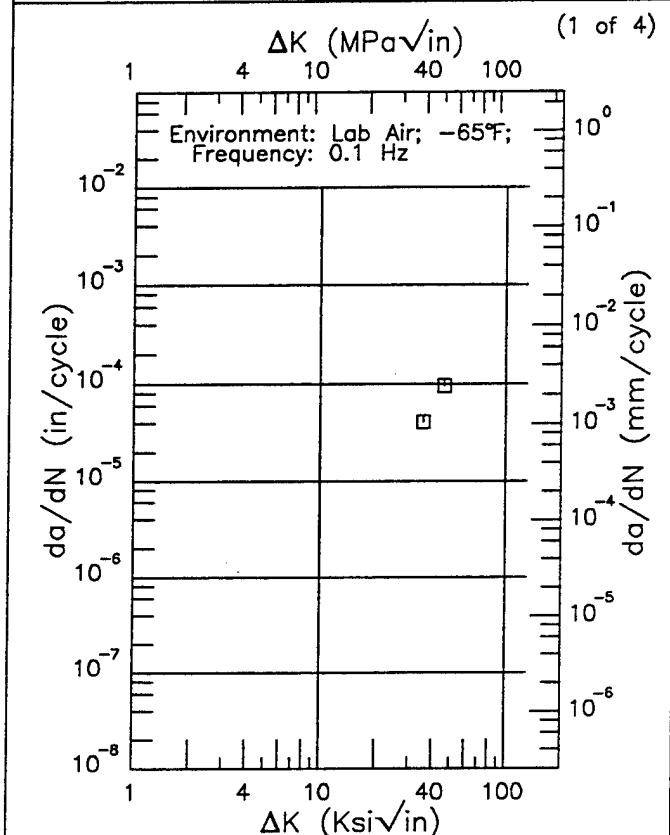
NOTES: (1) ALPHA PRECIPITATE IN BETA MATRIX STRAIGHTNESS OF CRACK FRONT MAY NOT MEET ASTM E399-72 REQUIREMENTS

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EF BETA III

Condition/Ht: STA
 Form: 1 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1

Yield Strength:
 Ult. Strength:
 Specimen Thk: 1 in.
 Specimen Width: 2.55 in.
 Ref: 88144



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
26.74 (min)	31.6
30.	33.4
35.	40.3
40.	52.2
50.	97.2
60.	190.
68.42 (max)	333.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
26.74 (min)	31.6
30.	33.4
35.	40.3
40.	52.2
50.	97.2
60.	190.
68.42 (max)	333.

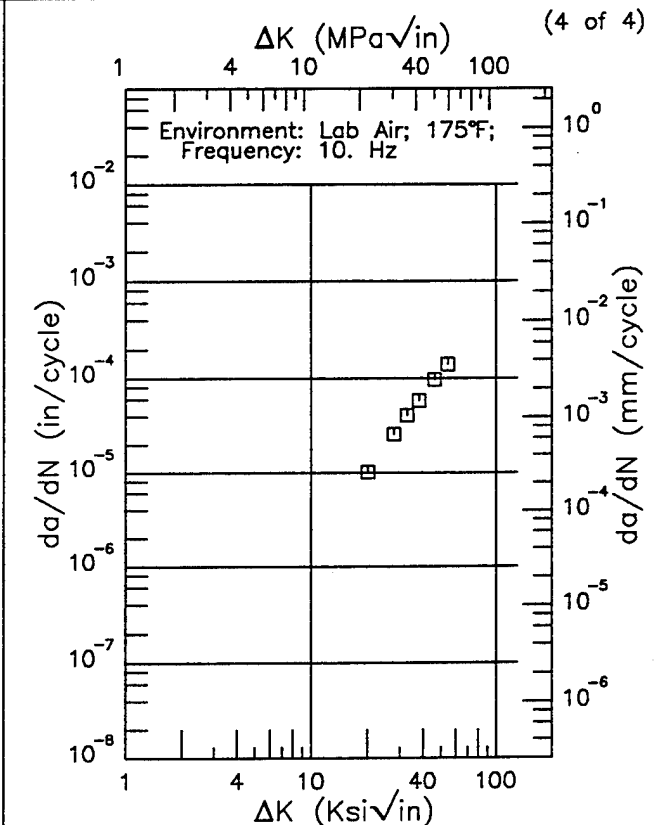
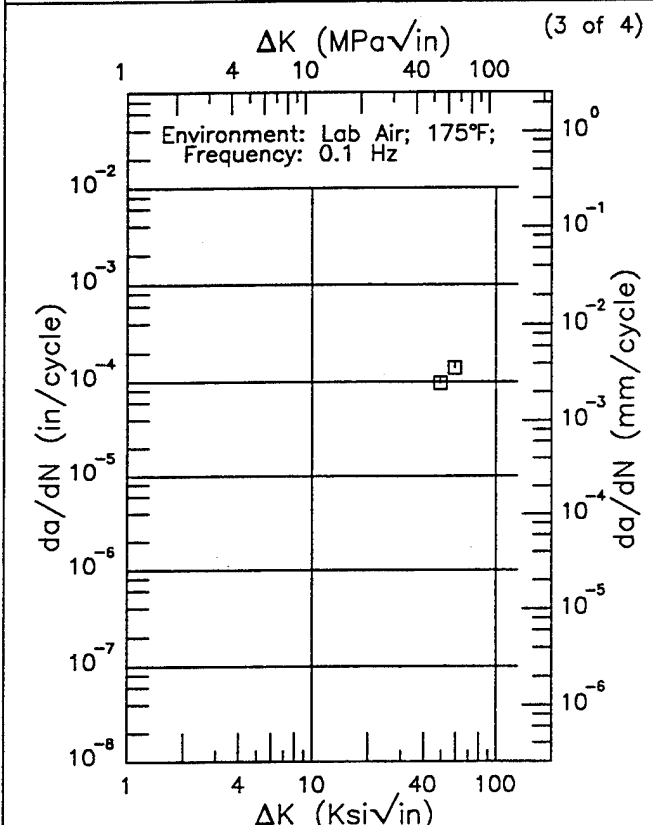
RMS % Error	Life Prediction Ratio Summary
	0. .5 .8 1.25 2.---

RMS % Error	Life Prediction Ratio Summary
31.39	0. .5 .8 1.25 2.---

Figure 6.3.3.1.1

Condition/Ht: STA
 Form: 1 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1

Yield Strength:
 Ult. Strength:
 Specimen Thk: 1 in.
 Specimen Width: 2.55 in.
 Ref: 88144



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
---------------------	-------------------------------

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
---------------------	-------------------------------

RMS % Error	Life Prediction Ratio Summary

RMS % Error	Life Prediction Ratio Summary

Figure 6.3.3.1.1 (Concluded)

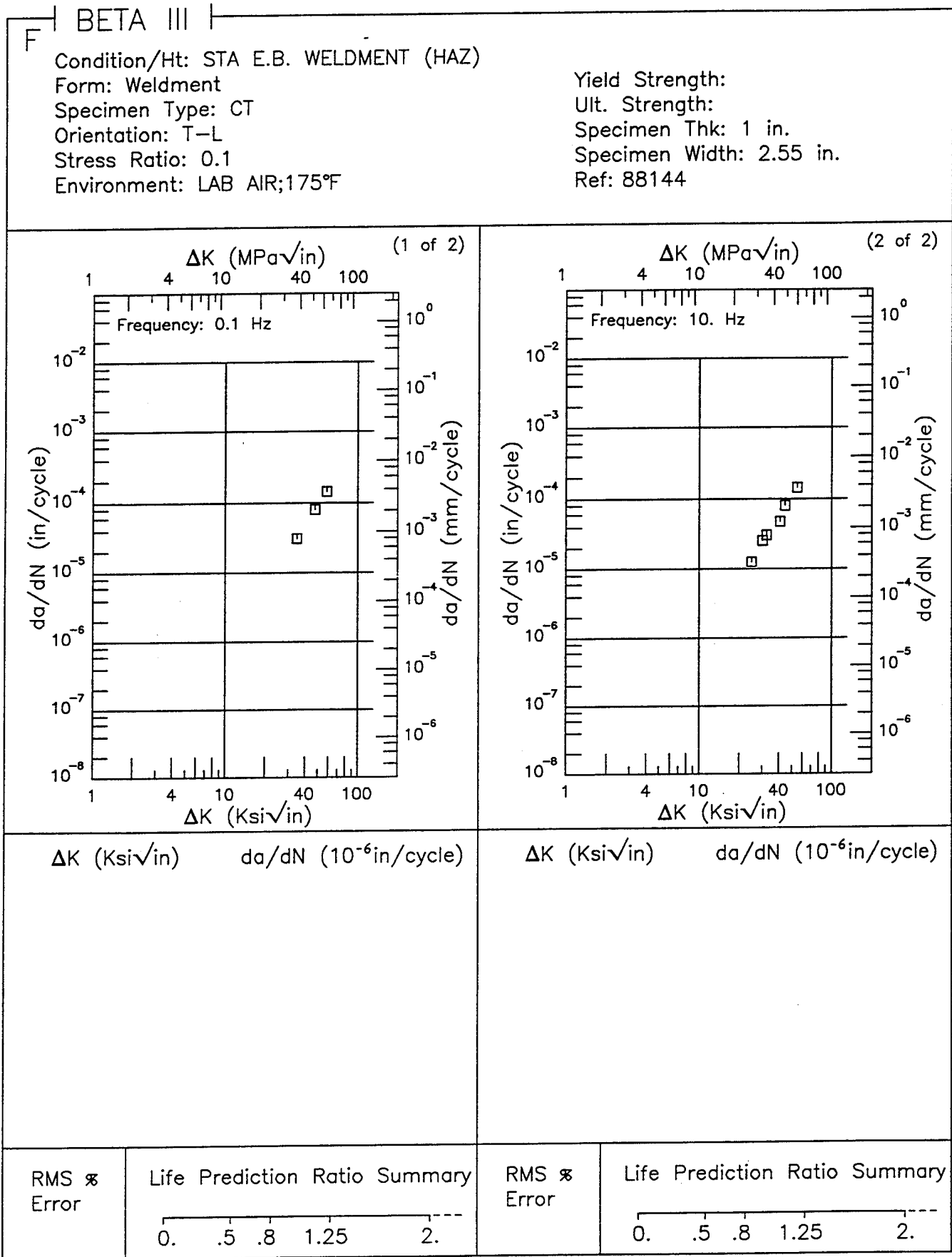


Figure 6.3.3.1.2

BETA III EF

Condition/Ht: STA E.B. WELDMENT (WELD ZONE)

Form: Weldment

Specimen Type: CT

Orientation: T-L

Stress Ratio: 0.1

Yield Strength:

Ult. Strength:

Specimen Thk: 1 in.

Specimen Width: 2.55 in.

Ref: 88144

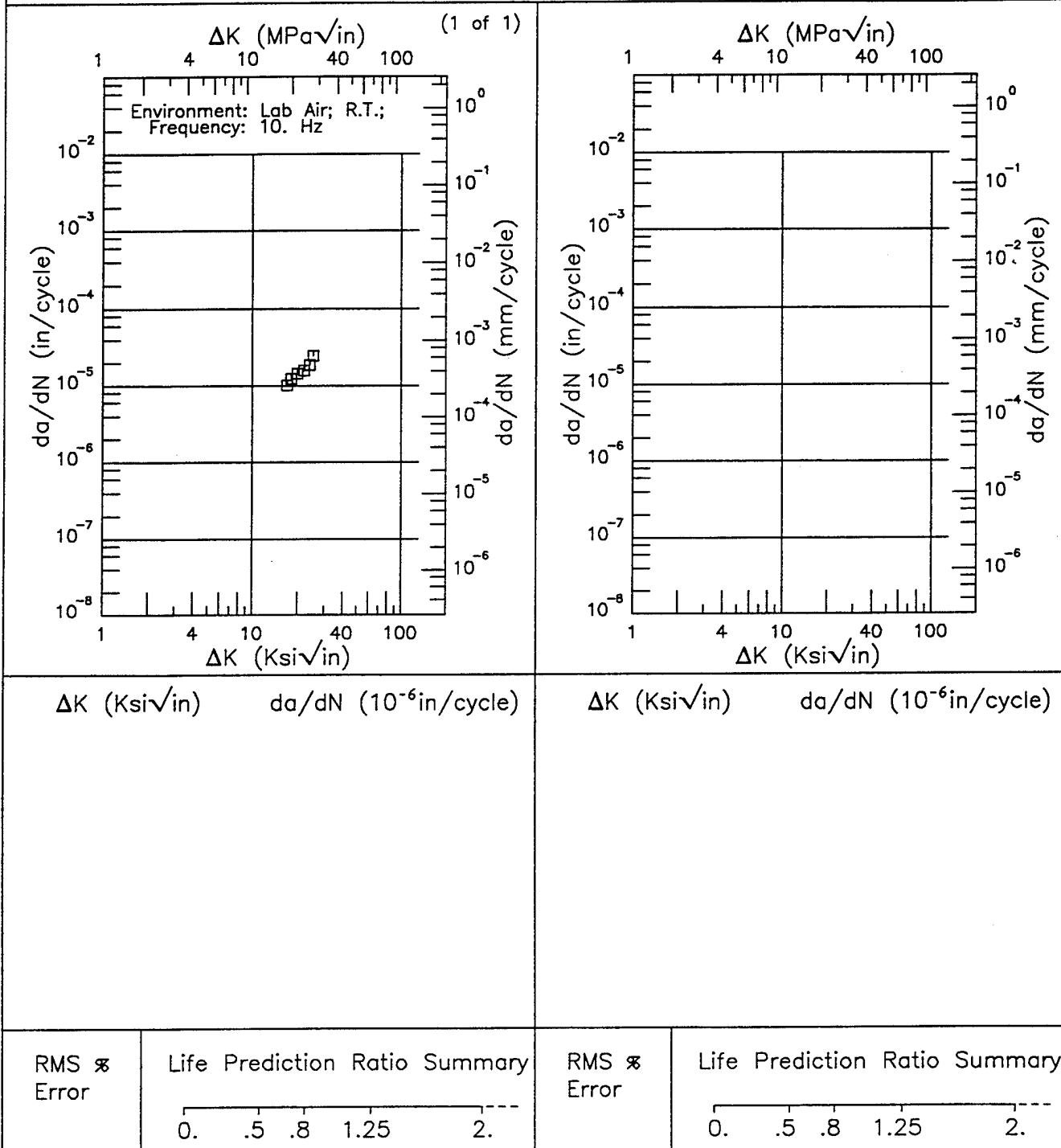
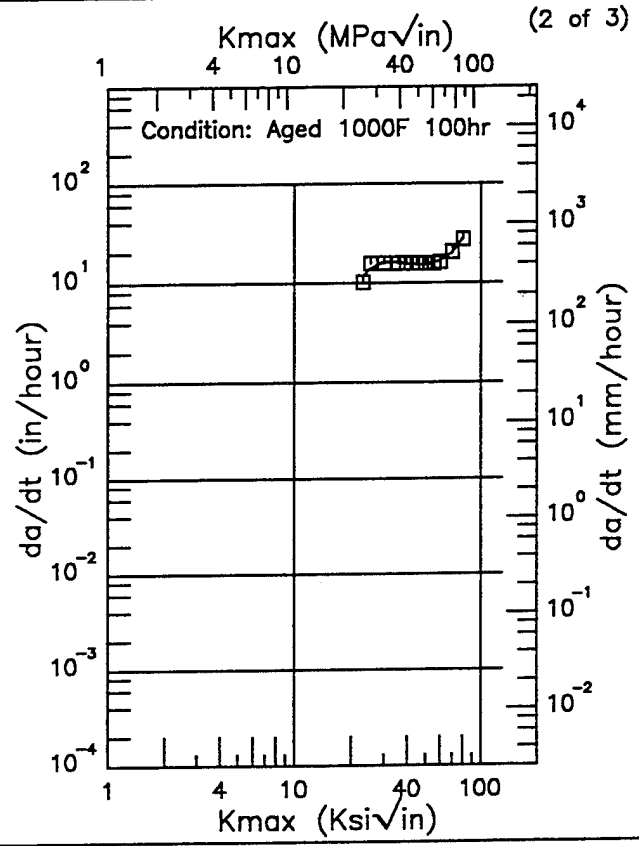
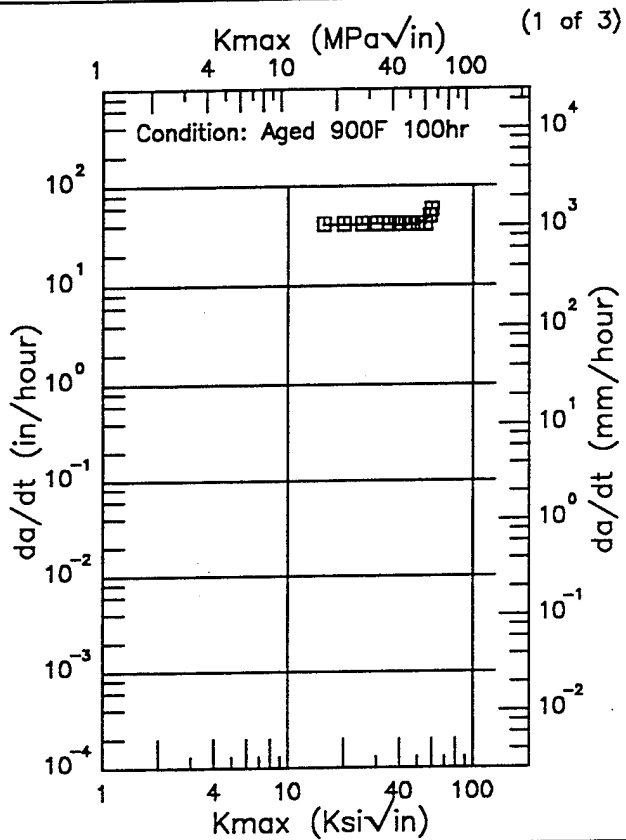


Figure 6.3.3.1.3

BETA III

Condition/Ht:
 Environment: 0.6M KCl; 75°F
 Specimen Type: SENT
 Orientation:
 Yield Strength:
 Ult. Strength:

Specimen Thk:
 Specimen Width:
 A₀:
 K_{Isc}:
 Ref: 82651



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
15.50 (min)	40679.
16.	40511.
20.	40187.
25.	40656.
30.	40950.
35.	40700.
40.	39904.
50.	40560.
59.50 (max)	49207.

Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
23.00 (min)	11248.
25.	13585.
30.	16206.
35.	16046.
40.	15283.
50.	15101.
60.	16587.
70.	19711.
80.00 (max)	27111.

RMS %
 Error
 5.76

RMS %
 Error
 5.46

Figure 6.3.3.2.1

BETA III

Condition/Ht:
 Environment: 0.6M KCl; 75°F
 Specimen Type: SENT
 Orientation:
 Yield Strength:
 Ult. Strength:

Specimen Thk:
 Specimen Width:
 A_o:
 K_{Isc}:
 Ref: 82651

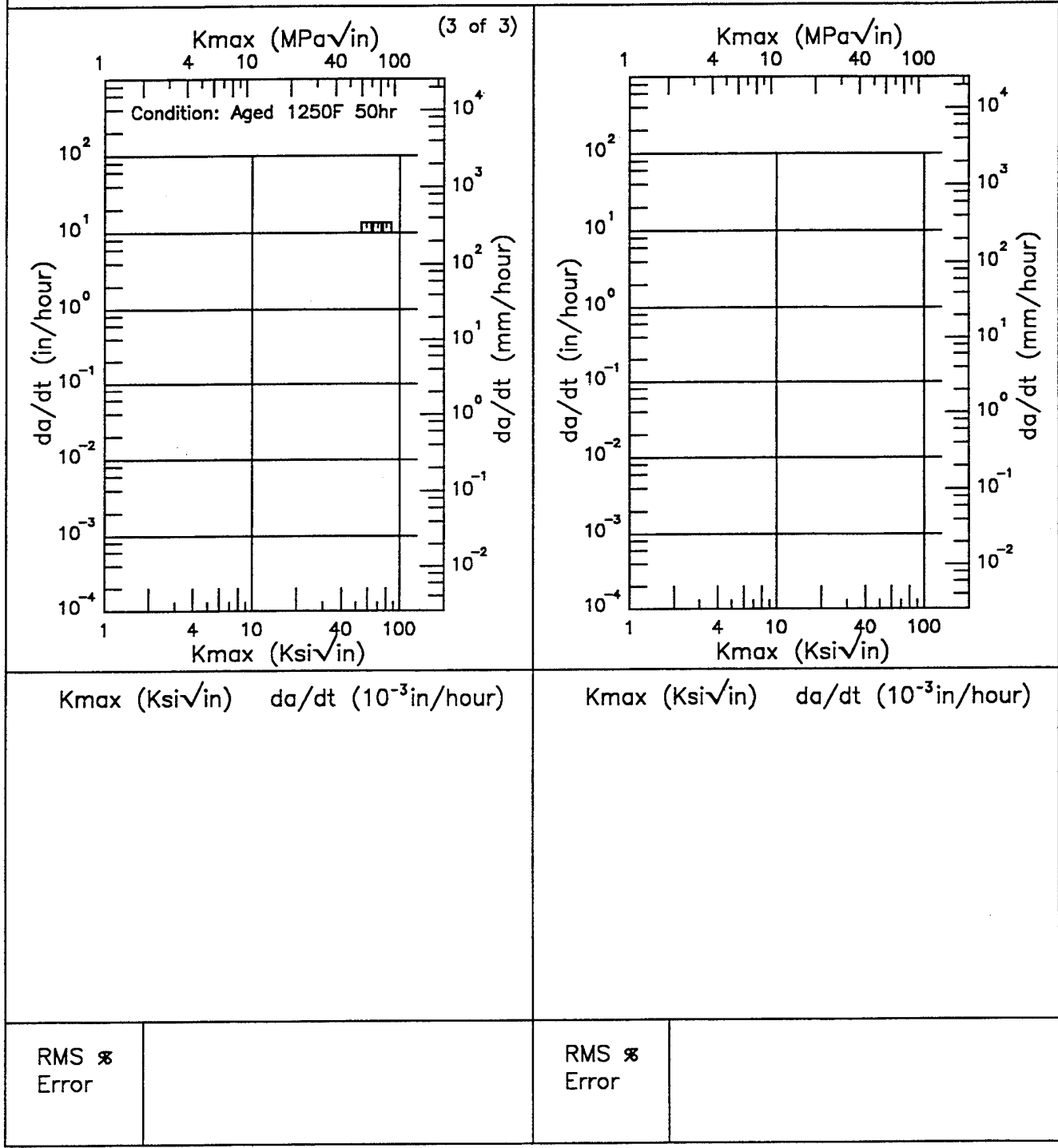
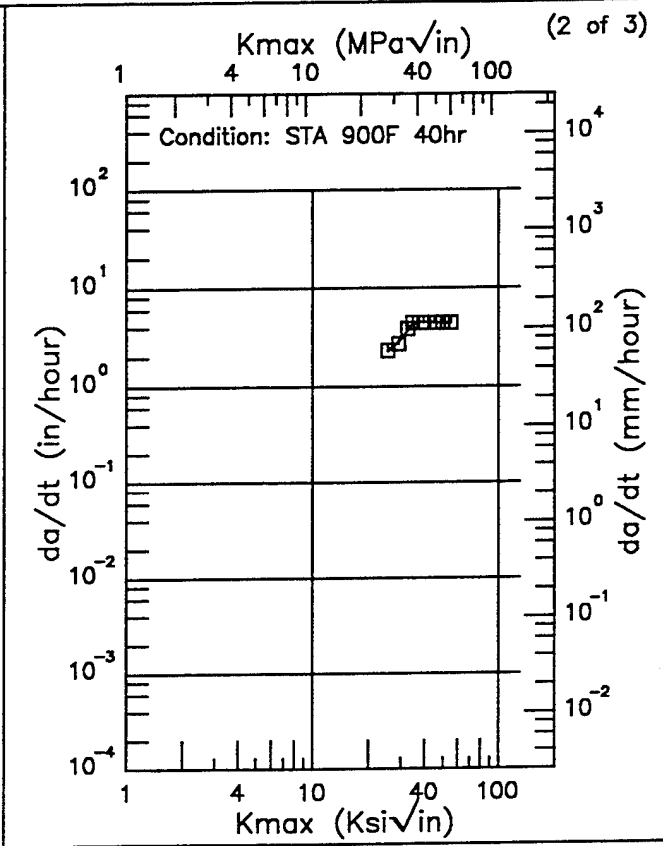
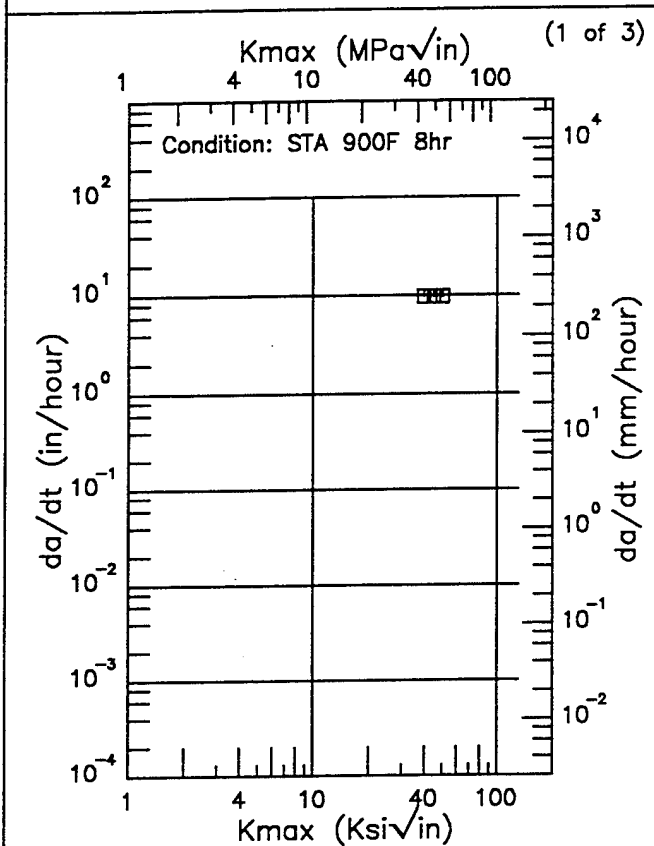


Figure 6.3.3.2.1 (Concluded)

BETA III

Condition/Ht:
 Environment: 0.6M KCl
 Specimen Type: SENT
 Orientation:
 Yield Strength:
 Ult. Strength:

Specimen Thk:
 Specimen Width:
 A₀:
 K_{Isc}:
 Ref: 82651



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
25.50 (min)	2289.
30.	2995.
35.	4400.
40.	4516.
50.	4466.
55.00 (max)	4381.

Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
25.50 (min)	2289.
30.	2995.
35.	4400.
40.	4516.
50.	4466.
55.00 (max)	4381.

RMS %
 Error

RMS %
 Error
 2.79

Figure 6.3.3.2.2

BETA III

Condition/Ht:
 Environment: 0.6M KCl
 Specimen Type: SENT
 Orientation:
 Yield Strength:
 Ult. Strength:

Specimen Thk:
 Specimen Width:
 A₀:
 K_{Isc}:
 Ref: 82651

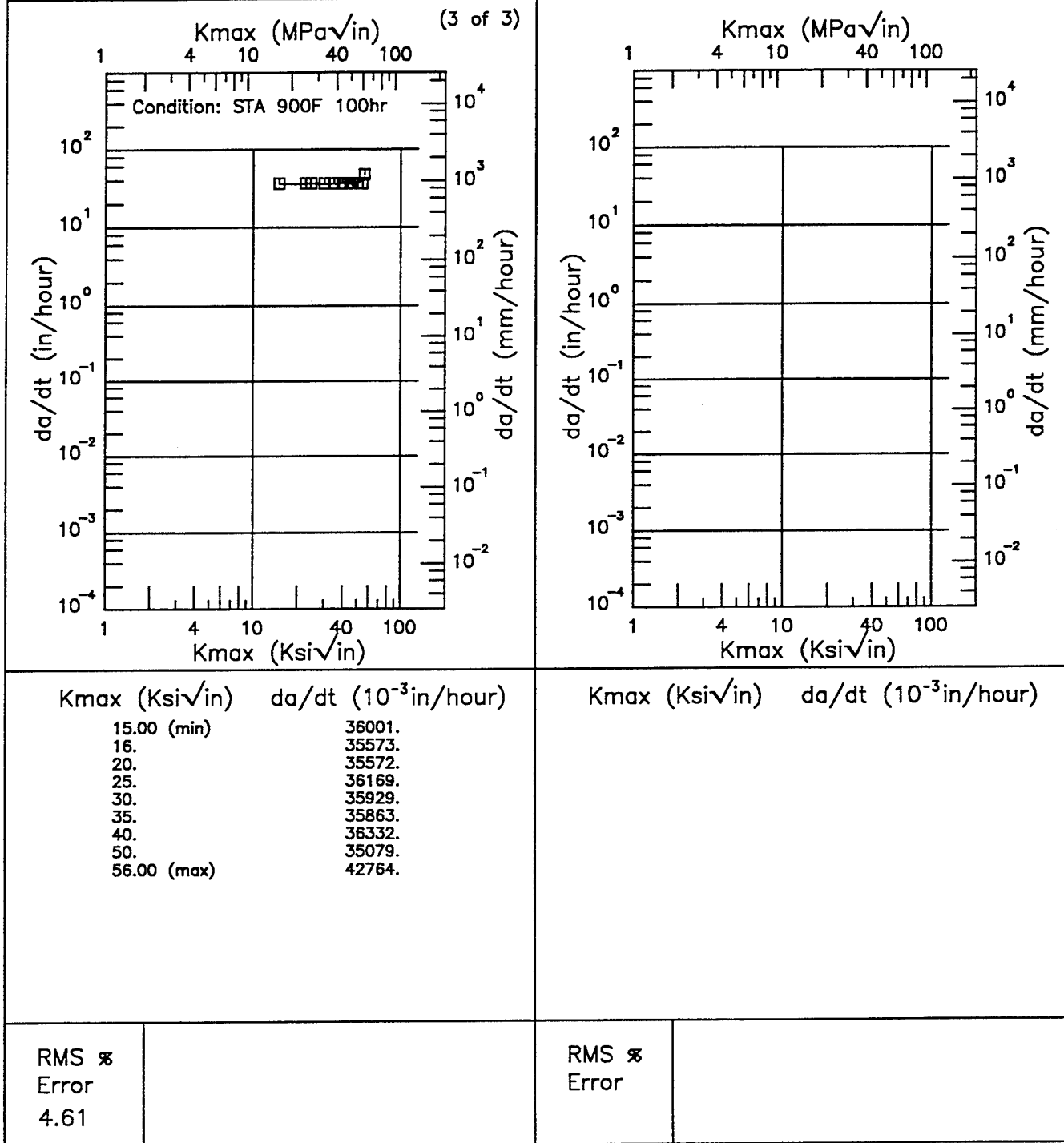


Figure 6.3.3.2.2 (Concluded)

BETA III

(1 of 1)

TABLE 6.3.3.3
 K_{Isc} SUMMARY FOR TITANIUM ALLOY BETA III

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K_Q (Ksi/in)	K_{Isc} (Ksi/in)	Test Time (min)	Test Date	Reference
						Design	Width (in)	Thick (in)							
Beta Stab +Aged 900F 1HR	S	R.T.	---	---	.6M KCl -1500MV	SENT	---	---	---	---	65	55*	---	1970	82651
						SENT	---	---	---	---	65	28*	---	1970	82651
						SENT	---	---	---	---	65	16*	---	1970	82651
						SENT	---	---	---	---	65	14*	---	1970	82651
						SENT	---	---	---	---	65	21*	---	1970	82651
						SENT	---	---	---	---	65	24*	---	1970	82651
						SENT	---	---	---	---	65	25*	---	1970	82651
						SENT	---	---	---	---	65	25*	---	1970	82651

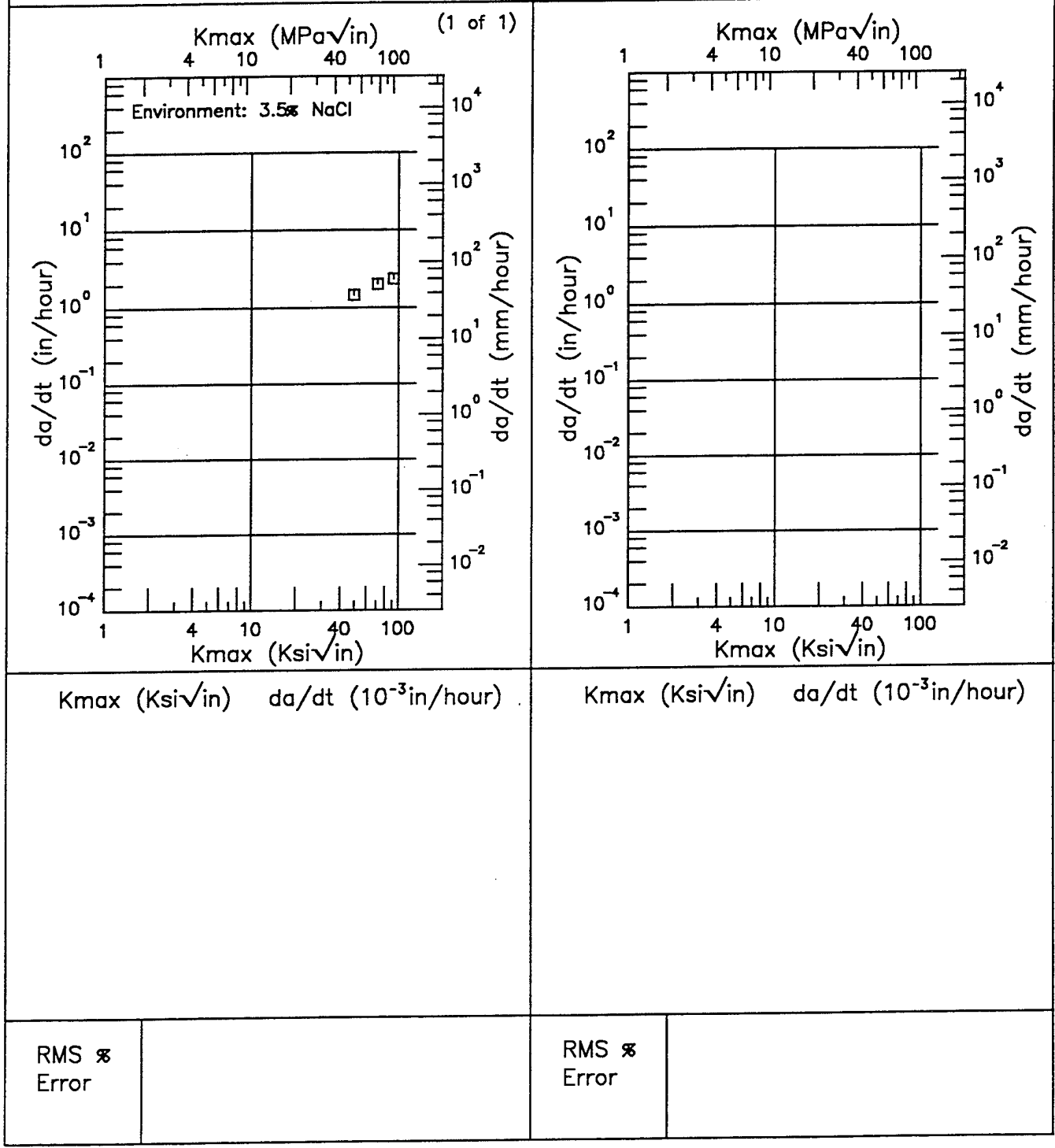
* specimen thickness does not meet minimum requirements of $2.5 \left(\frac{K_{Isc}^2}{\sigma_{YS}} \right)$

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BETA Ti

Condition/Ht: BETA STABILIZED
 Form: 0.16 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation:
 Yield Strength: 136 ksi
 Ult. Strength:

Specimen Thk: 0.16 in.
 Specimen Width: 8 in.
 A₀:
 K_ISCC: 68 ksi
 Ref: 77456



RMS %
Error

RMS %
Error

Figure 6.4.3.2

TABLE 6.5.2.1

1 of 1

TITANIUM CORONA 5 K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	2.5° (K_{Ic}/\sqrt{YS}) ¹ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (ksi \cdot $\sqrt{in.}$)	K_{Ic} MEAN	STAN DEV		
ALPHA-BETA FORGED & LOW ANNEAL & AGE	Forging	2.00	R.T.	---	136.3	---	2.000	---	---	0.57	64.84	---	---	---	R1005

IMI-834

1 of 1

TABLE 6.6.1.2

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 IMI-834 AT ROOM TEMPERATURE

ORIENTATION: C-R ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
1864F 2HRS; OQ TO R.T.; AGE 1161P 2HRS	DISK	0.1	1	2.6	6.0	10.0	20.0	60.0	100.0	
				0.22				7.72		

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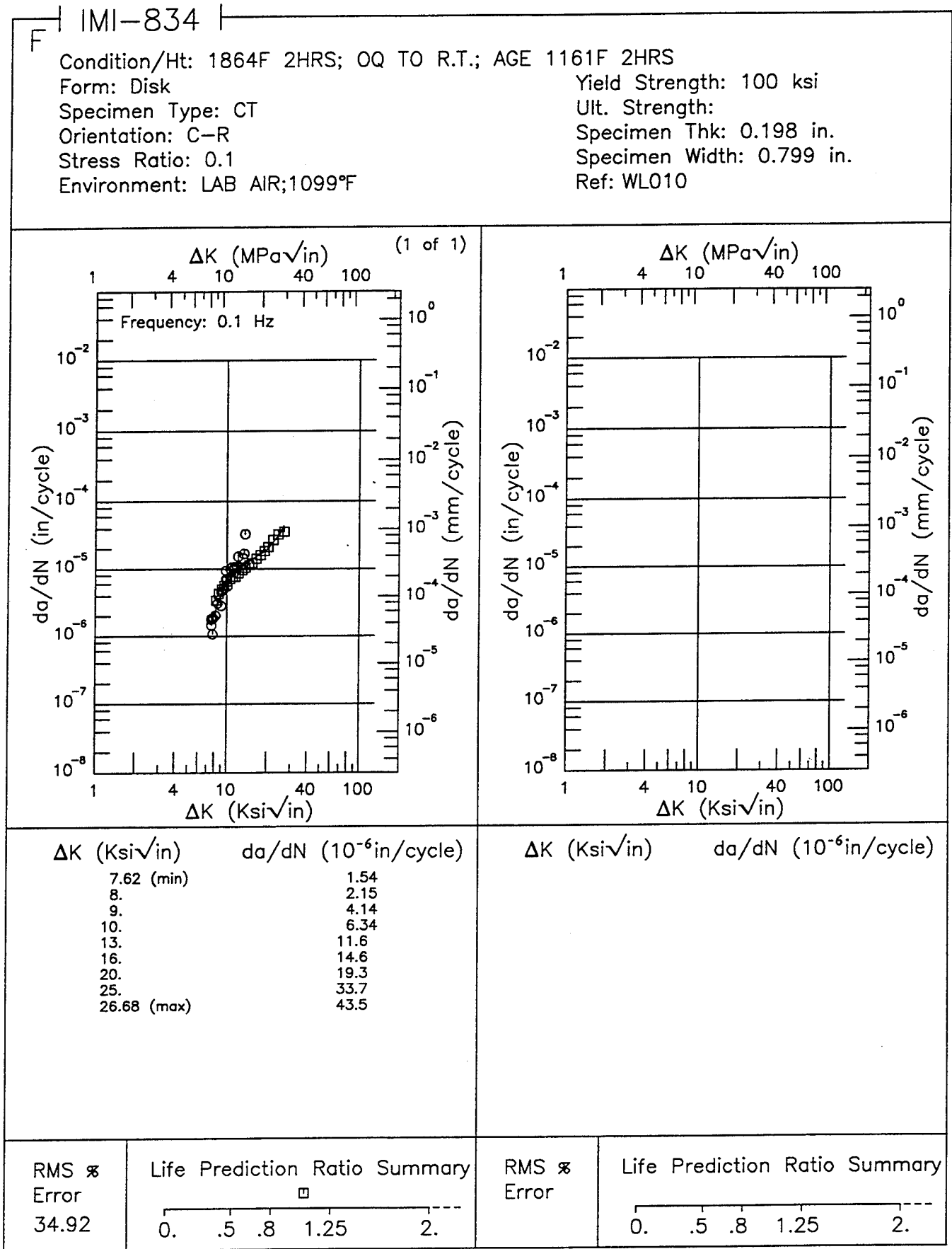


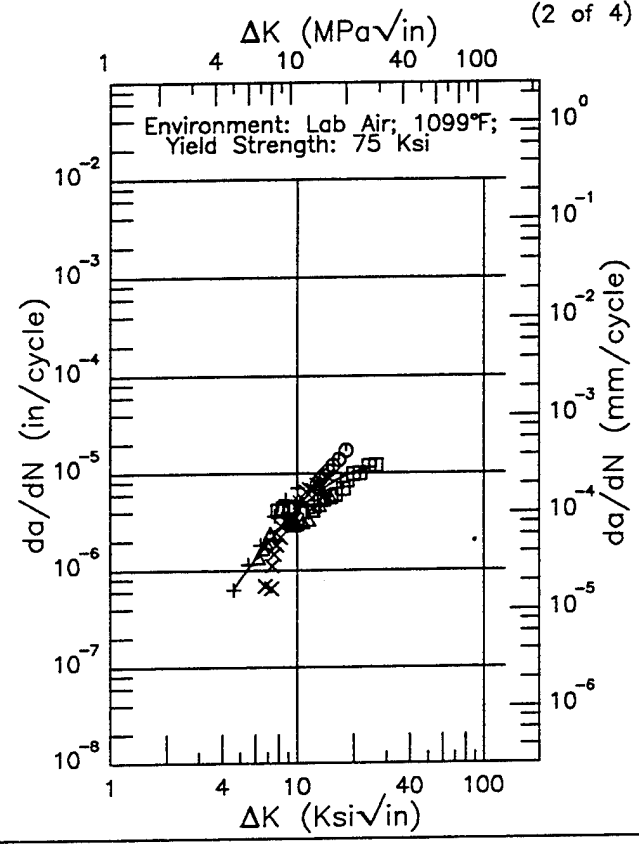
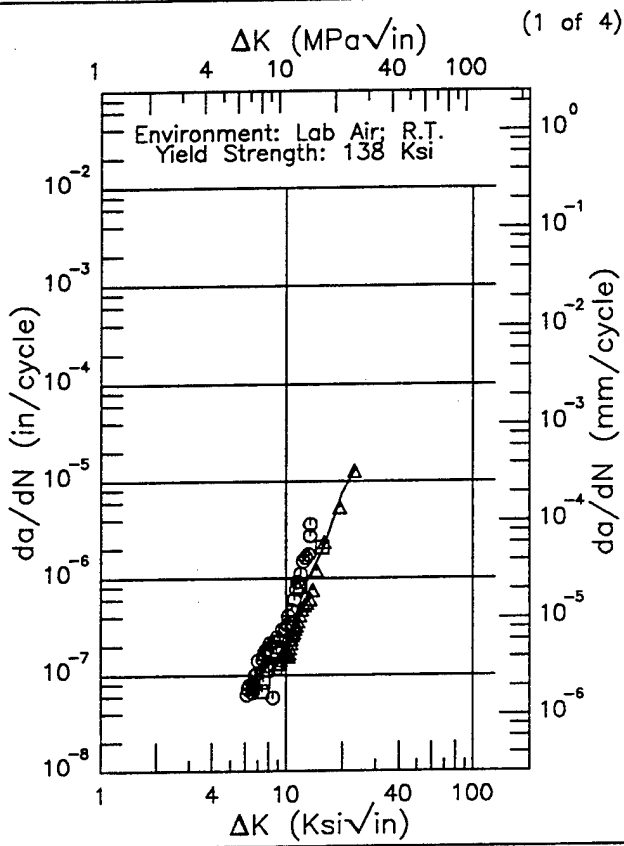
Figure 6.6.3.1.1

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IMI-834

Condition/Ht: 1864F 2HRS; OQ TO R.T.; AGE 1161F 2HRS
 Form: Disk
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.1
 Frequency: 1 Hz

Yield Strength: 75 - 138 ksi
 Ult. Strength:
 Specimen Thk: 0.171 - 0.196 in.
 Specimen Width: 0.796 - 0.8 in.
 Ref: WL010



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
6.11 (min)	0.0669
7.	0.101
8.	0.127
9.	0.158
10.	0.216
13.	0.944
16.	2.15
20.	7.72
23.04 (max)	12.7

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
4.60 (min)	0.640
5.	0.808
6.	1.31
7.	1.91
8.	2.60
9.	3.33
10.	4.10
13.	6.39
16.	8.38
20.	10.3
25.	11.4
26.25 (max)	11.5

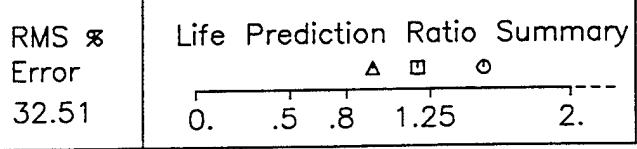
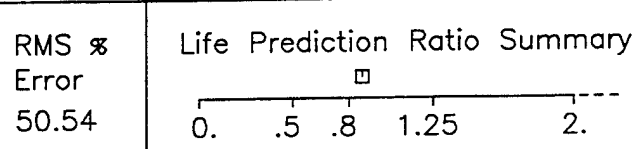
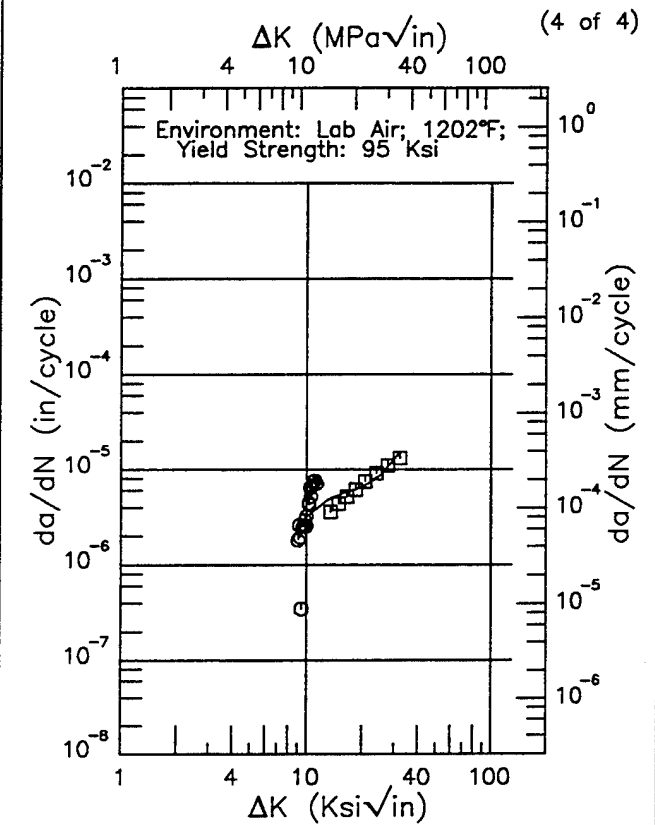
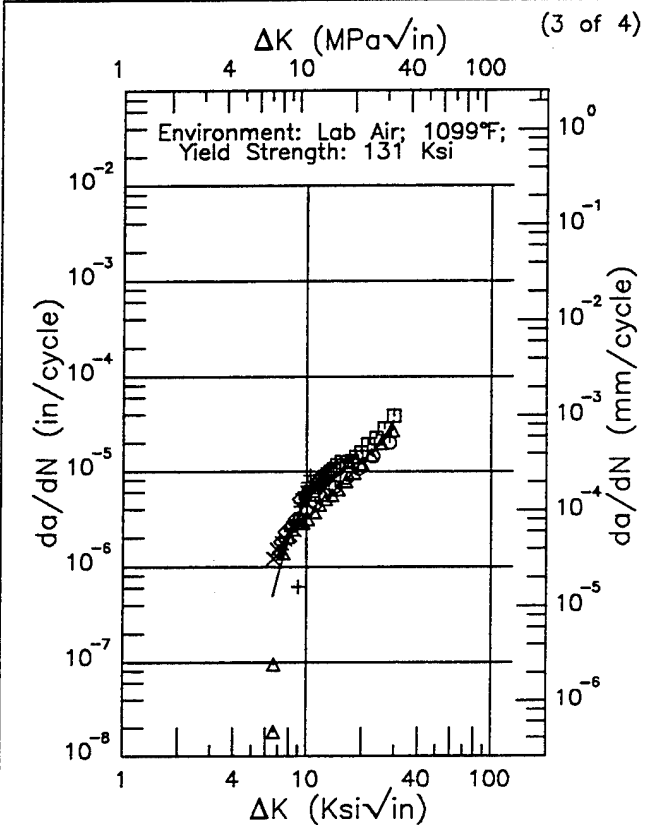


Figure 6.6.3.1.2

IMI-834 E

Condition/Ht: 1864F 2HRS; OQ TO R.T.; AGE 1161F 2HRS
 Form: Disk
 Specimen Type: CT
 Orientation: C-R
 Stress Ratio: 0.1
 Frequency: 1 Hz

Yield Strength: 75 - 138 ksi
 Ult. Strength:
 Specimen Thk: 0.171 - 0.196 in.
 Specimen Width: 0.796 - 0.8 in.
 Ref: WL010



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
6.56 (min)	0.500
7.	0.839
8.	1.97
9.	3.44
10.	4.99
13.	8.54
16.	10.5
20.	13.0
25.	19.5
29.62 (max)	34.2

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.97 (min)	2.34
9.	2.36
10.	3.15
13.	4.80
16.	5.64
20.	6.54
25.	8.52
30.	12.8
31.92 (max)	15.5

RMS % Error	Life Prediction Ratio Summary
38.87	

RMS % Error	Life Prediction Ratio Summary
49.26	

Figure 6.6.3.1.2 (Concluded)

E*

1 of 1

TABLE 6.7.1.1
MEAN PLANE STRAIN FRACTURE TOUGHNESS
FOR TITANIUM ALLOY TI-6 AL-4V AT ROOM TEMPERATURE

Product Form	Condition/Heat Treatment	K_{Ic} (ksi \sqrt{in})								
		Specimen Orientation								
		L-T			T-L			S-L		
Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n		
Plate	1740F 1 HR AC	61.6	1.6	3	---	---	---	---	---	
	STA-1740F 1 HR AC 1000F 8HR AC	55.3	1.5	3	---	---	---	---	---	

TABLE 6.8.1.2

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 TI-10-2-3 AT ROOM TEMPERATURE**

ENVIRONMENT: Lab Air

ORIENTATION: L-T

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Kst \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
UNSPECIFIED	UNSPECIFIED	0.05	10							
		0.1	10			1.16				
		0.33	10			0.65				

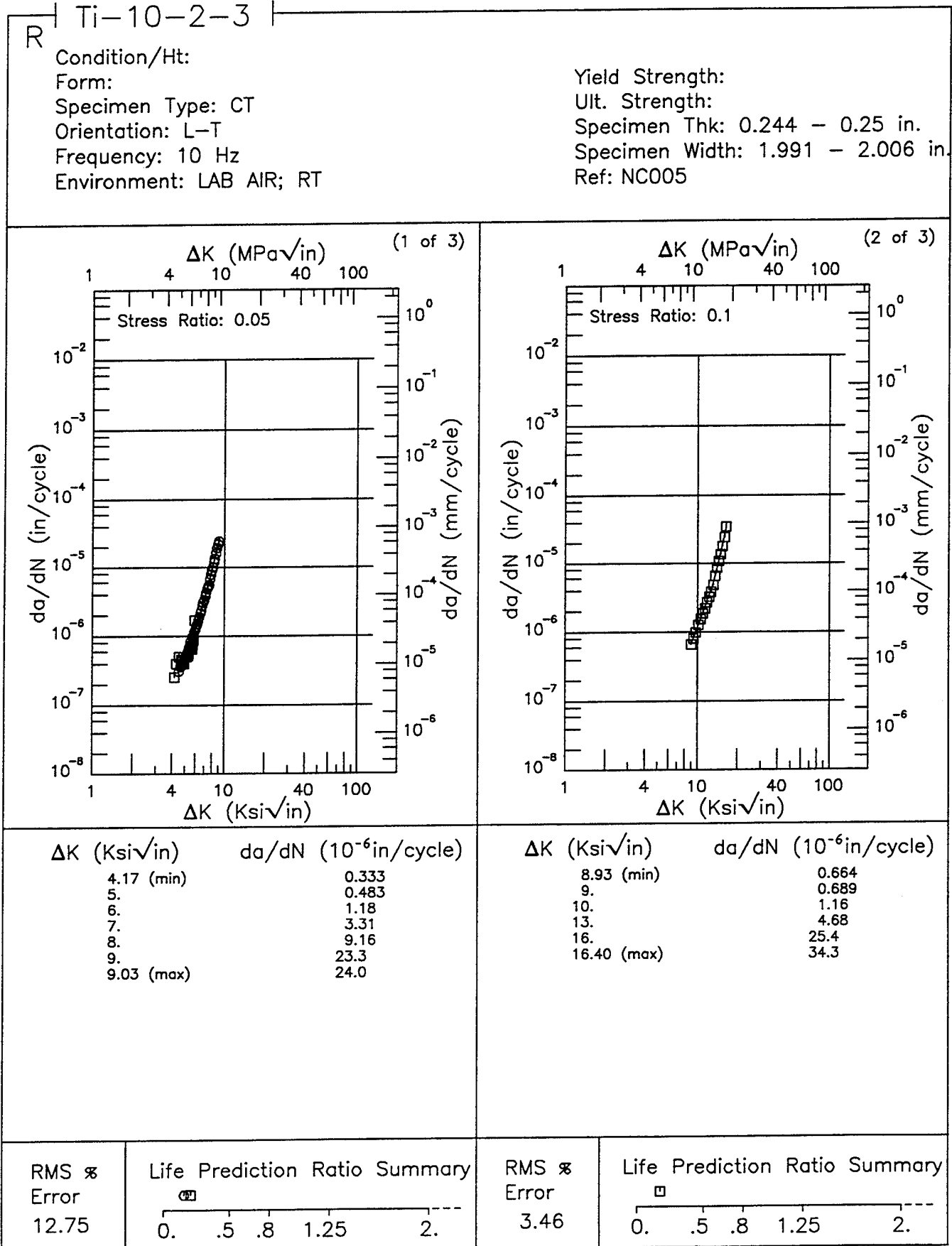
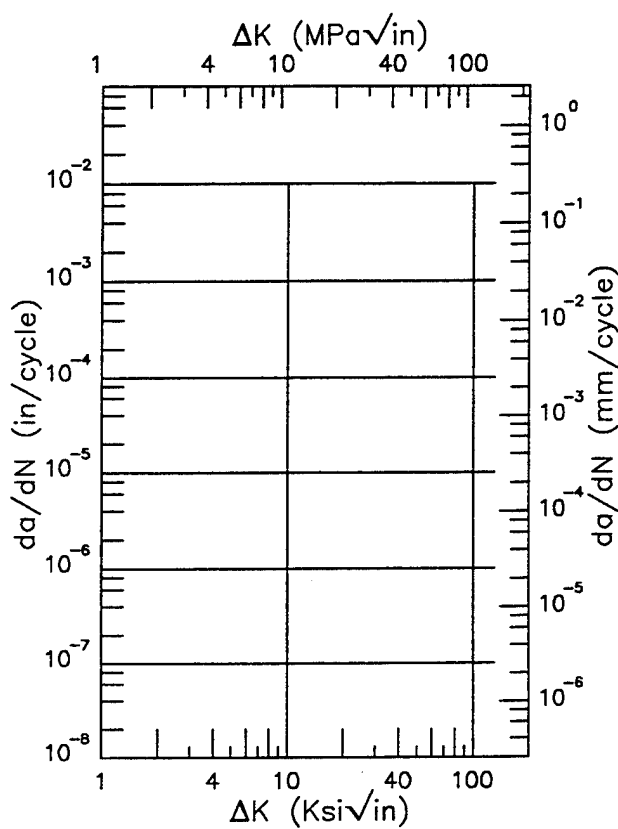
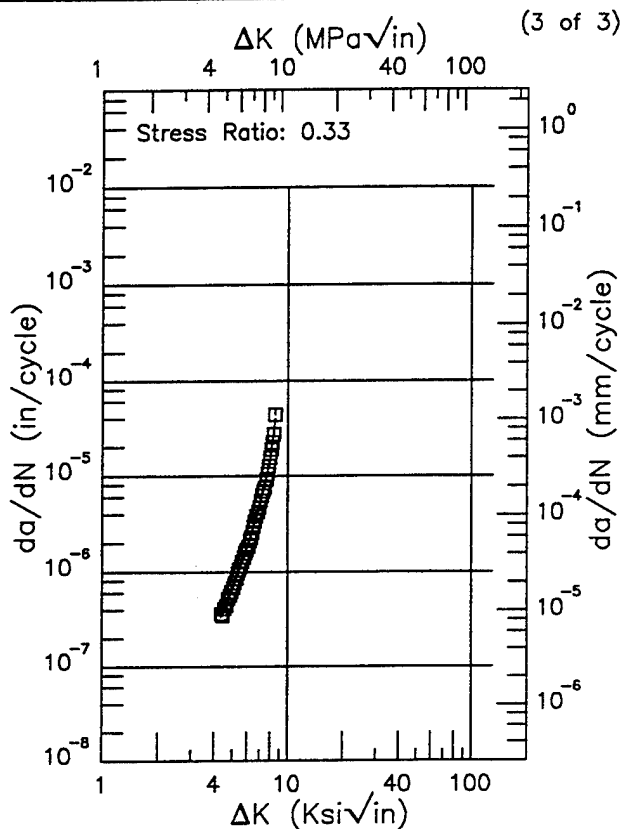


Figure 6.8.3.1

Ti-10-2-3 R

Condition/Ht:
 Form:
 Specimen Type: CT
 Orientation: L-T
 Frequency: 10 Hz
 Environment: LAB AIR; RT

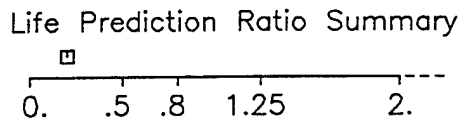
Yield Strength:
 Ult. Strength:
 Specimen Thk: 0.244 - 0.25 in.
 Specimen Width: 1.991 - 2.006 in.
 Ref: NC005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
4.36 (min)	0.331
5.	0.652
6.	1.75
7.	4.85
8.	15.3
8.53 (max)	40.0

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS %
 Error
 4.46



RMS %
 Error

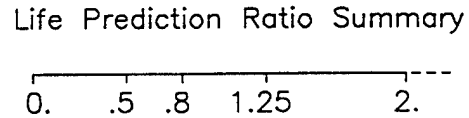


Figure 6.8.3.1 (Concluded)

Ti-4Al-3Mo-1V

(1 of 1)

TABLE 6.9.3.3
 K_{Isc} SUMMARY FOR TITANIUM ALLOY Ti-4Al-3Mo-1V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K_q (Ksi√in)	K_{Isc} (Ksi√in)	Test Time (min)	Test Date	Reference
						Design	Width (in)	Thick (in)							
Mill Annealed	P	R.T.	L-S	---	3.5% NaCl	---	---	---	0.5	---	117	105*	---	1969	75386

* specimen thickness does not meet minimum requirements of $2.5 \left(\frac{K_{Isc}}{\sigma_{YS}} \right)^2$

TABLE 6.10.1.2.1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
Ti-5-2.5 ELI AT ROOM TEMPERATURE

ORIENTATION: R-C

ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
UNSPECIFIED	DISK	0.05	0.1				11.41			
		0.05	10				11.62	142.07		
		0.5	0.1			2.35	16.79			
		0.5	10			2.76	17.77			

Ti-5-2.5 ELI

1 of 1

TABLE 6.10.1.2.2

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-5-2.5 ELI AT ROOM TEMPERATURE

ORIENTATION: C-R ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
UNSPECIFIED	DISK	0.05	0.1				12.29		
		0.05	10		1.88	19.49	261.66	108.63	
		0.5	0.1			2.79	15.32		
		0.5	10			2.81	24.16		

TABLE 6.10.1.2.3

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-5-2.5 ELI AT ROOM TEMPERATURE

ORIENTATION: CS ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
UNSPECIFIED	DISK	0.05	10						
		0.5	0.1			0.9	7.78		
		0.5	10			0.48	8.85		

Ti-5-2.5 ELI

1 of 1

TABLE 6.10.1.2.4

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
Ti-5-2.5 ELI AT ROOM TEMPERATURE

ENVIRONMENT: Lab Air

ORIENTATION: SC

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
UNSPECIFIED	DISK	0.05	0.1				9.9	200.55	
		0.05	10				10.05	196.41	
		0.5	0.1			1.8	16.77		
		0.5	10			3	18.45		

TABLE 6.10.1.2.5

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-5-2.5 ELI AT ROOM TEMPERATURE

ENVIRONMENT: Lab Air

ORIENTATION: RS

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
UNSPECIFIED	DISK	0.05	0.1			0.44	3.78		
		0.05	10			0.08	3.48	71.92	
		0.5	10			0.93	13.57		

TI-5-2.5 ELI

1 of 1

TABLE 6.10.1.2.6

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 TI-5-2.5 ELI AT ROOM TEMPERATURE

ENVIRONMENT: Lab Air

ORIENTATION: SR

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level ($K_{SI}(\sqrt{in})$)						
				2.5	5.0	10.0	20.0	50.0	100.0	
UNSPECIFIED	DISK	0.05	0.1					8.45		
		0.05	10					7.15	194.49	
		0.5	0.1			1.9		14.85		
		0.5	10			1.89		20.85		

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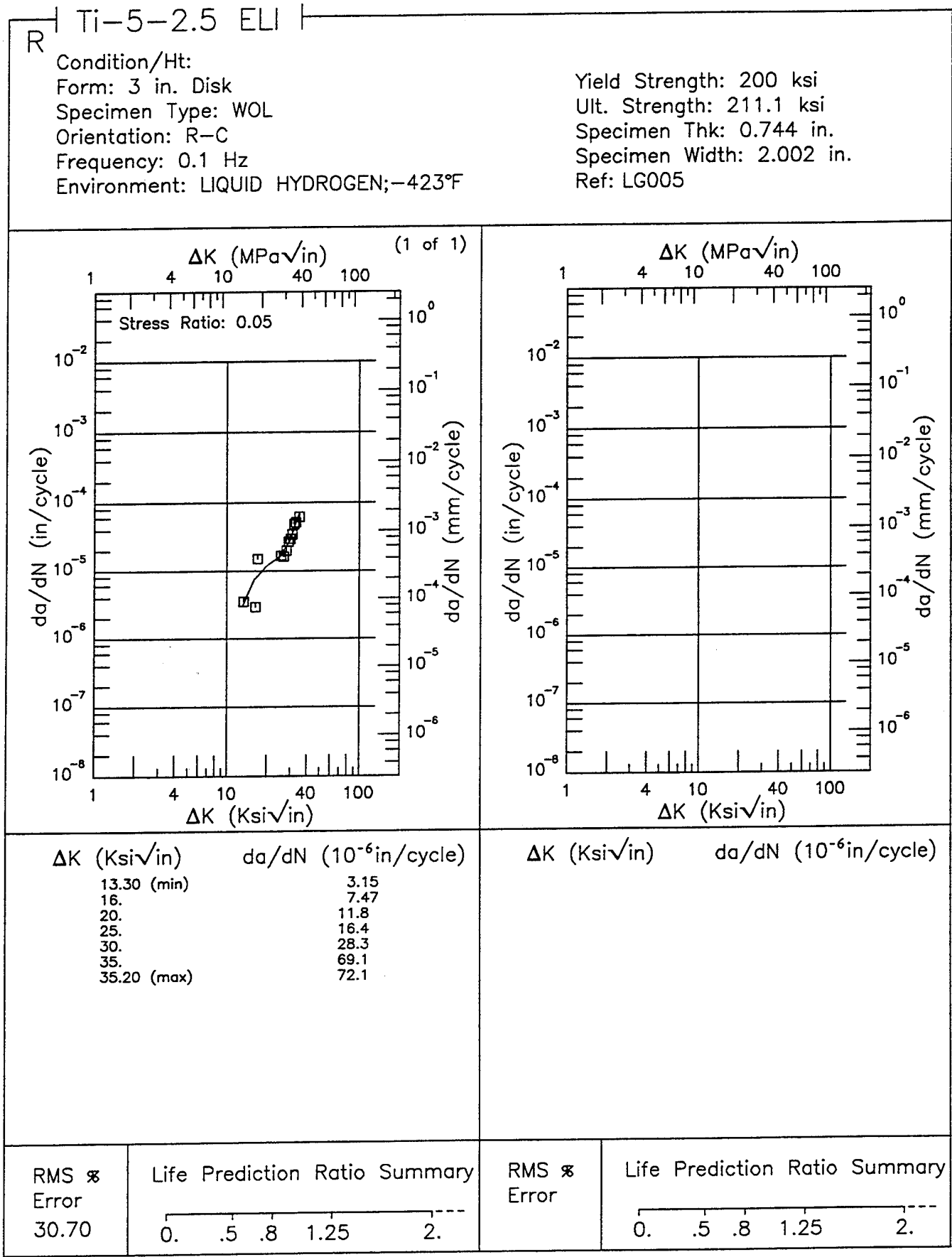
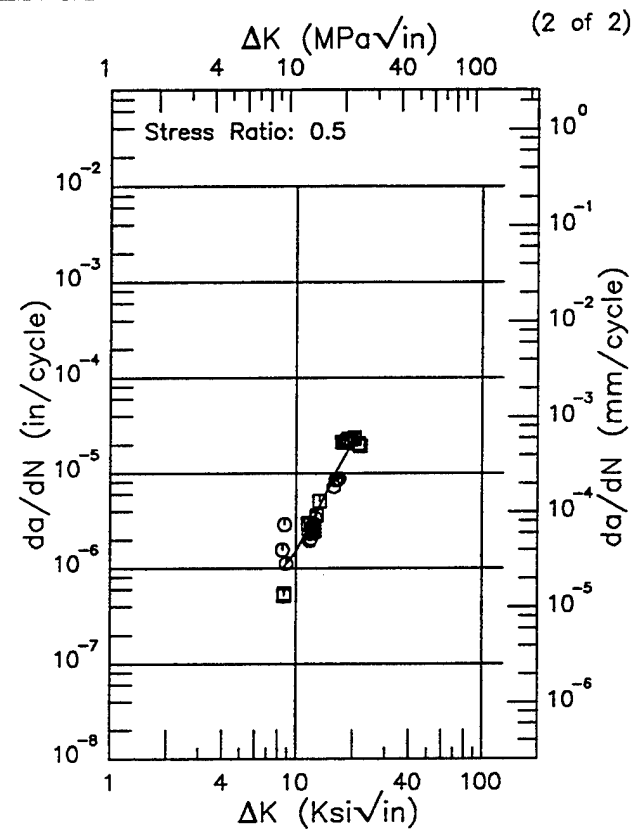
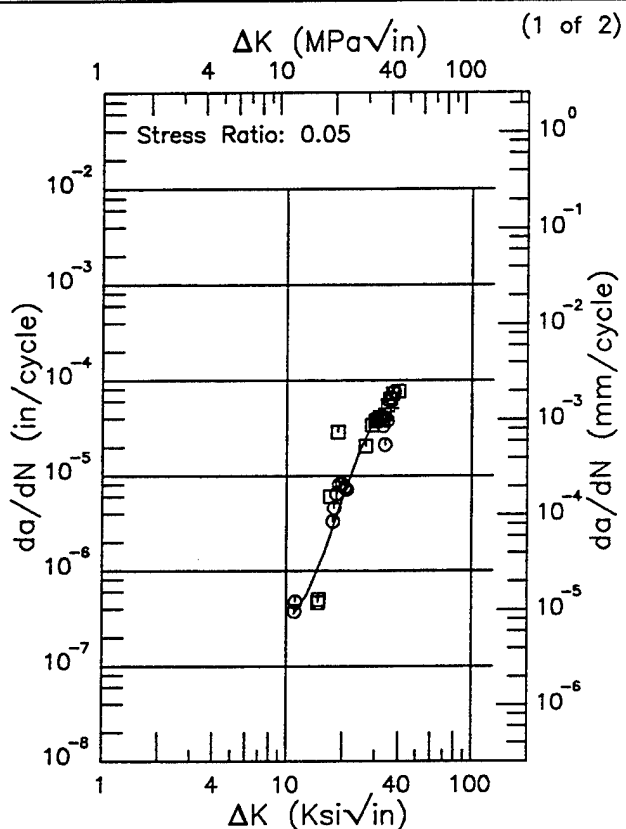


Figure 6.10.3.1.1

Ti-5-2.5 ELI R

Condition/Ht:
 Form: 3 in. Disk
 Specimen Type: WOL
 Orientation: R-C
 Frequency: 10 Hz
 Environment: LIQUID HYDROGEN; -423°F

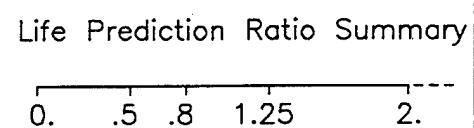
Yield Strength: 200 ksi
 Ult. Strength: 211.1 ksi
 Specimen Thk: 0.745 - 0.748 in.
 Specimen Width: 2.003 - 2.011 in.
 Ref: LG005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.98 (min)	0.356
13.	0.579
16.	1.57
20.	5.72
25.	19.4
30.	40.3
35.	54.0
39.97 (max)	50.5

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.44 (min)	0.990
9.	1.13
10.	1.51
13.	4.01
16.	9.48
20.	21.3
21.84 (max)	27.1

RMS % Error
>100.0



RMS % Error
41.66

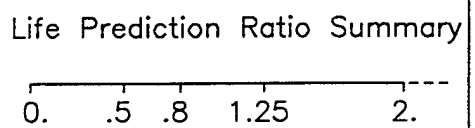


Figure 6.10.3.1.2

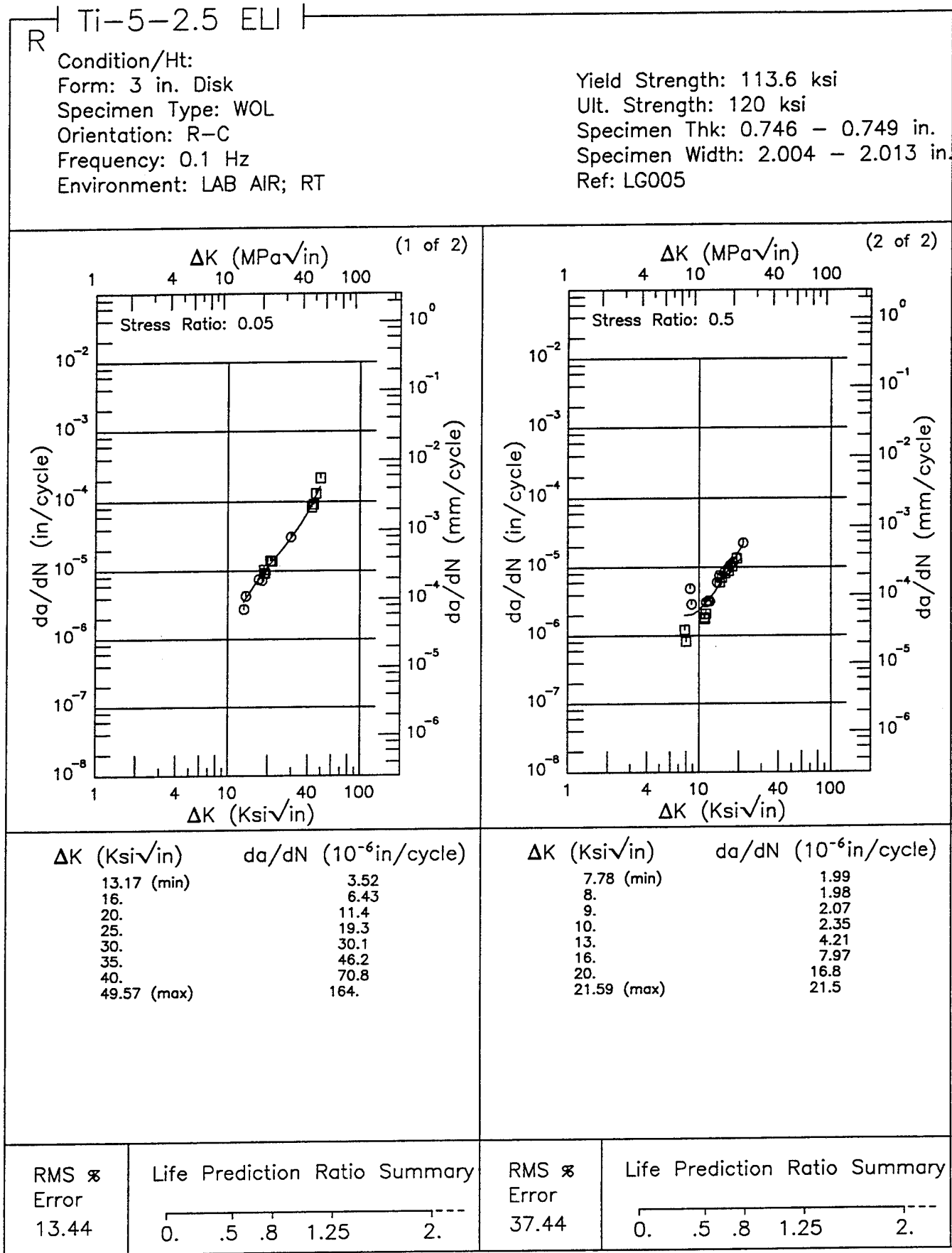
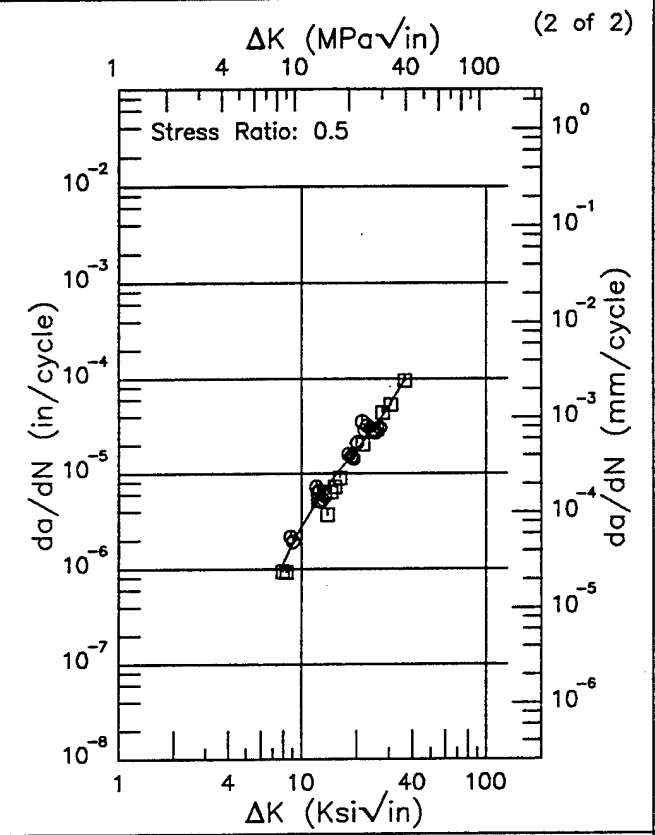
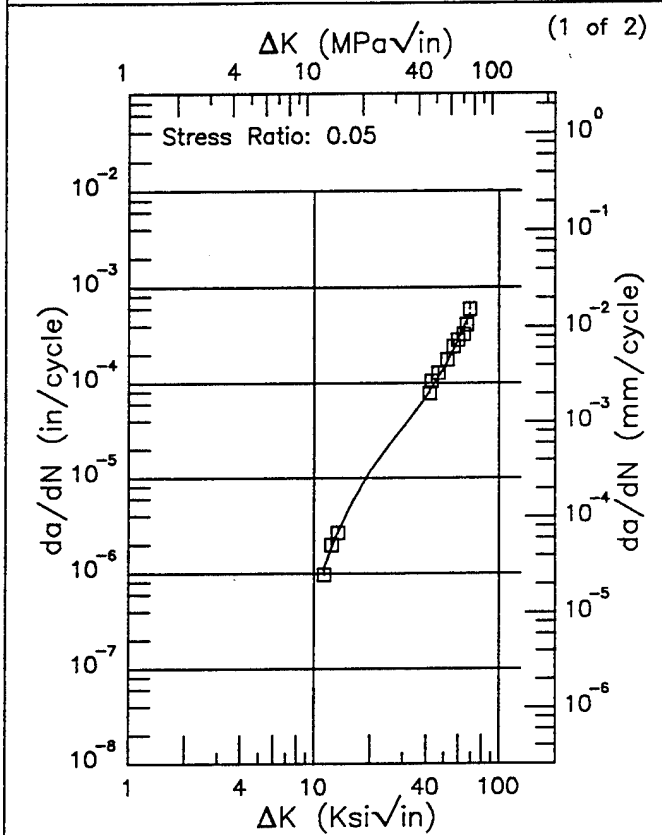


Figure 6.10.3.1.3

Ti-5-2.5 ELI R

Condition/Ht:
 Form: 3 in. Disk
 Specimen Type: WOL
 Orientation: R-C
 Frequency: 10 Hz
 Environment: LAB AIR; RT

Yield Strength: 113.6 ksi
 Ult. Strength: 120 ksi
 Specimen Thk: 0.745 - 0.749 in.
 Specimen Width: 2 - 2.005 in.
 Ref: LG005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
11.23 (min)	1.09
13.	2.32
16.	5.54
20.	11.6
25.	21.5
30.	33.9
35.	50.0
40.	71.5
50.	142.
60.	283.
69.25 (max)	542.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.91 (min)	1.09
8.	1.14
9.	1.87
10.	2.76
13.	6.17
16.	10.5
20.	17.8
25.	30.9
30.	51.8
35.	86.6
36.03 (max)	96.3

RMS % Error	Life Prediction Ratio Summary
9.74	

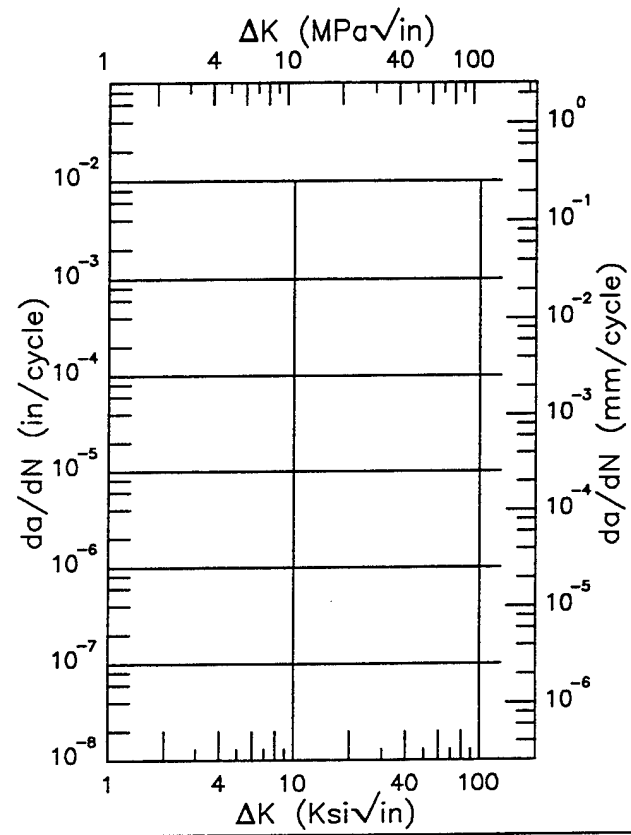
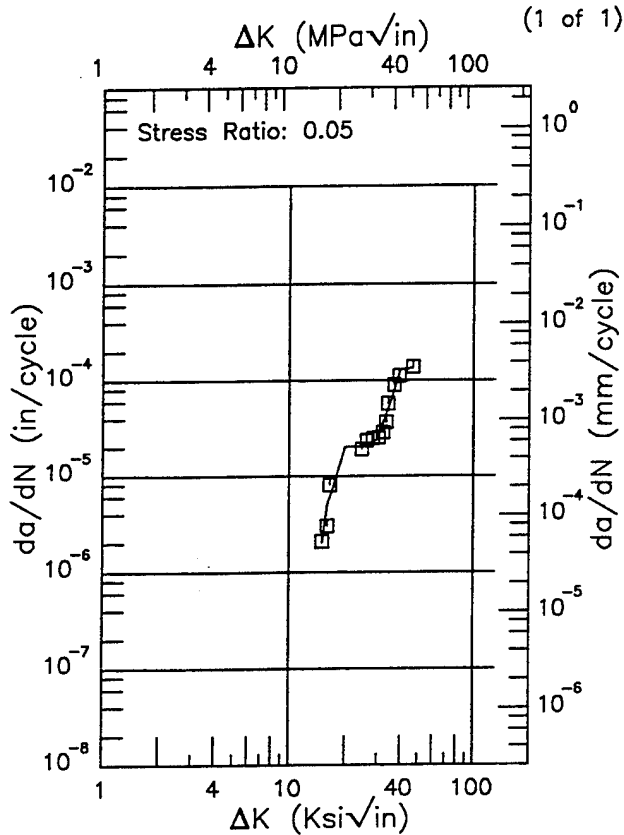
RMS % Error	Life Prediction Ratio Summary
23.13	

Figure 6.10.3.1.4

R | Ti-5-2.5 ELI |

Condition/Ht:
 Form: 3 in. Disk
 Specimen Type: WOL
 Orientation: C-R
 Frequency: 0.1 Hz
 Environment: LIQUID HYDROGEN; -423°F

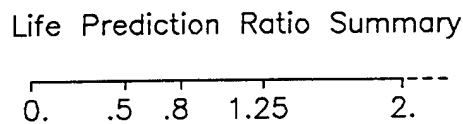
Yield Strength: 200 ksi
 Ult. Strength: 211.1 ksi
 Specimen Thk: 0.749 in.
 Specimen Width: 2.004 in.
 Ref: LG005



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
14.99 (min)	2.03
16.	5.06
20.	20.2
25.	20.4
30.	24.5
35.	58.2
40.	130.
46.57 (max)	135.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
--------------------------------------	-------------------------------

RMS %
 Error
 15.84



RMS %
 Error

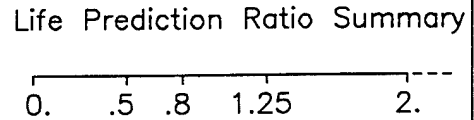
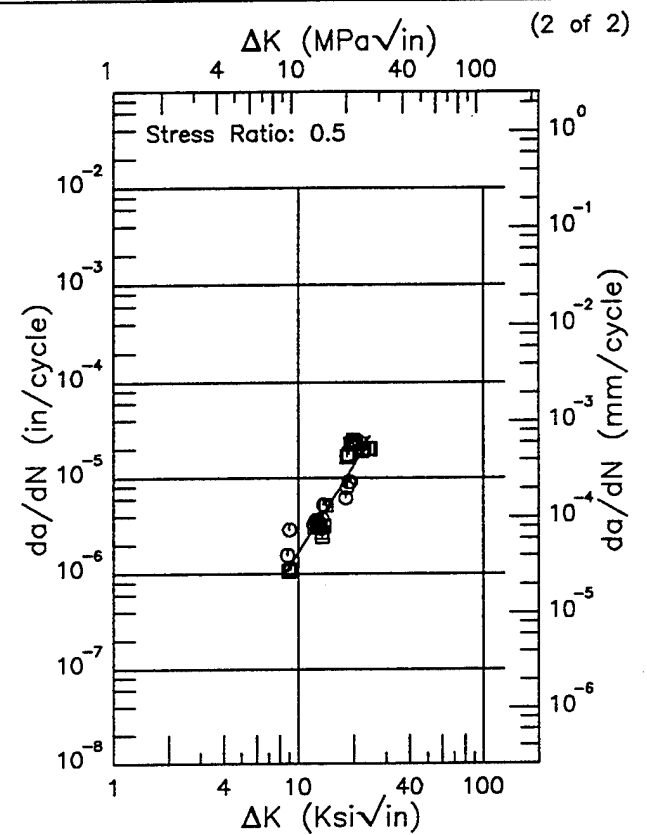
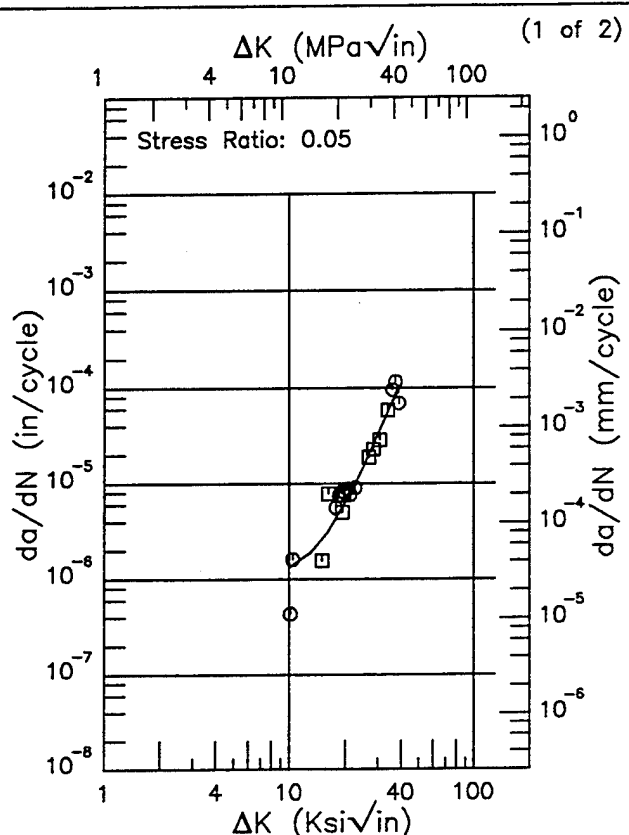


Figure 6.10.3.1.5

Ti-5-2.5 ELI R

Condition/Ht:
 Form: 3 in. Disk
 Specimen Type: WOL
 Orientation: C-R
 Frequency: 10 Hz
 Environment: LIQUID HYDROGEN; -423°F

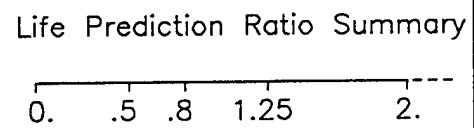
Yield Strength: 200 ksi
 Ult. Strength: 211.1 ksi
 Specimen Thk: 0.744 - 0.747 in.
 Specimen Width: 2.004 - 2.008 in.
 Ref: LG005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.10 (min)	1.34
13.	1.87
16.	3.08
20.	6.32
25.	15.0
30.	32.7
35.	65.4
38.83 (max)	105.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.73 (min)	1.08
9.	1.19
10.	1.65
13.	3.62
16.	6.73
20.	13.6
24.45 (max)	27.3

RMS %
 Error
 44.37



RMS %
 Error
 41.95

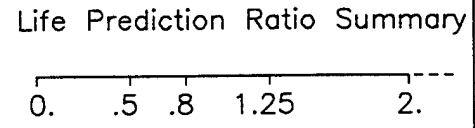


Figure 6.10.3.1.6

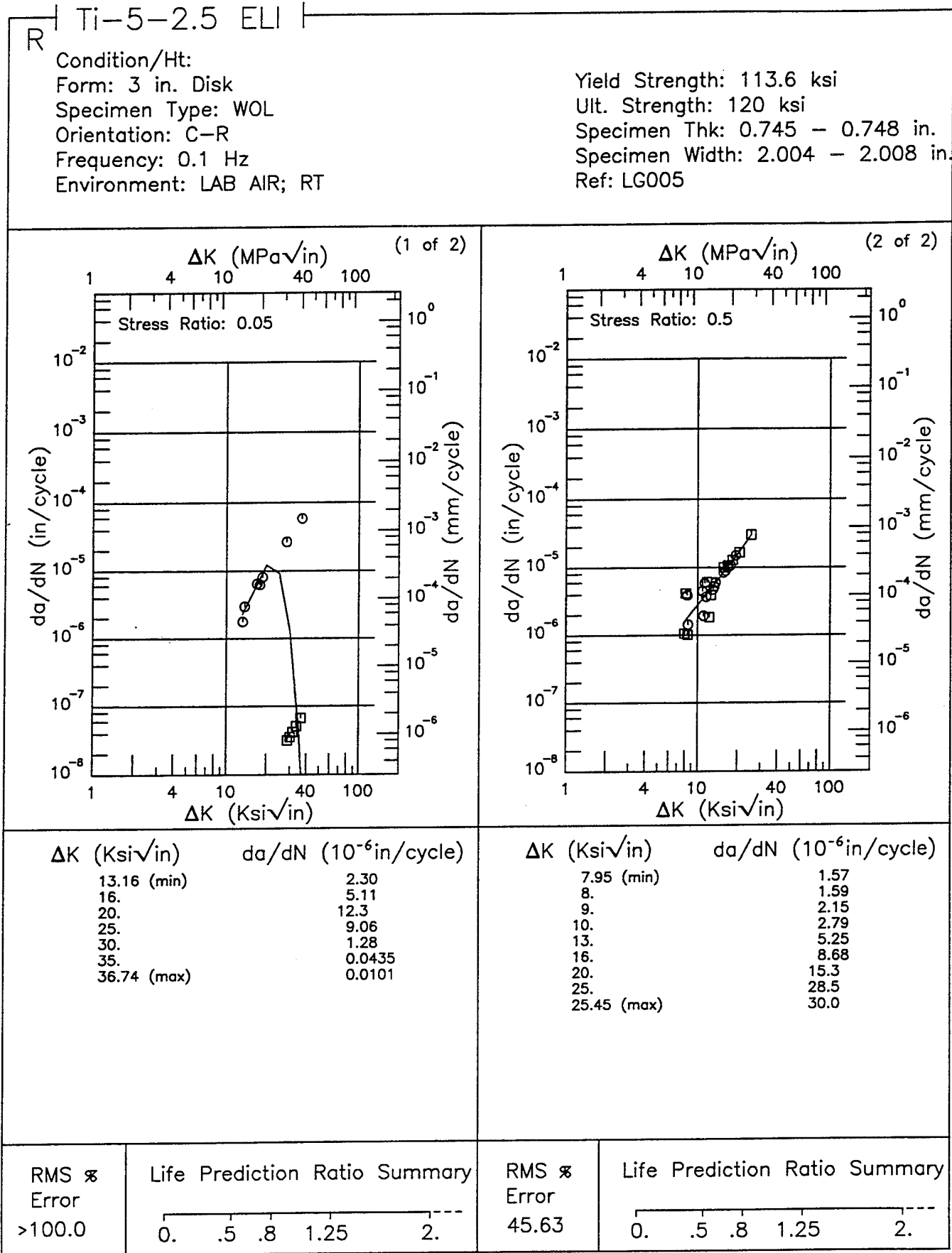
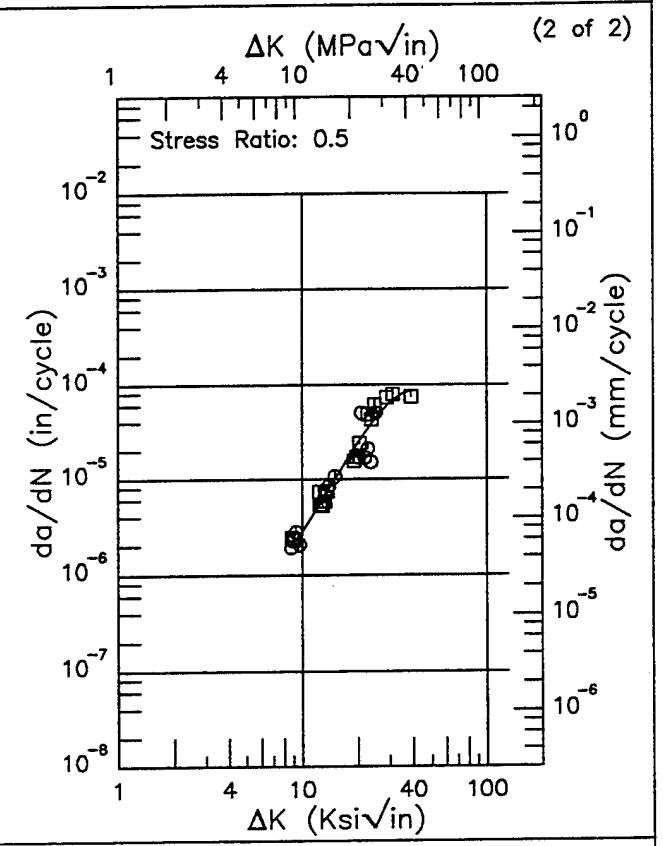
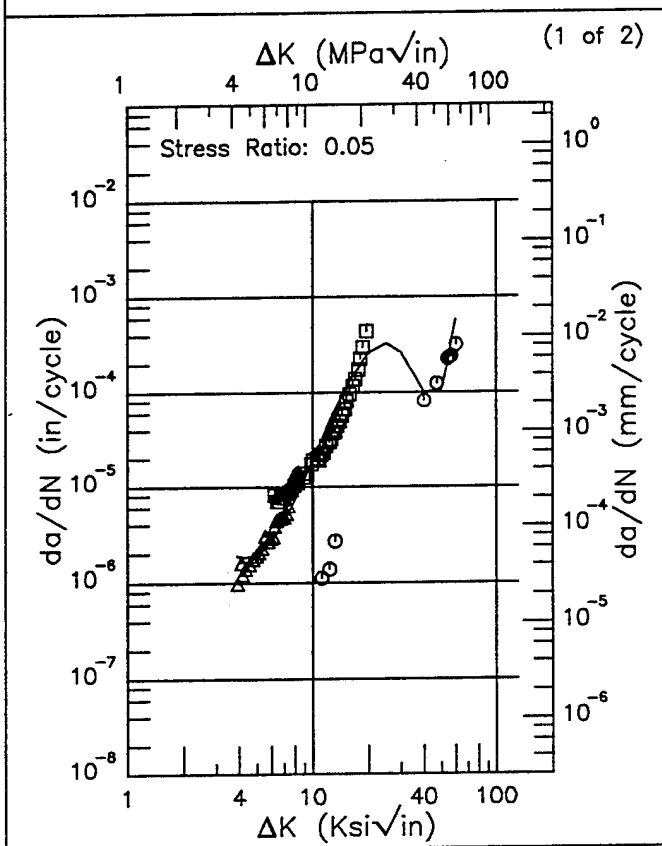


Figure 6.10.3.1.7

Ti-5-2.5 ELI R

Condition/Ht:
 Form: 3 in. Disk
 Specimen Type: WOL
 Orientation: C-R
 Frequency: 10 Hz
 Environment: LAB AIR; RT

Yield Strength: 113.6 ksi
 Ult. Strength: 120 ksi
 Specimen Thk: 0.744 - 0.753 in.
 Specimen Width: 2.002 - 2.01 in.
 Ref: LG005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
3.89 (min)	1.90
4.	1.83
5.	1.88
6.	2.69
7.	4.37
8.	7.32
9.	12.1
10.	19.5
13.	63.1
16.	142.
20.	262.
25.	325.
30.	262.
35.	159.
40.	100.
50.	107.
59.86 (max)	577.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.70 (min)	1.96
9.	2.13
10.	2.81
13.	6.18
16.	12.1
20.	24.2
25.	44.3
30.	64.8
35.	80.3
38.87 (max)	86.7

RMS % Error 50.67	Life Prediction Ratio Summary
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RMS % Error 29.98	Life Prediction Ratio Summary
----------------------	-----------------------------------

Figure 6.10.3.1.8

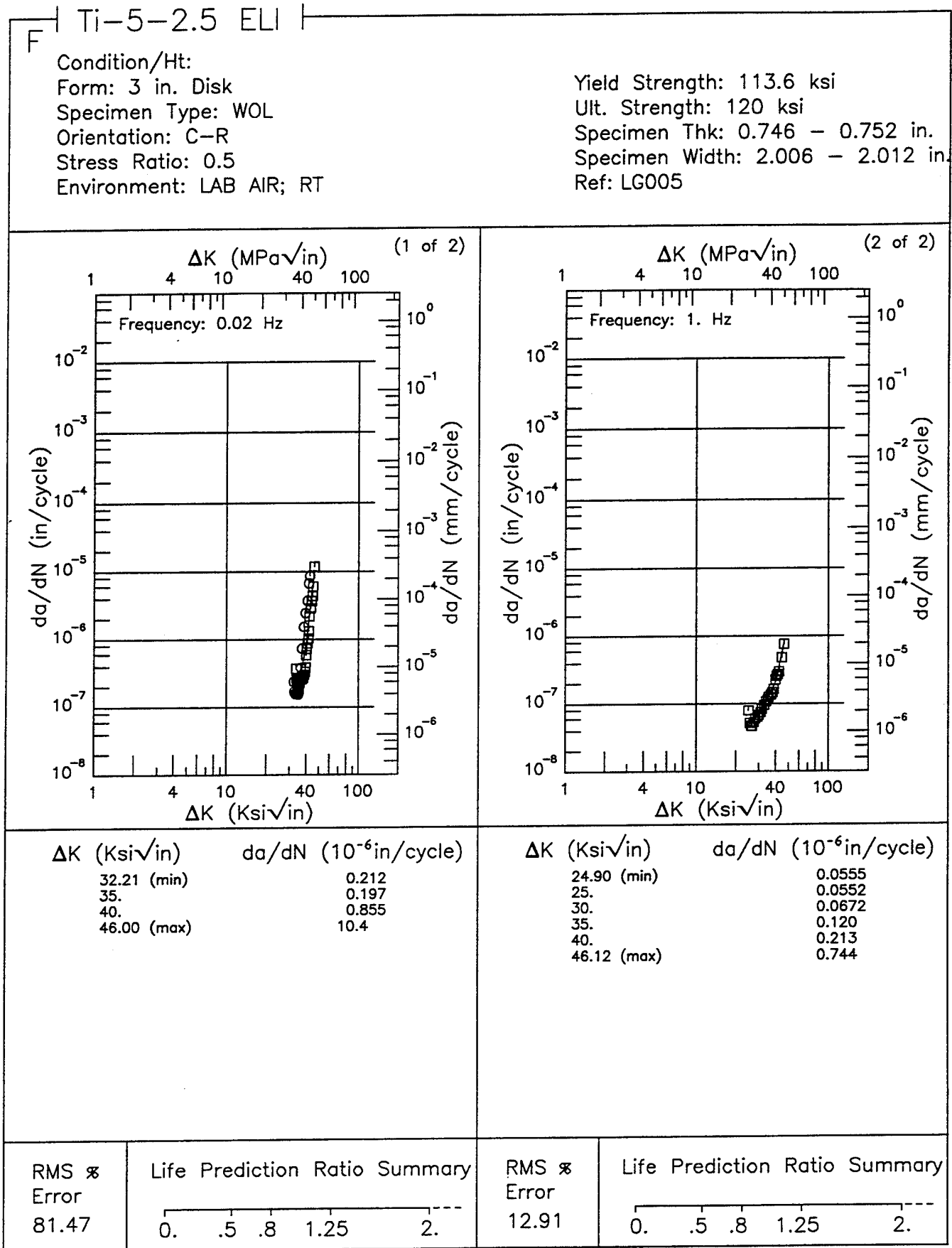
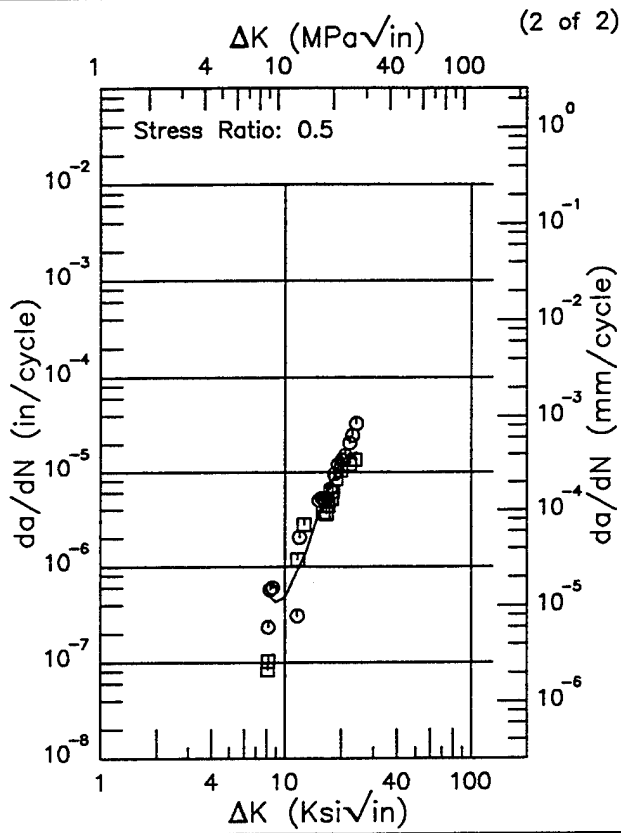
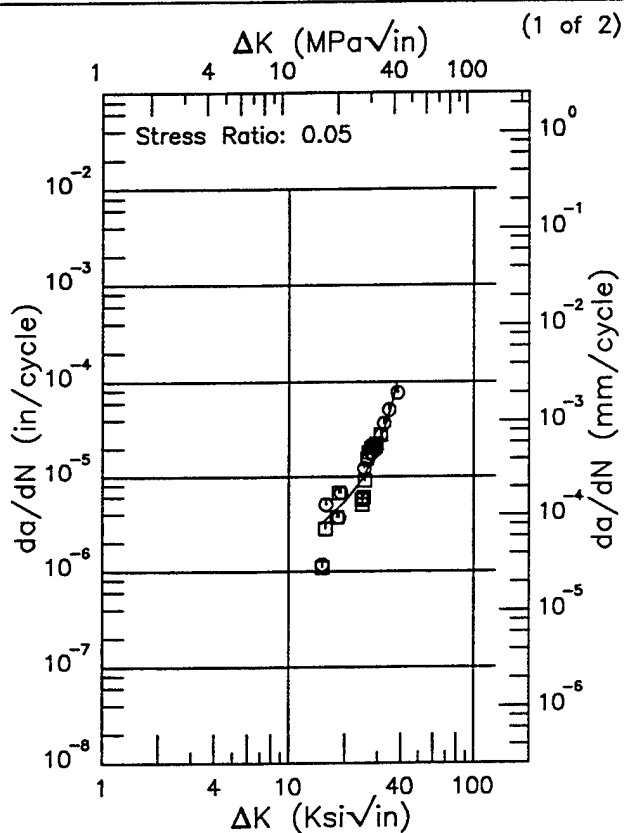


Figure 6.10.3.1.9

Ti-5-2.5 ELI R

Condition/Ht:
 Form: 3 in. Disk
 Specimen Type: WOL
 Orientation: C-S
 Frequency: 10 Hz
 Environment: LIQUID HYDROGEN; -423°F

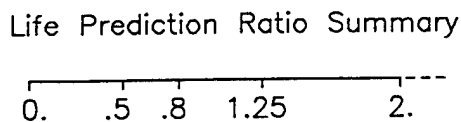
Yield Strength: 200 ksi
 Ult. Strength: 211.1 ksi
 Specimen Thk: 0.745 - 0.747 in.
 Specimen Width: 2.006 - 2.012 in.
 Ref: LG005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
15.02 (min)	3.18
16.	3.56
20.	5.40
25.	9.69
30.	20.5
35.	50.8
38.28 (max)	99.7

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.03 (min)	0.514
9.	0.426
10.	0.480
13.	1.45
16.	4.81
20.	13.3
24.09 (max)	16.4

RMS %
 Error
 32.43



RMS %
 Error
 47.38

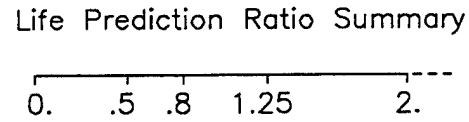


Figure 6.10.3.1.10

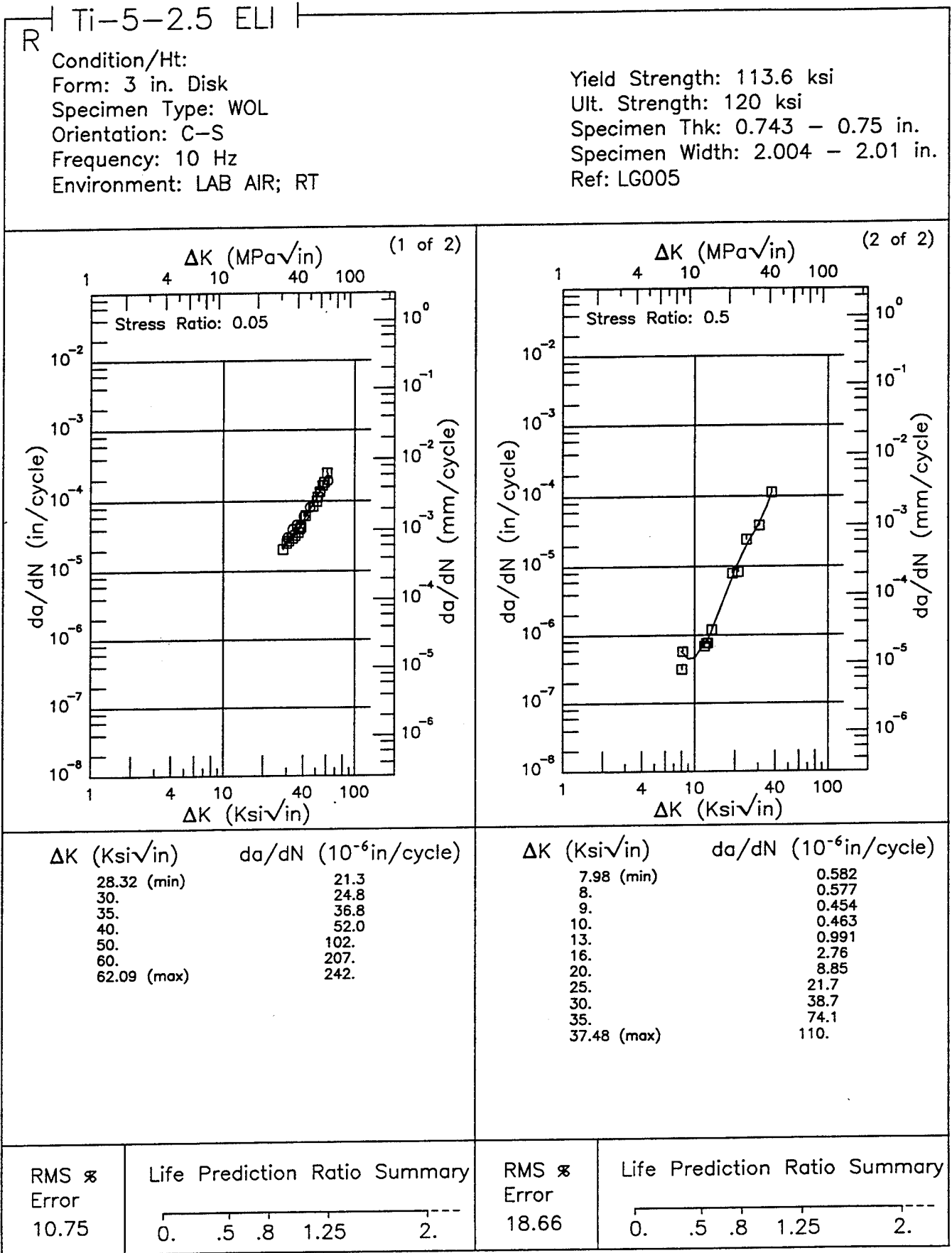


Figure 6.10.3.1.11

Ti-5-2.5 ELI R

Condition/Ht:
 Form: 3 in. Disk
 Specimen Type: WOL
 Orientation: C-S
 Frequency: 0.1 Hz
 Environment: LAB AIR; RT

Yield Strength: 113.6 ksi
 Ult. Strength: 120 ksi
 Specimen Thk: 0.749 in.
 Specimen Width: 2.01 in.
 Ref: LG005

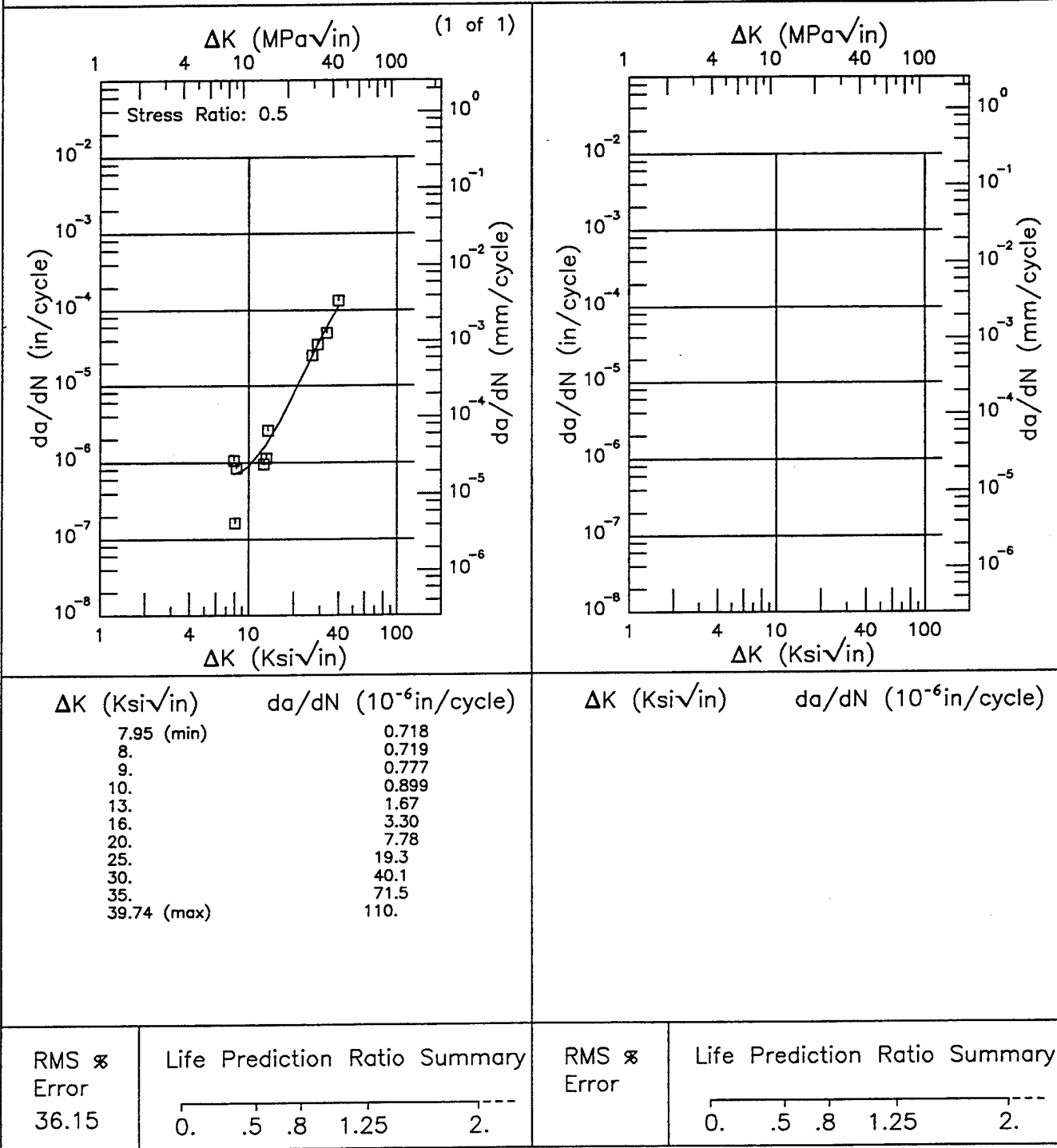


Figure 6.10.3.1.12

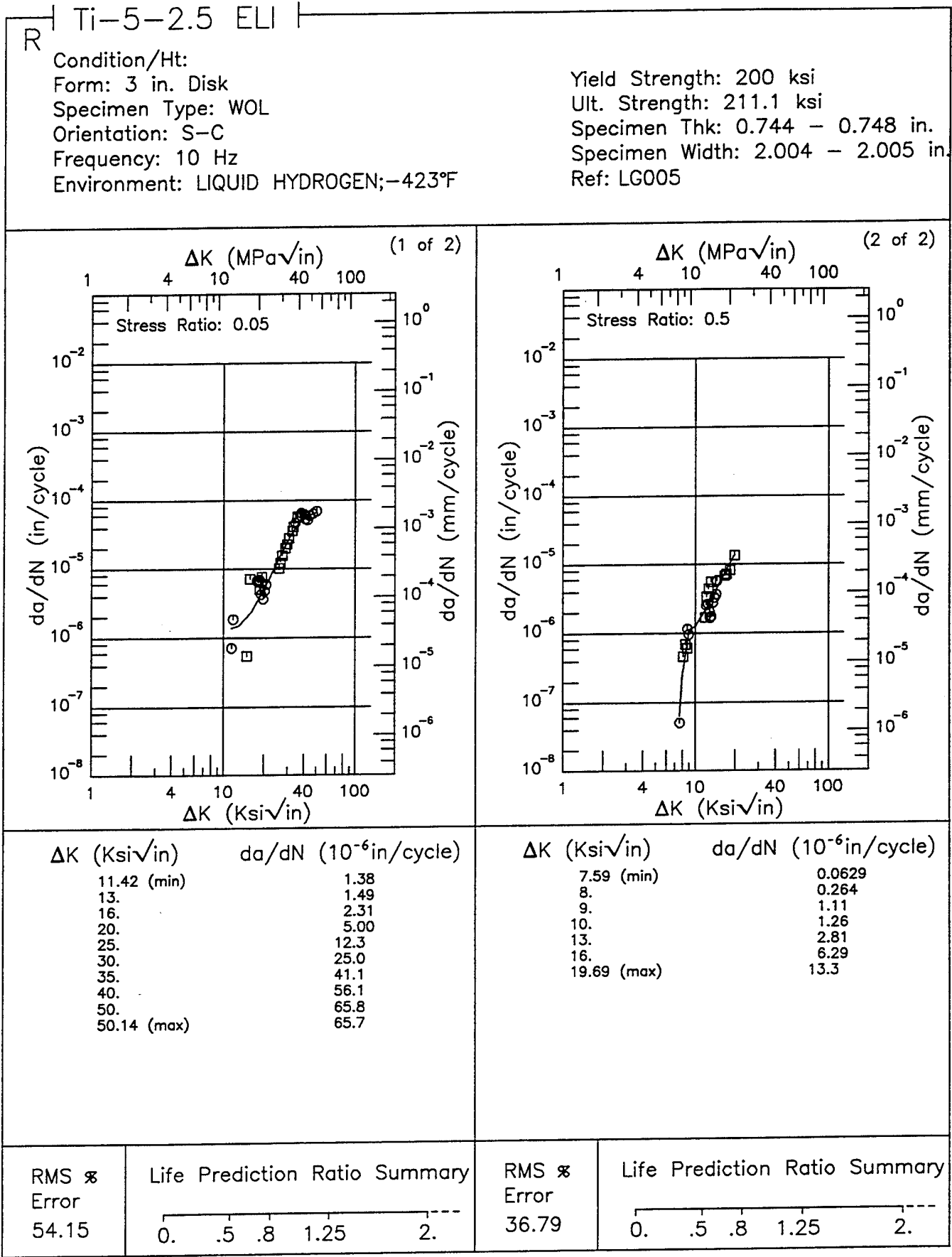
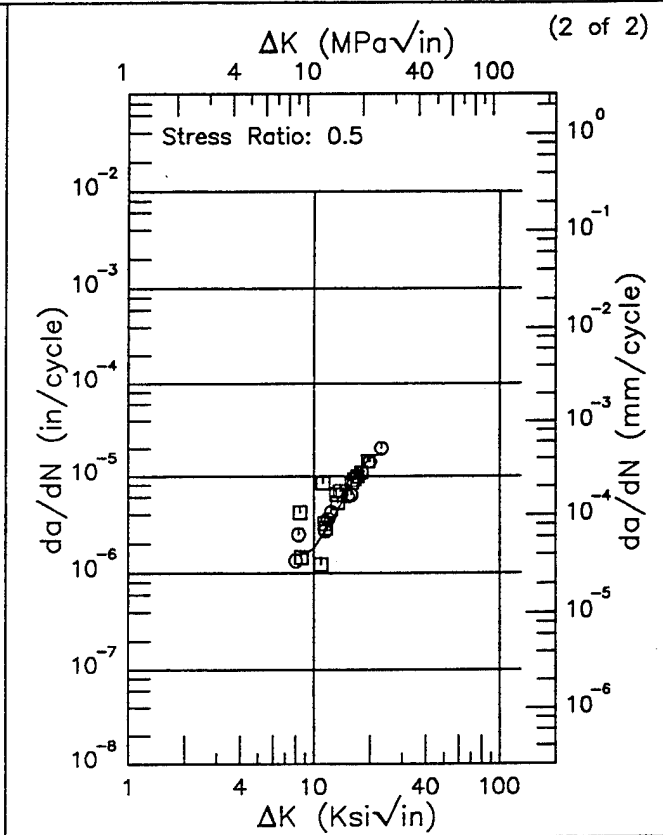
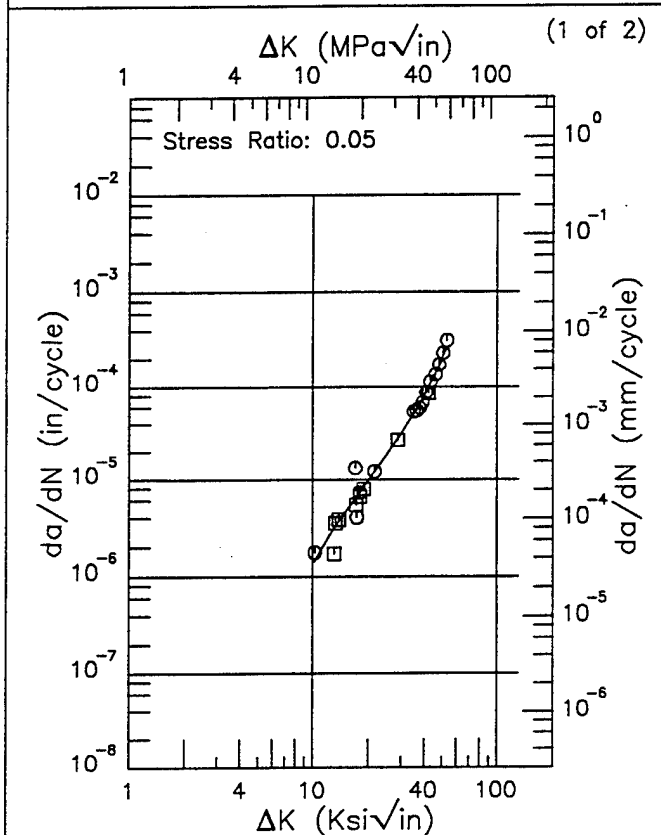


Figure 6.10.3.1.13

Ti-5-2.5 ELI R

Condition/Ht:
 Form: 3 in. Disk
 Specimen Type: WOL
 Orientation: S-C
 Frequency: 0.1 Hz
 Environment: LAB AIR; RT

Yield Strength: 113.6 ksi
 Ult. Strength: 120 ksi
 Specimen Thk: 0.743 - 0.748 in.
 Specimen Width: 2.005 - 2.01 in.
 Ref: LG005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.22 (min)	1.42
13.	3.01
16.	5.41
20.	9.90
25.	18.3
30.	31.4
35.	51.5
40.	82.3
50.	201.
52.94 (max)	258.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.97 (min)	1.71
8.	1.70
9.	1.59
10.	1.80
13.	3.92
16.	8.52
20.	15.8
22.95 (max)	17.5

RMS % Error	Life Prediction Ratio Summary
27.79	

RMS % Error	Life Prediction Ratio Summary
60.56	

Figure 6.10.3.1.14

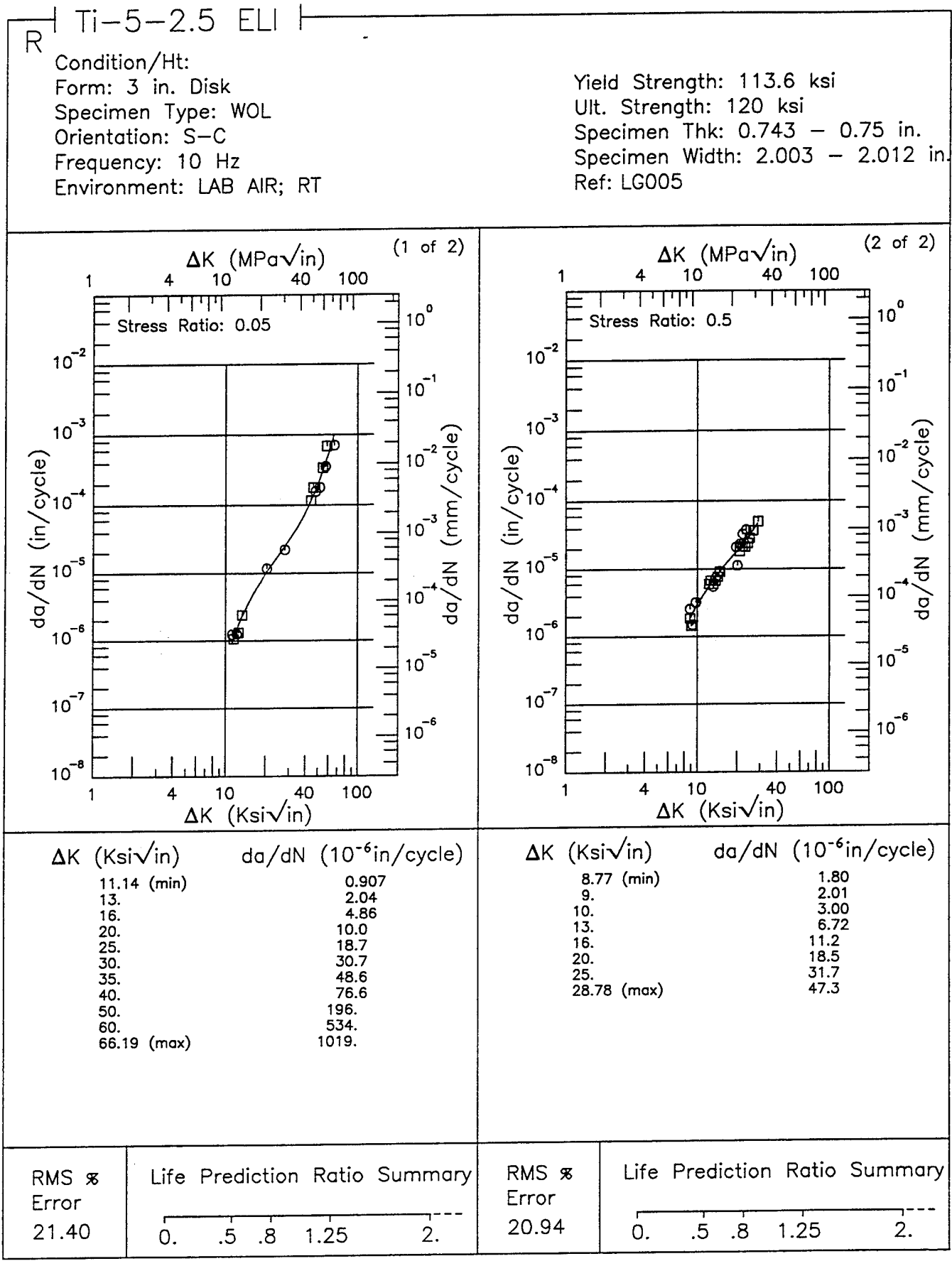
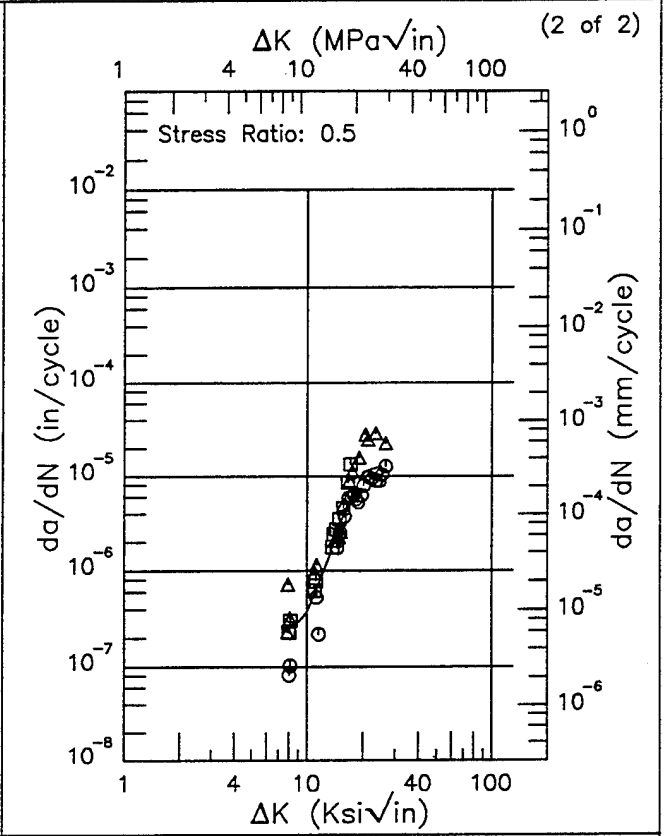
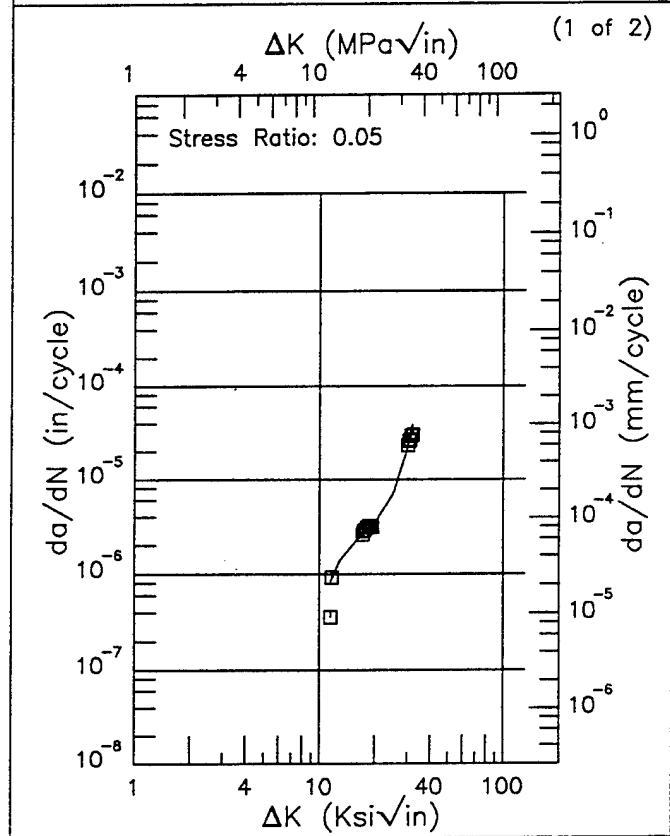


Figure 6.10.3.1.15

Ti-5-2.5 ELI R

Condition/Ht:
 Form: 3 in. Disk
 Specimen Type: WOL
 Orientation: R-S
 Frequency: 10 Hz
 Environment: LIQUID HYDROGEN; -423°F

Yield Strength: 200 ksi
 Ult. Strength: 211.1 ksi
 Specimen Thk: 0.745 - 0.749 in.
 Specimen Width: 1.997 - 2.004 in.
 Ref: LG005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
11.47 (min)	0.833
13.	1.39
16.	2.29
20.	3.43
25.	7.02
30.	21.7
32.00 (max)	38.3

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.79 (min)	0.306
8.	0.291
9.	0.291
10.	0.371
13.	1.29
16.	4.23
20.	11.1
25.	12.7
26.40 (max)	10.9

RMS % Error 17.32	Life Prediction Ratio Summary
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RMS % Error 58.06	Life Prediction Ratio Summary
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Figure 6.10.3.1.16

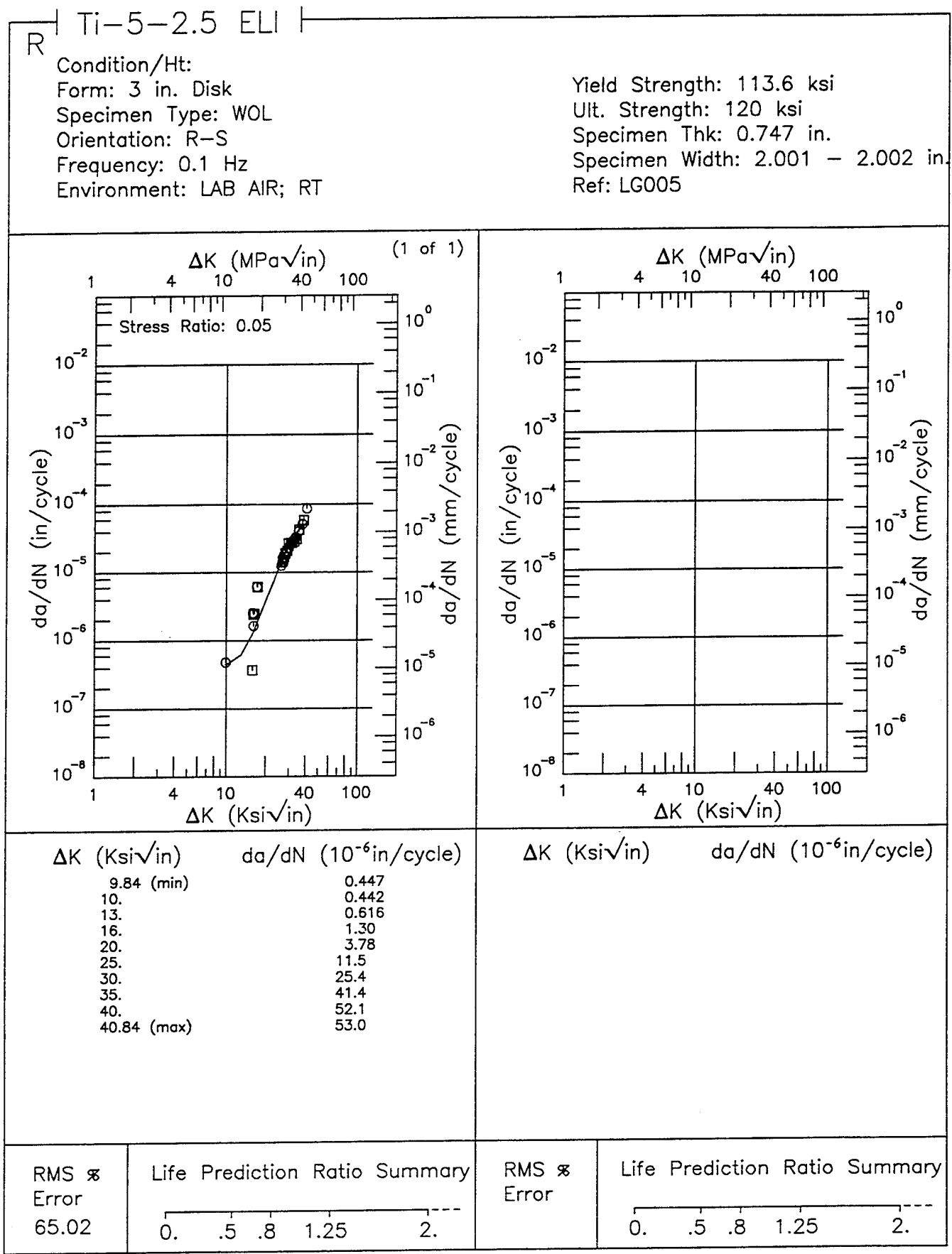
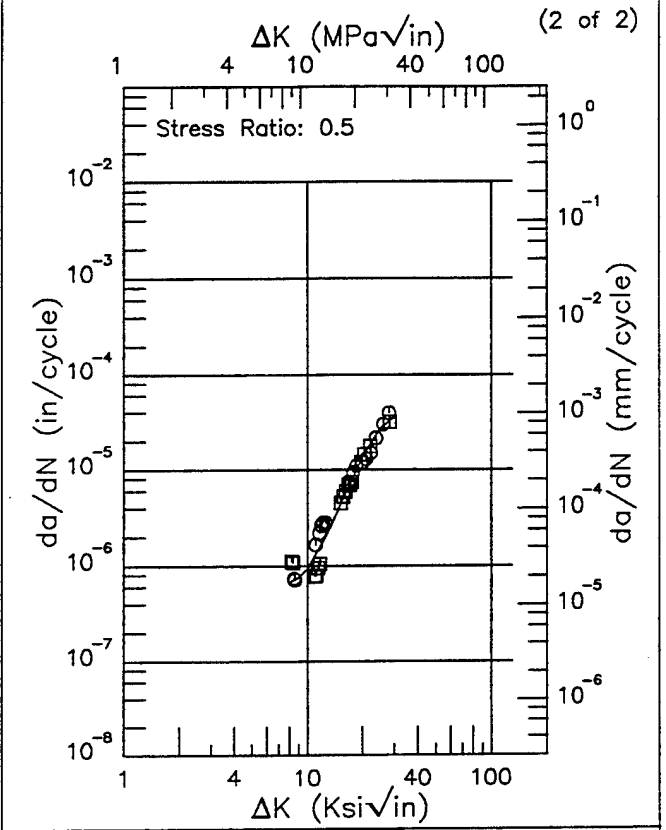
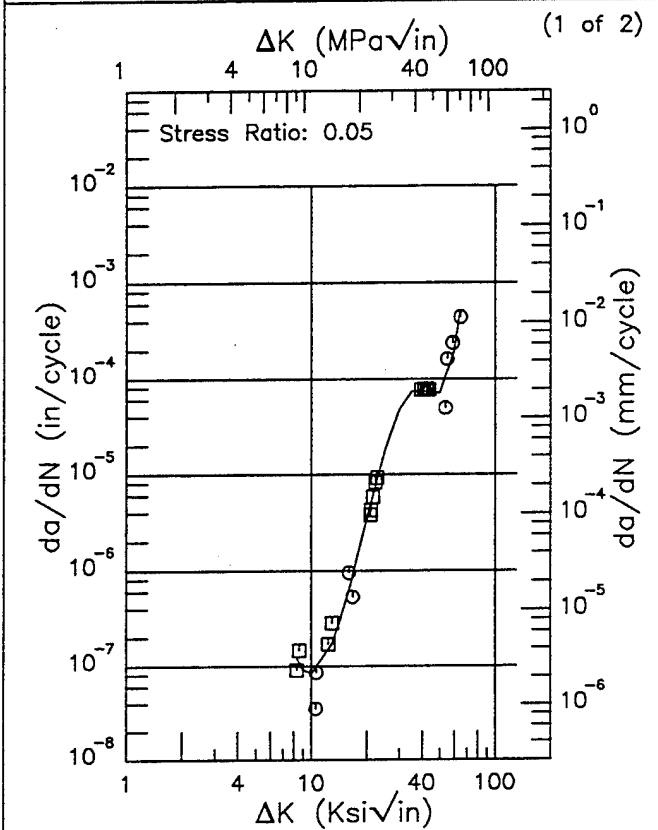


Figure 6.10.3.1.17

Ti-5-2.5 ELI R

Condition/Ht:
 Form: 3 in. Disk
 Specimen Type: WOL
 Orientation: R-S
 Frequency: 10 Hz
 Environment: LAB AIR; RT

Yield Strength: 113.6 ksi
 Ult. Strength: 120 ksi
 Specimen Thk: 0.744 - 0.748 in.
 Specimen Width: 1.996 - 2.002 in.
 Ref: LG005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.28 (min)	0.125
9.	0.0929
10.	0.0830
13.	0.175
16.	0.639
20.	3.48
25.	17.6
30.	47.5
35.	73.4
40.	73.7
50.	71.9
60.	212.
64.25 (max)	515.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.18 (min)	0.691
9.	0.757
10.	0.933
13.	2.25
16.	5.44
20.	13.6
25.	26.6
28.08 (max)	31.9

RMS % Error 31.83	Life Prediction Ratio Summary
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RMS % Error 27.93	Life Prediction Ratio Summary
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Figure 6.10.3.1.18

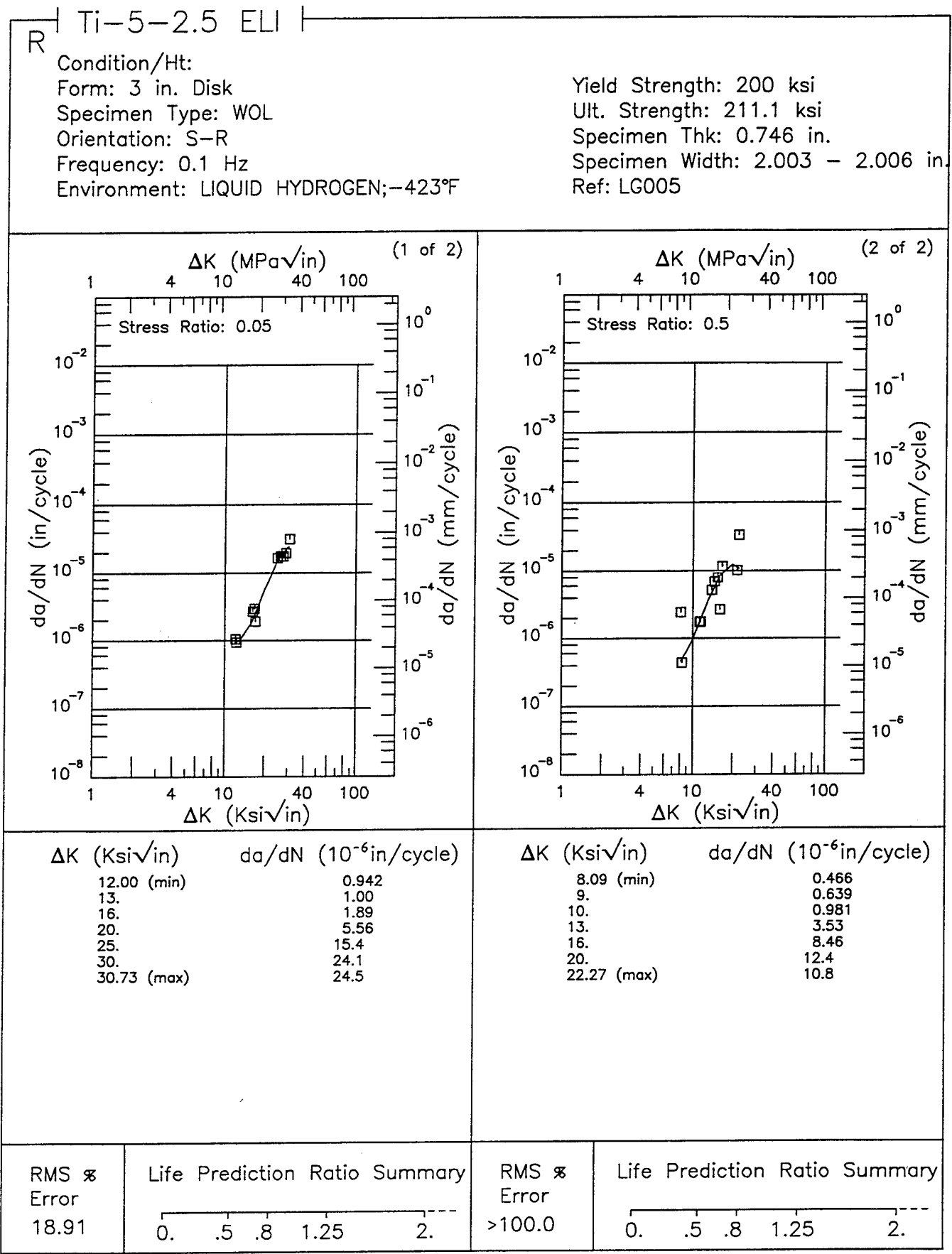
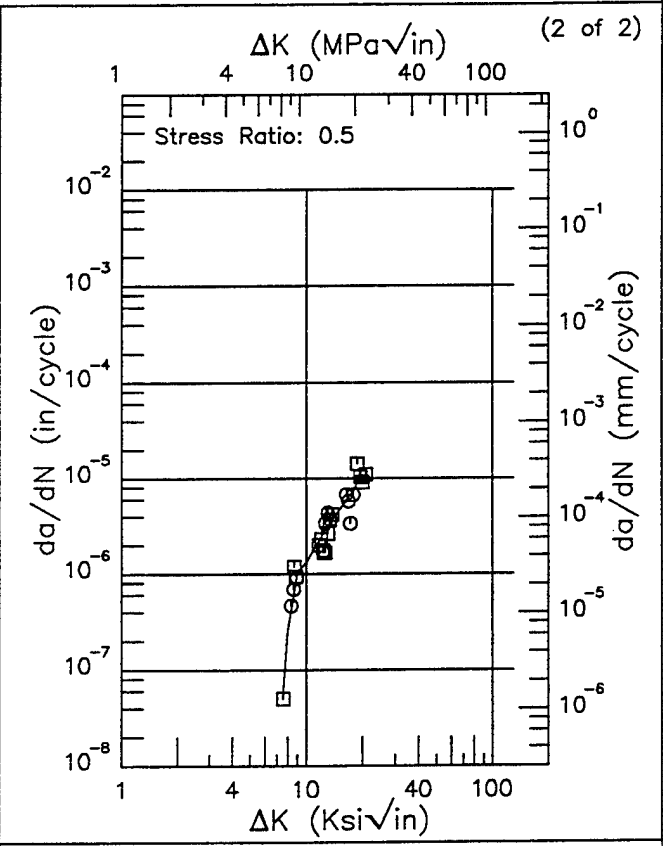
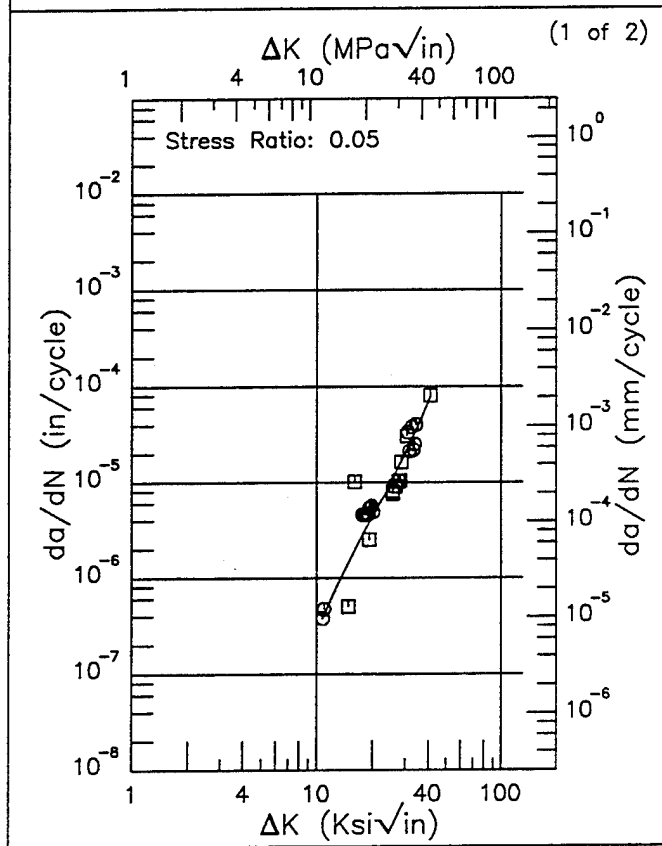


Figure 6.10.3.1.19

Ti-5-2.5 ELI R

Condition/Ht:
 Form: 3 in. Disk
 Specimen Type: WOL
 Orientation: S-R
 Frequency: 10 Hz
 Environment: LIQUID HYDROGEN; -423°F

Yield Strength: 200 ksi
 Ult. Strength: 211.1 ksi
 Specimen Thk: 0.744 - 0.748 in.
 Specimen Width: 2.004 - 2.011 in.
 Ref: LG005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.74 (min)	0.376
13.	0.863
16.	1.95
20.	4.42
25.	10.0
30.	20.2
35.	38.4
40.	70.6
40.98 (max)	79.4

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.49 (min)	0.0505
8.	0.261
9.	1.02
10.	1.32
13.	3.13
16.	5.42
20.	11.0
20.73 (max)	9.71

RMS % Error	Life Prediction Ratio Summary
90.92	0. .5 .8 1.25 2.

RMS % Error	Life Prediction Ratio Summary
32.25	0. .5 .8 1.25 2.

Figure 6.10.3.1.20

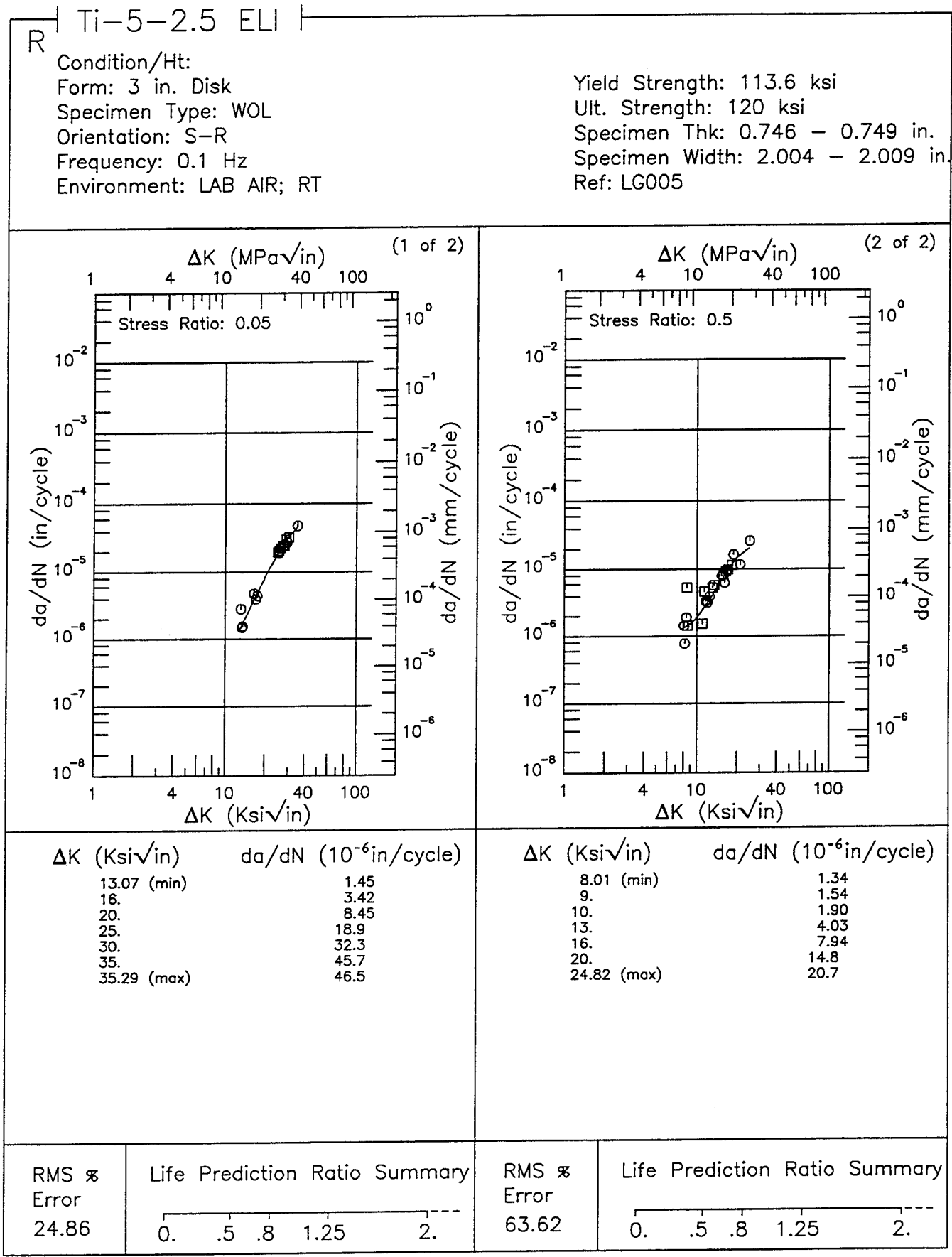
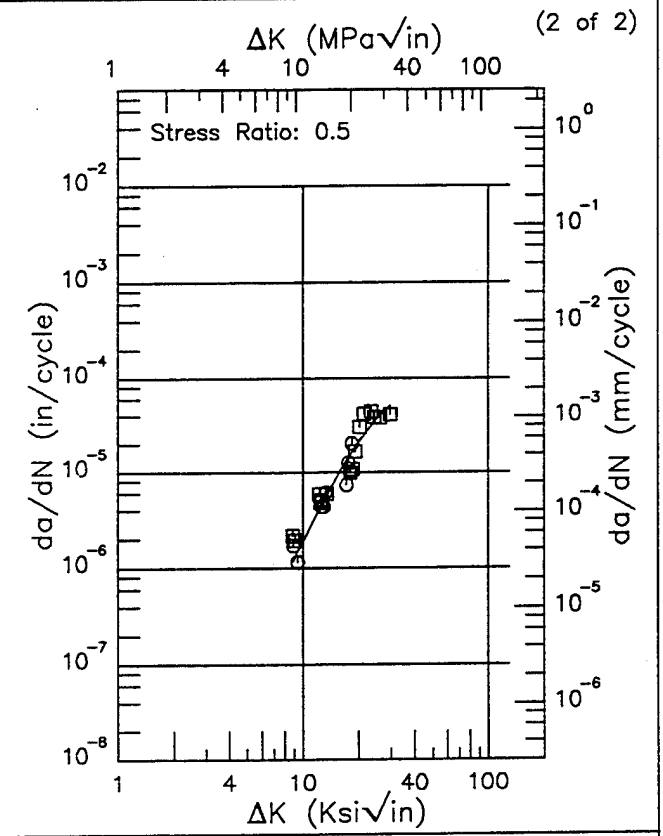
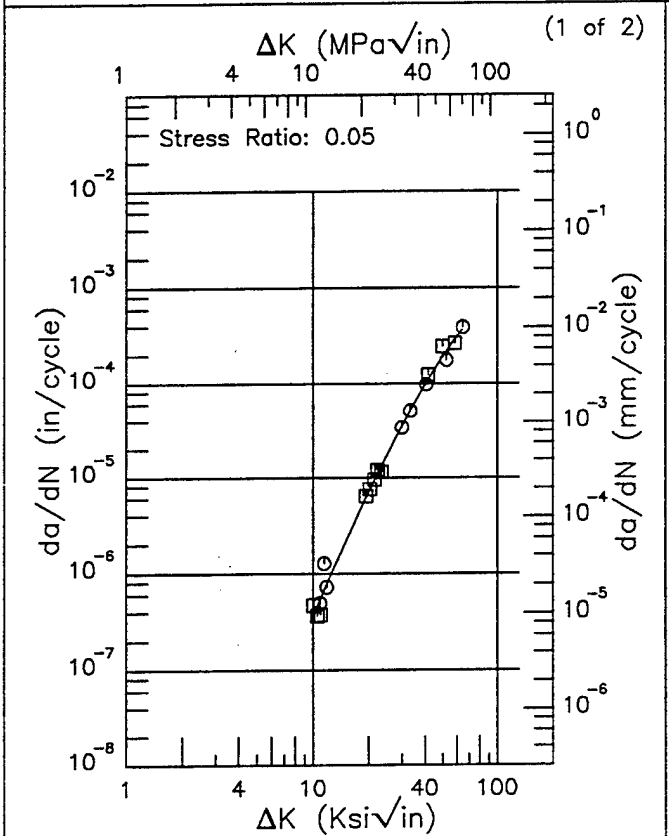


Figure 6.10.3.1.21

Ti-5-2.5 ELI R

Condition/Ht:
 Form: 3 in. Disk
 Specimen Type: WOL
 Orientation: S-R
 Frequency: 10 Hz
 Environment: LAB AIR; RT

Yield Strength: 113.6 ksi
 Ult. Strength: 120 ksi
 Specimen Thk: 0.745 - 0.747 in.
 Specimen Width: 2.004 - 2.01 in.
 Ref: LG005



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
10.06 (min)	0.381
13.	1.15
16.	2.79
20.	7.15
25.	17.7
30.	35.6
35.	62.3
40.	98.0
50.	194.
60.	315.
64.24 (max)	369.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
8.80 (min)	1.16
9.	1.26
10.	1.89
13.	5.06
16.	10.5
20.	20.8
25.	36.7
29.54 (max)	50.6

RMS % Error 25.23	Life Prediction Ratio Summary
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RMS % Error 34.86	Life Prediction Ratio Summary
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Figure 6.10.3.1.22

Ti-5Al-2.5Sn

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TABLE 6.11.1.2.1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-5Al-2.5Sn AT ROOM TEMPERATURE

ORIENTATION: L-T

ENVIRONMENT: 3.5% NaCl

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi/ $\sqrt{\text{in}}$)						
				2.6	6.0	10.0	20.0	60.0	100.0	
ANNEALED	SHEET	0.1	50							
		0.1-0.67	30-55			8.79	28.64	160.44		
		0.67	55		0.29	7.38				

TABLE 6.11.1.2.2

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-5Al-2.5Sn AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: Distilled Water

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
ANNEALED	SHEET	0.1	30						
		0.1	60				11.77		
		0.07	55			0.25	3.38		

Ti-5Al-2.5Sn

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TABLE 6.11.1.2.3

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-5Al-2.5Sn AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: Dry Argon

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-5} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
ANNEALED	SHEET	0.1	30				5.66	94.72	
		0.1	60				4.77		
		0.67	55-58.3			0.03	0.27		

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TABLE 6.11.1.2.4

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-5Al-2.5Sn AT ROOM TEMPERATURE

ORIENTATION: L-T

ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi $\sqrt{\text{in}}$)					
				2.5	5.0	10.0	20.0	50.0	100.0
ANNEALED	SHEET	0.1	30				11.56	123.8	
		0.1	50				11.69		
		0.67	55			0.15	2.17		

Ti-5Al-2.5Sn

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TABLE 6.11.1.2.5

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-5Al-2.5Sn AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: 3.5% NaCl

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi/ \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
ANNEALED	SHEET	0.1	30							
		0.1	50				24.35			
		0.67	54.2-55			0.7	12.52			

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TABLE 6.11.1.2.6

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
Ti-5Al-2.5Sn AT ROOM TEMPERATURE

ORIENTATION: T-L

ENVIRONMENT: Distilled Water

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi $\sqrt{\text{in}}$)					
				2.5	5.0	10.0	20.0	50.0	100.0
ANNEALED	SHEET	0.1	30				12.51	130.45	
		0.1	50				11.97		
		0.07	55			0.36	3.74		

Ti-5Al-2.5Sn

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TABLE 6.11.1.2.7

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-5Al-2.5Sn AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: Dry Argon

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
ANNEALED	SHEET	0.1	30							
		0.1	50-53.3				5.38	114.1		
		0.67	55			0.49				

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TABLE 6.11.1.2.8

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-5Al-2.5Sn AT ROOM TEMPERATURE

ORIENTATION: T-L

ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi $\sqrt{\text{in}}$)					
				2.5	6.0	10.0	20.0	50.0	100.0
ANNEALED	SHEET	0.1	30				11.18	140.04	
		0.1	50				11.76		
		0.67	55-58.3		0.16	3.08			

Ti-5Al-2.5Sn

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TABLE 6.11.2.2

TI-5AL-2.5SN K _C																					
CONDITION HEAT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN		CRACK LENGTH		GROSS STRESS			K _{app}			K _C			DATE	REFER	
	FORM	THICK (in.)				WIDTH (in.)	THICK (in.)	INIT (in.)	FINAL (in.)	ONSET (KSI)	MAX (KSI)	K _{app} MEAN	K _{app} STAN DEV	K _C MEAN	K _C STAN DEV	K _C (KSI/in.)	K _C MEAN	K _C STAN DEV			
BUCKLING OF CRACK EDGES RETAINED																					
ANNEALED	Sheet	0.06	1.000	0.062	0.080	---	---	131.10	46.66	---	---	---	---	---	---	---	---	---	---	1971	80104
		0.06	1.000	0.063	0.410	---	---	63.90	57.34	---	---	---	---	---	---	---	---	---	---	1971	80104
		0.06	1.000	0.064	0.480	---	---	54.60	55.63	55.1	6.0	---	---	---	---	---	---	---	---	1971	80104
		0.06	1.000	0.064	0.170	---	---	115.60	60.77	---	---	---	---	---	---	---	---	---	---	1971	80104
ANNEALED	Sheet	0.10	1.000	0.111	0.120	---	---	161.30	70.66*	---	---	---	---	---	---	---	---	---	---	1971	80104
		0.10	1.000	0.113	0.360	---	---	90.30	73.90	74.2	0.4	---	---	---	---	---	---	---	---	1971	80104
		0.10	1.000	0.113	0.240	---	---	116.90	74.44	---	---	---	---	---	---	---	---	---	---	1971	80104
ANNEALED	Sheet	0.06	2.000	0.062	0.080	---	---	147.10	52.20	---	---	---	---	---	---	---	---	---	---	1971	80104
		0.06	2.000	0.063	0.140	---	---	123.00	57.86	---	---	---	---	---	---	---	---	---	---	1971	80104
		0.06	2.000	0.063	0.790	---	---	46.80	57.80	59.0	4.9	---	---	---	---	---	---	---	---	1971	80104
		0.06	2.000	0.063	0.180	---	---	115.10	61.51	---	---	---	---	---	---	---	---	---	---	1971	80104
		0.06	2.000	0.064	0.420	---	---	76.30	65.39	---	---	---	---	---	---	---	---	---	---	1971	80104
		0.10	2.000	0.112	0.130	---	---	158.30	71.72*	---	---	---	---	---	---	---	---	---	---	---	1971
ANNEALED	Sheet	0.10	2.000	0.113	0.380	---	---	97.30	76.89	77.6	1.1	---	---	---	---	---	---	---	---	1971	80104
		0.10	2.000	0.114	0.270	---	---	119.00	78.38	---	---	---	---	---	---	---	---	---	---	1971	80104
		0.02	3.000	0.018	0.500	---	---	110.10	99.28	---	---	---	---	---	---	---	---	---	---	1967	68968
ANNEALED	Sheet	0.02	3.000	0.019	0.540	---	---	103.60	97.37	---	---	---	---	---	---	---	---	---	---	1967	68968
		0.02	3.000	0.019	1.040	---	---	69.00	95.36	95.4	2.8	115.6	4.9	122.41	111.85	---	---	---	---	1967	68968
		0.02	3.000	0.019	0.150	---	---	165.40	80.41*	---	---	---	---	---	---	---	---	---	---	1967	68968

* NOTE: NET SECTION STRESS EXCEEDS 90% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

TABLE 6.11.2.2 (CONTINUED)

TI-5AL-2.5SN K _C																					
CONDITION HEAT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K _{app}			K _C			DATE	REFER		
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	INIT (in.) 2a _i	FINAL (in.) 2a _f	ONSET (Ksi) σ _s	MAX (Ksi) σ _{max}	K _{app} (Ksi√in.)	K _{app} MEAN	STAN DEV	K _C (Ksi√in.)	K _C MEAN	STAN DEV				
BUCKLING OF CRACK EDGES RESTRAINED																					
ANNEALED Cont'd	Sheet Cont'd	0.02	203.5	3.000	0.019	1.000	---	---	69.30	93.33	---	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	3.000	0.019	1.020	1.240	---	---	72.20	98.51	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	3.000	0.019	0.320	0.500	---	---	129.90	92.75	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	3.000	0.019	0.340	0.670	---	---	127.90	94.22	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	3.000	0.019	0.120	0.300	---	---	181.00	78.66*	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	3.000	0.019	0.510	0.730	---	---	107.60	98.06	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	3.000	0.020	0.150	0.330	---	---	172.70	83.96*	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	3.000	0.020	0.270	0.470	---	---	138.90	90.91	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	3.000	0.020	1.030	1.260	---	---	68.60	94.20	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	3.000	0.064	0.990	---	---	---	48.50	64.89	---	---	---	---	---	---	---	1971	80104	
ANNEALED	Sheet	0.10	211.0	3.000	0.116	1.060	---	---	55.20	76.77	---	---	---	---	---	---	---	1971	80104		
ANNEALED	Sheet	0.10	211.0	3.000	0.116	0.990	---	---	58.30	78.01	---	---	---	---	---	---	---	1971	80104		
ANNEALED	Sheet	0.02	203.5	6.000	0.017	2.000	2.390	---	50.10	95.42	---	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	6.000	0.018	2.010	2.230	---	49.00	93.63	---	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	6.000	0.018	0.130	0.250	---	---	181.60	82.09*	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	6.000	0.018	0.130	0.220	---	---	191.90	86.74*	---	---	---	---	---	---	---	1967	68968	
		0.02	203.5	6.000	0.018	0.490	0.800	---	---	105.90	93.29	---	---	---	---	---	---	---	---	1967	68968
		0.02	203.5	6.000	0.018	1.000	1.330	---	---	73.70	93.98	---	---	---	---	---	---	---	---	1967	68968
		0.02	203.5	6.000	0.018	2.000	2.420	---	---	48.80	92.95	---	---	---	---	---	---	---	---	1967	68968
		0.02	203.5	6.000	0.018	2.000	2.420	---	---	48.80	92.95	---	---	---	---	---	---	---	---	1967	68968

* NOTE: NET SECTION STRESS EXCEEDS 80% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

TI-5Al-2.5Sn

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TABLE 6.11.2.2 (CONTINUED)

TI-5AL-2.5SN K _C																			
CONDITION HEAT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K _{app}			K _C			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	INIT (in.) S _a	FINAL (in.) S _f	ONSET (Ksi) σ _o	MAX (Ksi) σ _{max}	K _{app} (Ksi/in.)	K _{app} MEAN	STAN DEV	K _C (Ksi/in.)	K _C MEAN	STAN DEV		
ANNEALED Cont'd	Sheet Cont'd	0.02	203.5	6.000	0.018	0.490	0.640	...	110.50	97.34	...	111.58	1967	68968			
		0.02	203.5	6.000	0.019	0.240	0.640	...	156.40	96.12	...	157.92*	1967	68968			
		0.02	203.5	6.000	0.019	0.500	0.720	...	109.10	97.10	...	117.07	1967	68968			
		0.02	203.5	6.000	0.019	1.000	1.340	...	74.10	94.49	...	110.94	Cont'd	Cont'd	1967	68968			
		0.02	203.5	6.000	0.019	0.250	0.530	...	153.30	96.17	...	140.55*	1967	68968			
		0.02	203.5	6.000	0.019	1.020	1.210	...	71.40	92.02	...	100.98	1967	68968			
		0.02	203.5	6.000	0.019	0.250	0.610	...	157.20	98.62*	...	154.87*	1967	68968			
		0.02	203.5	12.000	0.018	0.240	0.650	...	154.40	94.82	...	156.30*	1967	68968			
		0.02	203.5	12.000	0.018	0.500	0.690	...	103.60	91.91	...	108.08	1967	68968			
		0.02	203.5	12.000	0.018	0.490	0.620	...	104.30	91.60	...	103.10	1967	68968			
		0.02	203.5	12.000	0.018	0.890	1.190	...	76.00	94.68	...	104.54	1967	68968			
		0.02	203.5	12.000	0.019	4.000	4.450	...	35.40	95.35	...	102.42	1967	68968			
ANNEALED	Sheet	0.02	203.5	12.000	0.019	0.130	0.380	...	178.40	80.62*	...	137.92*	1967	68968			
		0.02	203.5	12.000	0.019	0.490	0.660	...	108.60	95.38	...	110.76	107.0	9.0	1967	68968			
		0.02	203.5	12.000	0.019	2.000	2.410	...	52.10	93.96	...	110.76	1967	68968			
		0.02	203.5	12.000	0.019	1.000	1.170	...	71.30	89.75	...	110.76	1967	68968			
		0.02	203.5	12.000	0.019	2.000	2.460	...	51.20	92.34	...	103.97	1967	68968			
		0.02	203.5	12.000	0.020	4.000	4.040	...	39.20	105.59	...	103.34	1967	68968			
		0.02	203.5	12.000	0.020	0.120	0.260	...	163.90	79.85*	...	106.28	1967	68968			
		0.02	203.5	12.000	0.020	0.120	0.260	...	163.90	79.85*	...	117.56*	1967	68968			
		0.02	203.5	12.000	0.020	0.120	0.260	...	163.90	79.85*	...	117.56*	1967	68968			
		0.02	203.5	12.000	0.020	0.120	0.260	...	163.90	79.85*	...	117.56*	1967	68968			
		0.02	203.5	12.000	0.020	0.120	0.260	...	163.90	79.85*	...	117.56*	1967	68968			

HUCKLING OF CRACK SIDES RESTRAINED

* NOTE: NET SECTION STRESS EXCEEDS 80% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

TABLE 6.11.2.2 (CONCLUDED)

Ti-5Al-2.5Sn K _C																				
CONDITION HEAT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K _{app}			K _C			DATE	REFER	
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	INIT (in.) 2a _i	FINAL (in.) 2a _f	ONSET (Ksi) σ _o	MAX (Ksi) σ _{max}	K _{app} (Ksi√in.)	K _{app} MEAN	STAN DEV	K _C (Ksi√in.)	K _C MEAN	STAN DEV			
ANNEALED Cont'd	Sheet Cont'd	0.02	12,000	0.020	4,020	4,350	...	35.20	95.12	100.26	1967	68968	BUCKLING OF CRACK EDGES RETAINED	Cont'd	Cont'd	Cont'd	Cont'd	Cont'd	1967	68968
		0.02	12,000	0.020	0.240	0.470	153.20	94.09	131.76	1967	68968									
		0.02	12,000	0.020	0.120	0.310	181.10	76.63*	126.43*	1967	68968									
		0.02	12,000	0.021	0.240	0.400	151.30	92.92	120.01	1967	68968									
		0.02	12,000	0.021	0.990	1.370	70.90	88.79	104.85	1967	68968									
		0.02	12,000	0.021	2,000	2,330	51.60	93.06	101.08	1967	68968									
		0.02	15,930	0.016	4,980	5,480	32.80	97.69	103.92	1967	68968									
		0.02	15,930	0.019	4,980	5,280	29.20	86.97	90.29	1967	68968									
		0.06	3,000	0.064	1,000	...	44.90	60.47	...	1971	80104									
		0.02	15,940	0.018	5,000	5,400	44.60	133.16	139.93	1967	68968									
		0.02	16,230	0.020	4,980	5,610	44.80	133.11	143.73	1967	68968									
		0.20	13,880	0.202	4,550	...	68.70	196.87*	...	1966	66218									
0.02	15,950	0.018	5,080	6,780	54.50	163.94	200.71*	1967	68968											
0.02	16,390	0.018	5,070	6,660	58.70	176.16	211.66*	1967	68968											
0.02	12,000	0.015	1,130	1,390	64.70	86.67	96.40	1966	66103											
0.02	12,000	0.015	0.330	0.710	112.40	80.96	118.96	1966	66103											
0.03	12,000	0.036	0.280	1,070	129.00	85.58	168.07	1966	66103											
0.03	12,000	0.036	1,200	1,840	73.70	101.81	127.14	1966	66103											

* NOTE: NET SECTION STRESS EXCEEDS 80% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

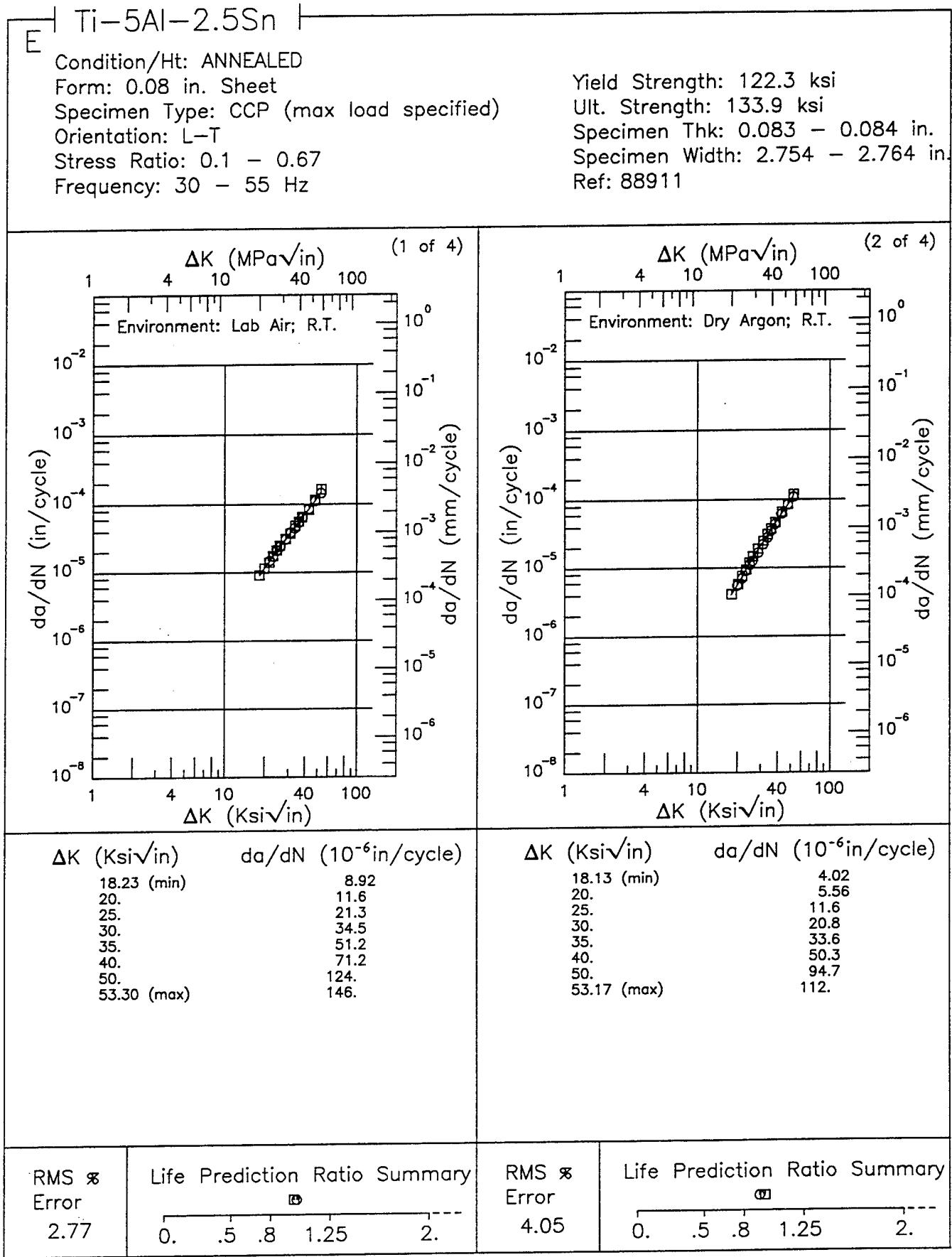
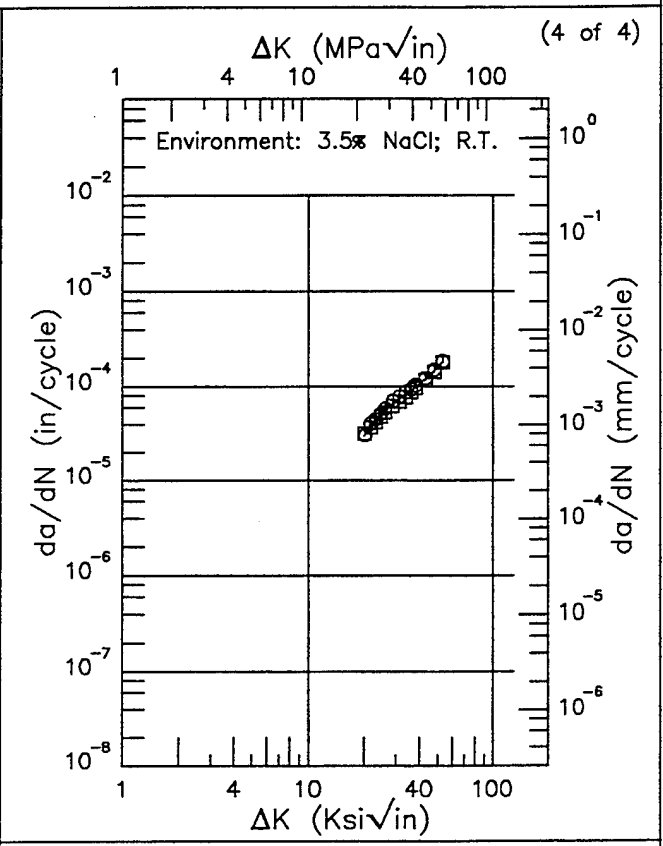
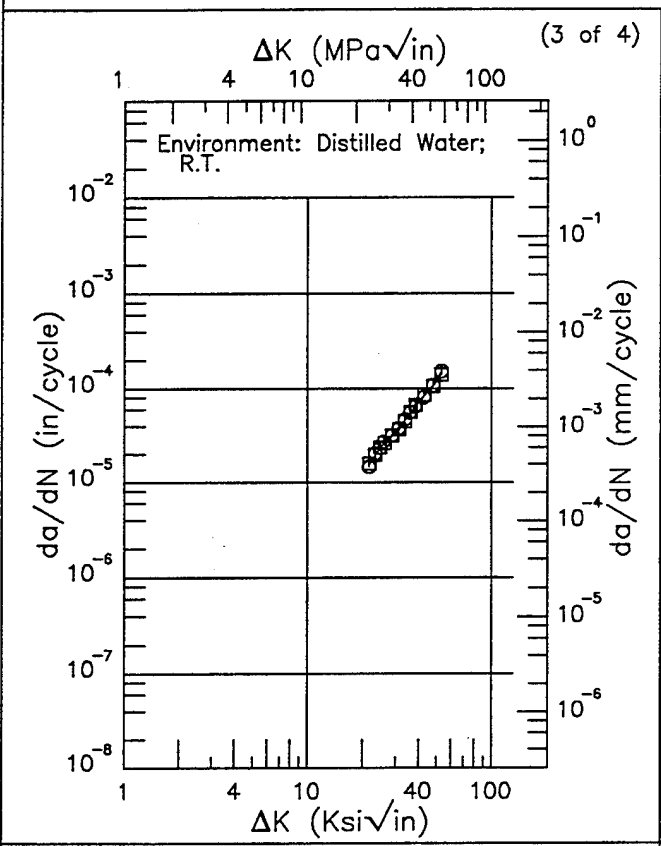


Figure 6.11.3.1.1

Ti-5Al-2.5Sn E

Condition/Ht: ANNEALED
 Form: 0.08 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.1
 Frequency: 30 Hz

Yield Strength: 122.3 ksi
 Ult. Strength: 133.9 ksi
 Specimen Thk: 0.083 - 0.084 in.
 Specimen Width: 2.754 - 2.764 in.
 Ref: 88911



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
21.49 (min)	15.4
25.	24.0
30.	34.9
35.	50.7
40.	72.6
50.	119.
53.14 (max)	152.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
20.00 (min)	29.9
25.	51.4
30.	69.2
35.	86.0
40.	105.
50.	157.
53.17 (max)	179.

RMS % Error 3.20	Life Prediction Ratio Summary □
---------------------	------------------------------------

RMS % Error 4.68	Life Prediction Ratio Summary □□
---------------------	-------------------------------------

Figure 6.11.3.1.1 (Concluded)

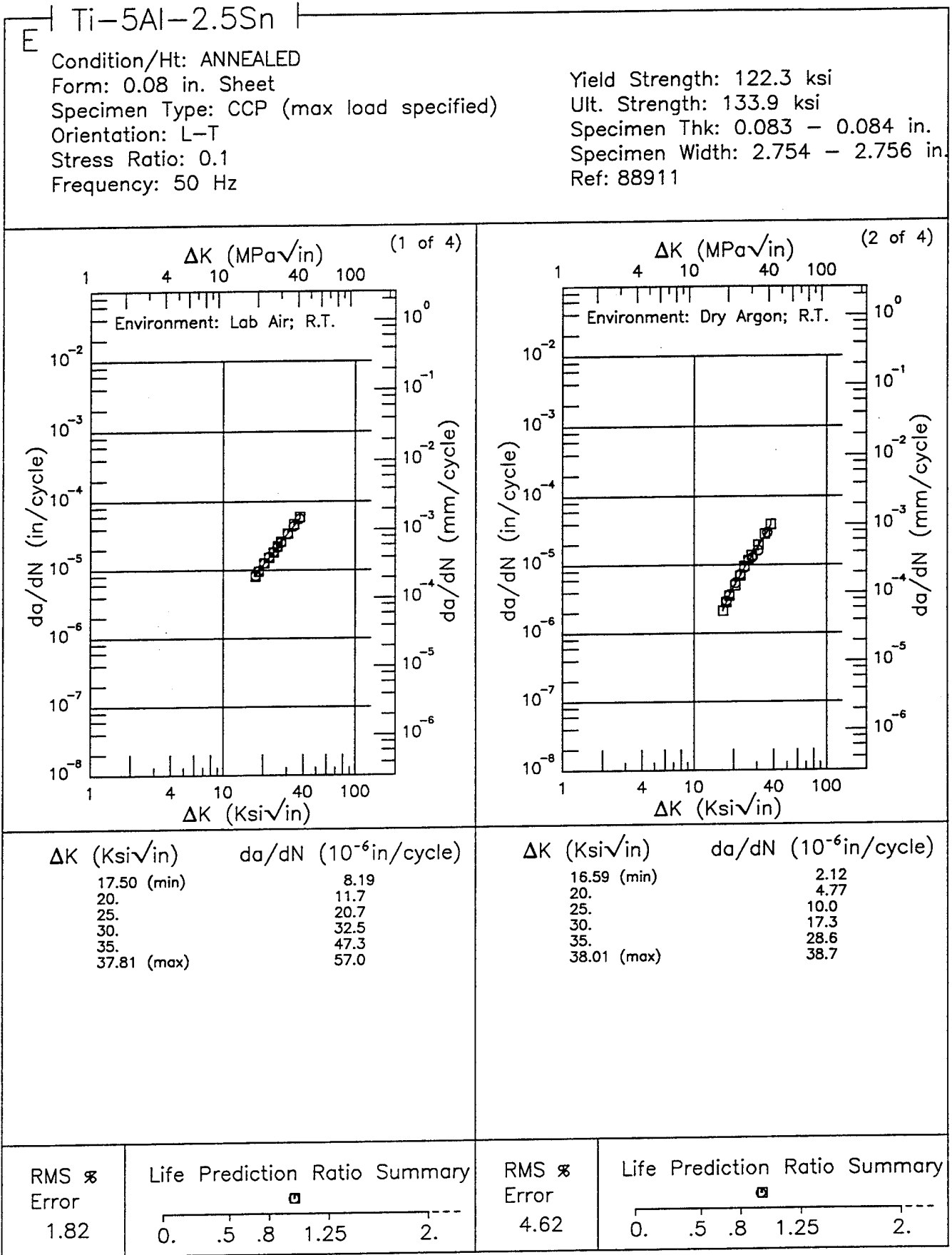


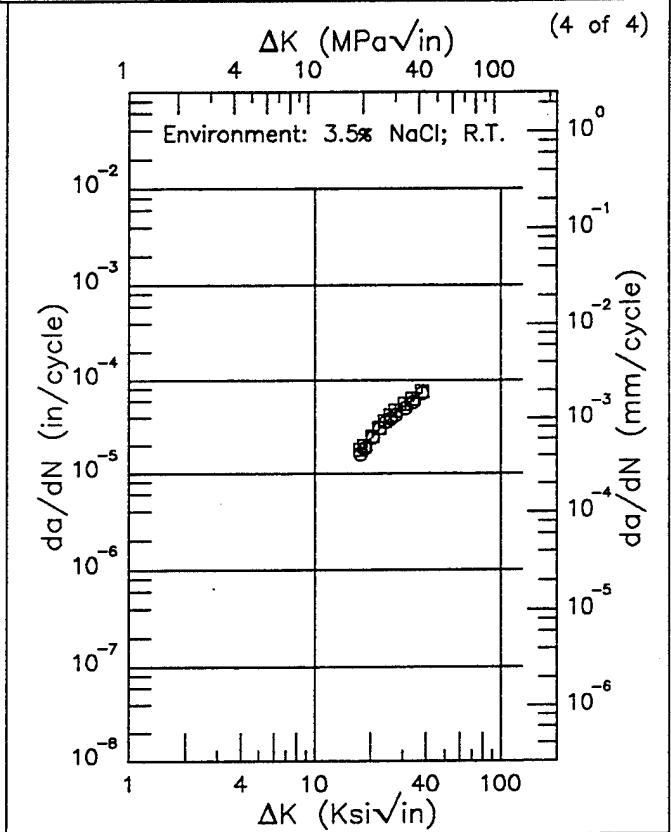
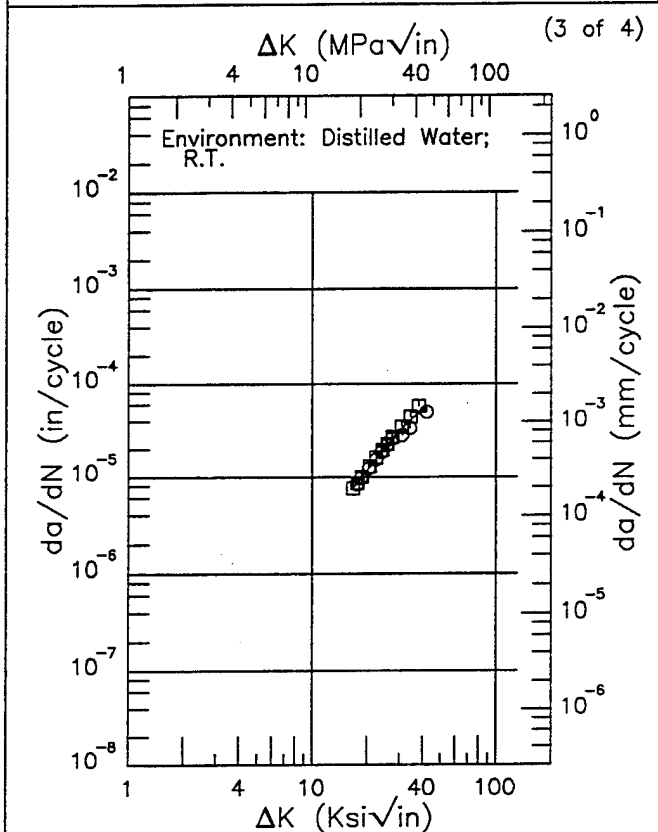
Figure 6.11.3.1.2

Ti-5Al-2.5Sn

E

Condition/Ht: ANNEALED
 Form: 0.08 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.1
 Frequency: 50 Hz

Yield Strength: 122.3 ksi
 Ult. Strength: 133.9 ksi
 Specimen Thk: 0.083 - 0.084 in.
 Specimen Width: 2.754 - 2.756 in.
 Ref: 88911



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.62 (min)	7.49
20.	11.8
25.	20.4
30.	31.1
35.	42.4
40.	52.7
42.00 (max)	56.3

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
17.57 (min)	16.3
20.	23.5
25.	37.5
30.	50.5
35.	64.2
38.00 (max)	73.6

RMS % Error	Life Prediction Ratio Summary
7.49	

RMS % Error	Life Prediction Ratio Summary
4.55	

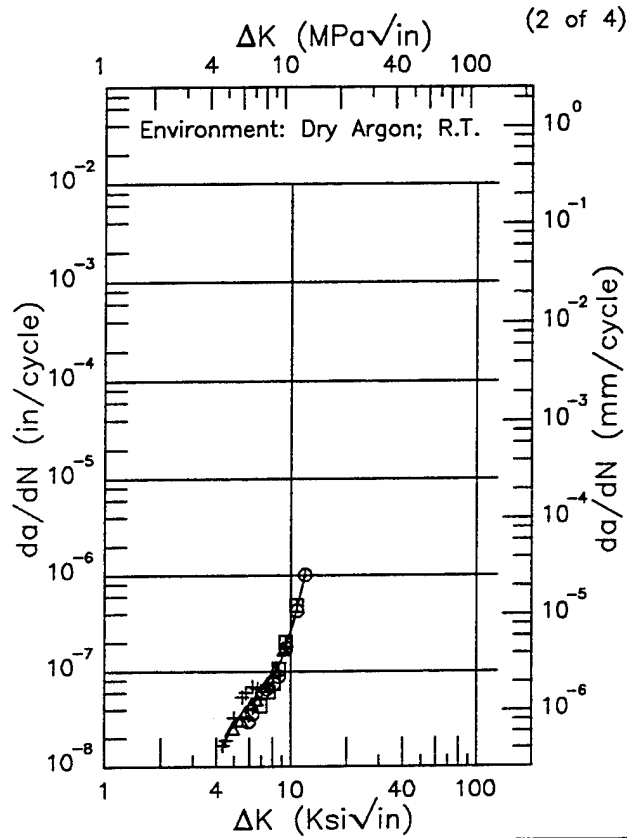
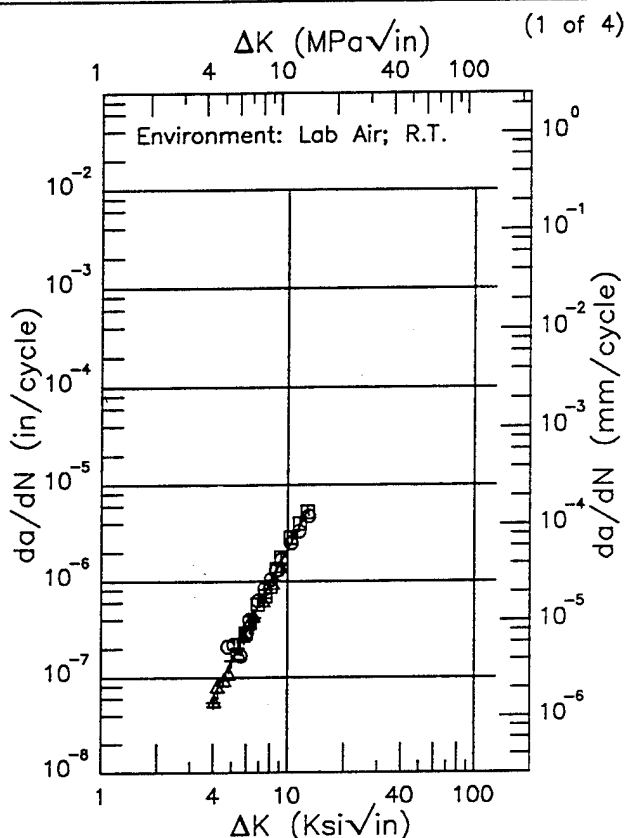
Figure 6.11.3.1.2 (Concluded)

Ti-5Al-2.5Sn

E

Condition/Ht: ANNEALED
 Form: 0.08 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.67
 Frequency: 55 - 58.3 Hz

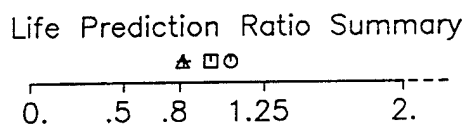
Yield Strength: 122.3 ksi
 Ult. Strength: 133.9 ksi
 Specimen Thk: 0.083 - 0.085 in.
 Specimen Width: 2.752 - 2.756 in.
 Ref: 88911



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
3.99 (min)	0.0604
4.	0.0610
5.	0.151
6.	0.310
7.	0.563
8.	0.937
9.	1.46
10.	2.17
12.86 (max)	5.55

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
4.31 (min)	0.0150
5.	0.0301
6.	0.0443
7.	0.0559
8.	0.0796
9.	0.138
10.	0.269
11.87 (max)	0.999

RMS \times
 Error
 14.65



RMS \times
 Error
 18.52

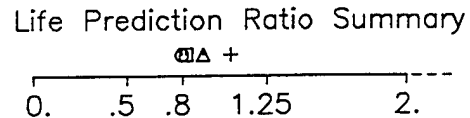


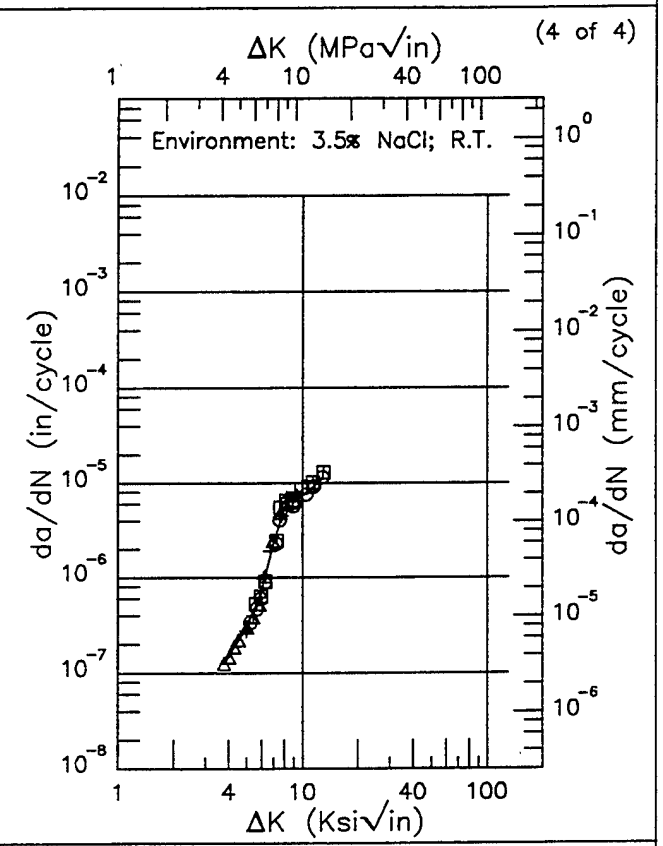
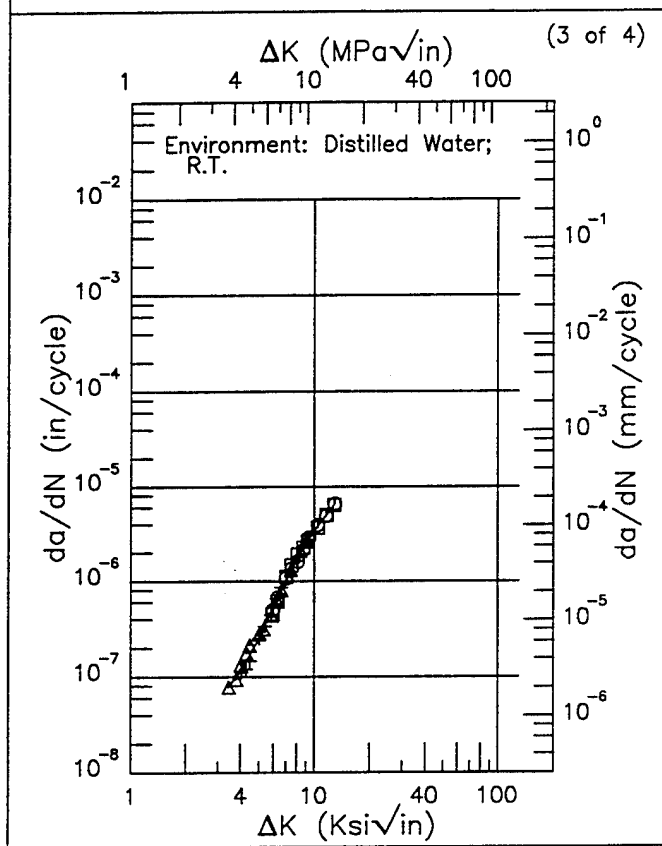
Figure 6.11.3.1.3

Ti-5Al-2.5Sn

E

Condition/Ht: ANNEALED
 Form: 0.08 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.67
 Frequency: 55 - 58.3 Hz

Yield Strength: 122.3 ksi
 Ult. Strength: 133.9 ksi
 Specimen Thk: 0.083 - 0.085 in.
 Specimen Width: 2.752 - 2.756 in.
 Ref: 88911



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
3.44 (min)	0.0777
3.5	0.0804
4.	0.113
5.	0.249
6.	0.530
7.	1.01
8.	1.68
9.	2.49
10.	3.38
12.87 (max)	6.36

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
3.77 (min)	0.124
4.	0.149
5.	0.295
6.	0.723
7.	2.29
8.	5.45
9.	7.25
10.	7.38
12.93 (max)	11.3

RMS % Error	Life Prediction Ratio Summary
8.05	

RMS % Error	Life Prediction Ratio Summary
12.68	

Figure 6.11.3.1.3 (Concluded)

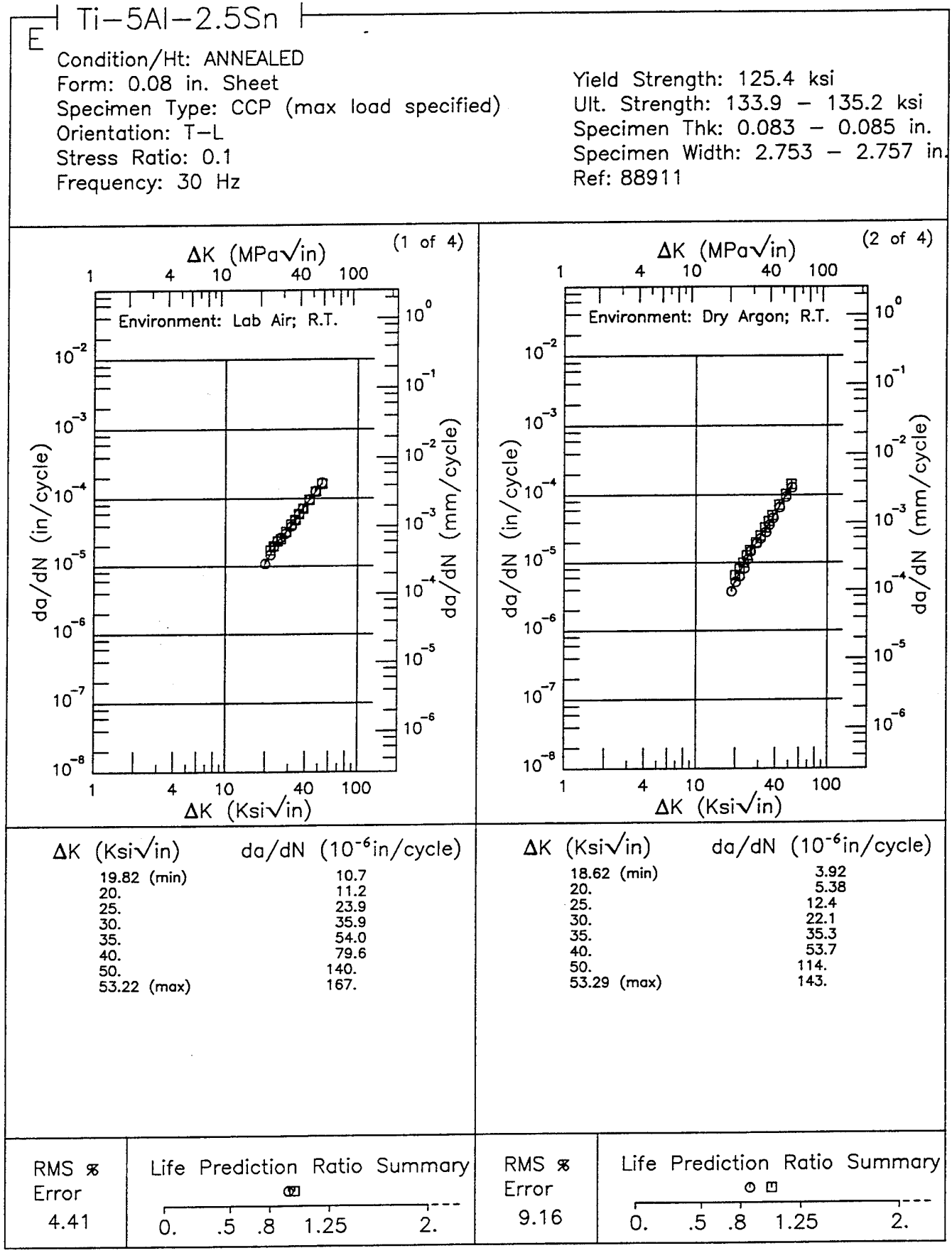
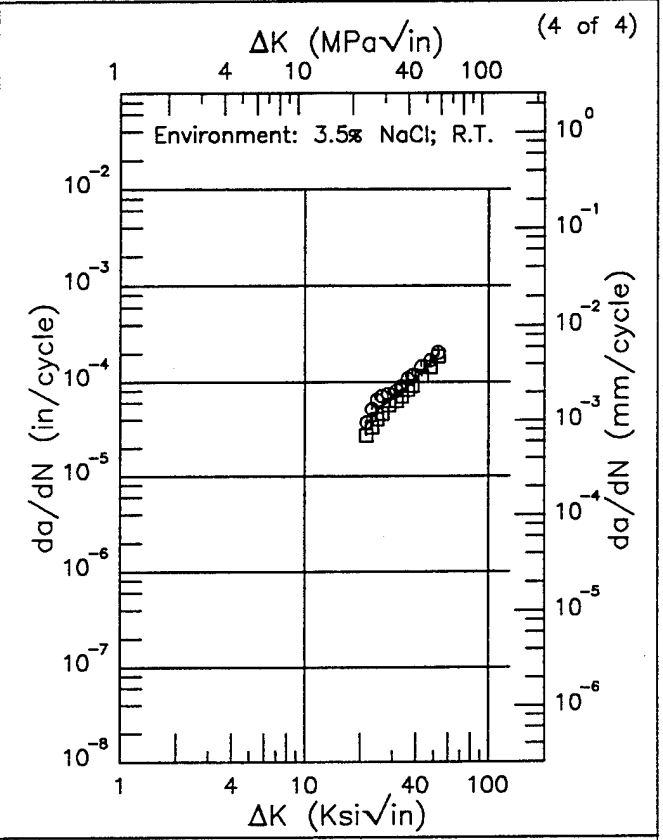
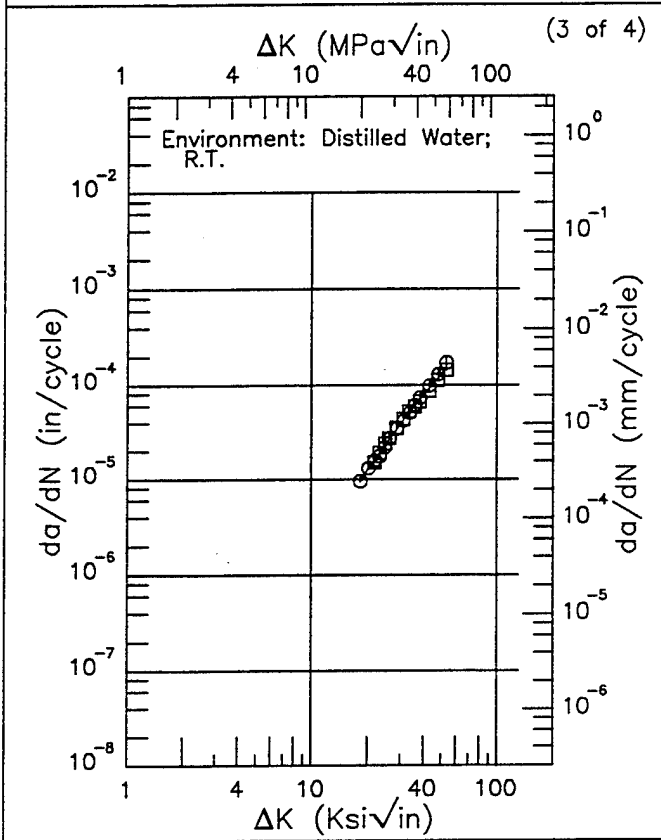


Figure 6.11.3.1.4

Ti-5Al-2.5Sn E

Condition/Ht: ANNEALED
 Form: 0.08 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: T-L
 Stress Ratio: 0.1
 Frequency: 30 Hz

Yield Strength: 125.4 ksi
 Ult. Strength: 133.9 - 135.2 ksi
 Specimen Thk: 0.083 - 0.085 in.
 Specimen Width: 2.753 - 2.757 in.
 Ref: 88911



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
18.18 (min)	9.47
20.	12.5
25.	23.6
30.	38.6
35.	57.3
40.	79.0
50.	130.
53.08 (max)	148.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
21.52 (min)	34.7
25.	50.0
30.	68.0
35.	85.7
40.	109.
50.	177.
53.13 (max)	201.

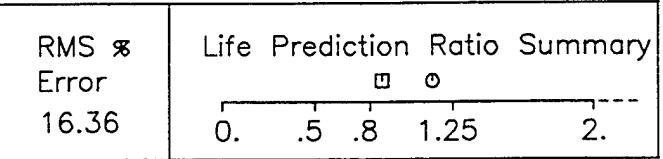
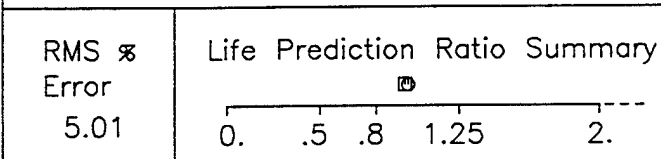


Figure 6.11.3.1.4 (Concluded)

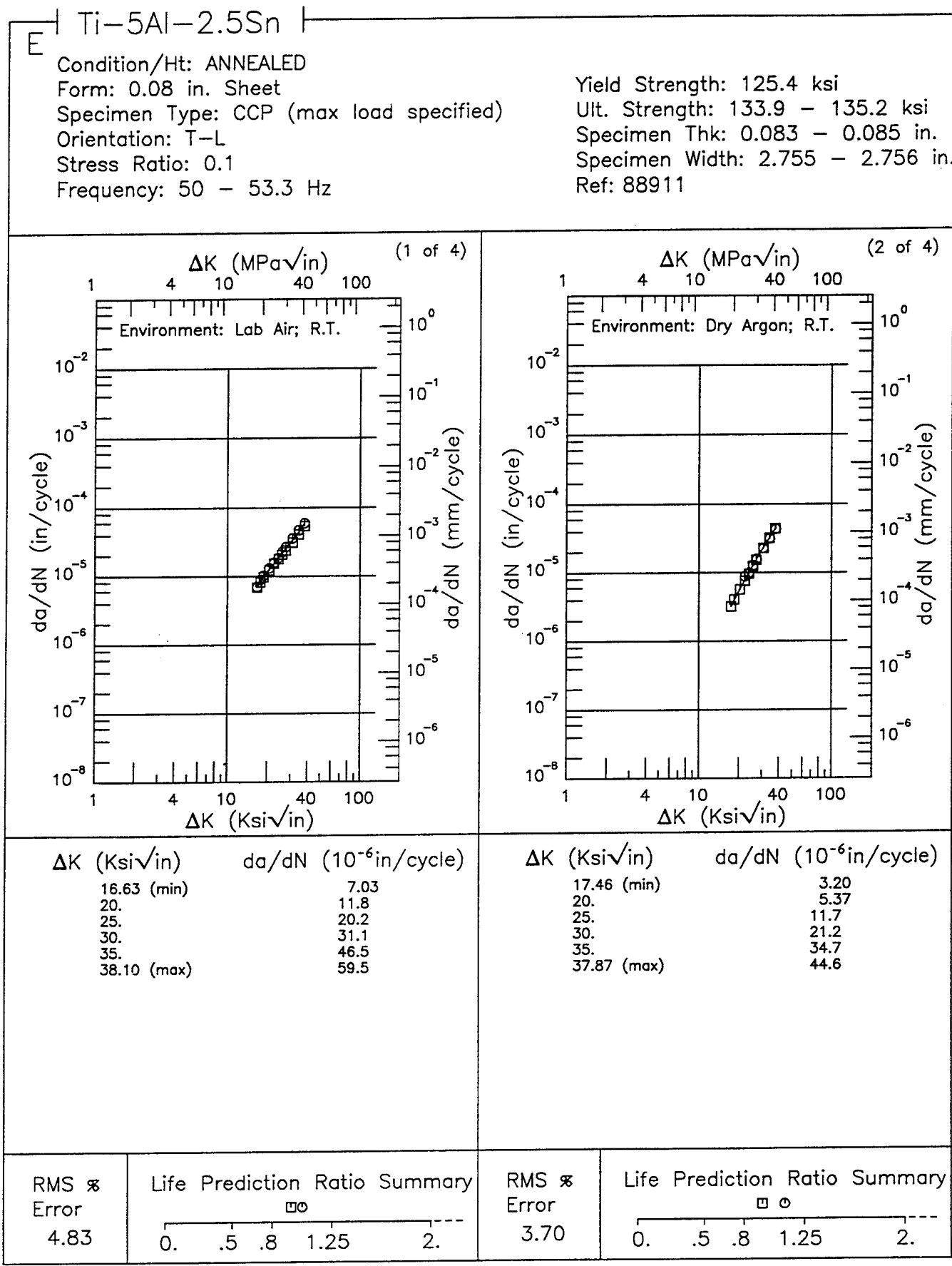
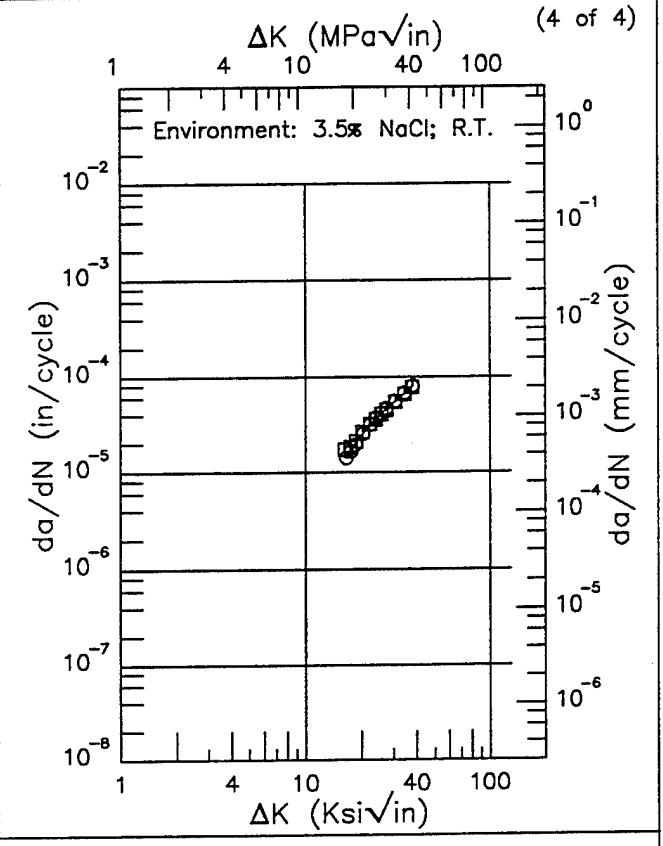
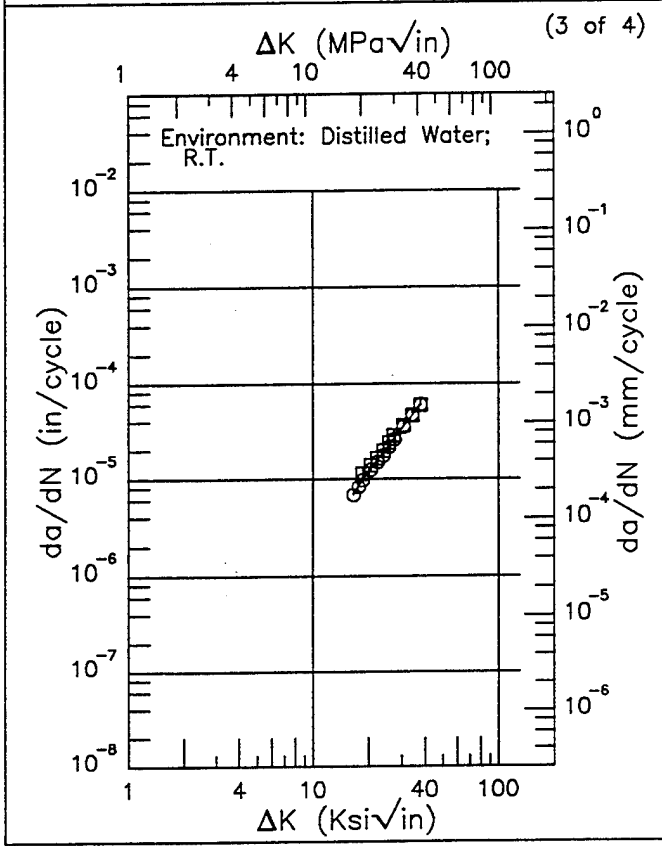


Figure 6.11.3.1.5

Ti-5Al-2.5Sn E

Condition/Ht: ANNEALED
 Form: 0.08 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: T-L
 Stress Ratio: 0.1
 Frequency: 50 - 53.3 Hz

Yield Strength: 125.4 ksi
 Ult. Strength: 133.9 - 135.2 ksi
 Specimen Thk: 0.083 - 0.085 in.
 Specimen Width: 2.755 - 2.756 in.
 Ref: 88911



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
16.57 (min)	7.01
20.	12.0
25.	21.3
30.	33.4
35.	49.2
37.79 (max)	59.9

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
16.28 (min)	14.2
20.	24.4
25.	39.3
30.	53.4
35.	69.6
37.84 (max)	79.0

RMS % Error	Life Prediction Ratio Summary
5.76	

RMS % Error	Life Prediction Ratio Summary
5.01	

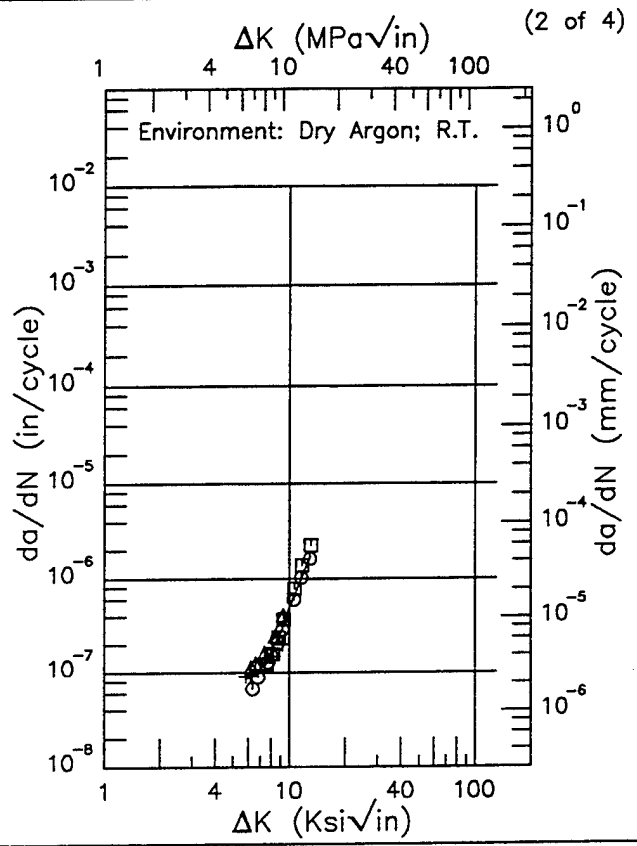
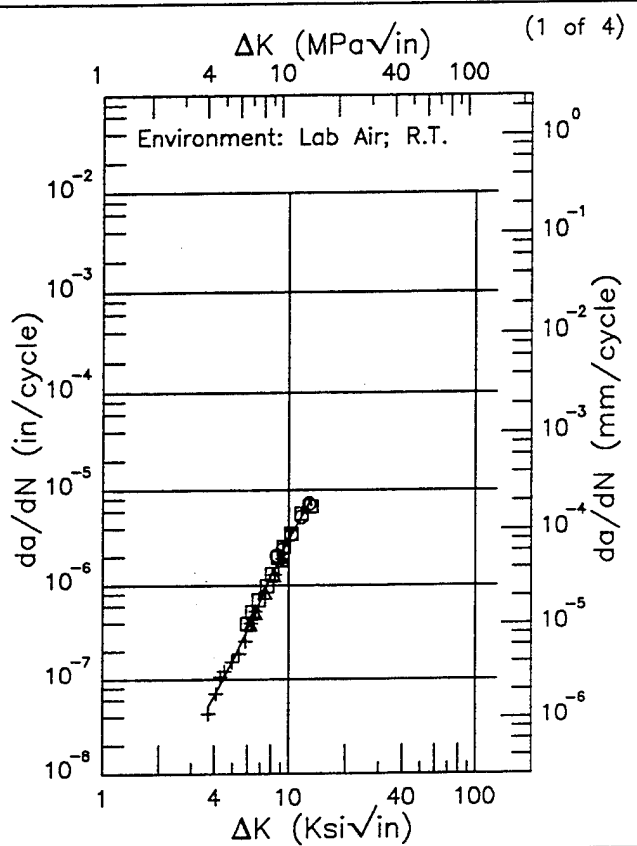
Figure 6.11.3.1.5 (Concluded)

Ti-5Al-2.5Sn

E

Condition/Ht: ANNEALED
 Form: 0.08 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: T-L
 Stress Ratio: 0.67
 Frequency: 54.2 - 58.3 Hz

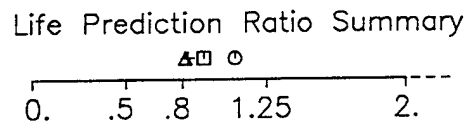
Yield Strength: 125.4 ksi
 Ult. Strength: 133.9 - 135.2 ksi
 Specimen Thk: 0.083 - 0.085 in.
 Specimen Width: 2.753 - 2.757 in.
 Ref: 88911



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
3.69 (min)	0.0501
4.	0.0662
5.	0.158
6.	0.347
7.	0.685
8.	1.23
9.	2.02
10.	3.08
13.	7.57
13.07 (max)	7.68

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
5.83 (min)	0.0860
6.	0.0875
7.	0.110
8.	0.163
9.	0.271
10.	0.490
13.	1.73
13.02 (max)	1.73

RMS % Error
11.37



RMS % Error
18.25

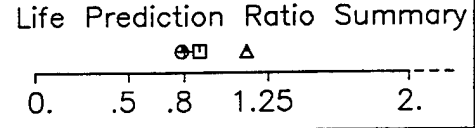


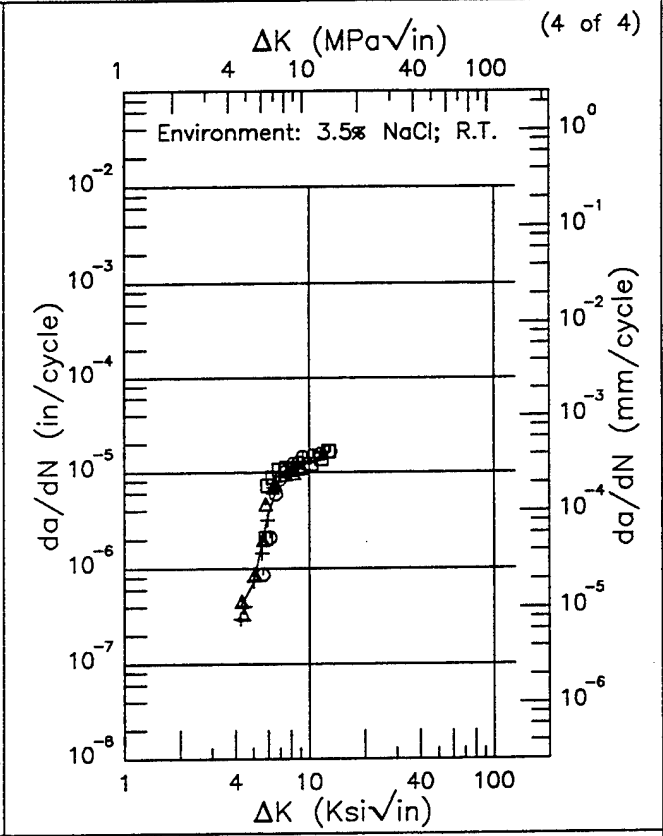
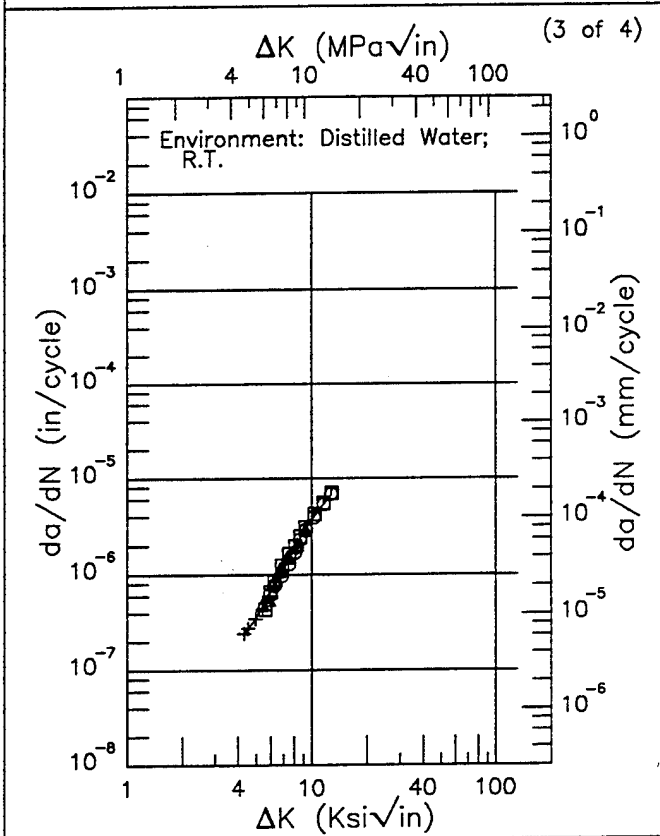
Figure 6.11.3.1.6

Ti-5Al-2.5Sn

E

Condition/Ht: ANNEALED
 Form: 0.08 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: T-L
 Stress Ratio: 0.67
 Frequency: 54.2 - 58.3 Hz

Yield Strength: 125.4 ksi
 Ult. Strength: 133.9 - 135.2 ksi
 Specimen Thk: 0.083 - 0.085 in.
 Specimen Width: 2.753 - 2.757 in.
 Ref: 88911



ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
4.28 (min)	0.237
5.	0.362
6.	0.657
7.	1.13
8.	1.81
9.	2.70
10.	3.74
12.74 (max)	6.70

ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
4.23 (min)	0.395
5.	0.698
6.	3.65
7.	9.00
8.	11.0
9.	11.6
10.	12.5
12.91 (max)	16.4

RMS % Error	Life Prediction Ratio Summary
7.18	

RMS % Error	Life Prediction Ratio Summary
30.21	

Figure 6.11.3.1.6 (Concluded)

Ti-6-2-2-2-2

1 of 1

TABLE 6.12.1.2.1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
Ti-6-2-2-2-2 AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: 3.5% NaCl

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
ST	PLATE	0.1	1			2.08	48.4	626.69	
					0.3	4.87	31.94		
STA	PLATE	0.1	1				75.53		
					0.46	6.14	16.94		

1 of 1

TABLE 6.12.1.2.2

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
TI-6-2-2-2-2 AT ROOM TEMPERATURE

ENVIRONMENT: H.H.A.

ORIENTATION: L-T

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	6.0	10.0	20.0	50.0	100.0
ST	PLATE	0.1	1			2.38	13.58	285.45	
		0.1	20			1.45	11.42	104.74	
STA	PLATE	0.1	1			1.93	17.04	418.61	
		0.1	20			0.17	13.68		

Ti-6-2-2-2-2

1 of 1

TABLE 6.12.2.1

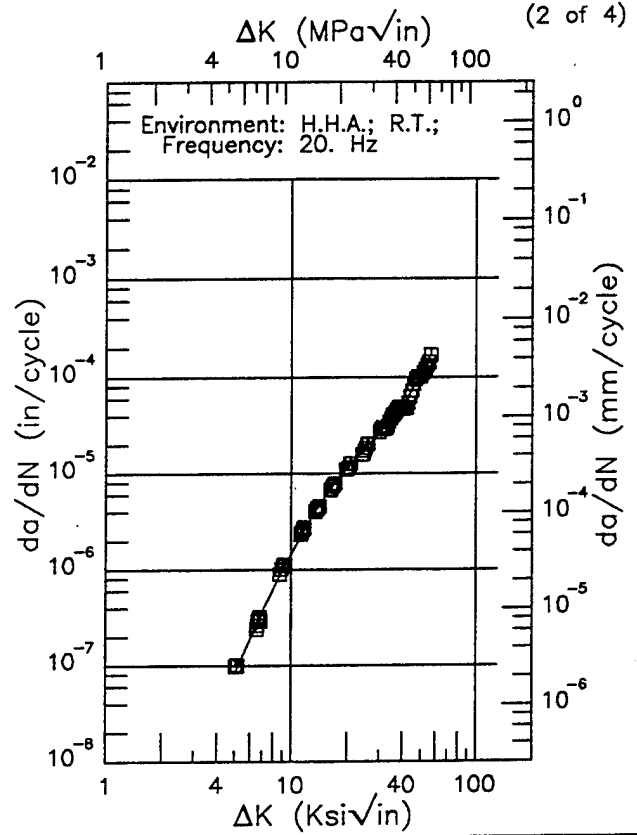
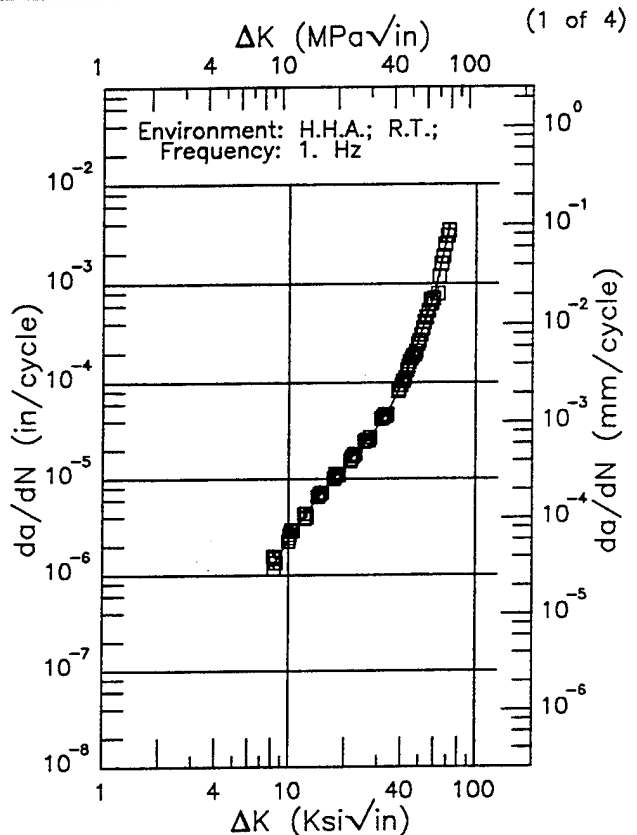
TITANIUM Ti-6Al-2Sn-2Zr-2Mo-2Cr-0.26Si K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic}/TYS)^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (KSI/in.)	K_{Ic} MEAN	STAN DEV		
1740F 1 HR AC	Plate	0.62	R.T.	L-T	148.0	2.000	0.625	CT	1.000	0.45	62.90	61.6	1.6	1974	88186
		0.62			148.0	2.000	0.625	CT	1.000	0.41	59.80			1974	88186
		0.62			148.0	2.000	0.626	CT	1.000	0.44	62.00			1974	88186
STA-1740F 1 HR AC 1000F 8HR AC	Plate	0.62	R.T.	L-T	157.0	2.000	0.625	CT	1.000	0.30	54.30	65.3	1.5	1974	88186
		0.62			157.0	2.000	0.624	CT	1.000	0.33	57.00			1974	88186
		0.62			157.0	2.000	0.625	CT	1.000	0.30	54.70			1974	88186

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EF | Ti-6-2-2-2-2 |

Condition/Ht: ST
 Form: 0.63 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.1

Yield Strength: 157 ksi
 Ult. Strength:
 Specimen Thk: 0.151 - 0.152 in.
 Specimen Width: 3 in.
 Ref: 86844



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.23 (min)	1.23
9.	1.68
10.	2.36
13.	4.93
16.	8.17
20.	13.6
25.	22.5
30.	35.5
35.	57.1
40.	95.5
50.	285.
60.	788.
70.	3037.
71.40 (max)	3918.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
5.05 (min)	0.0855
6.	0.184
7.	0.355
8.	0.614
9.	0.976
10.	1.45
13.	3.58
16.	6.61
20.	11.4
25.	18.1
30.	26.1
35.	36.6
40.	51.4
50.	105.
56.48 (max)	154.

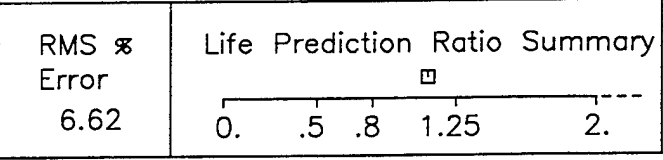
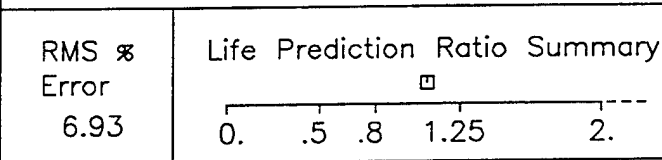
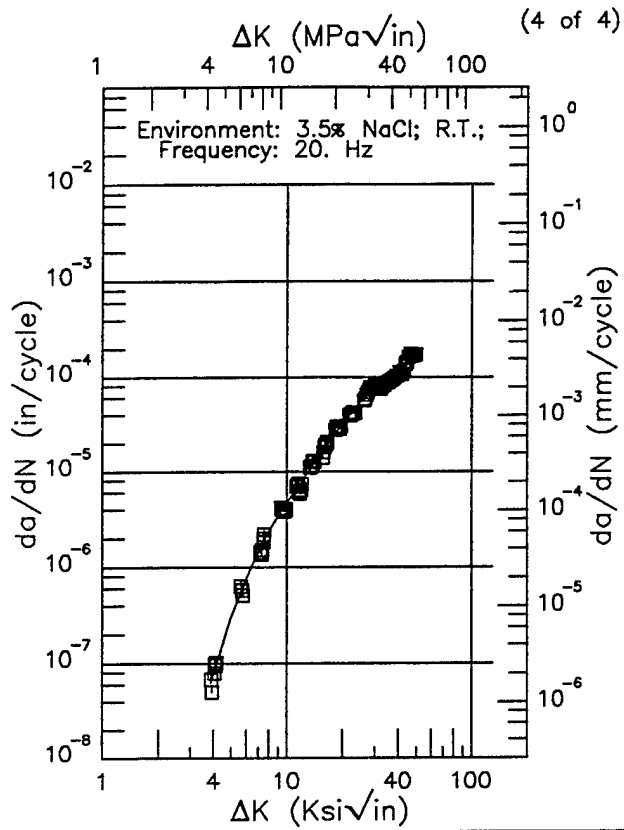
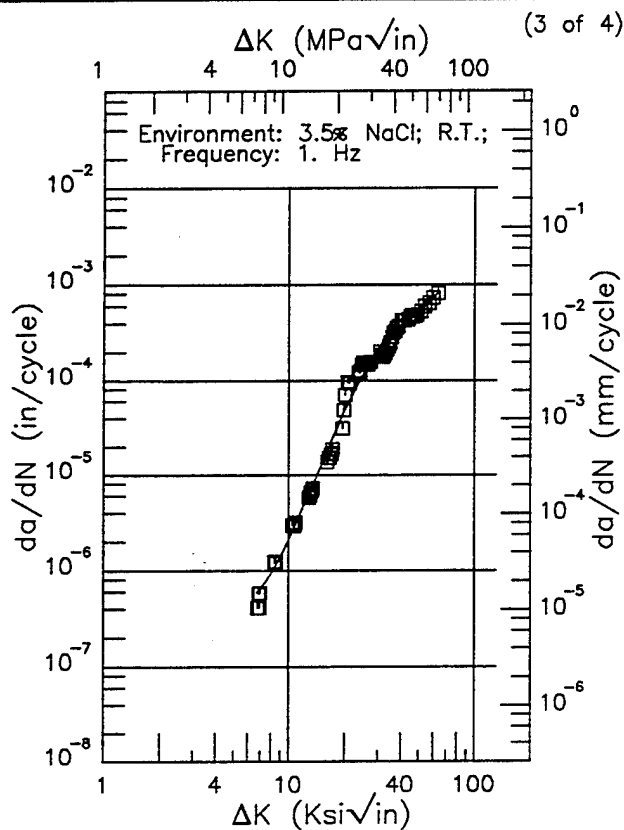


Figure 6.12.3.1.1

Ti-6-2-2-2-2 EF

Condition/Ht: ST
 Form: 0.63 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.1

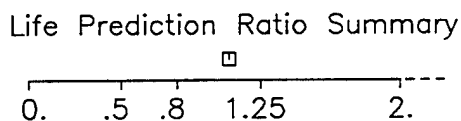
Yield Strength: 157 ksi
 Ult. Strength:
 Specimen Thk: 0.151 - 0.152 in.
 Specimen Width: 3 in.
 Ref: 86844



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
6.77 (min)	0.596
7.	0.642
8.	0.924
9.	1.38
10.	2.08
13.	6.63
16.	17.6
20.	48.4
25.	116.
30.	201.
35.	289.
40.	373.
50.	527.
60.	727.
63.48 (max)	831.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
3.89 (min)	0.0620
4.	0.0755
5.	0.301
6.	0.749
7.	1.43
8.	2.32
9.	3.41
10.	4.67
13.	9.67
16.	17.1
20.	31.9
25.	54.5
30.	73.6
35.	89.6
40.	109.
49.43 (max)	193.

RMS %
 Error
 21.60



RMS %
 Error
 9.15

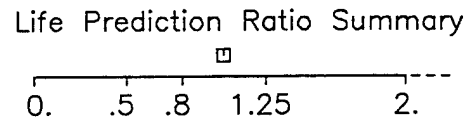
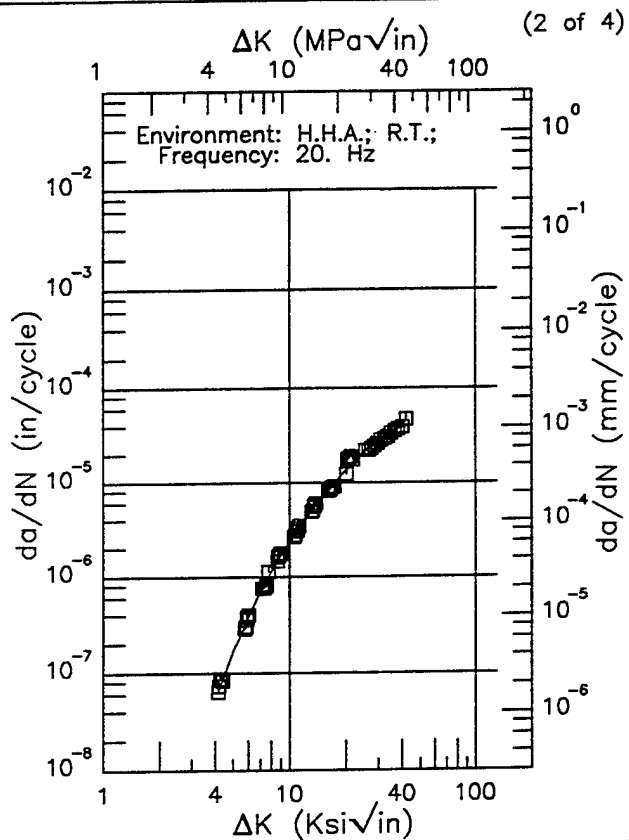
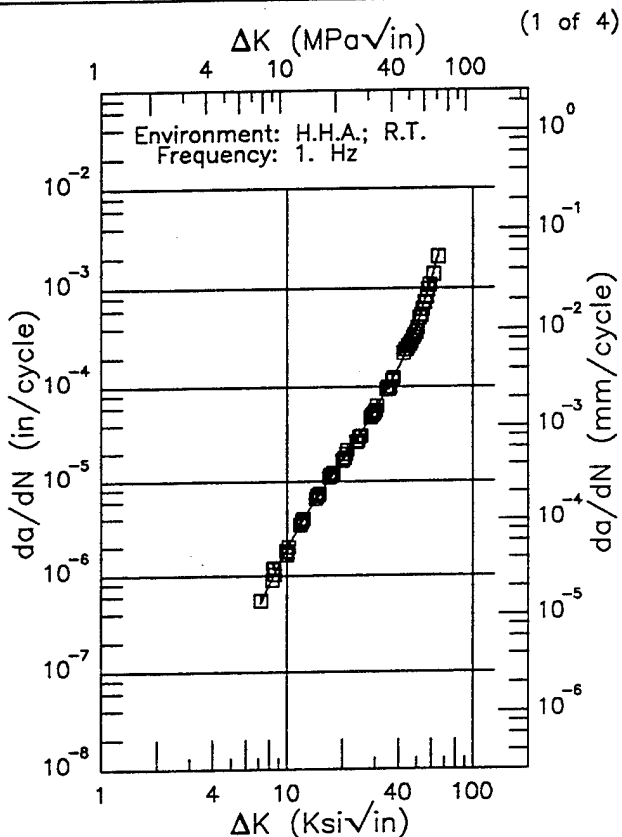


Figure 6.12.3.1.1 (Concluded)

EF | Ti-6-2-2-2-2 |

Condition/Ht: STA
 Form: 0.63 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.1

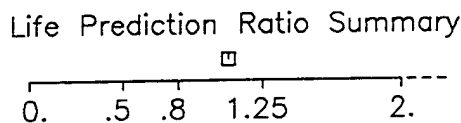
Yield Strength: 157 ksi
 Ult. Strength:
 Specimen Thk: 0.143 - 0.147 in.
 Specimen Width: 3 in.
 Ref: 86844



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
7.18 (min)	0.509
8.	0.813
9.	1.30
10.	1.93
13.	4.69
16.	8.86
20.	17.0
25.	32.8
30.	57.7
35.	96.7
40.	159.
50.	419.
60.	1188.
64.83 (max)	2082.

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
4.13 (min)	0.0622
5.	0.166
6.	0.377
7.	0.700
8.	1.14
9.	1.68
10.	2.34
13.	4.87
16.	8.19
20.	13.7
25.	21.1
30.	27.3
35.	32.4
40.	40.3
41.87 (max)	44.9

RMS $\%$
 Error
 6.11



RMS $\%$
 Error
 8.28

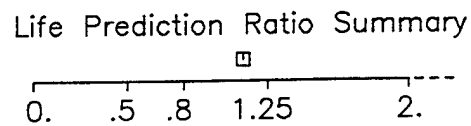


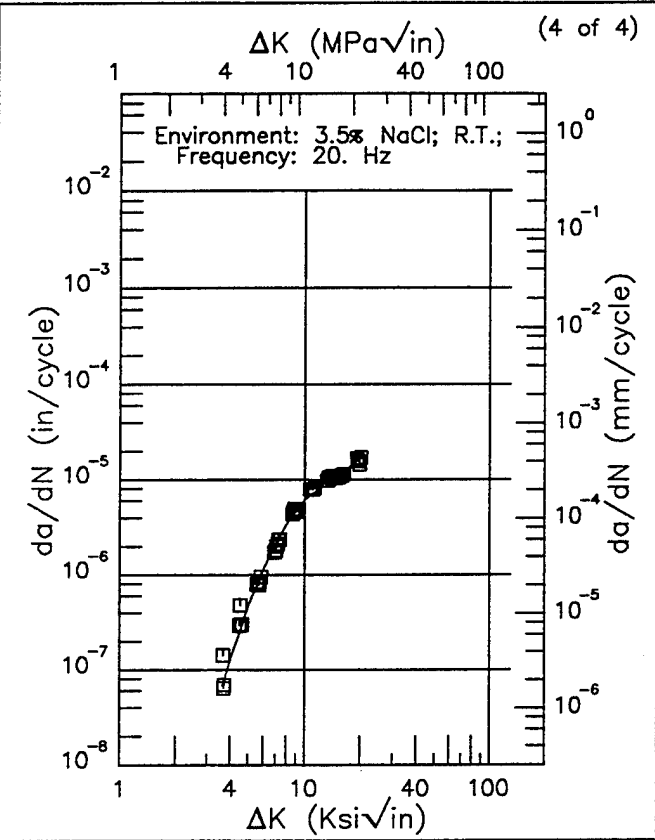
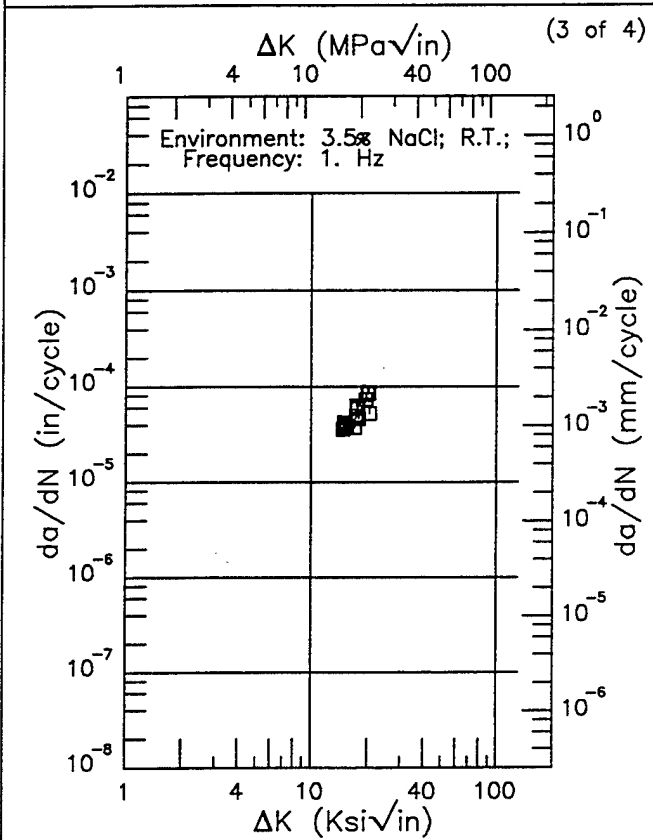
Figure 6.12.3.1.2

Ti-6-2-2-2-2

EF

Condition/Ht: STA
 Form: 0.63 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.1

Yield Strength: 157 ksi
 Ult. Strength:
 Specimen Thk: 0.143 - 0.147 in.
 Specimen Width: 3 in.
 Ref: 86844



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
14.74 (min)	38.5
16.	37.7
20.	75.5
20.37 (max)	69.3

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
3.64 (min)	0.0686
4.	0.126
5.	0.456
6.	1.11
7.	2.10
8.	3.35
9.	4.74
10.	6.14
13.	9.43
16.	11.8
20.	16.9
20.05 (max)	17.0

RMS % Error	Life Prediction Ratio Summary
13.97	

RMS % Error	Life Prediction Ratio Summary
20.69	

Figure 6.12.3.1.2 (Concluded)

Ti-6-2-4-2

1 of 1

TABLE 6.13.1.2

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6-2-4-2 AT ROOM TEMPERATURE

ORIENTATION: C-R ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
1700F 1HR AC 1100F 8HR AC	FORGING	0.1	0.16							
		0.5	0.16		10					
					2.3					

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EF | Ti-6-2-4-2 |

Condition/Ht: 1790F 1HR AC 1100F 8HR AC
 Form: 2 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Stress Ratio: 0.1

Yield Strength: 139.4 - 140.9 ksi
 Ult. Strength: 151.6 - 152.3 ksi
 Specimen Thk: 0.073 - 0.083 in.
 Specimen Width: 1.75 in.
 Ref: PW002

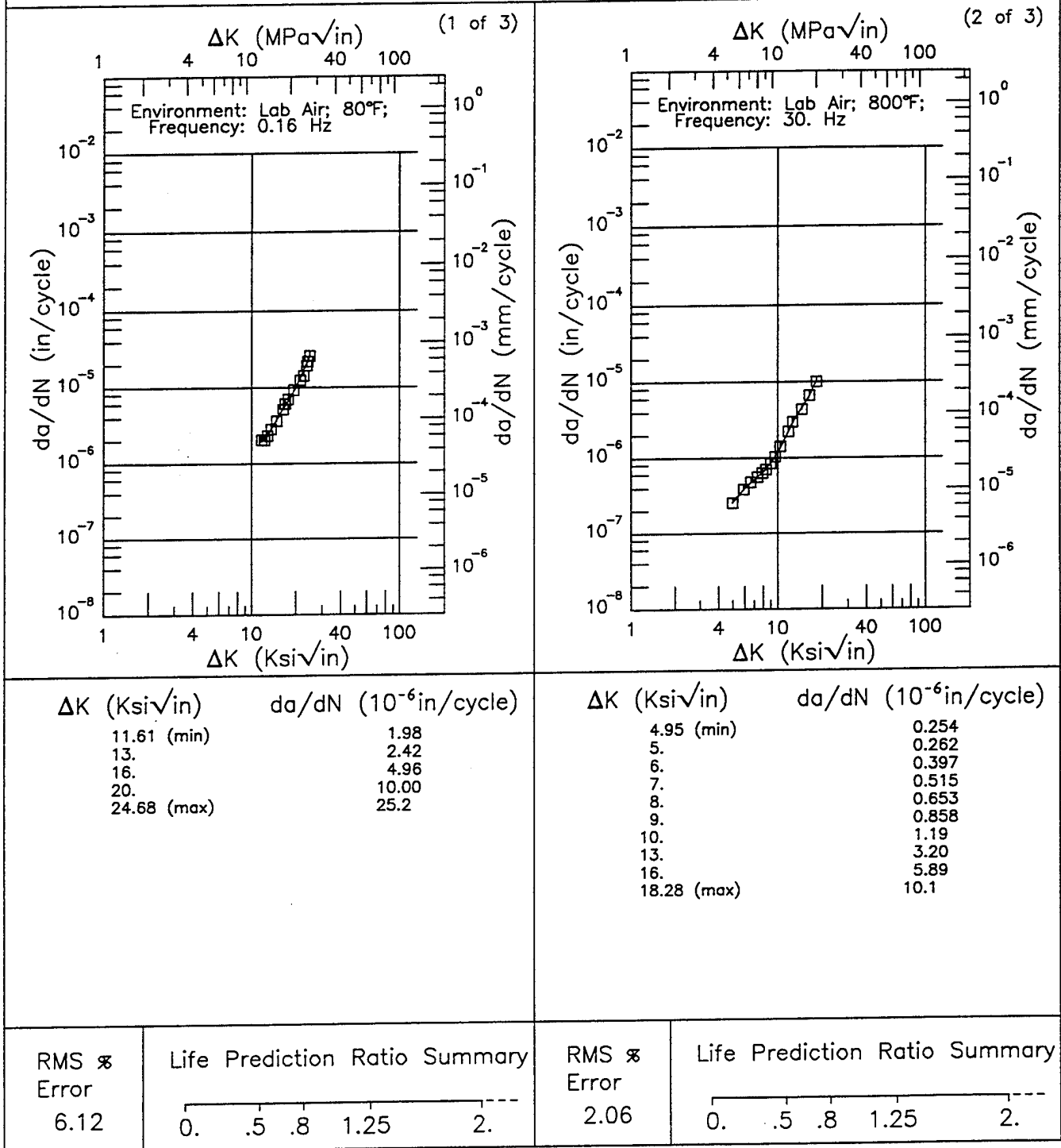


Figure 6.13.3.1.1

Ti-6-2-4-2 EF

Condition/Ht: 1790F 1HR AC 1100F 8HR AC
 Form: 2 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Stress Ratio: 0.1

Yield Strength: 139.4 - 140.9 ksi
 Ult. Strength: 151.6 - 152.3 ksi
 Specimen Thk: 0.073 - 0.083 in.
 Specimen Width: 1.75 in.
 Ref: PW002

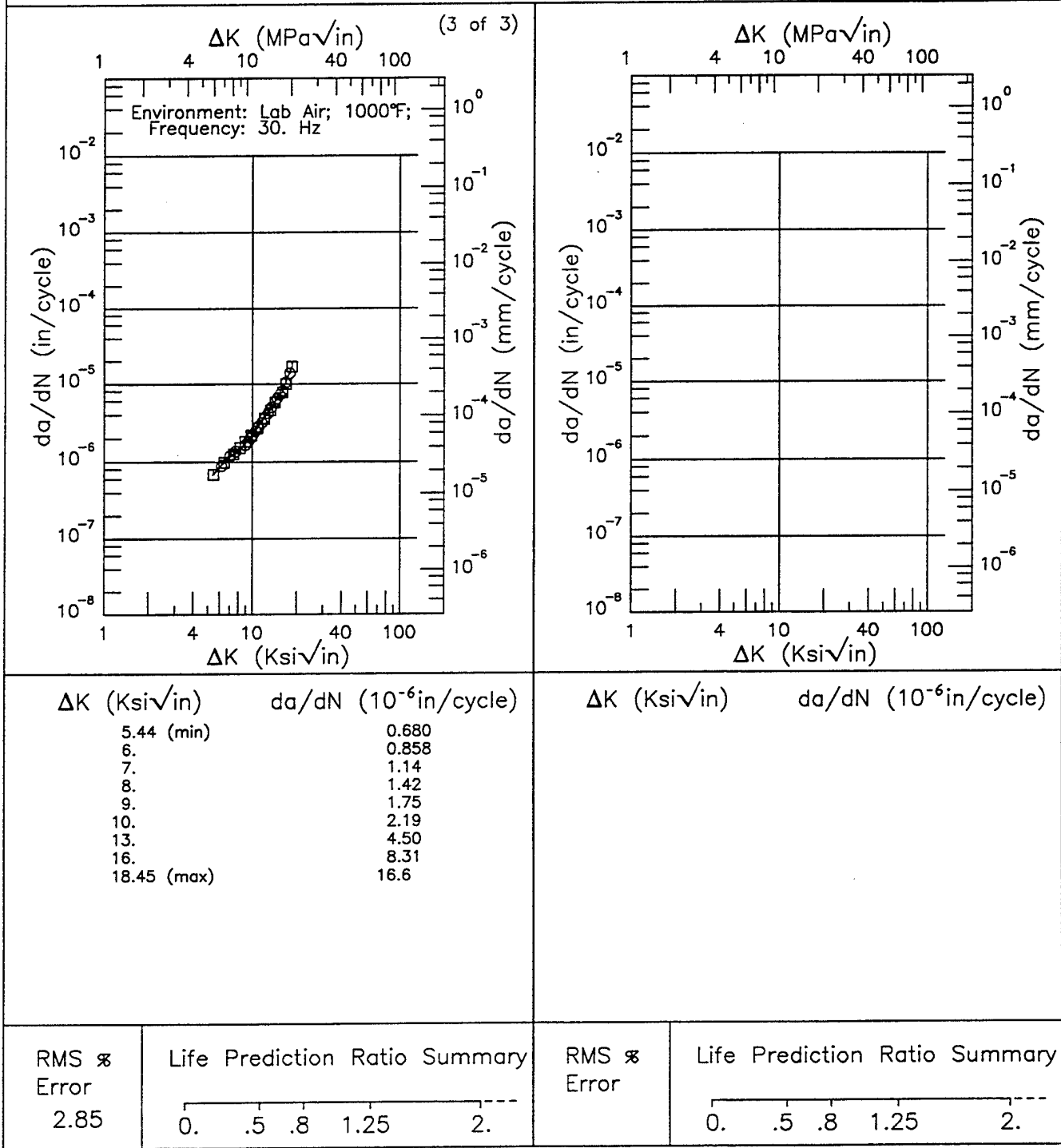
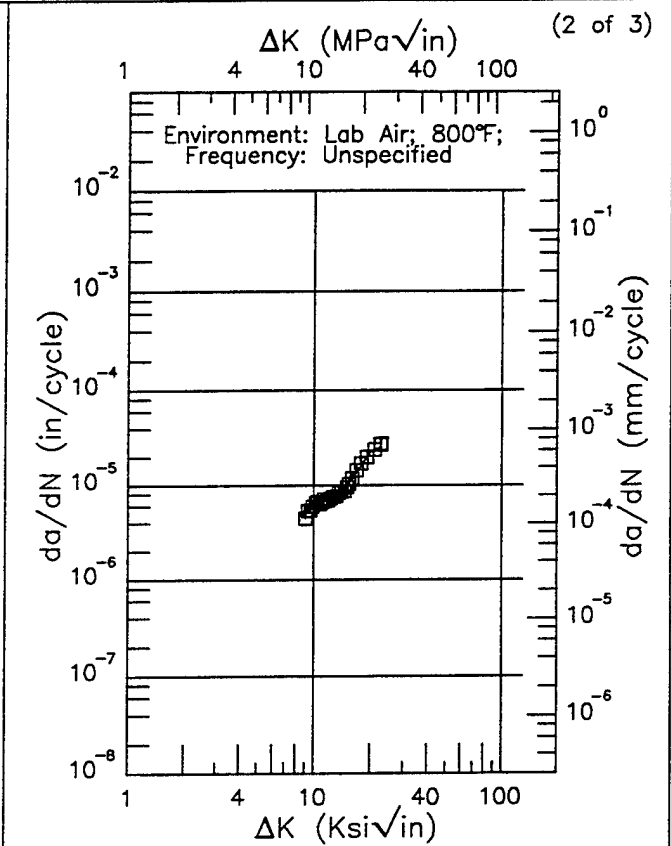
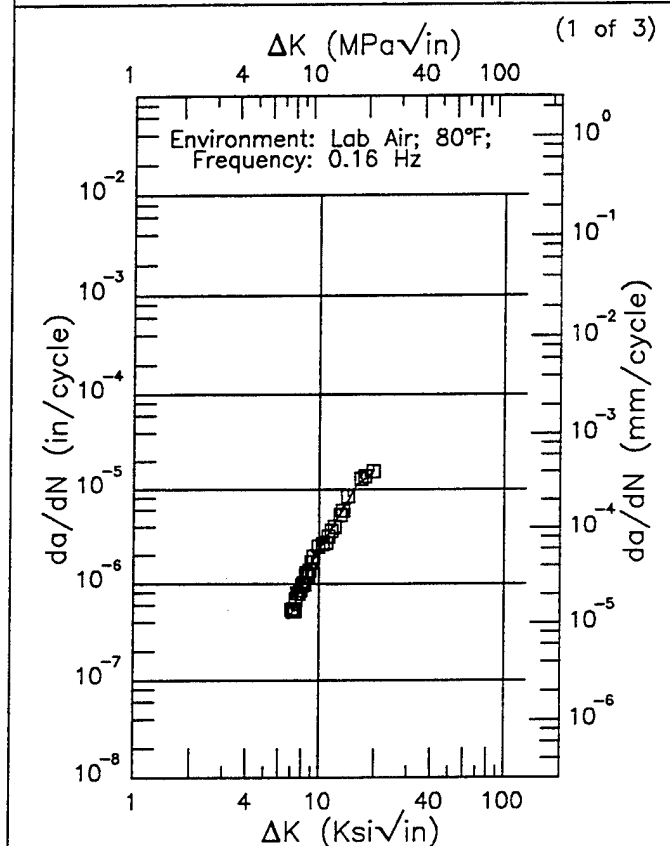


Figure 6.13.3.1.1 (Concluded)

EF | Ti-6-2-4-2 |

Condition/Ht: 1790F 1HR AC 1100F 8HR AC
 Form: 2 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Stress Ratio: 0.5

Yield Strength: 139.4 - 140.9 ksi
 Ult. Strength: 151.6 - 152.3 ksi
 Specimen Thk: 0.074 - 0.082 in.
 Specimen Width: 1.75 in.
 Ref: PW002



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
7.05 (min)	0.466
8.	0.951
9.	1.58
10.	2.30
13.	5.37
16.	11.4
19.52 (max)	14.8

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
8.97 (min)	4.87
9.	4.90
10.	5.88
13.	7.76
16.	11.4
20.	22.6
22.81 (max)	25.6

RMS $\%$ Error	Life Prediction Ratio Summary
7.60	

RMS $\%$ Error	Life Prediction Ratio Summary
4.99	

Figure 6.13.3.1.2

Ti-6-2-4-2 EF

Condition/Ht: 1790F 1HR AC 1100F 8HR AC
 Form: 2 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Stress Ratio: 0.5

Yield Strength: 139.4 - 140.9 ksi
 Ult. Strength: 151.6 - 152.3 ksi
 Specimen Thk: 0.074 - 0.082 in.
 Specimen Width: 1.75 in.
 Ref: PW002

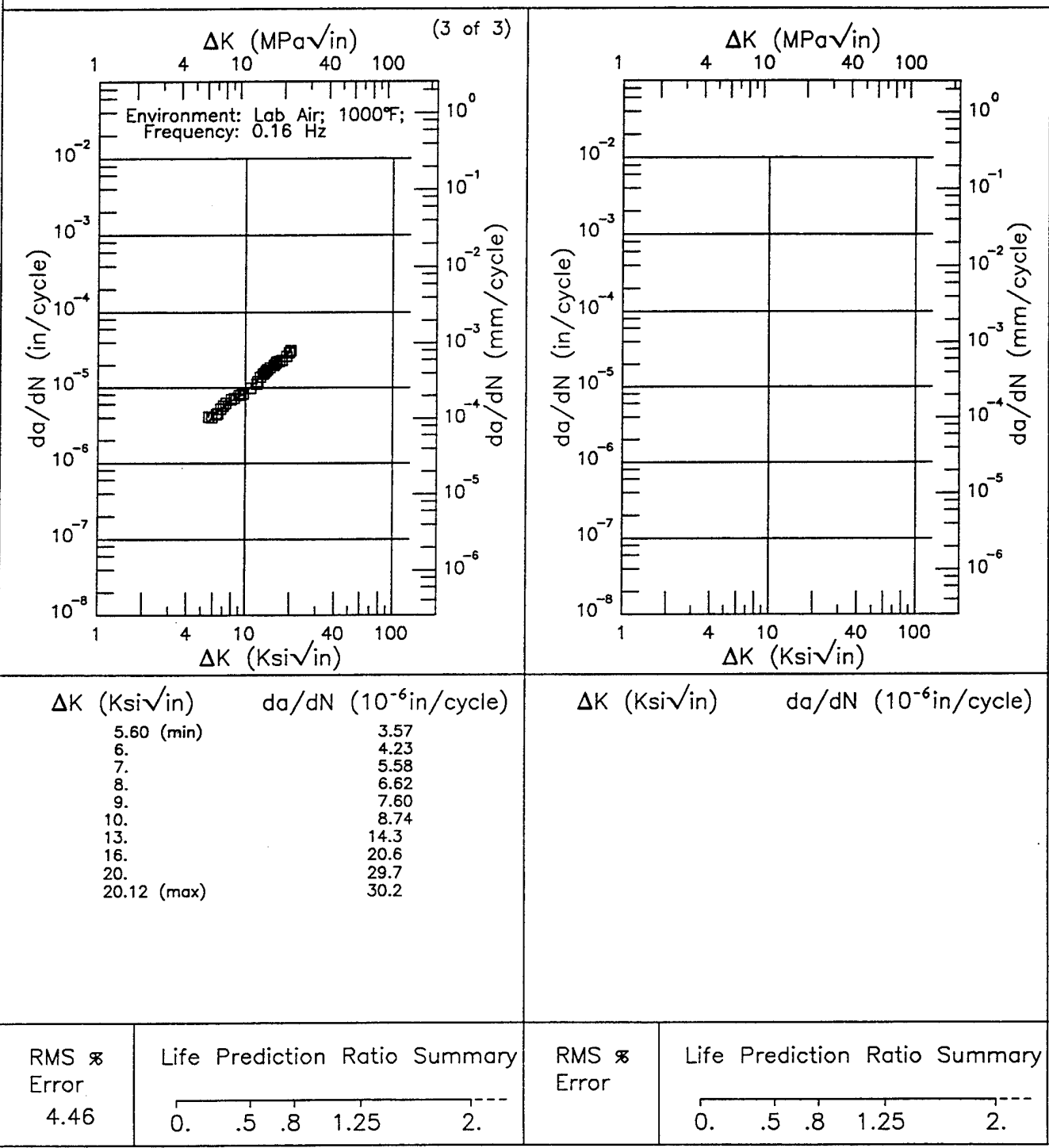


Figure 6.13.3.1.2 (Concluded)

EF | Ti-6-2-4-2 |

Condition/Ht: 1790F 1HR AC 1100F 8HR AC
 Form: 2 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Stress Ratio: 0.7

Yield Strength: 135.5 - 140.9 ksi
 Ult. Strength: 148.5 - 152.3 ksi
 Specimen Thk: 0.08 - 0.081 in.
 Specimen Width: 1.75 in.
 Ref: PW002

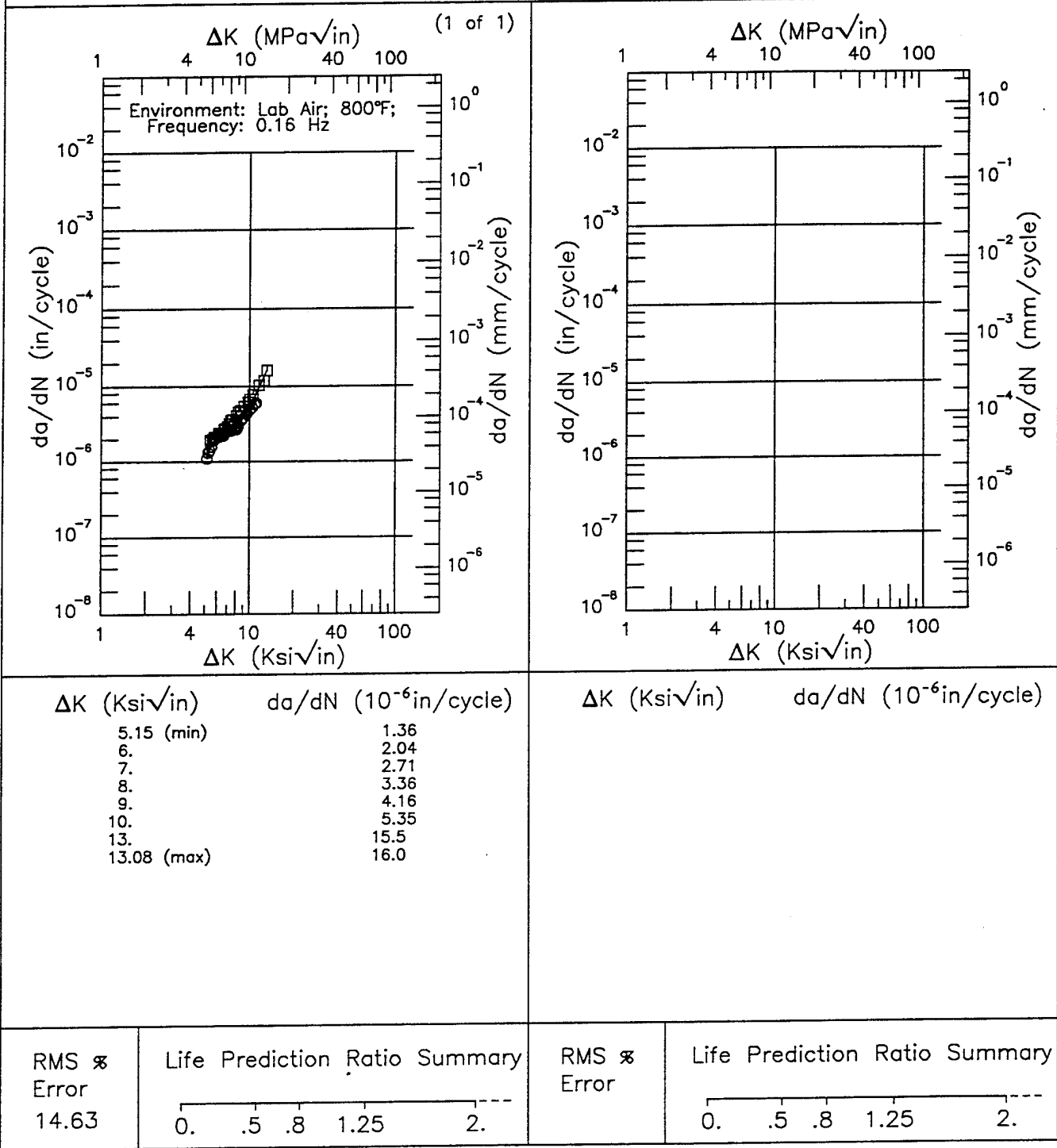


Figure 6.13.3.1.3

TABLE 6.14.1.2.1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6-2-4-2 ELI AT ROOM TEMPERATURE

ENVIRONMENT: Lab Air

ORIENTATION: L-T

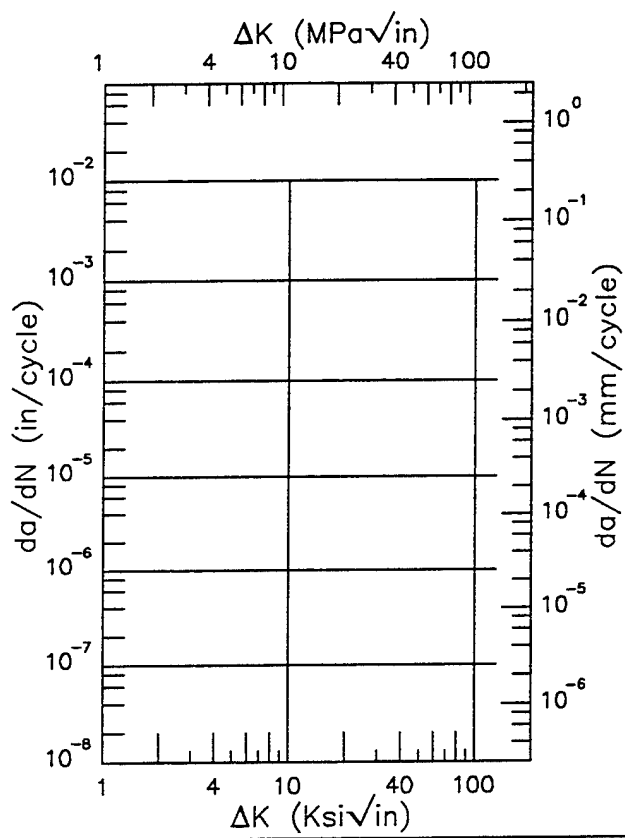
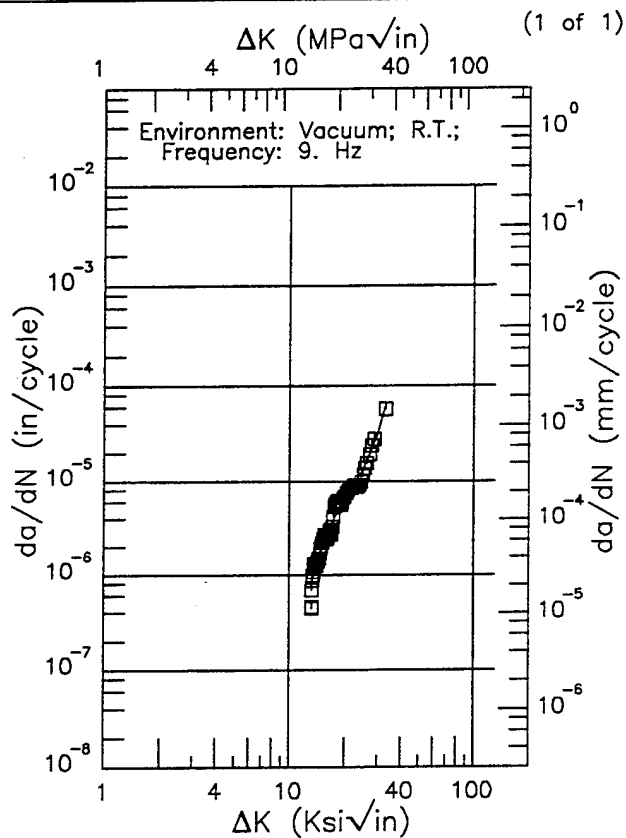
CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.6	6.0	10.0	20.0	50.0	100.0	
ANNEAL 1450F 1HR AC	PLATE	-0.3	9							
		0.1	9				21.3			
		0.5	9				13.49			

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EF | Ti-6-2-4-2 ELI |

Condition/Ht: ANNEAL 1450F 1HR AC
 Form: 0.75 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.1

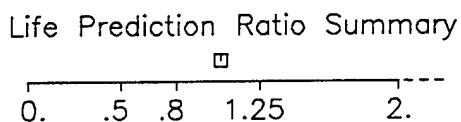
Yield Strength: 140 ksi
 Ult. Strength: 149 ksi
 Specimen Thk: 0.187 in.
 Specimen Width: 1.5 in.
 Ref: MA016



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.18 (min)	0.825
16.	2.89
20.	6.63
25.	13.3
30.	31.5
32.99 (max)	60.9

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS %
 Error
 15.85



RMS %
 Error

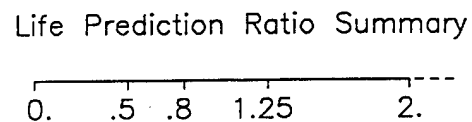


Figure 6.14.3.1.1

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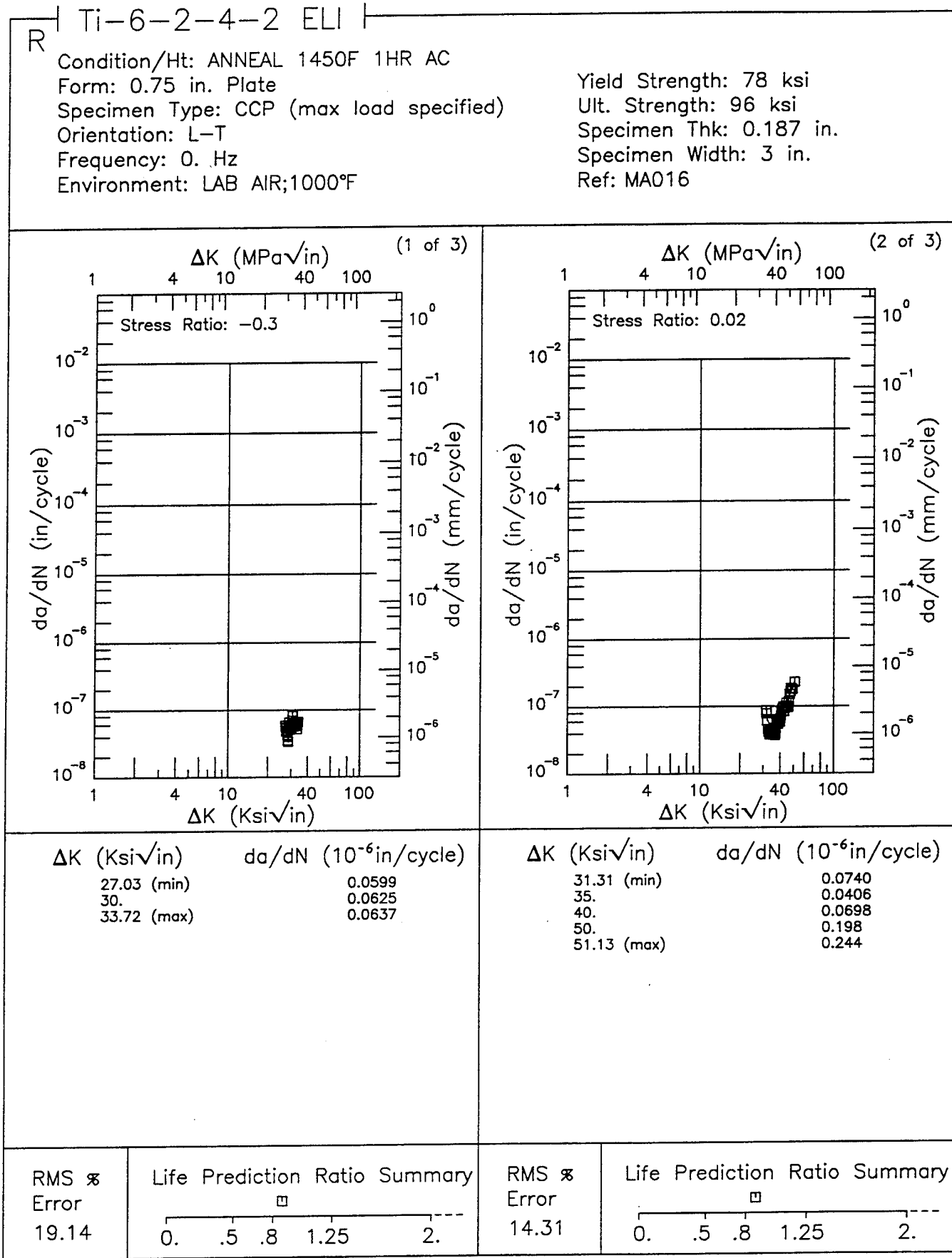


Figure 6.14.3.1.2

Ti-6-2-4-2 ELI R

Condition/Ht: ANNEAL 1450F 1HR AC
 Form: 0.75 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 0. Hz
 Environment: LAB AIR;1000°F

Yield Strength: 78 ksi
 Ult. Strength: 96 ksi
 Specimen Thk: 0.187 in.
 Specimen Width: 3 in.
 Ref: MA016

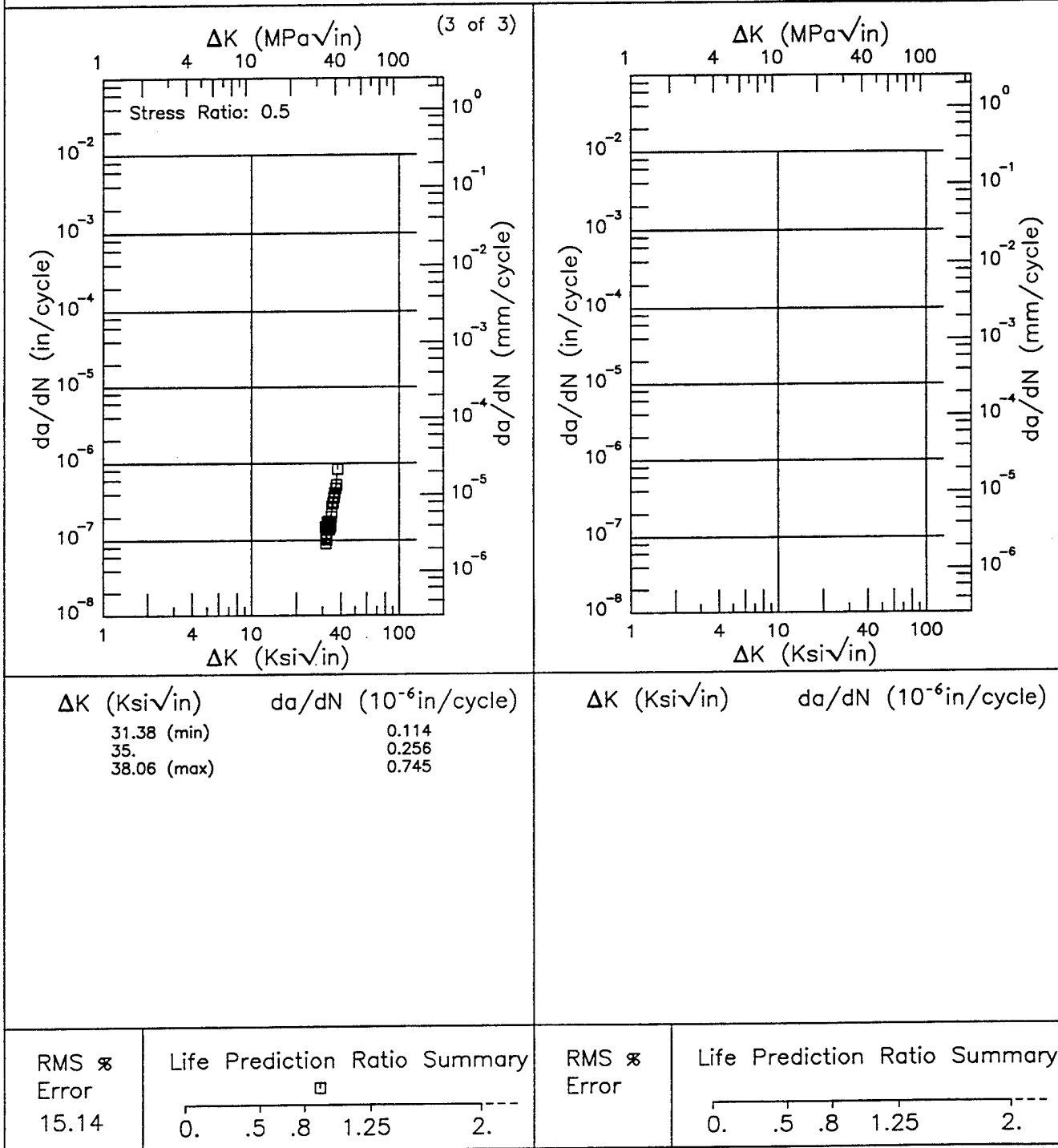


Figure 6.14.3.1.2 (Concluded)

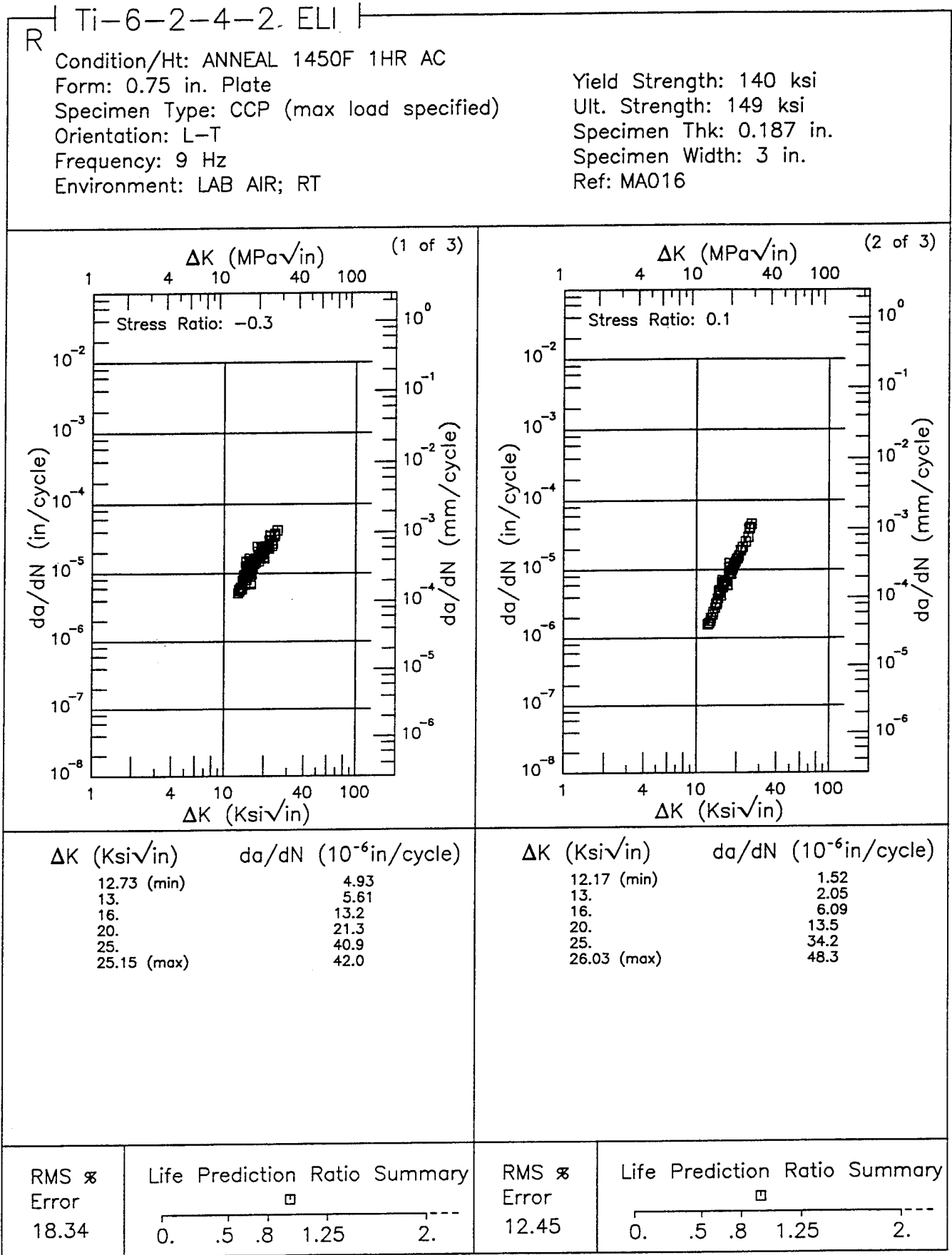


Figure 6.14.3.1.3

Ti-6-2-4-2 ELI R

Condition/Ht: ANNEAL 1450F 1HR AC
 Form: 0.75 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 9 Hz
 Environment: LAB AIR; RT

Yield Strength: 140 ksi
 Ult. Strength: 149 ksi
 Specimen Thk: 0.187 in.
 Specimen Width: 3 in.
 Ref: MA016

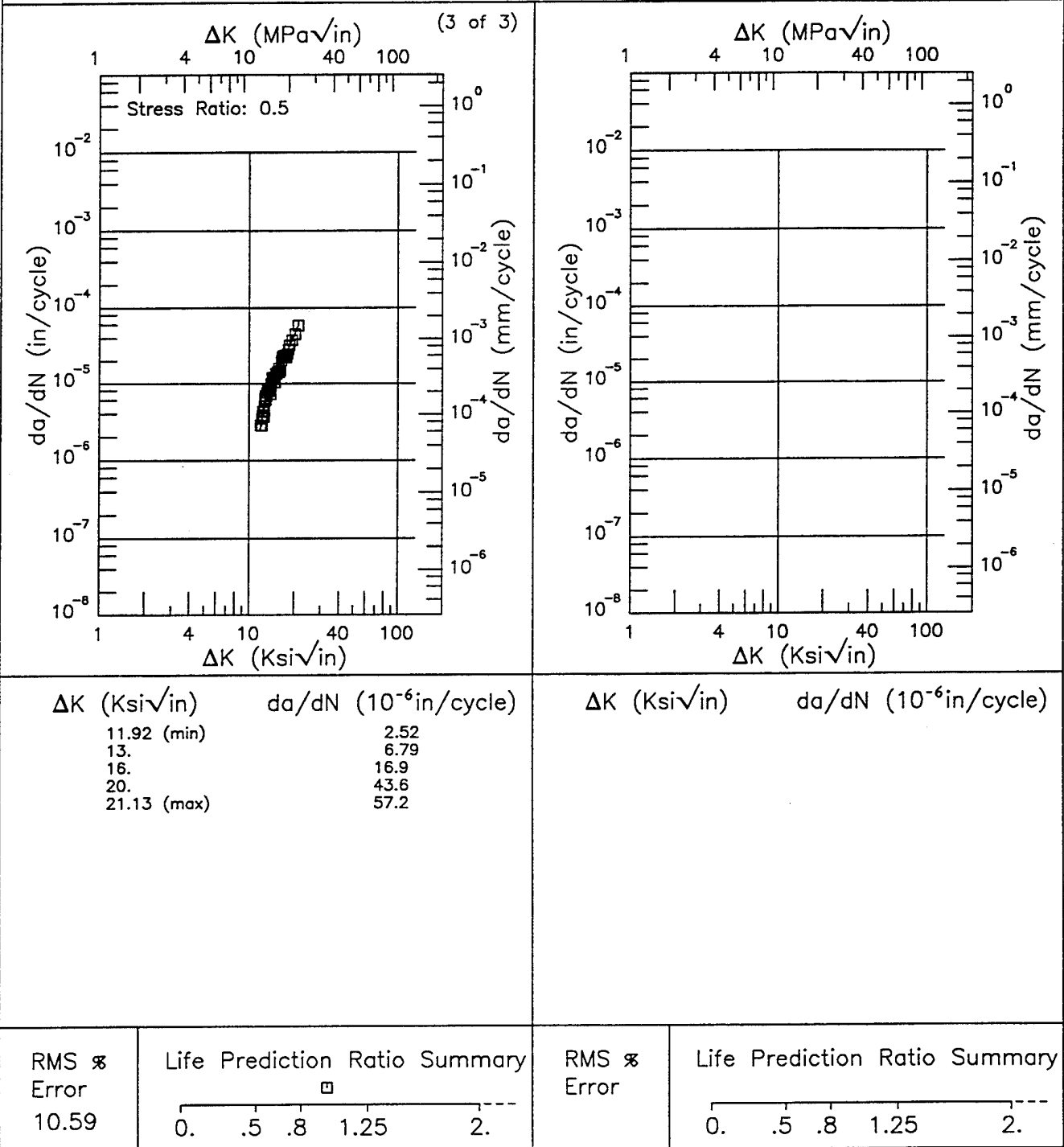
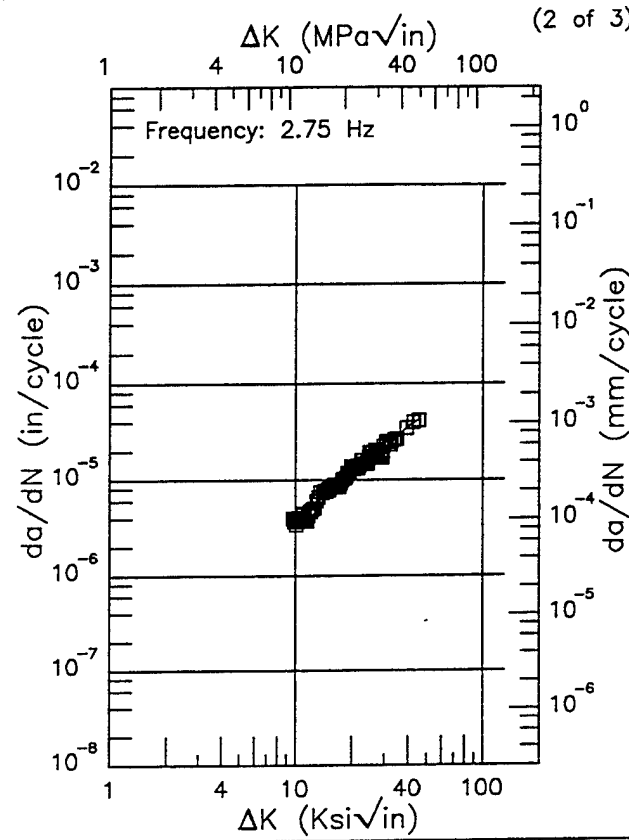
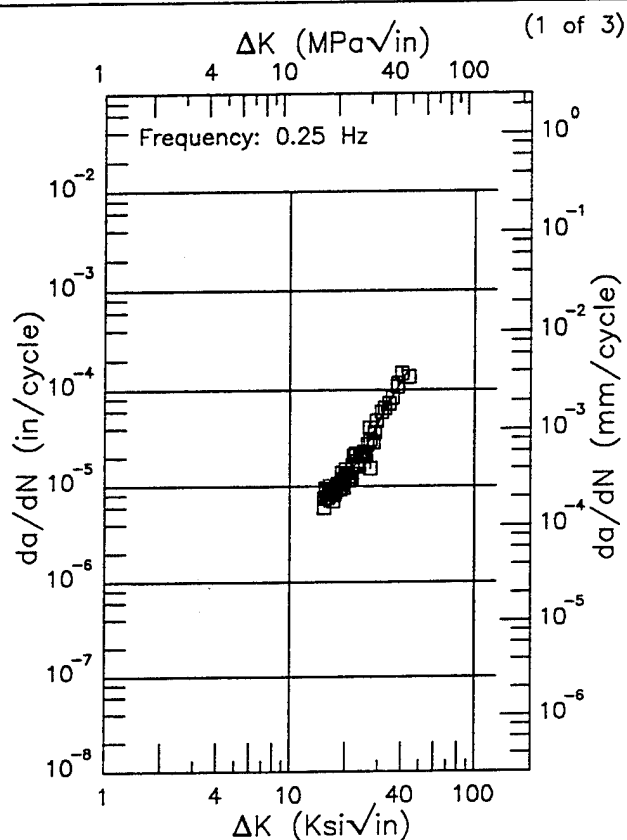


Figure 6.14.3.1.3 (Concluded)

F | Ti-6-2-4-2 ELI |

Condition/Ht: ANNEAL 1450F 1HR AC
 Form: 0.75 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.02
 Environment: LAB AIR;1000°F

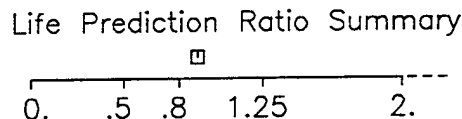
Yield Strength: 78 ksi
 Ult. Strength: 96 ksi
 Specimen Thk: 0.187 in.
 Specimen Width: 3 - 3.002 in.
 Ref: MA016



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
15.33 (min)	8.44
16.	8.46
20.	11.6
25.	22.8
30.	44.9
35.	79.5
40.	123.
43.90 (max)	158.

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
9.66 (min)	3.35
10.	3.56
13.	5.78
16.	8.54
20.	12.4
25.	16.9
30.	21.7
35.	27.7
40.	35.8
45.26 (max)	41.5

RMS $\%$
 Error
 15.34



RMS $\%$
 Error
 7.50

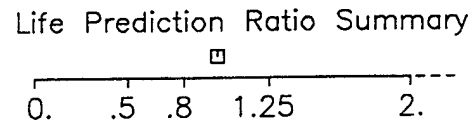


Figure 6.14.3.1.4

Ti-6-2-4-2 ELI F

Condition/Ht: ANNEAL 1450F 1HR AC
 Form: 0.75 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.02
 Environment: LAB AIR;1000°F

Yield Strength: 78 ksi
 Ult. Strength: 96 ksi
 Specimen Thk: 0.187 in.
 Specimen Width: 3 - 3.002 in.
 Ref: MA016

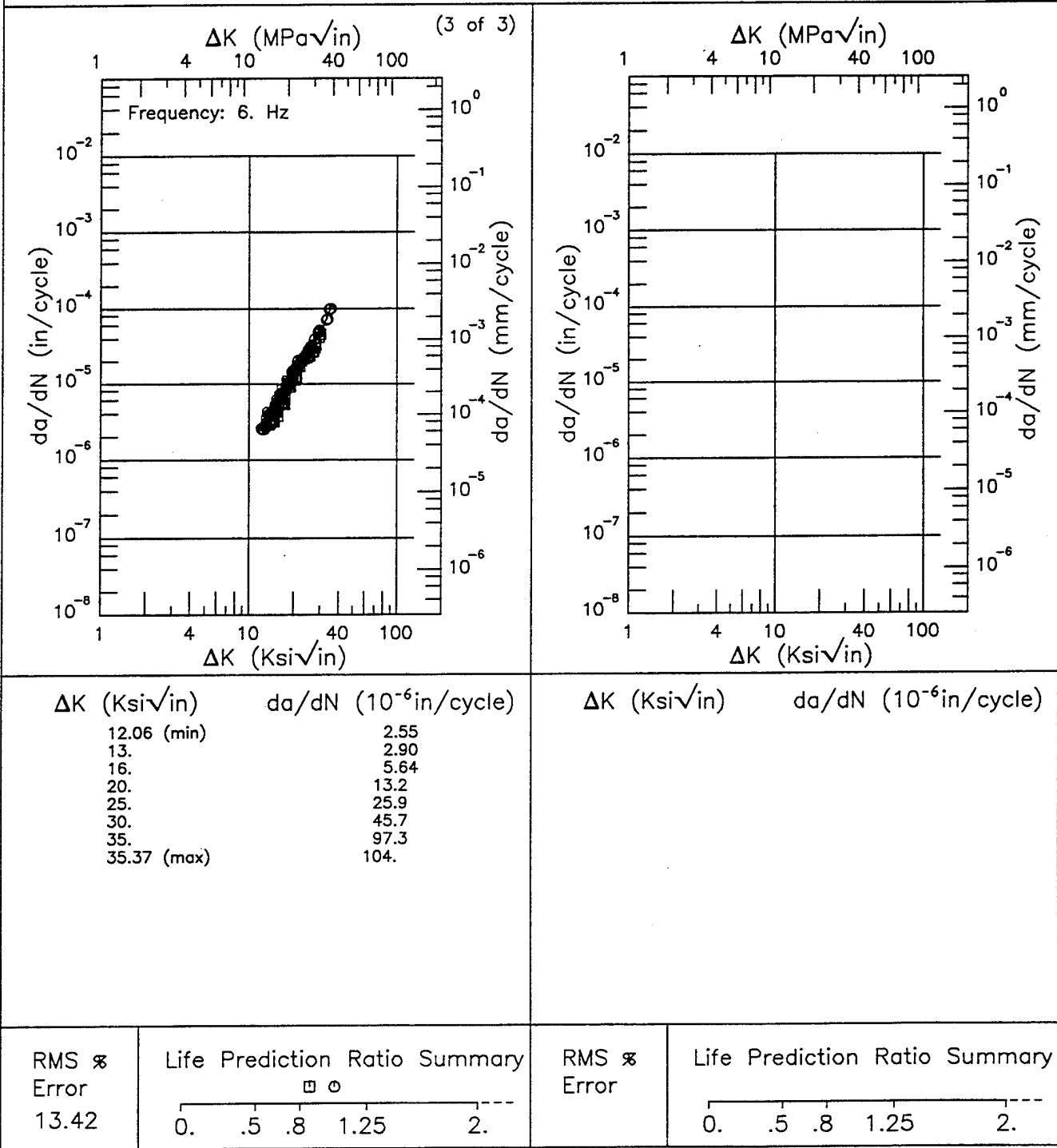
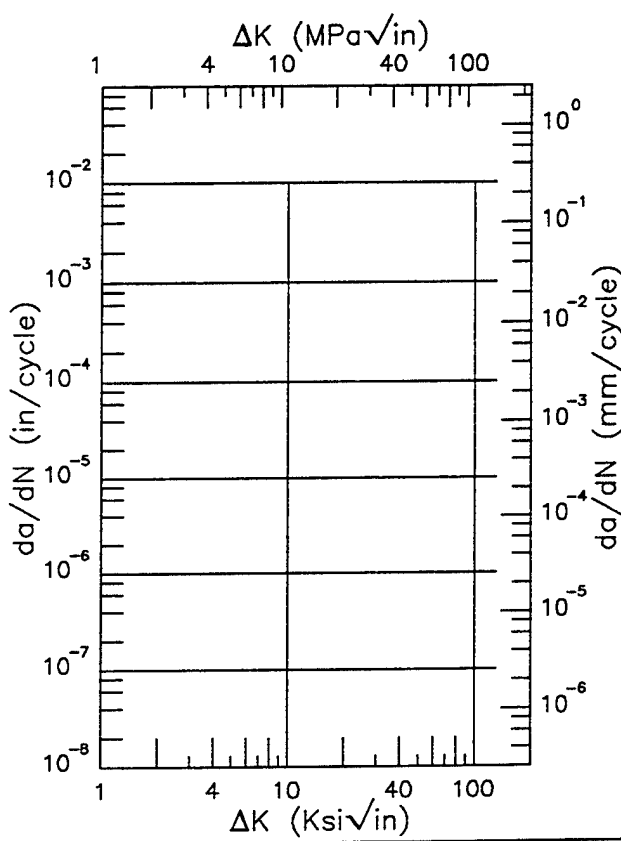
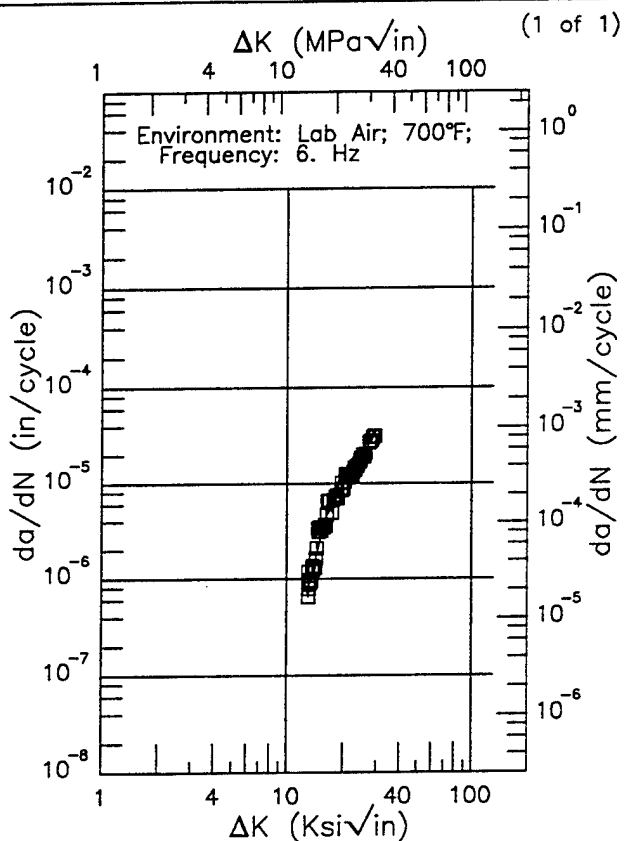


Figure 6.14.3.1.4 (Concluded)

EF | Ti-6-2-4-2 ELI |

Condition/Ht: ANNEAL 1450F 1HR AC
 Form: 0.75 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.02

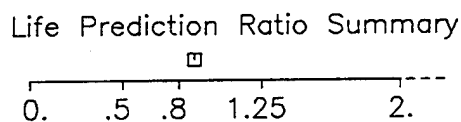
Yield Strength: 95 ksi
 Ult. Strength: 109 ksi
 Specimen Thk: 0.187 in.
 Specimen Width: 3.001 in.
 Ref: MA016



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
12.93 (min)	0.712
13.	0.758
16.	4.20
20.	9.90
25.	17.8
29.36 (max)	35.5

ΔK (Ksi $\sqrt{\text{in}}$) da/dN (10^{-6} in/cycle)

RMS %
 Error
 17.14



RMS %
 Error

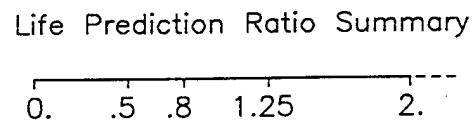


Figure 6.14.3.1.5

Ti-6-2-4-2 ELI EF

Condition/Ht: ANNEAL 1450F 1HR AC
 Form: 0.75 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.5

Yield Strength: 78 ksi
 Ult. Strength: 96 ksi
 Specimen Thk: 0.187 in.
 Specimen Width: 3 in.
 Ref: MA016

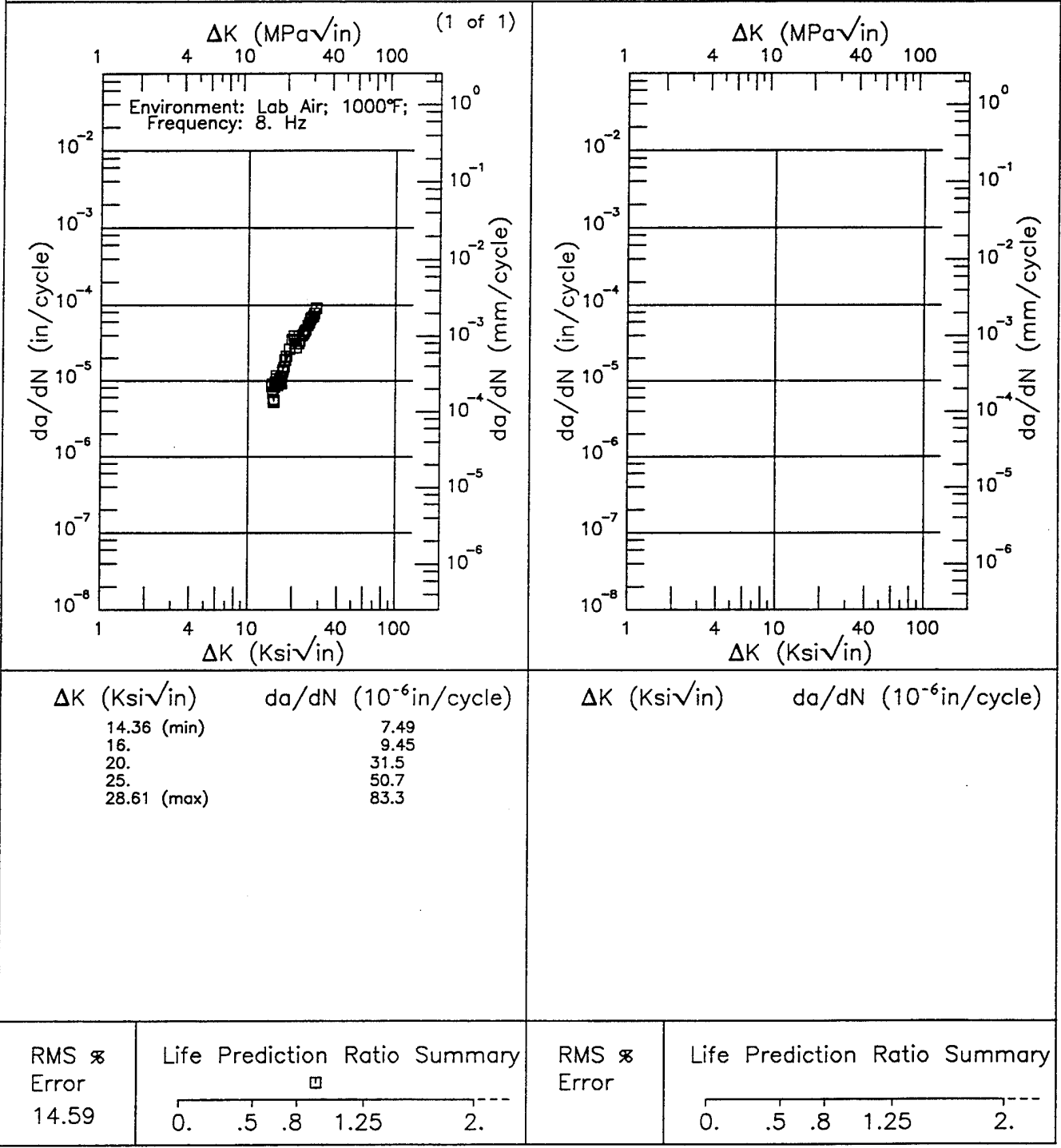


Figure 6.14.3.1.6

TI-6-2-4-6

1 of 1

TABLE 6.15.1.2.1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 TI-6-2-4-6 AT ROOM TEMPERATURE

ORIENTATION: L-T		ENVIRONMENT: Lab Air				
CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)		
				ΔK Level (Kst \sqrt{in})		
UNSPECIFIED	EXTRUSION	0.1	20	2.5	5.0	10.0
				20.0	50.0	100.0
				0.82	9.45	

TABLE 6.15.1.2.2

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
Ti-6-2-4-6 AT ROOM TEMPERATURE**

ORIENTATION: C-R ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
1690F 2HRS AC 1550F 2HRS OQ 1100F 8HRS AC	FORGING	0.1	30		0.1	1.01			
		0.7	0.16	0.06	0.55				

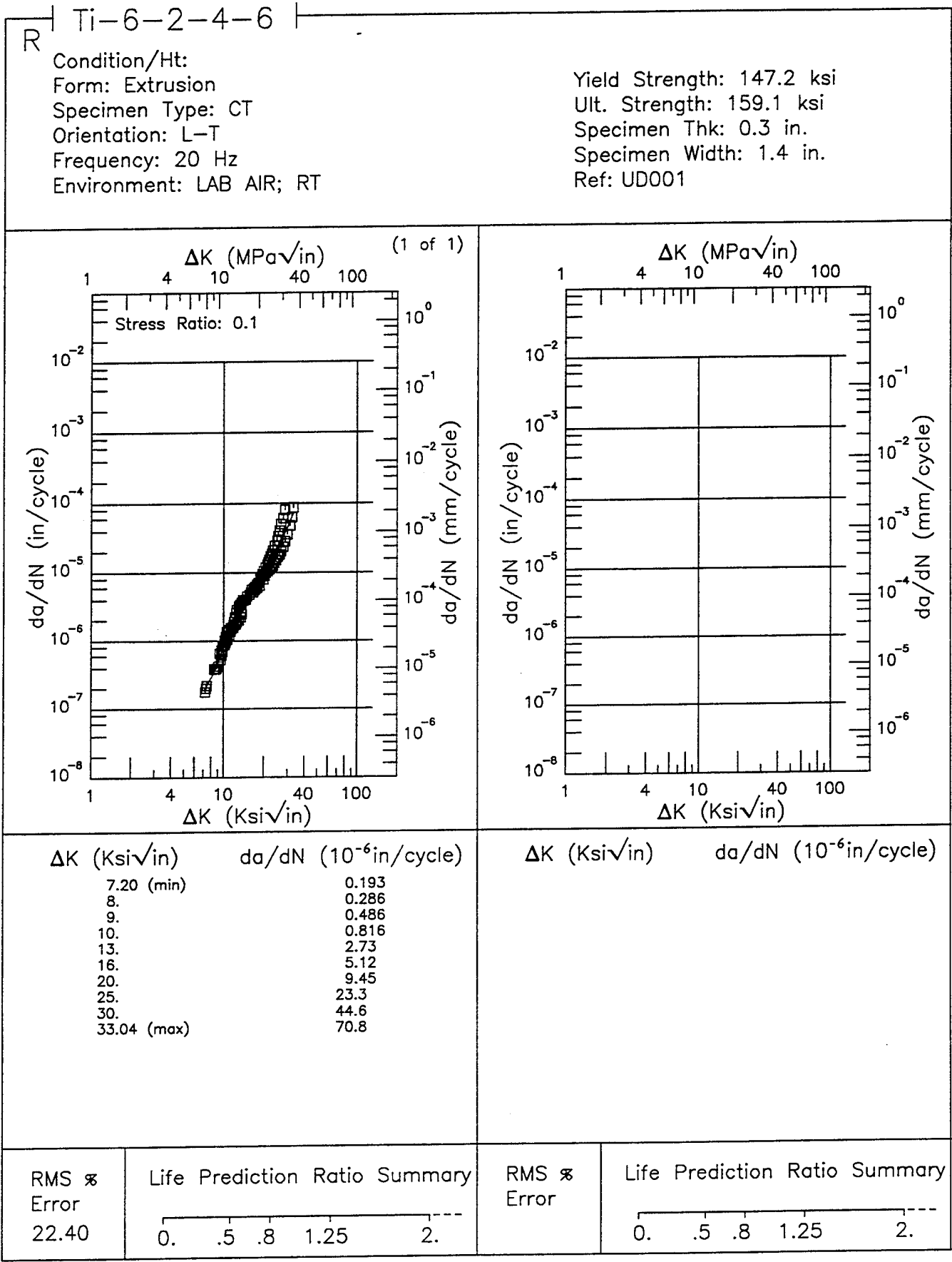


Figure 6.15.3.1.1

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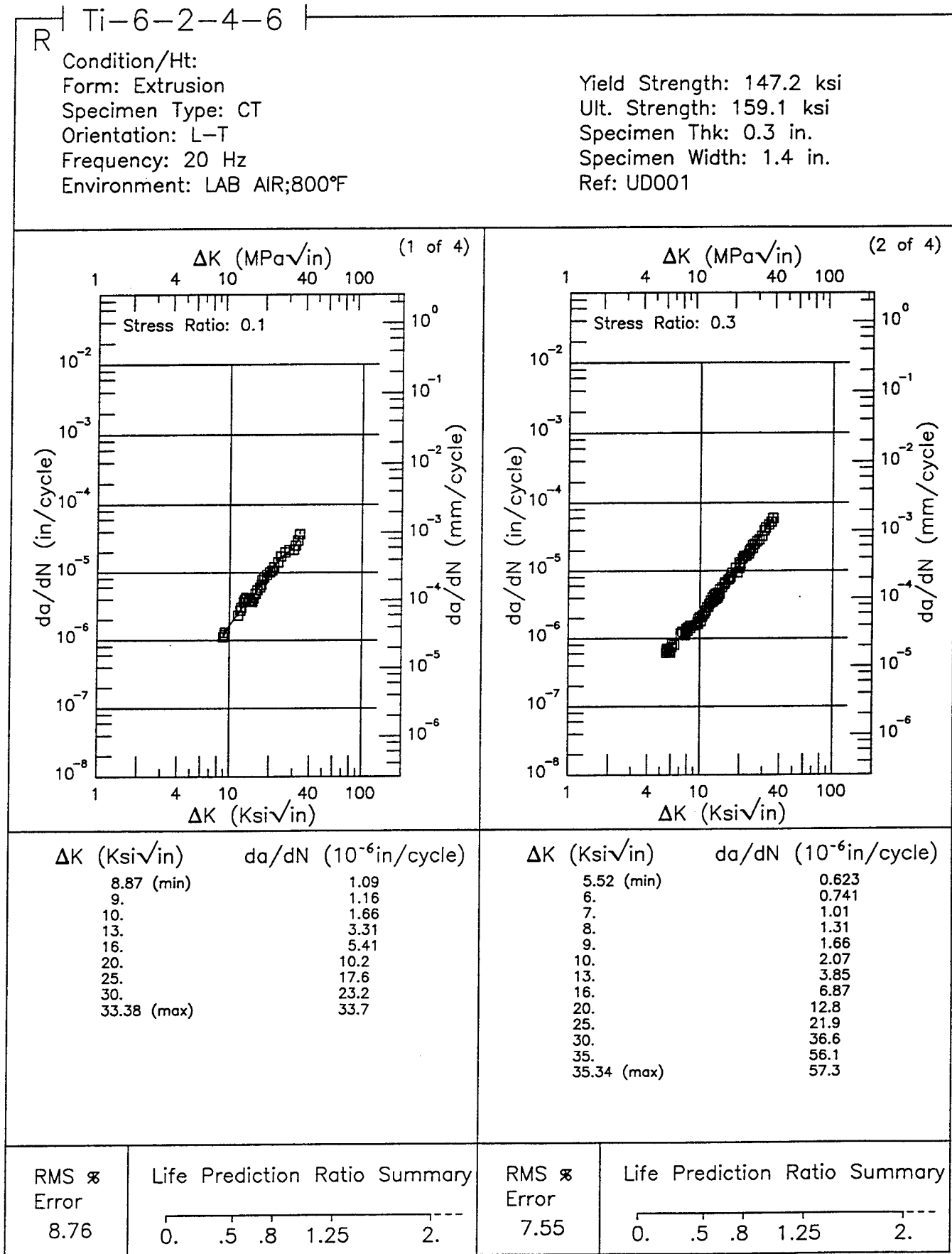
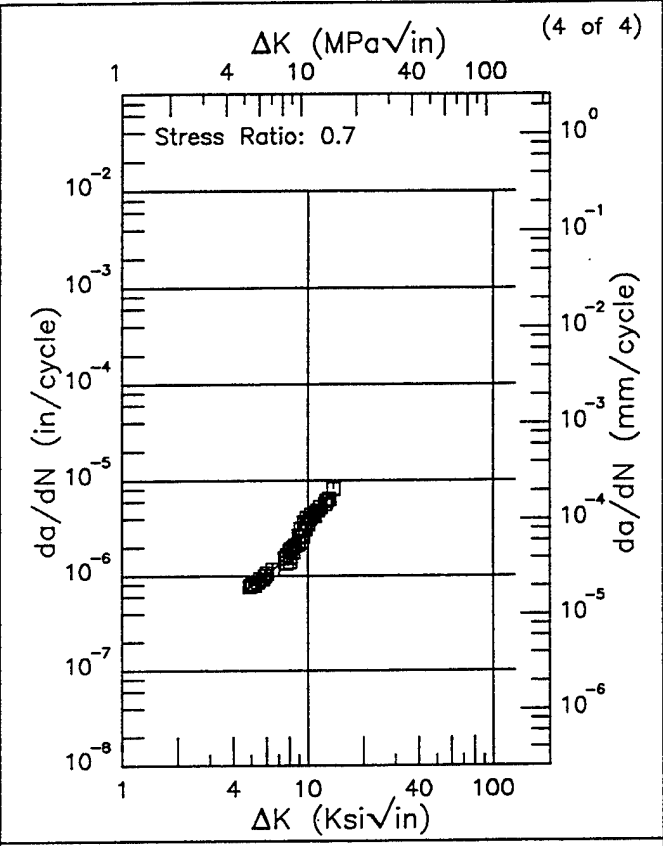
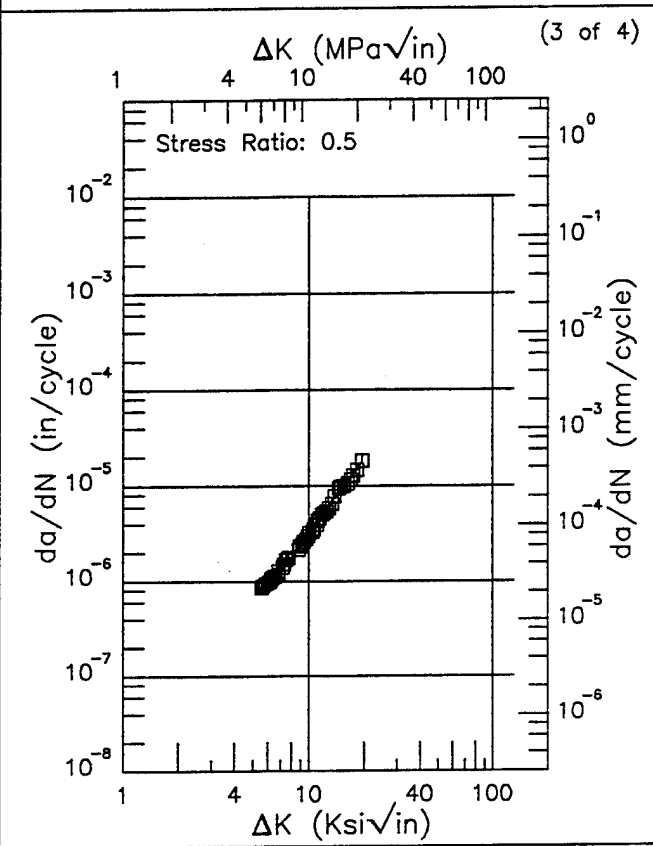


Figure 6.15.3.1.2

Ti-6-2-4-6 R

Condition/Ht:
 Form: Extrusion
 Specimen Type: CT
 Orientation: L-T
 Frequency: 20 Hz
 Environment: LAB AIR;800°F

Yield Strength: 147.2 ksi
 Ult. Strength: 159.1 ksi
 Specimen Thk: 0.3 in.
 Specimen Width: 1.4 in.
 Ref: UD001



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
5.54 (min)	0.814
6.	0.976
7.	1.35
8.	1.78
9.	2.31
10.	3.01
13.	6.52
16.	10.7
19.33 (max)	17.8

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
4.89 (min)	0.784
5.	0.798
6.	0.995
7.	1.32
8.	1.80
9.	2.48
10.	3.42
13.	6.74
13.67 (max)	7.16

RMS % Error	Life Prediction Ratio Summary
4.18	0. .5 .8 1.25 2.

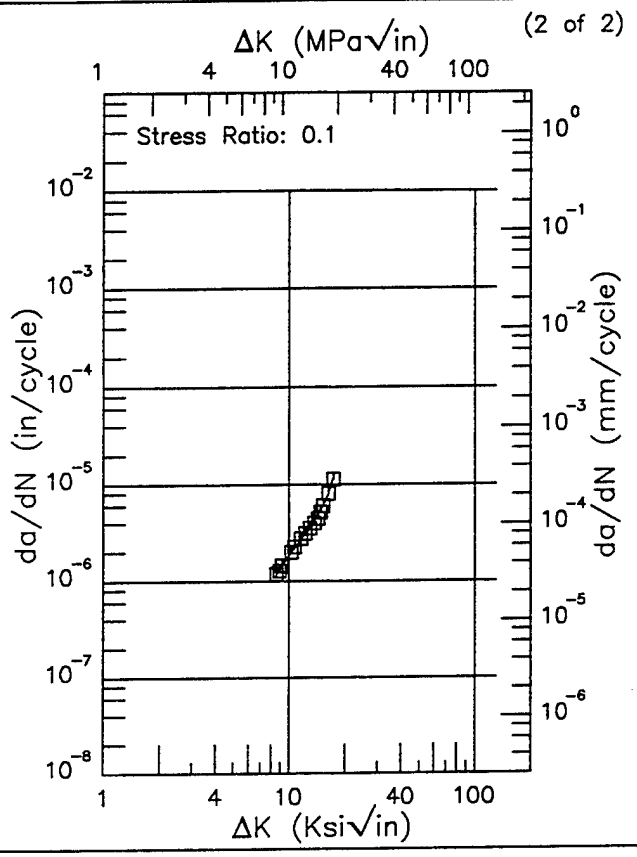
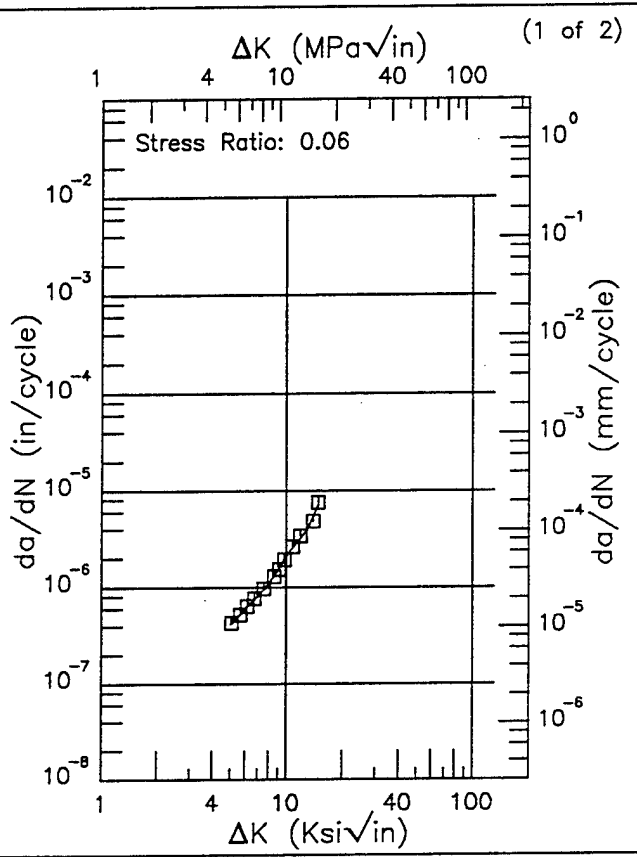
RMS % Error	Life Prediction Ratio Summary
9.87	0. .5 .8 1.25 2.

Figure 6.15.3.1.2 (Concluded)

R | Ti-6-2-4-6 |

Condition/Ht: 1690F 2HRS AC 1550F 2HRS OQ 1100F 8HRS AC
 Form: 1.8 - 2.8 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Frequency: 30 Hz
 Environment: LAB AIR;800°F

Yield Strength: 165.5 - 166 ksi
 Ult. Strength: 180.8 - 182.3 ksi
 Specimen Thk: 0.077 - 0.082 in.
 Specimen Width: 1.75 in.
 Ref: PW002



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
5.05 (min)	0.417
6.	0.603
7.	0.805
8.	1.08
9.	1.52
10.	2.12
13.	4.00
14.76 (max)	7.28

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.51 (min)	1.13
9.	1.35
10.	1.83
13.	3.55
16.	7.15
17.28 (max)	11.4

RMS % Error	Life Prediction Ratio Summary
3.56	0. .5 .8 1.25 2. ---

RMS % Error	Life Prediction Ratio Summary
4.35	0. .5 .8 1.25 2. ---

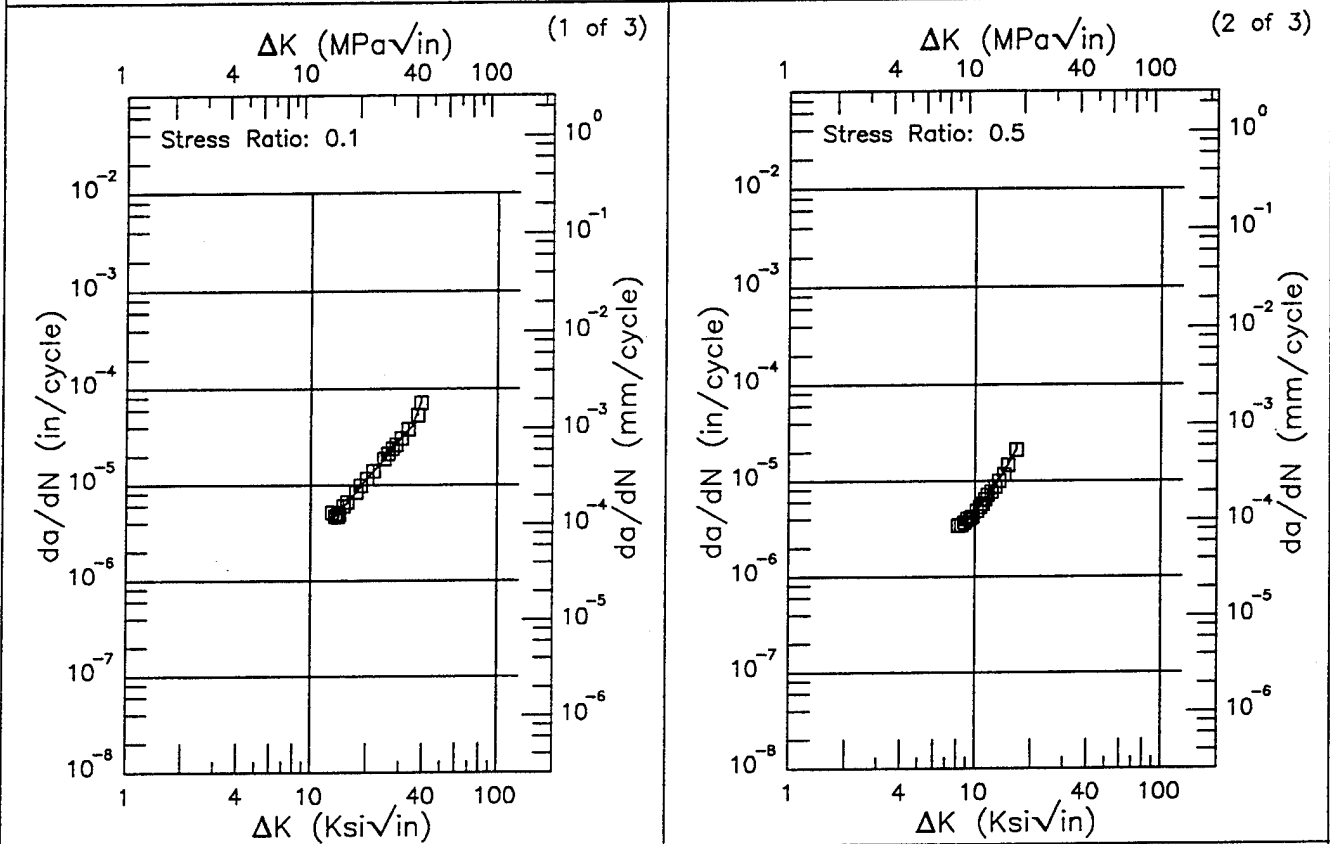
Figure 6.15.3.1.3

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R | Ti-6-2-4-6 |

Condition/Ht: 1690F 2HRS AC 1550F 2HRS OQ 1100F 8HRS AC
 Form: 1.8 - 2.8 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Frequency: 0.2 Hz
 Environment: LAB AIR;800°F

Yield Strength: 165.5 - 166 ksi
 Ult. Strength: 180.8 - 182.3 ksi
 Specimen Thk: 0.079 - 0.083 in.
 Specimen Width: 1.75 in.
 Ref: PW002



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)	ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
13.03 (min)	4.53	8.16 (min)	3.36
16.	6.74	9.	3.77
25.	11.8	10.	4.62
30.	18.8	13.	9.03
35.	29.7	16.	18.1
39.00 (max)	41.4	16.58 (max)	21.3
	69.8		

RMS % Error 5.05	Life Prediction Ratio Summary 0. .5 .8 1.25 2.	RMS % Error 2.60	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
---------------------	---	---------------------	---

Figure 6.15.3.1.4

Ti-6-2-4-6 R

Condition/Ht: 1690F 2HRS AC 1550F 2HRS OQ 1100F 8HRS AC
 Form: 1.8 - 2.8 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Frequency: 0.2 Hz
 Environment: LAB AIR;800°F

Yield Strength: 165.5 - 166 ksi
 Ult. Strength: 180.8 - 182.3 ksi
 Specimen Thk: 0.079 - 0.083 in.
 Specimen Width: 1.75 in.
 Ref: PW002

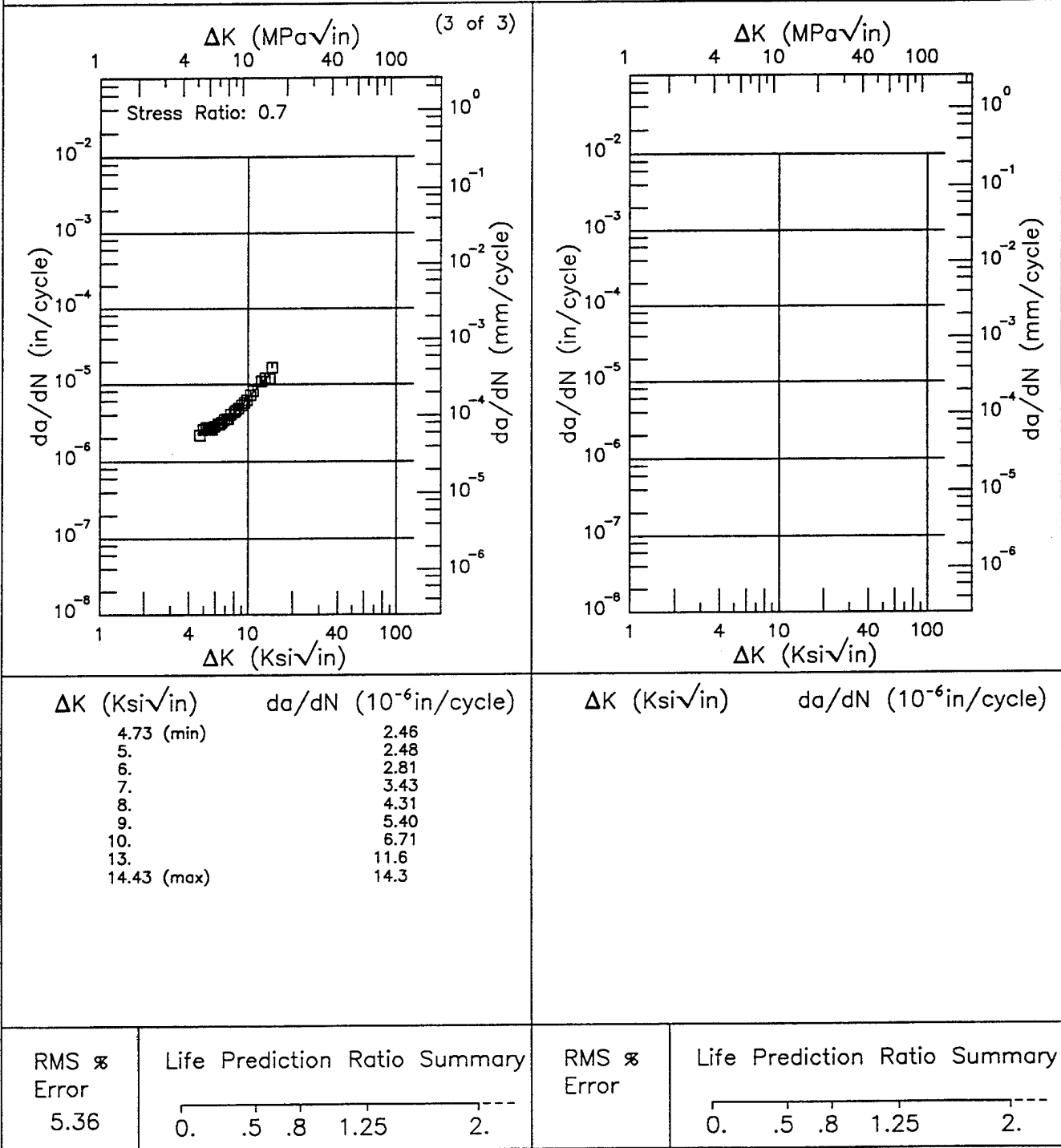
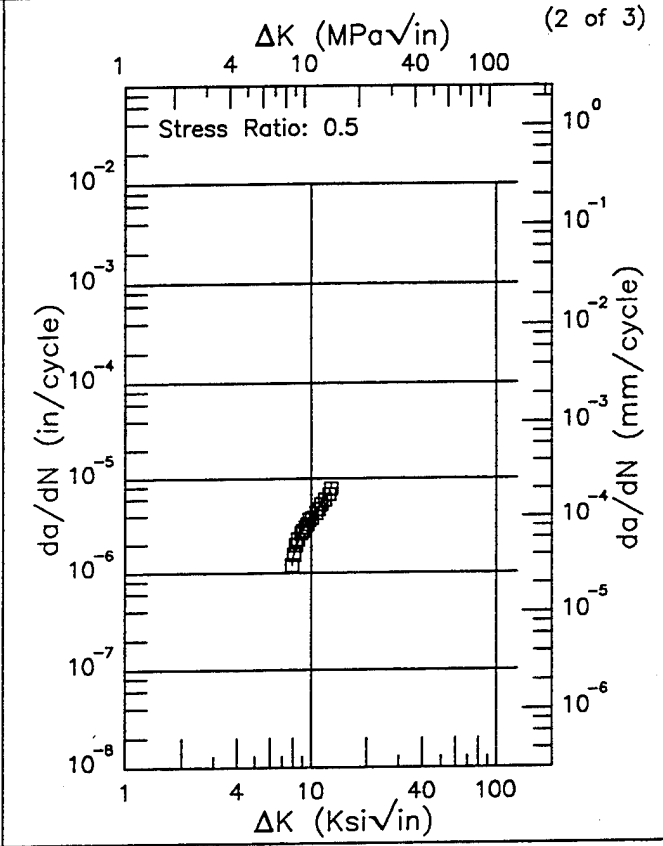
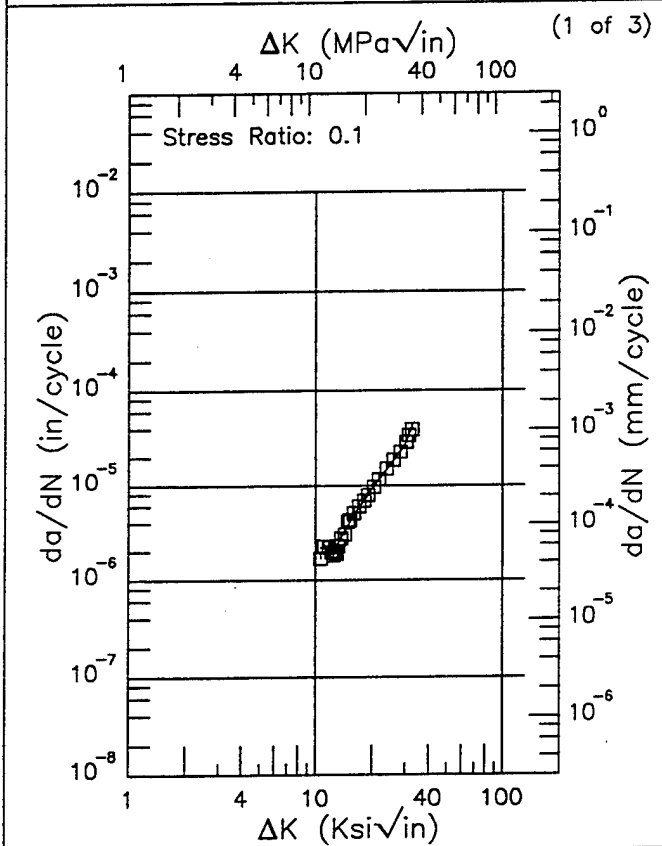


Figure 6.15.3.1.4 (Concluded)

R | Ti-6-2-4-6 |

Condition/Ht: 1690F 2HRS AC 1550F 2HRS OQ 1100F 8HRS AC
 Form: 2.8 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Frequency: 0.2 Hz
 Environment: LAB AIR;600°F

Yield Strength: 166 ksi
 Ult. Strength: 182.3 ksi
 Specimen Thk: 0.08 - 0.081 in.
 Specimen Width: 1.75 in.
 Ref: PW002



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.60 (min)	2.18
13.	2.30
16.	4.93
20.	9.23
25.	17.0
30.	27.1
32.52 (max)	40.6

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.88 (min)	1.29
8.	1.48
9.	2.85
10.	3.69
12.83 (max)	7.34

RMS % Error	Life Prediction Ratio Summary
9.18	

RMS % Error	Life Prediction Ratio Summary
4.98	

Figure 6.15.3.1.5

Ti-6-2-4-6

R

Condition/Ht: 1690F 2HRS AC 1550F 2HRS OQ
 Form: 2.8 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Frequency: 0.2 Hz
 Environment: LAB AIR;600°F

Yield Strength: 166 ksi
 Ult. Strength: 182.3 ksi
 Specimen Thk: 0.08 - 0.081 in.
 Specimen Width: 1.75 in.
 Ref: PW002

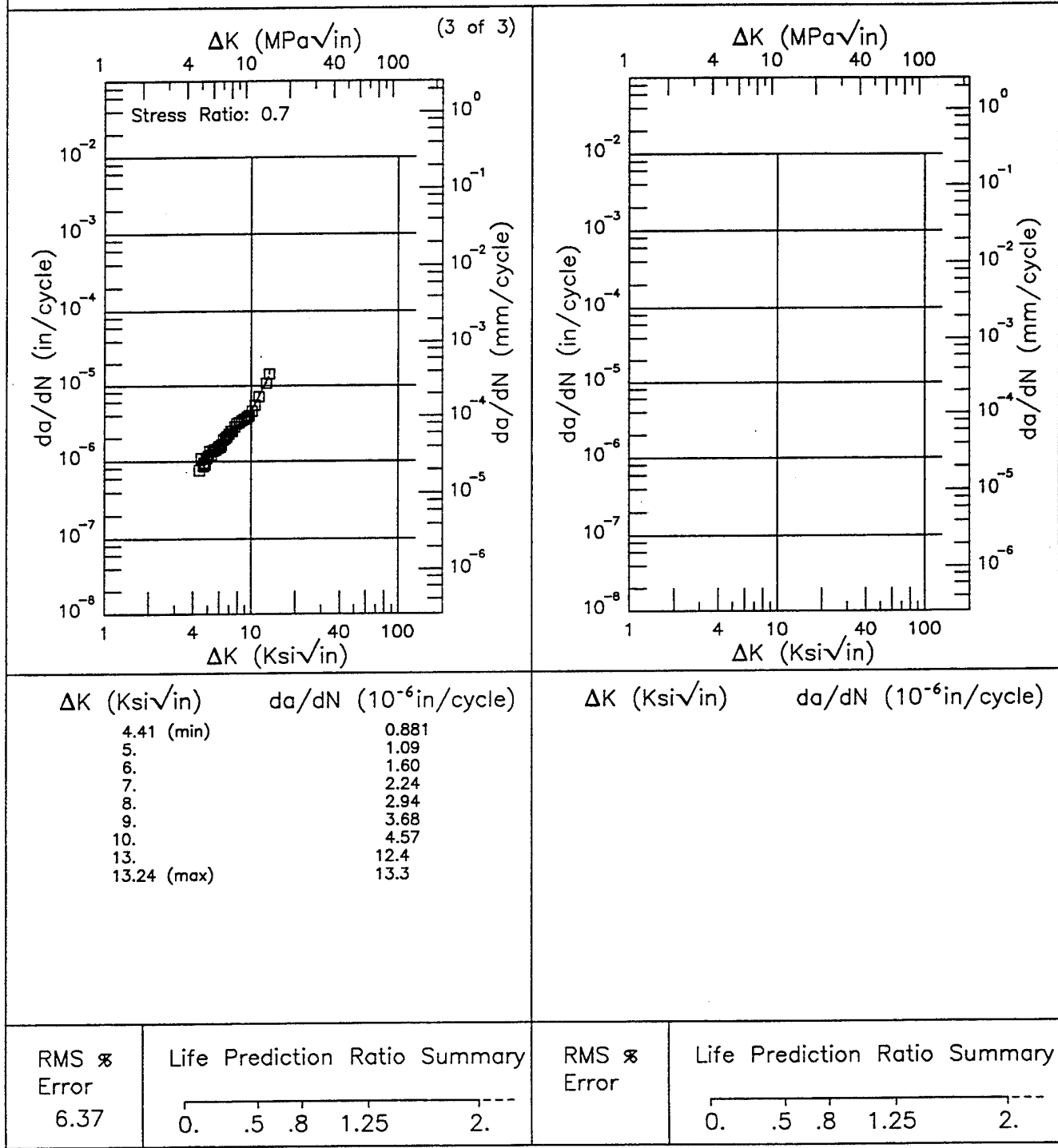


Figure 6.15.3.1.5 (Concluded)

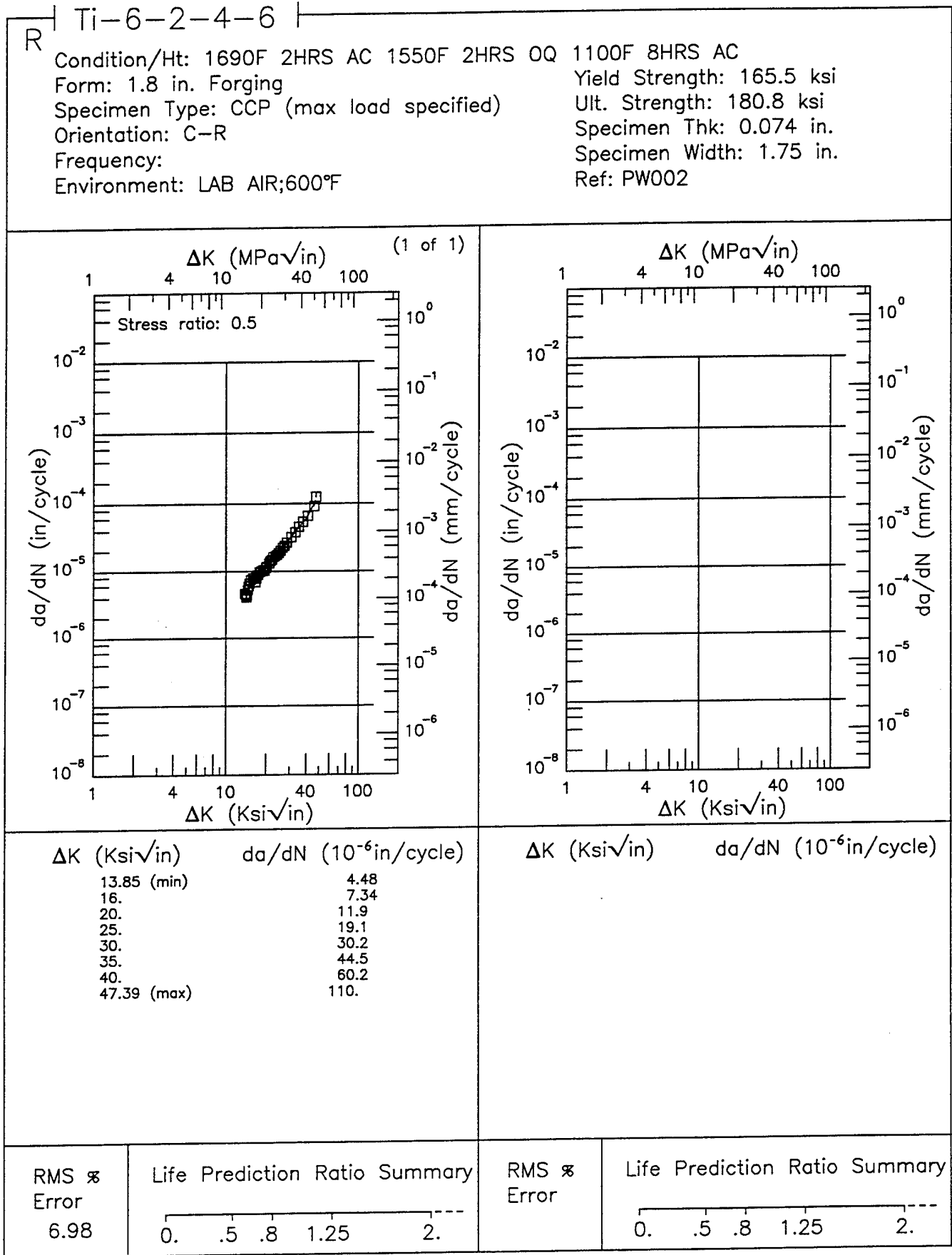


Figure 6.15.3.1.6

Ti-6-2-4-6 R

Condition/Ht: 1690F 2HRS AC 1550F 2HRS OQ 1100F 8HRS AC
 Form: 2.8 in. Forging Yield Strength: 166 ksi
 Specimen Type: CCP (max load specified) Ult. Strength: 182.3 ksi
 Orientation: C-R Specimen Thk: 0.079 - 0.08 in.
 Frequency: 0.2 Hz Specimen Width: 1.75 in.
 Environment: LAB AIR; RT Ref: PW002

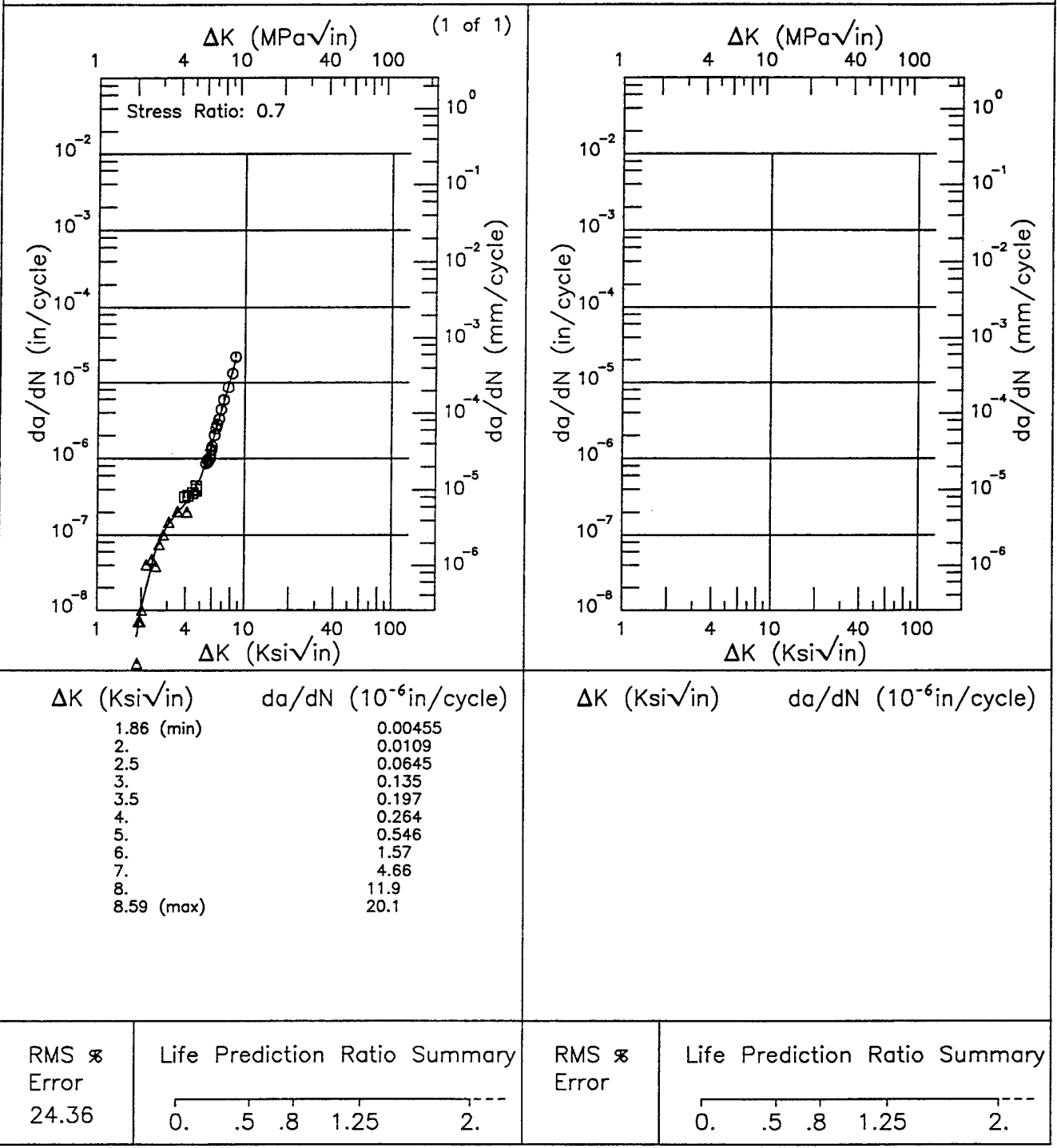


Figure 6.15.3.1.7

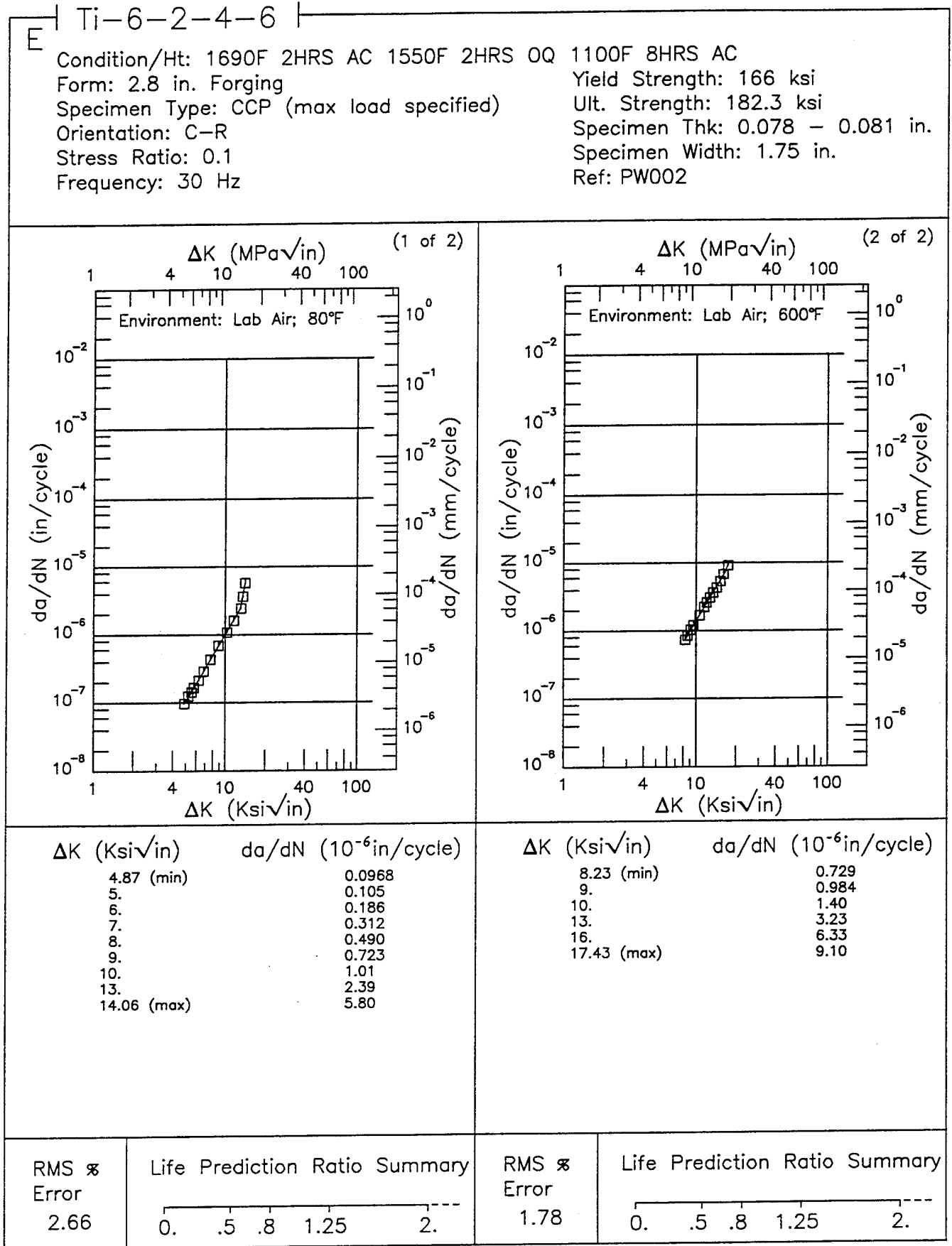


Figure 6.15.3.1.8

TABLE 6.16.1.1

**MEAN PLANE STRAIN FRACTURE TOUGHNESS
FOR TITANIUM ALLOY Ti-6Al-4V AT ROOM TEMPERATURE**

Product Form	Condition/Heat Treatment	K_{Ic} ($ksi\sqrt{in}$)									
		Specimen Orientation					Specimen Orientation				
		L-T		T-L			L-T		T-L		
Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n
Plate	1750F 1HR FC TO 1100F	---	---	---	91.5	2.1	2	---	---	---	---
	1750F 1HR FC TO RT	71.8	3.2	2	91.6	1.3	2	---	---	---	---
	1750F 2HR WQ 1000F 2HR AC 1300F 2HR AC STA	41.4	2.3	2	---	---	---	---	---	---	---
	ANNEALED 1375F 3HR AC	60.4	5.5	2	---	---	---	---	---	---	---
	BETA PROCESSED - MA	94.9	4.8	3	---	---	---	---	---	---	---
	MA	74.4	32.6	3	91.6	24.4	7	---	---	---	---
	RA	82.8	7.8	22	80.8	10.8	22	---	---	---	---
	STA	---	---	---	42.6	2.	3	---	---	---	---
	1700F 6HR AC 1400F 6HR AC	75.9	4.2	6	81.2	5.8	6	---	---	---	---
	1750F 1HR WQ 1000F 4HR	---	---	---	79.3	4.9	3	---	---	---	---
Forging	AB FORGED-MA ALPHA-BETA FORGED MA	---	---	---	35.4	2.7	4	---	---	---	---
	ANNEALED	70.8	15.9	4	67.3	13.6	6	---	---	---	---
	ANNEALED 1300F 4HR AC	58.1	1.2	3	62.2	3.	3	68.1	1.	2	

Ti-6Al-4V

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TABLE 6.16.1.1 (CONCLUDED)
MEAN PLANE STRAIN FRACTURE TOUGHNESS
FOR TITANIUM ALLOY Ti-6Al-4V AT ROOM TEMPERATURE

Product Form	Condition/Heat Treatment	K_{Ic} (ksi \sqrt{in})											
		Specimen Orientation						Specimen Orientation					
		L-T			T-L			S-L			S-L		
Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n		
Forging (Cont'd)	B FORGED-MA BETA FORGED MA 1300F 2HR AC	70.6	4.9	3	---	---	---	---	---	---	---	---	
	B FORGED-MA BETA FORGED MA 1300F 2HR AC	---	---	---	71.	0.4	3	73.9	2.5	2	2	2	
	MA 1300F 2HR AC	50.9	6.9	4	49.5	3.9	3	43.6	5.8	3	3	3	
Extrusion	RA	83.6	5.5	41	83.9	6.9	50	88.9	3.2	9	9	9	
	ANNEALED	82.6	5.3	5	85.2	6.5	6	---	---	---	---	---	
	MA	83.5	3.1	5	87.5	4.1	6	---	---	---	---	---	
Forged Bar	AS RECEIVED	57.1	10.4	14	54.9	10.8	21	---	---	---	---	---	
	B FORGED BETA FORGED REHEATED TO 1950F DRAWN TO SIZE	---	---	---	42.6	4.3	4	---	---	---	---	---	
	ANNEALED	79.6	9.6	2	---	---	---	---	---	---	---	---	
Billet	ANNEALED 1000F 2HR AC	50.9	0.6	2	---	---	---	---	---	---	---	---	
	DBA	68.2	9.7	9	64.2	11.8	13	---	---	---	---	---	
	MA 1300F 2HR AC	84.	3.4	3	---	---	---	---	---	---	---	---	

TABLE 6.16.1.2.1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: L-S ENVIRONMENT: H.H.A.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
RA	PLATE	0.1	10	2.5	5.0	10.0	20.0	50.0	100.0
						1.11	12.82		

Ti-6Al-4V

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TABLE 6.16.1.2.2

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: Distilled Water

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi $\sqrt{\text{in}}$)					
				2.5	5.0	10.0	20.0	50.0	100.0
ANNEALED	FORGING	0.1	1			0.37	14.31		
						0.39	7.49		
	EXTRUSION	0.8	1		0.22	1.37			

TABLE 6.16.1.2.3

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: Dry Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
BA	PLATE	0.1	6			0.15	8.39	111.47		
		0.3	6			0.29	10.1			
		0.5	6			1.15	11.38			
BETA PROCESSED - MA	SHEET	0.1	10							52.46
	PLATE	0.1	10				4.72	42.1		
		0.08	1				16.7	757.11		
DBTC	PLATE	0.3	1			0.6	18.69			
		0.3	1				12.85	281.01		
		0.08	1-6			1.67	13.5			
MA	SHEET	0.3	6			2.27				
		0.5	6			2.3	21.08			
		0.3	1				69.42			
MA	PLATE	0.3	1				21.56			
		0.3	1-6			0.43	16.1	361.47		
		0.08	6				11.99			

Ti-6Al-4V

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TABLE 6.16.1.2.3 (CONCLUDED)

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: L-T

ENVIRONMENT: Dry Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
MA	EXTRUSION	0.08	6			0.34	12.64			
		0.3	6				16.36			
		0.5	6					16.88		
RA	PLATE	0.08	6			0.44	10.09	204.33		
		0.3	6			1.46	13.34			
		0.5	6			2.16	18.1			
	0.7	6			3.21					
	FORGING	0.08	1			0.39	5.83	172.77		
		0.08	6					6.58		
0.3		6			1.16	15.47	338.75			

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TABLE 6.16.1.2.4

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: L-T

ENVIRONMENT: H.H.A.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)														
				ΔK Level (Ksi \sqrt{in})														
				2.5	5.0	10.0	20.0	50.0	100.0									
BA	PLATE	0.1	0.1															
		0.3	0.1			3.75	111.63											
		0.5	0.1			9.28												

Ti-6Al-4V

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TABLE 6.16.1.2.5

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: JP4

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.6	5.0	10.0	20.0	50.0	100.0	
BA	FORGING	0.02	0.1-20							
MA	SHEET	0.08	6		1.56		13.92			
RA	PLATE	0.08	1		0.04		12.39			
	FORGING	0.08	1				4.95			

TABLE 6.16.1.2.6

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: L-T

ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi/ $\sqrt{\text{in}}$)						
				2.5	6.0	10.0	20.0	50.0	100.0	
ALPHA-BETA FORGE-ANNEALED	FORGING	0.1	30			1.02				
		0.1	30		0.05					
		0.1	30		0.05					
		-1	5			0.87	10.33			
		0.1	5			0.29	9.53			
		0.1	15				11.12			
ANNEALED	FORGING	0.1	20			0.38				
		0.4	10		0.05	1.32	16.33			
		0.4	20		0.02	1.09	17.6			
		0.8	15		0.07					
		-1	5			0.41	7.29			
		0.1	8			0.34	11.63			
EXTRUSION		0.1	15						8.68	242.97
		0.1	20			0.18				
		0.4	15			0.03	16.54			

Ti-6Al-4V

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TABLE 6.16.1.2.6 (CONTINUED)

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
TI-6Al-4V AT ROOM TEMPERATURE

ENVIRONMENT: Lab Air

ORIENTATION: L-T

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
ANNEALED	EXTRUSION	0.8	10		0.13	1.68				
		0.8	15		0.1	1.37				
		0.8	30		0.08	1				
ANNEALED AT 1375F 3HRS AC	BILLET	0.02	10-20			0.27	11.05			
		0.02	10-20			0.25	11.03			
		0.02	0.1-20				2.55	105.12		
BETA PROCESSED - MA	PLATE	0.1	1				0.87			
		0.5	1			1.84				
		-1	10			1.14	12.91	322.86		
MA	PLATE	0.02	0.1-20			0.32	10.8			
		0.02	1-27			0.11				
		0.02	0.1-30			0.15	6.77			
		0.02	5-30			0.02	0.4	15.79		
		0.04	20						93.19	
		0.05	20				6.34			

TABLE 6.16.1.2.6 (CONCLUDED)

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
MA	PLATE	0.3	20			0.99			
		0.5	10			8.51	97.18		
	FORGING	0.02	1-30					8.18	292.11
		0.1	10			4.04			
	EXTRUSION	0.1	1-10					9.35	189.23
		0.1	1-20					12.2	225.55
		0.3	10					0.39	8.82
	UNSPECIFIED	0.55	10					0.84	

Ti-6Al-4V

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TABLE 6.16.1.2.7

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: S.S.W.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
BA	FORGING	0.02	0.1-20							
MA	EXTRUSION	0.1	1-10				4.83	164.25		
							11.38	270.5		

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TABLE 6.16.1.2.8

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: L-T

ENVIRONMENT: S.T.W.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level ($K\sqrt{\text{in}}$)					
				2.5	5.0	10.0	20.0	50.0	100.0
BA	PLATE	0.1	1				11.02	160.87	
	PLATE	0.08	1			0.69	13.62		
MA	SHEET	0.08	1			3.24			
		0.3	1				76.32		
	PLATE	0.3	1			0.63	34.25		
	EXTRUSION	0.1	1-10				10.89	210.05	
RA	PLATE	0.08	0.1-1			0.71	12.91		
		0.08	1			1.19	22.42	264.77	
		0.08	1			0.37	8.03		
	FORGING	0.3	1			2.2	24.11		
		0.5	1			4.18			
		0.08	1			1.58	38.76		

Ti-6Al-4V

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TABLE 6.16.1.2.9

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
TI-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: T-S

ENVIRONMENT: 3.5% NaCl

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	6.0	10.0	20.0	50.0	100.0	
BA	PLATE	0.1	10							
RA	PLATE	0.1	10			0.62	18.38			
										49.71

TABLE 6.16.1.2.10

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: 3.5% NaCl

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Kst/ $\sqrt{\text{in}}$)					
				2.5	5.0	10.0	20.0	50.0	100.0
BA	PLATE	0.1	1				37.41	255.71	
		0.1	10		0.79	13.63	102.98		
		0.	15		0.63				
		0.1	0.1		3.5	44.65			
		0.1	0.1						
		0.1	10		4.91	61.29			
RA	PLATE	0.1	10				67.53		
		0.5	0.1		2.03				
		0.5	1		2.67	114.91			
		0.5	10		34.23	170.88			
		0.1	1			29.74			
		0.1	10			9.28			
RA(FAST COOLED)	PLATE	0.1	1						
STRESS RELIEVED E.B. WELDMENT (HAZ)	WELDMENT	0.1	10						
STRESS RELIEVED E.B. WELDMENT (WELD ZONE)	WELDMENT	0.1	0.1						

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TABLE 6.16.1.2.11

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: ALT JP4/H₂O (D)

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Kst/ \sqrt{in})						
RA	PLATE	0.5	1	2.5	5.0	10.0	20.0	50.0	100.0	
						1.68	37.77			

TABLE 6.16.1.2.12

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: Distilled Water

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
ANNEALED	FORGING	0.1	1		0.15	2.64	28.49		
		0.8	1		8.14	17.51			
	EXTRUSION	0.1	1				7.72		
		0.8	1		0.34	2.04			
BA	PLATE	0.1	0.1					123.05	
		0.	16			0.43			
RA	PLATE	0.1	0.1				22.62	270.27	
		0.5	0.1				21.8		
		0.5	1			2.77	28.03		
STRESS RELIEVED E.B. WELDMENT (WELD ZONE)	WELDMENT	0.1	0.1						
		0.1	10						

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TABLE 6.16.1.2.13

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ENVIRONMENT: Dry Air

ORIENTATION: T-L

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
DB	PLATE	0.08	6			0.33	12.47			
		0.3	6				34.69			
DBTC	PLATE	0.08	1				7.71	161.59		
		0.3	1			0.49	12.28	276.36		
MA	SHEET	0.08	6				9.53			
	EXTRUSION	0.08	6				12.12			
RA	PLATE	0.08	6			0.43	10.76			
		0.08	6			0.46	11.32	232.73		
		0.1	0.1					224.79		
	FORGING	0.5	0.1			1.08	20.39			
		0.5	1				18.21			
		0.08	6				7.57	137.41		
		0.5	6			2.62	19.72			

TABLE 6.16.1.2.14

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 TI-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: F.C.S.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
RA	PLATE	0.08	1	2.5	5.0	10.0	20.0	50.0	100.0
							9.1		

Ti-6Al-4V

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TABLE 6.16.1.2.15

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: H.H.A.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
RA	PLATE	0.1	10			3.86	22.11		
		0.1	10				14.61		
		0.5	10		0.34	3.66	65.81		
		0.5	10			6.41	26.62		

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TABLE 6.16.1.2.16

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE**

ORIENTATION: T-L

ENVIRONMENT: JP4

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi $\sqrt{\text{in}}$)						
				2.5	5.0	10.0	20.0	50.0	100.0	
BA	FORGING	0.02	0.1-20				2.09	97.77		
RA	PLATE	0.08	1		0.69		14.92			
STRESS RELIEVED E.B. WELDMENT (HAZ)	WELDMENT	0.1	0.1				14.55	621		
		0.1	10				9.36			

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TABLE 6.16.1.2.17

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
Ti-6Al-4V AT ROOM TEMPERATURE**

ORIENTATION: T-L

ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)								
				ΔK Level (Ksi \sqrt{in})								
				2.5	5.0	10.0	20.0	50.0	100.0			
ANNEALED	FORGING	0.1	5									
		0.1	20			0.65						
		0.4	10		0.1	2.41						
		0.4	20		0.05	1.41						
		0.8	15		0.23							
		0.8	30		0.25	6.32						
	EXTRUSION	0.1	5-10			0.34				7.95		
		0.1	15							6.84	161.87	
		0.1	20		0.01	0.2						
		0.4	5-15		0.06	0.84				14.05		
		0.4	15		0.03	0.4				13.39		
		0.8	30		0.08	0.87						
AS WELDED E.B. WELDMENT (HAZ)	WELDMENT	0.1	10						6.52			
AS WELDED E.B. WELDMENT (WELD ZONE)	WELDMENT	0.1	10						5.94			
BA	FORGING	0.02	0.1-20						1.93	94.89		

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TABLE 6.16.1.2.17 (CONCLUDED)

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: T-L

ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi $\sqrt{\text{in}}$)						
				2.5	6.0	10.0	20.0	50.0	100.0	
MA	EXTRUSION	0.1	5-20							
RA	PLATE	0.1	10				13.75	276.81		
STRESS RELIEVED E.B. WELDMENT (HAZ)	WELDMENT	0.1	10				23.53			
STRESS RELIEVED E.B. WELDMENT (WELD ZONE)	WELDMENT	0.1	10				11.35	508.19		
							10.93			

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TABLE 6.16.1.2.18

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: S.S.W.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
BA	FORGING	0.02	0.1-20							
MA	EXTRUSION	0.1	1-10				3.76	126.14		
							17.18	209.03		

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TABLE 6.16.1.2.19

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
Ti-6Al-4V AT ROOM TEMPERATURE**

ORIENTATION: T-L ENVIRONMENT: S.T.W.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi $\sqrt{\text{in}}$)						
				2.6	6.0	10.0	20.0	50.0	100.0	
BA	PLATE	0.1	0.1			0.58	6.43	400.25		
							36.67	185.25		
							20.7	98.95		
DB	PLATE	0.08	1				7.9	225.95		
DB + 2DBTC	PLATE	0.08	1				7.62	249.01		
DB + 4DBTC	PLATE	0.08	1				9.51	170.63		
DBT + PC	PLATE	0.08	1				11.06			
DBTC	PLATE	0.08	1			2.81	17.87			
MA	EXTRUSION	0.1	1-10			0.62	11.14			
							25.92	354.12		
RA	PLATE	0.08	1			0.65				
						2.08	48.68			
						0.63	10.46	294.78		
						0.1	0.1		820.85	
						0.1	0.1			

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TABLE 6.16.1.2.19 (CONCLUDED)

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: S.T.W.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Kst/\sqrt{in})					
				2.5	6.0	10.0	20.0	50.0	100.0
RA	PLATE	0.5	1			2.01	51.26		
		0.08	1			0.75	16.9	235.07	
	FORGING	0.08	1			0.62	14.24	185.08	
		0.5	1			3.41	58.02		

TABLE 6.16.1.2.20

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: STW/JP4

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
DBTC	PLATE	0.08	1	2.5	5.0	10.0	20.0	50.0	100.0

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TABLE 6.16.1.2.21

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: WATER SAT JP4

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
RA	PLATE	0.1	0.1						
		0.5				17.23			
		0.5			1.55	19.08			

TABLE 6.16.1.2.22

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: S-T ENVIRONMENT: Dry Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)								
				ΔK Level (Ksi \sqrt{in})								
DB + 2B7C	PLATE	0.08	6	2.6	6.0	10.0	20.0	50.0	100.0			

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TABLE 6.16.1.2.23

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: S-T

ENVIRONMENT: S.T.W.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
DB	PLATE	0.08	1							
DB + TR	PLATE	0.08	1			0.72	19.87			9.96

TABLE 6.16.1.2.24

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: R-C ENVIRONMENT: Argon

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Kst/ \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
1550F 4HRS FC 1000F 4HRS Argon COOLED	FORGING	0.1	10				4.31		
1750F 4HRS Argon COOLED	FORGING	0.1	10				4.38		
1950F 4HRS WQ 1000F 4HRS Argon COOLED	FORGING	0.1	10				2.52	147.7	

TABLE 6.16.1.2.25

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: C-R ENVIRONMENT: Argon

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Kst $_{1/2}$ in)					
				2.5	5.0	10.0	20.0	50.0	100.0
1750F 4HRS Argon COOLED	FORGING	0.1	10				5		
1950F 4HRS WQ 1000F 4HRS Argon COOLED	FORGING	0.1	10				3.63		

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TABLE 6.16.1.2.26

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 TI-6Al-4V AT ROOM TEMPERATURE

ENVIRONMENT: Lab Air

ORIENTATION: C-R

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (HZ)	FCGR (10^{-6} in/cycle)												
				ΔK Level (Ksi/in)												
				2.5	5.0	10.0	20.0	50.0	100.0							
1775F 1HR WQ 1675F 1HR WQ 1000F 4HR AC	DISK	0.05	0.33-10													
		0.03	0.33			0.82	11.45									
1775F 1HR WQ 1675F 1HR WQ 1000F-1200F 2-8HR AC	DISK	0.03	0.5				9.93									
		0.25	0.33						22.1							
		0.25	0.5								17.73					
		-1	0.16									1.14				
UNSPECIFIED	UNSPECIFIED	-0.5	0.16								1.21					
		0.1	20									10.35				
		0.3	20									1.66	16.26			
		0.5	20									1.82				
		0.7	20										0.15			

TABLE 6.16.1.2.27

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V AT ROOM TEMPERATURE

ORIENTATION: UNSPECIFIED ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	$FCGR (10^{-6} \text{ in/cycle})$					
				$\Delta K \text{ Level (Ksi}/\sqrt{\text{in}})$					
				2.5	5.0	10.0	20.0	50.0	100.0
BA	SHEET	0.1	10			1.84	14.44	122.21	
		0.1	10			4.66			
		0.3	10				14.91	210.93	
HIP 1650F 16 KSI UNSPECIFIED	CASTING UNSPECIFIED	0.1	0.1-20			3.87	26.35		
		0.1	30			0.16	5.23		

TABLE 6.16.2.1

TITANIUM Ti-6Al-4V K_{10}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{10}/\sqrt{a})^2$ (in.)	K_{10}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{10} (KSI) $\cdot \sqrt{(in.)}$	K_{10} MEAN	STAN DEV		
1450F 1HR AC	Plate	1.00	R.T.	T-L	146.0	—	1.000	CT	—	0.23	44.00	—	—	1981	NR001
		1.40			118.0	3.006	1.317	CT	1.605	0.97	74.10			1973	85857
1700F 6 HR AC 1400F 6 HR AC	Forging	1.40	R.T.	L-T	119.0	3.002	1.365	CT	1.615	1.16	81.20	76.9	4.2	1973	85857
		1.40				1.251	CT	1.587	0.91	71.60	1973			85857	
		1.40				1.350	CT	1.588	1.16	81.10	1973			85857	
		1.40				1.301	CT	1.583	0.94	73.00	1973			85857	
		1.40				1.290	CT	1.604	1.07	74.40	1973			85857	
		1.40				1.357	CT	1.564	1.17	86.30	1973			85857	
1700F 6 HR AC 1400F 6 HR AC	Forging	1.40	R.T.	T-L	126.0	3.002	1.311	CT	1.599	0.91	76.00	81.2	5.8	1973	85857
		1.40				1.353	CT	1.617	1.18	87.50	1973			85857	
		1.40				1.325	CT	1.500	0.88	74.80	1973			85857	
		1.40				1.279	CT	1.568	1.09	85.20	1973			85857	
		1.40				1.336	CT	1.632	0.92	77.10	1973			85857	
		3.00				2.000	WOL-CT EQ.	1.998	0.43	66.30	1966			76411	
1750F 1 HR WQ 1000F 4 HR	Forging	3.00	-75	T-L	159.0	5.100	2.000	WOL-CT EQ.	2.091	0.48	70.00	68.2	2.6	1966	76411
		3.00			159.0	5.100	2.000	WOL-CT EQ.	2.088	0.63	76.80			1966	76411
1750F 1 HR WQ 1000F 4 HR	Forging	3.00	-40	T-L	153.0	5.100	2.000	WOL-CT EQ.	2.069	0.47	66.20	71.5	7.5	1966	76411
		3.00			147.0	5.100	2.000	WOL-CT EQ.	1.998	0.51	66.40			1966	76411
1750F 1 HR WQ 1000F 4 HR	Forging	3.00	0	T-L	147.0	5.100	2.000	WOL-CT EQ.	2.081	0.61	72.90	69.7	4.6	1966	76411
		3.00			147.0	5.100	2.000	WOL-CT EQ.	2.081	0.61	72.90			1966	76411

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TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic}/\sqrt{YS})^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (KSI $\cdot \sqrt{\text{in.}}$)	K_{Ic} MEAN	STAN DEV		
1750F 1 HR WQ 1000F 4 HR	Forging	3.00	32	T-L	148.0	5.100	2.000	WOL-CT EQ.	2.041	0.47	64.40	65.0	0.8	1966	76411
		3.00				5.100	2.000	WOL-CT EQ.	2.011	0.49	65.50	65.0	0.8	1966	76411
1750F 1 HR WQ 1000F 4 HR	Forging	3.00	R.T.	T-L	140.0	5.100	2.000	WOL-CT EQ.	1.966	0.91	84.90	78.3	4.9	1966	76411
		3.00				5.100	2.000	WOL-CT EQ.	2.019	0.76	76.80	78.3	4.9	1966	76411
1750F 1 HR WQ 1000F 4 HR	Forging	3.00	100	T-L	133.0	5.100	2.000	WOL-CT EQ.	1.969	0.75	72.90	71.1	2.5	1966	76411
		3.00				5.100	2.000	WOL-CT EQ.	2.049	0.68	69.30	71.1	2.5	1966	76411
1750F 1 HR WQ 1000F 4 HR	Forging	3.00	160	T-L	127.0	5.100	2.000	WOL-CT EQ.	1.970	0.85	74.00	78.0	5.7	1966	76411
		3.00				5.100	2.000	WOL-CT EQ.	2.033	1.00	82.00	78.0	5.7	1966	76411
1750F 1HR FC TO 1100F AC	Plate	1.50	R.T.	L-T	120.0	4.005	1.501	CT	1.991	1.20	83.00	--	--	1973	85836
1750F 1HR FC TO 1100F AC	Plate	1.50	R.T.	T-L	120.0	4.004	1.501	CT	1.888	1.41	90.00	91.5	2.1	1973	85836
		1.50				4.000	1.500	CT	--	1.50	93.00	91.5	2.1	1974	89004
1750F 1HR FC TO RT	Plate	1.50	R.T.	L-T	120.0	4.004	1.502	CT	1.717	0.84	69.50	71.9	3.2	1973	85836
		1.50				4.004	1.501	CT	1.718	0.95	74.00	71.9	3.2	1973	85836
1750F 1HR FC TO RT	Plate	1.50	R.T.	T-L	120.0	4.008	1.502	CT	1.956	1.43	90.60	91.6	1.3	1973	85836
		1.50				4.003	1.502	CT	1.986	1.49	92.50	91.6	1.3	1973	85836
1750F 2 HR FC TO 900F AT 100F/HR AC	Forging	3.00	R.T.	L-T	115.0	3.005	1.500	CT	1.716	1.20	79.80	--	--	1973	88440
1750F 2 HR FC TO 900F AT 100F/HR AC	Forging	3.00	R.T.	T-L	130.0	3.005	1.500	CT	1.561	1.21	90.50	--	--	1973	88440
		0.62				3.501	0.634	CT	1.865	0.17	39.80	--	--	1973	85836
1300F 2 HR AC STA	Plate	0.62	R.T.	L-T	150.0	3.501	0.633	CT	1.880	0.20	43.00	41.4	2.3	1973	85836

TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}																
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic}/TYS)^2$ (in.)	K_{Ic}			DATE	REFER	
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Ksi) $\sqrt{(in.)}$	K_{Ic} MEAN	STAN DEV			
AB FORGED-MA ALPHA-BETA- FORGED MILL ANNEALED	---	---	R.T.	---	133.0	2.500	1.250	1.250	CT	1.250	0.57	63.60		1973	90584 (1)	
	---	---			133.0	2.500	1.250	1.250	CT	1.250	0.50	59.40		1973	90584 (1)	
	---	---			134.0	2.500	1.250	1.250	CT	1.250	0.62	66.50		1973	90584 (1)	
	---	---			134.0	2.500	1.250	1.250	CT	1.250	0.41	54.10		1973	90584 (1)	
	---	---			134.0	2.500	1.250	1.250	CT	1.250	0.43	55.90	58.5	4.4	1973	90584 (1)
	---	---			134.0	2.500	1.250	1.250	CT	1.250	0.44	56.30		1973	90584 (1)	
	2.75	---			136.0	2.000	1.000	1.000	CT	1.000	0.44	57.20		1974	88962 (2)	
	2.75	---			136.0	2.000	1.000	1.000	CT	1.000	0.41	55.20		1974	88962 (2)	
	2.25	---			145.0	2.000	1.000	1.000	CT	1.065	0.18	38.60		1973	86688	
	2.25	---			145.0	3.000	1.500	1.500	CT	1.566	0.12	32.20		1973	86688	
AB FORGED-MA ALPHA-BETA- FORGED MILL ANNEALED	2.25	---	R.T.	T-L	145.0	2.000	1.000	1.000	CT	1.071	0.15	36.00	35.4	2.7	1973	86688
	2.25	---			145.0	2.000	1.000	1.000	CT	1.074	0.14	34.90		1973	86688	
	---	---			128.0	2.500	1.250	1.250	CT	1.250	0.74	69.90		1973	90584 (1)	
	---	---			128.0	2.500	1.250	1.250	CT	1.250	0.79	72.10		1973	90584 (1)	
AB FORGED-RA ALPHA-BETA FORGED RECRYSTALLIZED ANNEAL 1700F 4 HR FC TO 1000F AC	---	---	R.T.	---	128.0	2.500	1.250	1.250	CT	1.250	0.72	68.60		1973	90584 (1)	
	---	---			128.0	2.500	1.250	1.250	CT	1.250	0.70	68.00	67.8	3.3	1973	90584 (1)
	---	---			128.0	2.500	1.250	1.250	CT	1.250	0.74	69.60		1973	90584 (1)	
	---	---			128.0	2.500	1.250	1.250	CT	1.250	0.70	67.70		1973	90584 (1)	
	---	---			132.0	2.500	1.250	1.250	CT	1.250	0.60	64.50		1973	90584 (1)	
	---	---			132.0	2.500	1.250	1.250	CT	1.250	0.54	61.70		1973	90584 (1)	

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TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{I0}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{I0}/\sqrt{A})^2$ (in.)	K_{I0}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{I0} (Ksi) $\cdot \sqrt{A}$ (in.)	K_{I0} MEAN	STAN DEV		
ANNEALED	Forging	1.00	-65	L-T	153.3	2.003	1.001	CT	1.020	0.36	59.20	54.9	5.9	1987	DA006
		1.00				2.006	0.999	CT	1.074	0.36	58.50			1987	DA006
		3.00				2.000	1.000	CT	1.023	0.23	48.10			1987	DA007
ANNEALED	Forging	3.00	-65	T-L	154.5	2.000	0.958	CT	1.046	0.22	46.30	46.2	1.5	1987	DA007
		3.00				2.000	0.998	CT	1.043	0.24	47.60			1987	DA007
		1.50				3.010	1.491	CT	1.656	0.16	44.10			1987	DA006
		1.50				3.012	1.491	CT	1.628	0.20	46.87			1987	DA006
		3.00				3.001	1.501	CT	1.568	1.41	85.70			1973	85034
ANNEALED	Forging	3.00	R.T.	L-T	119.0	2.998	1.500	CT	1.556	1.22	83.10	70.7	15.9	1973	85034
		1.50				3.011	1.502	CT	1.573	0.50	58.90			1987	DA006
		1.50				3.011	1.502	CT	1.568	0.44	55.30			1987	DA006
		3.00				3.003	1.500	CT	1.549	1.46	90.40			1973	85034
ANNEALED	Forging	3.00	R.T.	T-L	120.0	3.001	1.494	CT	1.618	1.01	76.40	67.3	13.6	1973	85034
		3.00				3.007	1.495	CT	1.566	0.48	61.40			1987	DA007
		3.00				3.005	1.495	CT	1.558	0.52	63.60			1987	DA007
		1.50				3.010	1.506	CT	1.589	0.39	57.10			1987	DA006
		1.50				3.012	1.501	CT	1.625	0.36	54.70			1987	DA006

TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K _{1c}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	2.5° (K _{1c} /TYS) ^a (in.)	K _{1c}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K _{1c} (Ksi) √(in.)	K _{1c} MEAN	STAN DEV		
ANNEALED	Extrusion	1.75	3.000	1.498	CT	146.8	3.000	1.498	1.571	0.71	78.20		1987	DA007	
		1.75	3.010	1.497	CT	146.8	3.010	1.497	1.556	0.70	77.50		1987	DA007	
		1.50	3.005	1.501	CT	152.5	3.005	1.501	1.422	0.57	72.90	75.9	2.4	1987	DA006
		1.50	3.007	1.510	CT	152.5	3.007	1.510	1.427	0.61	75.10		1987	DA006	
ANNEALED	Extrusion	1.75	3.000	1.498	CT	149.2	3.000	1.498	1.554	0.70	79.20		1987	DA007	
		1.75	3.006	1.499	CT	149.2	3.006	1.499	1.598	0.67	77.10		1987	DA007	
		1.50	3.007	1.504	CT	157.2	3.007	1.504	1.431	0.58	75.50	76.9	1.7	1987	DA006
		1.50	3.008	1.502	CT	157.2	3.008	1.502	1.425	0.58	75.90		1987	DA006	
ANNEALED	Extrusion	4.00	4.003	1.629	CT	122.0	4.006	1.624	2.027	1.42	92.00		1973	85836 (1)	
		1.75	3.005	1.500	CT	127.2	3.005	1.500	1.589	1.00	80.60		1987	DA007	
		1.75	3.007	1.500	CT	127.2	3.007	1.500	1.577	1.00	80.30	82.6	5.3	1987	DA007
		1.50	3.008	1.501	CT	131.6	3.008	1.501	1.637	0.93	80.20		1987	DA006	
ANNEALED	Extrusion	1.50	3.007	1.502	CT	131.6	3.007	1.502	1.437	0.92	79.90		1987	DA006	
		4.00	4.003	1.629	CT	122.0	4.003	1.629	2.027	1.51	94.90		1973	85836 (1)	
		4.00	4.005	1.625	CT	122.0	4.005	1.625	1.995	1.41	91.70		1973	85836 (1)	
		1.75	3.005	1.500	CT	127.9	3.005	1.500	1.571	1.06	83.40		1987	DA007	
ANNEALED	Extrusion	1.75	3.009	1.498	CT	127.9	3.009	1.498	1.627	1.02	81.80	85.2	6.5	1987	DA007
		1.50	2.007	1.502	CT	131.4	2.007	1.502	1.439	0.91	79.20		1987	DA006	
		1.50	3.008	1.493	CT	131.4	3.008	1.493	1.465	0.83	80.30		1987	DA006	
		6.00	2.501	1.250	CT	123.0	2.501	1.250	1.249	0.87	72.80		1975	MA003	
ANNEALED 1000F 2 HR AC	Billet	6.00	2.500	1.251	CT	123.0	2.500	1.251	1.241	1.23	86.40	79.6	9.6	1975	MA003
		2.30	2.498	1.250	CT	145.0	2.498	1.250	1.264	0.30	50.50		1971	84360	
		2.30	2.500	1.249	CT	145.0	2.500	1.249	1.250	0.31	51.30	50.9	0.6	1971	84360

NOTES: (1) COMPOSITION (WT PERCENT) 6.35Al, 4.31V, 0.22C, 0.16Fe, 0.09N, 0.06H, 0.16O

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TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	2.5° $(K_{Ic}/TBS)^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (KSI $\sqrt{in.}$)	K_{Ic} MEAN	STAN DEV		
ANNEALED 1800F 4 HR AC	Forging	2.30	-65	L-T	145.0	1.500	0.750	CT	---	0.28	48.40	56.8	11.8	1974	89504
		2.30													
ANNEALED 1300F 4 HR AC	Forging	2.30	-65	T-L	151.0	1.500	0.750	CT	---	0.28	50.40	67.8	11.4	1974	89504
		2.30													
		2.30													
ANNEALED 1300F 4 HR AC	Forging	2.30	-65	S-T	146.0	1.500	0.750	CT	---	0.50	65.00	56.9	8.0	1974	89504
		2.30													
		2.30													
ANNEALED 1300F 4 HR AC	Forging	2.30	-65	S-L	146.0	1.500	0.750	CT	---	0.38	56.60	57.7	2.4	1974	89504
		2.30													
		2.30													
ANNEALED 1300F 4 HR AC	Forging	2.30	R.T.	L-T	129.0	1.500	0.750	CT	---	0.49	57.20	58.1	1.2	1974	89504
		2.30													
		2.30													
ANNEALED 1300F 4 HR AC	Forging	2.30	R.T.	T-L	132.0	1.500	0.750	CT	---	0.61	65.20	62.2	3.0	1974	89504
		2.30													
		2.30													
ANNEALED 1300F 4 HR AC	Forging	2.30	R.T.	S-T	128.0	1.500	0.750	CT	---	0.49	56.90	56.1	1.1	1974	89504
		2.30													
		2.30													

TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}														
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	2.5° $(K_{Ic}/TYS)^2$ (in.)	K_{Ic} (ksi $\sqrt{in.}$)	K_{Ic}		REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN				K_{Ic} MEAN	STAN DEV	
ANNEALED 1300F 4 HR AC	Forging	2.30	R.T.	S-L	128.0	1.500	0.750	CT	---	0.69	67.40	68.1	1.0	89504
		2.30				1.500	0.750	CT	---	0.72	69.80		89504	
ANNEALED 1375F 3 HR AC	Plate	2.75	R.T.	L-T	129.0	2.499	1.250	CT	1.224	0.48	56.59	60.4	5.5	MA003
		2.75				2.497	1.250	CT	1.200	0.62	64.30		MA003	
ANNEALED 2200F 2 HR	Forging	1.00	R.T.	---	140.0	2.000	1.000	CT	1.000	0.50	62.80	57.5	3.9	89862 (1)
		1.00				2.000	1.000	CT	1.000	0.43	58.20		89862 (1)	
		1.00				2.000	1.000	CT	1.000	0.39	55.20		89862 (1)	
		1.00				2.000	1.000	CT	1.000	0.44	59.80		89862 (1)	
		1.00				2.000	1.000	CT	1.000	0.35	52.40		89862 (1)	
		---				2.000	1.002	CT	1.020	0.61	61.10		90012	
AS RECEIVED	Forged Bar	---	R.T.	L-T	128.0	2.006	1.000	CT	1.074	0.59	61.30	57.1	10.4	90012
		3.50				1.998	1.001	CT	1.019	0.62	62.50		90012	
		1.90				1.997	0.625	CT	0.994	0.59	61.30		90012	
		3.50				1.996	1.002	CT	1.006	0.48	56.60		90012	
		---				1.998	1.000	CT	1.029	0.79	72.90		90012	
		---				2.000	1.000	CT	1.053	0.53	60.10		90012	
		3.50				1.998	0.998	CT	1.036	0.72	70.60		90012	
		1.30				2.000	0.563	CT	1.005	0.32	47.70		90012	
		---				1.998	1.003	CT	1.013	0.61	66.10		90012	
		1.50				1.998	1.003	CT	1.063	0.20	38.40		90012	
		2.50				2.000	0.999	CT	1.032	0.37	52.50		90012	
		2.60				2.001	1.000	CT	1.021	0.24	42.60		90012	
2.70	2.001	0.999	CT	1.022	0.26	45.70		90012						

NOTES: (1) INTERMEDIATE GRAIN SIZE

Ti-6Al-4V

TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}																
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	σ_{TS} (K _{TS}) ^a (in.)	K_{Ic}					
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (K _{TS}) ^a (in.)	K_{Ic} MEAN	STAN DEV	DATE	REFER	
AS RECEIVED	Forged Bar	3.50	3.50	R.T.	T-L	127.0	1.998	0.998	CT	1.064	0.68	66.40	54.9	10.8	1974	90012
		---	---			128.0	1.998	0.752	CT	1.023	0.42	52.50			1974	90012
		1.80	1.80			129.0	1.997	0.779	CT	1.018	0.66	66.50			1974	90012
		1.50	1.50			130.0	1.998	0.502	CT	1.000	0.41	52.60			1974	90012
		---	---			130.0	1.999	0.645	CT	0.998	0.43	54.00			1974	90012
		2.00	2.00			131.0	1.998	1.003	CT	1.023	0.53	60.10			1974	90012
		1.50	1.50			132.0	1.999	1.002	CT	1.055	0.16	33.80			1974	90012
		---	---			133.0	2.000	1.002	CT	1.027	0.69	69.90			1974	90012
		---	---			133.0	1.998	1.001	CT	1.024	0.50	59.40			1974	90012
		3.50	3.50			133.0	1.998	1.005	CT	1.045	0.56	62.70			1974	90012
		3.50	3.50			134.0	2.000	1.000	CT	1.028	0.58	64.50			1974	90012
		1.50	1.50			135.0	1.996	0.560	CT	0.994	0.45	57.40			1974	90012
		---	---			135.0	2.003	1.001	CT	1.023	0.46	57.80			1974	90012
		---	---			139.0	2.001	1.002	CT	1.016	0.44	56.20			1974	90012
		---	---			139.0	2.000	0.654	CT	1.010	0.37	53.50			1974	90012
		---	---			139.0	1.998	1.003	CT	1.033	0.35	52.10			1974	90012
		2.80	2.80			140.0	2.002	0.999	CT	1.014	0.14	32.80			1974	90012
1.80	1.80	141.0	2.002	1.001	CT	1.033	0.18	37.40	1974	90012						
1.80	1.80	142.0	1.999	0.999	CT	1.048	0.36	53.70	1974	90012						
1.00	1.00	142.0	1.993	1.001	CT	1.054	0.55	66.80	1974	90012						
2.50	2.50	145.0	2.000	0.999	CT	1.028	0.19	40.10	1974	90012						

TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	$2.8 \cdot (K_{Ic}/\sqrt{TS})^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (KSI) $(\sqrt{in.})$	K_{Ic} MEAN	STAN DEV		
AS RECEIVED-AB (ALPHA-BETA FORGED)	Forged Bar	3.50	R.T.	L-T	127.0	2.000	1.000	CT	1.023	0.82	77.10	--	1974	90012	
		3.50													
AS RECEIVED-AB (ALPHA-BETA FORGED)	Forged Bar	2.25	R.T.	T-L	135.0	2.000	1.000	CT	1.046	0.90	46.80	--	1973	86688	
		2.25													
		2.25													
		2.25													
B FORGED BETA FORGED REHEATED TO 1950F DRAWN TO SIZE ANNEALED 1300F	Forged Bar	2.25	R.T.	T-L	135.0	2.000	1.000	CT	1.088	0.28	45.60	42.6	1973	86688	
		2.25													
		2.25													
		2.25													
B FORGED-MA BETA FORGED MILL ANNEALED 1300F 2 HR AC	Forging	2.75	R.T.	...	131.0	2.000	1.000	CT	1.000	0.88	77.70	75.3	1974	88962 (1)	
		2.75													
		2.00													
		2.00													
B FORGED-MA BETA FORGED MILL ANNEALED 1300F 2 HR AC	Forging	2.00	R.T.	L-T	137.0	2.000	1.000	CT	...	0.69	66.40	70.6	1971	80538	
		2.00													
		2.00													
		2.00													
B FORGED-MA BETA FORGED MILL ANNEALED 1300F 2 HR AC	Forging	2.00	R.T.	T-L	137.0	2.000	1.000	CT	...	0.77	76.00	71.0	1971	80538	
		2.00													
		2.00													
		2.00													
B FORGED-MA BETA FORGED MILL ANNEALED 1300F 2 HR AC	Forging	2.00	R.T.	T-L	131.0	2.000	1.000	CT	...	0.84	69.30	71.0	1971	80538	
		2.00													
		2.00													
		2.00													
B FORGED-MA BETA FORGED MILL ANNEALED 1300F 2 HR AC	Forging	2.00	R.T.	S-L	132.0	2.000	1.000	CT	...	0.74	72.10	73.9	1971	80538	
		2.00													
		2.00													
		2.00													

NOTES: (1) COMP. DISK

TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic}/TTS)^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (KSI) $\sqrt{(in.)}$	K_{Ic} MEAN	STAN DEV		
BB AB FIN-SOMA BETA BLOCKED ALPHA-BETA FINISHED 30% REDUCTION MILL ANNEALED 1300F 2 HR AC	Forging	2.50	R.T.	--	127.0	2.000	1.000	CT	1.000	1.00	80.50	78.2	3.3	1974	88962
		2.50				2.000	1.000	CT	1.000	0.89	75.90			88962	
BB AB FIN-MA BETA BLOCKED ALPHA-BETA FINISHED MILL ANNEALED	Forging	--	R.T.	--	132.0	2.500	1.250	CT	1.250	0.78	73.50	72.3	5.5	1973	90584 (1)
		--				2.500	1.250	CT	1.250	0.89	76.70			90584 (1)	
		--				2.500	1.250	CT	1.250	0.89	65.10			90584 (1)	
		--				2.500	1.250	CT	1.250	0.72	72.10			90584 (1)	
BB AB FIN-RA BETA BLOCKED ALPHA-BETA FINISHED RECRYSTALLIZED ANNEAL 1700F 4HR FC TO 1000F AC	Forging	--	R.T.	--	135.0	2.500	1.250	CT	1.250	0.89	80.70	75.1	4.3	1973	90584 (1)
		--				2.500	1.250	CT	1.250	0.76	74.40			90584 (1)	
		--				2.500	1.250	CT	1.250	0.74	74.80			90584 (1)	
		--				2.500	1.250	CT	1.250	0.66	70.30			90584 (1)	
BB AB FIN10STO BETA-BLOCKED ALPHA-BETA FINISHED 10% REDUCTION SOLUTION TREATED & OVERAGED 1750F 1HR WQ 1300F 2HR AC	Forging	2.50	R.T.	--	136.0	2.000	1.000	CT	1.000	0.84	79.00	81.7	3.4	1974	88962
		2.50				2.000	1.000	CT	1.000	0.89	85.50			88962	
		2.50				2.000	1.000	CT	1.000	0.88	80.50			88962	
BB AB FIN30STO BETA-BLOCKED ALPHA-BETA FINISHED 30% REDUCTION SOLUTION TREATED & OVERAGED 1750F 1 HR WQ 1300F 2HR AC	Forging	2.50	R.T.	--	137.0	2.000	1.000	CT	1.000	0.81	78.00	73.1	5.3	1974	88962
		2.50				2.000	1.000	CT	1.000	0.60	67.40			88962	
		2.50				2.000	1.000	CT	1.000	0.72	73.80			88962	

NOTES: (1) F-14 OUTBOARD COVER

TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	$2.5^\circ (K_{Ic}/\sqrt{YS})^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (KSI) $\cdot \sqrt{(in.)}$	K_{Ic} MEAN	STAN DEV		
BB B FIN-10MA BETA BLOCKED BETA FINISHED 10% REDUCTION MILL ANNEALED 1300F 2HR AC	Forging	2.50	R.T.	---	128.0	2.000	1.000	CT	1.000	0.85	74.50	73.3	1.9	1974	88962
		2.50									74.40				
		2.50									71.10				
BB B FIN10STOA BETA BLOCKED BETA FINISHED 10% REDUCTION SOLUTION TREATED & OVERAGED 1750F 1HR WQ 1300F 2HR AC	Forging	2.50	R.T.	---	138.0	2.000	1.000	CT	1.000	0.86	85.60	86.1	0.5	1974	88962
		2.50									86.40				
		2.50									86.40				
BETA ANNEALED	Plate	2.50	-65	L-T	125.0	4.003	1.991	CT	2.116	2.06	113.50	108.4	2.8	1974	88575
		2.50				2.000	2.162		110.50						
		2.50				3.997	2.072		109.10						
		2.50				4.003	1.998		106.90						
		2.50				4.009	2.005		108.00						
		1.00				2.550	1.000		WOL-CT EQ.		79.20				
BETA ANNEALED PLATE EB WELDED THEN BETA ANNEALED, EB WELD IN PLANE OF FRACTURE	Plate	2.50	R.T.	---	116.0	1.994	0.999	CT	1.133	0.90	69.80	69.2	0.9	1974	88575
		2.50				1.992	0.981		1.140		68.50				
		3.00				3.000	1.500		---		99.00				
BETA PROCESSED MILL ANNEALED	Plate	3.00	R.T.	L-T	130.4	3.000	1.500	CT	---	1.35	96.00	94.9	4.8	1975	UD008
		3.00				3.000	1.500		---		96.00				
		3.00				3.000	1.500		---		89.60				

TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}																
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic}/T)^2$ (in.)	K_{Ic}			DATE	REFER	
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (KSI $\sqrt{\text{in.}}$)	K_{Ic} MEAN	STAN DEV			
DIFFUSION BOND ANNEALED - 1750F 1.5-4HR FC TO 900F AT 100F/HR	Billet	3.50	R.T.	L-T	127.0	2.000	1.000	CT	1.072	0.93	77.50	68.2	9.7	1974	90012	
		1.00				2.001	1.002	CT	1.092	0.98	80.70			1974	90012	
		---				2.002	1.004	CT	1.032	0.75	70.80			1974	90012	
	Billet	1.50	R.T.	L-T	132.0	2.000	1.003	CT	1.019	0.45	56.10	68.2	9.7	1974	90012	
		---				1.995	1.000	CT	1.083	0.63	76.30			1974	90012	
		3.50				1.999	1.003	CT	1.029	0.68	69.00			1974	90012	
	Billet	2.60	R.T.	L-T	133.0	1.996	0.975	CT	1.017	0.48	58.10	68.2	9.7	1974	90012	
		3.50				1.999	1.000	CT	1.030	0.69	70.40			1974	90012	
		2.70				2.000	1.001	CT	1.024	0.40	54.70			1974	90012	
	DIFFUSION BOND ANNEALED - 1750F 1.5-4HR FC TO 900F AT 100F/HR	Billet	---	R.T.	T-L	125.0	1.997	0.676	CT	0.992	0.65	64.00	64.2	11.8	1974	90012
			---				2.000	1.003	CT	1.032	0.80	70.80			1974	90012
			---				2.001	0.999	CT	1.043	0.90	76.90			1974	90012
		Billet	3.50	R.T.	T-L	131.0	1.995	1.000	CT	1.048	0.84	76.00	64.2	11.8	1974	90012
			3.50				1.998	0.999	CT	1.013	0.67	68.20			1974	90012
			2.00				1.998	1.000	CT	1.028	0.75	72.70			1974	90012
Billet		3.50	R.T.	T-L	133.0	2.000	0.999	CT	1.019	0.72	71.40	64.2	11.8	1974	90012	
		1.50				2.000	1.000	CT	1.043	0.39	53.10			1974	90012	
		---				2.000	1.001	CT	1.033	0.66	69.20			1974	90012	
Billet		2.50	R.T.	T-L	137.0	1.998	0.877	CT	1.021	0.33	49.50	64.2	11.8	1974	90012	
		2.60				1.997	0.979	CT	1.031	0.27	45.10			1974	90012	
		2.70				2.000	1.004	CT	1.018	0.25	44.60			1974	90012	
Billet		1.00	R.T.	T-L	142.0	2.001	1.000	CT	1.042	0.66	72.70	64.2	11.8	1974	90012	

TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic}/\sqrt{S})^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (KSI $\cdot \sqrt{\text{in.}}$)	K_{Ic} MEAN	STAN DEV		
DIFFUSION BONDED - 1700F 4HR FC TO 1400F AT 100F/HR THEN TO 900F IN 0.75HR	Plate	0.62	R.T.	---	120.0	4.005	2.002	CT	1.966	1.47	92.10	---	---	1973	85836
DIFFUSION BONDED - 1700F 4HR FC TO 1400F AT 100F/HR THEN TO 900F IN 0.75HR	Plate	0.62	R.T.	L-T	120.0	3.004	1.500	CT	1.574	1.11	79.90	---	---	1973	85836
DIFFUSION BONDED - 1700F 4HR FC TO 1400F AT 100F/HR THEN TO 900F IN 0.75HR	Plate	0.62	R.T.	T-L	120.0	3.005	1.501	CT	1.575	1.20	83.10	---	---	1973	85836
DIFFUSION BONDED - 1700F 4HR FC TO 1400F AT 100F/HR THEN TO 900F IN 0.75HR	Plate	1.50			120.0	4.000	1.500	CT	---	1.34	88.00	---	---	1974	89004
DIFFUSION BONDED - 1700F 4HR FC TO 1400F AT 100F/HR THEN TO 900F IN 0.75HR	Plate	1.50	R.T.	S-T	120.0	4.000	1.500	CT	---	1.16	82.00	85.7	3.2	1974	89004
DIFFUSION BONDED - 1700F 4HR FC TO 1400F AT 100F/HR THEN TO 900F IN 0.75HR	Plate	1.50			120.0	4.000	1.500	CT	---	1.31	87.00	---	---	1974	89004
DIFFUSION BONDED - 1700F 4HR FC TO 1400F AT 100F/HR THEN TO 900F IN 0.75HR	Plate	2.50	R.T.	S-L	120.0	4.000	1.500	CT	---	1.14	81.00	---	---	1974	89004
DIFFUSION BONDED - 1700F 4HR FC TO 1400F AT 100F/HR THEN TO 900F IN 0.75HR	Billet	0.62	R.T.	L-T	120.0	3.000	1.500	CT	---	1.11	80.00	---	---	1974	89004
DIFFUSION BONDED - 1700F 4HR FC TO 1400F AT 100F/HR THEN TO 900F IN 0.75HR	Billet	0.62	R.T.	T-L	120.0	3.000	1.500	CT	---	1.20	83.00	---	---	1974	89004

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TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	2.5° (K _{1c} /YIS) ¹ (in.)	K _{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K _{Ic} (Ksi) • √(in.)	K _{Ic} MEAN	STAN DEV		
DIFFUSION BONDED - 1700F 4HR FC TO 1400F AT 100F/HR THEN TO 900F IN 0.75HR	Billet	8.00	3.995	1.495	CT	1.926	1.32	87.40	86.0	3.6		1972	84306		
		8.00	3.999	1.457	CT	1.881	1.19	82.40							
		8.00	3.998	1.495	CT	1.920	1.83	87.80							
		8.00	3.998	1.495	CT	1.963	1.43	90.90							
		8.00	3.998	1.492	CT	1.969	1.45	91.60							
MA 10-20%ALPHA 10 TO 20% PRIMARY ALPHA MILL ANNEALED 1300F 2 HR AC	Forging	2.50	2.000	1.000	CT	1.000	0.67	73.10	69.4	3.2		1974	88962		
		2.50	2.000	1.000	CT	1.000	0.58	68.00							
		2.50	2.000	1.000	CT	1.000	0.57	67.10							
MA 40-50%ALPHA 40 TO 50% PRIMARY ALPHA MILL ANNEALED 1300F 2 HR AC	Forging	2.50	2.000	1.000	CT	1.000	0.70	71.50	69.1	3.4		1974	88962		
		2.50	2.000	1.000	CT	1.000	0.68	70.60							
		2.50	2.000	1.000	CT	1.000	0.58	65.20							
MA COARSE GRAIN 1300F 2 HR AC	Forging	14.00	2.000	1.000	CT	1.000	0.40	55.30	53.9	2.0		1974	88962		
		14.00	2.000	1.000	CT	1.000	0.36	53.20							
		14.00	2.000	1.000	CT	1.000	0.35	52.20							
		14.00	2.000	1.000	CT	1.000	0.42	56.70							
		14.00	2.000	1.000	CT	1.000	0.35	52.10							
MA FINE GRAIN 1300F 2 HR AC	Forging	6.00	2.000	1.000	CT	1.000	0.85	80.60	69.5	7.7		1974	88962		
		6.00	2.000	1.000	CT	1.000	0.76	76.10							
		6.00	2.000	1.000	CT	1.000	0.46	59.60							
		6.00	2.000	1.000	CT	1.000	0.62	69.10							
		6.00	2.000	1.000	CT	1.000	0.60	67.60							
6.00	2.000	1.000	CT	1.000	0.54	64.00									

TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}																		
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	$\frac{2.5 \cdot (K_{Ic} \sqrt{TS})^2}{(in.)}$	K_{Ic}			DATE	REFER			
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Ksi) $\sqrt{(in.)}$	K_{Ic} MEAN	STAN DEV					
MILL ANNEAL	Plate	1.50	R.T.	L-T	133.0	2.496	1.244	CT	1.930	0.45	56.59	55.6	1.3	1981	MA002			
		1.50				2.502	1.255				CT					1.297	0.42	54.70
MILL ANNEAL	Plate	1.00	R.T.	T-L	146.1	2.550	1.000	WOL-CT EQ.	--	0.17	38.20	--	--	1977	JEM01			
		2.00				4.000	2.004				CT	1.970	1.03			89.70		
MILL ANNEALED	Plate	2.00	-85	T-L	157.0	4.000	2.002	CT	1.980	0.94	85.70	87.9	2.0	1973	88144			
		2.00				4.000	2.003				CT					1.990	0.99	88.20
		2.00				4.000	2.003				CT					2.137	1.94	112.00
MILL ANNEALED	Plate	2.00	R.T.	L-T	127.0	4.000	2.003	CT	2.137	1.94	112.00	--	--	1972	85084			
		1.25				3.495	1.245				CT	1.741	1.16			95.30		
MILL ANNEALED	Plate	1.25	R.T.	T-L	119.4	3.499	1.245	CT	1.824	1.20	97.00	100.6	6.8	1972	84906			
		1.25				3.500	1.247				CT					1.817	1.15	94.80
		2.00				4.000	2.002				CT					2.038	1.53	98.70
		2.00				4.000	2.001				CT					2.097	1.94	112.00
		2.00				4.000	2.002				CT					2.048	1.75	105.50
MILL ANNEALED	Extrusion	4.00	R.T.	L-T	123.5	3.997	1.465	CT	1.957	1.21	87.60	83.5	3.1	1972	84306 (1)			
		1.80				3.934	1.578				CT					2.031	1.17	85.09
		1.80				3.930	1.578				CT					2.066	1.12	83.06
		1.80				3.995	1.577				CT					2.049	1.09	82.37
		1.80				3.995	1.578				CT					2.115	1.01	79.17

NOTES: (1) COMPOSITION (WT PERCENT) 6.51Al, 4.86V, 0.024C, 0.19Fe, 0.012V, 0.068H, 0.15O

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TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (ksi)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic}/YS)^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (ksi) $\sqrt{\text{in.}}$	K_{Ic} MEAN	STAN DEV		
MILL ANNEALED	Extrusion	4.00	R.T.	T-L	125.5	4.000	1.498	CT	1.947	1.30	92.50	87.5	4.1	1972	84906 (1)
		4.00												1972	84906 (1)
		1.80												1976	NC001
		1.80												1976	NC001
		1.80												1976	NC001
		1.80												1976	NC001
MILL ANNEALED 1300F 2 HR AC	Forging	2.00	R.T.	L-T	132.0	2.000	1.000	CT	---	0.30	45.40	47.7	2.9	1971	80538
		2.00												1971	80538
		2.00												1971	80538
		2.00												1971	80538
		2.00												1971	80538
		2.00												1971	80538
MILL ANNEALED 1300F 2 HR AC	Forging	2.00	R.T.	T-L	133.0	2.000	1.000	CT	---	0.30	46.50	49.5	3.9	1971	80538
		2.00												1971	80538
		2.00												1971	80538
		2.00												1971	80538
		2.00												1971	80538
		2.00												1971	80538
MILL ANNEALED 1300F 2 HR AC	Forging	2.00	R.T.	S-L	140.0	2.000	1.000	CT	---	0.31	49.50	43.8	5.8	1971	80538
		2.00												1971	80538
		2.00												1971	80538
		2.00												1971	80538
		2.00												1971	80538
		2.00												1971	80538
MILL ANNEALED 1300F 2 HR AC	Billet	2.30	R.T.	L-T	120.0	2.500	1.250	CT	1.277	1.13	80.30	84.0	3.4	1971	84960
		2.30												1971	84960
		2.30												1971	84960
		2.30												1971	84960
		2.30												1971	84960
		2.30												1971	84960
MILL ANNEALED 1300F 2 HR AC	Forging	4.50	R.T.	L-T	130.0	2.498	1.248	CT	1.355	0.54	60.59	---	---	1981	MA002
		1.50												1972	84906 (2)
RECRYSTALLIZE ANNEAL	Plate	1.50	-65	L-T	---	3.999	1.502	CT	2.044	0.69	60.40	---	---	1972	84906 (2)
RECRYSTALLIZE ANNEAL	Plate	1.50	-65	T-L	---	3.997	1.498	CT	2.110	1.03	77.00	---	---	1972	84906 (2)

NOTES: (1) COMPOSITION (WT PERCENT) 6.35AL, 4.31V, 0.22C, 0.16Fe, 0.09N, 0.06H, 0.16O
(2) 1700F 4HR FC TO 1400F AT 100F/HR COOL TO 900F IN .75HR

TABLE 6.16.2.1 (CONTINUED)

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TITANIUM TI-6Al-4V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic}/TYS)^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} MEAN	K_{Ic} STAN DEV			
RECRYSTALLIZE ANNEAL - 1700F 4HR FC TO 1400F AT 100F/HR COOL TO 900F IN .75HR	Plate	1.50	1.50	R.T.	L-T	118.0	4.005	1.502	CT	2.004	1.06	79.00	84306	1972	84306
		1.50	1.50			118.0	4.003	1.502	CT	2.021	0.97	74.90	84306	1972	84306
		2.00	2.00			119.0	6.002	1.780	CT	3.075	1.64	97.30	84306	1972	84306
		2.00	2.00			119.0	6.003	1.873	CT	3.138	1.62	96.60	84306	1972	84306
		2.50	2.50			120.0	4.000	2.000	CT	---	0.95	74.00	89004	1974	89004
		1.50	1.50			120.0	4.000	1.501	CT	2.103	1.16	81.60	85336	1973	85336
		1.25	1.25			120.0	---	1.250	CT	---	1.08	79.00	89004	1974	89004
		2.50	2.50			120.0	2.999	1.126	CT	1.526	0.83	69.00	85336	1973	85336
		1.25	1.25			120.0	---	1.250	CT	---	1.03	77.00	89004	1974	89004
		2.50	2.50			120.0	---	---	CT	---	1.38	89.00	89004	1974	89004
		1.25	1.25			120.0	---	1.250	CT	---	0.98	75.00	89004	1974	89004
		2.50	2.50			120.0	---	---	CT	---	1.31	87.00	89004	1974	89004
		1.25	1.25			120.0	---	1.250	CT	---	0.92	73.00	89004	1974	89004
		2.50	2.50			120.0	4.000	2.000	CT	---	1.17	82.00	89004	1974	89004
		1.25	1.25			120.0	---	1.250	CT	---	1.14	81.00	89004	1974	89004
		1.50	1.50			120.0	6.000	1.370	CT	---	1.11	80.00	89004	1974	89004
		2.50	2.50			120.0	4.000	2.000	CT	---	1.11	80.00	89004	1974	89004
		1.50	1.50			121.0	6.002	1.496	CT	3.080	1.44	91.00	84306	1972	84306
		1.50	1.50			121.0	3.500	1.500	CT	---	1.44	92.00	89004	1974	89004
		1.50	1.50			121.0	6.003	1.500	CT	3.103	1.30	86.60	84306	1972	84306
1.50	1.50	121.0	6.001	1.497	CT	3.065	1.41	90.20	84306	1972	84306				
1.50	1.50	121.0	3.500	1.500	CT	---	1.29	87.00	89004	1974	89004				

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TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	2.5° (σ_{TS} , TYS) (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (KSI) $\sqrt{(in.)}$	K_{Ic} MEAN	STAN DEV		
RECRYSTALLIZE ANNEAL = 1700F 4HR FC TO 1400F AT 100F/HR COOL TO 900F IN .75HR	Plate	2.50	3.000	1.108	CT	1.520	0.83	69.30	85836	1973	85836				
		1.25	--	1.250	CT	--	0.92	73.00	89004	1974	89004				
		2.50	4.000	2.000	CT	--	1.25	85.00	89004	1974	89004				
		1.25	--	1.250	CT	--	1.00	76.00	89004	1974	89004				
		2.50	4.000	2.000	CT	--	0.83	69.00	89004	1974	89004				
		2.50	--	--	CT	--	1.50	93.00	89004	1974	89004				
		2.00	6.001	2.069	CT	3.048	1.97	106.40	85836	1973	85836				
		1.25	--	1.250	CT	--	0.95	74.00	89004	1974	89004				
		1.50	6.000	1.370	CT	--	1.28	86.00	89004	1974	89004				
		1.50	4.001	1.498	CT	2.191	1.45	91.50	85836	1973	85836				
		1.25	--	1.250	CT	--	1.14	81.00	89004	1974	89004				
		2.50	4.000	2.000	CT	--	0.96	75.00	89004	1974	89004				
		1.50	3.000	1.500	CT	--	1.25	85.00	89004	1974	89004				
		2.50	--	--	CT	--	1.44	91.00	89004	1974	89004				
		2.50	3.000	1.127	CT	1.544	0.90	71.90	85836	1973	85836				
		2.00	5.999	1.997	CT	3.056	1.79	101.50	85836	1973	85836				
		2.50	2.999	1.374	CT	1.534	0.84	69.40	85836	1973	85836				
		2.50	--	--	CT	--	1.14	81.00	89004	1974	89004				
		1.25	--	1.250	CT	--	0.85	70.00	89004	1974	89004				
		2.50	4.000	2.000	CT	--	1.00	76.00	89004	1974	89004				
1.50	3.500	1.500	CT	--	1.18	83.00	89004	1974	89004						
1.00	2.550	1.000	WOL-CT EQ.	--	0.65	69.20	JEM01(1)	1977	JEM01(1)						

NOTES: (1) TYS APPROX. 120

TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic}/TYS)^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Ksi $\cdot\sqrt{in.}$)	K_{Ic} MEAN	STAN DEV		
RECRYSTALLIZE ANNEAL = 1700F 4HR FC TO 1400F AT 100F/HR COOL TO 900F IN 75HR					--	3.000	1.500	CT	--	1.03	77.00		1974	89004	
					--	4.000	2.020	CT	--	1.20	83.00		1974	89004	
					--	3.000	1.500	CT	--	0.92	73.00		1974	89004	
					--	4.000	1.790	CT	--	1.06	78.00		1974	89004	
					114.0	2.997	1.499	CT	1.528	1.43	86.10		1973	85034	
					115.0	2.999	1.500	CT	1.524	1.49	89.00		1973	85034	
					116.0	2.998	1.499	CT	1.535	1.30	83.60		1973	85034	
					117.0	2.999	1.503	CT	1.554	1.47	89.70		1973	85034	
					117.0	3.002	1.500	CT	1.539	1.05	76.00		1973	85034	
				R.T.	L-T	118.0	3.004	1.499	CT	1.540	1.46	90.30		1973	85034
						118.0	2.999	1.499	CT	1.526	1.19	81.30		1973	85034
						119.0	3.008	1.497	CT	1.536	1.46	90.90		1973	85034
						119.0	3.007	1.498	CT	1.563	1.49	91.90		1973	85867
						119.0	3.005	1.500	CT	1.552	1.36	87.80		1973	85867
						119.0	3.002	1.501	CT	1.533	1.39	88.90		1973	85034
						120.0	3.006	1.505	CT	1.561	1.41	90.10		1973	85867
					121.0	3.000	1.502	CT	1.541	0.96	75.10		1973	85034	
					121.0	3.004	1.376	CT	1.615	1.22	84.40		1973	85867	
					121.0	3.005	1.374	CT	1.588	1.27	86.10		1973	85867	

Ti-6Al-4V

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TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}																				
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	2.5° (K _{IC}) _{TS} ' (in.)	K_{Ic}			DATE	REFER					
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Ksi) (in.) √in.	K_{Ic} MEAN	STAN DEV							
RECRYSTALLIZE ANNEAL = 1700F 4HR FC TO 1400F AT 100F/HR COOL TO 900F IN .75HR Cont'd	Forging Cont'd	1.20	121.0	L-T Cont'd	R.T. Cont'd	3.010	1.345	CT	1.556	1.15	82.10	Cont'd	1973	86867						
		6.70	121.0												1.500	CT	1.548	1.08	78.70	
		2.20	121.0												3.003	1.502	CT	1.589	1.17	82.80
		2.20	121.0												3.002	1.504	CT	1.555	1.24	85.10
		2.20	122.0												3.003	1.499	CT	1.633	1.18	83.90
		1.20	122.0												3.009	1.318	CT	1.583	1.12	81.70
		1.50	123.0												2.998	1.417	CT	1.593	0.95	76.70
		1.20	123.0												3.006	1.377	CT	1.605	1.17	84.30
		1.50	123.0												3.002	1.377	CT	1.578	0.80	73.20
		2.20	124.0												3.003	1.501	CT	1.607	1.16	84.40
		1.50	124.0												3.004	1.409	CT	1.577	0.94	76.00
		3.20	124.0												3.001	1.504	CT	1.583	1.49	95.70
		1.50	124.0												2.999	1.397	CT	1.585	0.98	77.80
		2.50	124.0												3.005	1.504	CT	1.549	1.06	81.50
		1.20	125.0												3.001	1.375	CT	1.612	1.14	84.60
		2.20	126.0												3.003	1.495	CT	1.591	1.11	83.80
		1.50	127.0												3.001	1.248	CT	1.542	1.17	86.70
		1.50	127.0												3.003	1.250	CT	1.527	1.24	89.50
		1.50	127.0												3.001	1.250	CT	1.583	1.20	88.00
		1.50	128.0												2.999	1.248	CT	1.598	1.07	84.40
1.50	129.0	2.999	1.249	CT	1.440	1.08	84.80													
1.50	130.0	3.002	1.372	CT	1.560	0.92	78.80													

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TABLE 6.16.2.1 (CONTINUED)

TITANIUM Ti-6Al-4V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	2.5° $(K_{Ic}/\sqrt{S})^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Ksi) $\sqrt{(in.)}$	K_{Ic} MEAN	STAN DEV		
RECRYSTALLIZE ANNEAL - 1700F 4HR FC TO 1400F AT 100F/HR COOL TO 800F IN .75HR Cont'd	Forging Cont'd	4.75	125.0		1.502	3.001	1.502	CT	1.541	1.24	87.90		1973	85034	
		2.20	125.0		1.498	3.006	1.598	CT	1.593	1.03	80.30		1973	85857	
		3.20	126.0		1.508	3.000	1.589	CT	1.589	1.30	90.70		1973	85857	
		2.25	126.0		1.500	2.998	1.516	CT	1.516	1.02	80.50		1973	85034	
		2.25	126.0		1.502	2.995	1.595	CT	1.595	1.31	91.30		1973	85034	
		3.20	126.0		1.501	3.000	1.521	CT	1.521	1.47	96.50		1973	85857	
		2.20	127.0		1.502	3.008	1.783	CT	1.783	1.38	94.40		1973	85857	
		2.20	127.0		1.500	3.002	1.577	CT	1.577	1.03	81.40		1973	85857	
		2.25	127.0		1.500	3.001	1.552	CT	1.552	1.34	93.10		1973	85034	
		2.20	128.0		1.500	3.006	1.601	CT	1.601	0.88	76.90		1973	85857	
		1.50	128.0	R.T. Cont'd	1.398	3.002	1.595	CT	1.595	0.97	79.70		1973	85034	
		1.50	129.0	T-L Cont'd	1.424	3.003	1.583	CT	1.583	0.83	74.20	Cont'd	1973	85034	
		1.50	129.0		1.377	3.001	1.576	CT	1.576	0.85	75.40		1973	85034	
		1.50	129.0		1.387	3.000	1.586	CT	1.586	1.02	82.20		1973	85034	
		1.50	129.0		1.355	2.999	1.596	CT	1.596	1.11	85.90		1973	85034	
		1.50	130.0		1.388	3.001	1.564	CT	1.564	1.16	86.60		1973	85034	
		1.20	130.0		1.347	3.004	1.587	CT	1.587	1.26	92.50		1973	85857	
		1.50	130.0		1.389	3.004	1.587	CT	1.587	0.99	81.70		1973	85034	
		1.20	130.0		1.344	3.008	1.653	CT	1.653	1.27	92.60		1973	85857	
		2.25	130.0		1.500	3.000	1.522	CT	1.522	1.32	94.50		1973	85034	
1.20	130.0		1.319	3.010	1.649	CT	1.649	1.22	90.80		1973	85857			
2.20	131.0		1.497	3.006	1.592	CT	1.592	0.95	80.60		1973	85857			

TABLE 6.16.2.1 (CONCLUDED)

TITANIUM Ti-6Al-4V K_{Ic}																
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (ksi)	SPECIMEN			CRACK LENGTH (in.) A	S.S. * (K_{Ic}/TYS) ¹ (in.)	K_{Ic}			DATE	REFER	
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (ksi $\sqrt{in.}$)	K_{Ic} MEAN	STAN DEV			
RECRYSTALLIZE ANNEAL - 1700F 4HR FC TO 1400F AT 100F/HR COOL TO 900F IN .75HR Cont'd	Forging Cont'd	1.20	131.0	3.005	1.375	CT	1.599	0.85	76.30	Cont'd	Cont'd	Cont'd	1973	85857		
		1.50	132.0	3.000	1.409	CT	1.583	0.81	76.00				1973	85034		
		1.20	133.0	3.006	1.374	CT	1.577	0.86	77.90				1973	85857		
		1.20	134.0	3.009	1.380	CT	1.625	1.15	91.00				1973	85857		
		2.25	134.0	2.998	1.499	CT	1.537	1.04	86.50				1973	85034		
		1.20	136.0	3.007	1.377	CT	1.585	1.08	89.60				1973	85857		
		--	--	3.000	1.500	CT	--	1.20	83.00				1974	89004 (1)		
		4.60	118.0	3.003	1.428	CT	1.559	1.42	89.30				1973	85634		
		6.70	121.0	3.001	1.502	CT	1.552	1.46	92.50				1973	85034		
		4.75	122.0	3.000	1.500	CT	1.573	1.37	90.20				1973	85034		
RECRYSTALLIZE ANNEAL - 1700F 4HR FC TO 1400F AT 100F/HR COOL TO 900F IN .75HR	Forging	6.70	122.0	3.000	1.502	CT	1.540	1.40	91.10	3.2	86.9	86.9	1973	85034		
		4.60	122.0	3.001	1.500	CT	1.540	1.26	86.70				1973	85634		
		6.70	122.0	2.999	1.501	CT	1.532	1.37	90.40				1973	85034		
		4.75	124.0	3.002	1.502	CT	1.544	1.37	91.70				1973	85034		
		6.70	125.0	3.001	1.501	CT	1.543	1.17	85.60				1973	85034		
		0.62	160.0	3.503	0.634	CT	1.801	0.18	43.50				1973	86836		
		0.62	160.0	3.501	0.632	CT	1.790	0.19	44.00				1973	86836		
		0.62	160.0	3.503	0.633	CT	1.831	0.16	40.30				1973	85836		
		2.00	126.9	--	2.000	--	--	1.32	92.32				--	--	--	R1005
		2.00	132.0	--	2.000	--	--	1.00	83.52				--	--	--	R1005
STA STOA	Plate Forging	2.50	142.0	2.000	1.000	CT	1.000	0.44	59.80	2.0	42.6	42.6	1974	86962		
		2.50	142.0	2.000	1.000	CT	1.000	0.47	61.40				1974	86962		
		2.50	142.0	2.000	1.000	CT	1.000	0.40	56.90				1974	86962		
		2.50	142.0	2.000	1.000	CT	1.000	0.40	56.90				1974	86962		

NOTES: (1) TYS APPROX. 120

TABLE 6.16.2.2

TI-6AL-4V K _C																			
CONDITION HEAT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K _{app}			K _C			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.)	THICK (in.)	INIT (in.)	FINAL (in.)	ONSET (Ksi)	MAX (Ksi)	K _{app} (Ksi√in.)	K _{app} MEAN	STAN DEV	K _C (Ksi√in.)	K _C MEAN	STAN DEV		
BUCKLING OF CRACK EDGES NOT RESTRAINED																			
1300F 1HR AC	Forging	---	R.T.	L-T	147.1	0.392	2.020	3.180	...	51.90	99.57	...	141.72	1964	58782		
		---				0.392	2.020	3.200	...	53.50	102.64	...	101.1	2.2	146.98	144.4	3.7	1964	58782
		---				0.732	3.000	5.000	...	54.40	126.94	...	---	---	190.38*	---	---	---	1964
1300F 1HR AC	Forging	---	R.T.	L-T	147.1	0.998	3.000	5.200	...	28.30	66.05	...	103.28	---	---	1964	58782		
BUCKLING OF CRACK EDGES RESTRAINED																			
ANNEALED	Sheet	0.04	-109	L-T	163.0	0.040	2.000	73.60	135.72	1963	54304		
		0.04				0.040	1.040	...	94.10	121.54	1963	54304	
		0.04				0.040	0.850	...	101.00	117.52	1963	54304
		0.04				0.040	1.540	...	77.20	122.89	121.4	7.0	1963	54304
		0.04				0.040	3.060	...	47.40	114.42	1963	54304
		0.04				0.040	4.840	...	33.60	121.50	1963	54304
ANNEALED	Sheet	0.04	R.T.	L-T	137.3	0.040	4.440	35.30	116.22	1963	54304		
		0.04				0.040	1.550	...	95.50	152.56*	1963	54304	
		0.04				0.040	3.030	...	61.20	146.71	1963	54304	
		0.04				0.040	2.020	...	78.60	145.78	143.2	4.3	1963	54304
		0.04				0.040	0.950	...	108.30	133.46*	1963	54304
		0.04				0.040	4.050	...	47.50	143.18	1963	54304

* NOTE: NET SECTION STRESS EXCEEDS 80% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

TABLE 6.16.2.2 (CONTINUED)

TI-6AL-4V K _C																																																																
CONDITION HEAT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K _{app}			K _C			DATE	REFER																																													
	FORM	THICK (in.)				WIDTH (in.)	THICK (in.)	INIT (in.)	FINAL (in.)	ONSET (Ksi)	MAX (Ksi)	K _{app} (Ksi/in.)	K _{app} MEAN	STAN DEV	K _C (Ksi/in.)	K _C MEAN	STAN DEV																																															
BUCKLING OF CRACK EDGES RESTRAINED																																																																
ANNEALED	Cont'd	Sheet	0.04	R.T.	L-T	137.3	8.000	0.040	5.030	...	36.20	137.12	Cont'd	...	Cont'd	...	Cont'd	1963	54304																																													
		Cont'd	0.04	Cont'd	Cont'd	137.3	8.000	0.040	0.480	...	126.00	109.65*	Cont'd	...	Cont'd	...	Cont'd	1963	54304																																													
ANNEALED	Sheet	0.04	0.04	550	L-T	96.7	8.000	0.040	2.020	...	80.50	149.31*	1963	54304																																													
																				0.04	8.000	0.040	1.260	...	91.00	130.02*	1963	54304																												
																																					0.04	8.000	0.040	0.570	...	101.20	96.06*	1963	54304													
																																																				0.04	8.000	0.040	0.950	...	94.90	116.95*	1963	54304
BUCKLING OF CRACK EDGES UNKNOWN																																																																
MA	Plate	0.25	R.T.	L-T	130.0	9.630	0.280	3.900	5.220	25.60	51.70	140.00	...	182.38*	1971	83984																																														
MA	Plate	0.25	R.T.	L-T	130.0	9.630	0.286	5.780	6.860	18.70	37.80	148.60	...	187.79*	1971	83984																																														
MA	Plate	0.25	R.T.	L-T	130.0	9.630	0.293	1.920	...	36.50	85.10	161.62*	1971	83984																																														
MA	Plate	0.25	R.T.	L-T	130.0	16.120	0.260	8.020	10.580	31.30	45.10	189.99	...	254.46*	1971	83984																																														
MA	Plate	0.25	R.T.	L-T	130.0	16.120	0.275	3.990	6.100	31.10	67.50	175.67	...	229.66*	1971	83984																																														
MA	Plate	0.25	R.T.	L-T	130.0	16.110	0.285	12.93	13.550	11.30	15.40	125.65	...	142.95	1971	83984																																														
MA	Plate	0.25	R.T.	L-T	130.0	18.000	0.265	4.540	...	53.90	75.30	209.36	1971	83984																																														
MA	Plate	0.25	R.T.	L-T	130.0	18.000	0.265	4.640	...	35.60	75.40	212.33	210.8	1971	83984																																														
MA	Plate	0.25	R.T.	L-T	130.0	32.160	0.262	6.020	...	39.20	75.20	236.38	1971	83984																																														

* NOTE: NET SECTION STRESS EXCEEDS 80% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

TABLE 6.16.2.2 (CONTINUED)

TI-6AL-4V K _C																				
CONDITION HEAT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN		CRACK LENGTH		GROSS STRESS			K _{app}			K _C			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	INIT (in.) 2a _i	FINAL (in.) 2a _f	ONSET (KSI) σ _s	MAX (KSI) σ _m	K _{app} (KSI/in.)	K _{app} MEAN	STAN DEV	K _C (KSI/in.)	K _C MEAN	STAN DEV			
BUCKLING OF CRACK EDGES UNKNOWN																				
MA	Plate	0.25	130.0		0.270	32.150	12.72	17.000	33.40	47.00	233.00	295.74	257.3	64.3	1971	83984				
		0.25	130.0	R.T.	0.273	32.140	19.20	21.700	14.20	26.20	187.15	218.88	212.5	23.3	1971	83984				
		0.25	130.0		0.275	32.130	8.010	---	30.60	58.90	217.31	---	---	---	1971	83984				
MA	Plate	0.25	130.0	R.T.	0.288	32.160	8.070	---	40.20	55.40	205.27	---	---	1971	83984					
BUCKLING OF CRACK EDGES RESTRAINED																				
MA	Sheet	0.05	136.5		0.047	24.070	10.00	---	---	41.60	184.97	---	---	1964	57573					
		0.05	133.8		0.048	24.020	6.000	6.560	51.80	52.10	166.39	175.22	---	1964	57573					
		0.05	136.5		0.049	23.900	5.990	---	---	47.30	150.98	---	---	1964	57573					
		0.05	136.5		0.049	24.060	6.000	10.500	43.60	49.60	168.39	228.95	---	1964	57573					
		0.05	136.5		0.049	24.060	6.000	---	---	60.70	193.83	---	---	1964	57573					
		0.05	136.5		0.050	23.990	6.000	---	---	47.50	151.72	---	---	1964	57573					
		0.05	133.8		0.050	24.050	6.000	---	---	41.30	51.50	164.46	---	---	1964	57573				
		0.05	136.5	R.T.	0.050	24.060	3.000	---	---	78.80	172.72	167.0	12.8	196.4	19.9	1964	57573			
		0.05	136.5		0.050	24.070	6.000	---	---	56.60	180.74	---	---	---	---	1964	57573			
		0.05	136.7		0.050	24.070	6.000	---	---	53.20	169.88	---	---	---	---	1964	57573			
		0.05	136.7		0.052	24.070	6.010	7.350	41.10	51.80	165.57	---	---	---	---	1964	57573			
		0.05	136.7		0.052	24.070	6.010	6.750	43.80	51.40	164.29	---	---	---	---	1964	57573			
0.05	136.0		0.054	24.080	6.000	8.120	38.40	47.00	150.08	---	---	---	---	1964	57573					
0.05	136.0		0.055	24.080	6.010	8.170	45.90	51.50	164.60	---	---	---	---	1964	57573					

TABLE 6.16.2.2 (CONCLUDED)

CONDITION HEAT TREAT		PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K _{app}			K _c			DATE	REFER	
		FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	INIT (in.) S ₀	FINAL (in.) S _f	ONSET (Ksi) σ ₀	MAX (Ksi) σ _{max}	K _{app} (Ksi/in.)	K _{app} MEAN	STAN DEV	K _c (Ksi/in.)	K _c MEAN	STAN DEV			
BUCKLING OF CRACK SIDES RESTRAINED																					
MA	Sheet	0.20	129.3	24.060	0.212	6.000	---	---	81.00	258.66*	---	---	---	---	---	---	---	---	---	1964	57573
		0.20	129.3	24.070	0.212	3.010	---	---	93.10	204.41*	---	---	---	---	---	---	---	---	---	1964	57573
		0.20	129.3	24.070	0.212	6.010	---	---	65.60	70.60	225.66	---	---	---	---	---	---	---	---	1964	57573
MA	Sheet	0.20	129.3	24.070	0.218	6.000	---	---	76.40	243.96	---	---	---	---	---	---	---	---	---	1964	57573
		0.20	129.3	24.080	0.220	10.00	---	---	55.30	246.86	244.9	1.3	---	---	---	---	---	---	1964	57573	
MA	Sheet	0.05	136.7	24.370	0.062	5.990	---	---	54.20	172.73	---	---	---	---	---	---	---	---	1964	57573	
MA	Sheet	0.05	136.5	24.070	0.061	5.990	7.100	43.30	47.50	151.53	---	---	---	---	167.72	---	---	---	1964	57573	
MA	Sheet	0.05	163.3	8.000	0.060	1.980	3.130	95.00	84.70	165.28	---	---	---	---	207.78*	---	---	---	1964	57573	
		0.05	163.3	8.010	0.060	1.970	2.540	72.50	77.30	141.29	---	---	---	---	164.74	159.4	7.5	---	1964	57573	
		0.05	164.3	8.010	0.062	1.920	2.370	66.60	75.50	135.97	144.2	10.0	---	---	154.07	---	---	---	1964	57573	
MA	Sheet	0.03	127.0	8.040	0.025	1.980	2.600	61.20	75.60	138.54	---	---	---	---	163.45*	---	---	---	1964	57573	
MA	Sheet	0.05	136.0	8.010	0.062	2.000	2.220	90.00	95.30	175.72*	---	---	---	---	186.89*	---	---	---	1964	57573	
MA	Sheet	0.13	139.7	8.060	0.127	1.980	2.740	48.00	103.70	190.02*	---	---	---	---	231.93*	---	---	---	1964	57573	
MA	Sheet	0.03	80.2	8.030	0.025	2.060	2.460	75.70	77.70	145.73*	---	---	---	---	162.23*	---	---	---	1964	57573	
MA	Sheet	0.05	81.7	8.020	0.061	2.000	2.090	63.40	75.40	139.01*	---	---	---	---	142.64*	---	---	---	1964	57573	

* NOTE: NET SECTION STRESS EXCEEDS 80% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

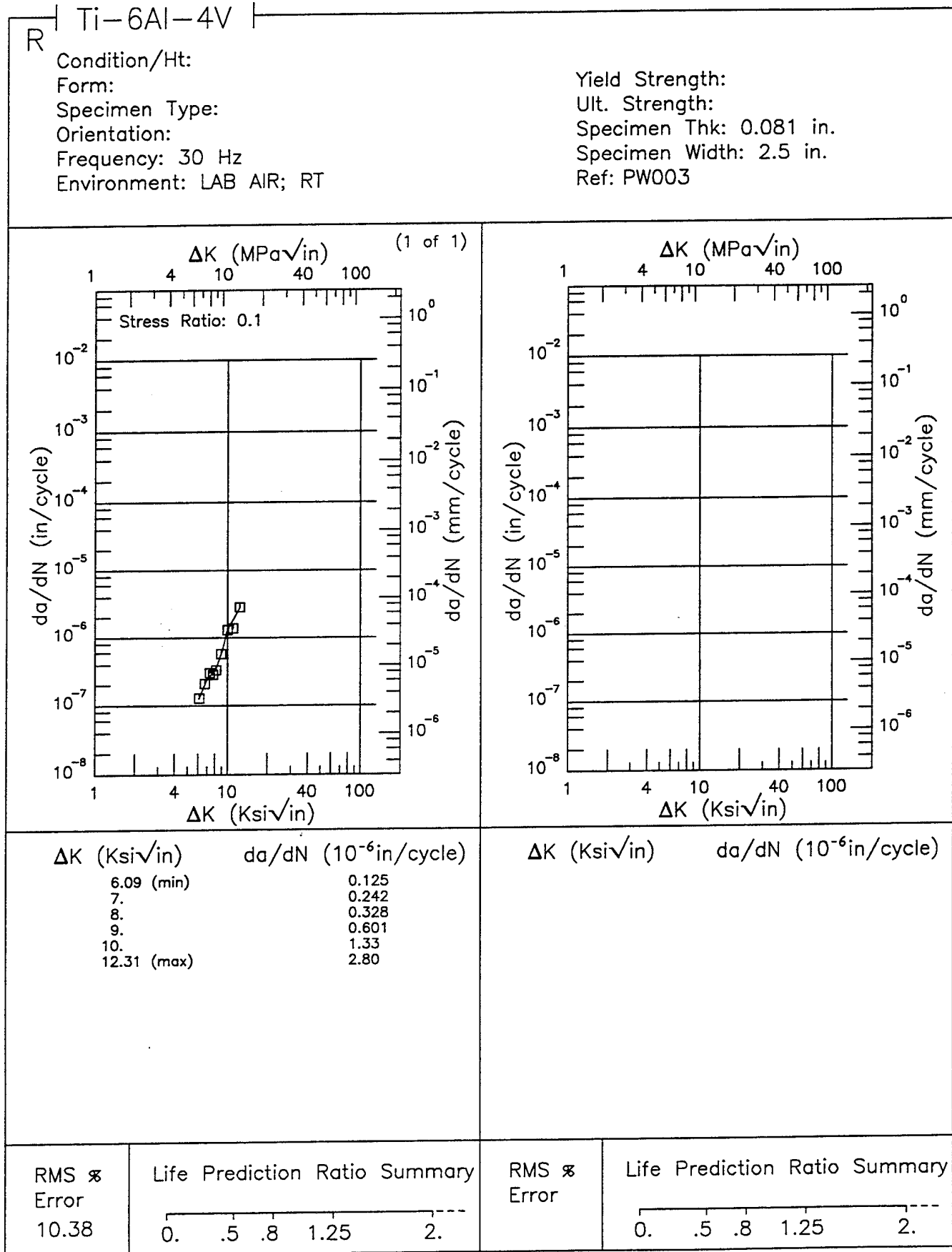


Figure 6.16.3.1.1

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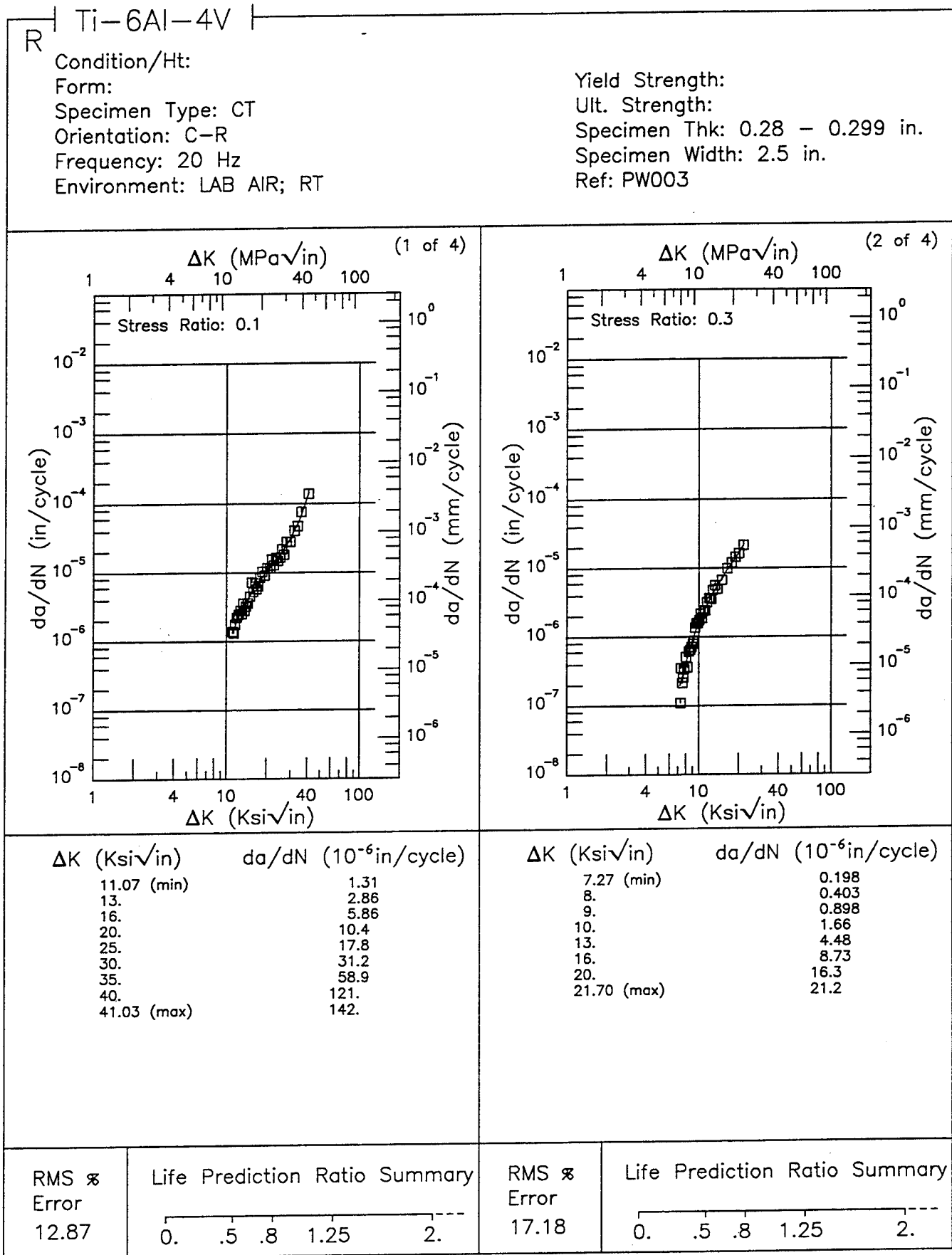
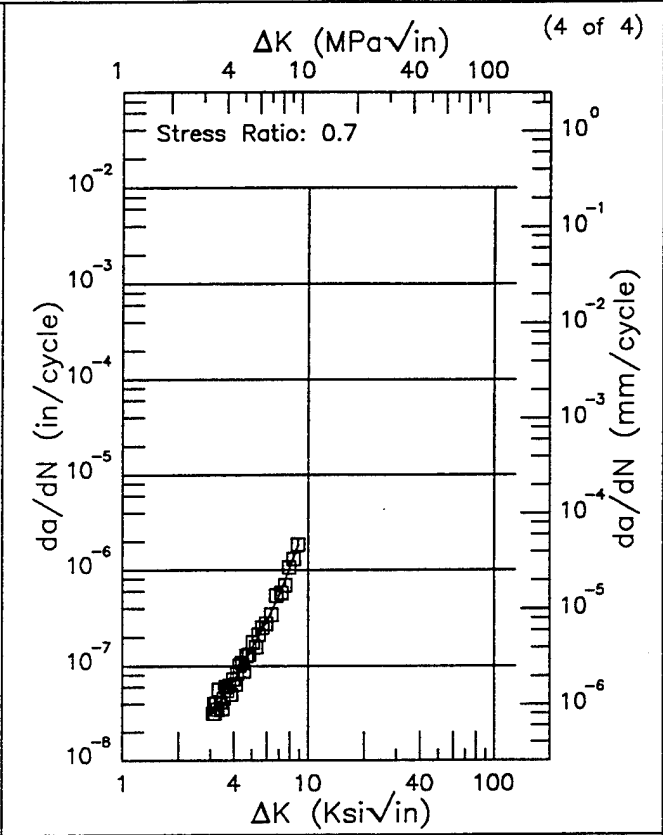
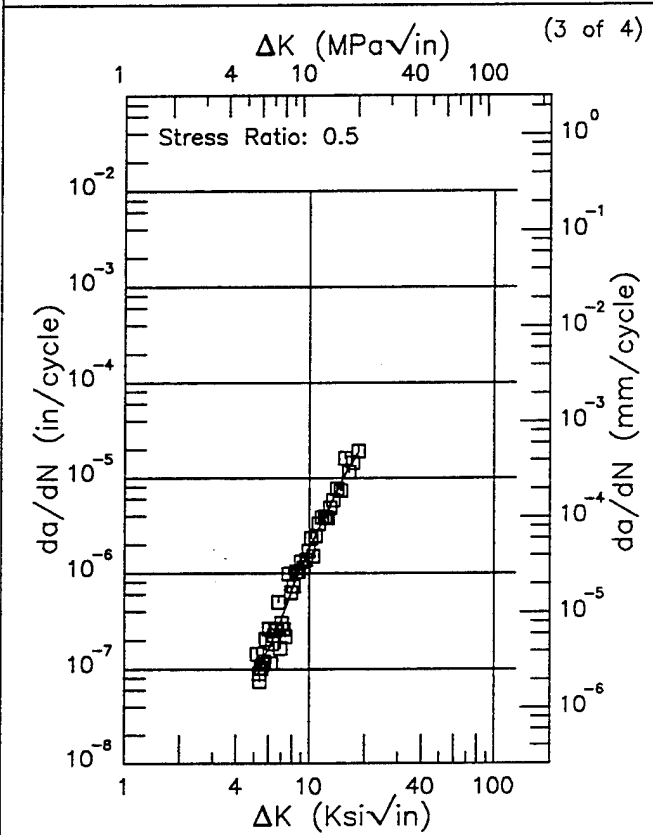


Figure 6.16.3.1.2

Ti-6Al-4V R

Condition/Ht:
 Form:
 Specimen Type: CT
 Orientation: C-R
 Frequency: 20 Hz
 Environment: LAB AIR; RT

Yield Strength:
 Ult. Strength:
 Specimen Thk: 0.28 - 0.299 in.
 Specimen Width: 2.5 in.
 Ref: PW003



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
5.18 (min)	0.102
6.	0.161
7.	0.333
8.	0.664
9.	1.16
10.	1.82
13.	5.28
16.	12.3
18.26 (max)	18.2

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
3.07 (min)	0.0347
3.5	0.0467
4.	0.0702
5.	0.154
6.	0.295
7.	0.545
8.	1.04
8.80 (max)	1.82

RMS % Error	Life Prediction Ratio Summary
28.66	0. .5 .8 1.25 2.

RMS % Error	Life Prediction Ratio Summary
12.39	0. .5 .8 1.25 2.

Figure 6.16.3.1.2 (Concluded)

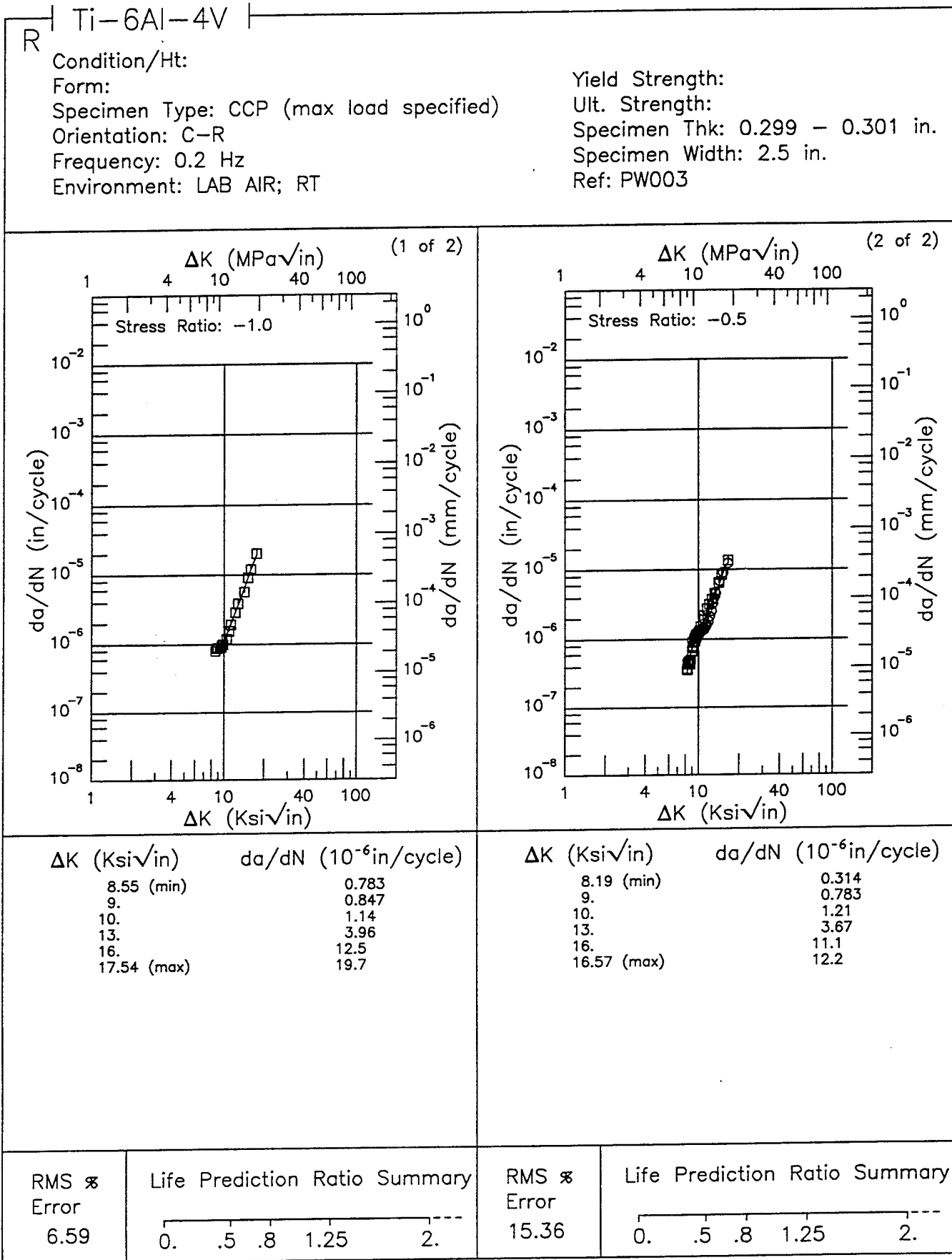


Figure 6.16.3.1.3

Ti-6Al-4V R

Condition/Ht:
 Form:
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Frequency: 0.2 Hz
 Environment: LAB AIR;300°F

Yield Strength:
 Ult. Strength:
 Specimen Thk: 0.297 - 0.302 in.
 Specimen Width: 2.5 in.
 Ref: PW003

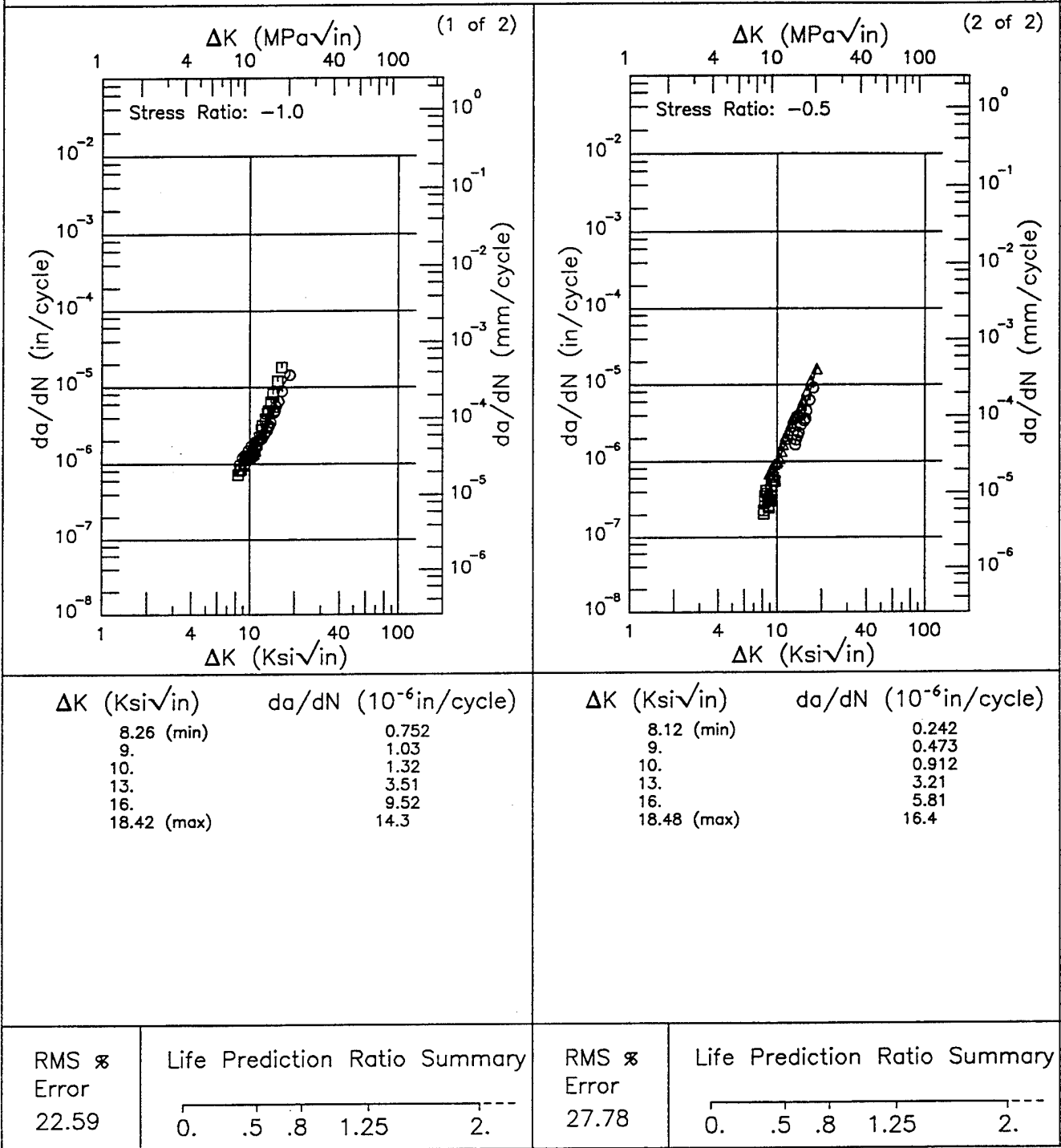


Figure 6.16.3.1.4

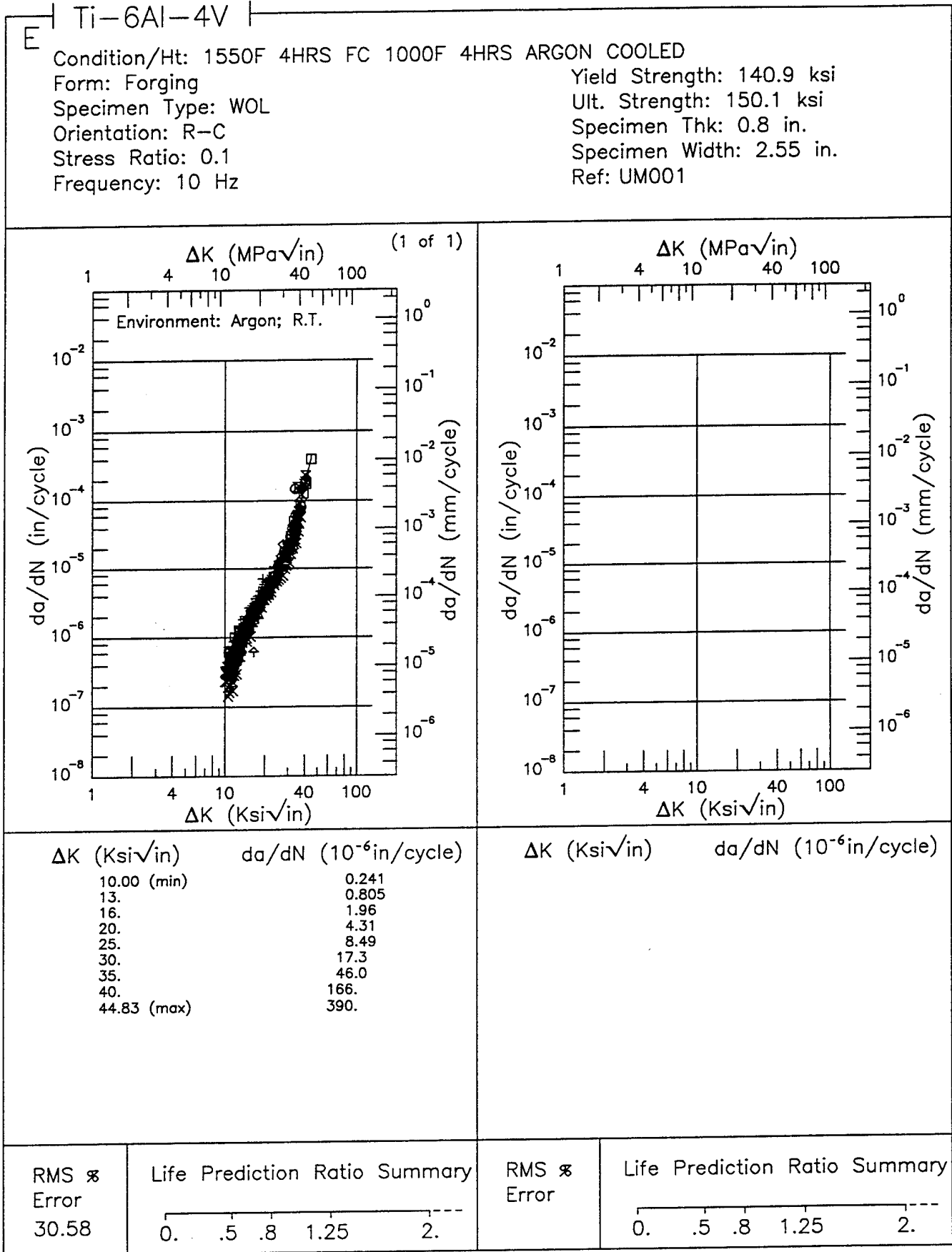


Figure 6.16.3.1.5

Ti-6Al-4V

E

Condition/Ht: 1750F 4HRS ARGON COOLED
 Form: Forging
 Specimen Type: WOL
 Orientation: R-C
 Stress Ratio: 0.1
 Frequency: 10 Hz

Yield Strength: 135.7 ksi
 Ult. Strength: 144.9 ksi
 Specimen Thk: 0.8 in.
 Specimen Width: 2.55 in.
 Ref: UM001

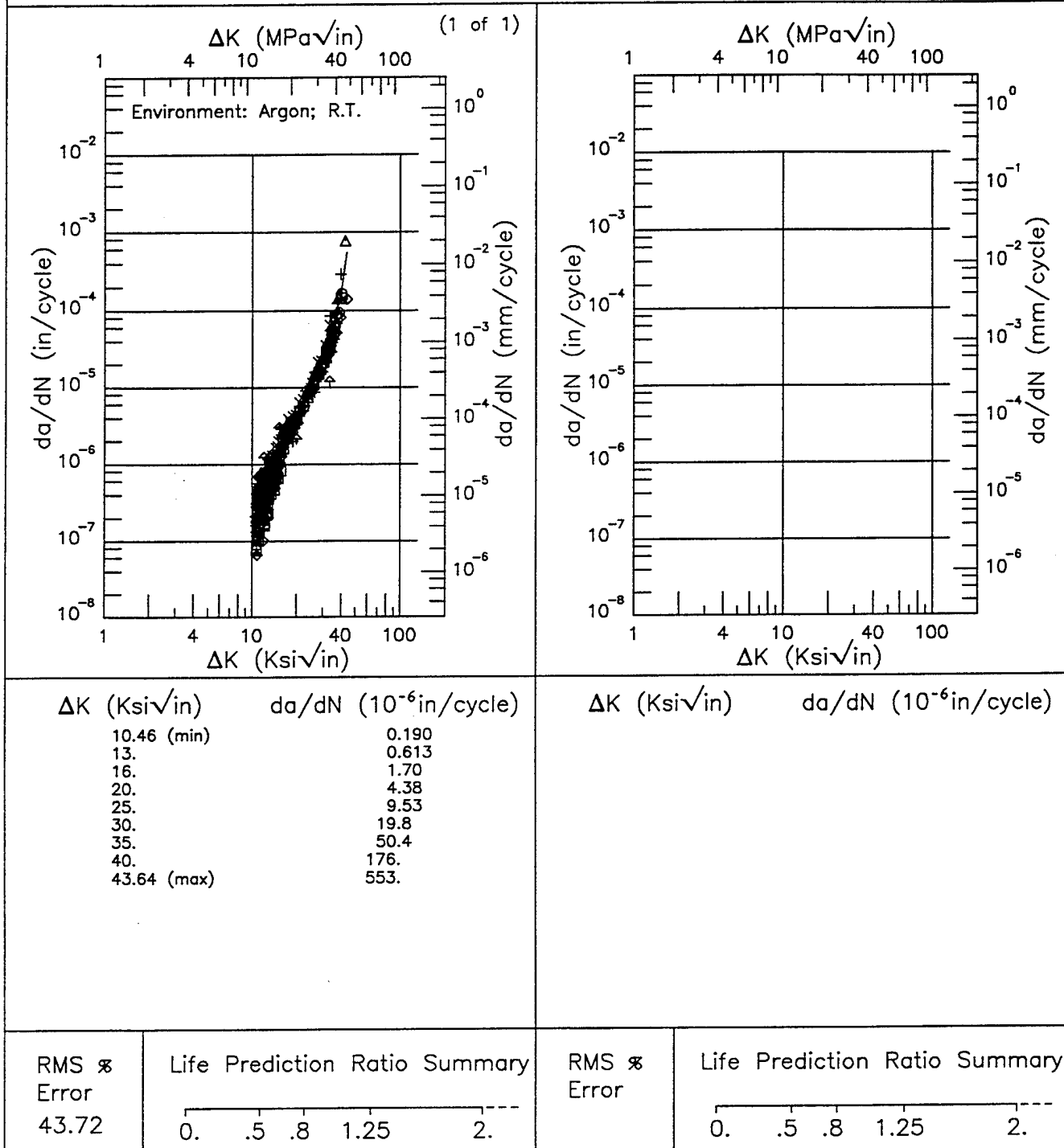


Figure 6.16.3.1.6

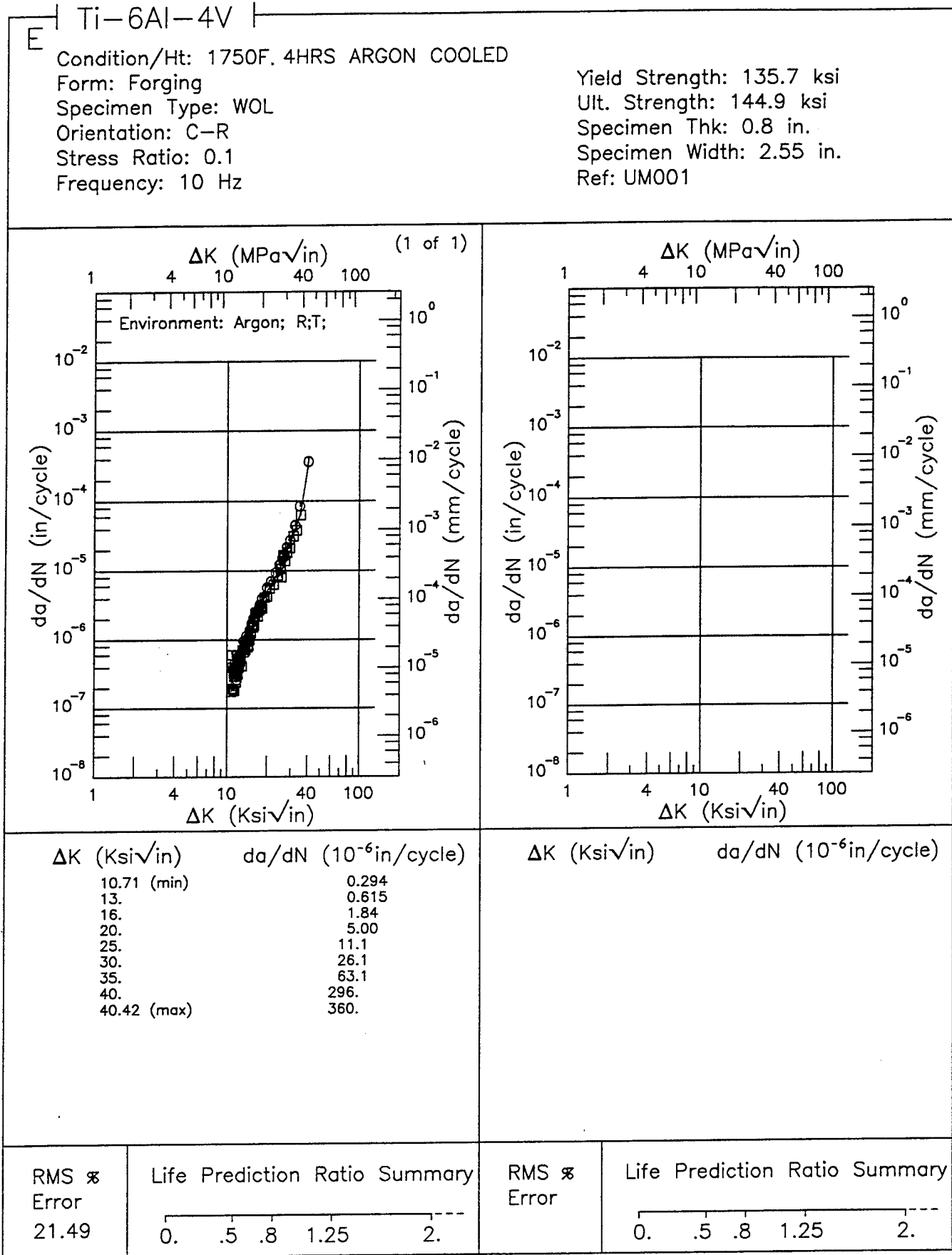


Figure 6.16.3.1.7

Ti-6Al-4V

E

Condition/Ht: 1775F 1HR WQ 1675F 1HR WQ 1000F 4HR AC

Form: 0.5 in. Disk

Specimen Type: CCP (max load specified)

Orientation: C-R

Stress Ratio: 0.05

Frequency: 0.3 - 10 Hz

Yield Strength: 120 ksi

Ult. Strength: 130 ksi

Specimen Thk: 0.12 in.

Specimen Width: 1.98 in.

Ref: GE003

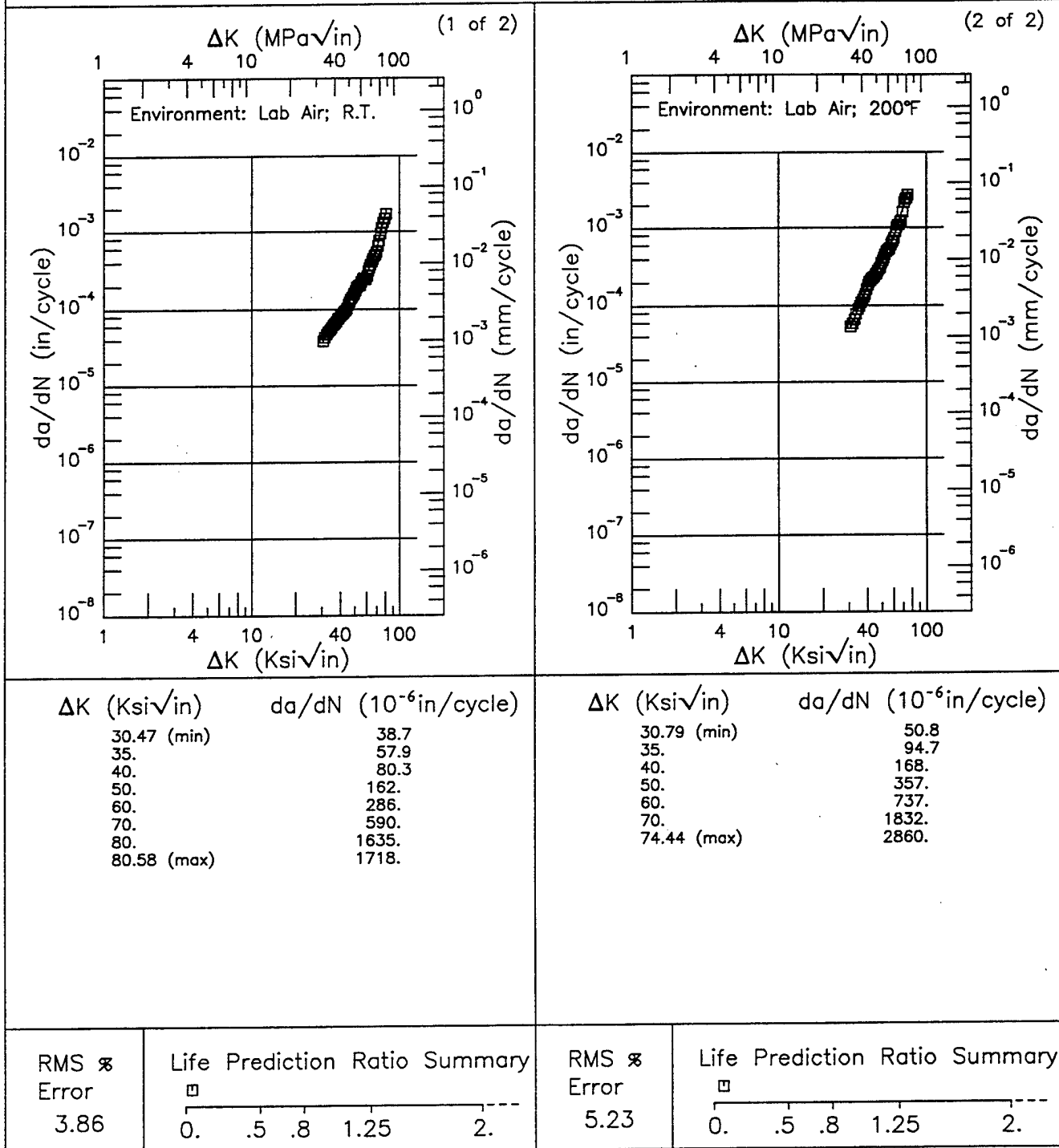
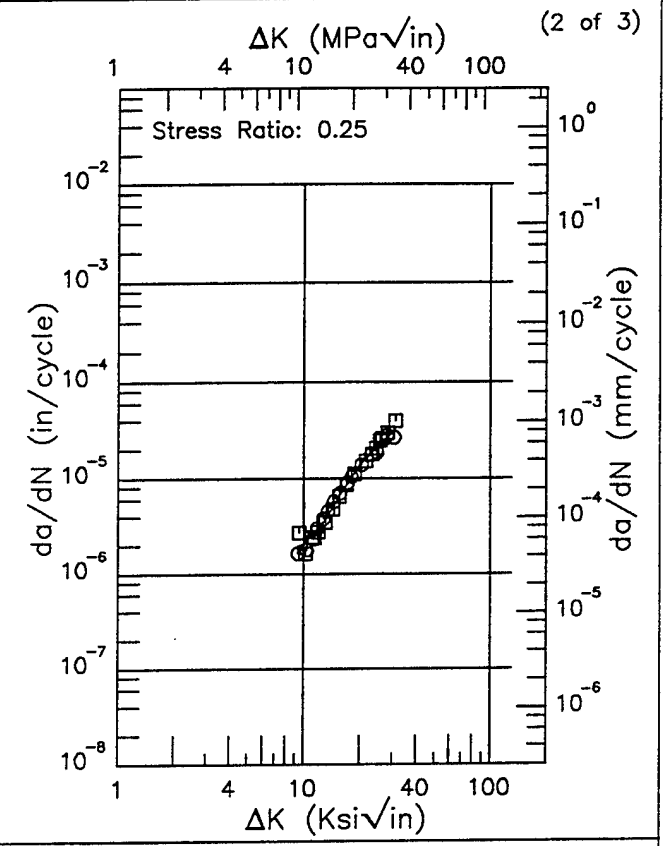
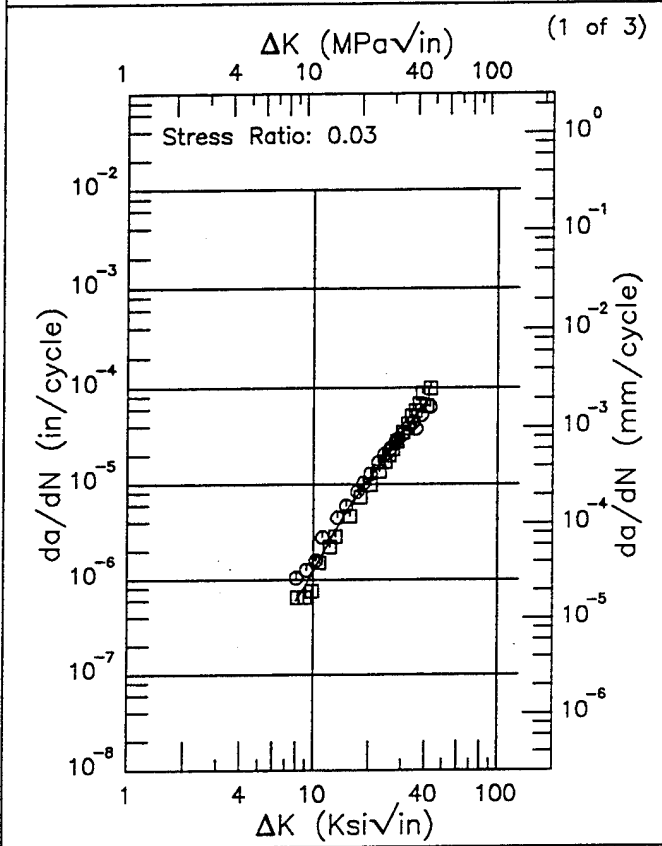


Figure 6.16.3.1.8

R | Ti-6Al-4V |

Condition/Ht: 1775F 1HR WQ 1675F 1HR WQ 1000F-1200F 2-8HR AC
 Form: 0.94 - 1.15 in. Disk Yield Strength: 145 - 150 ksi
 Specimen Type: KB Bar Ult. Strength:
 Orientation: C-R Specimen Thk: 0.251 - 0.253 in.
 Frequency: 0.3 Hz Specimen Width: 0.995 - 1.002 in.
 Environment: LAB AIR;600°F Ref: GE007



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.01 (min)	0.600
9.	0.864
10.	1.24
13.	3.17
16.	6.31
20.	11.5
25.	19.7
30.	32.2
35.	51.2
40.	68.1
42.82 (max)	65.3

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
9.28 (min)	1.96
10.	2.06
13.	3.61
16.	7.04
20.	13.7
25.	21.5
30.	30.1
30.82 (max)	31.9

RMS % Error 22.00	Life Prediction Ratio Summary 0. 0.5 .8 1.25 2. ---
----------------------	--

RMS % Error 12.30	Life Prediction Ratio Summary 0. .5 .8 1.25 2. ---
----------------------	---

Figure 6.16.3.1.9

Ti-6Al-4V R

Condition/Ht: 1775F 1HR WQ 1675F 1HR WQ 1000F-1200F 2-8HR AC
 Form: 0.94 - 1.15 in. Disk
 Specimen Type: KB Bar
 Orientation: C-R
 Frequency: 0.3 Hz
 Environment: LAB AIR;600°F
 Yield Strength: 145 - 150 ksi
 Ult. Strength:
 Specimen Thk: 0.251 - 0.253 in.
 Specimen Width: 0.995 - 1.002 in.
 Ref: GE007

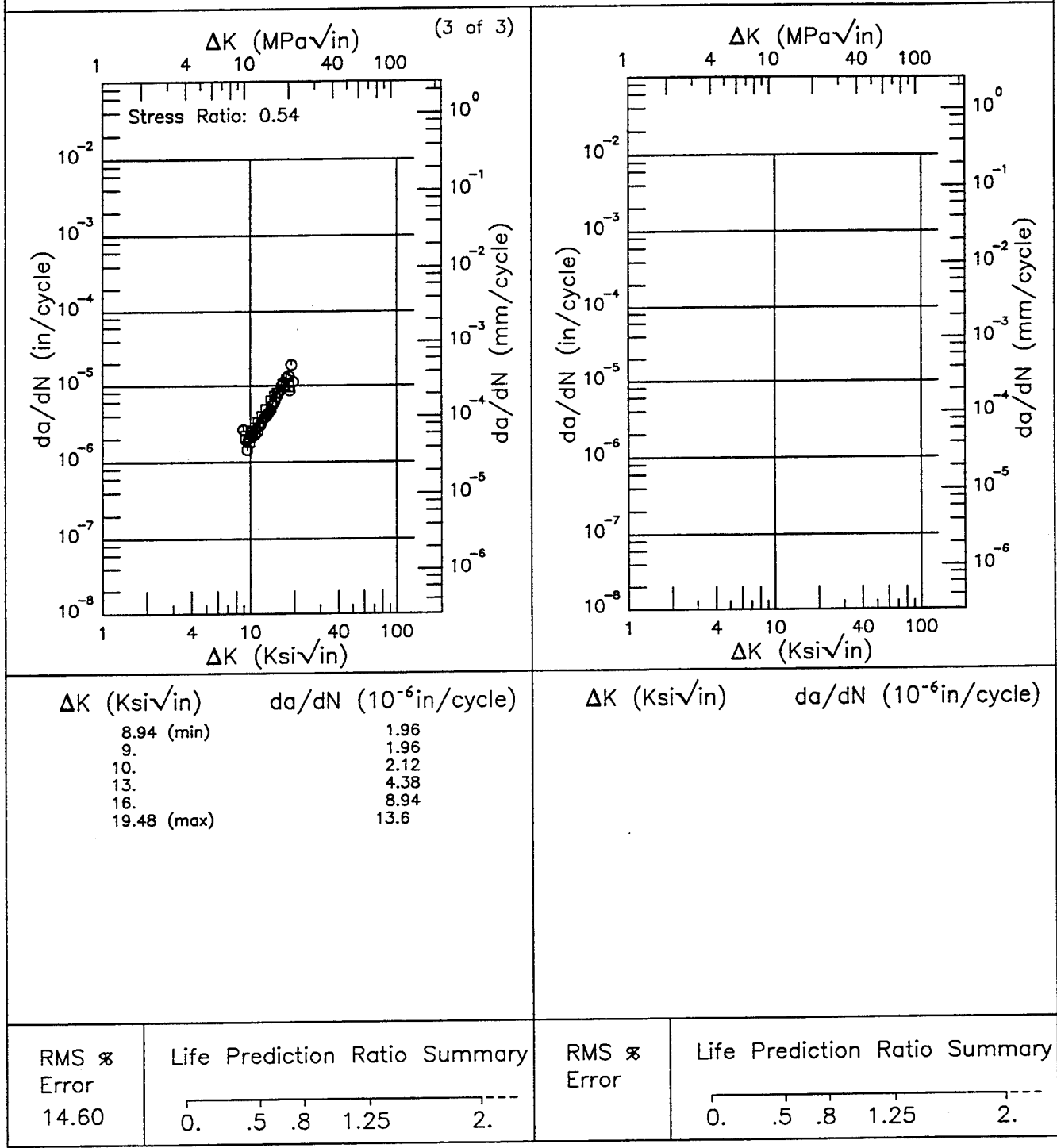
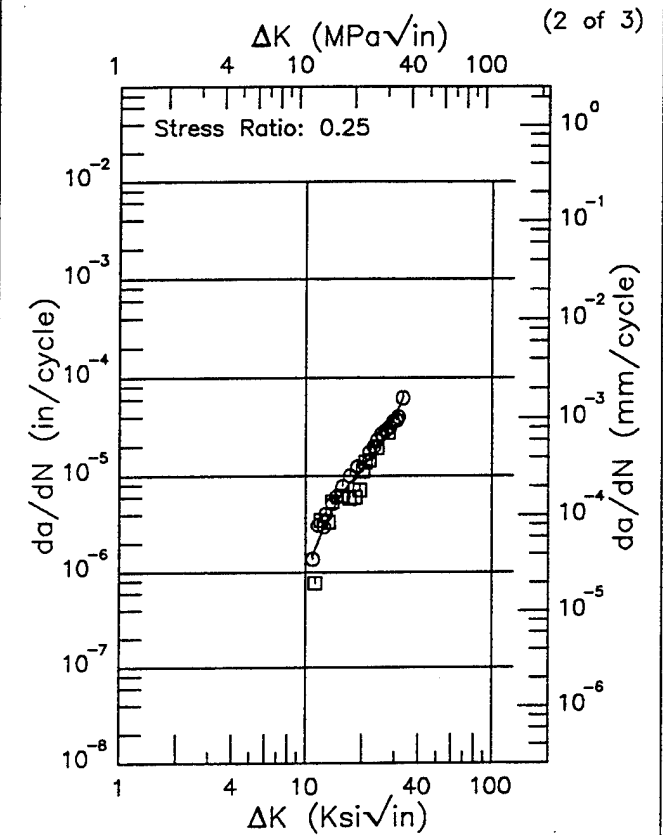
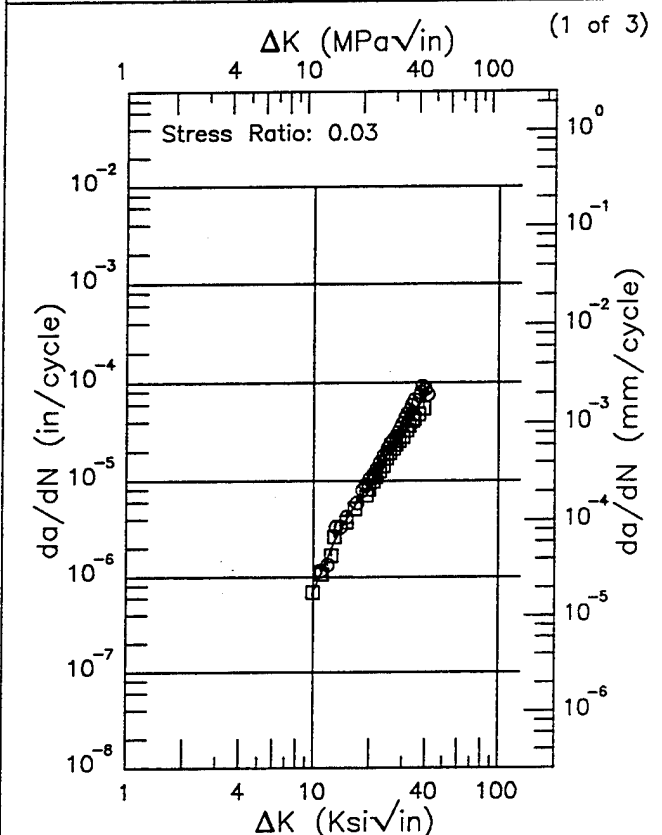


Figure 6.16.3.1.9 (Concluded)

R

Ti-6Al-4V

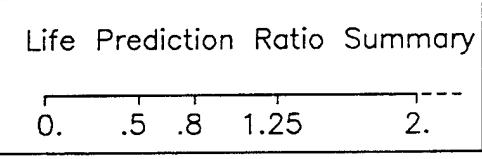
Condition/Ht: 1775F 1HR WQ 1675F 1HR WQ 1000F-1200F 2-8HR AC
 Form: 0.94 in. Disk Yield Strength: 150 ksi
 Specimen Type: KB Bar Ult. Strength:
 Orientation: C-R Specimen Thk: 0.252 in.
 Frequency: 0.3 Hz Specimen Width: 0.991 - 0.999 in.
 Environment: LAB AIR;300°F Ref: GE007



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
9.83 (min)	0.644
10.	0.709
13.	2.40
16.	4.92
20.	9.75
25.	19.7
30.	34.6
35.	53.3
40.	76.6
41.12 (max)	82.6

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.90 (min)	1.46
13.	3.46
16.	6.88
20.	11.9
25.	20.9
30.	39.8
32.83 (max)	60.4

RMS % Error
14.98



RMS % Error
22.63

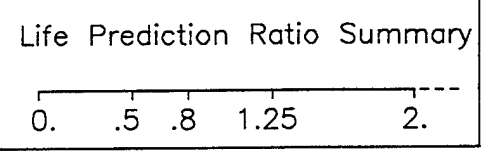


Figure 6.16.3.1.10

Ti-6Al-4V R

Condition/Ht: 1775F 1HR WQ 1675F 1HR WQ 1000F-1200F 2-8HR AC
 Form: 0.94 in. Disk
 Specimen Type: KB Bar
 Orientation: C-R
 Frequency: 0.3 Hz
 Environment: LAB AIR;300°F

Yield Strength: 150 ksi
 Ult. Strength:
 Specimen Thk: 0.252 in.
 Specimen Width: 0.991 - 0.999 in.
 Ref: GE007

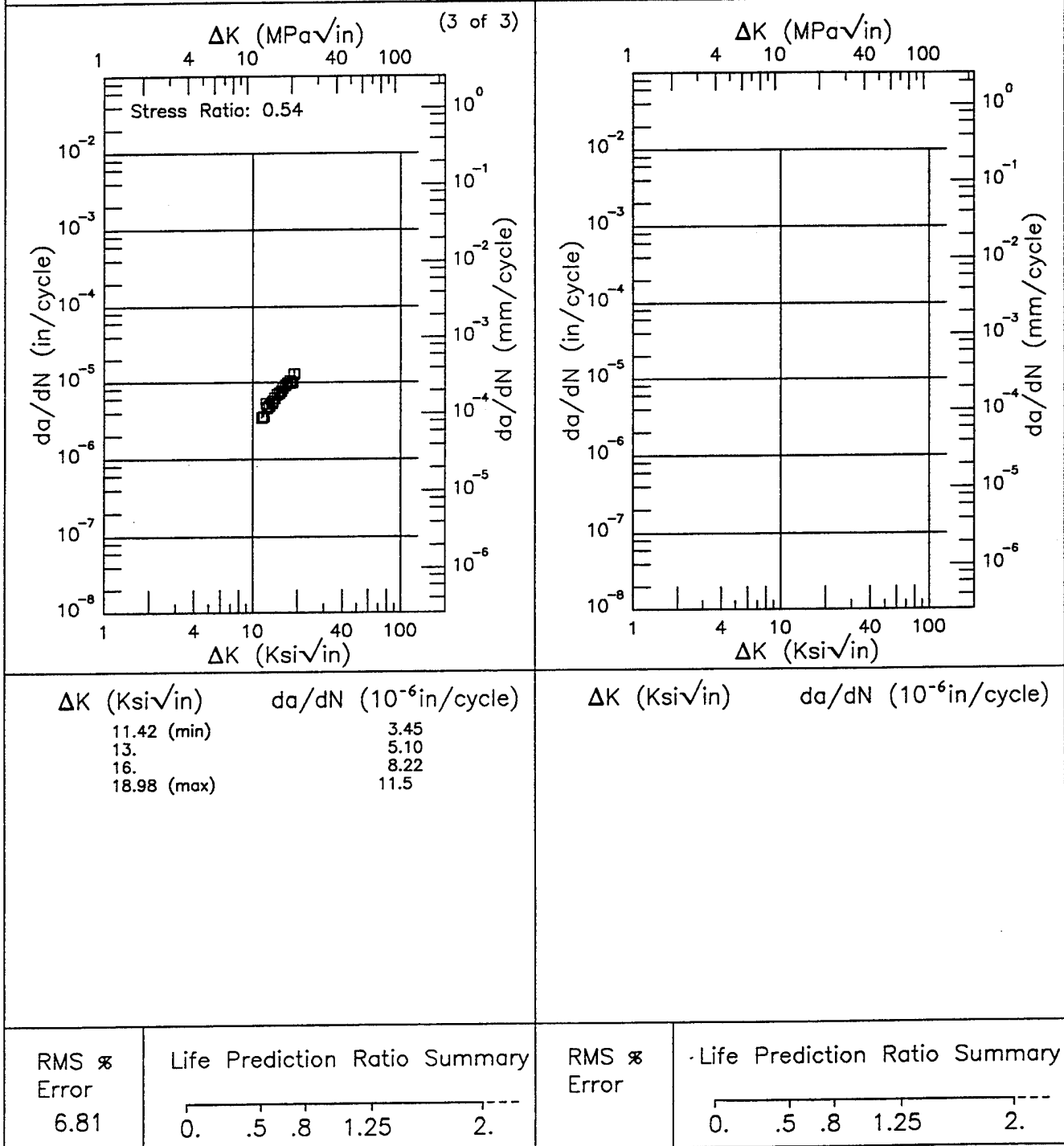


Figure 6.16.3.1.10 (Concluded)

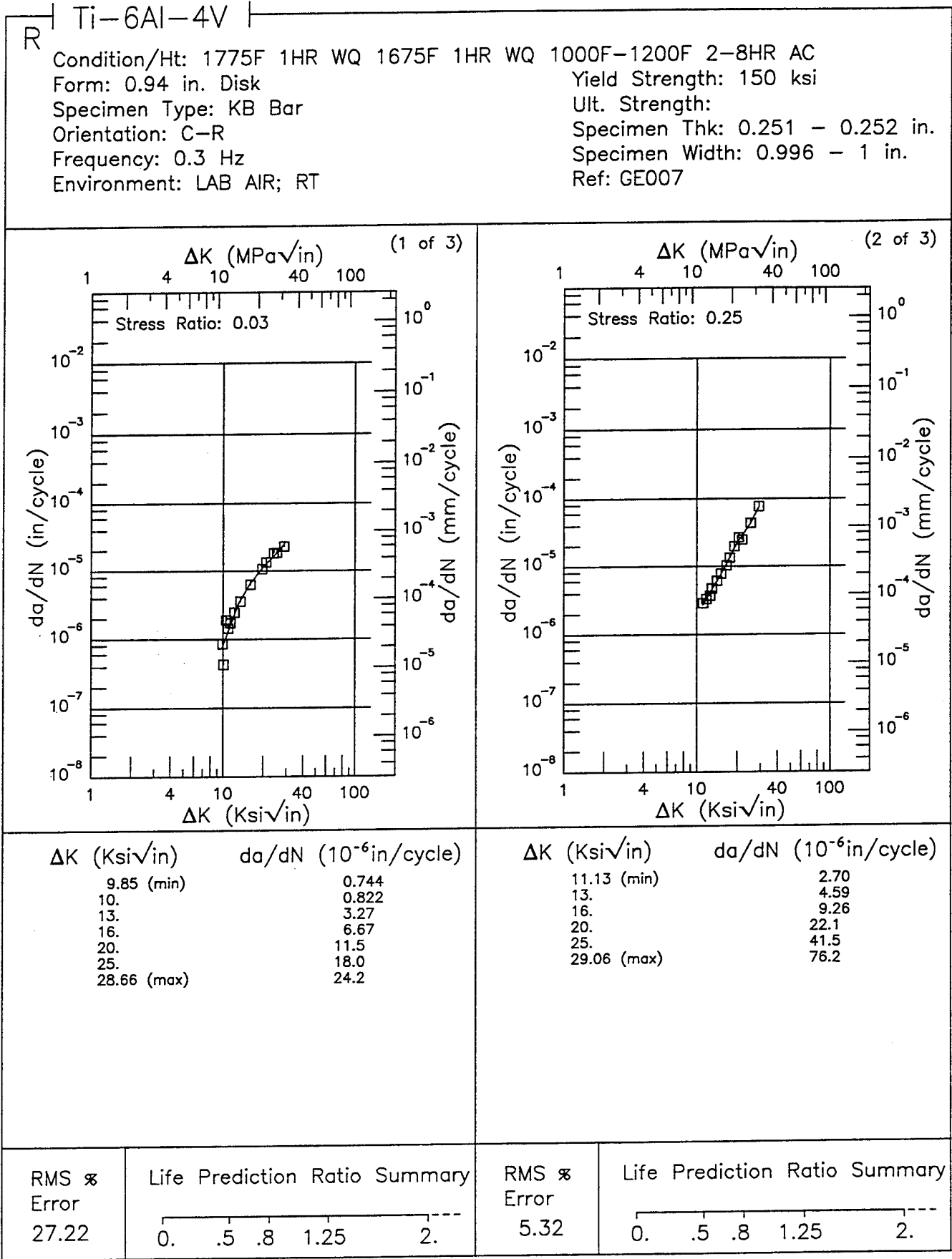
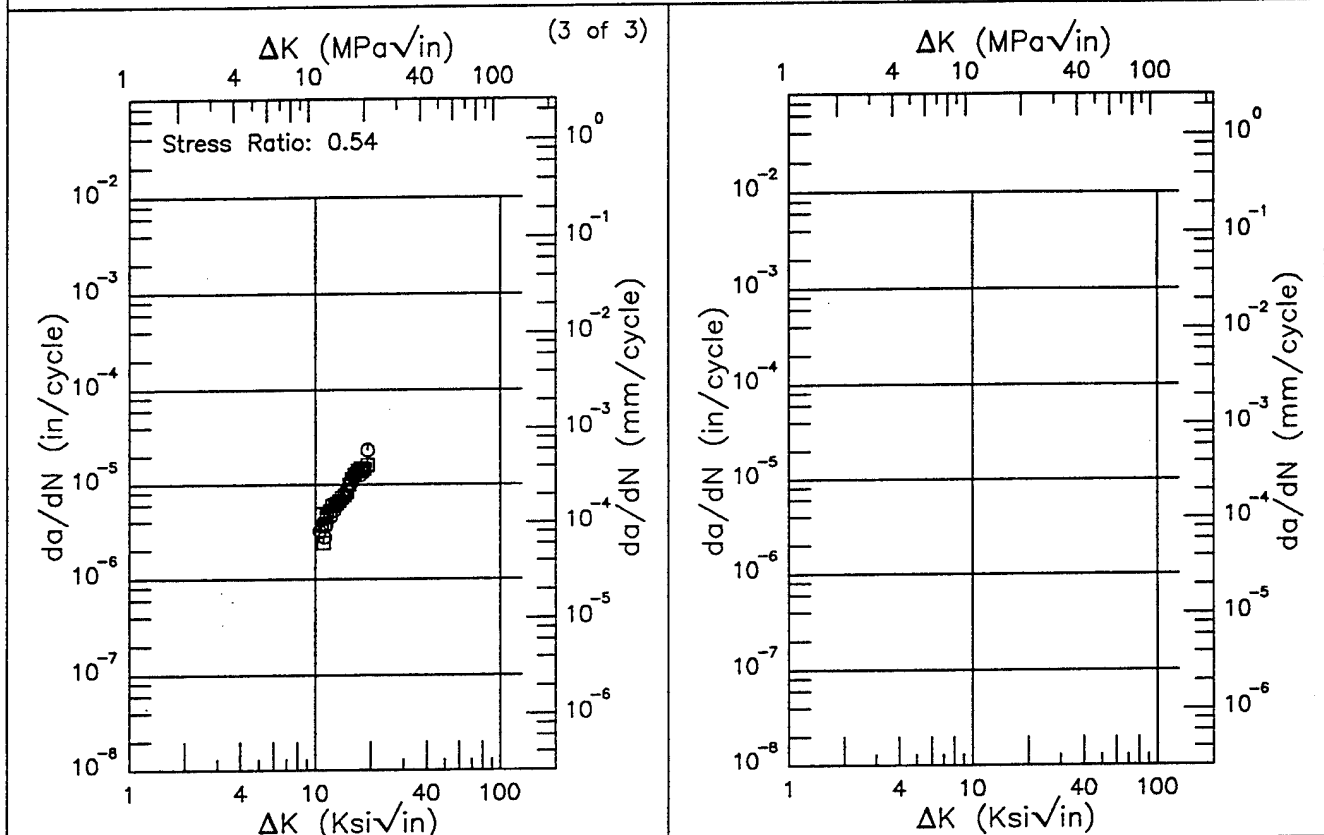


Figure 6.16.3.1.11

Ti-6Al-4V R

Condition/Ht: 1775F 1HR WQ 1675F 1HR WQ 1000F-1200F 2-8HR AC
 Form: 0.94 in. Disk
 Specimen Type: KB Bar
 Orientation: C-R
 Frequency: 0.3 Hz
 Environment: LAB AIR; RT

Yield Strength: 150 ksi
 Ult. Strength:
 Specimen Thk: 0.251 - 0.252 in.
 Specimen Width: 0.996 - 1 in.
 Ref: GE007



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)	ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.48 (min)	3.27		
13.	6.41		
16.	12.1		
19.06 (max)	16.6		

RMS % Error 13.84	Life Prediction Ratio Summary 0. .5 .8 1.25 2.	RMS % Error	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---	-------------	---

Figure 6.16.3.1.11 (Concluded)

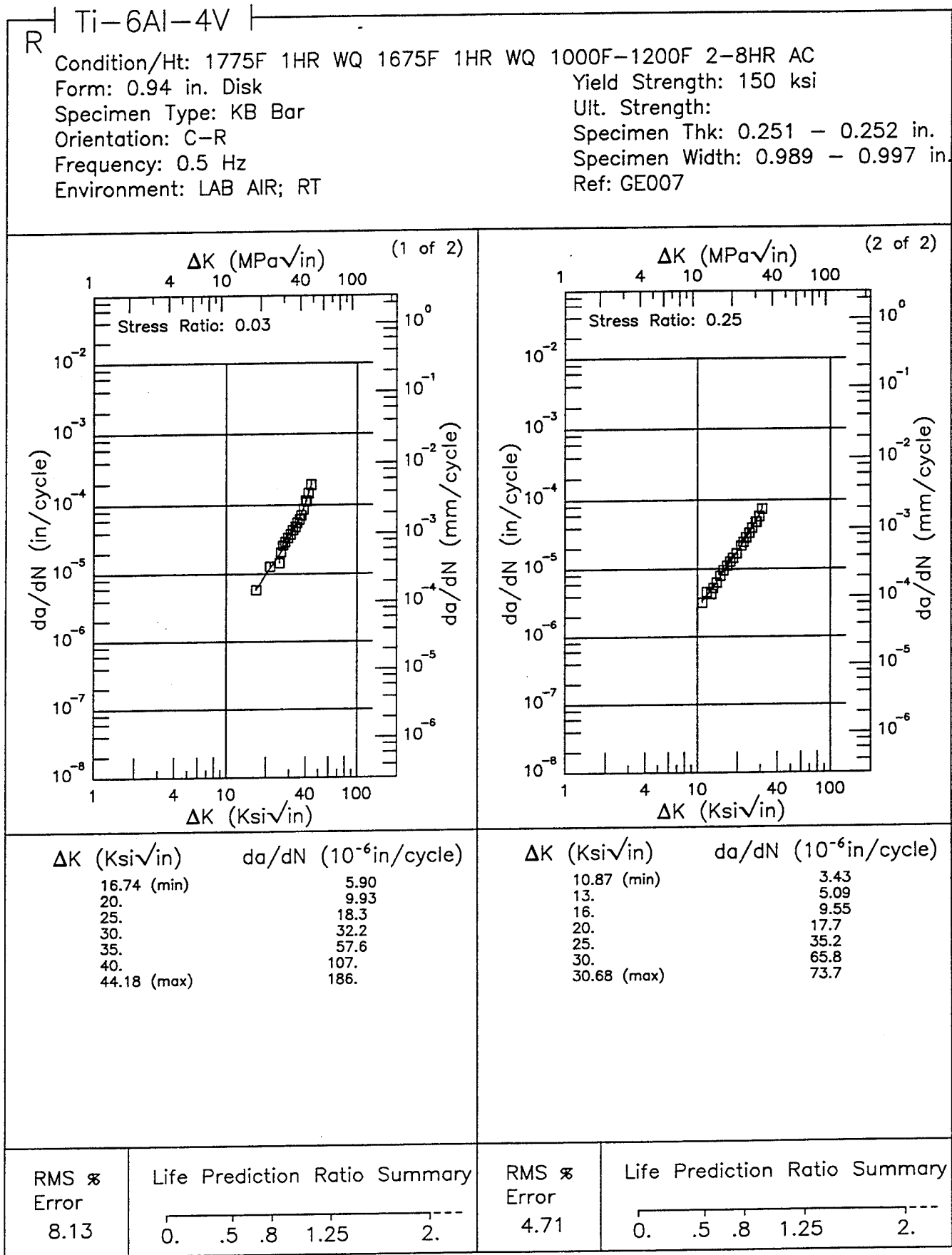


Figure 6.16.3.1.12

Ti-6Al-4V E

Condition/Ht: 1950F 4HRS WQ 1000F 4HRS ARGON COOLED
 Form: Forging
 Specimen Type: WOL
 Orientation: R-C
 Stress Ratio: 0.1
 Frequency: 10 Hz

Yield Strength: 136.4 ksi
 Ult. Strength: 147.8 ksi
 Specimen Thk: 0.8 in.
 Specimen Width: 2.55 in.
 Ref: UM001

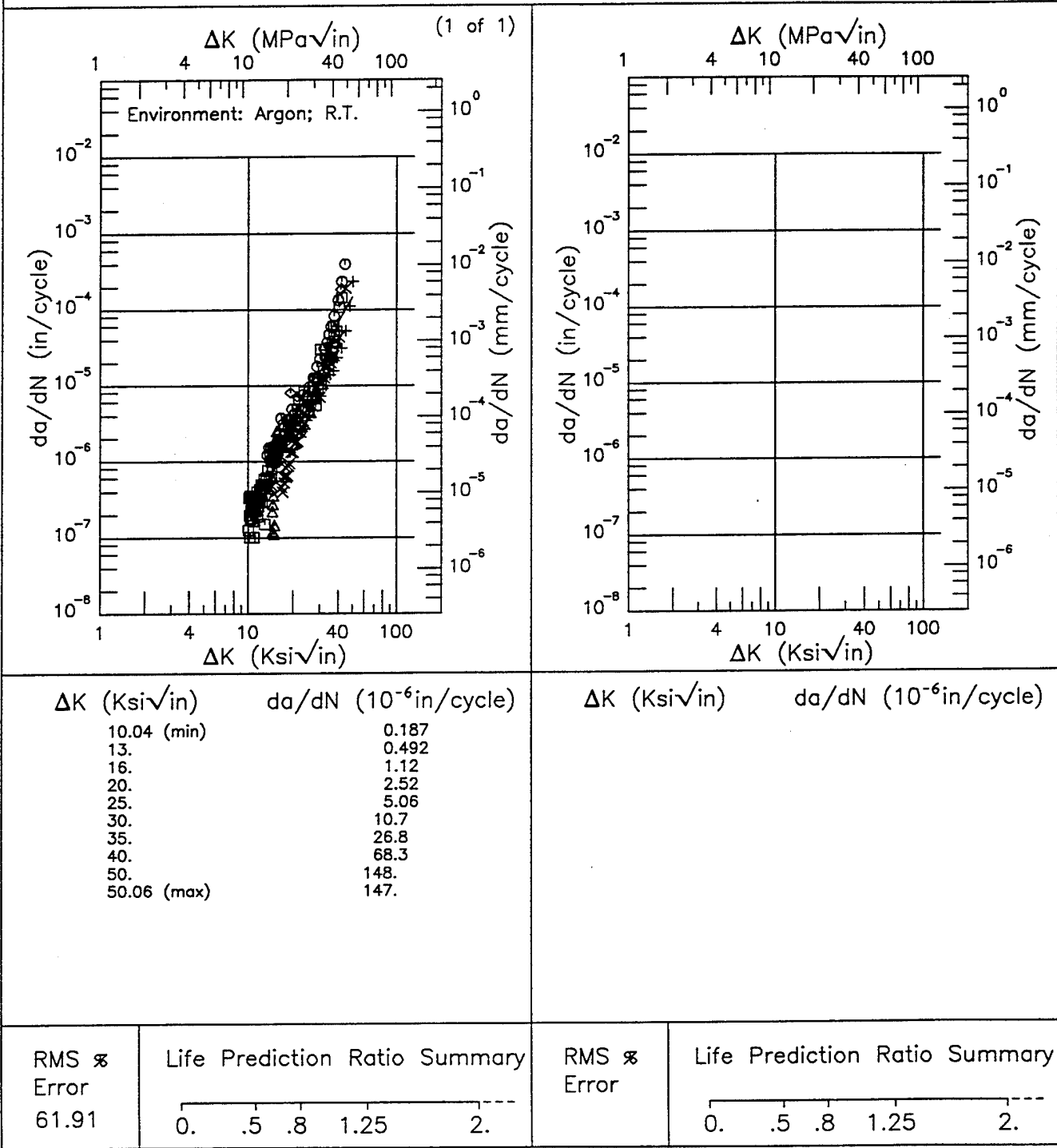


Figure 6.16.3.1.13

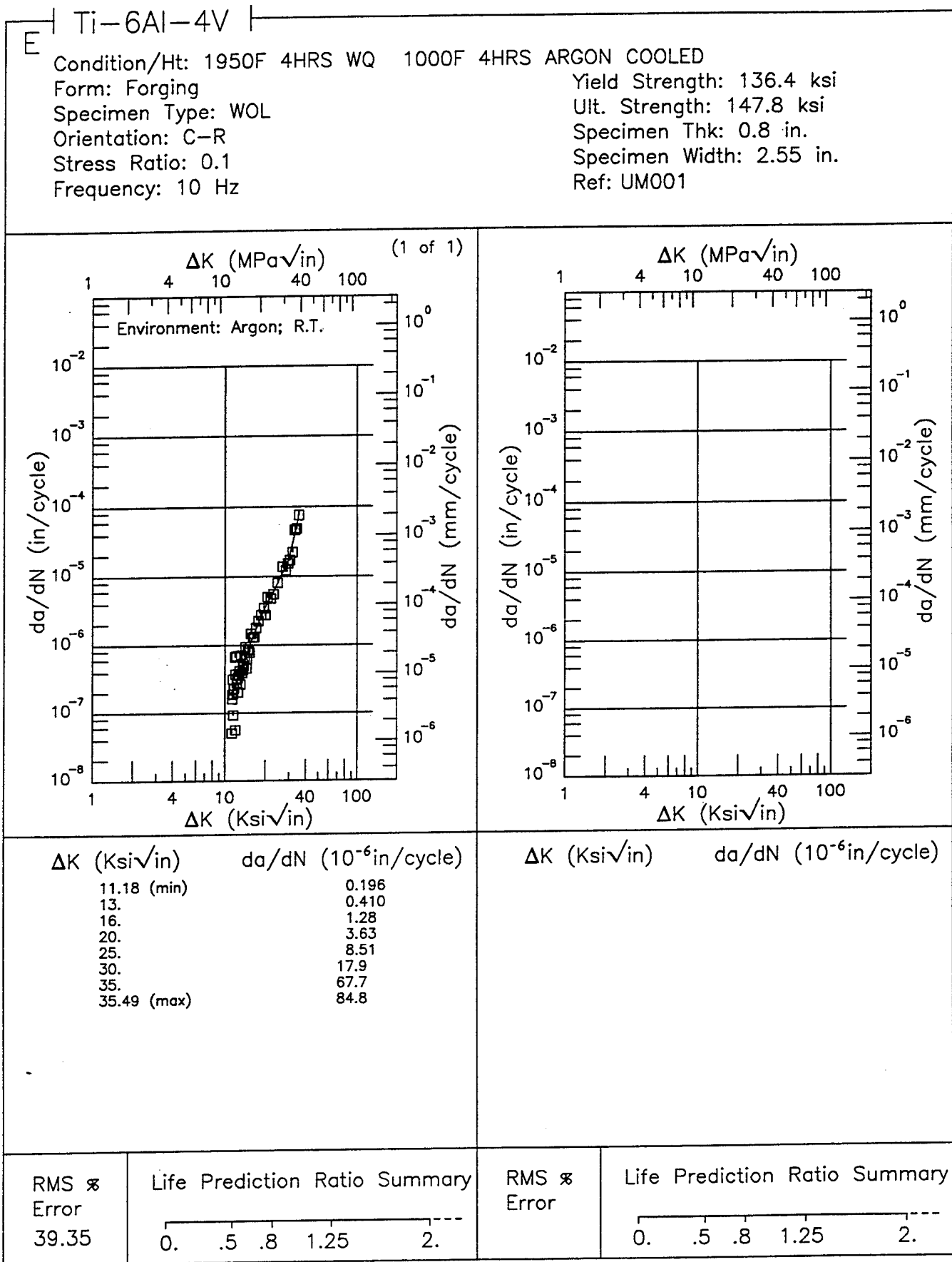


Figure 6.16.3.1.14

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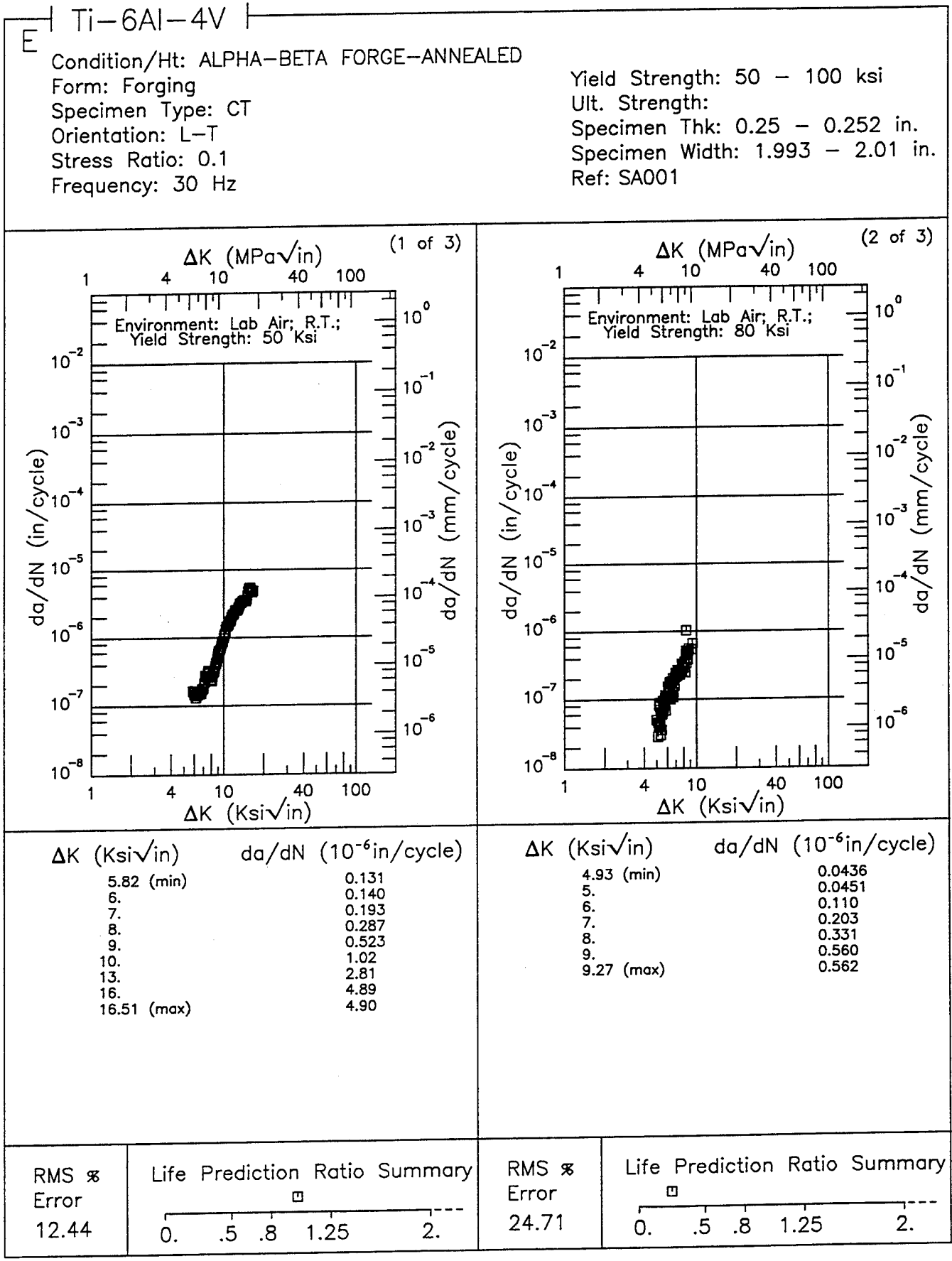


Figure 6.16.3.1.15

Ti-6Al-4V E

Condition/Ht: ALPHA-BETA FORGE-ANNEALED
 Form: Forging
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.1
 Frequency: 30 Hz

Yield Strength: 50 - 100 ksi
 Ult. Strength:
 Specimen Thk: 0.25 - 0.252 in.
 Specimen Width: 1.993 - 2.01 in.
 Ref: SA001

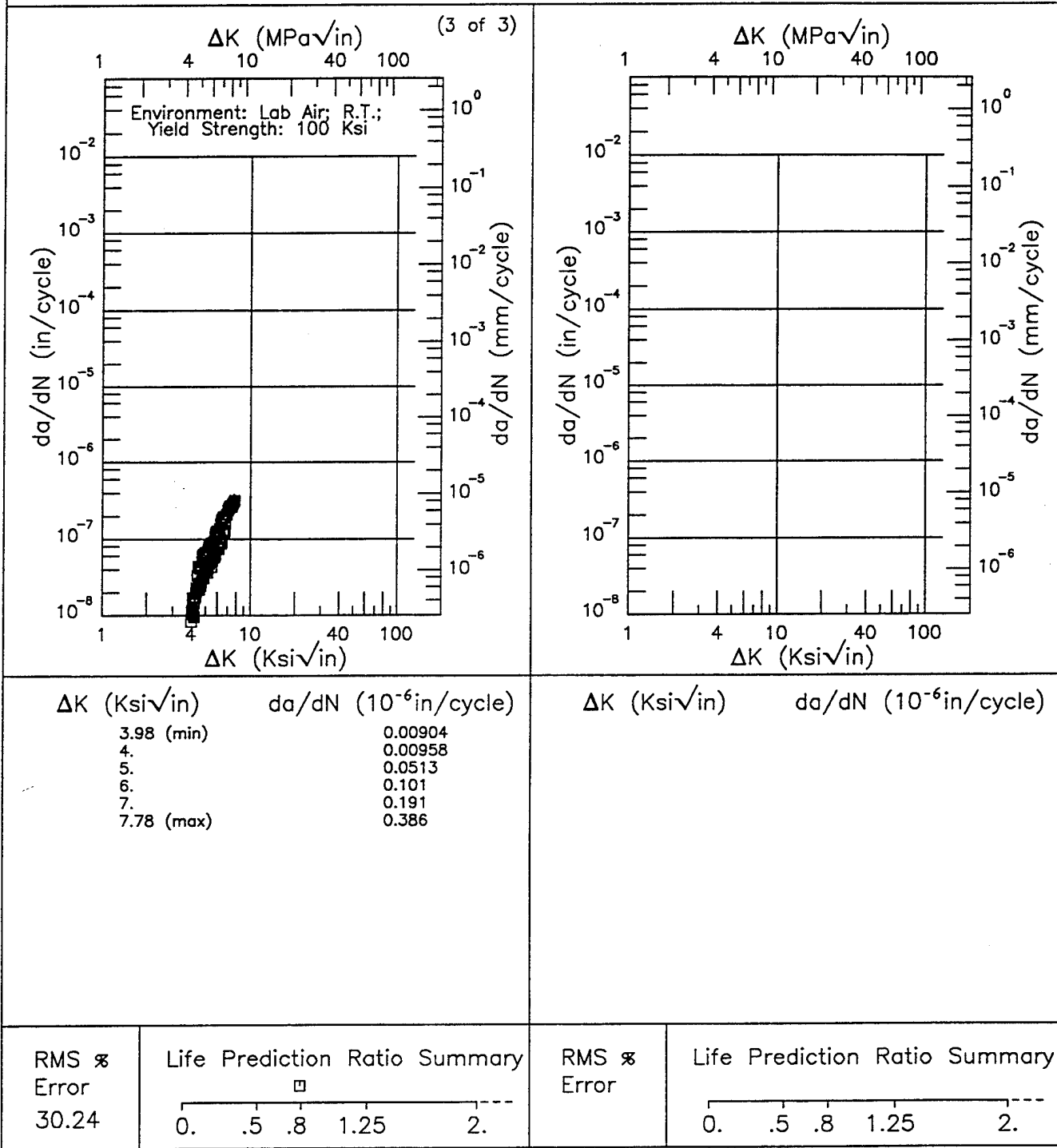


Figure 6.16.3.1.15 (Concluded)

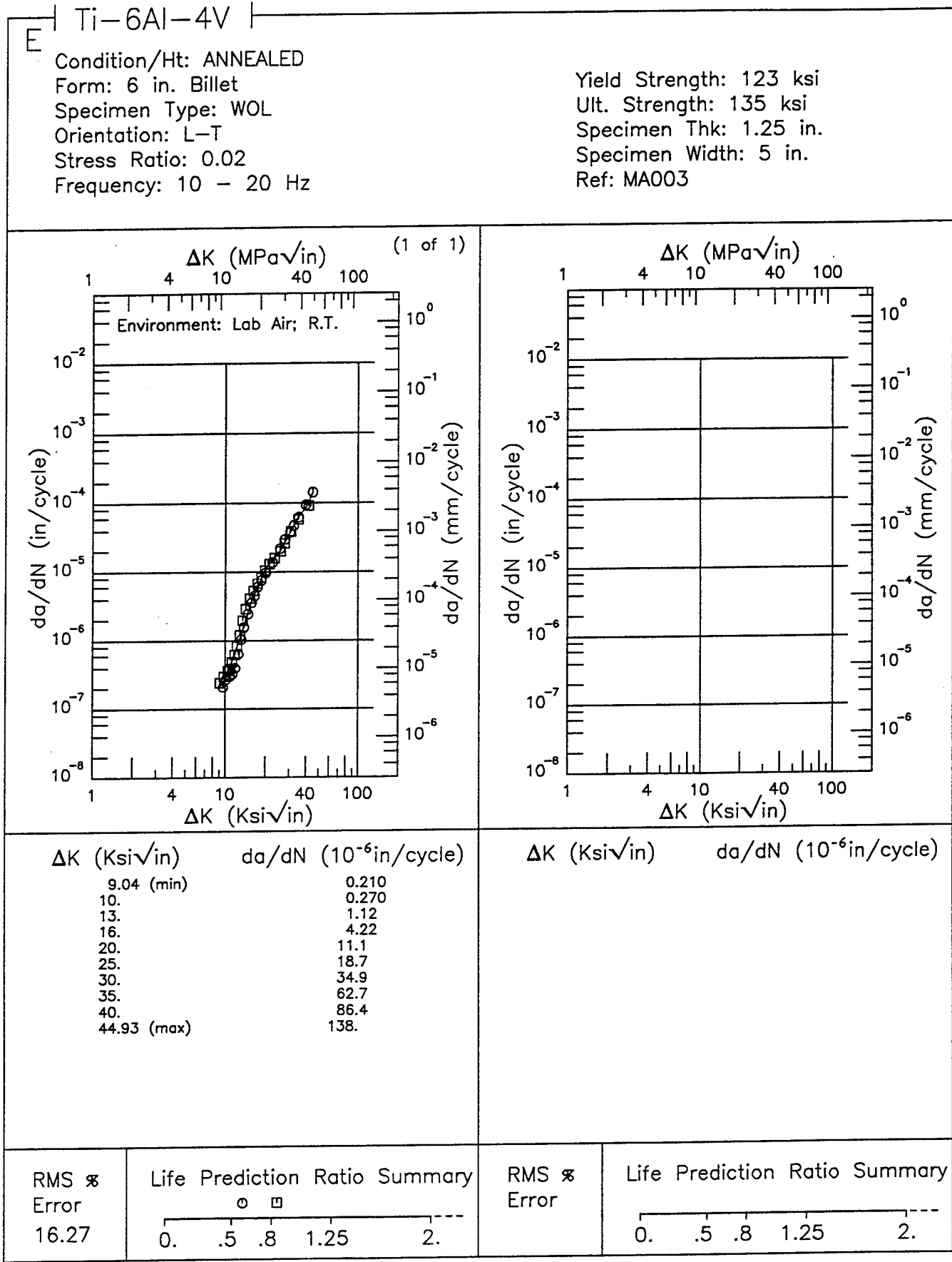
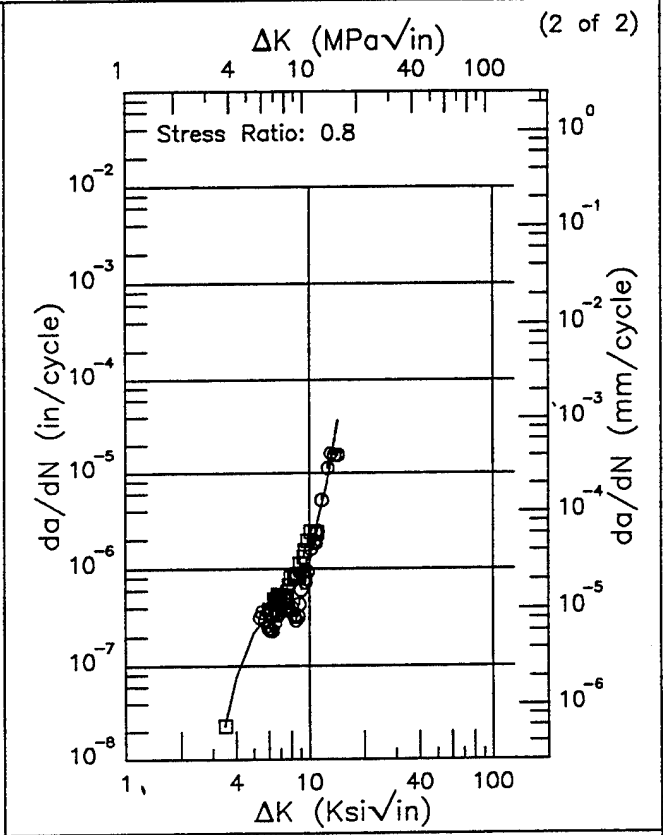
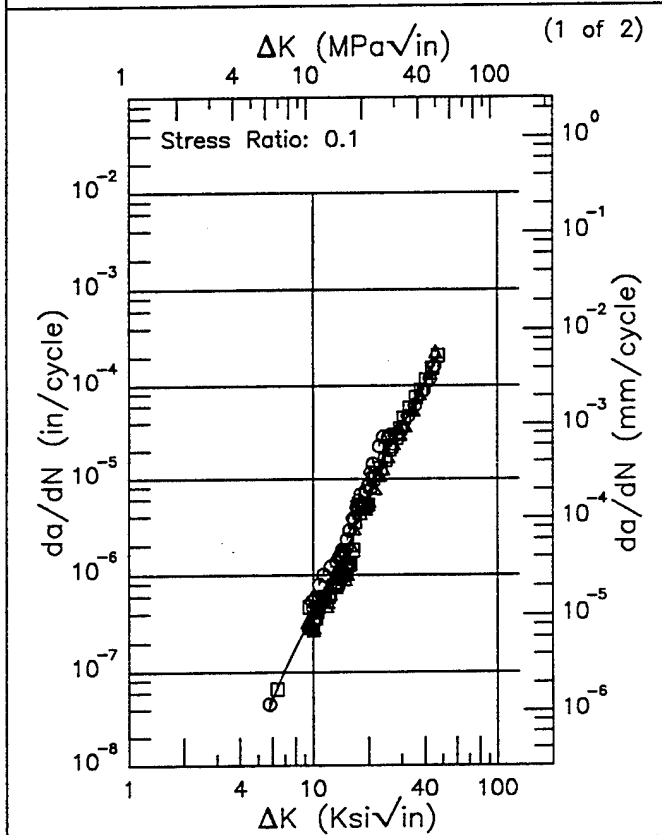


Figure 6.16.3.1.16

Ti-6Al-4V R

Condition/Ht: ANNEALED
 Form: 1.75 in. Extrusion
 Specimen Type: CT
 Orientation: L-T
 Frequency: 1 Hz
 Environment: DIST WATER; RT

Yield Strength: 127.2 - 131.6 ksi
 Ult. Strength:
 Specimen Thk: 0.247 - 0.252 in.
 Specimen Width: 2 - 2.008 in.
 Ref: DA006;DA007



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
5.77 (min)	0.0411
6.	0.0508
7.	0.106
8.	0.181
9.	0.275
10.	0.391
13.	0.969
16.	2.33
20.	7.49
25.	21.9
30.	37.9
35.	57.5
40.	102.
46.98 (max)	206.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
3.48 (min)	0.0230
3.5	0.0245
4.	0.0750
5.	0.223
6.	0.348
7.	0.453
8.	0.595
9.	0.853
10.	1.37
13.	11.6
14.23 (max)	36.8

RMS % Error	Life Prediction Ratio Summary
26.52	

RMS % Error	Life Prediction Ratio Summary
29.39	

Figure 6.16.3.1.17

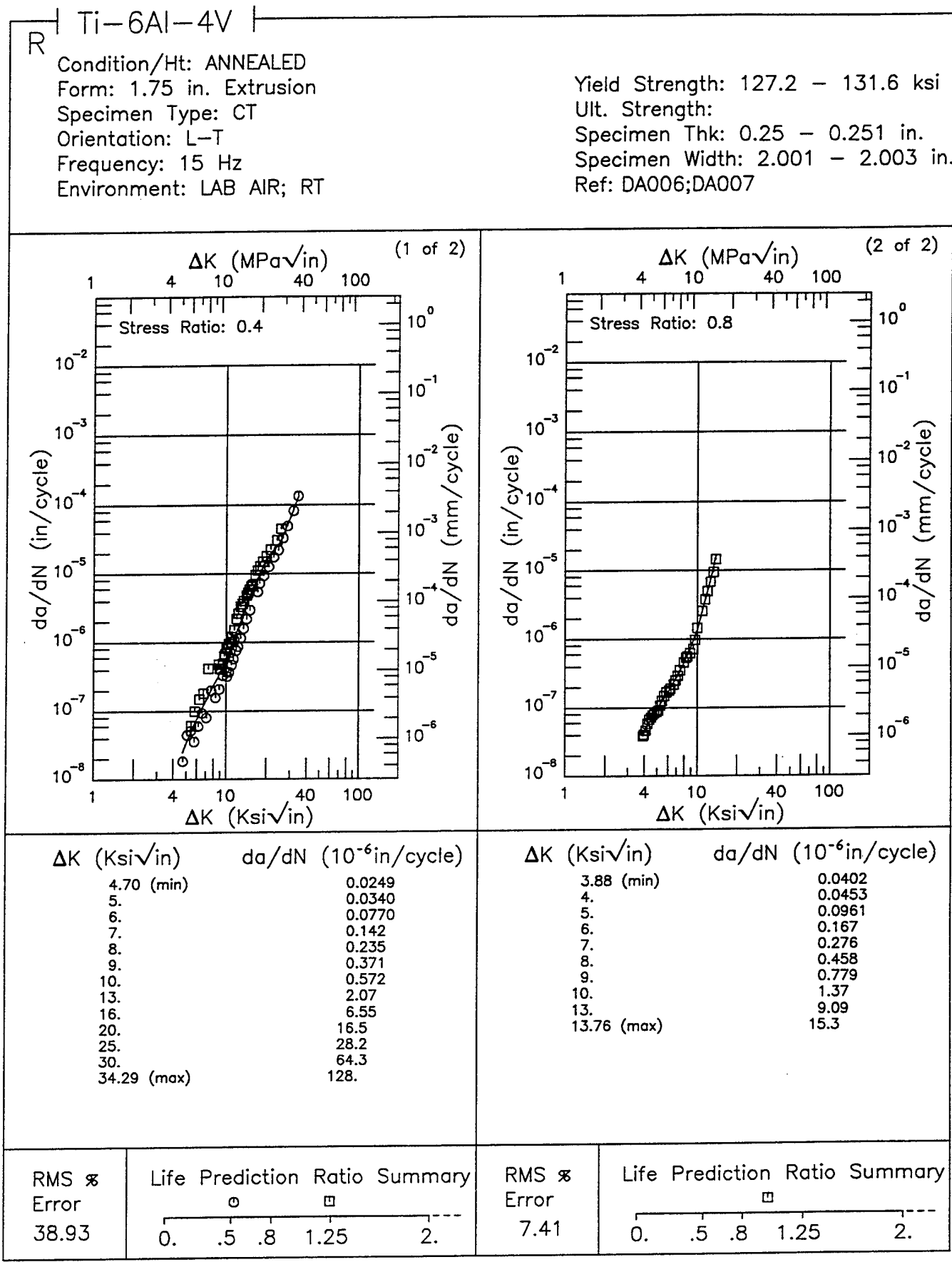


Figure 6.16.3.1.18

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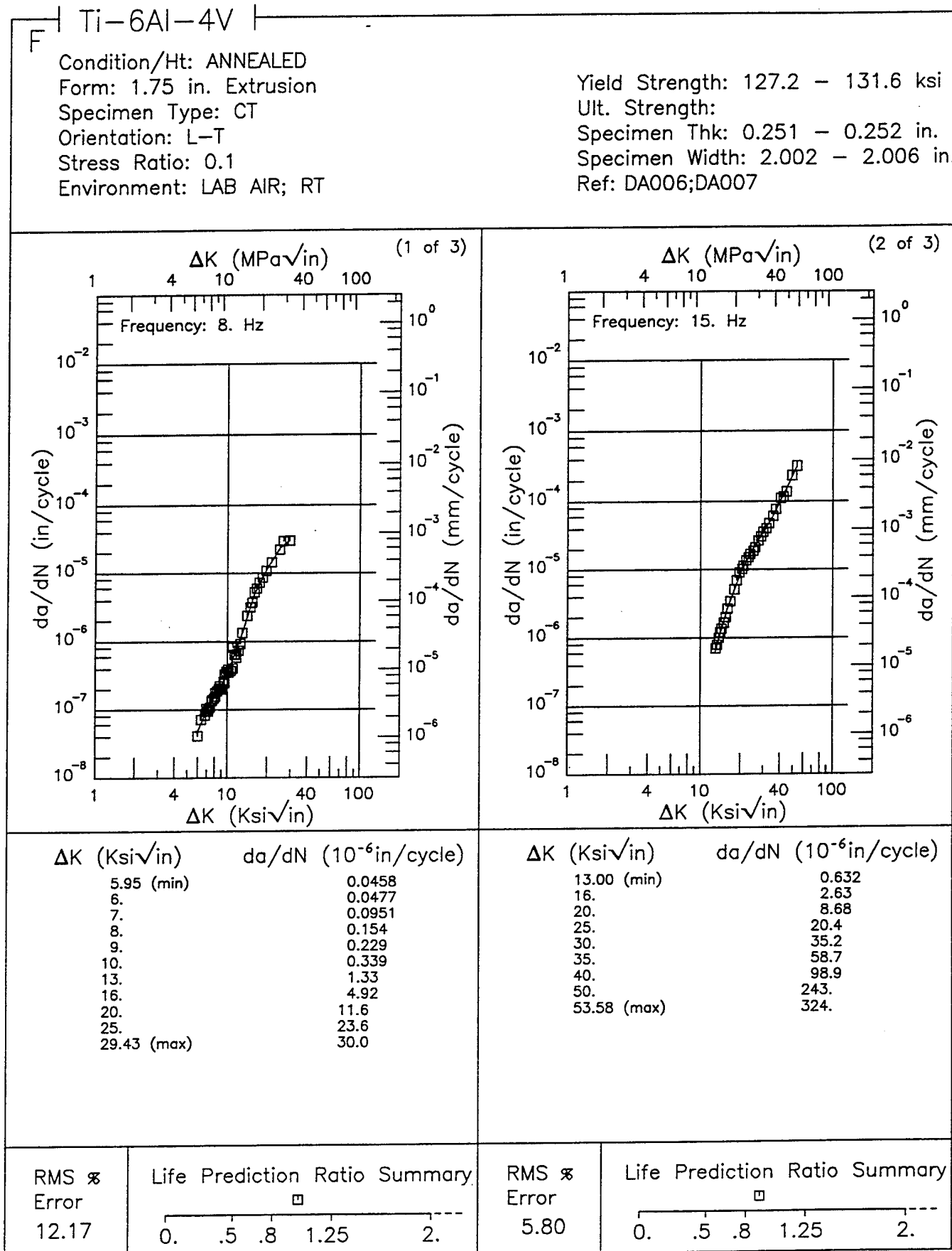
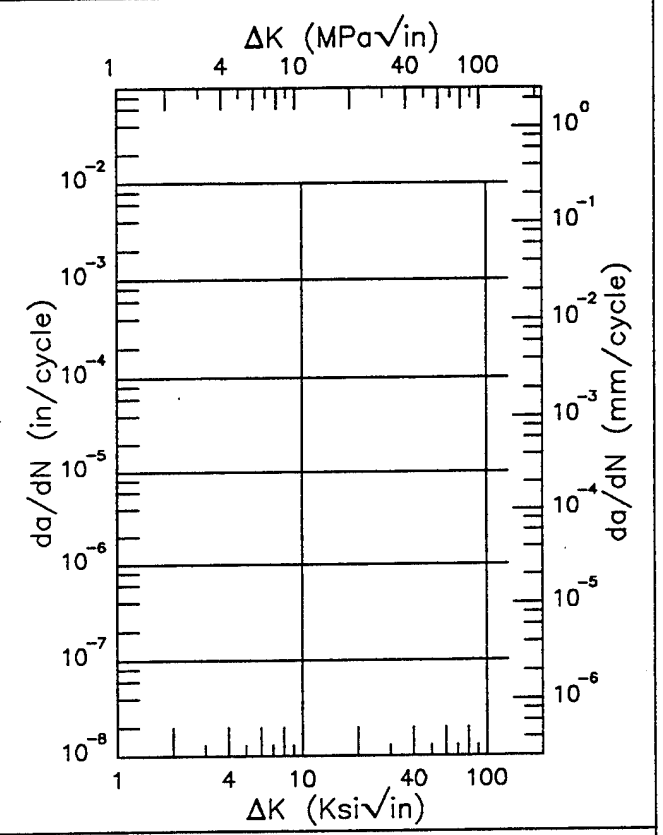
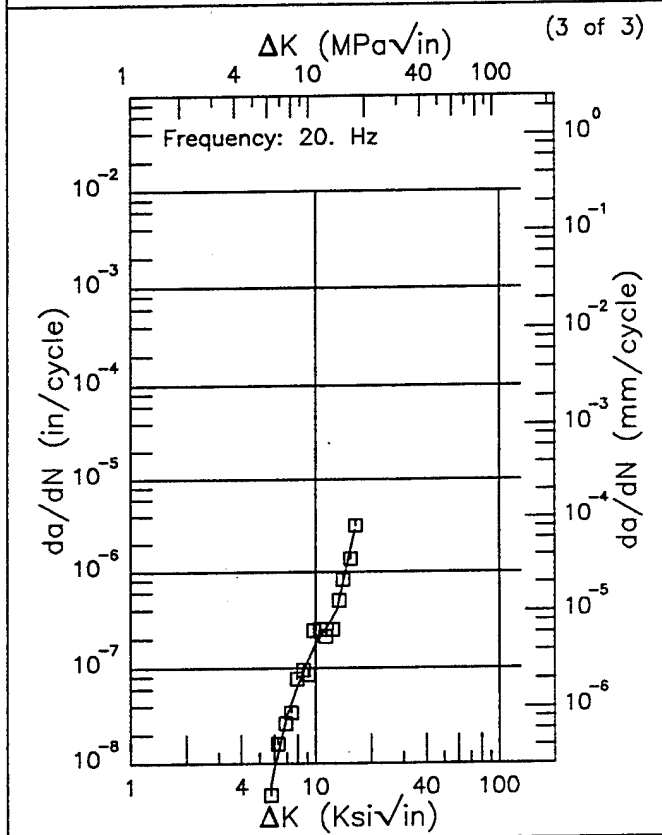


Figure 6.16.3.1.19

Ti-6Al-4V F

Condition/Ht: ANNEALED
 Form: 1.75 in. Extrusion
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.1
 Environment: LAB AIR; RT

Yield Strength: 127.2 - 131.6 ksi
 Ult. Strength:
 Specimen Thk: 0.251 - 0.252 in.
 Specimen Width: 2.002 - 2.006 in.
 Ref: DA006;DA007



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
5.74 (min)	0.00539
6.	0.00891
7.	0.0317
8.	0.0561
9.	0.116
10.	0.180
13.	0.391
16.	2.52
16.26 (max)	2.83

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)

RMS % Error	Life Prediction Ratio Summary
21.89	0. .5 .8 1.25 2.

RMS % Error	Life Prediction Ratio Summary
	0. .5 .8 1.25 2.

Figure 6.16.3.1.19 (Concluded)

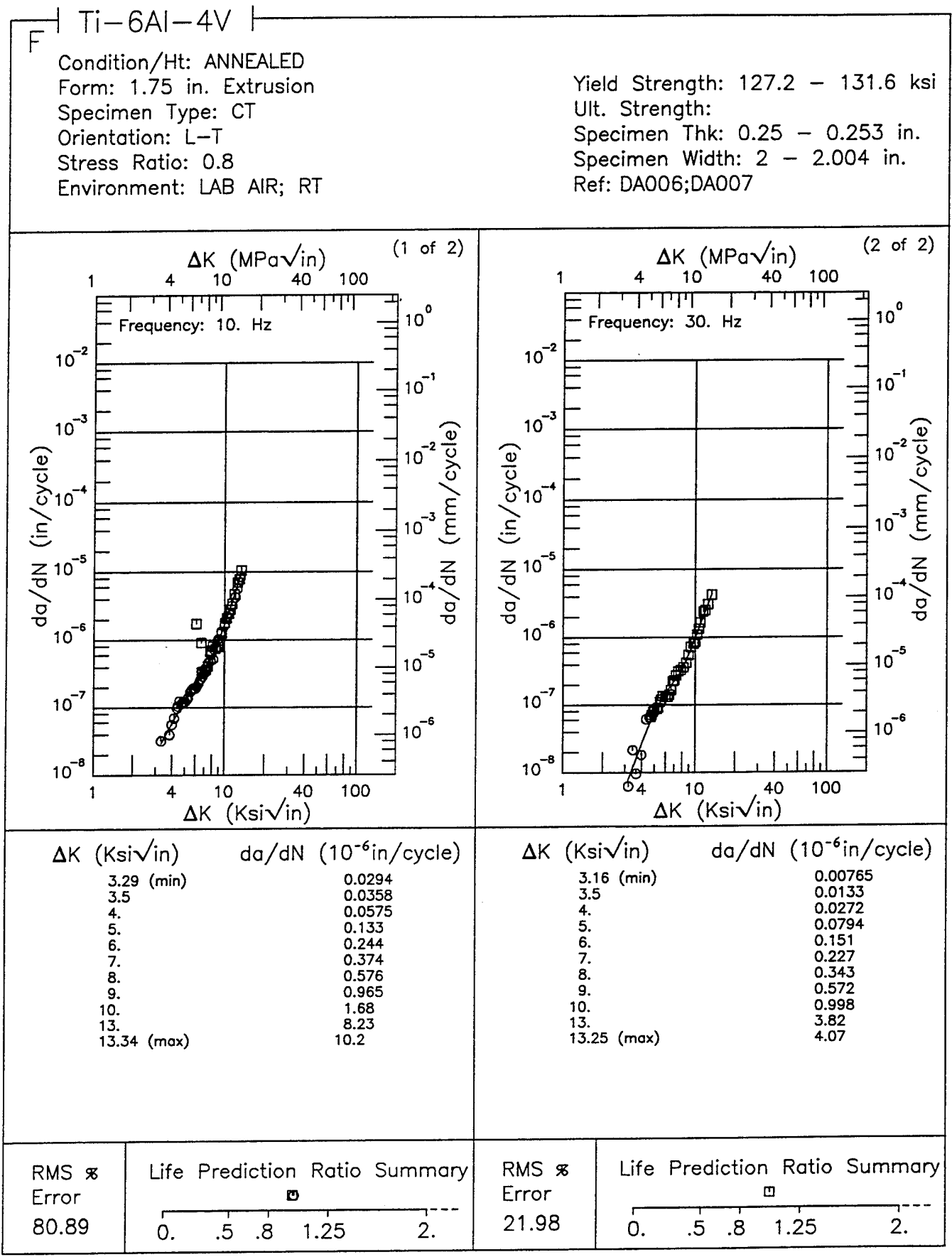


Figure 6.16.3.1.20

Ti-6Al-4V R

Condition/Ht: ANNEALED
 Form: 1.75 in. Extrusion
 Specimen Type: CT
 Orientation: T-L
 Frequency: 5 - 15 Hz
 Environment: LAB AIR; RT

Yield Strength: 131.4 ksi
 Ult. Strength:
 Specimen Thk: 0.249 - 0.25 in.
 Specimen Width: 2.002 - 2.006 in.
 Ref: DA006

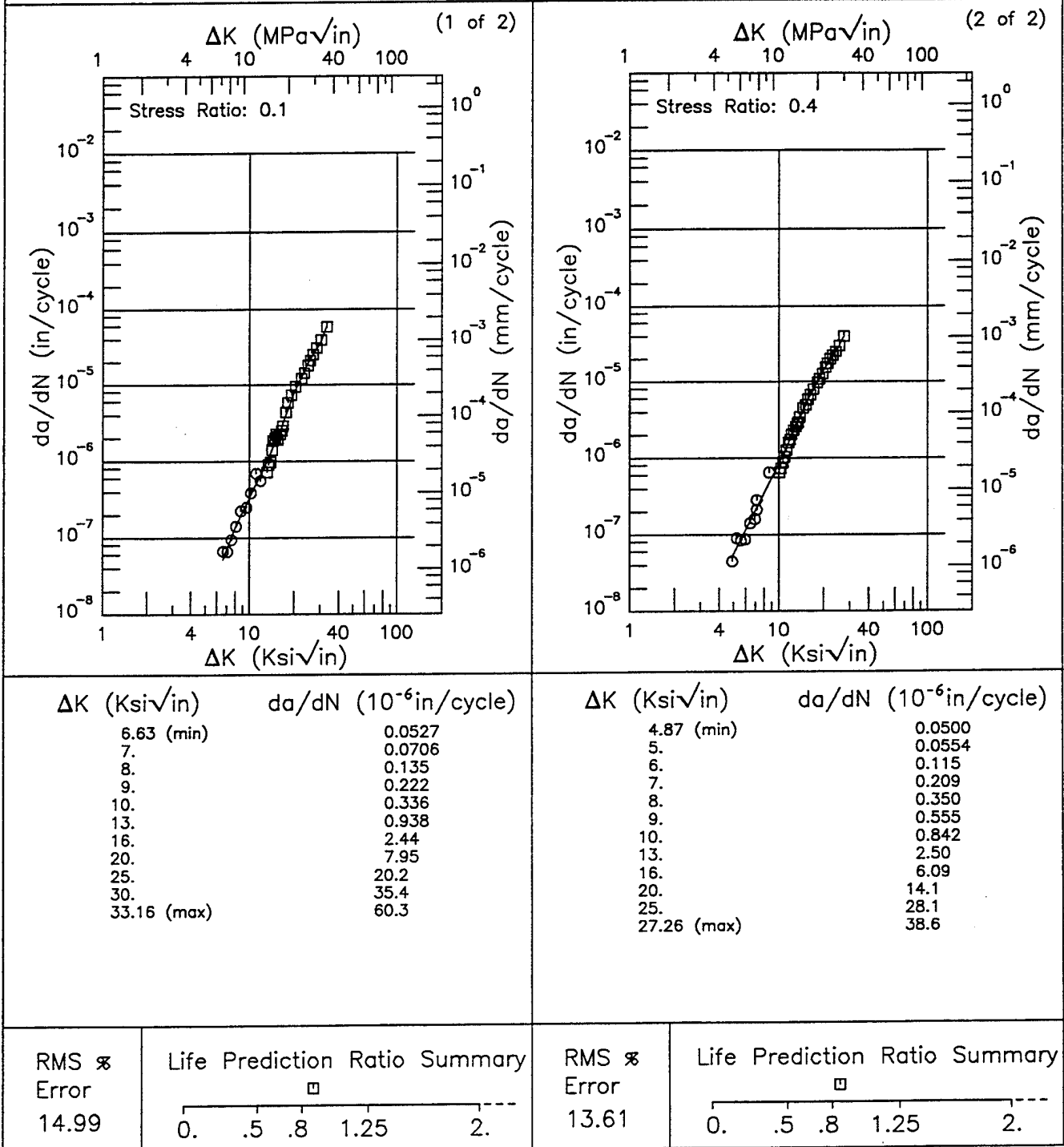


Figure 6.16.3.1.21

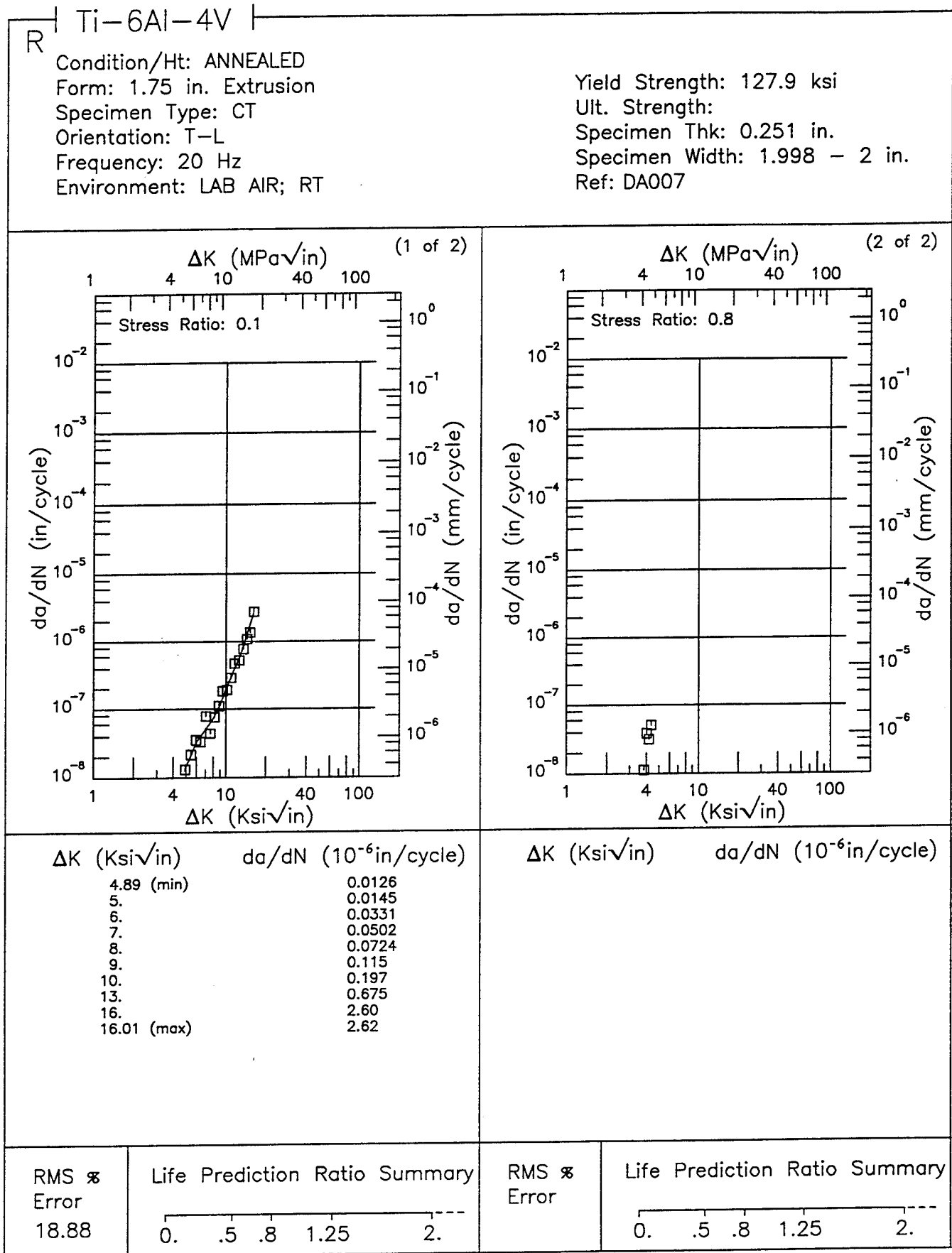
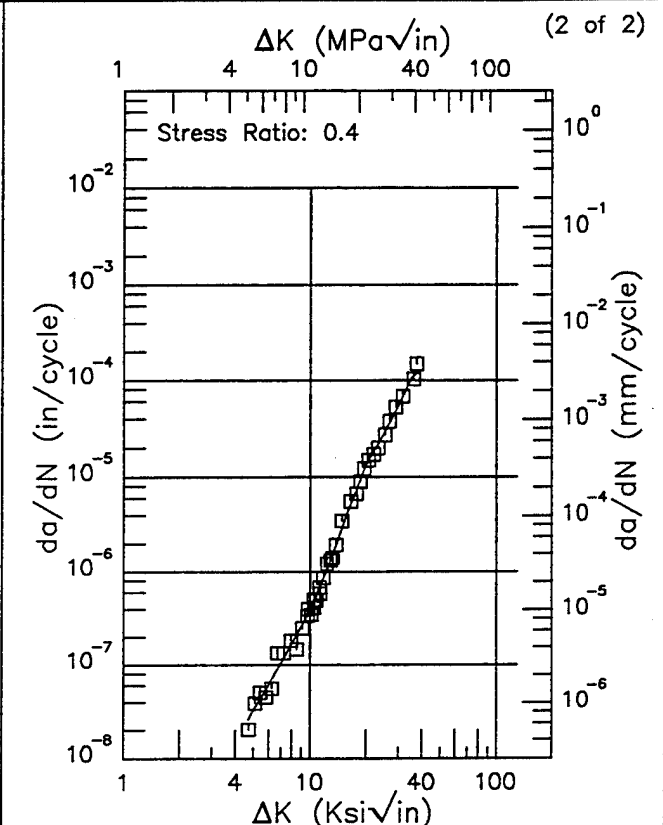
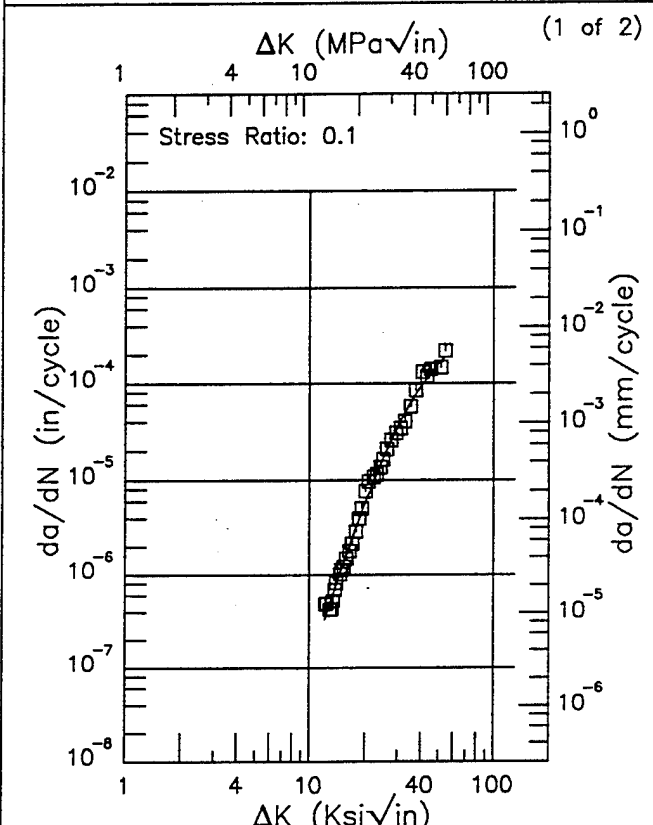


Figure 6.16.3.1.22

Ti-6Al-4V R

Condition/Ht: ANNEALED
 Form: 1.75 in. Extrusion
 Specimen Type: CT
 Orientation: T-L
 Frequency: 15 Hz
 Environment: LAB AIR; RT

Yield Strength: 127.9 ksi
 Ult. Strength:
 Specimen Thk: 0.25 - 0.251 in.
 Specimen Width: 2.002 - 2.003 in.
 Ref: DA007



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
12.19 (min)	0.330
13.	0.498
16.	1.74
20.	5.84
25.	16.9
30.	35.7
35.	61.9
40.	93.4
50.	162.
54.07 (max)	188.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
4.69 (min)	0.0257
5.	0.0323
6.	0.0613
7.	0.104
8.	0.167
9.	0.260
10.	0.398
13.	1.37
16.	4.34
20.	13.4
25.	28.1
30.	54.2
35.	105.
37.43 (max)	119.

RMS % Error	Life Prediction Ratio Summary
16.54	

RMS % Error	Life Prediction Ratio Summary
15.11	

Figure 6.16.3.1.23

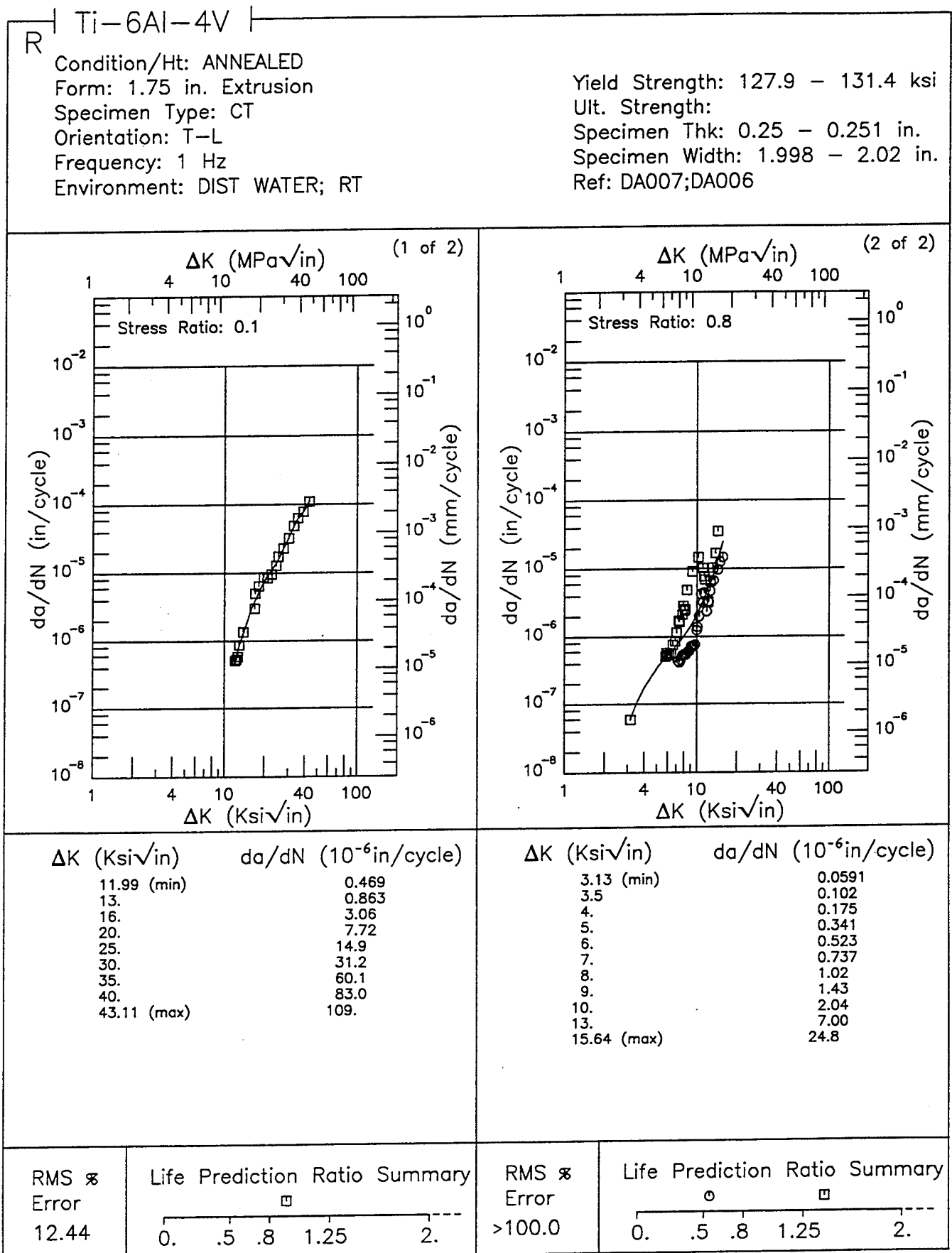


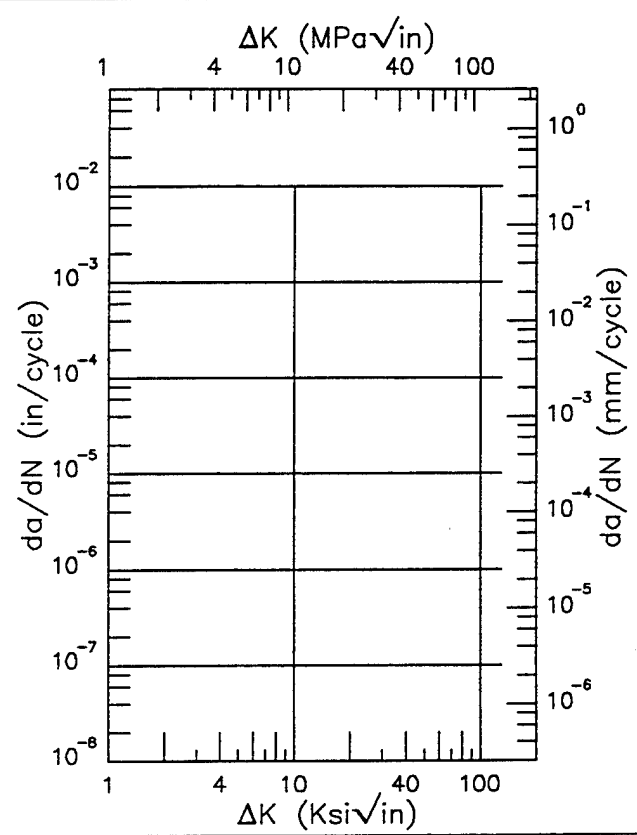
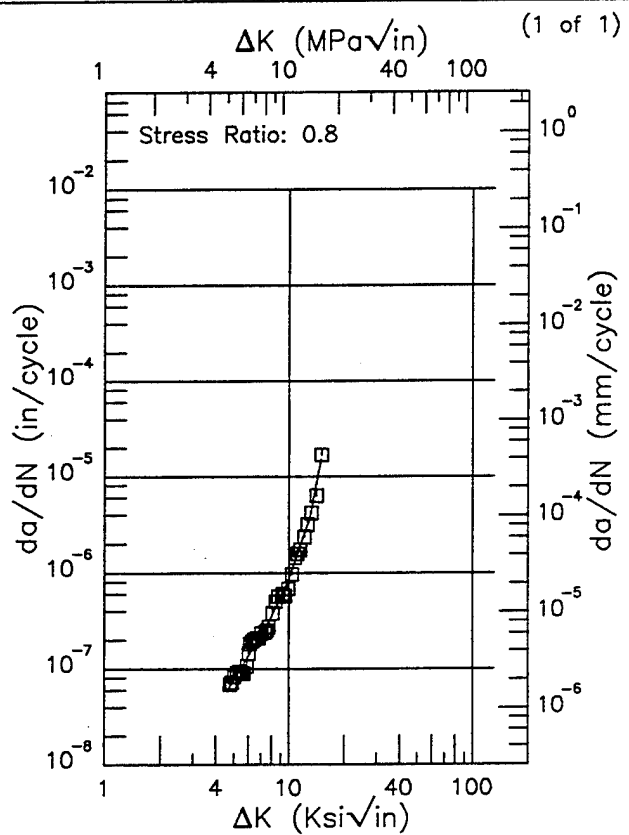
Figure 6.16.3.1.24

Ti-6Al-4V

R

Condition/Ht: ANNEALED
 Form: 1.75 in. Extrusion
 Specimen Type: CT
 Orientation: T-L
 Frequency: 30 Hz
 Environment: LAB AIR; RT

Yield Strength: 127.9 ksi
 Ult. Strength:
 Specimen Thk: 0.251 in.
 Specimen Width: 2 in.
 Ref: DA007



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
4.75 (min)	0.0646
5.	0.0765
6.	0.135
7.	0.221
8.	0.350
9.	0.552
10.	0.872
13.	3.62
14.97 (max)	15.2

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

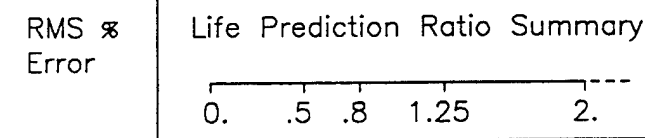
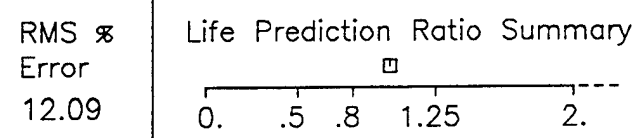


Figure 6.16.3.1.25

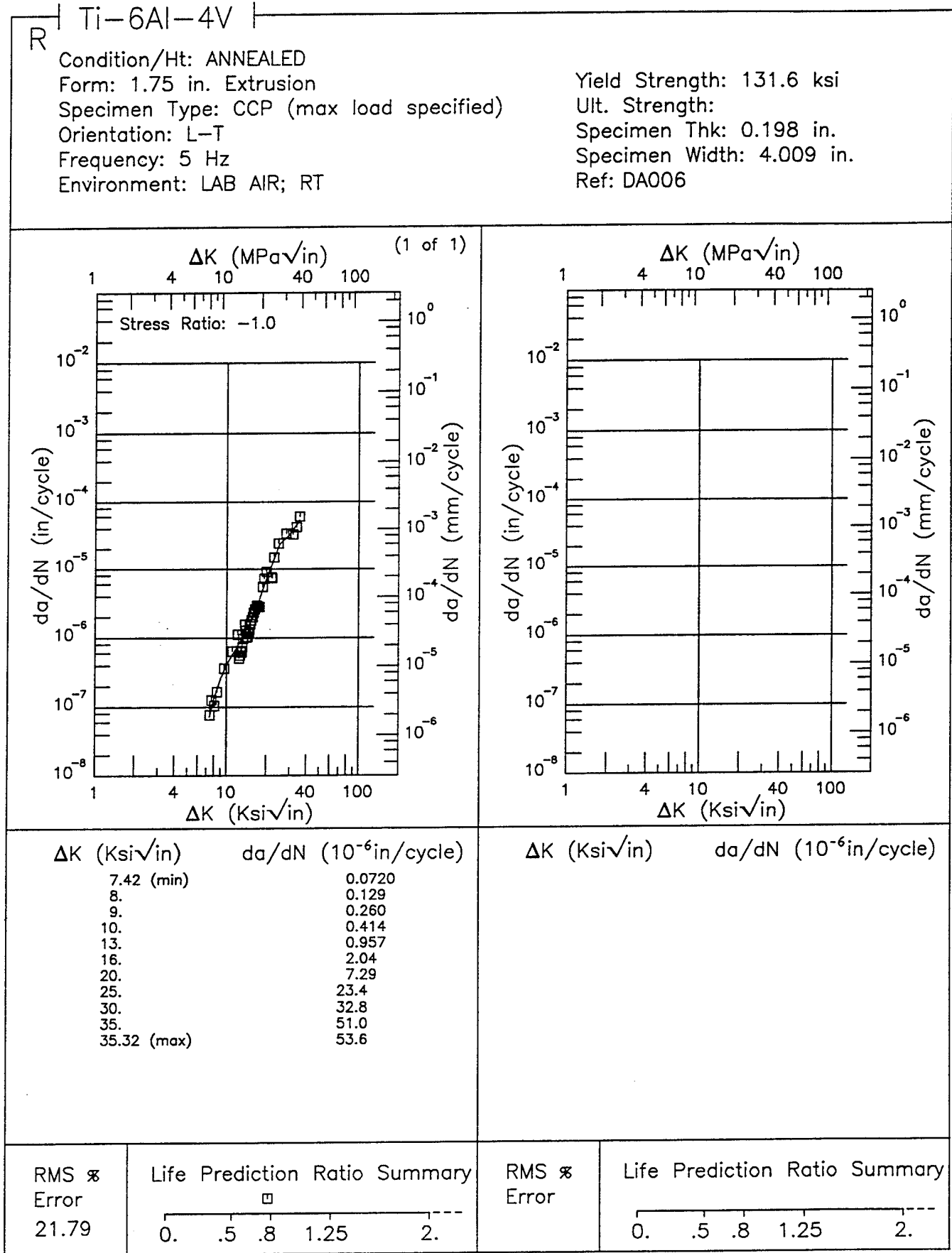
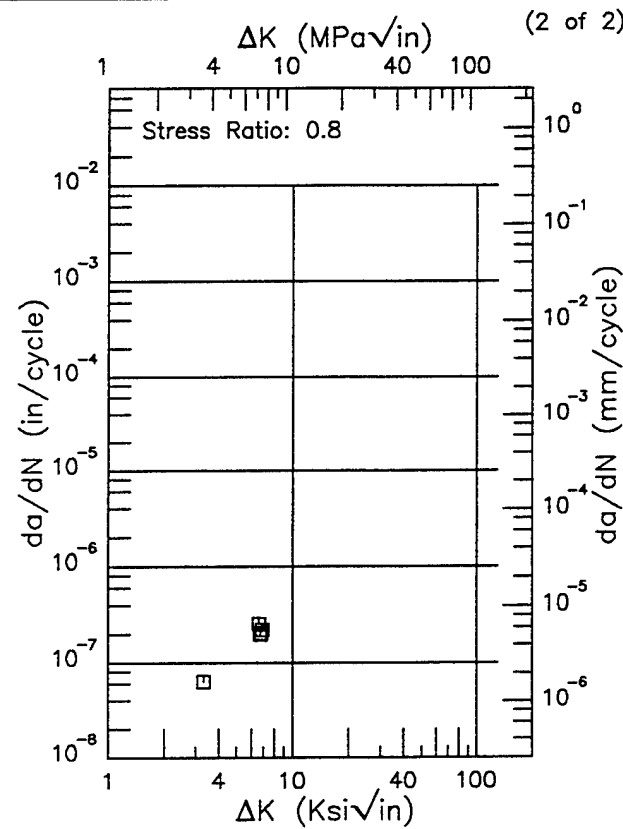
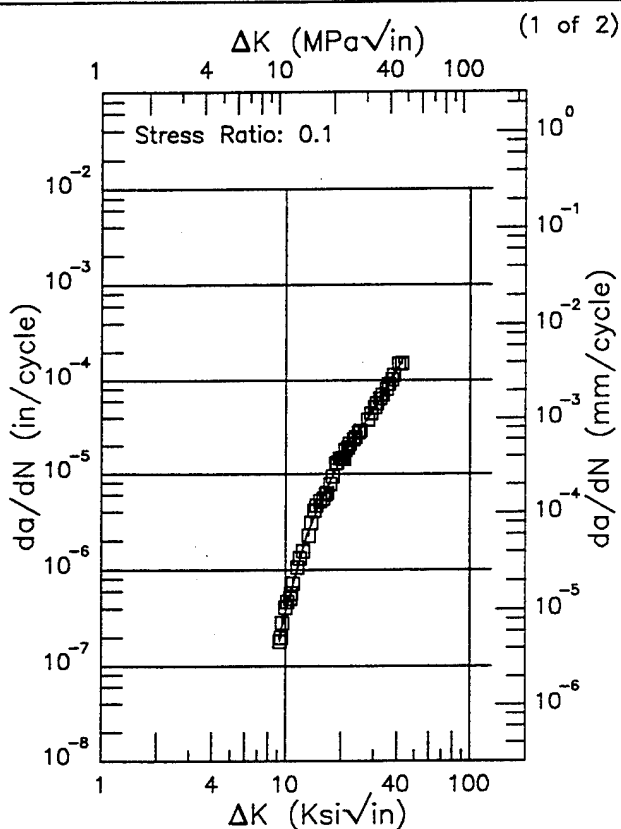


Figure 6.16.3.1.26

Ti-6Al-4V R

Condition/Ht: ANNEALED
 Form: 3 in. Forging
 Specimen Type: CT
 Orientation: L-T
 Frequency: 1 Hz
 Environment: DIST WATER; RT

Yield Strength: 131.7 ksi
 Ult. Strength:
 Specimen Thk: 0.245 - 0.246 in.
 Specimen Width: 2.007 - 2.009 in.
 Ref: DA006



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
9.25 (min)	0.196
10.	0.375
13.	2.19
16.	6.12
20.	14.3
25.	28.7
30.	49.0
35.	81.0
40.	130.
42.62 (max)	159.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
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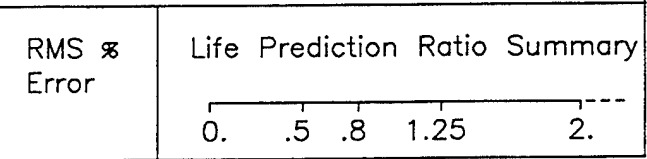
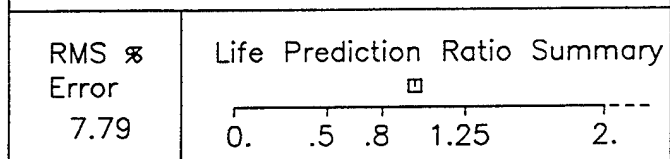


Figure 6.16.3.1.27

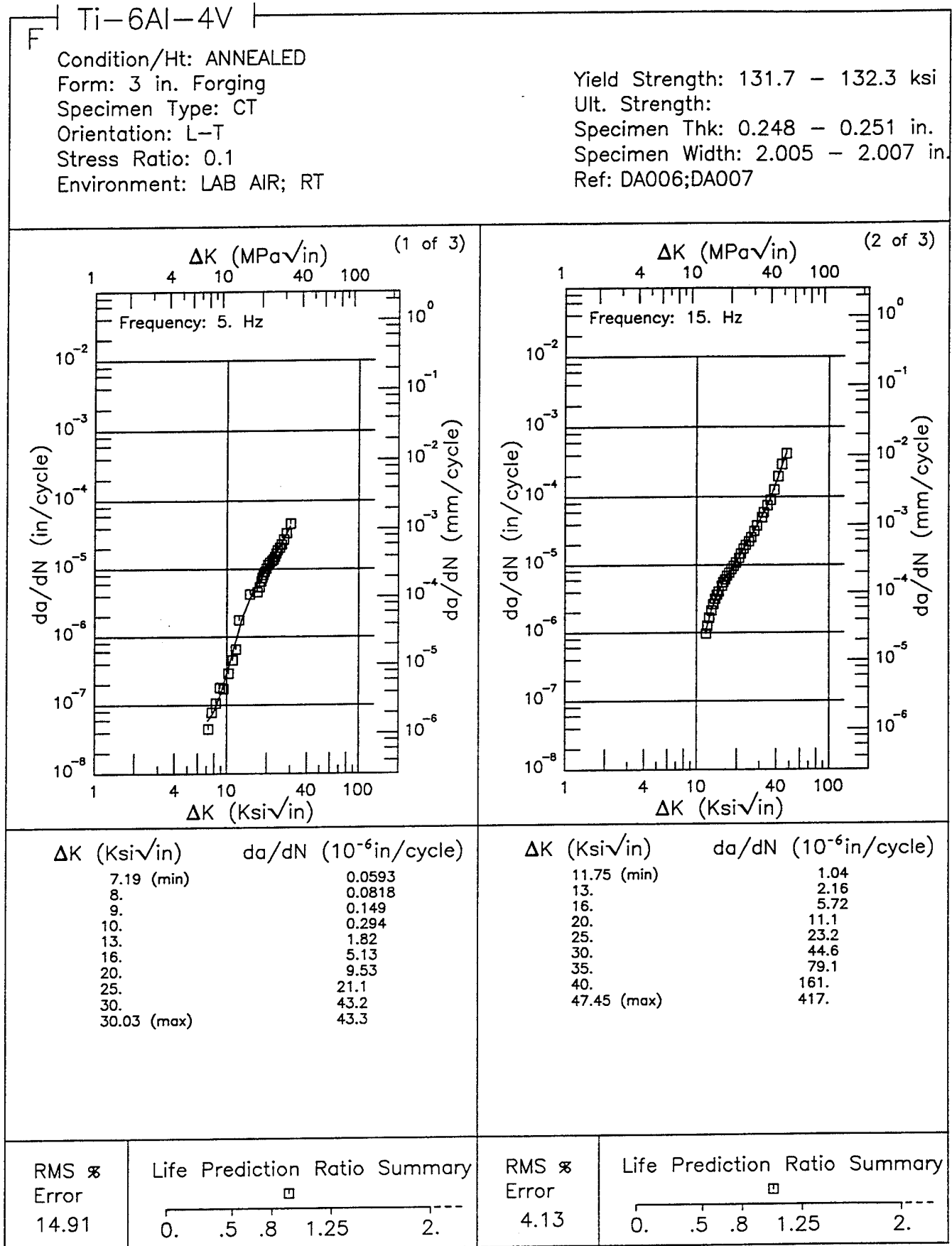
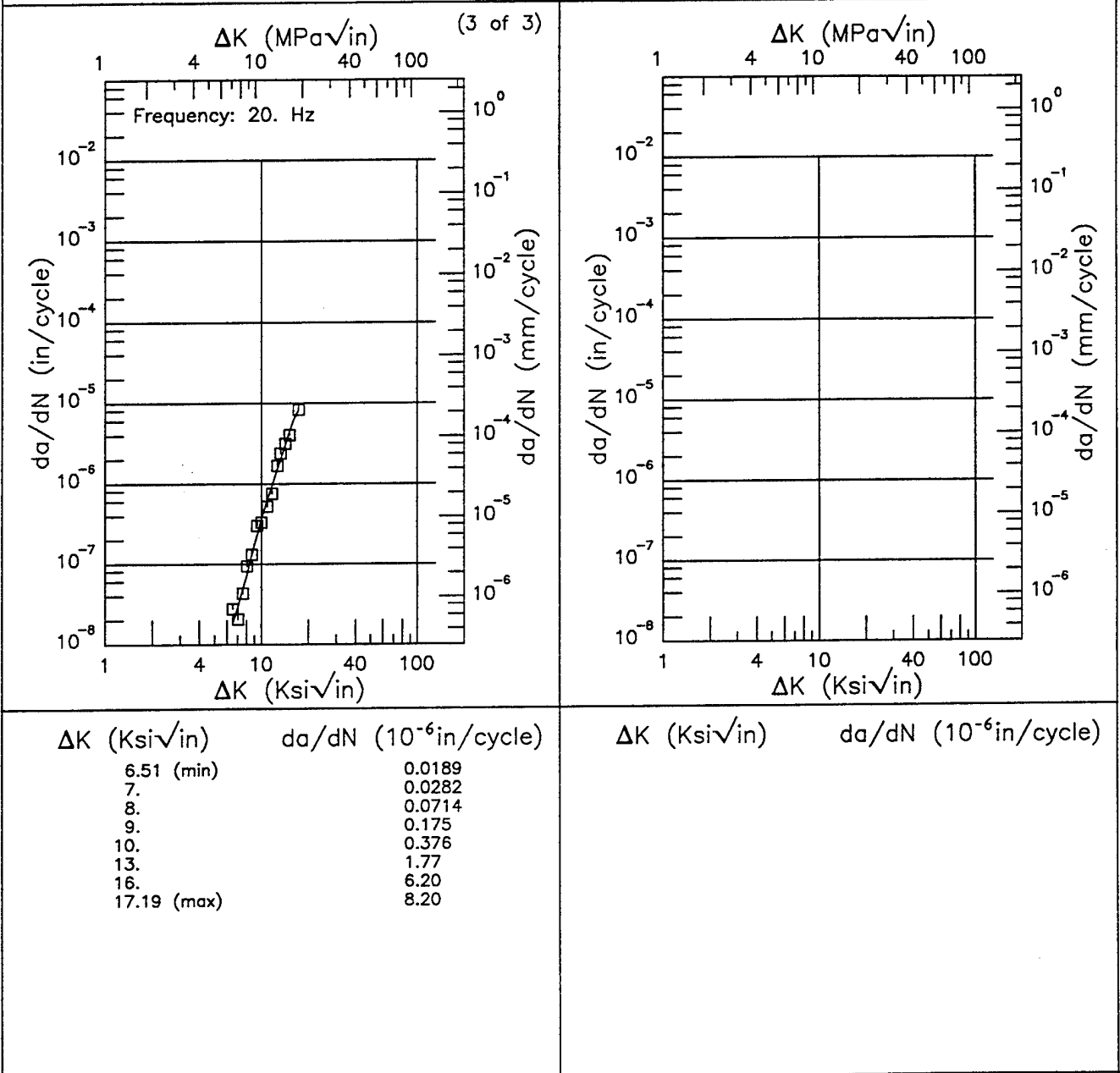


Figure 6.16.3.1.28

Ti-6Al-4V F

Condition/Ht: ANNEALED
 Form: 3 in. Forging
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.1
 Environment: LAB AIR; RT

Yield Strength: 131.7 - 132.3 ksi
 Ult. Strength:
 Specimen Thk: 0.248 - 0.251 in.
 Specimen Width: 2.005 - 2.007 in.
 Ref: DA006;DA007



RMS % Error 20.95	Life Prediction Ratio Summary 0. .5 .8 1.25 2.	RMS % Error	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
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Figure 6.16.3.1.28 (Concluded)

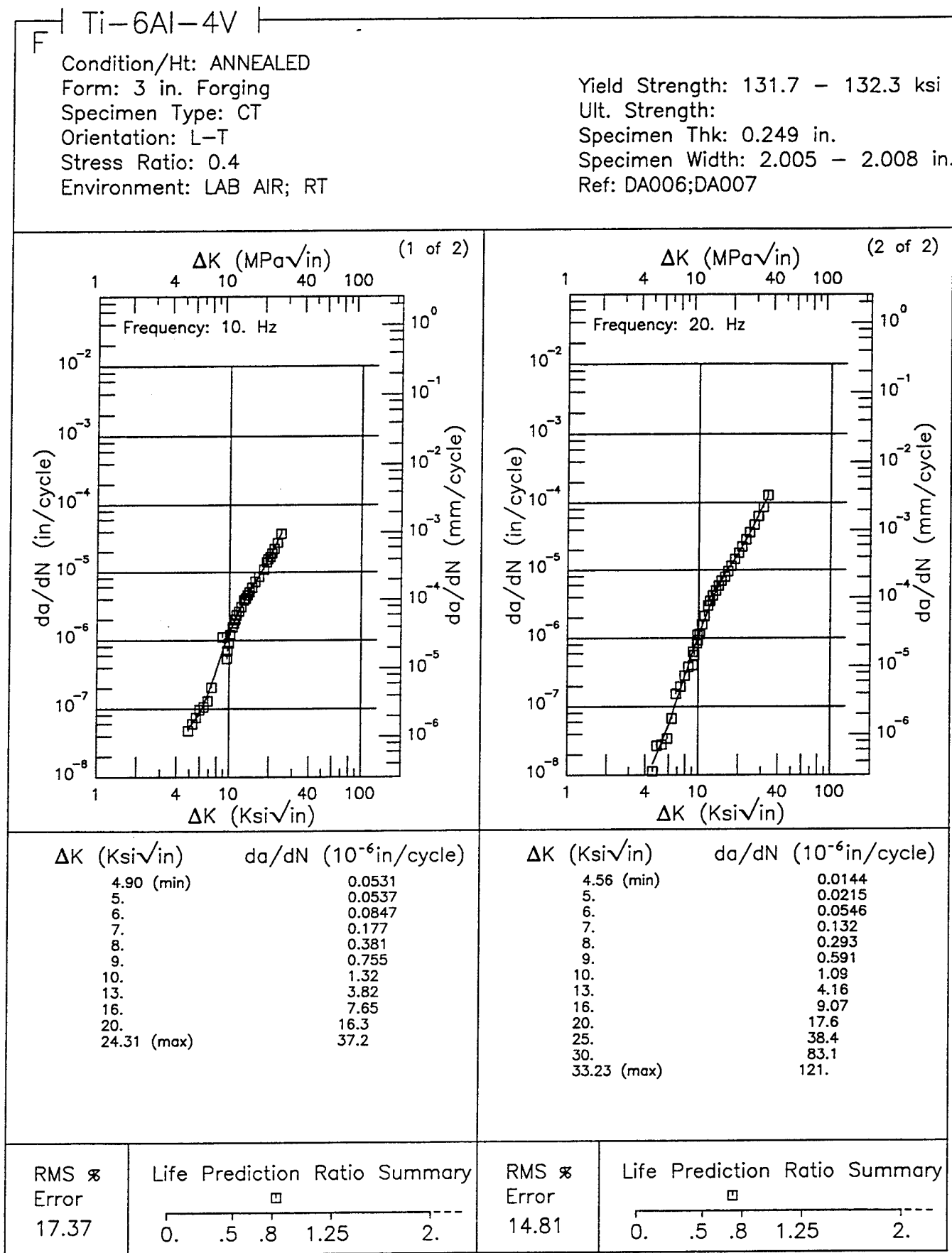


Figure 6.16.3.1.29

Ti-6Al-4V F

Condition/Ht: ANNEALED
 Form: 3 in. Forging
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.8
 Environment: LAB AIR; RT

Yield Strength: 131.7 - 132.3 ksi
 Ult. Strength:
 Specimen Thk: 0.248 - 0.249 in.
 Specimen Width: 2.005 - 2.006 in.
 Ref: DA006;DA007

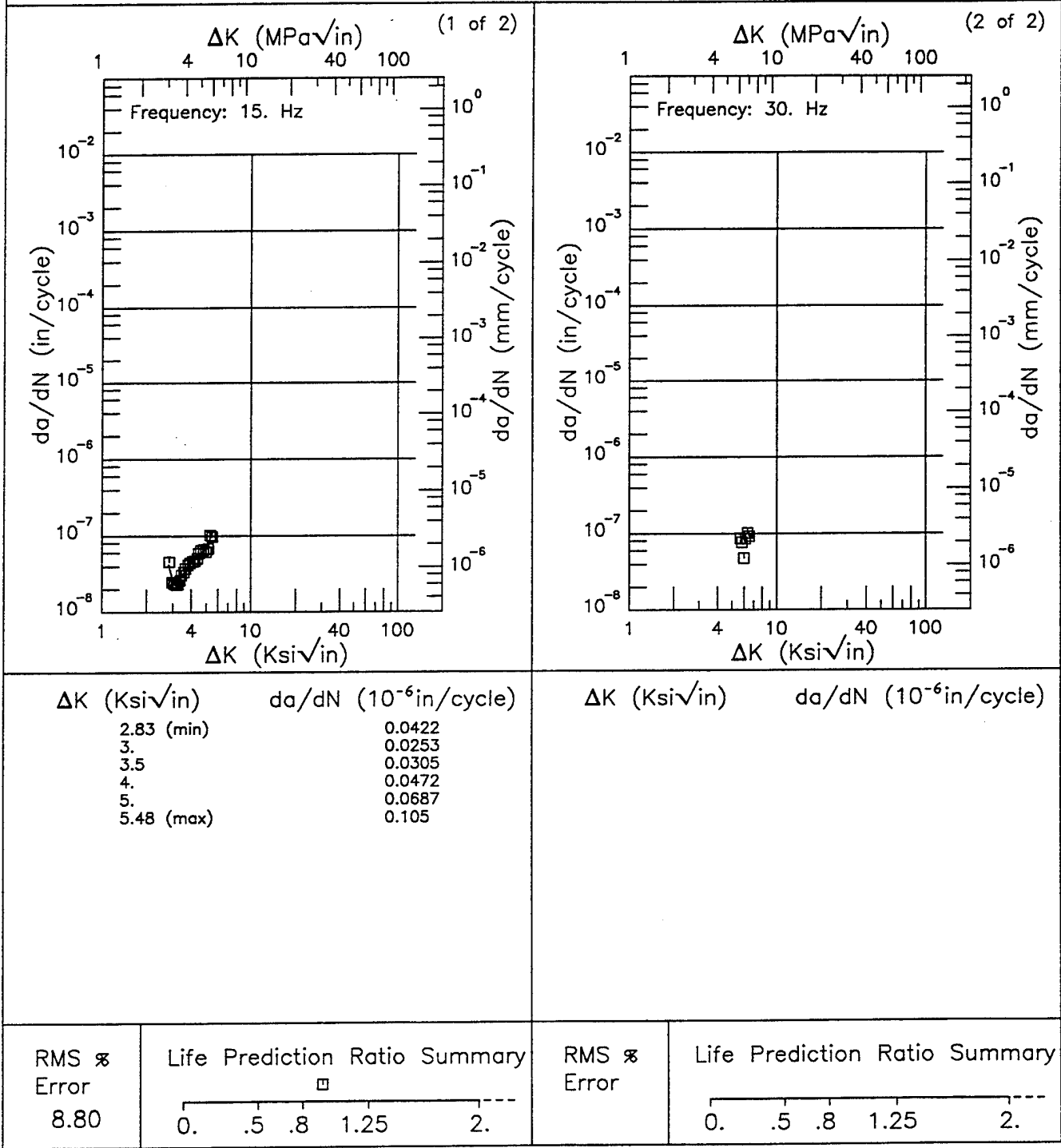


Figure 6.16.3.1.30

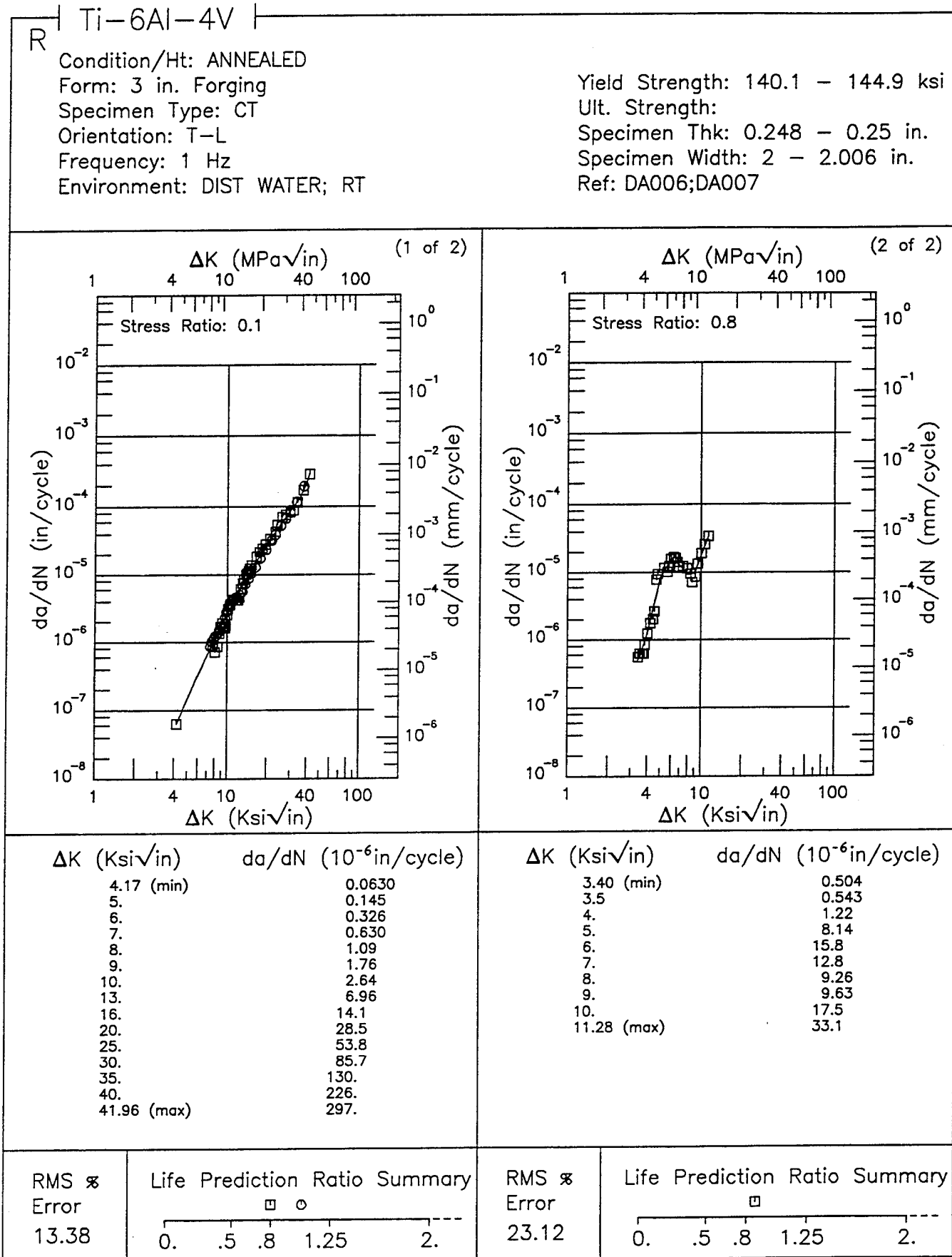


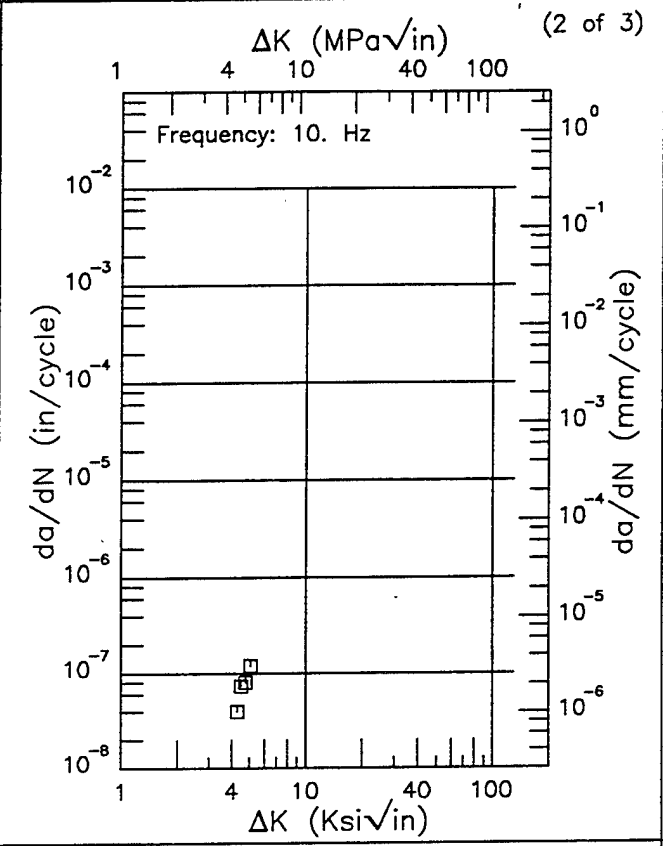
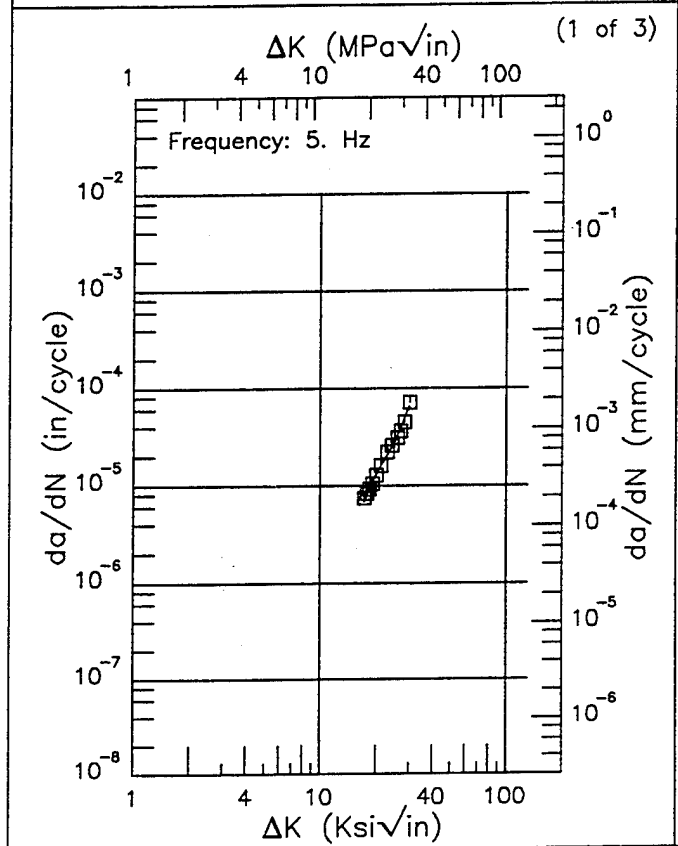
Figure 6.16.3.1.31

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F | Ti-6Al-4V |

Condition/Ht: ANNEALED
 Form: 3 in. Forging
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1
 Environment: LAB AIR; RT

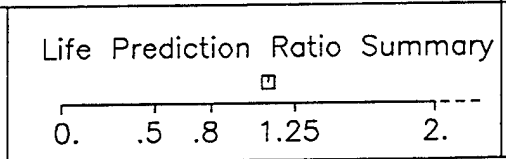
Yield Strength: 140.1 - 144.9 ksi
 Ult. Strength:
 Specimen Thk: 0.249 - 0.251 in.
 Specimen Width: 2.003 - 2.008 in.
 Ref: DA006;DA007



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
17.23 (min)	6.86
20.	13.1
25.	27.8
29.98 (max)	65.1

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
4	10
5	15
6	20
7	25
8	30
9	35
10	40

RMS % Error
5.66



RMS % Error

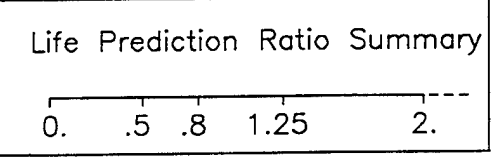
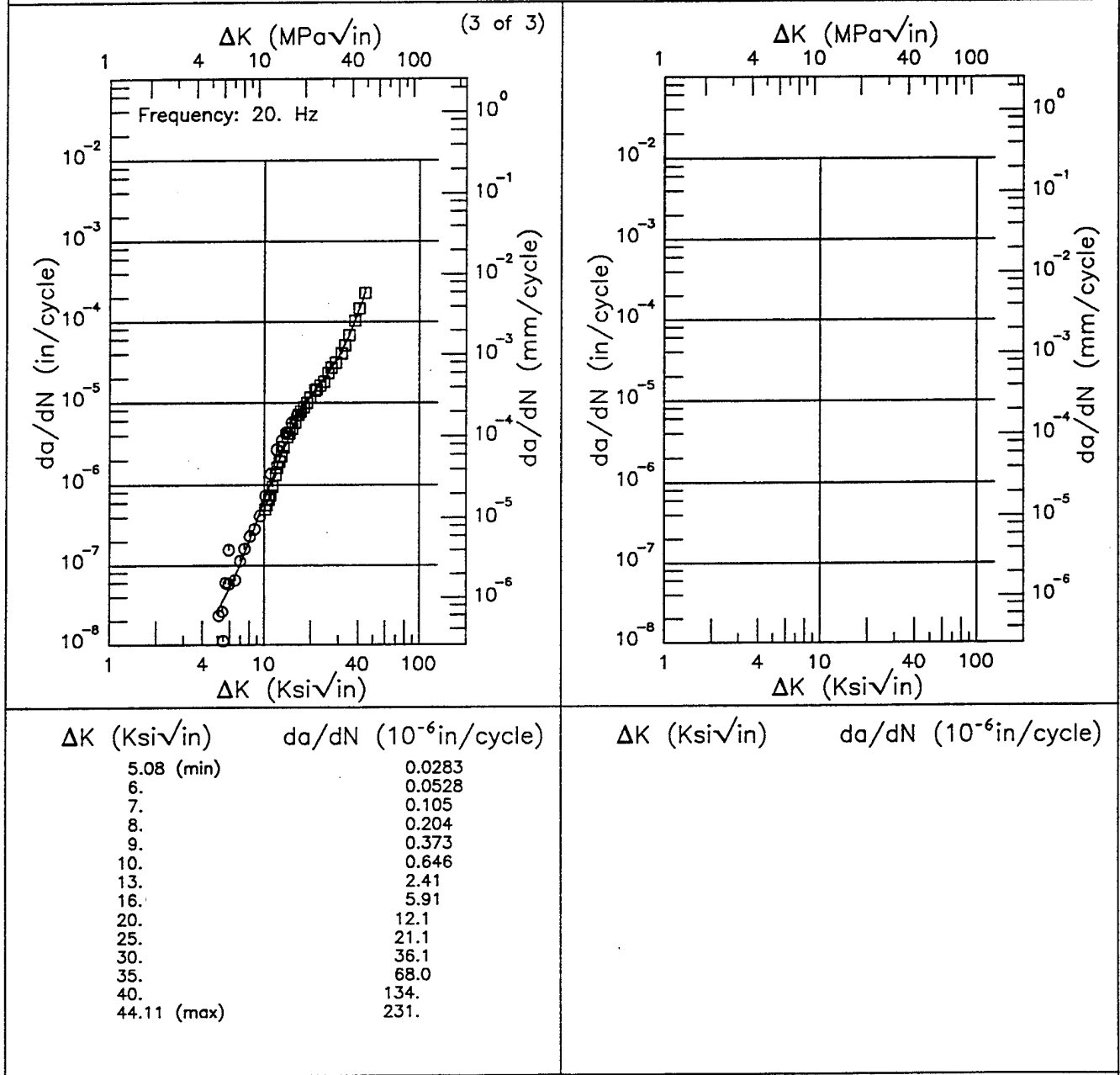


Figure 6.16.3.1.32

Ti-6Al-4V F

Condition/Ht: ANNEALED
 Form: 3 in. Forging
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1
 Environment: LAB AIR; RT

Yield Strength: 140.1 - 144.9 ksi
 Ult. Strength:
 Specimen Thk: 0.249 - 0.251 in.
 Specimen Width: 2.003 - 2.008 in.
 Ref: DA006;DA007



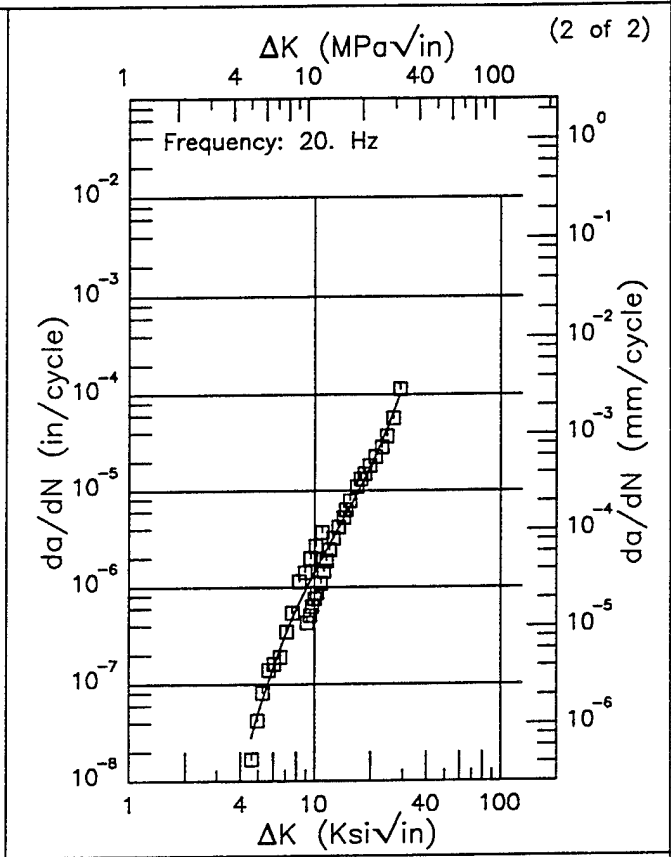
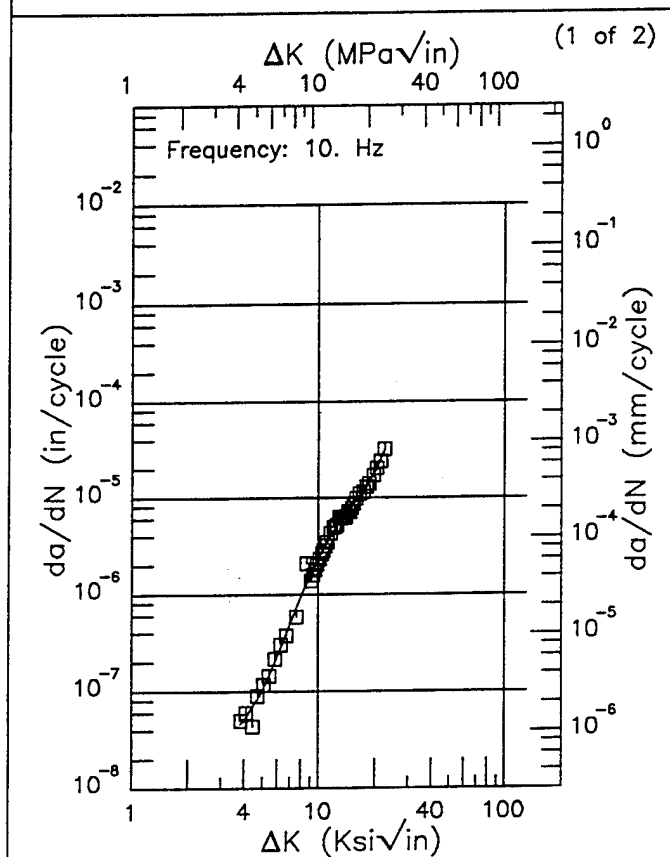
RMS % Error 37.27	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.	RMS % Error	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
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Figure 6.16.3.1.32 (Concluded)

F | Ti-6Al-4V |

Condition/Ht: ANNEALED
 Form: 3 in. Forging
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.4
 Environment: LAB AIR; RT

Yield Strength: 140.1 - 144.9 ksi
 Ult. Strength:
 Specimen Thk: 0.246 - 0.251 in.
 Specimen Width: 2.005 - 2.006 in.
 Ref: DA006;DA007



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
3.83 (min)	0.0461
4.	0.0498
5.	0.0994
6.	0.224
7.	0.480
8.	0.923
9.	1.57
10.	2.41
13.	5.71
16.	9.58
20.	18.6
22.36 (max)	31.7

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
4.58 (min)	0.0279
5.	0.0508
6.	0.149
7.	0.325
8.	0.589
9.	0.949
10.	1.41
13.	3.56
16.	7.37
20.	17.3
25.	47.3
28.75 (max)	98.8

RMS % Error 12.79	Life Prediction Ratio Summary
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RMS % Error 37.04	Life Prediction Ratio Summary
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Figure 6.16.3.1.33

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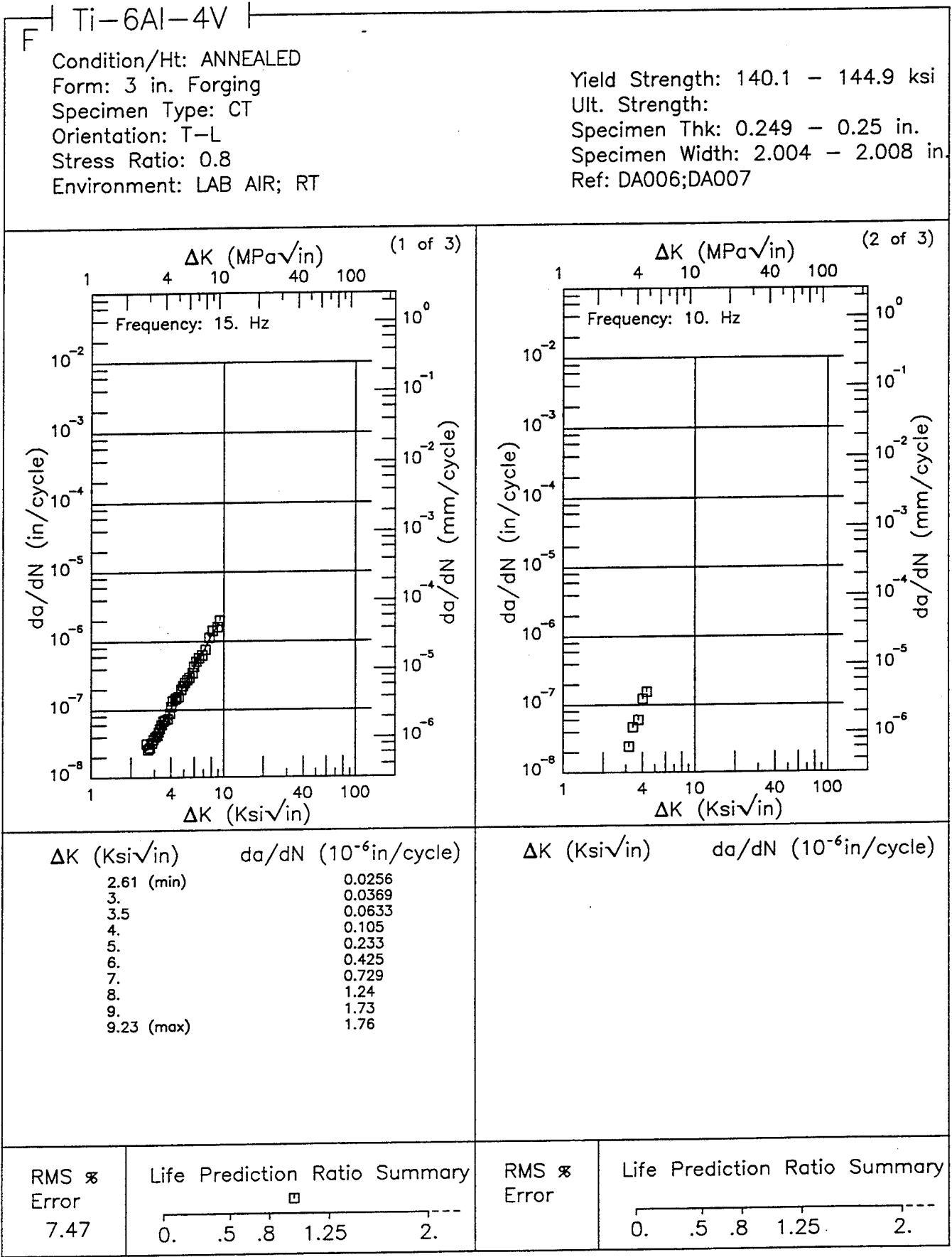
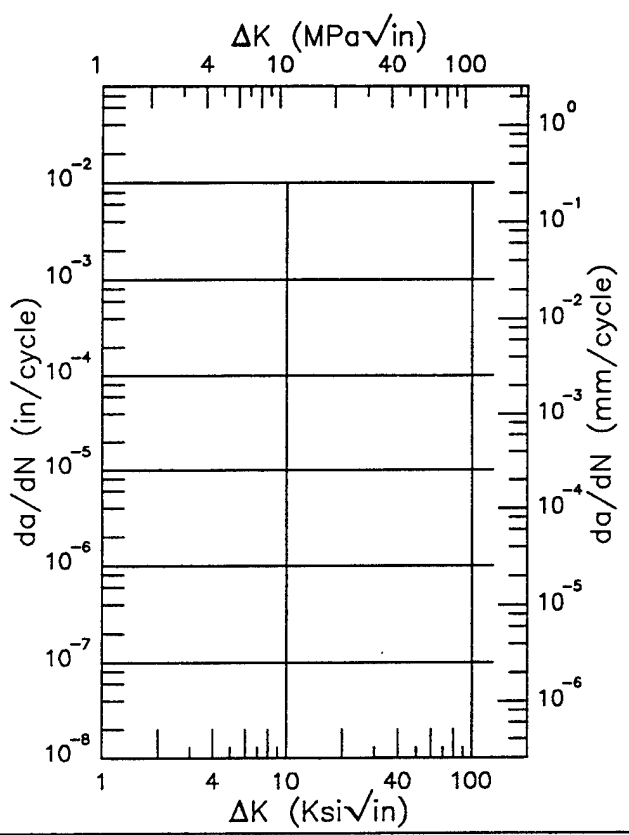
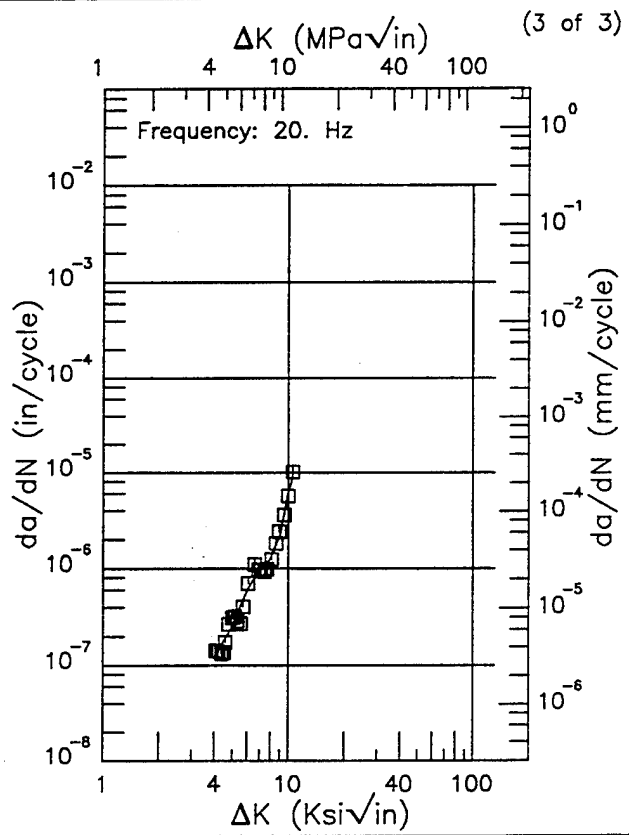


Figure 6.16.3.1.34

Ti-6Al-4V F

Condition/Ht: ANNEALED
 Form: 3 in. Forging
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.8
 Environment: LAB AIR; RT

Yield Strength: 140.1 - 144.9 ksi
 Ult. Strength:
 Specimen Thk: 0.249 - 0.25 in.
 Specimen Width: 2.004 - 2.008 in.
 Ref: DA006;DA007



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
4.06 (min)	0.128
5.	0.254
6.	0.592
7.	0.986
8.	1.26
9.	2.22
10.	6.32
10.53 (max)	9.59

ΔK (Ksi \sqrt{in}) da/dN (10^{-6} in/cycle)

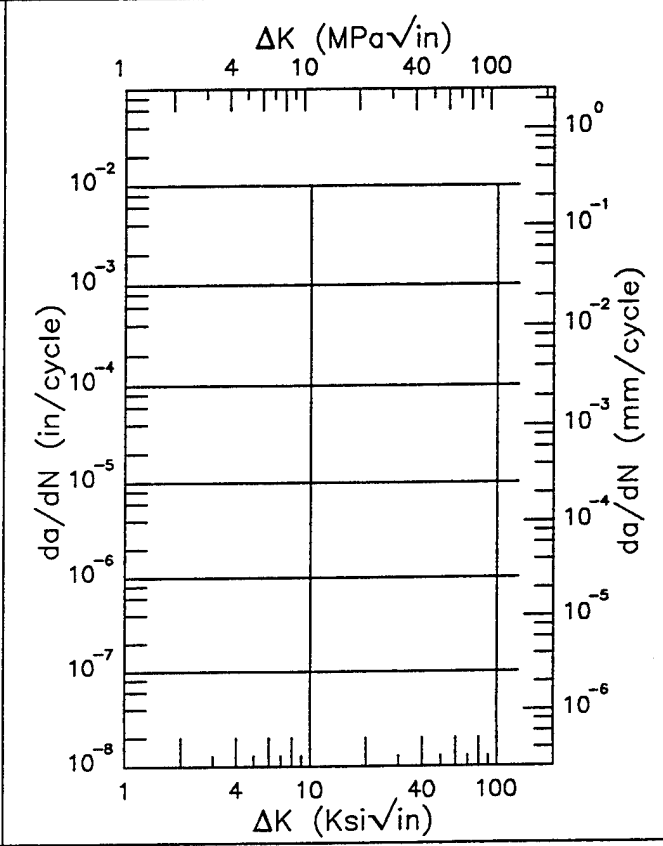
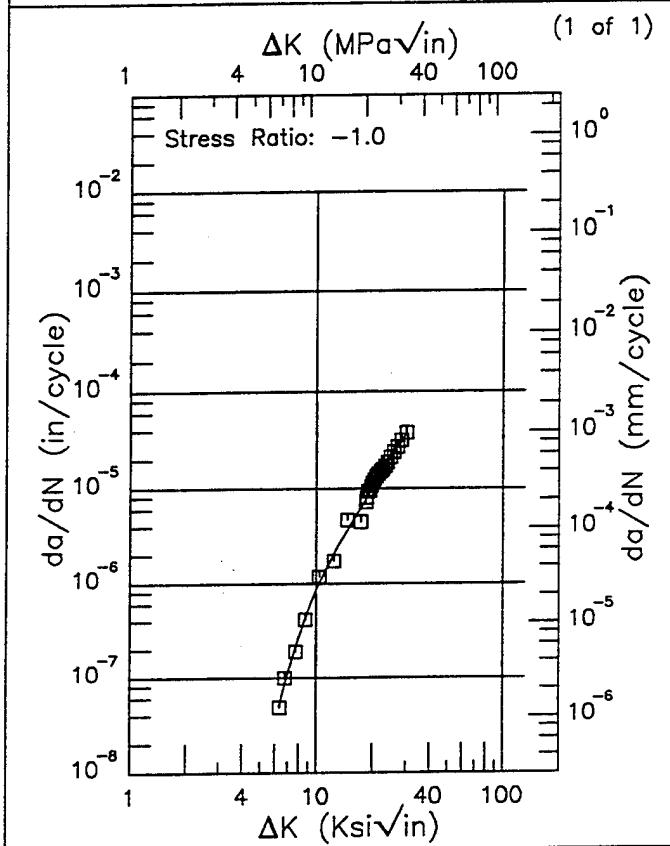
RMS % Error	Life Prediction Ratio Summary
16.59	

RMS % Error	Life Prediction Ratio Summary

Figure 6.16.3.1.34 (Concluded)

R | Ti-6Al-4V |
 Condition/Ht: ANNEALED
 Form: 3 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 5 Hz
 Environment: LAB AIR; RT

Yield Strength: 131.7 ksi
 Ult. Strength:
 Specimen Thk: 0.205 in.
 Specimen Width: 4.018 in.
 Ref: DA006



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
6.33 (min)	0.0501
7.	0.110
8.	0.268
9.	0.520
10.	0.868
13.	2.45
16.	4.89
20.	10.3
25.	21.9
30.	36.5
30.15 (max)	37.0

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
6.33 (min)	0.0501
7.	0.110
8.	0.268
9.	0.520
10.	0.868
13.	2.45
16.	4.89
20.	10.3
25.	21.9
30.	36.5
30.15 (max)	37.0

RMS % Error 11.51	Life Prediction Ratio Summary
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RMS % Error	Life Prediction Ratio Summary
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Figure 6.16.3.1.35

Ti-6Al-4V R

Condition/Ht: ANNEALED AT 1375F 3HRS AC
 Form: 2.75 in. Plate
 Specimen Type: WOL
 Orientation: L-T
 Frequency: 10 - 20 Hz
 Environment: LAB AIR; RT

Yield Strength: 129 ksi
 Ult. Strength: 139 ksi
 Specimen Thk: 1.25 in.
 Specimen Width: 5 in.
 Ref: MA003

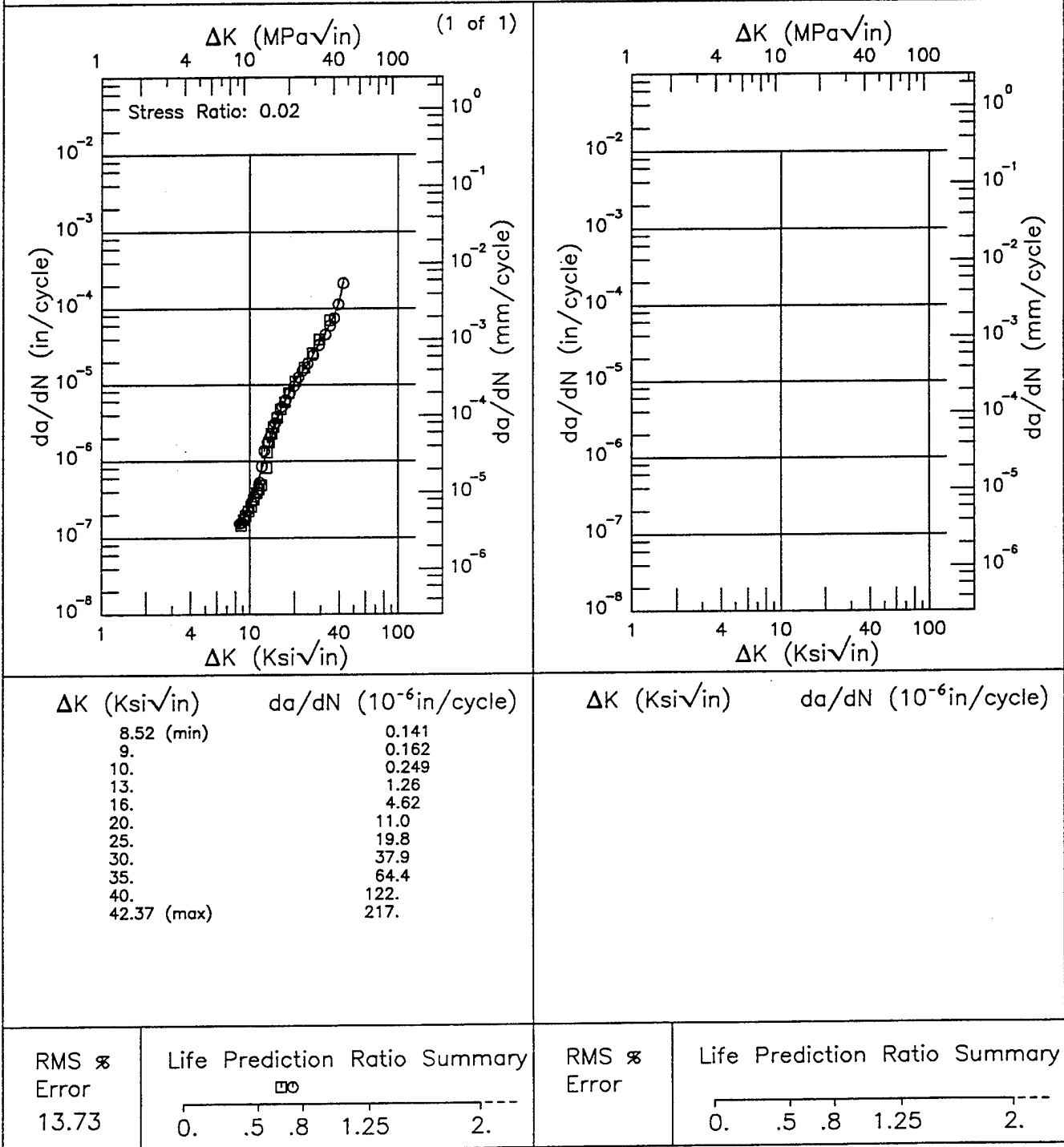
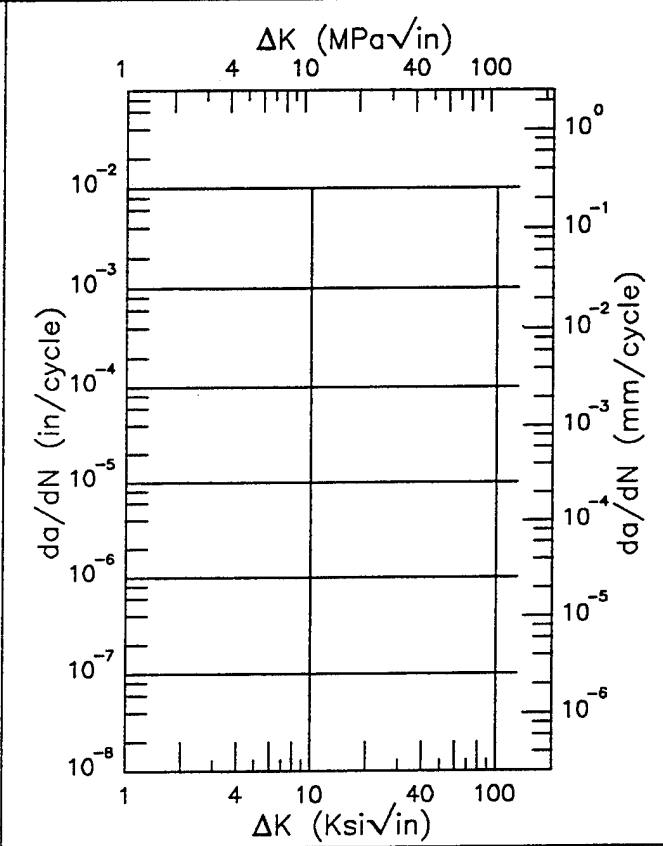
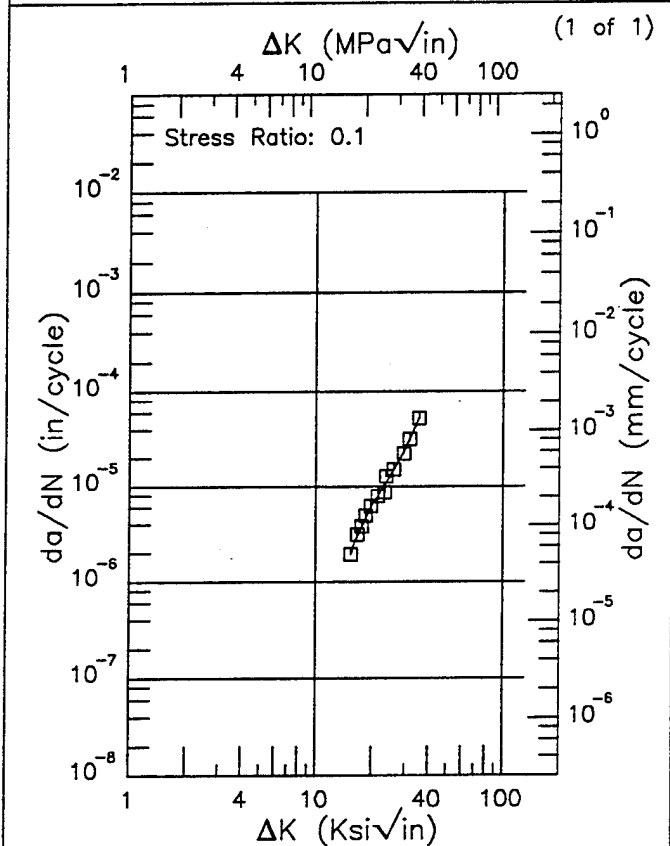


Figure 6.16.3.1.36

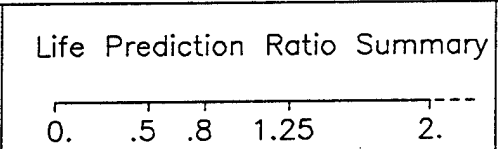
R | Ti-6Al-4V |
 Condition/Ht: AS WELDED E.B. WELDMENT (HAZ)
 Form: Weldment
 Specimen Type: CT
 Orientation: T-L
 Frequency: 10 Hz
 Environment: LAB AIR; RT
 Yield Strength:
 Ult. Strength:
 Specimen Thk: 1 in.
 Specimen Width: 2 in.
 Ref: 88144



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
15.35 (min)	1.98
16.	2.50
20.	6.52
25.	13.2
30.	24.7
35.	50.2
35.34 (max)	52.9

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
15.35 (min)	1.98
16.	2.50
20.	6.52
25.	13.2
30.	24.7
35.	50.2
35.34 (max)	52.9

RMS % Error
7.63



RMS % Error

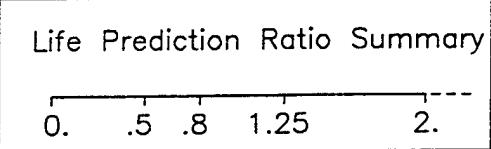


Figure 6.16.3.1.37

Ti-6Al-4V R

Condition/Ht: AS WELDED E.B. WELDMENT (WELD ZONE)

Form: Weldment

Specimen Type: CT

Orientation: T-L

Frequency: 10 Hz

Environment: LAB AIR; RT

Yield Strength:

Ult. Strength:

Specimen Thk: 1 in.

Specimen Width: 2 in.

Ref: 88144

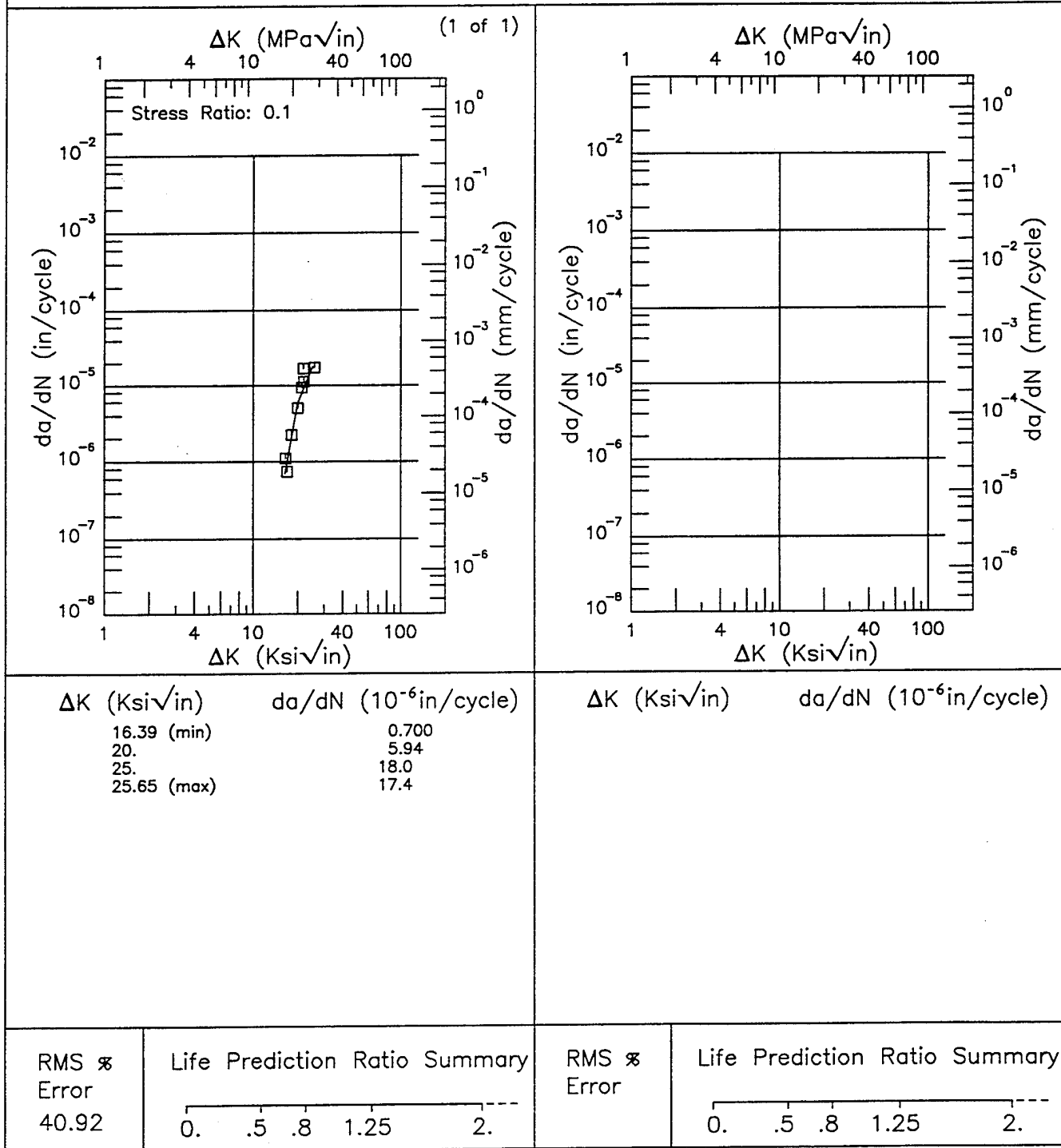


Figure 6.16.3.1.38

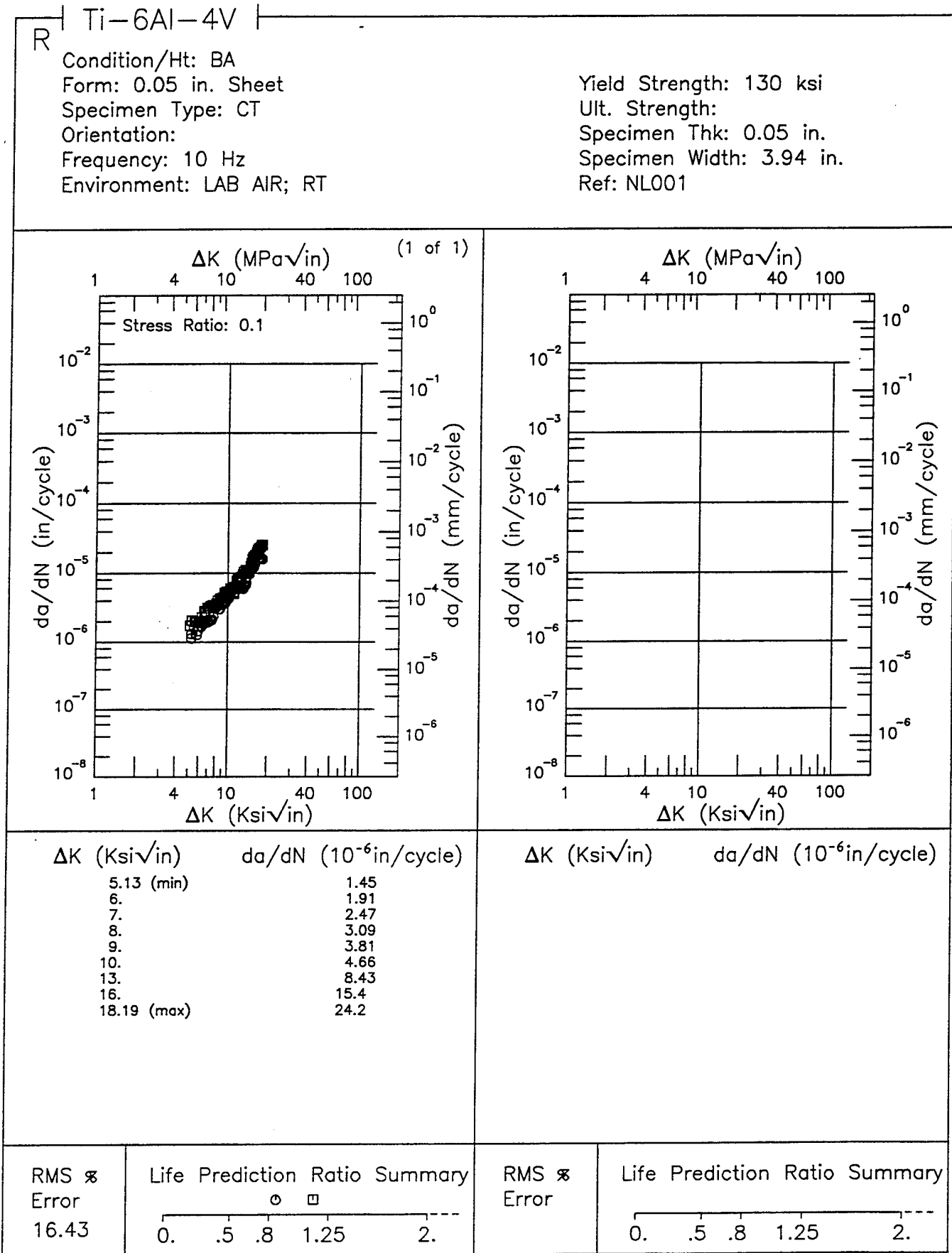


Figure 6.16.3.1.39

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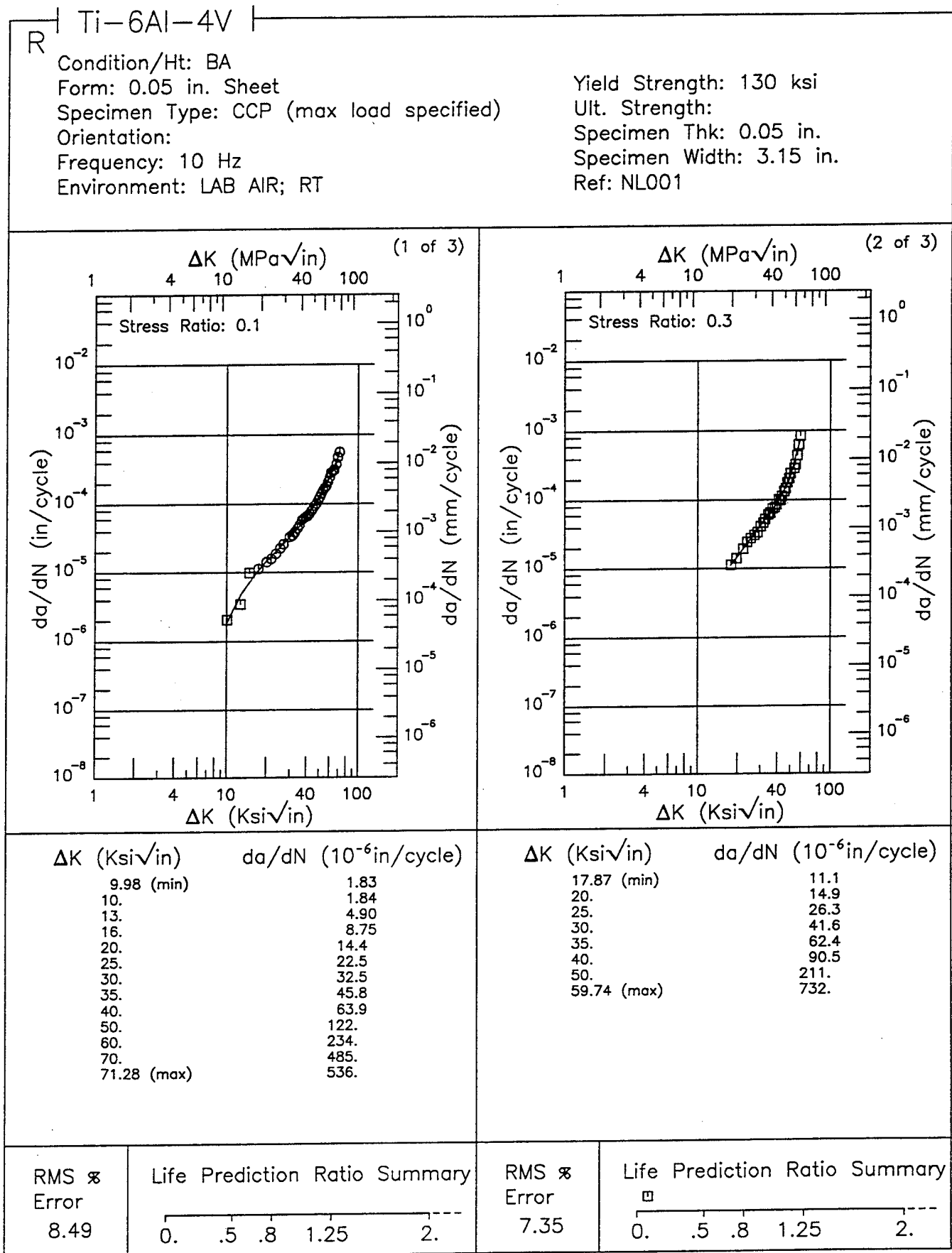
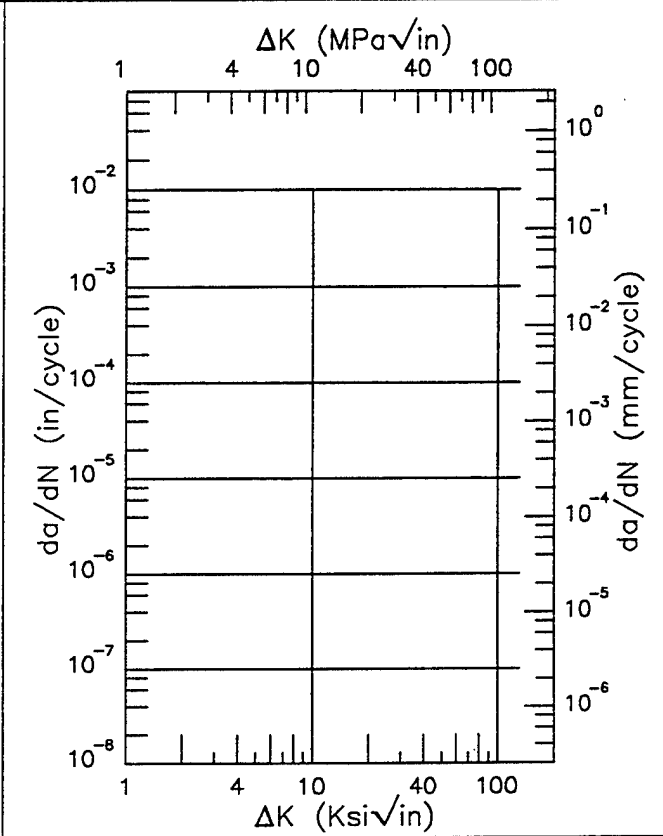
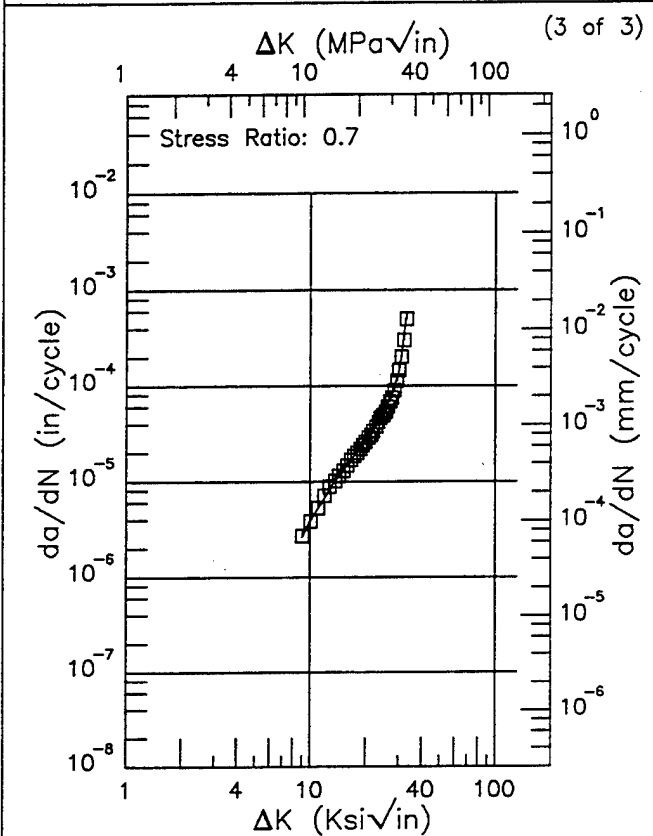


Figure 6.16.3.1.40

Ti-6Al-4V R

Condition/Ht: BA
 Form: 0.05 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation:
 Frequency: 10 Hz
 Environment: LAB AIR; RT

Yield Strength: 130 ksi
 Ult. Strength:
 Specimen Thk: 0.05 in.
 Specimen Width: 3.15 in.
 Ref: NL001



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
9.03 (min)	2.63
10.	3.87
13.	8.88
16.	15.3
20.	26.4
25.	50.5
30.	134.
33.00 (max)	515.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)

RMS % Error	Life Prediction Ratio Summary
5.36	

RMS % Error	Life Prediction Ratio Summary

Figure 6.16.3.1.40 (Concluded)

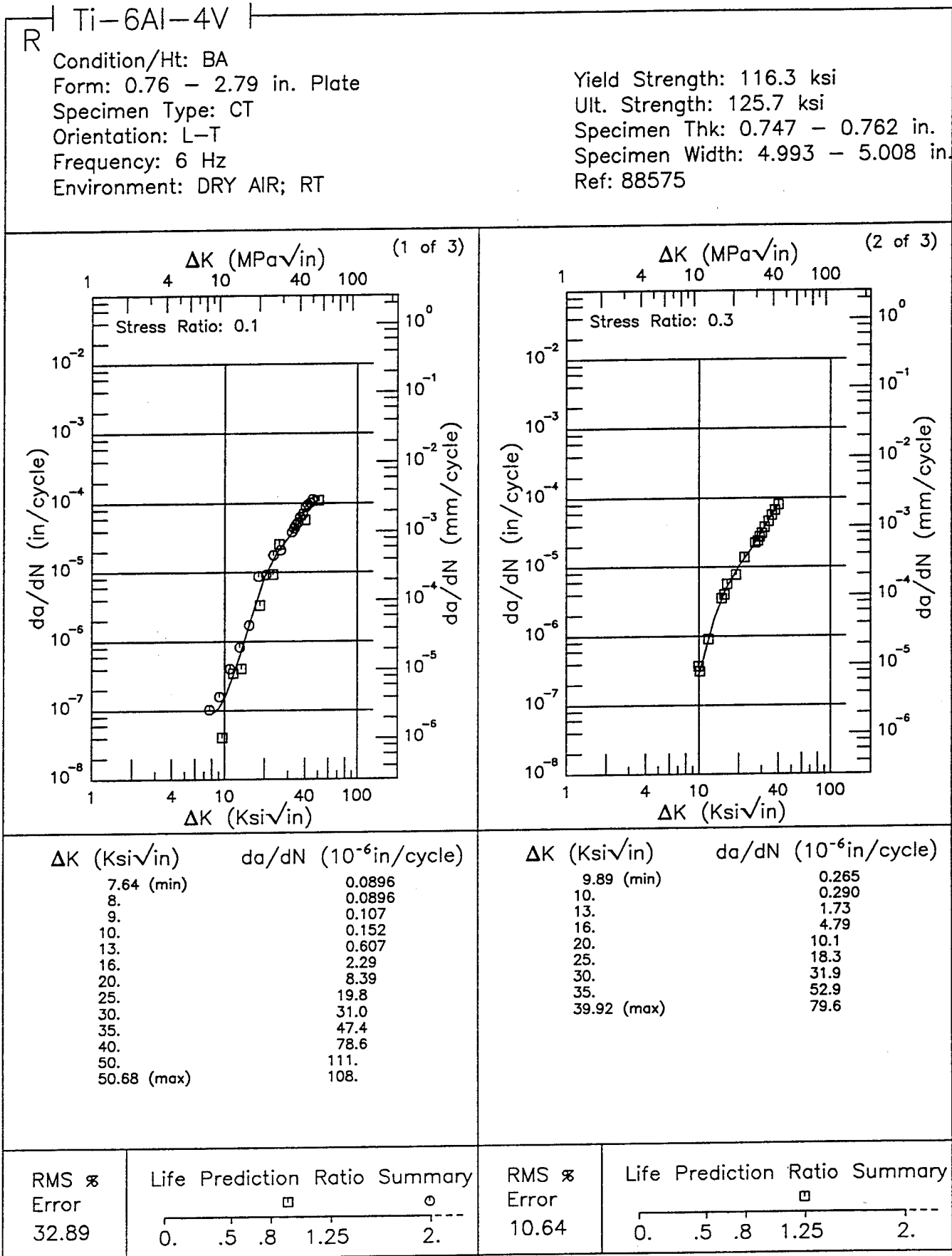


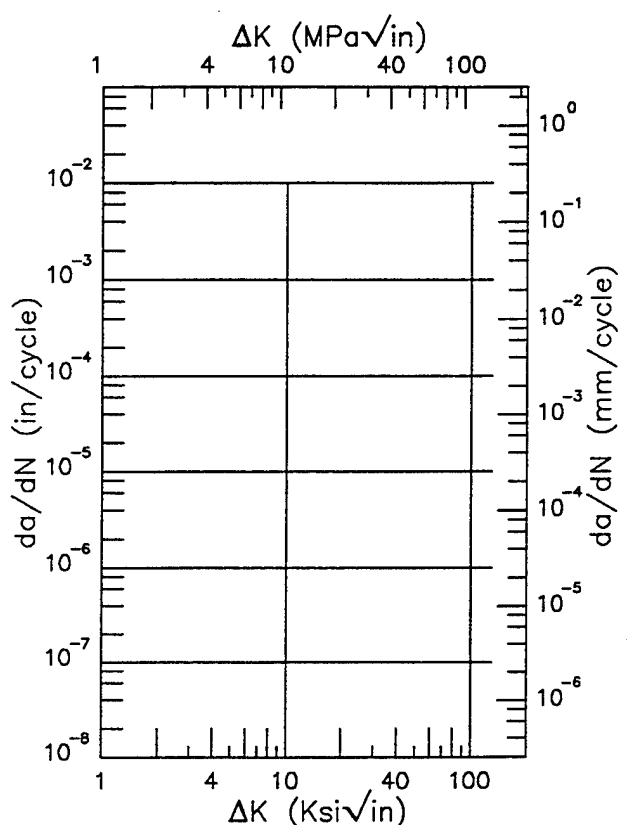
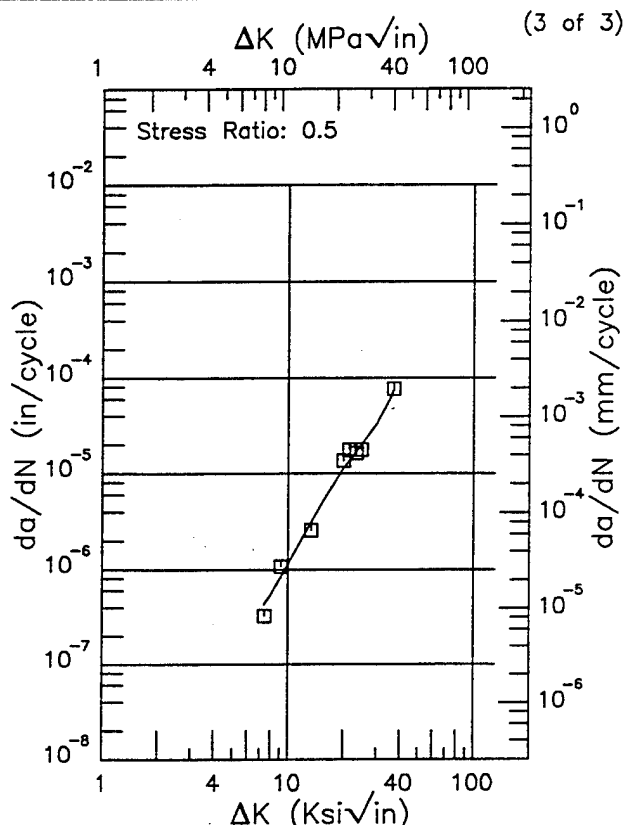
Figure 6.16.3.1.41

Ti-6Al-4V

R

Condition/Ht: BA
 Form: 0.76 - 2.79 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Frequency: 6 Hz
 Environment: DRY AIR; RT

Yield Strength: 116.3 ksi
 Ult. Strength: 125.7 ksi
 Specimen Thk: 0.747 - 0.762 in.
 Specimen Width: 4.993 - 5.008 in.
 Ref: 88575



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.43 (min)	0.428
8.	0.541
9.	0.800
10.	1.15
13.	2.86
16.	5.76
20.	11.4
25.	20.2
30.	33.0
35.	58.5
36.97 (max)	75.9

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS \times Error 19.61	Life Prediction Ratio Summary
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RMS \times Error	Life Prediction Ratio Summary
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Figure 6.16.3.1.41 (Concluded)

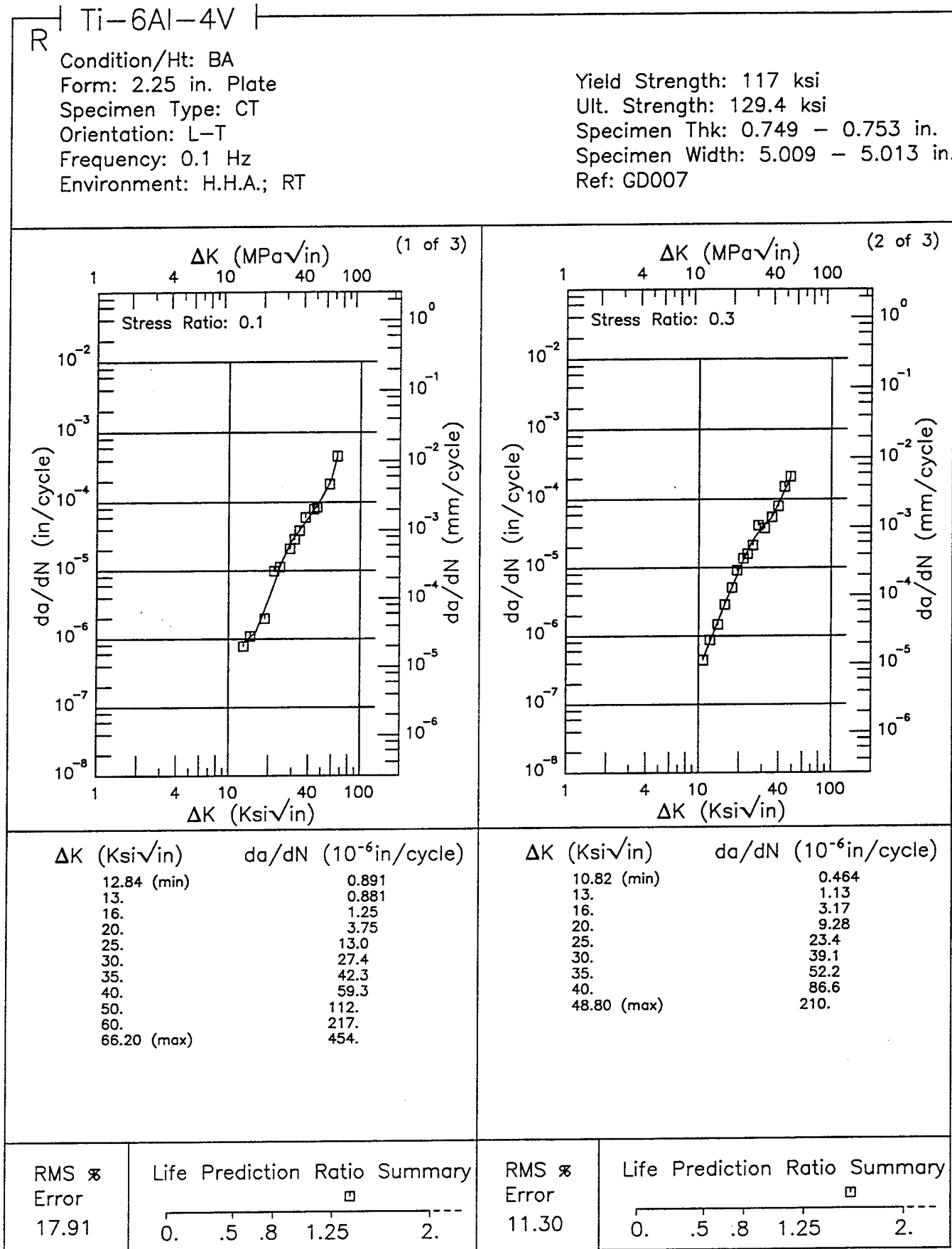


Figure 6.16.3.1.42

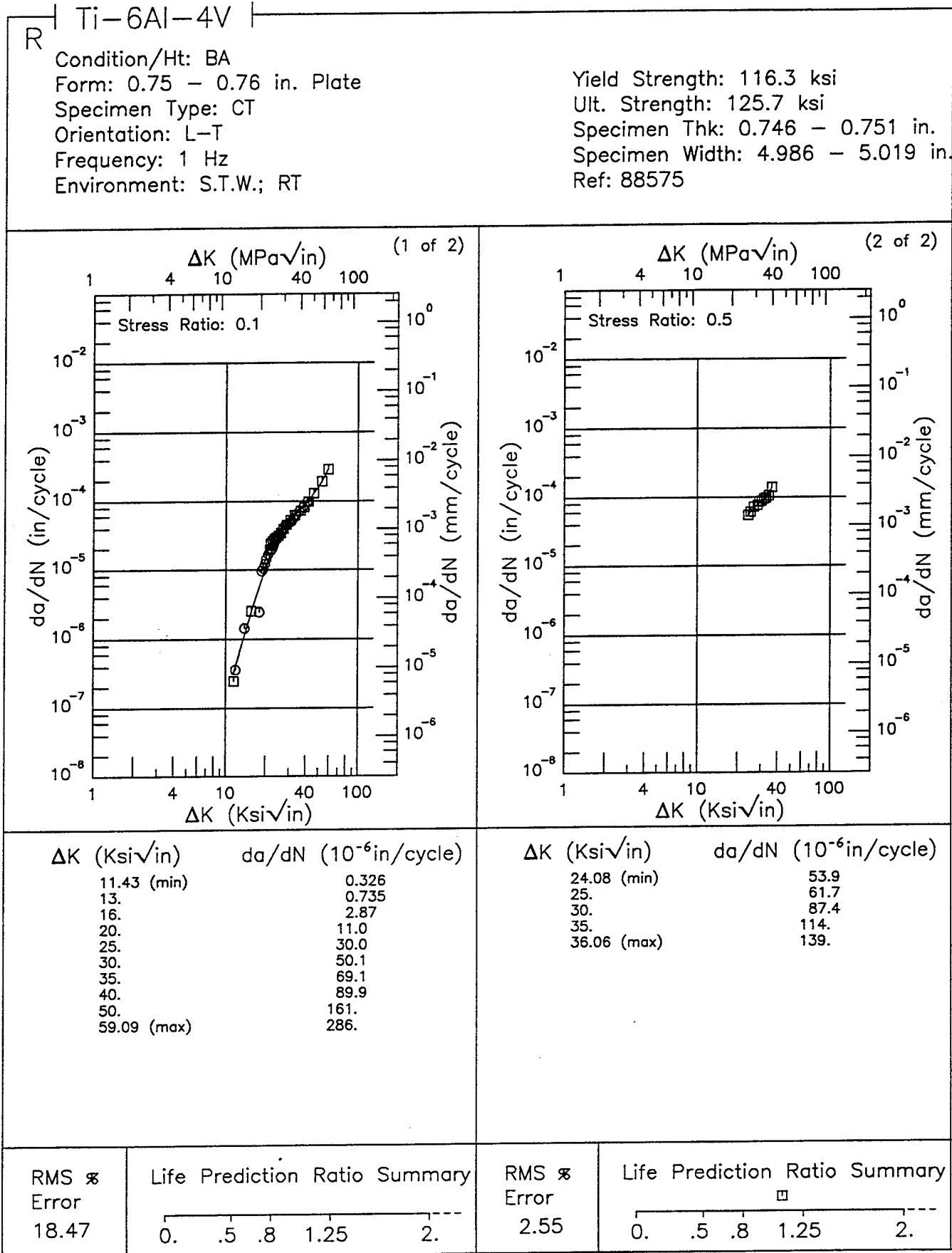


Figure 6.16.3.1.43

Ti-6Al-4V R

Condition/Ht: BA
 Form: 0.75 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Frequency: 0.1 Hz
 Environment: S.T.W.; RT

Yield Strength: 116.3 ksi
 Ult. Strength: 125.7 ksi
 Specimen Thk: 0.75 in.
 Specimen Width: 5.008 in.
 Ref: 88575

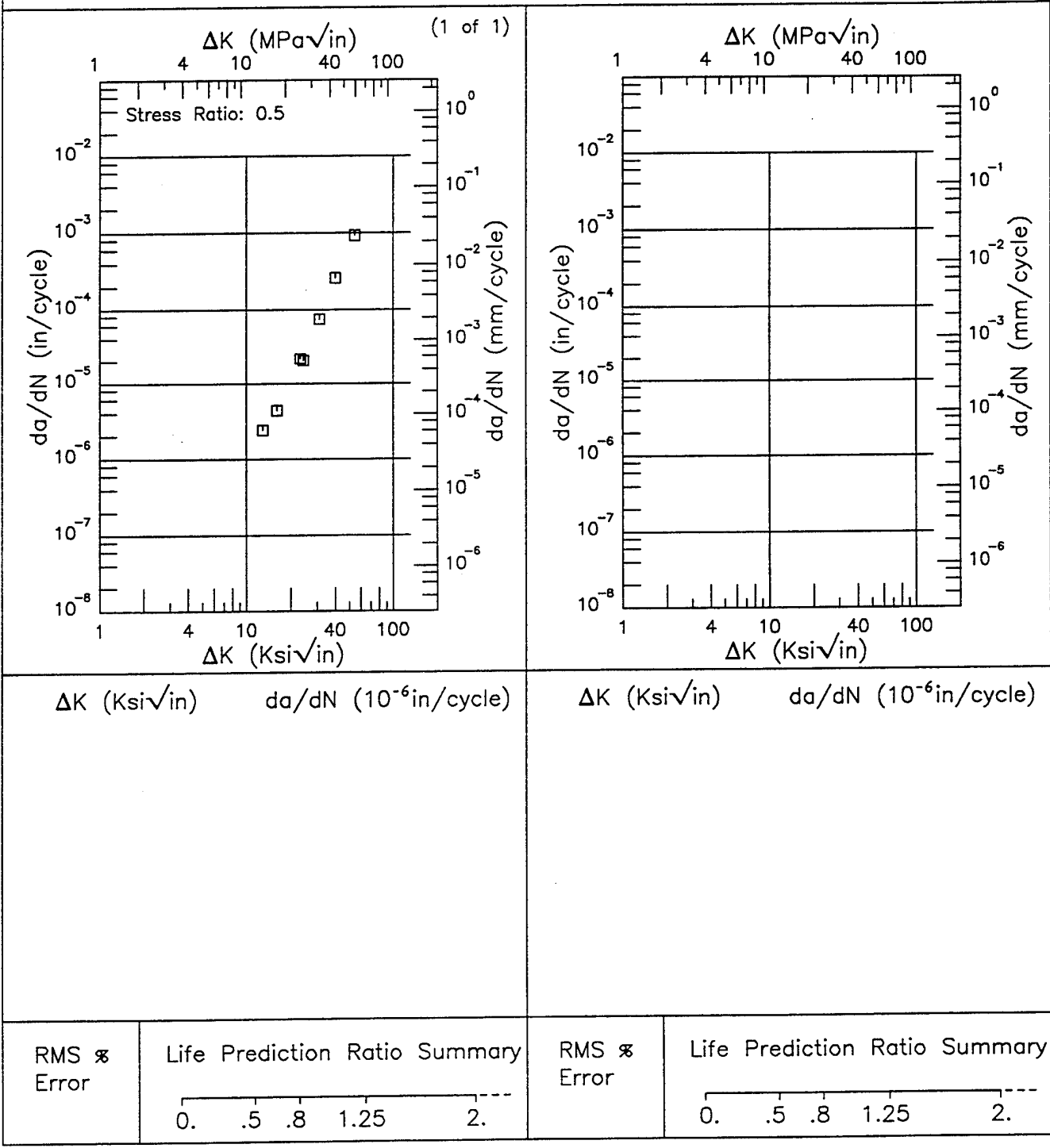
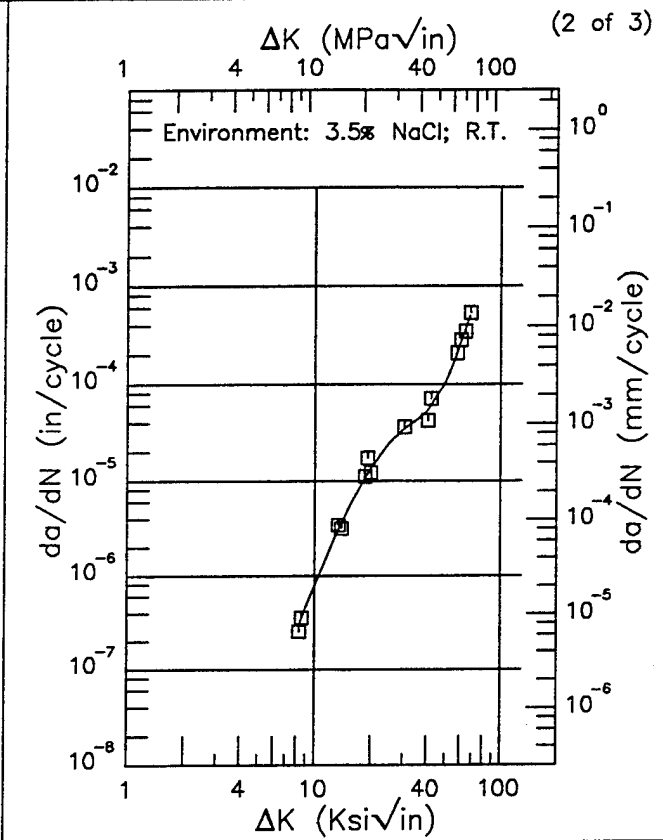
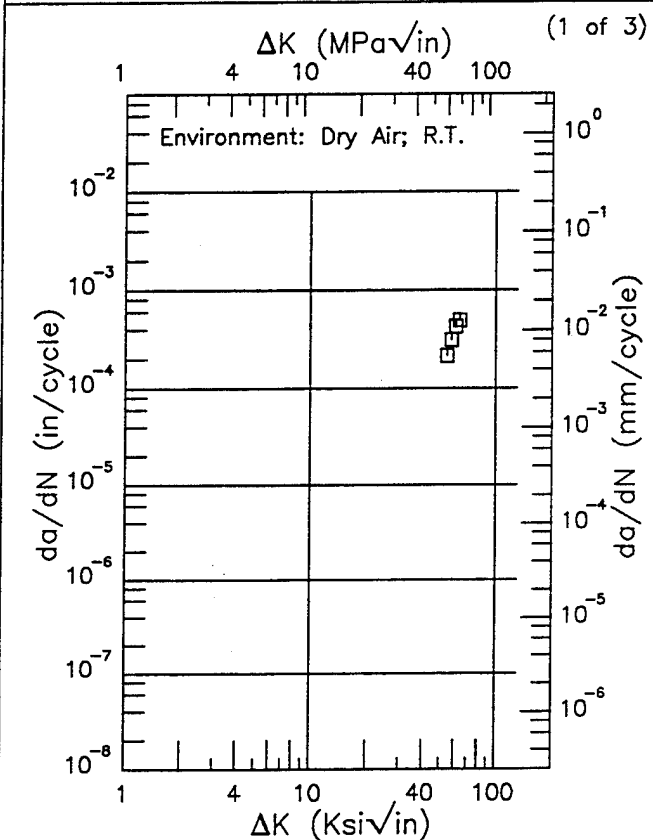


Figure 6.16.3.1.44

E | Ti-6Al-4V |

Condition/Ht: BA
 Form: 0.63 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1
 Frequency: 10 Hz

Yield Strength: 124.5 ksi
 Ult. Strength: 136.4 ksi
 Specimen Thk: 0.66 in.
 Specimen Width: 2.55 - 2.554 in.
 Ref: 90981



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.24 (min)	0.294
9.	0.464
10.	0.787
13.	2.69
16.	6.31
20.	13.6
25.	24.6
30.	34.1
35.	42.0
40.	52.4
50.	103.
60.	254.
68.19 (max)	499.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.24 (min)	0.294
9.	0.464
10.	0.787
13.	2.69
16.	6.31
20.	13.6
25.	24.6
30.	34.1
35.	42.0
40.	52.4
50.	103.
60.	254.
68.19 (max)	499.

RMS % Error	Life Prediction Ratio Summary
	0. .5 .8 1.25 2. ---

RMS % Error	Life Prediction Ratio Summary
16.33	0. .5 .8 1.25 2. --- □

Figure 6.16.3.1.45

Ti-6Al-4V E

Condition/Ht: BA
 Form: 0.63 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1
 Frequency: 10 Hz

Yield Strength: 124.5 ksi
 Ult. Strength: 136.4 ksi
 Specimen Thk: 0.66 in.
 Specimen Width: 2.55 - 2.554 in.
 Ref: 90981

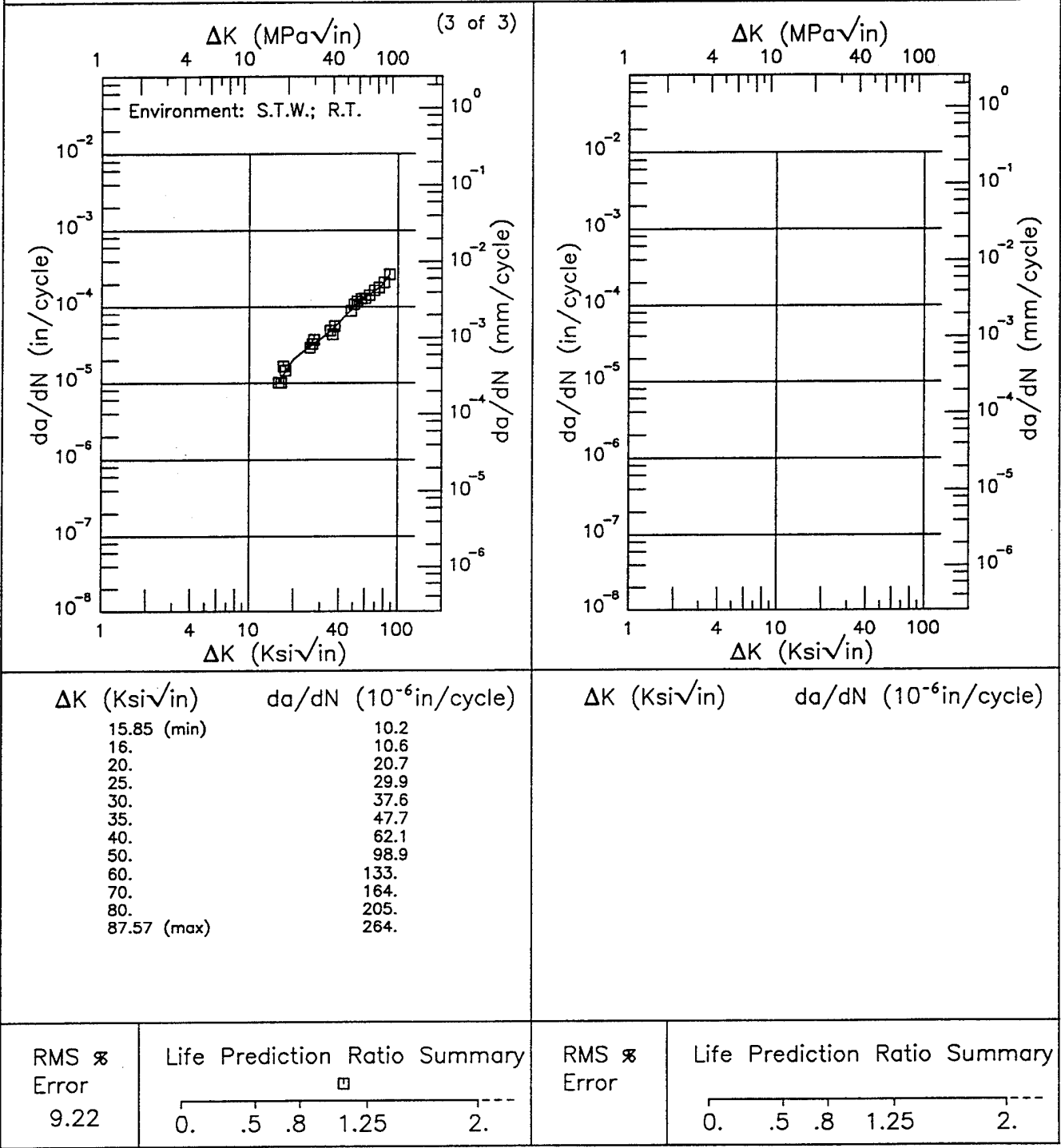


Figure 6.16.3.1.45 (Concluded)

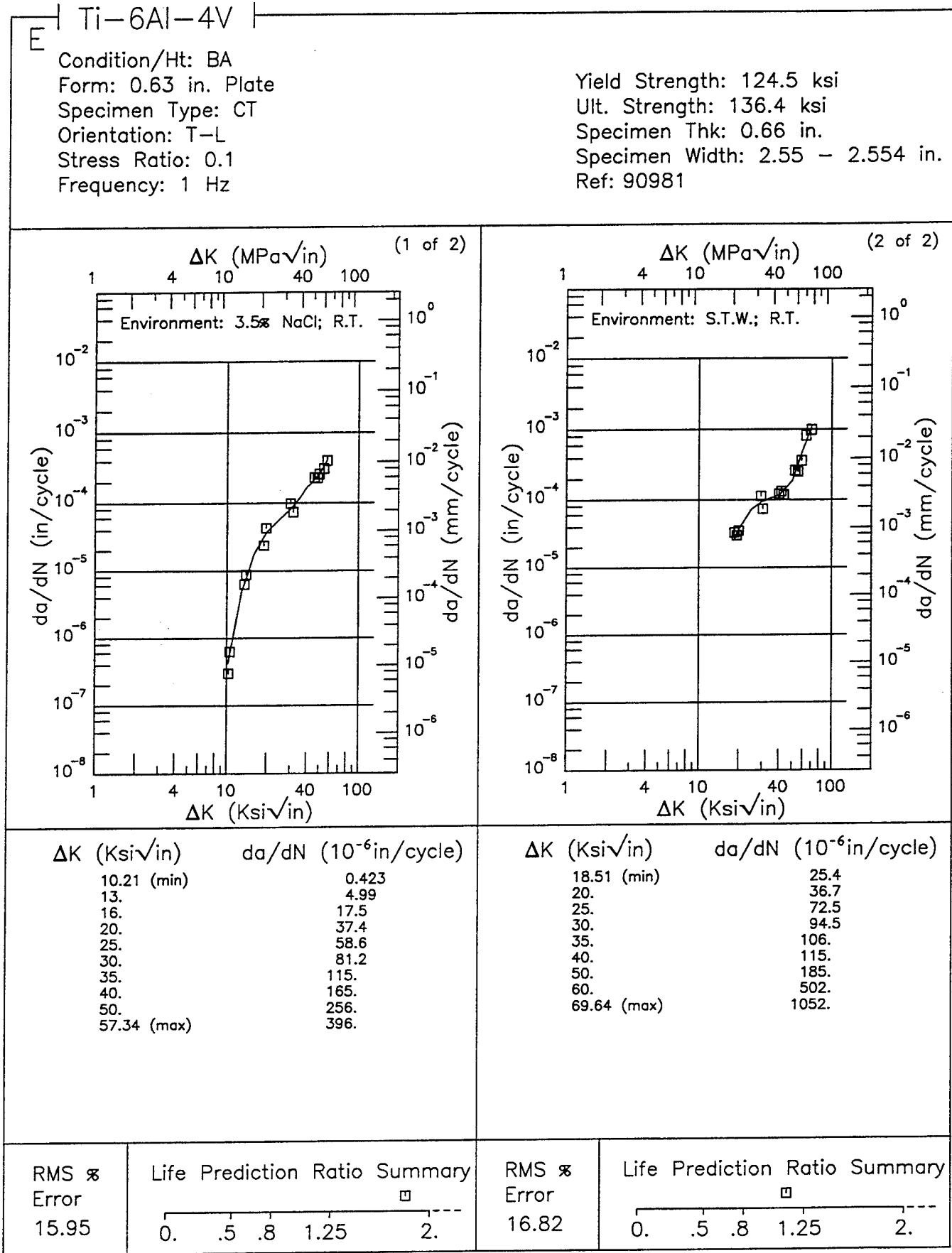
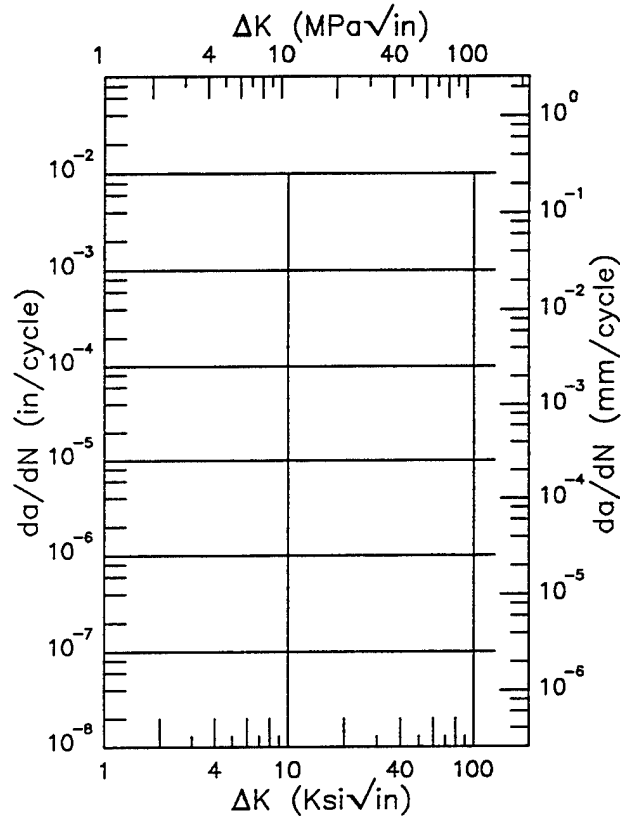
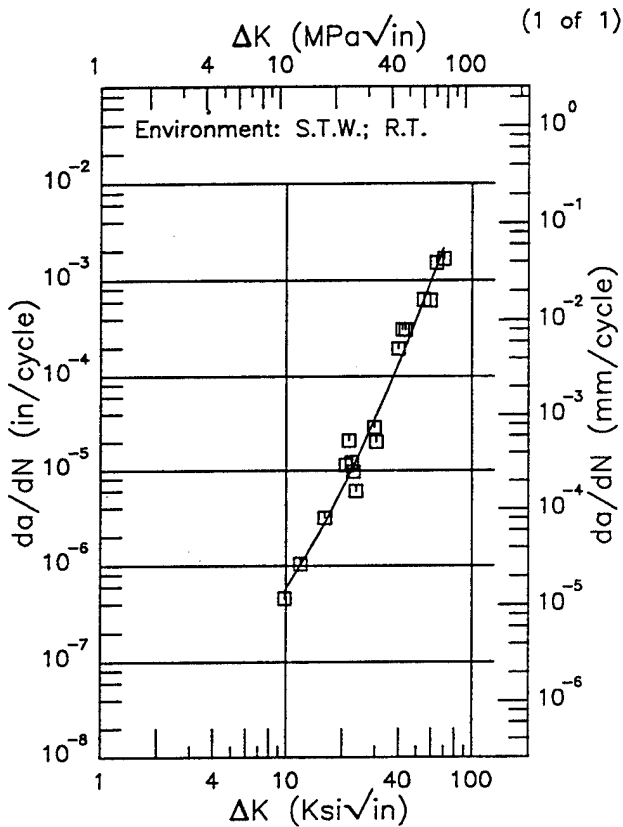


Figure 6.16.3.1.46

Ti-6Al-4V E

Condition/Ht: BA
 Form: 0.63 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1
 Frequency: 0.1 Hz

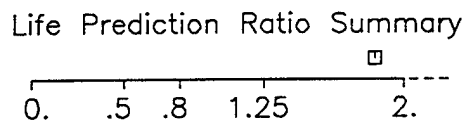
Yield Strength: 124.5 ksi
 Ult. Strength: 136.4 ksi
 Specimen Thk: 0.66 in.
 Specimen Width: 2.55 in.
 Ref: 90981



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
9.81 (min)	0.550
10.	0.581
13.	1.31
16.	2.73
20.	6.43
25.	16.2
30.	36.1
35.	72.9
40.	136.
50.	400.
60.	987.
69.91 (max)	2127.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
--------------------------------------	-------------------------------

RMS %
 Error
 47.90



RMS %
 Error

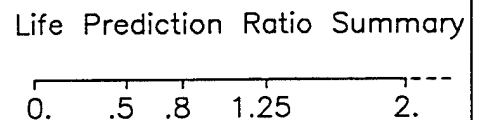


Figure 6.16.3.1.47

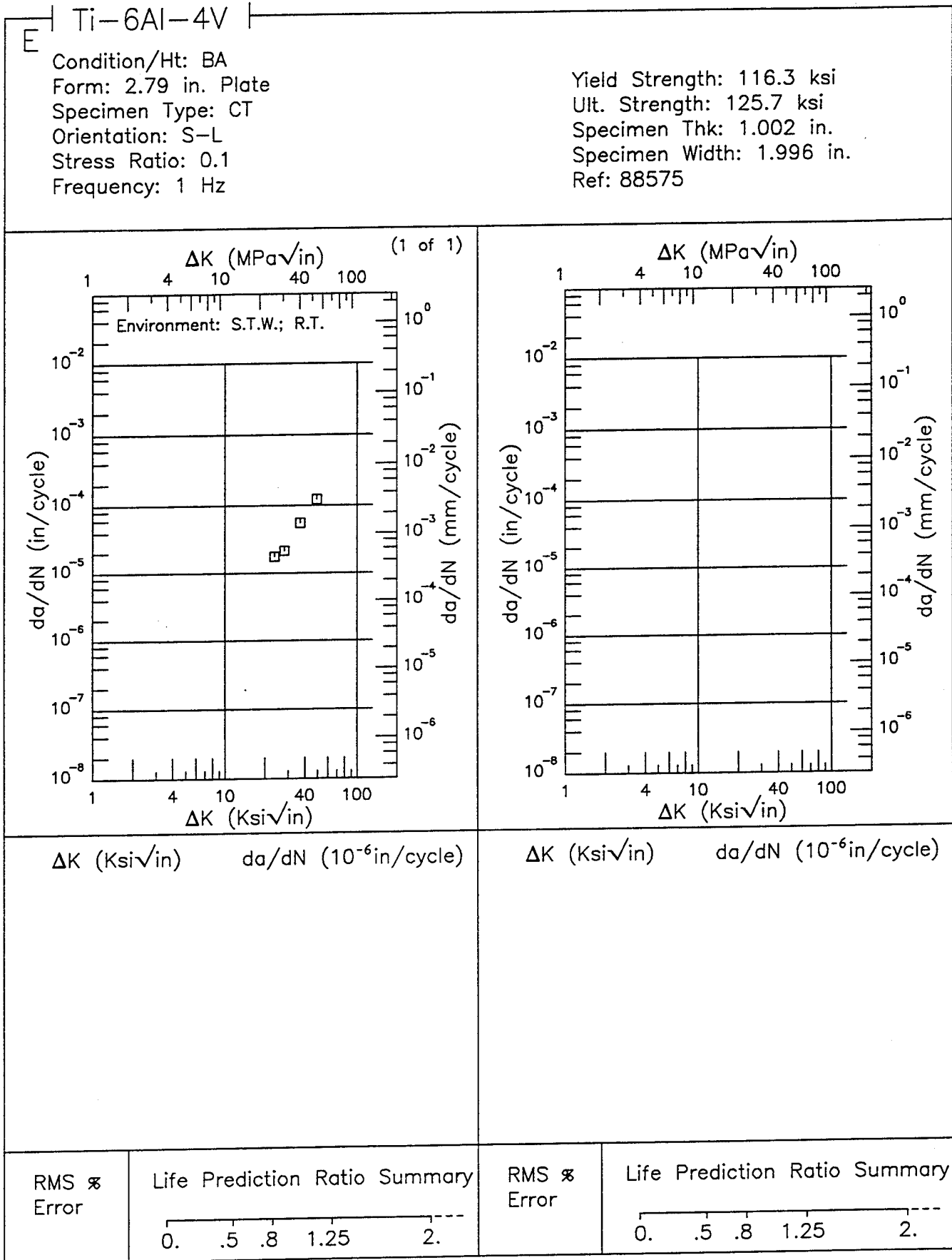
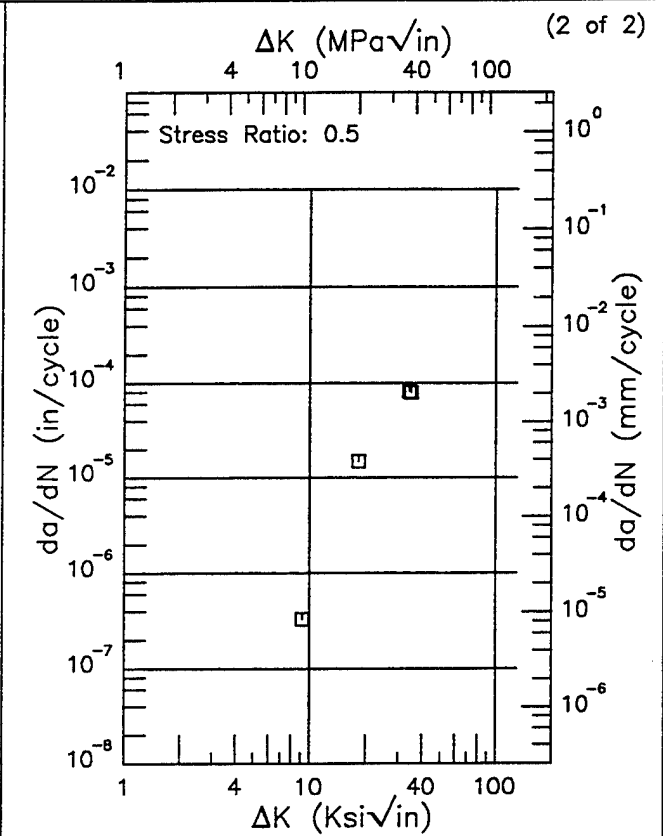
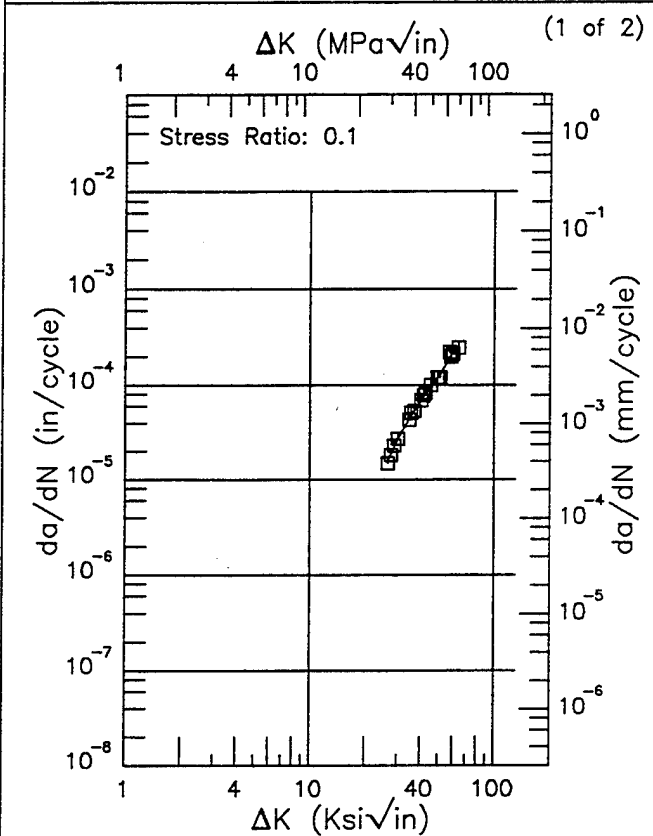


Figure 6.16.3.1.48

Ti-6Al-4V R

Condition/Ht: BA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Frequency: 0.1 Hz
 Environment: DIST WATER; RT

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
26.53 (min)	15.4
30.	26.2
35.	46.0
40.	68.0
50.	123.
60.	227.
63.96 (max)	244.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
26.53 (min)	15.4
30.	26.2
35.	46.0
40.	68.0
50.	123.
60.	227.
63.96 (max)	244.

RMS % Error	Life Prediction Ratio Summary
5.25	0. .5 .8 1.25 2. ---

RMS % Error	Life Prediction Ratio Summary
5.25	0. .5 .8 1.25 2. ---

Figure 6.16.3.1.49

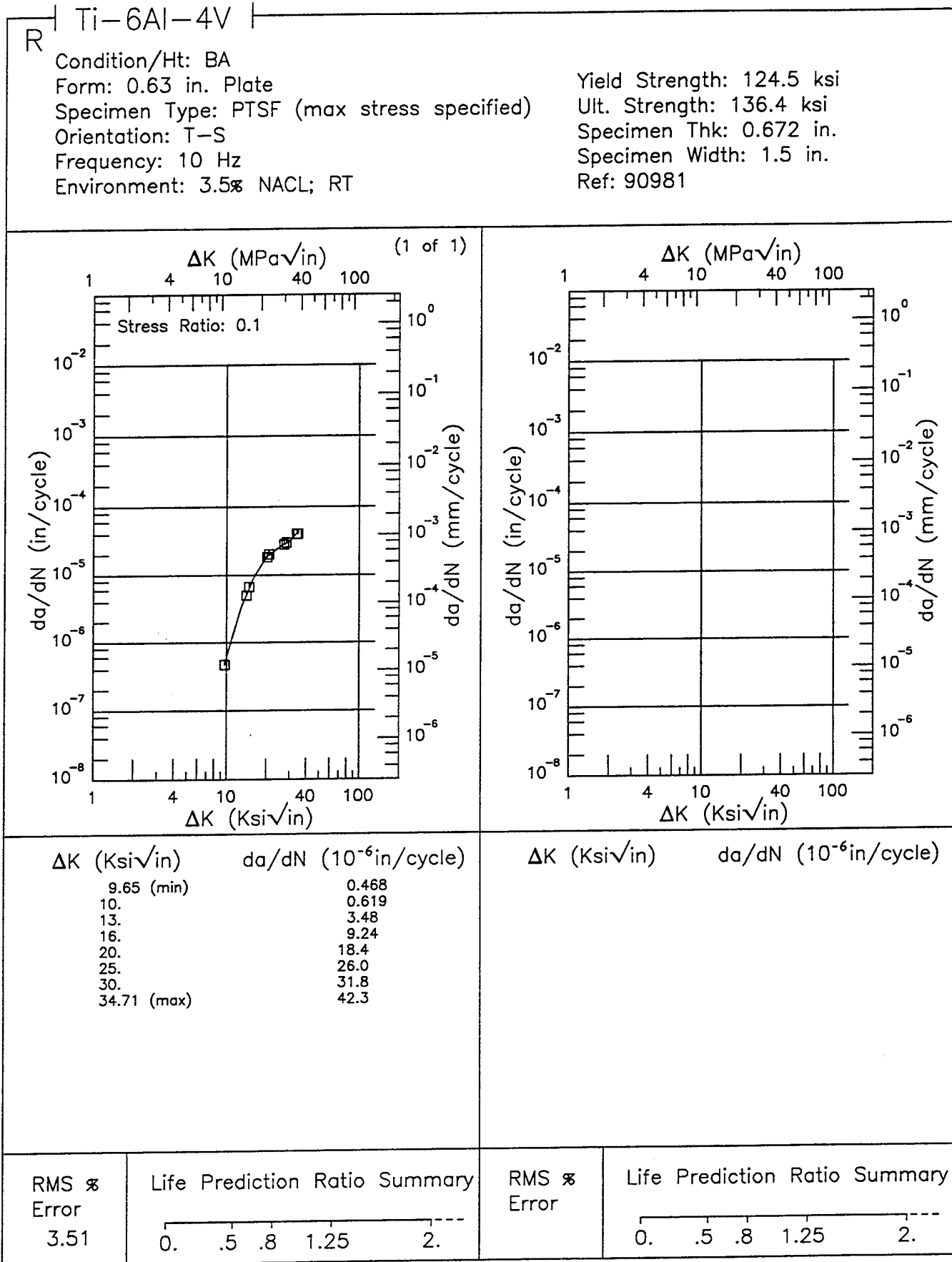


Figure 6.16.3.1.50

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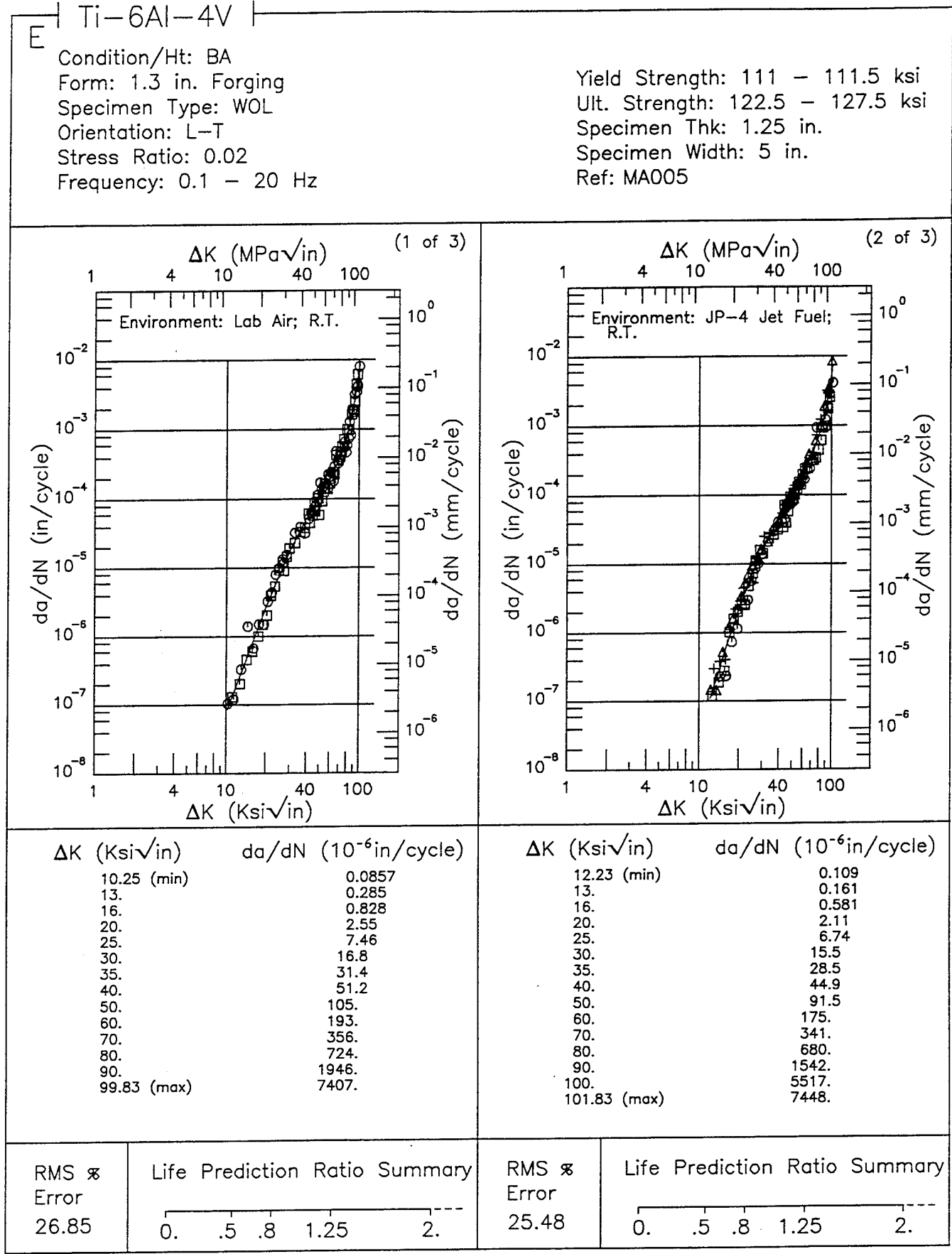
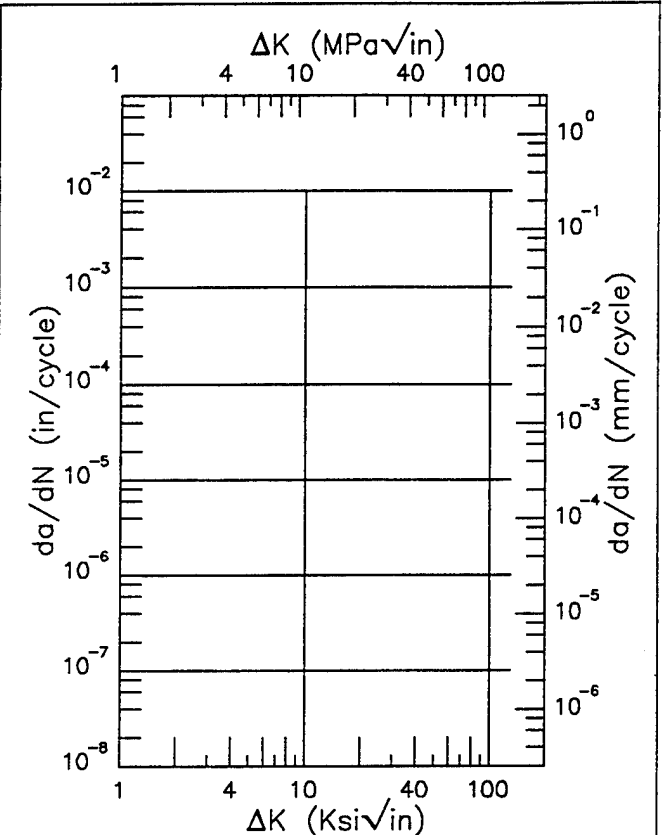
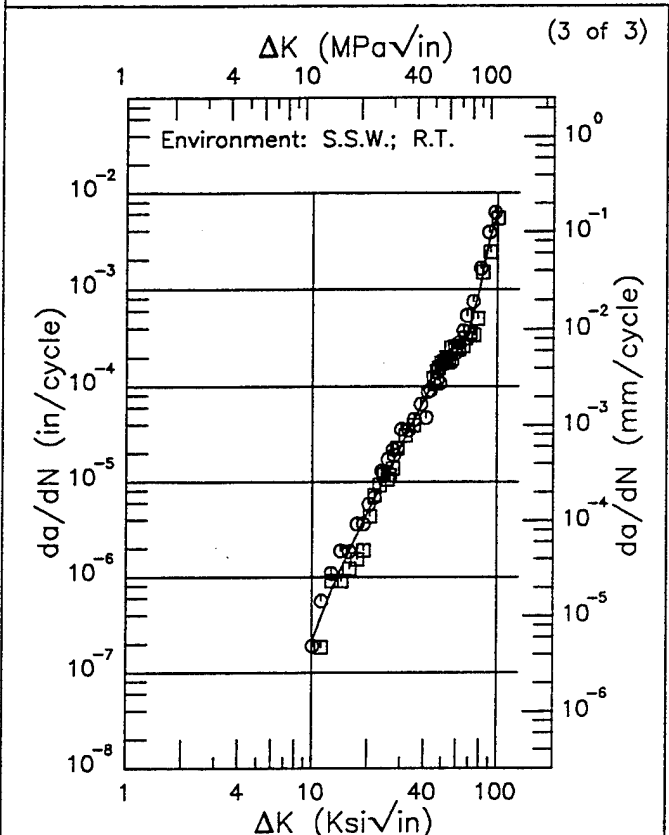


Figure 6.16.3.1.51

Ti-6Al-4V E

Condition/Ht: BA
 Form: 1.3 in. Forging
 Specimen Type: WOL
 Orientation: L-T
 Stress Ratio: 0.02
 Frequency: 0.1 - 20 Hz

Yield Strength: 111 - 111.5 ksi
 Ult. Strength: 122.5 - 127.5 ksi
 Specimen Thk: 1.25 in.
 Specimen Width: 5 in.
 Ref: MA005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.06 (min)	0.223
13.	0.775
16.	1.93
20.	4.83
25.	11.6
30.	23.6
35.	43.3
40.	72.7
50.	154.
60.	238.
70.	392.
80.	1007.
90.	3207.
99.93 (max)	6205.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.06 (min)	0.223
13.	0.775
16.	1.93
20.	4.83
25.	11.6
30.	23.6
35.	43.3
40.	72.7
50.	154.
60.	238.
70.	392.
80.	1007.
90.	3207.
99.93 (max)	6205.

RMS % Error	Life Prediction Ratio Summary
25.94	

RMS % Error	Life Prediction Ratio Summary

Figure 6.16.3.1.51 (Concluded)

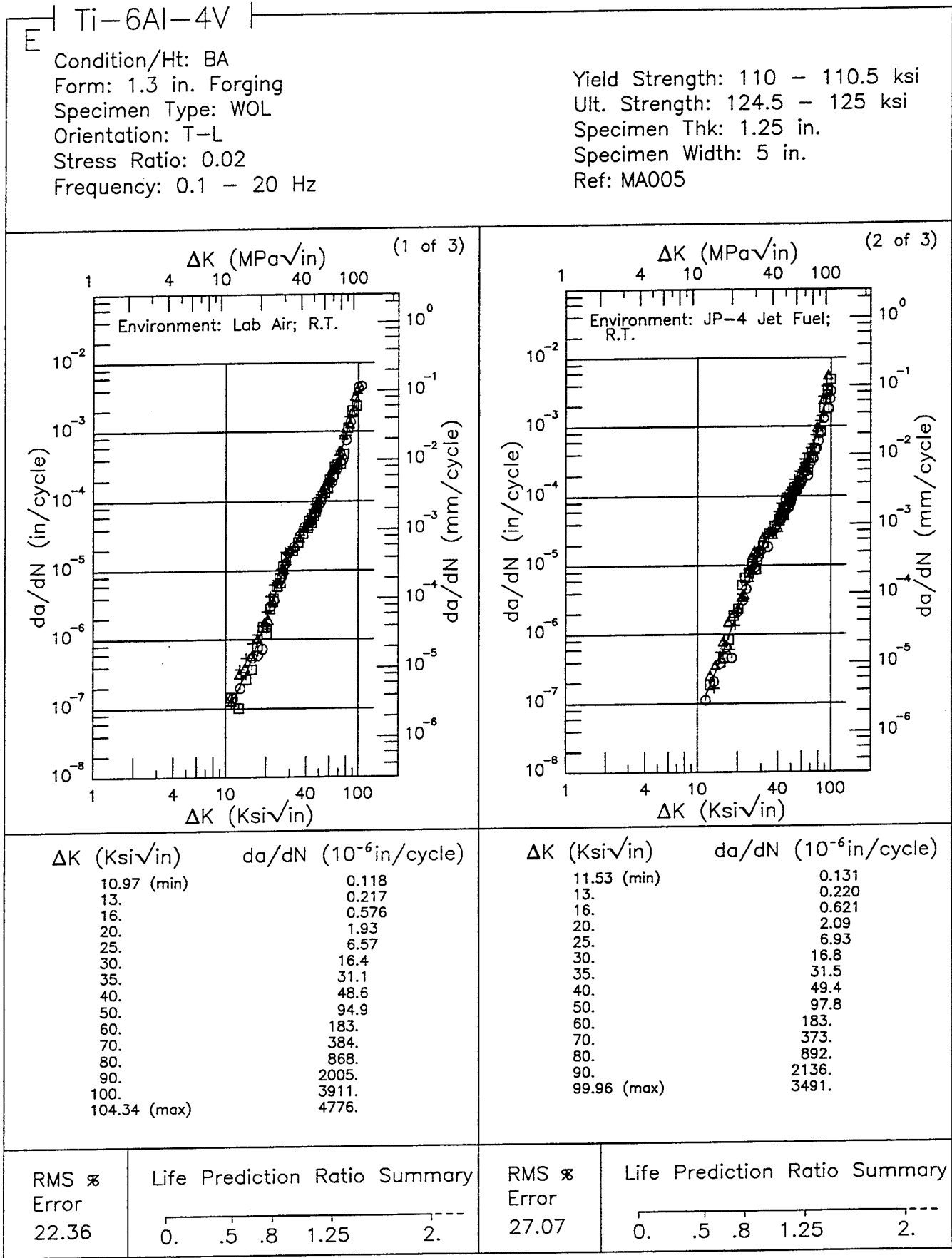
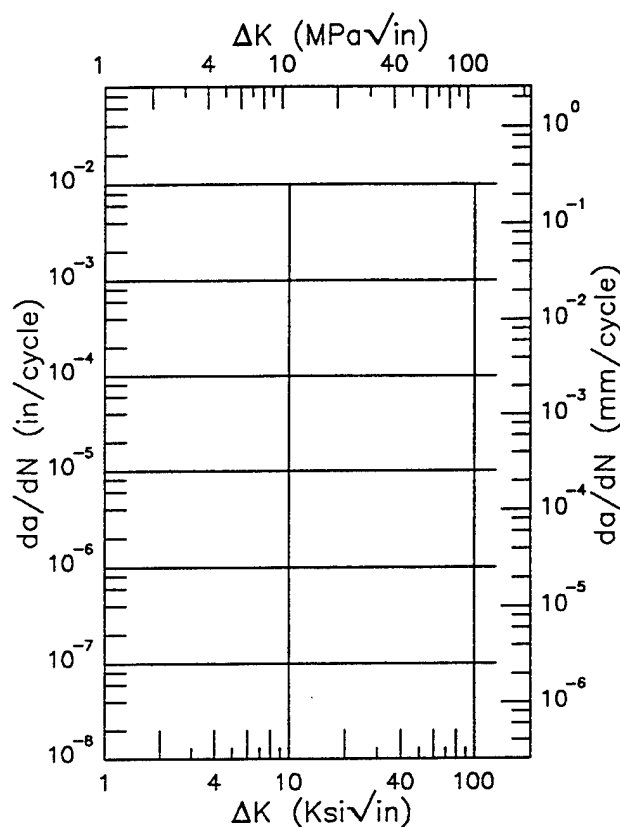
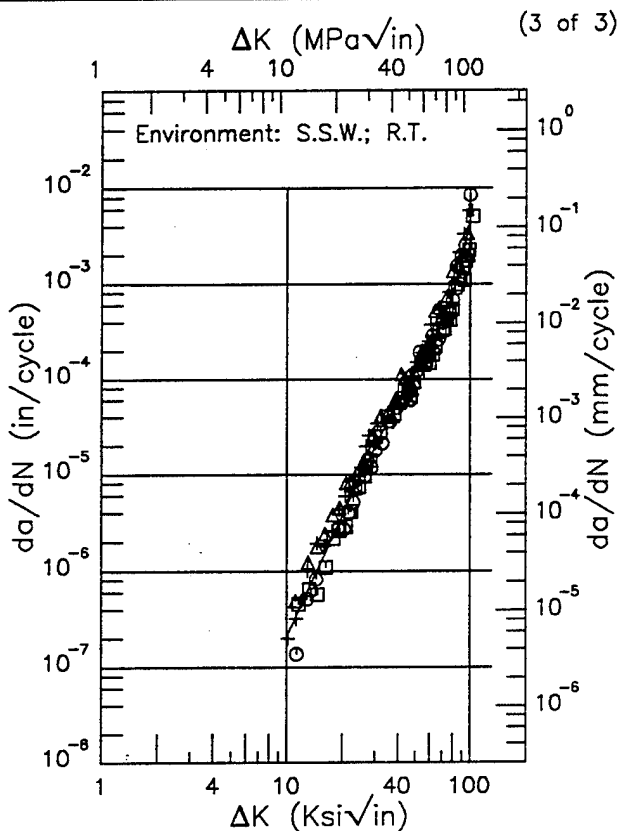


Figure 6.16.3.1.52

Ti-6Al-4V E

Condition/Ht: BA
 Form: 1.3 in. Forging
 Specimen Type: WOL
 Orientation: T-L
 Stress Ratio: 0.02
 Frequency: 0.1 - 20 Hz

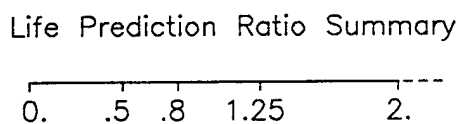
Yield Strength: 110 - 110.5 ksi
 Ult. Strength: 124.5 - 125 ksi
 Specimen Thk: 1.25 in.
 Specimen Width: 5 in.
 Ref: MA005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.11 (min)	0.217
13.	0.620
16.	1.48
20.	3.76
25.	9.37
30.	19.4
35.	35.0
40.	57.6
50.	126.
60.	229.
70.	399.
80.	781.
90.	1823.
100.	5139.
102.34 (max)	6727.

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS \times
 Error
 33.82



RMS \times
 Error

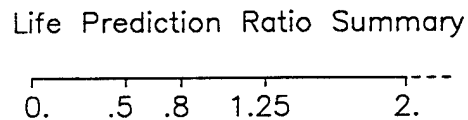


Figure 6.16.3.1.52 (Concluded)

Ti-6Al-4V

R

Condition/Ht: BETA PROCESSED - MA
 Form: 0.3 - 0.5 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Frequency: 10 Hz
 Environment: DRY AIR; RT

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 91332

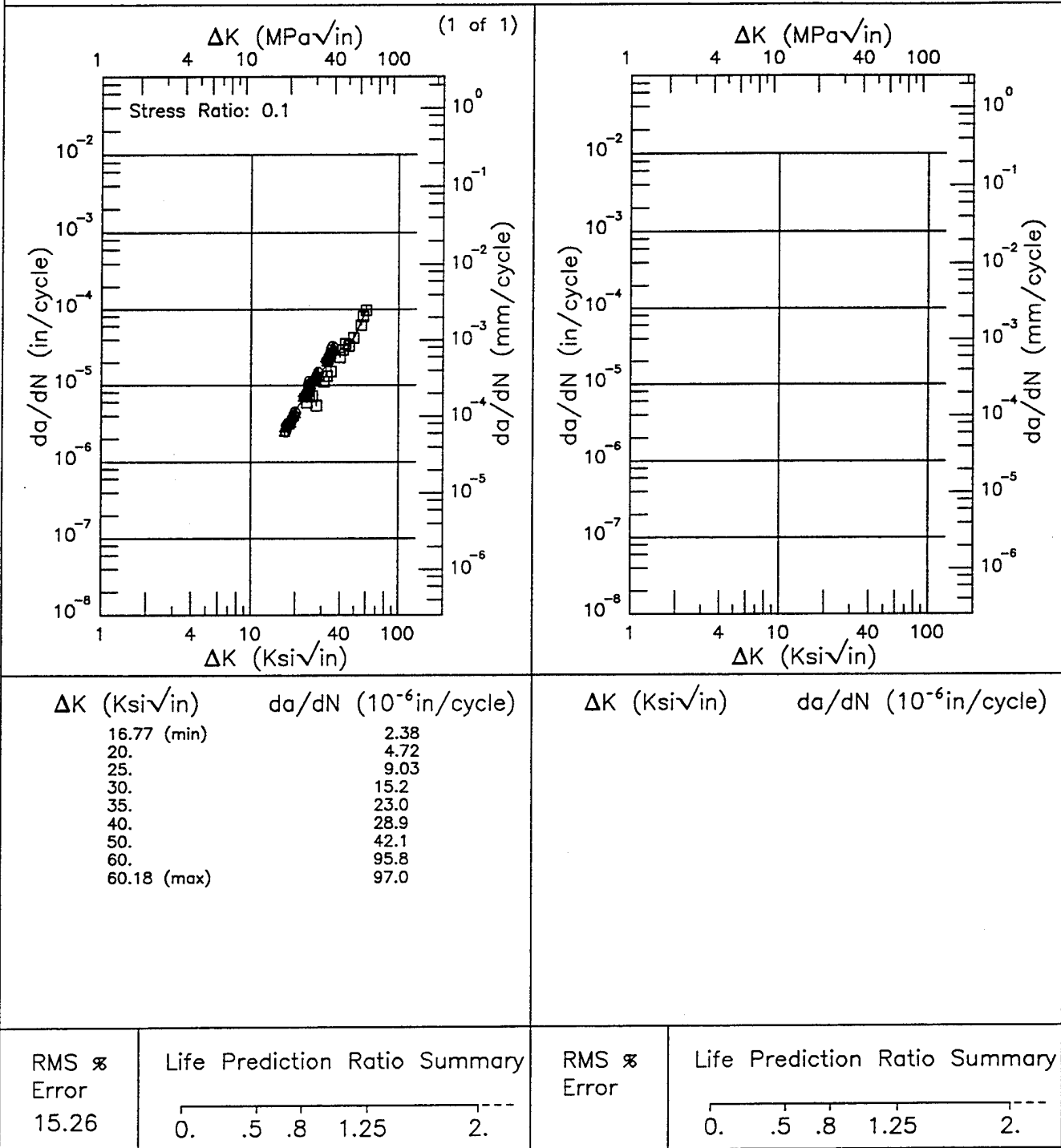


Figure 6.16.3.1.54

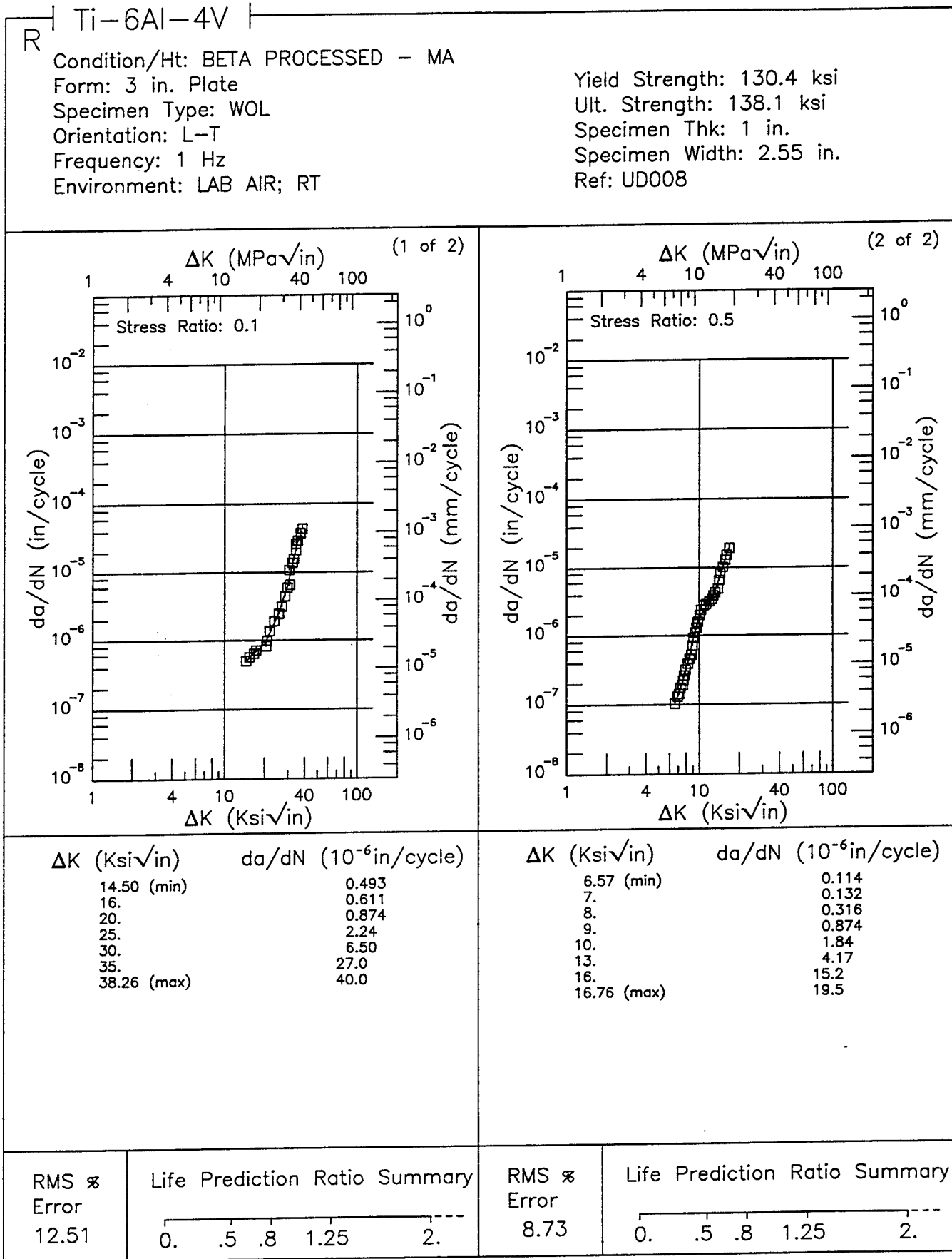
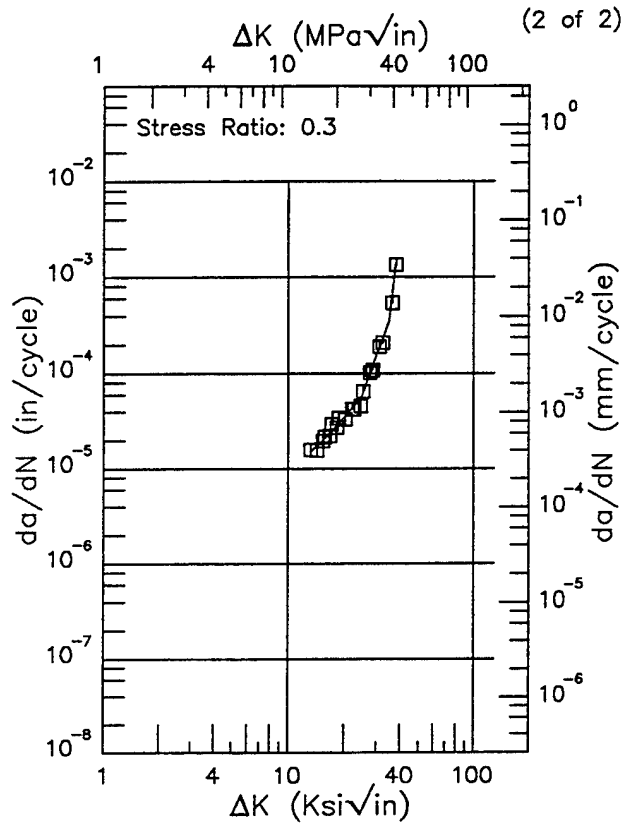
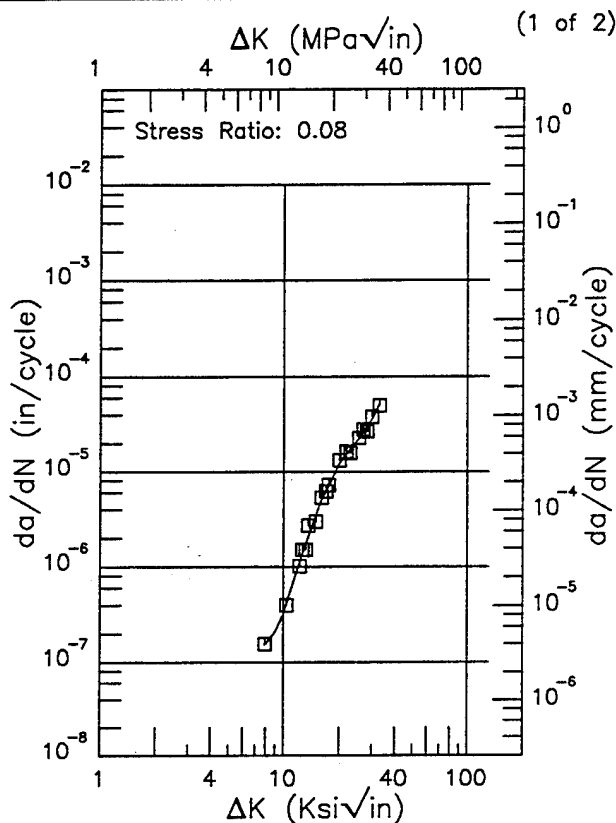


Figure 6.16.3.1.55

Ti-6Al-4V R

Condition/Ht: DB
 Form: 0.63 - 1.5 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Frequency: 6 Hz
 Environment: DRY AIR; RT

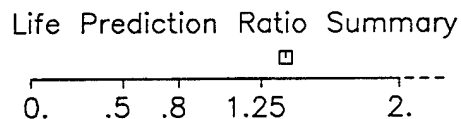
Yield Strength: 115 - 153 ksi
 Ult. Strength: 126 - 165 ksi
 Specimen Thk: 1 in.
 Specimen Width: 4.94 - 6 in.
 Ref: 88579



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.88 (min)	0.153
8.	0.156
9.	0.208
10.	0.331
13.	1.59
16.	5.14
20.	12.5
25.	21.5
30.	36.3
32.73 (max)	50.5

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.23 (min)	14.9
16.	21.1
20.	34.7
25.	57.3
30.	157.
35.	359.
37.82 (max)	1368.

RMS %
 Error
 10.63



RMS %
 Error
 8.89

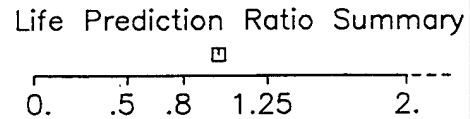


Figure 6.16.3.1.56

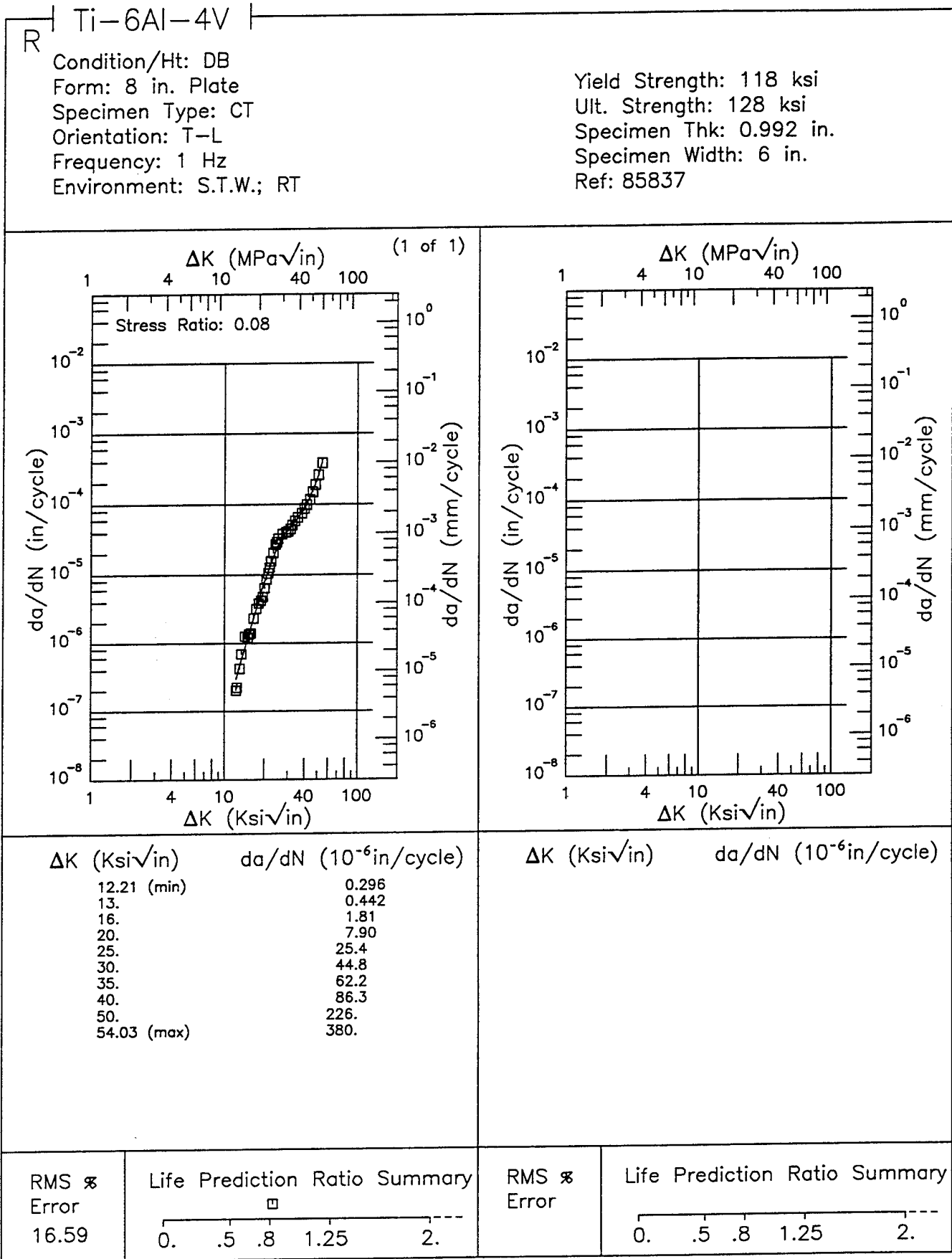


Figure 6.16.3.1.57

Ti-6Al-4V R

Condition/Ht: DB
 Form: 8 in. Plate
 Specimen Type: CT
 Orientation: S-T
 Frequency: 1 Hz
 Environment: S.T.W.; RT

Yield Strength: 114 ksi
 Ult. Strength: 128 ksi
 Specimen Thk: 0.993 in.
 Specimen Width: 4.94 in.
 Ref: 85837

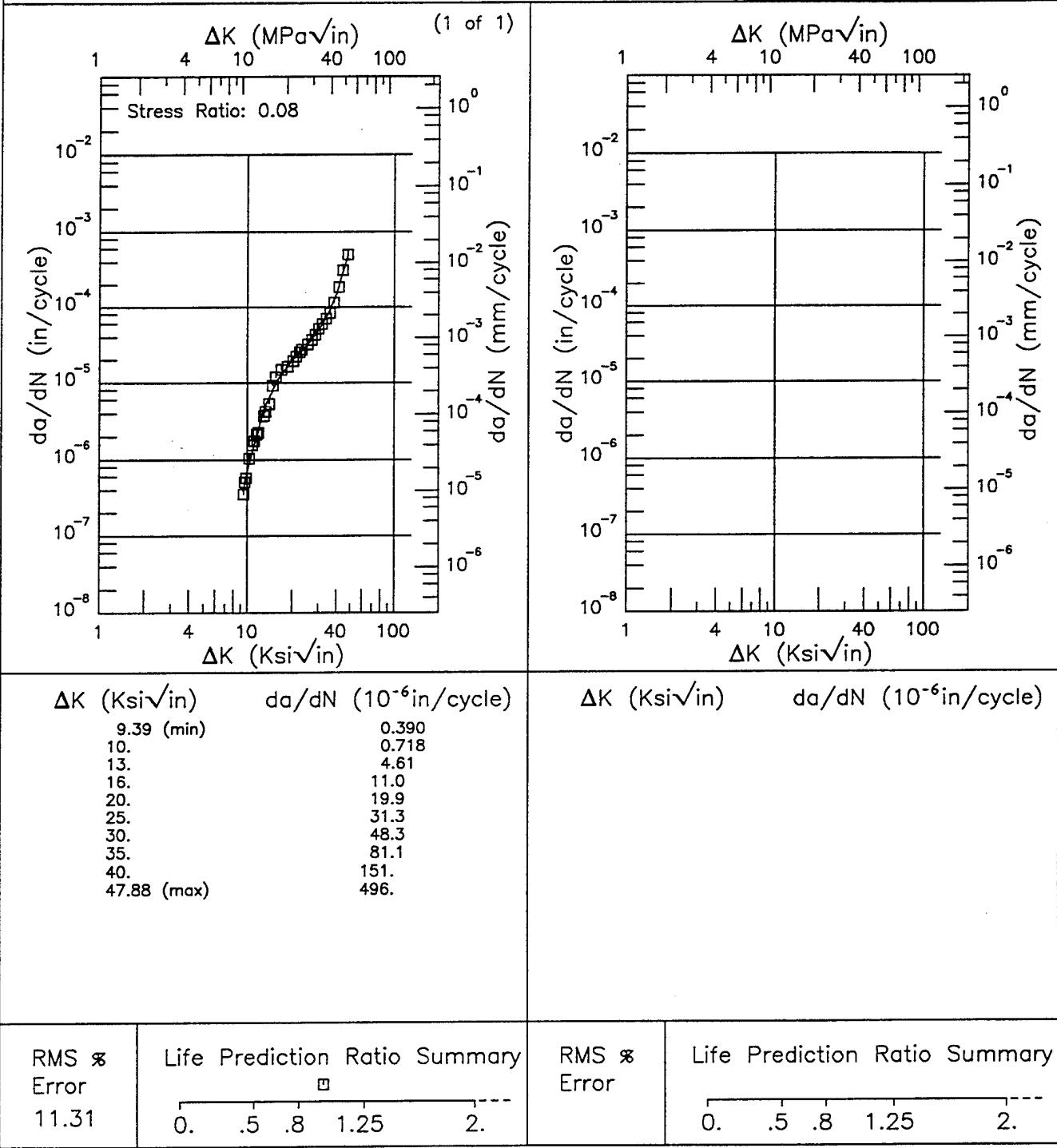
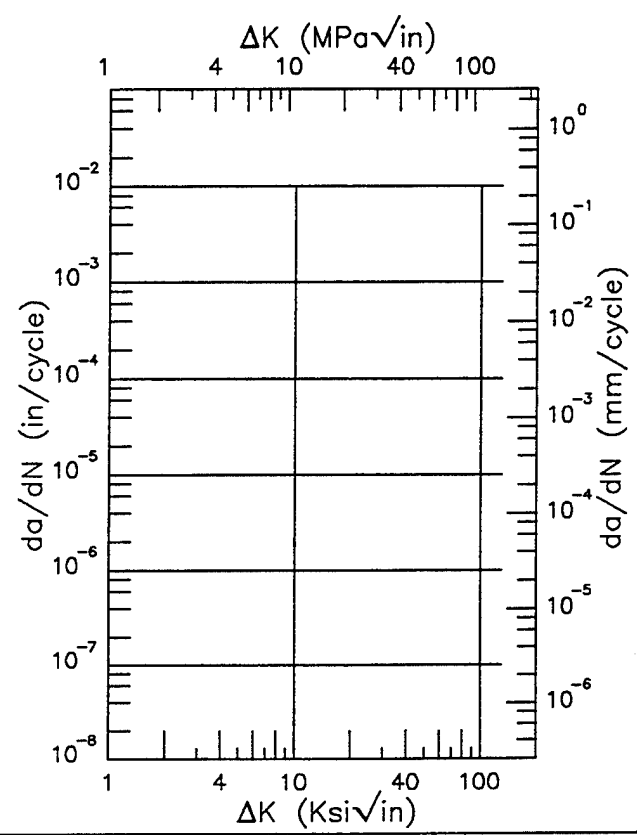
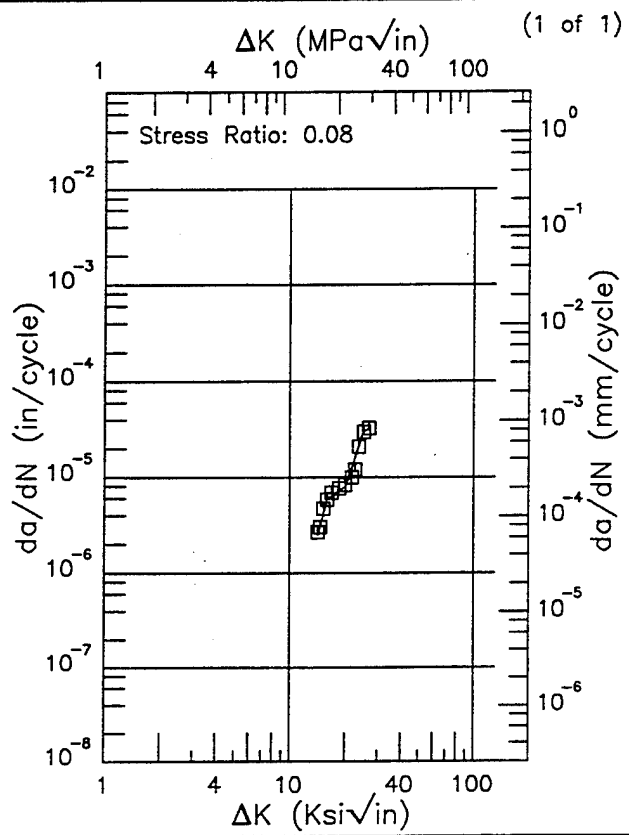


Figure 6.16.3.1.58

Ti-6Al-4V R

Condition/Ht: DB + 2DBTC
 Form: 8 in. Plate
 Specimen Type: CT
 Orientation: S-T
 Frequency: 6 Hz
 Environment: DRY AIR; RT

Yield Strength: 114 ksi
 Ult. Strength: 128 ksi
 Specimen Thk: 0.983 in.
 Specimen Width: 4.95 in.
 Ref: 85837



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
14.18 (min)	2.46
16.	5.88
20.	7.97
25.	29.9
26.83 (max)	32.1

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
14.18 (min)	2.46
16.	5.88
20.	7.97
25.	29.9
26.83 (max)	32.1

RMS % Error 11.46	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.
----------------------	--

RMS % Error	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.
-------------	--

Figure 6.16.3.1.60

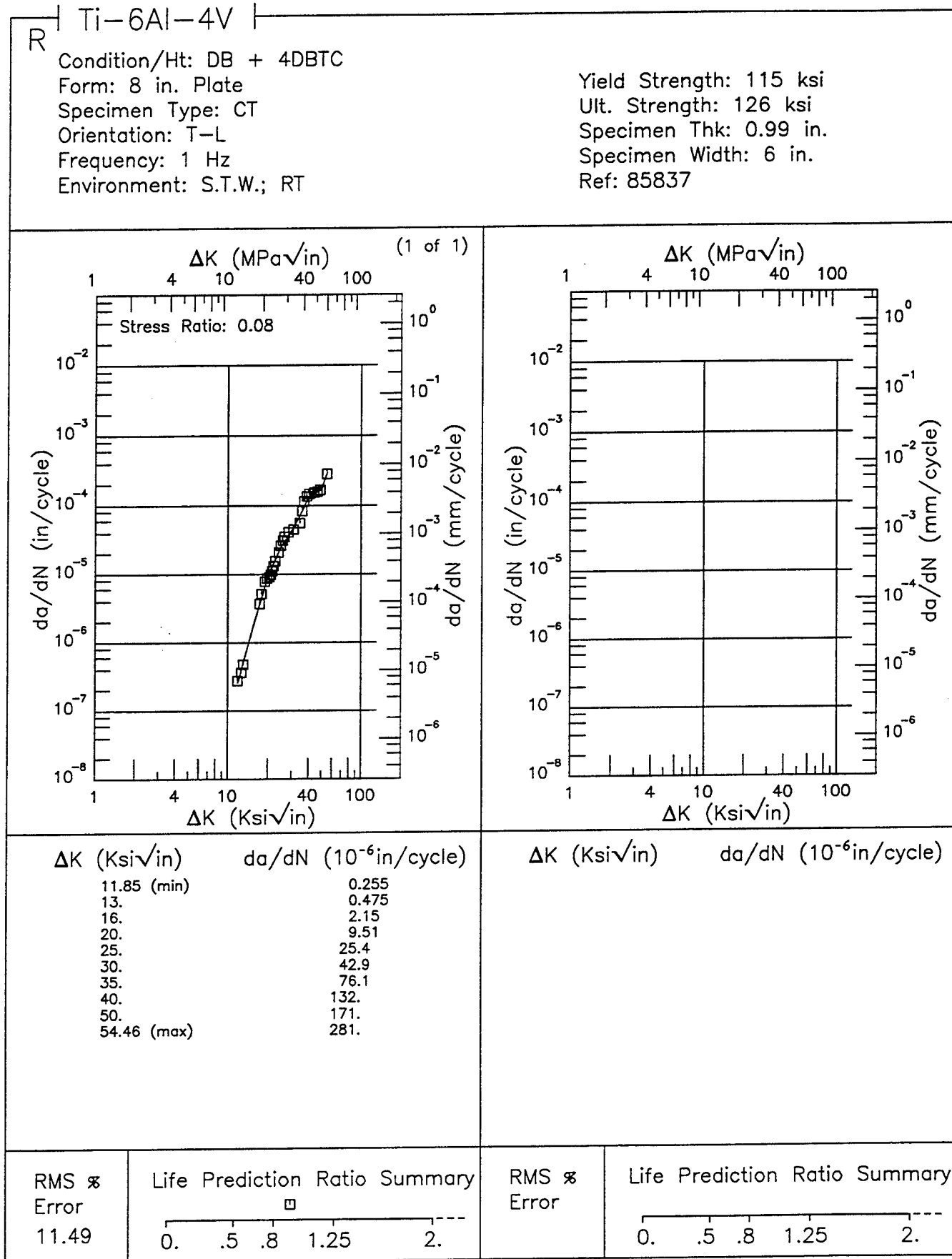


Figure 6.16.3.1.61

Ti-6Al-4V R

Condition/Ht: DB + TR
 Form: 1.5 in. Plate
 Specimen Type: CT
 Orientation: S-T
 Frequency: 1 Hz
 Environment: S.T.W.; RT

Yield Strength: 111 ksi
 Ult. Strength: 128 ksi
 Specimen Thk: 1 in.
 Specimen Width: 5 in.
 Ref: 88579

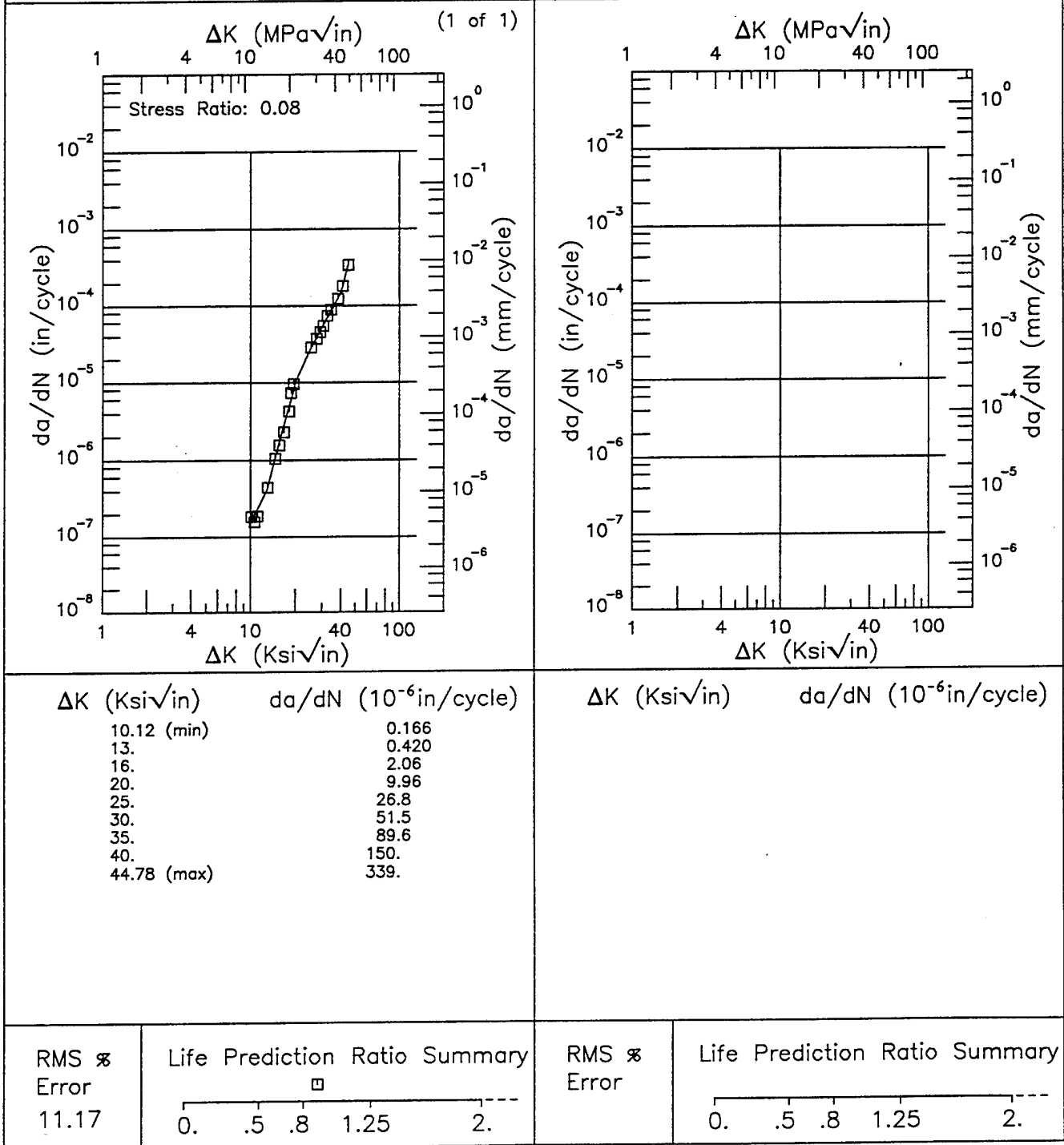


Figure 6.16.3.1.62

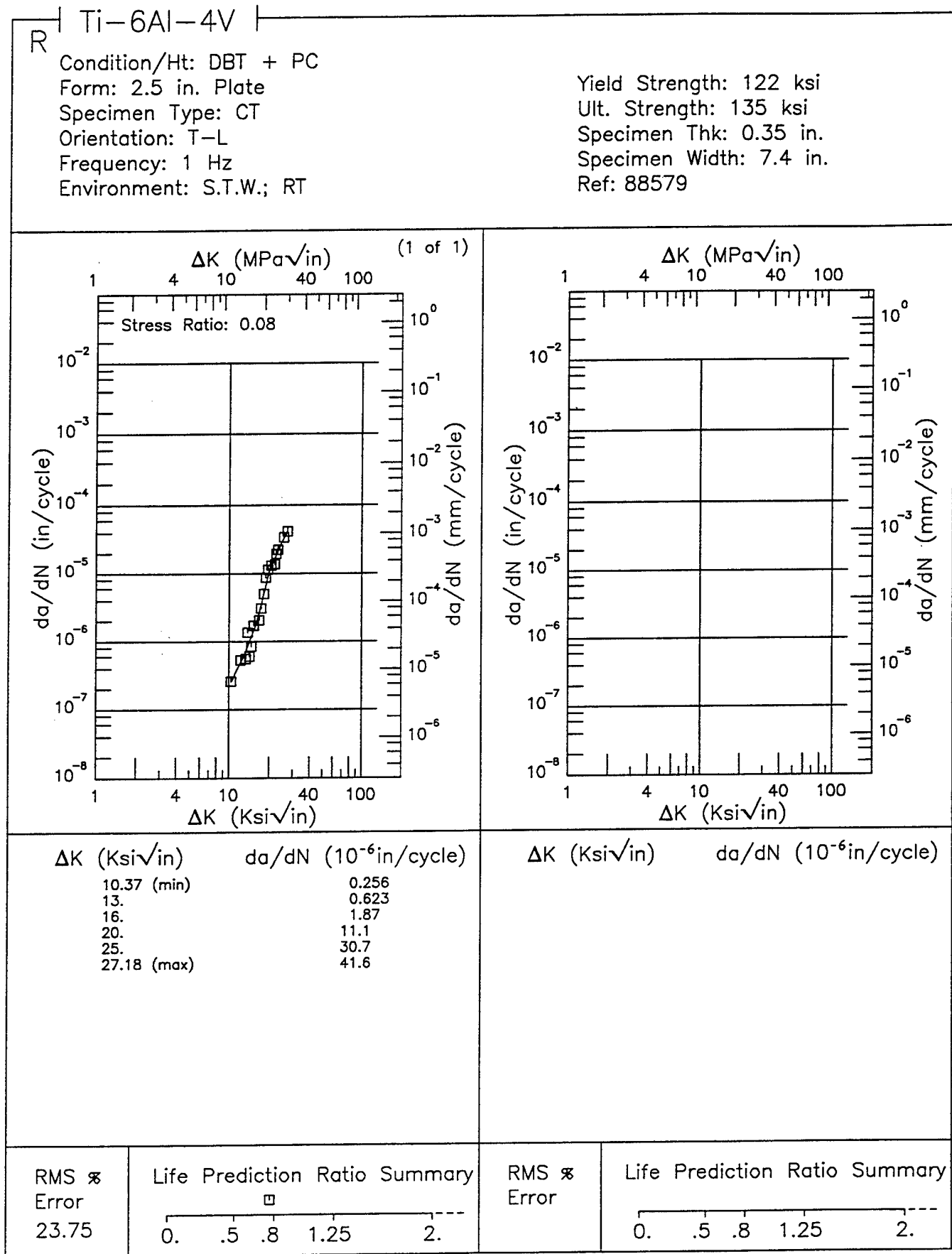


Figure 6.16.3.1.63

Ti-6Al-4V R

Condition/Ht: DBTC
 Form: 0.62 - 0.63 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Frequency: 1 Hz
 Environment: DRY AIR; RT

Yield Strength: 138 - 140 ksi
 Ult. Strength: 148 - 150 ksi
 Specimen Thk: 0.496 - 0.5 in.
 Specimen Width: 6 in.
 Ref: 88579;85837

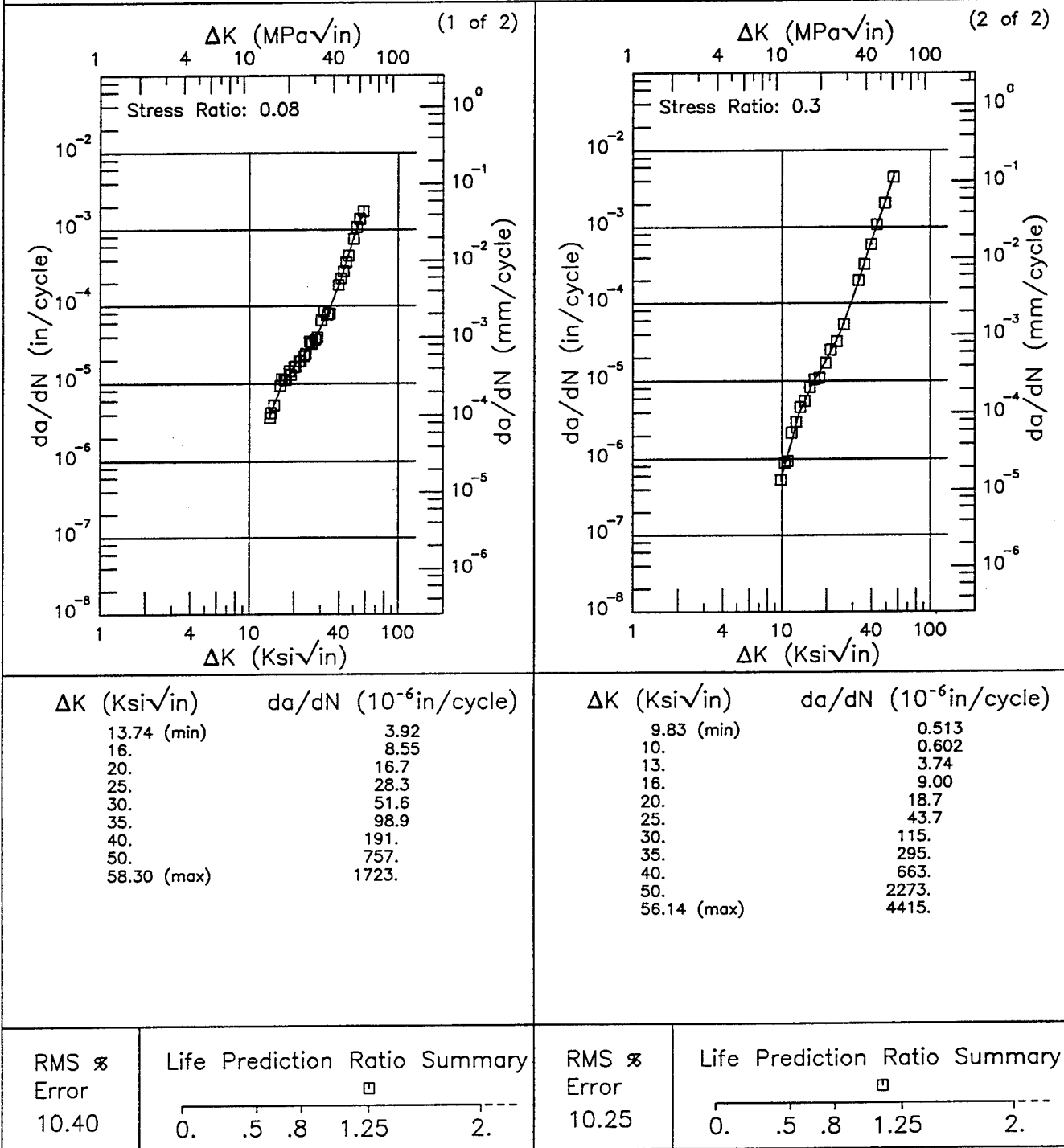


Figure 6.16.3.1.64

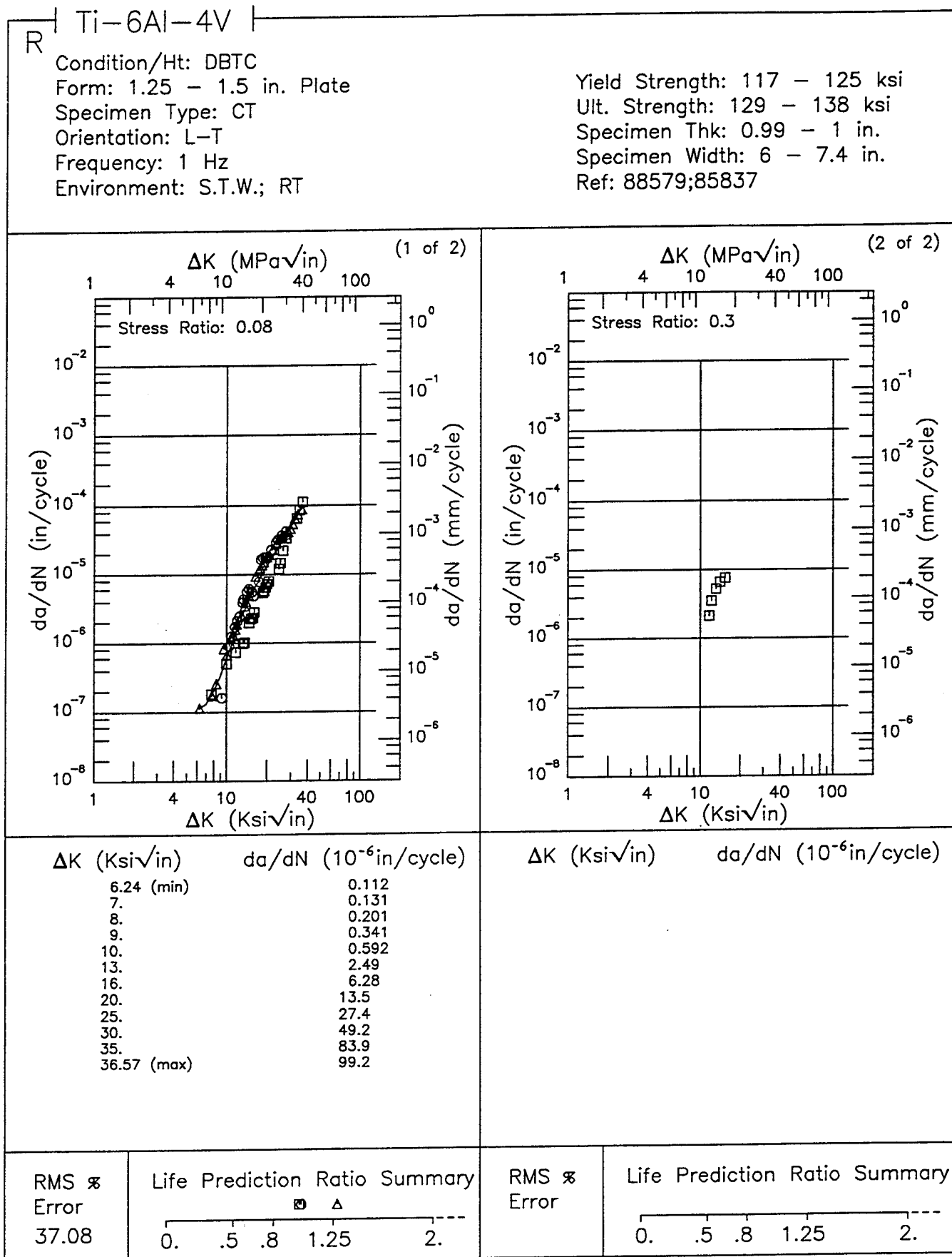
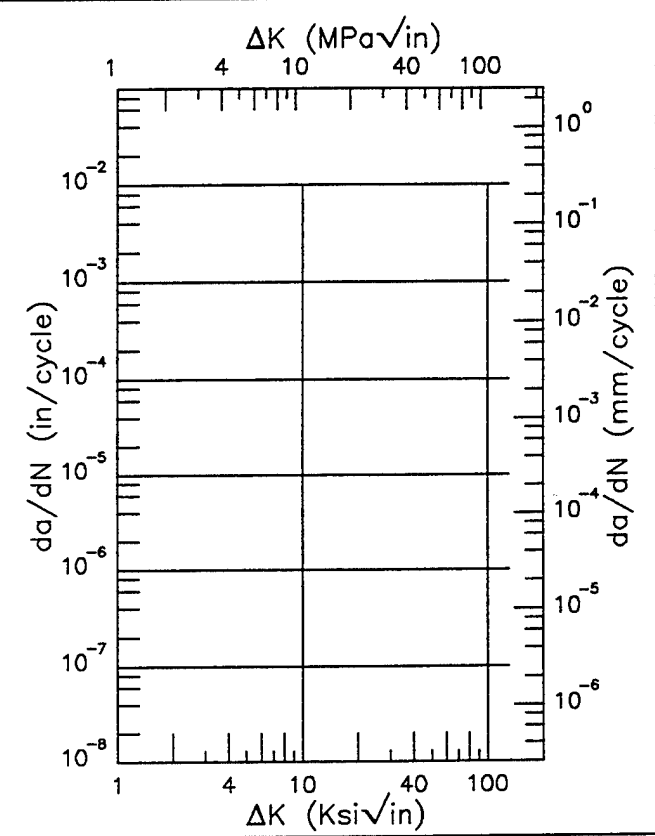
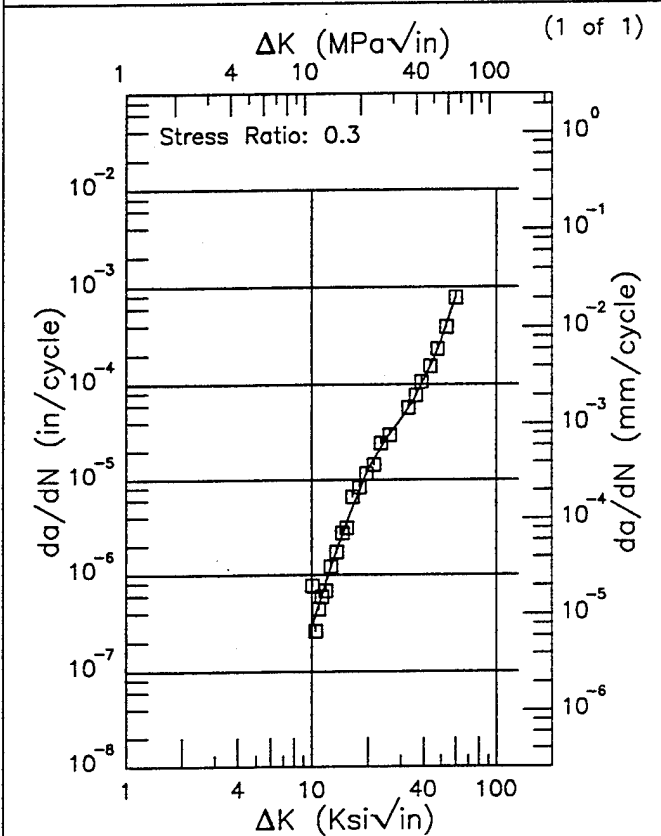


Figure 6.16.3.1.65

Ti-6Al-4V R

Condition/Ht: DBTC
 Form: 1.25 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Frequency: 1 Hz
 Environment: DRY AIR; RT

Yield Strength: 125 ksi
 Ult. Strength: 137 ksi
 Specimen Thk: 1 in.
 Specimen Width: 6 in.
 Ref: 88579



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.00 (min)	0.296
13.	1.39
16.	4.45
20.	12.8
25.	27.3
30.	43.3
35.	67.5
40.	111.
50.	281.
59.95 (max)	782.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.00 (min)	0.296
13.	1.39
16.	4.45
20.	12.8
25.	27.3
30.	43.3
35.	67.5
40.	111.
50.	281.
59.95 (max)	782.

RMS % Error 35.47	Life Prediction Ratio Summary
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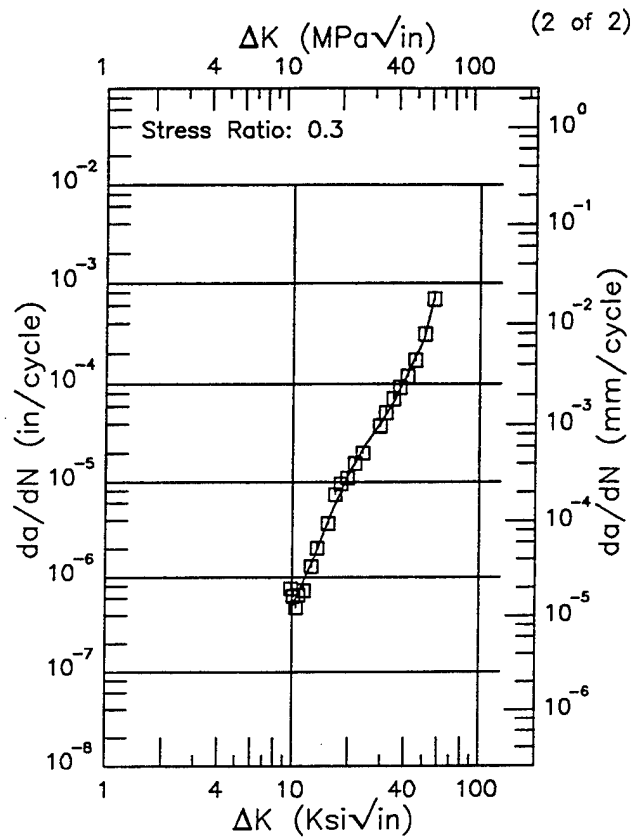
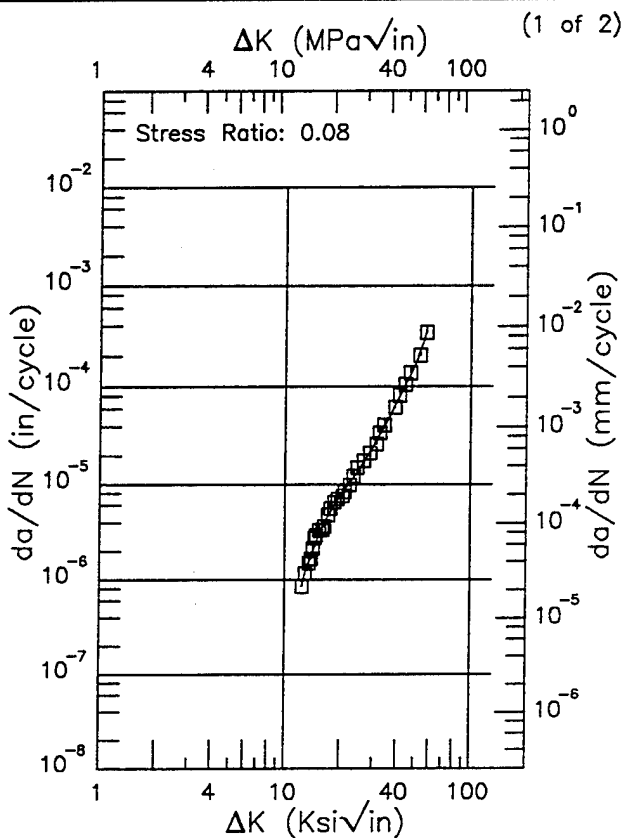
RMS % Error	Life Prediction Ratio Summary
-------------	-----------------------------------

Figure 6.16.3.1.66

R | Ti-6Al-4V |

Condition/Ht: DBTC
 Form: 1.25 - 1.5 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Frequency: 1 Hz
 Environment: DRY AIR; RT

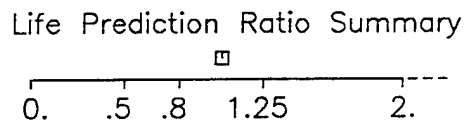
Yield Strength: 119 - 128 ksi
 Ult. Strength: 134 - 139 ksi
 Specimen Thk: 0.96 - 1 in.
 Specimen Width: 6 - 7.46 in.
 Ref: 88579



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
12.36 (min)	0.842
13.	1.22
15.	3.71
20.	7.71
25.	14.1
30.	25.2
35.	43.4
40.	71.4
50.	162.
57.13 (max)	334.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
9.82 (min)	0.458
10.	0.489
13.	1.56
16.	4.42
20.	12.3
25.	25.7
30.	41.8
35.	68.0
40.	113.
50.	276.
56.48 (max)	694.

RMS $\%$
 Error
 8.42



RMS $\%$
 Error
 18.51

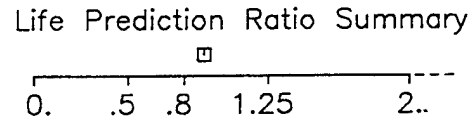


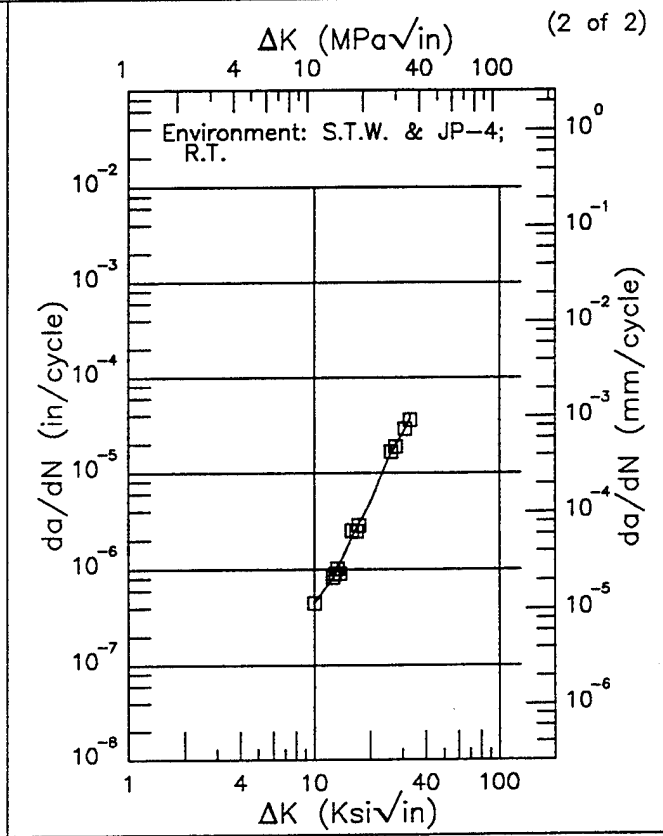
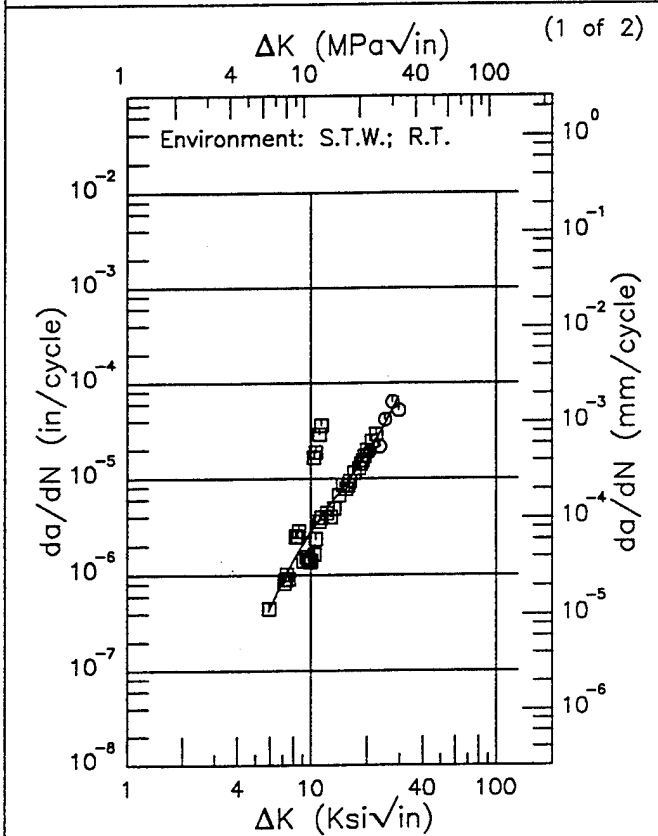
Figure 6.16.3.1.67

Ti-6Al-4V

E

Condition/Ht: DBTC
 Form: 1.5 - 2.5 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.08
 Frequency: 1 Hz

Yield Strength: 119 - 125 ksi
 Ult. Strength: 132 - 135 ksi
 Specimen Thk: 0.962 - 1 in.
 Specimen Width: 4.94 - 7.4 in.
 Ref: 85837;88579



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
5.90 (min)	0.421
6.	0.455
7.	0.866
8.	1.41
9.	2.06
10.	2.81
13.	5.66
16.	9.65
20.	17.9
25.	36.8
29.65 (max)	71.3

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
9.95 (min)	0.449
10.	0.449
13.	0.891
16.	2.19
20.	4.92
25.	13.9
30.	25.9
32.61 (max)	35.8

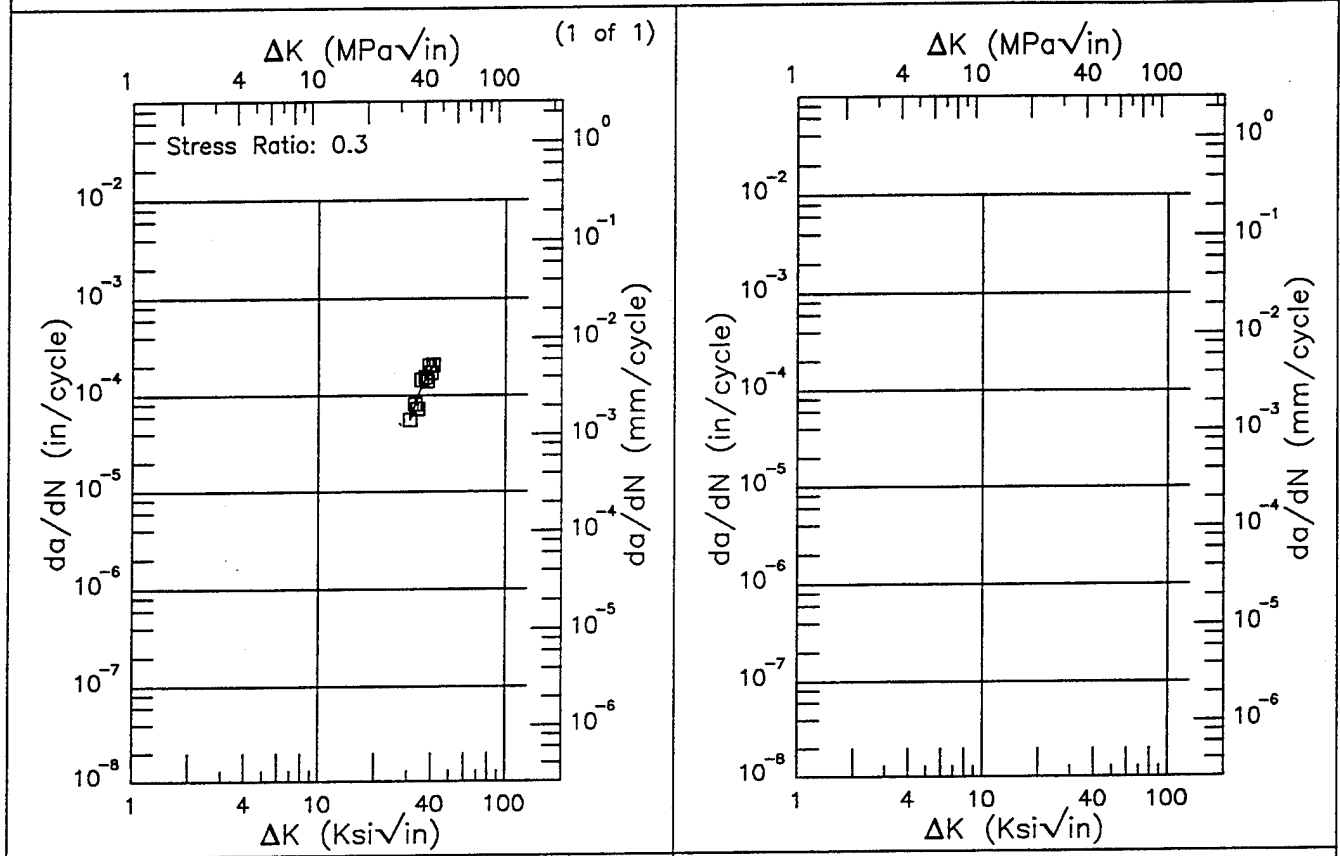
RMS % Error	Life Prediction Ratio Summary
>100.0	

RMS % Error	Life Prediction Ratio Summary
7.49	

Figure 6.16.3.1.68

R | Ti-6Al-4V |
 Condition/Ht: DBTC(RA)
 Form: 0.63 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Frequency: 1 Hz
 Environment: DRY AIR; RT

Yield Strength: 134 ksi
 Ult. Strength: 145 ksi
 Specimen Thk: 0.5 in.
 Specimen Width: 6 in.
 Ref: 88579



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)	ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
30.88 (min)	55.5		
35.	116.		
40.	196.		
40.98 (max)	194.		

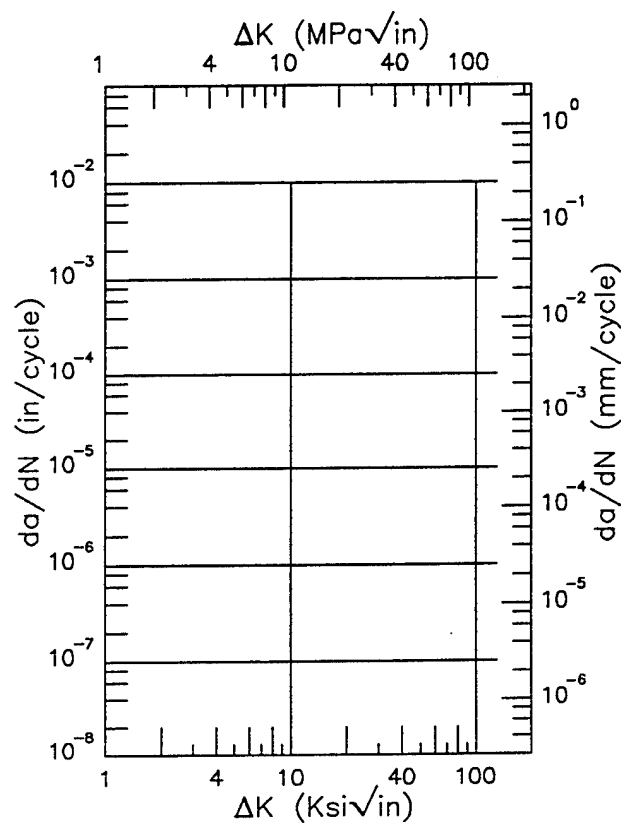
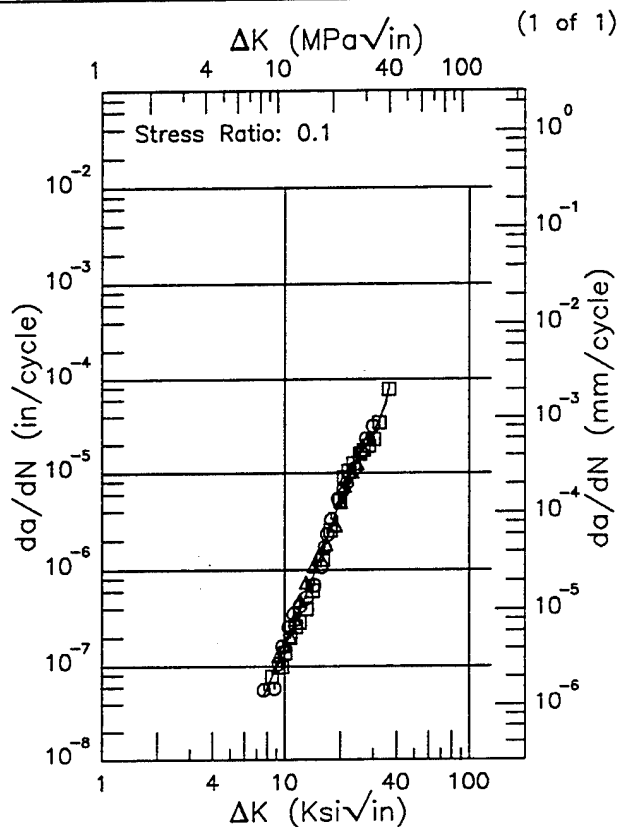
RMS % Error 13.38	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.	RMS % Error	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
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Figure 6.16.3.1.69

Ti-6Al-4V R

Condition/Ht: HIP 1650F 15 KSI
 Form: Casting
 Specimen Type: CT
 Orientation:
 Frequency: 0.1 - 20 Hz
 Environment: LAB AIR; RT

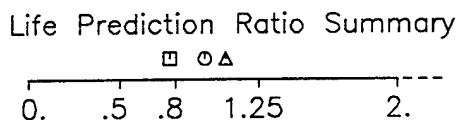
Yield Strength: 60 ksi
 Ult. Strength:
 Specimen Thk: 0.1 in.
 Specimen Width: 0.999 - 1 in.
 Ref: WL011



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
7.64 (min)	0.0490
8.	0.0597
9.	0.0993
10.	0.158
13.	0.529
16.	1.51
20.	5.23
25.	15.3
30.	25.5
35.	55.0
36.32 (max)	78.0

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
-------------------------------	-------------------------------

RMS %
 Error
 17.83



RMS %
 Error

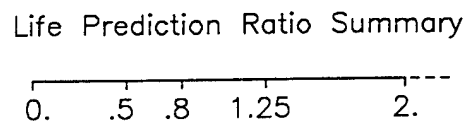


Figure 6.16.3.1.70

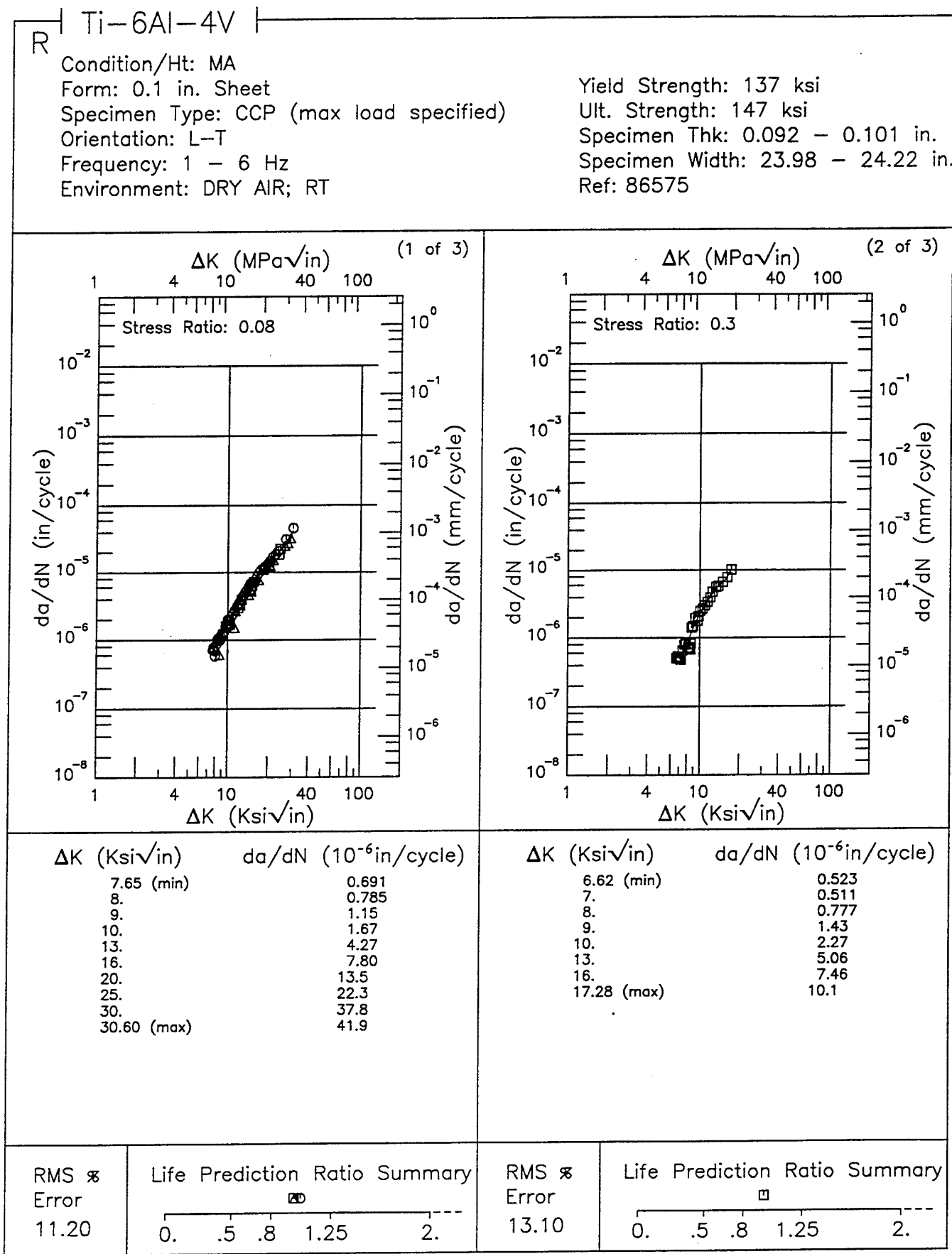


Figure 6.16.3.1.71

Ti-6Al-4V

R

Condition/Ht: MA
 Form: 0.1 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 1 - 6 Hz
 Environment: DRY AIR; RT

Yield Strength: 137 ksi
 Ult. Strength: 147 ksi
 Specimen Thk: 0.092 - 0.101 in.
 Specimen Width: 23.98 - 24.22 in.
 Ref: 86575

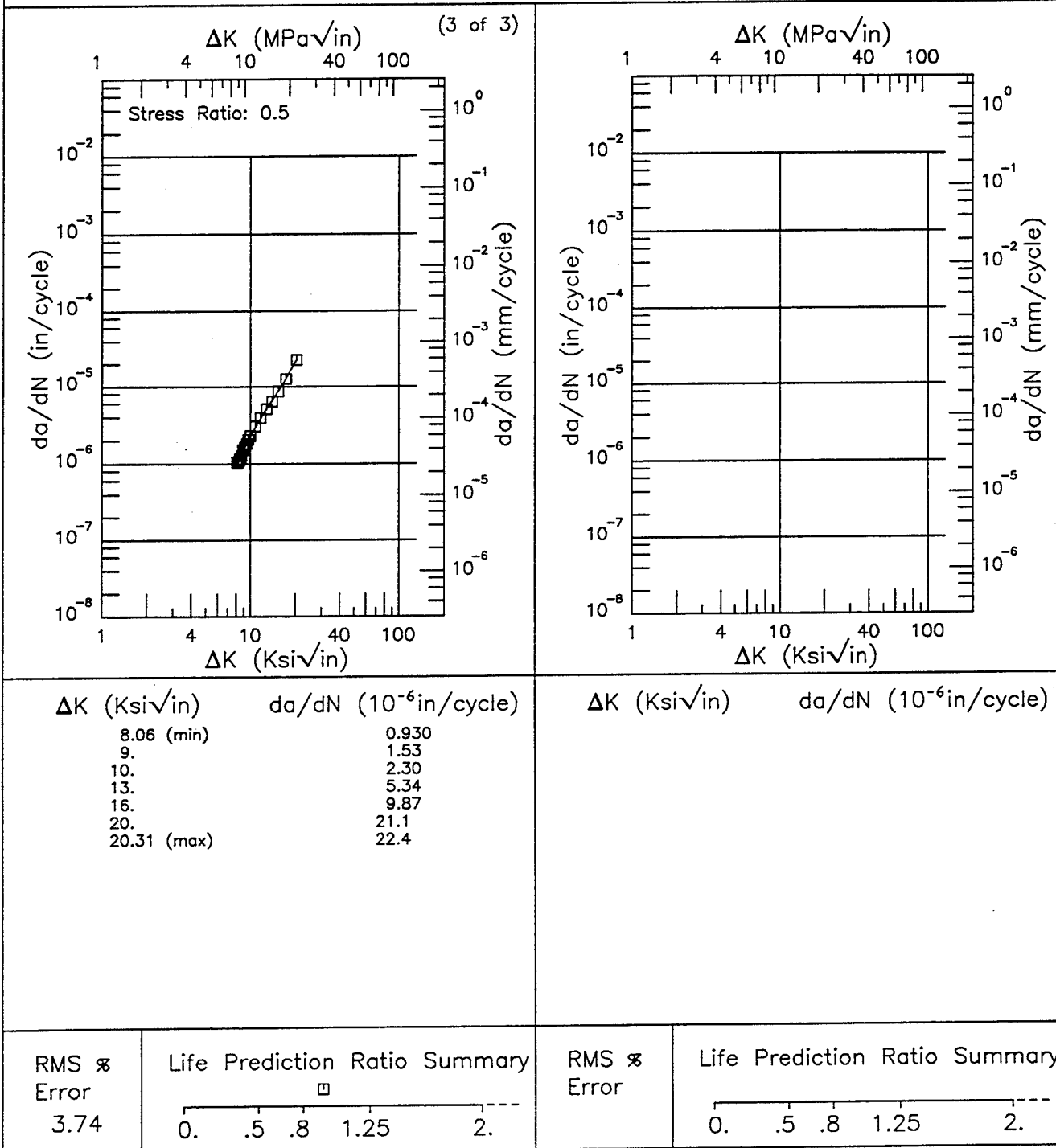


Figure 6.16.3.1.71 (Concluded)

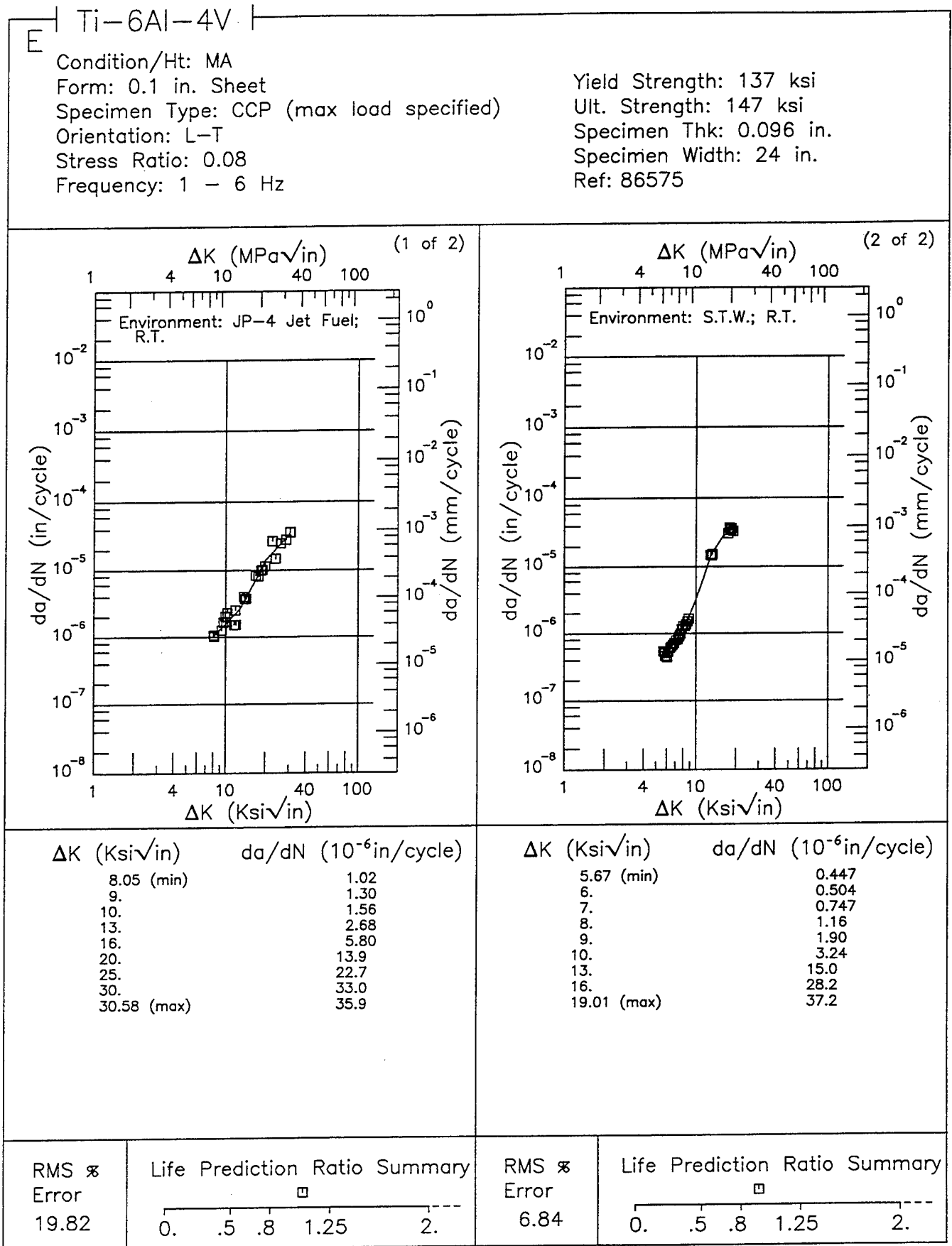


Figure 6.16.3.1.72

Ti-6Al-4V E

Condition/Ht: MA
 Form: 0.1 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: T-L
 Stress Ratio: 0.08
 Frequency: 6 Hz

Yield Strength: 143 ksi
 Ult. Strength: 151 ksi
 Specimen Thk: 0.096 - 0.097 in.
 Specimen Width: 24 in.
 Ref: 86575

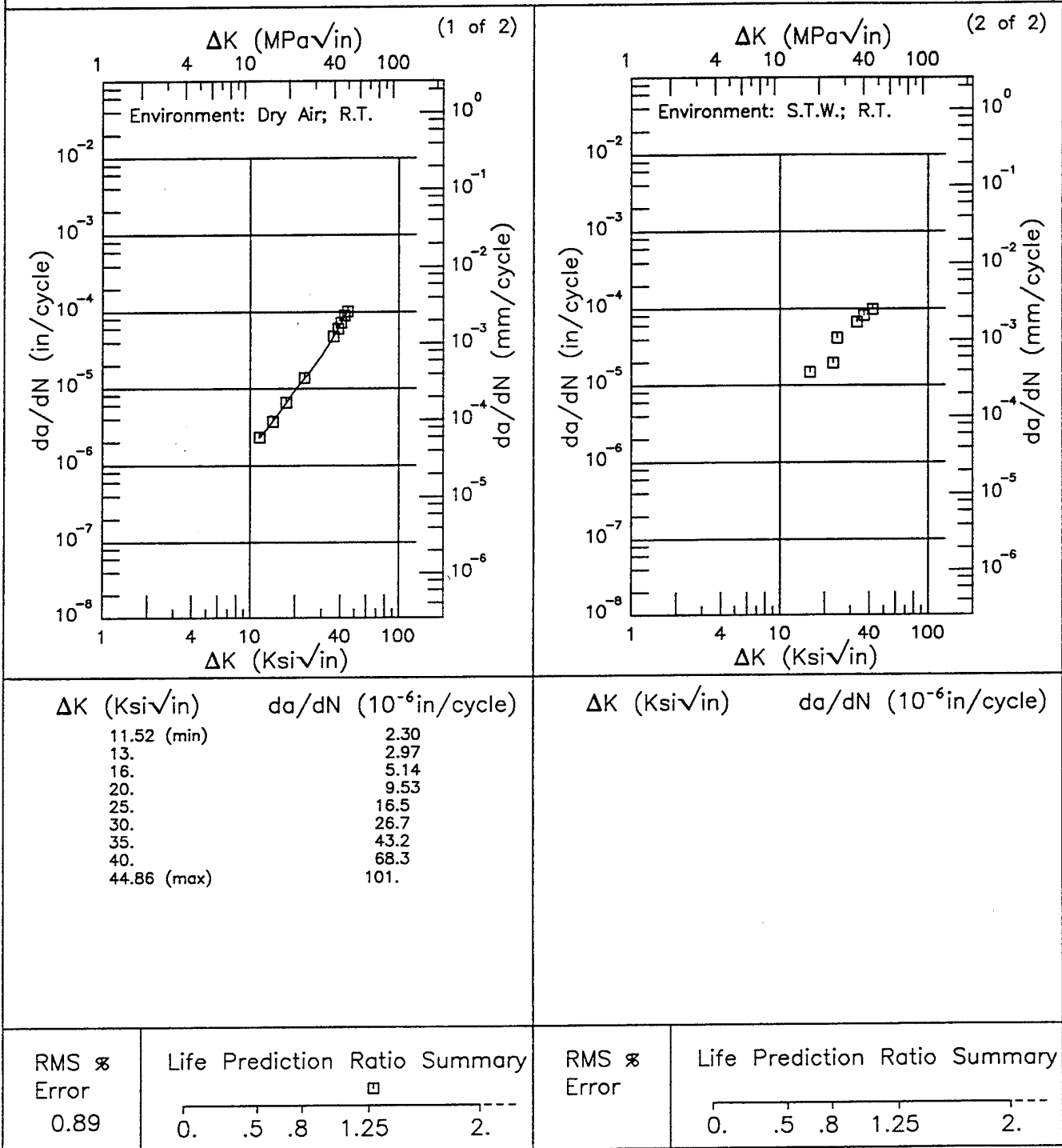


Figure 6.16.3.1.73

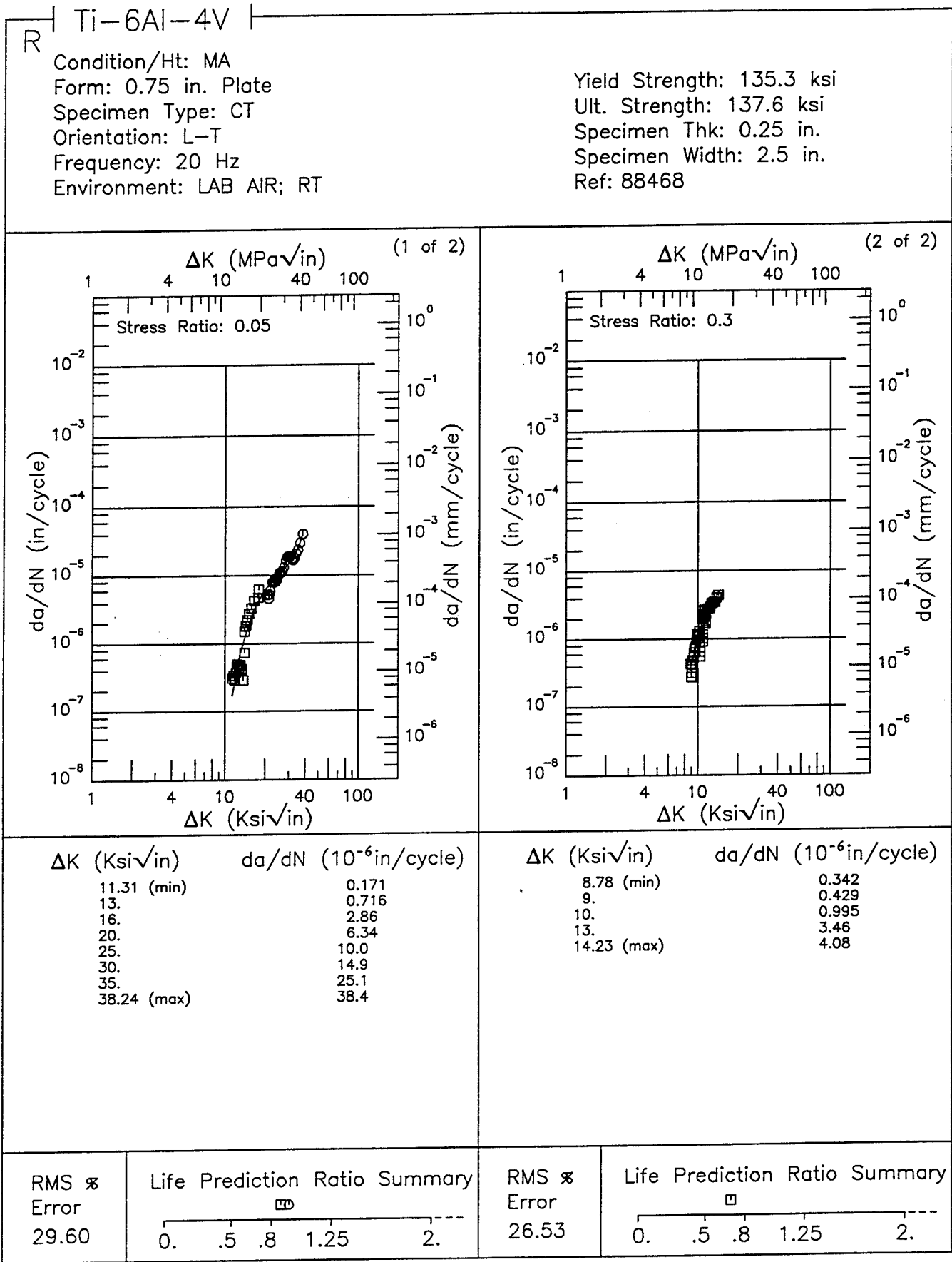


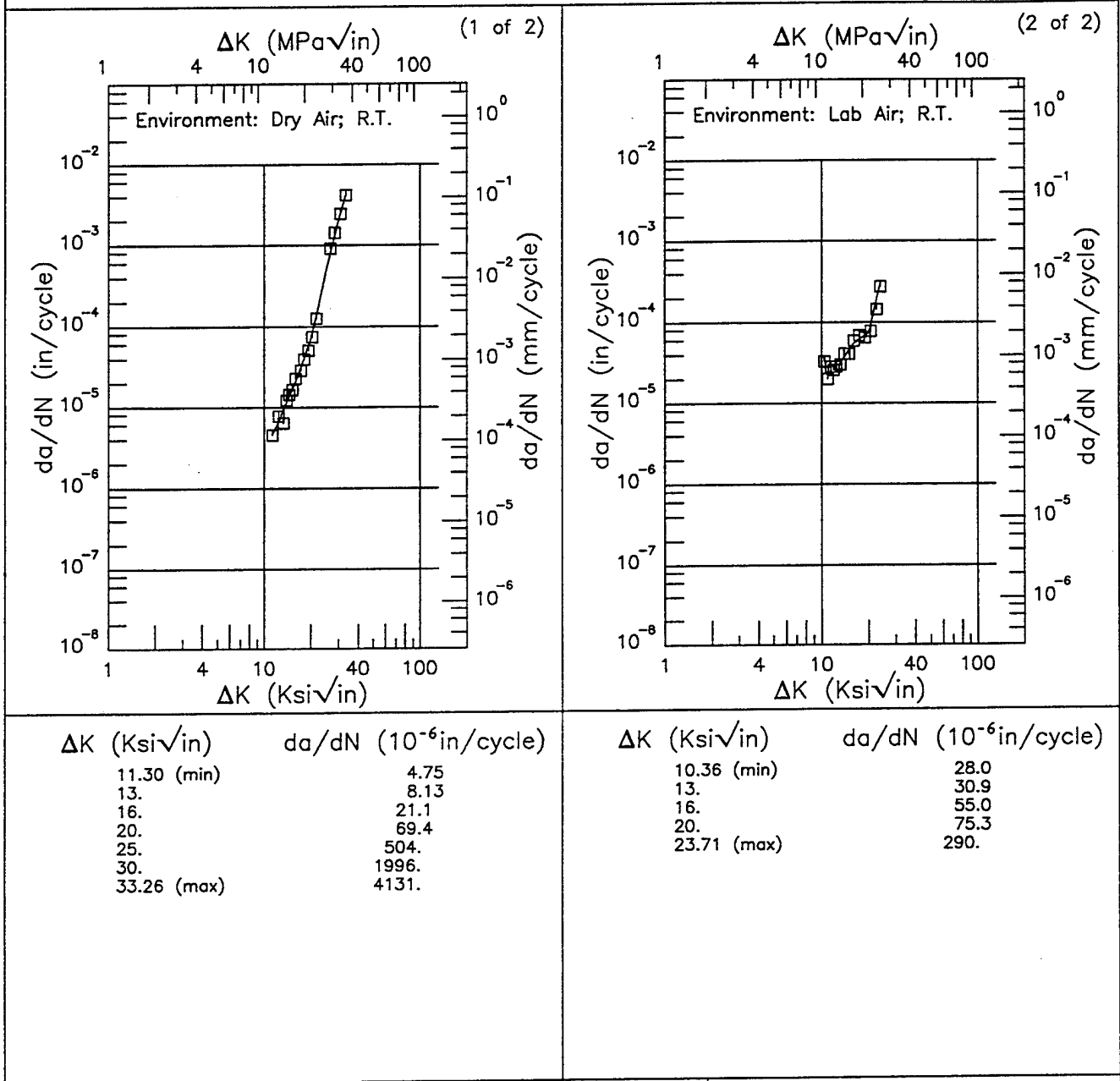
Figure 6.16.3.1.74

Ti-6Al-4V

E

Condition/Ht: MA
 Form: 0.63 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.3
 Frequency: 1 Hz

Yield Strength: 138 ksi
 Ult. Strength: 148 ksi
 Specimen Thk: 0.49 - 0.5 in.
 Specimen Width: 6 in.
 Ref: 88579



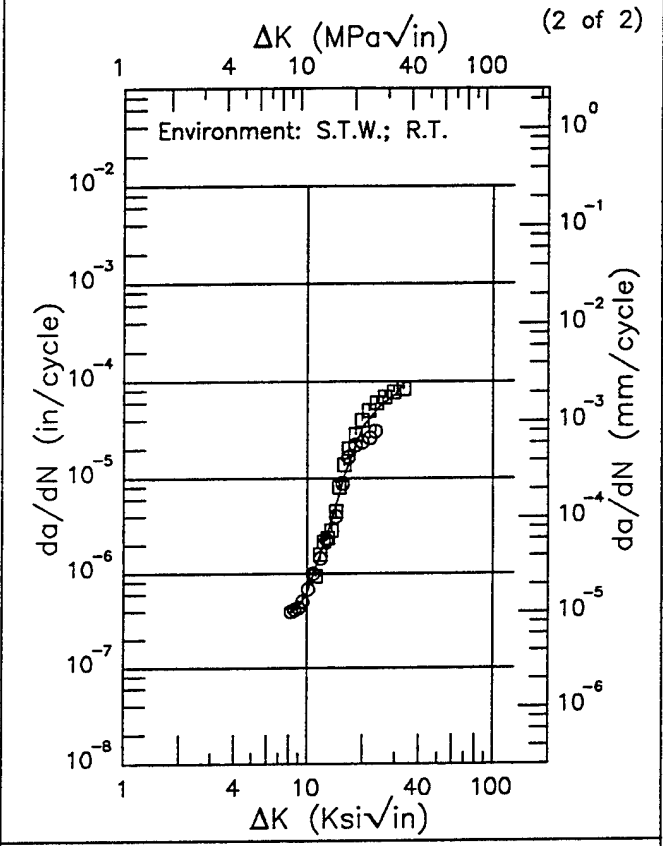
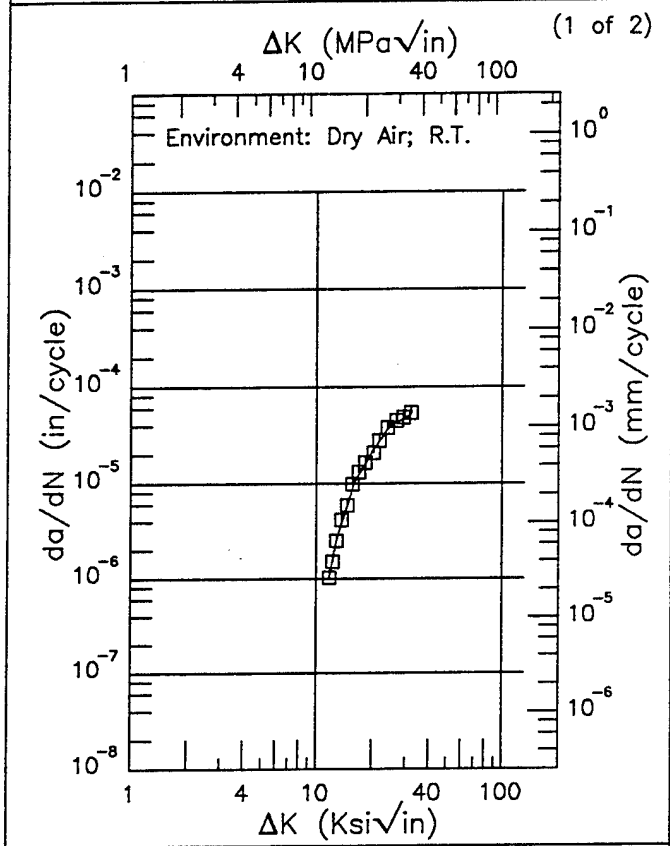
RMS % Error 10.90	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.	RMS % Error 14.05	Life Prediction Ratio Summary □ 0. .5 .8 1.25 2.
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Figure 6.16.3.1.75

E | **Ti-6Al-4V** |

Condition/Ht: MA
 Form: 1.25 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.3
 Frequency: 1 Hz

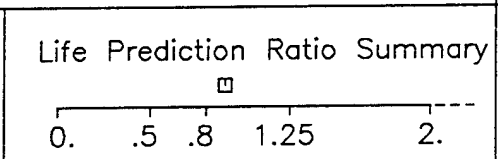
Yield Strength: 120 ksi
 Ult. Strength: 134 ksi
 Specimen Thk: 0.99 - 0.997 in.
 Specimen Width: 6 - 6.01 in.
 Ref: 85837



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
11.70 (min)	1.00
13.	2.76
16.	10.4
20.	21.6
25.	40.9
30.	47.5
32.09 (max)	53.8

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.19 (min)	0.388
9.	0.443
10.	0.629
13.	2.75
16.	10.7
20.	34.2
25.	57.1
30.	74.2
32.78 (max)	92.0

RMS % Error
5.42



RMS % Error
20.41

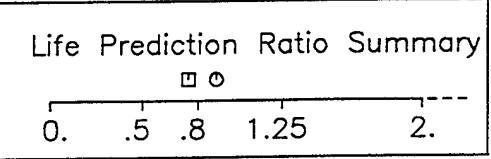
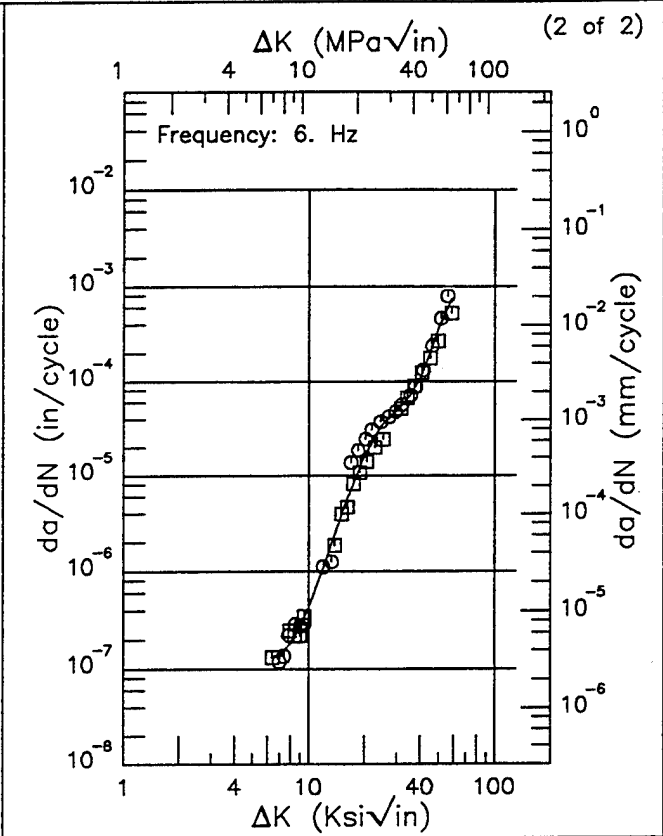
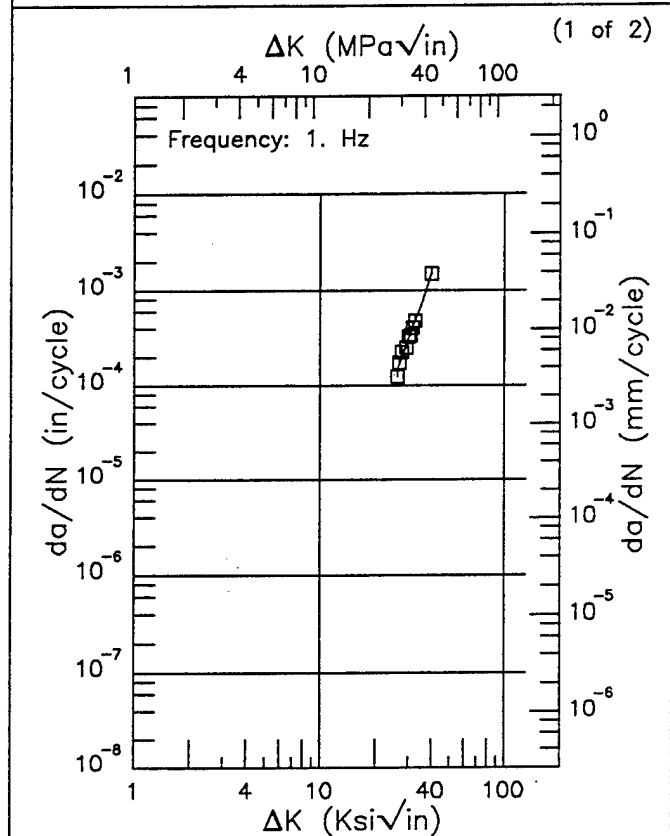


Figure 6.16.3.1.76

Ti-6Al-4V F

Condition/Ht: MA
 Form: 0.63 - 1.5 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.3
 Environment: DRY AIR; RT

Yield Strength: 120 - 138 ksi
 Ult. Strength: 134 - 148 ksi
 Specimen Thk: 0.67 - 1 in.
 Specimen Width: 6 in.
 Ref: 88579;84361;85837



ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
26.05 (min)	144.
30.	313.
35.	679.
39.74 (max)	1469.

ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
6.36 (min)	0.131
7.	0.139
8.	0.183
9.	0.272
10.	0.426
13.	1.67
16.	5.32
20.	16.1
25.	34.5
30.	49.6
35.	68.8
40.	108.
50.	361.
58.57 (max)	747.

RMS % Error	Life Prediction Ratio Summary
7.17	

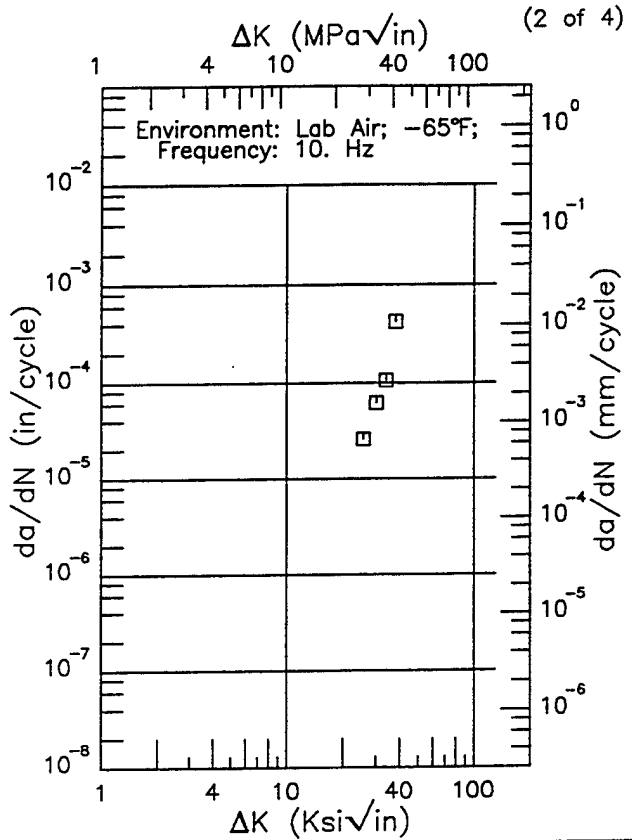
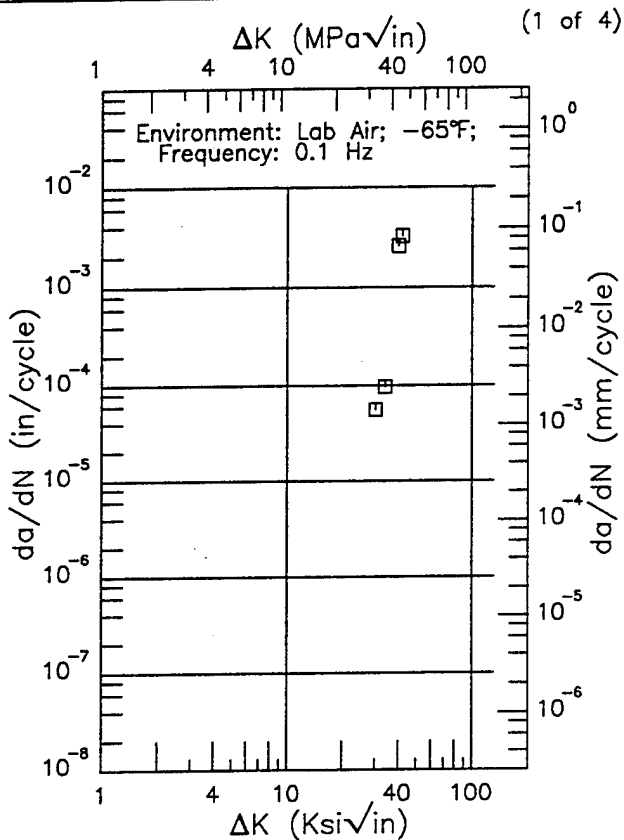
RMS % Error	Life Prediction Ratio Summary
26.46	

Figure 6.16.3.1.77

EF | Ti-6Al-4V |

Condition/Ht: MA
 Form: 1 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1

Yield Strength:
 Ult. Strength:
 Specimen Thk: 1 in.
 Specimen Width: 2.55 in.
 Ref: 88144

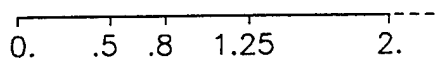


ΔK (Ksi√in) da/dN (10⁻⁶in/cycle)

ΔK (Ksi√in) da/dN (10⁻⁶in/cycle)

RMS %
 Error

Life Prediction Ratio Summary



RMS %
 Error

Life Prediction Ratio Summary

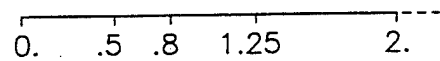
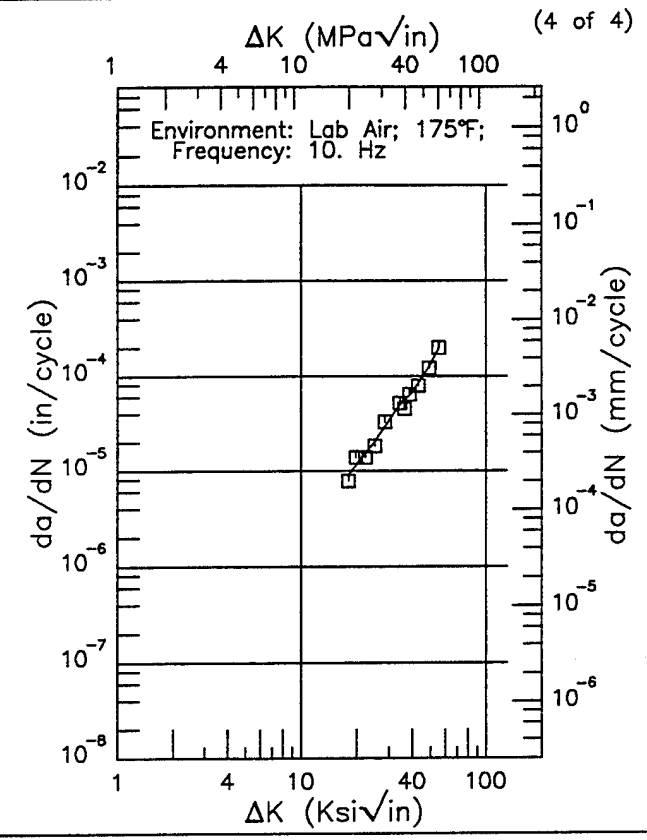
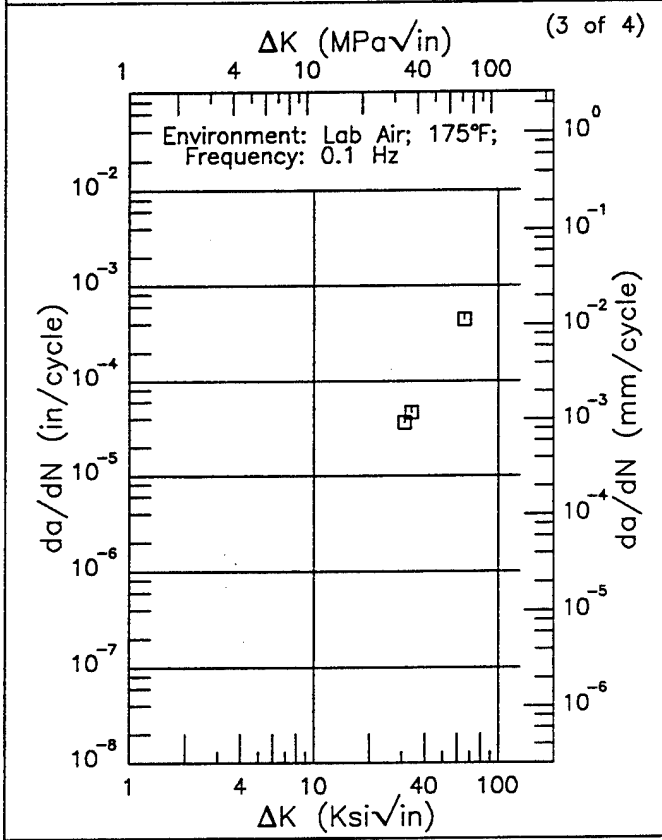


Figure 6.16.3.1.78

Ti-6Al-4V EF

Condition/Ht: MA
 Form: 1 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1

Yield Strength:
 Ult. Strength:
 Specimen Thk: 1 in.
 Specimen Width: 2.55 in.
 Ref: 88144



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
17.93 (min)	8.94
20.	11.7
25.	20.0
30.	33.9
35.	52.4
40.	66.7
50.	134.
55.07 (max)	197.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
17.93 (min)	8.94
20.	11.7
25.	20.0
30.	33.9
35.	52.4
40.	66.7
50.	134.
55.07 (max)	197.

RMS \times Error	Life Prediction Ratio Summary 0. .5 .8 1.25 2.-----
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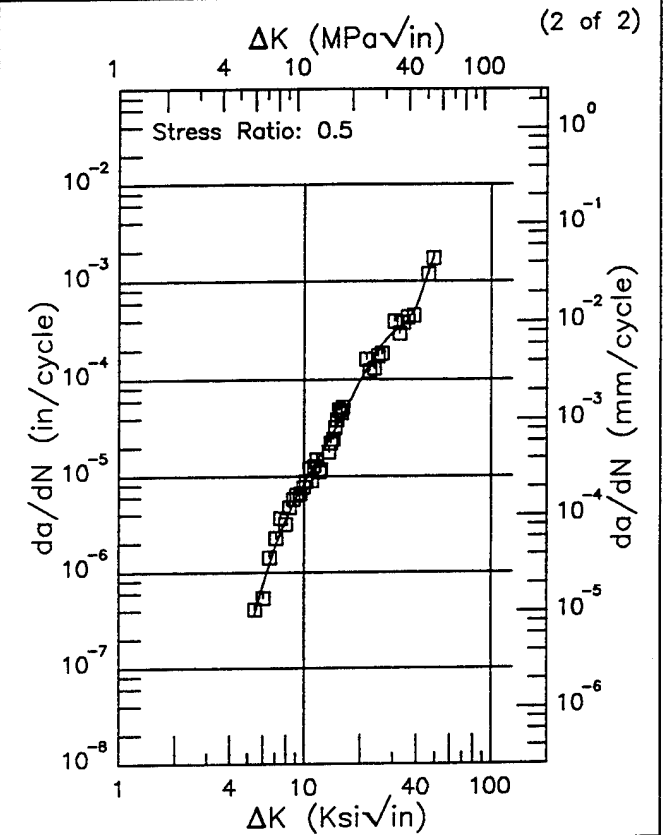
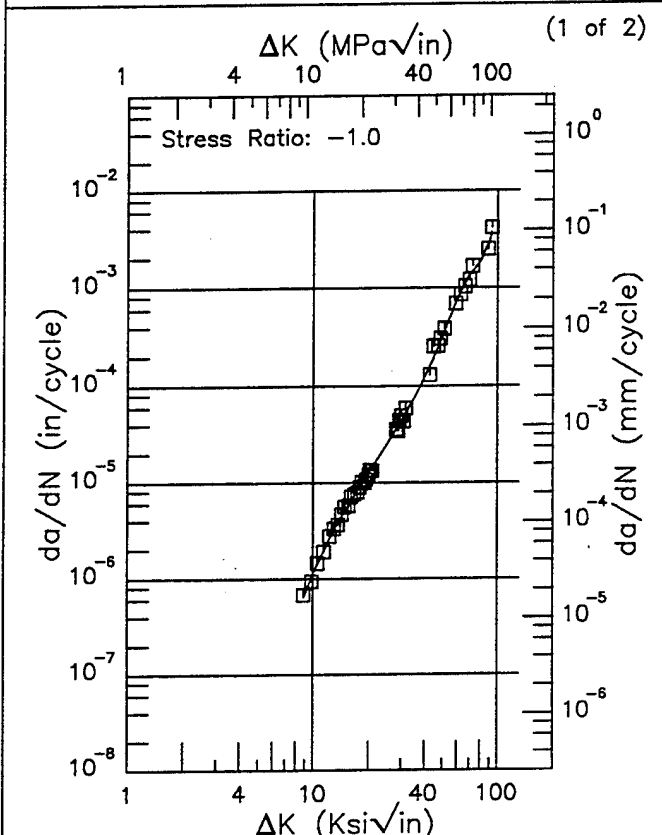
RMS \times Error 11.38	Life Prediction Ratio Summary 0. .5 .8 1.25 2.-----
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Figure 6.16.3.1.78 (Concluded)

R | Ti-6Al-4V |

Condition/Ht: MA
 Form: 1 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 10 Hz
 Environment: LAB AIR; RT

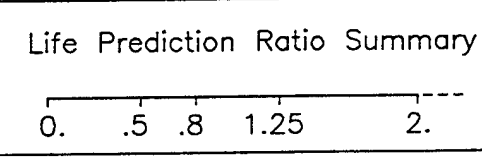
Yield Strength: 132.5 ksi
 Ult. Strength: 138 ksi
 Specimen Thk: 0.25 in.
 Specimen Width: 4 in.
 Ref: MA006



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.85 (min)	0.640
9.	0.695
10.	1.14
13.	3.25
16.	6.53
20.	12.9
25.	25.1
30.	44.4
35.	75.1
40.	124.
50.	323.
60.	767.
70.	1356.
80.	1789.
90.	2992.
92.48 (max)	3717.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
5.43 (min)	0.359
6.	0.727
7.	1.82
8.	3.50
9.	5.74
10.	8.51
13.	20.4
16.	41.3
20.	97.2
25.	208.
30.	300.
35.	379.
40.	552.
49.27 (max)	1771.

RMS % Error
9.11



RMS % Error
18.52

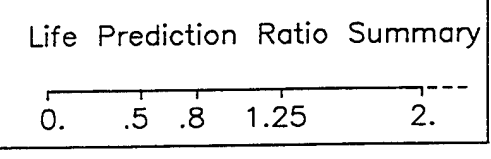
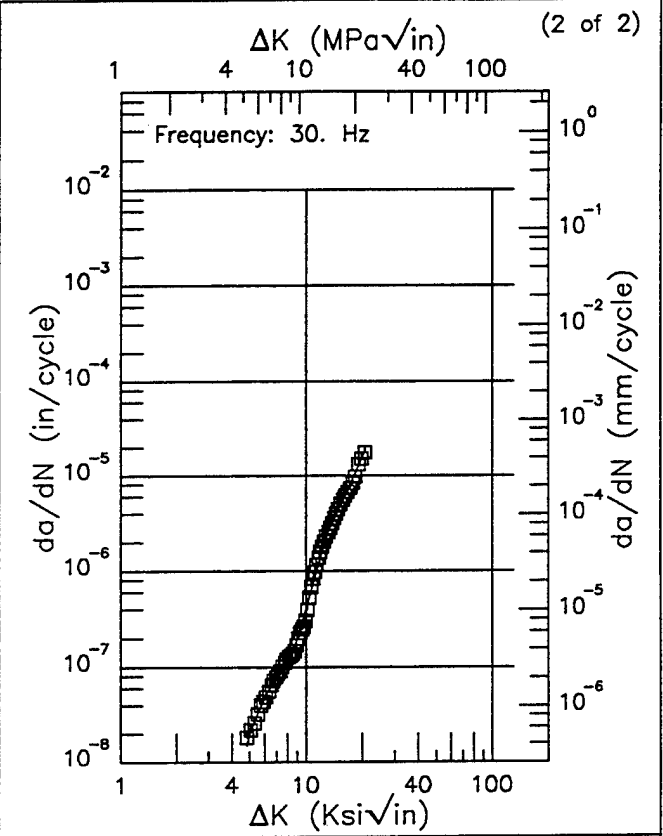
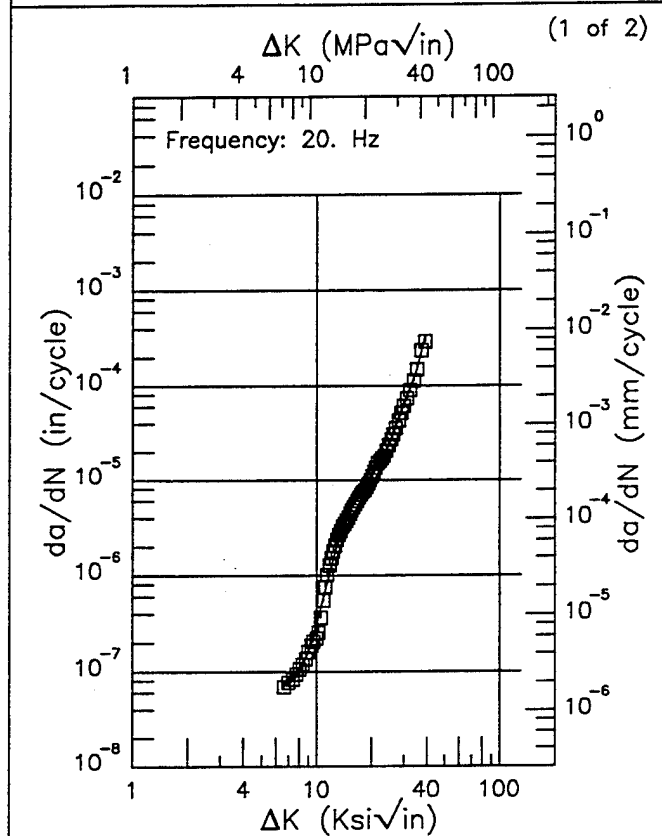


Figure 6.16.3.1.79

Ti-6Al-4V F

Condition/Ht: MA
 Form: 0.38 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.02
 Environment: LAB AIR; RT

Yield Strength: 137 ksi
 Ult. Strength: 145 ksi
 Specimen Thk: 0.241 - 0.242 in.
 Specimen Width: 3.952 - 3.953 in.
 Ref: MA002



ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
6.63 (min)	0.0805
7.	0.0750
8.	0.0933
9.	0.161
10.	0.317
13.	2.08
16.	5.96
20.	10.8
25.	24.1
30.	62.8
35.	144.
38.83 (max)	322.

ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
4.82 (min)	0.0149
5.	0.0198
6.	0.0494
7.	0.0759
8.	0.113
9.	0.198
10.	0.404
13.	2.48
16.	5.73
20.	15.8
20.57 (max)	18.2

RMS % Error	Life Prediction Ratio Summary
12.01	0. .5 .8 1.25 2.

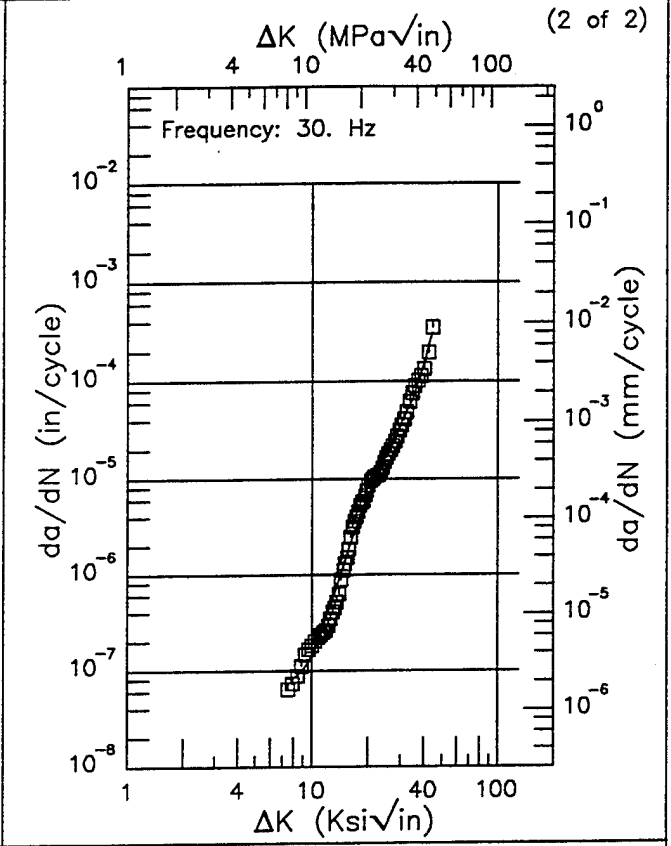
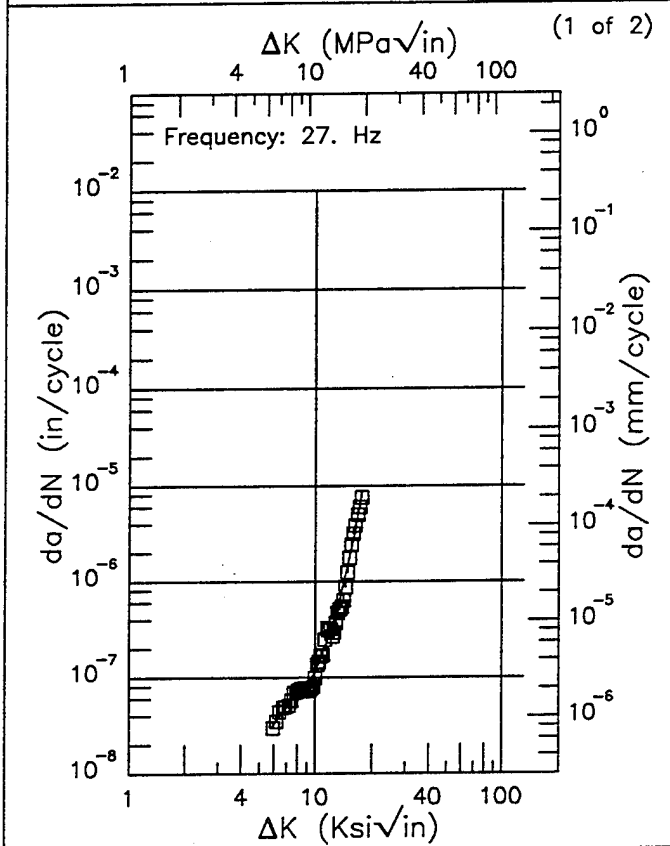
RMS % Error	Life Prediction Ratio Summary
9.55	0. .5 .8 1.25 2.

Figure 6.16.3.1.80

F | Ti-6Al-4V |

Condition/Ht: MA
 Form: 1.5 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.02
 Environment: LAB AIR; RT

Yield Strength: 133 ksi
 Ult. Strength: 142 ksi
 Specimen Thk: 0.188 - 0.191 in.
 Specimen Width: 3.957 - 3.964 in.
 Ref: MA002



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
5.91 (min)	0.0306
6.	0.0325
7.	0.0509
8.	0.0658
9.	0.0835
10.	0.113
13.	0.450
16.	3.03
17.49 (max)	8.73

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.47 (min)	0.0829
8.	0.0833
9.	0.102
10.	0.145
13.	0.563
16.	1.98
20.	6.77
25.	18.4
30.	35.8
35.	68.5
40.	148.
44.29 (max)	306.

RMS % Error 14.69	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
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RMS % Error 18.82	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

Figure 6.16.3.1.81

Ti-6Al-4V EF

Condition/Ht: MA
 Form: 0.75 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.04

Yield Strength: 135.3 ksi
 Ult. Strength: 137.6 ksi
 Specimen Thk: 0.29 in.
 Specimen Width: 6.011 in.
 Ref: 88468

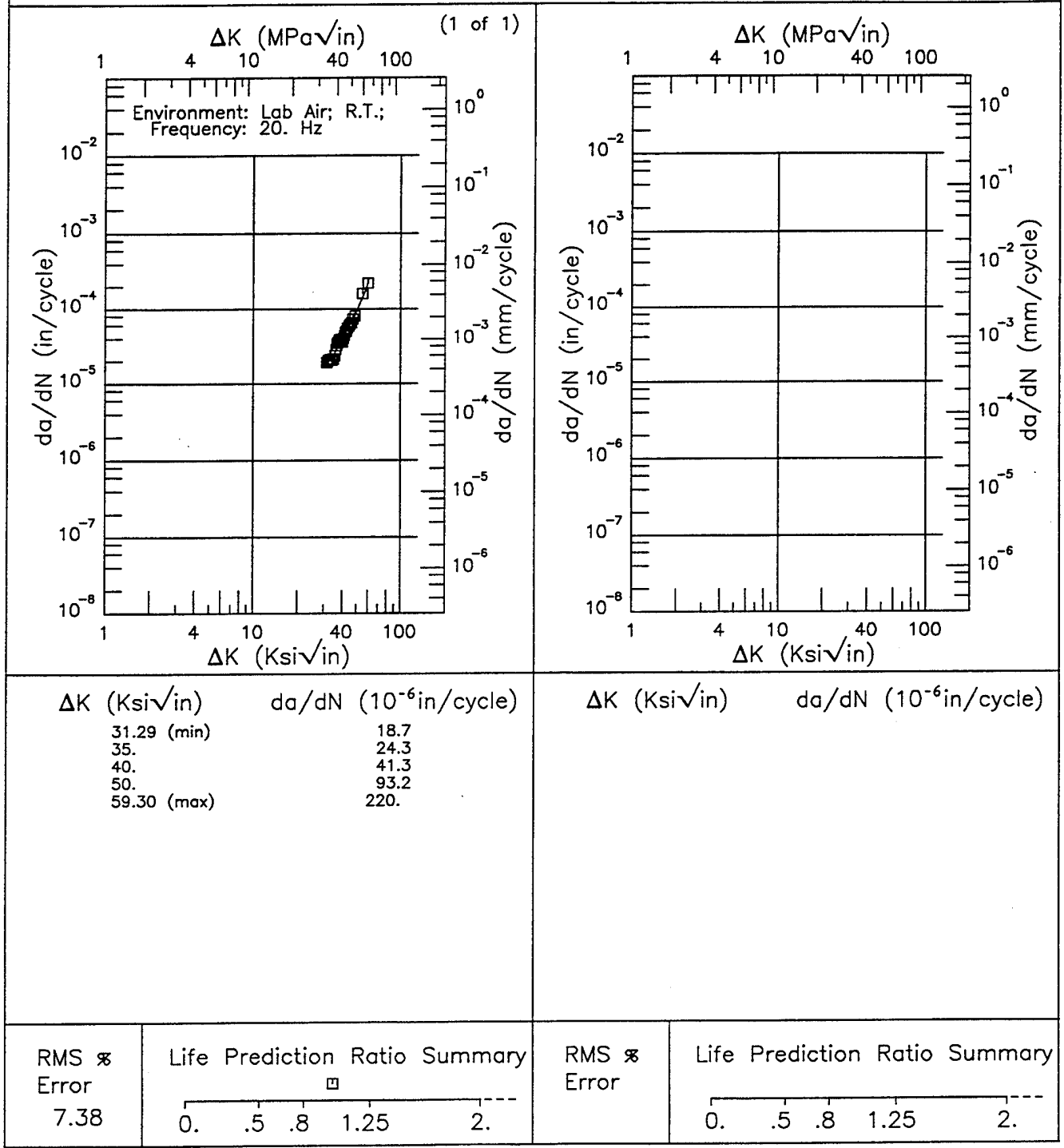


Figure 6.16.3.1.82

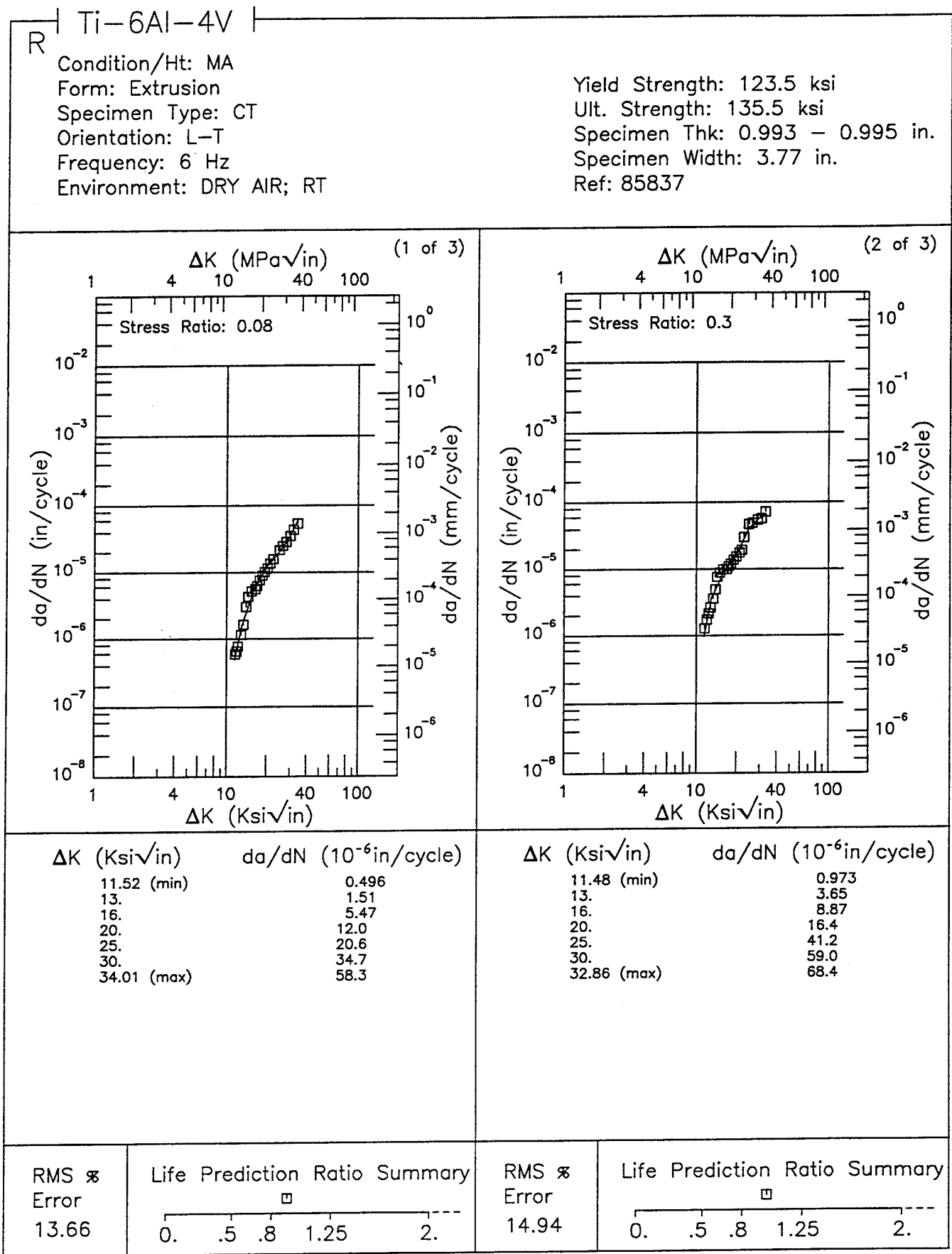


Figure 6.16.3.1.83

Ti-6Al-4V R

Condition/Ht: MA
 Form: Extrusion
 Specimen Type: CT
 Orientation: L-T
 Frequency: 6 Hz
 Environment: DRY AIR; RT

Yield Strength: 123.5 ksi
 Ult. Strength: 135.5 ksi
 Specimen Thk: 0.993 - 0.995 in.
 Specimen Width: 3.77 in.
 Ref: 85837

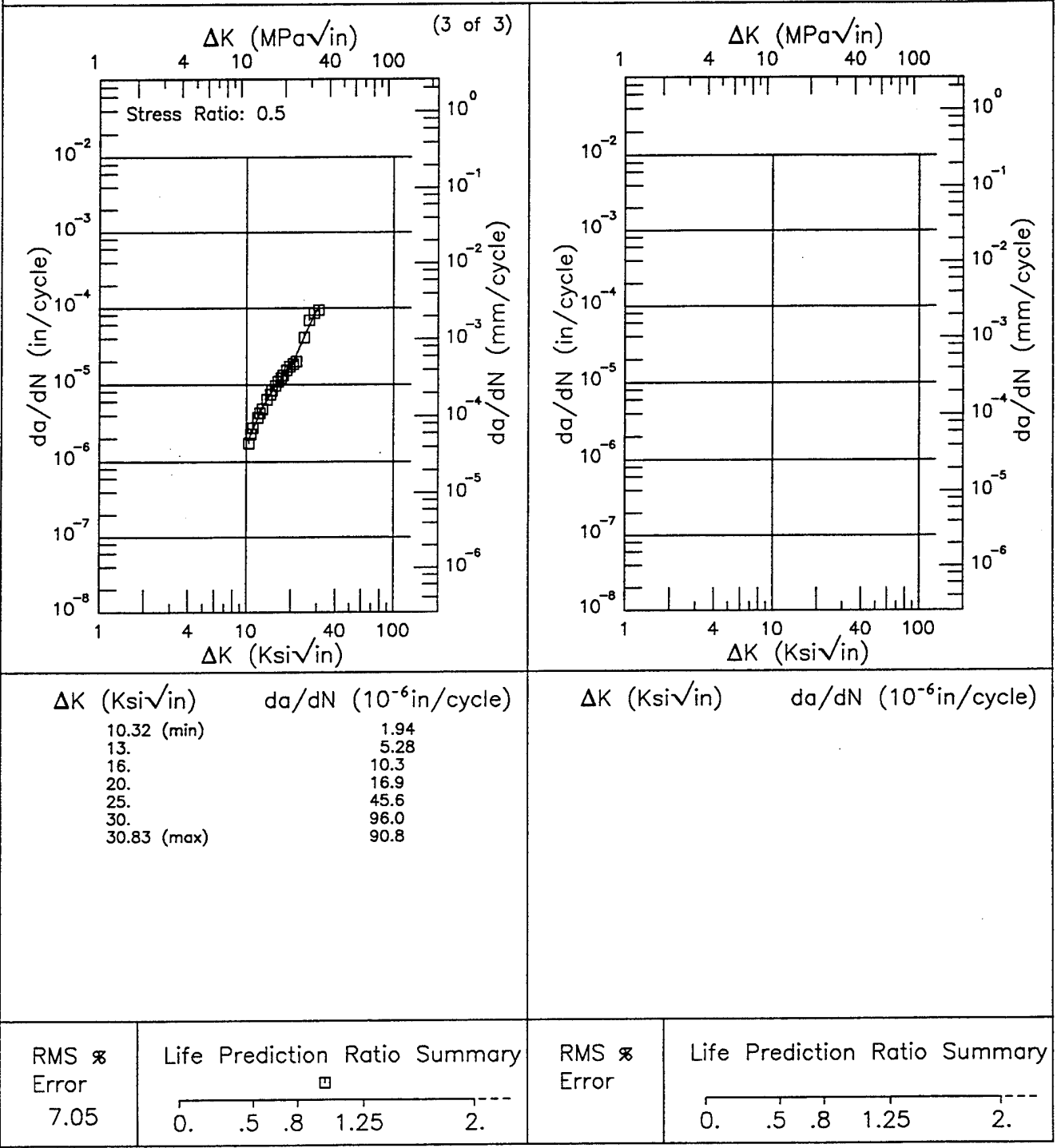


Figure 6.16.3.1.83 (Concluded)

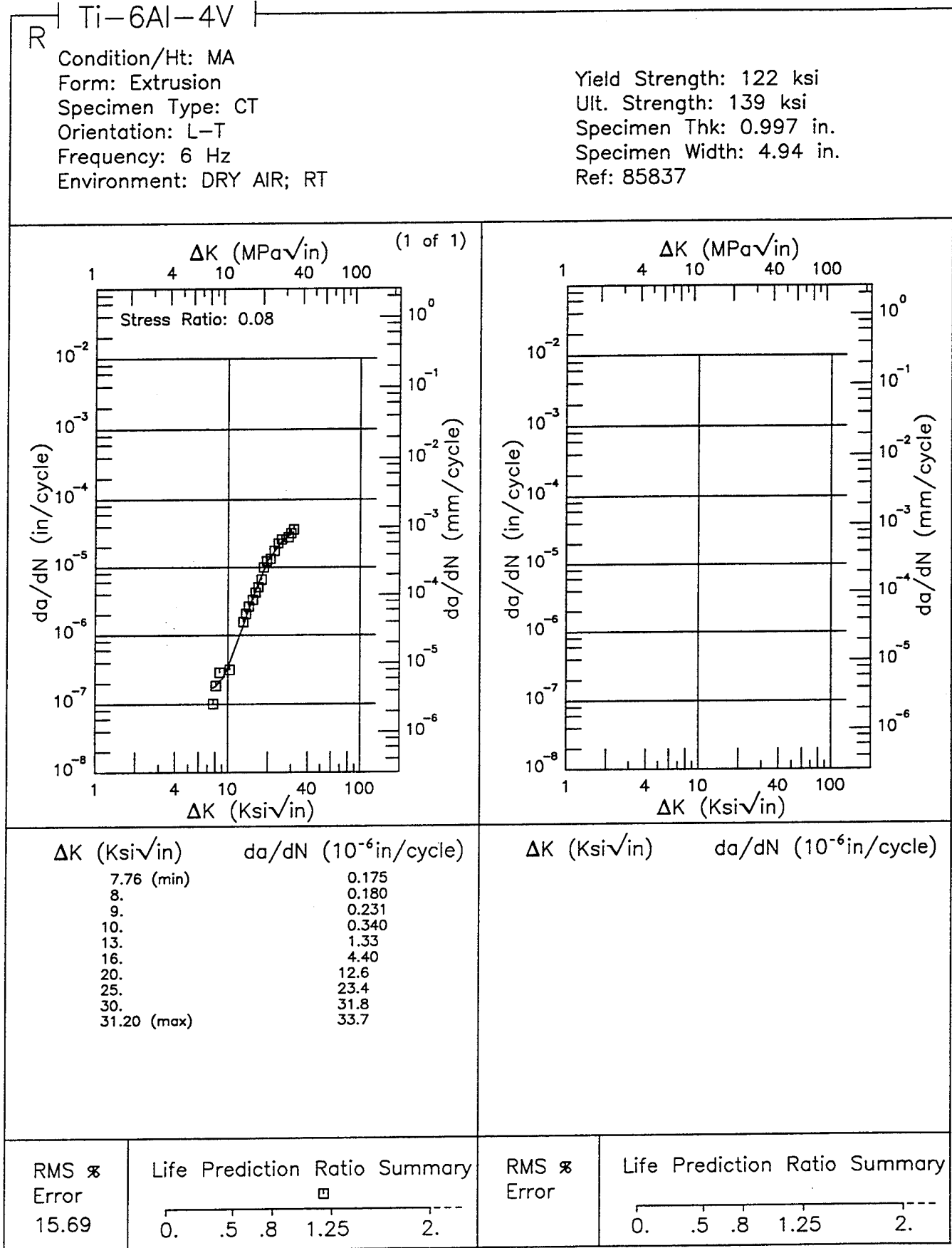
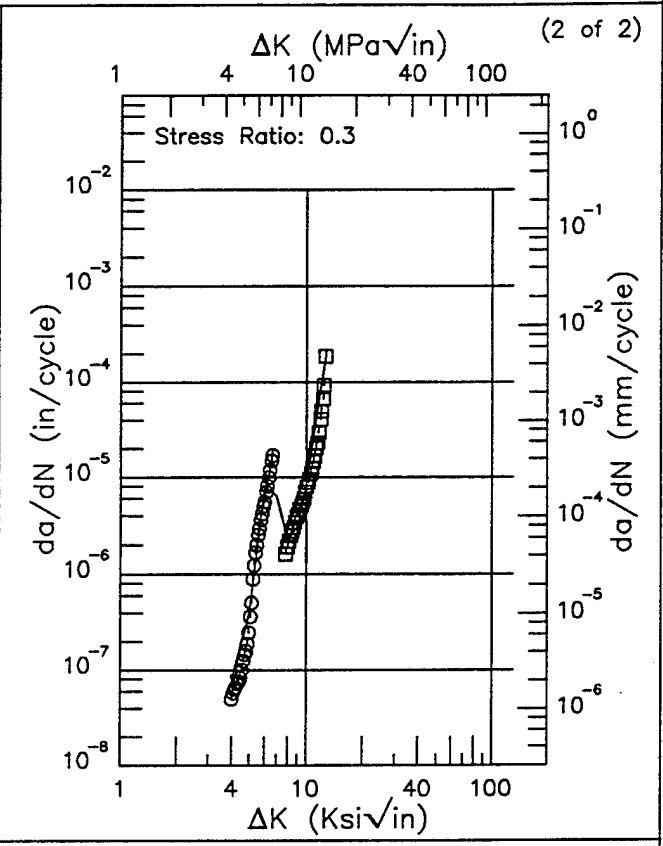
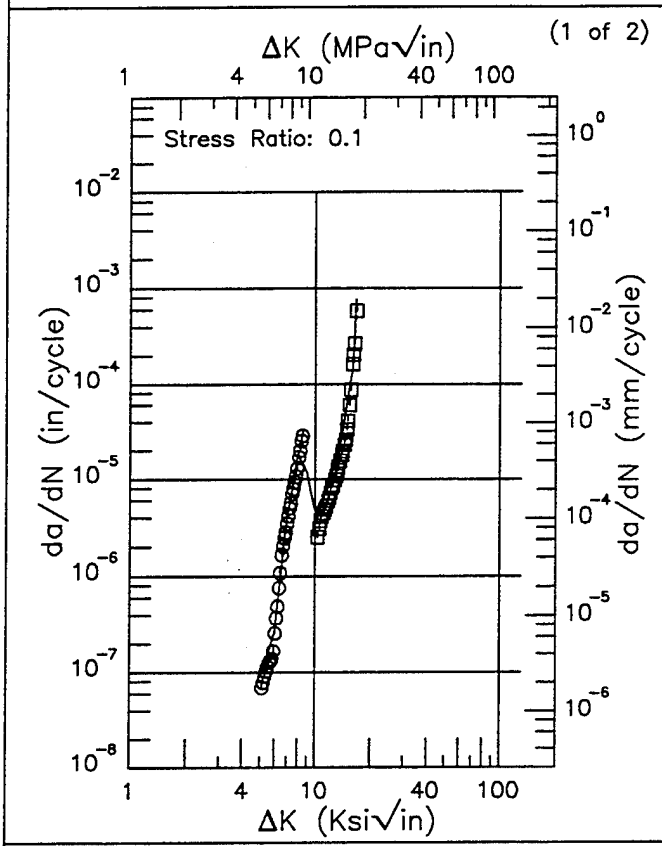


Figure 6.16.3.1.84

Ti-6Al-4V R

Condition/Ht: MA
 Form: Extrusion
 Specimen Type: CT
 Orientation: L-T
 Frequency: 10 Hz
 Environment: LAB AIR; RT

Yield Strength:
 Ult. Strength:
 Specimen Thk: 0.251 in.
 Specimen Width: 2 in.
 Ref: NC005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
5.10 (min)	0.0946
6.	0.259
7.	4.01
8.	18.2
9.	11.9
10.	4.64
13.	11.9
16.	152.
16.69 (max)	778.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
3.97 (min)	0.0878
4.	0.0802
5.	0.394
6.	8.21
7.	6.36
8.	2.22
9.	3.07
10.	8.82
12.66 (max)	210.

RMS % Error 29.84	Life Prediction Ratio Summary
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RMS % Error 40.20	Life Prediction Ratio Summary
----------------------	-----------------------------------

Figure 6.16.3.1.85

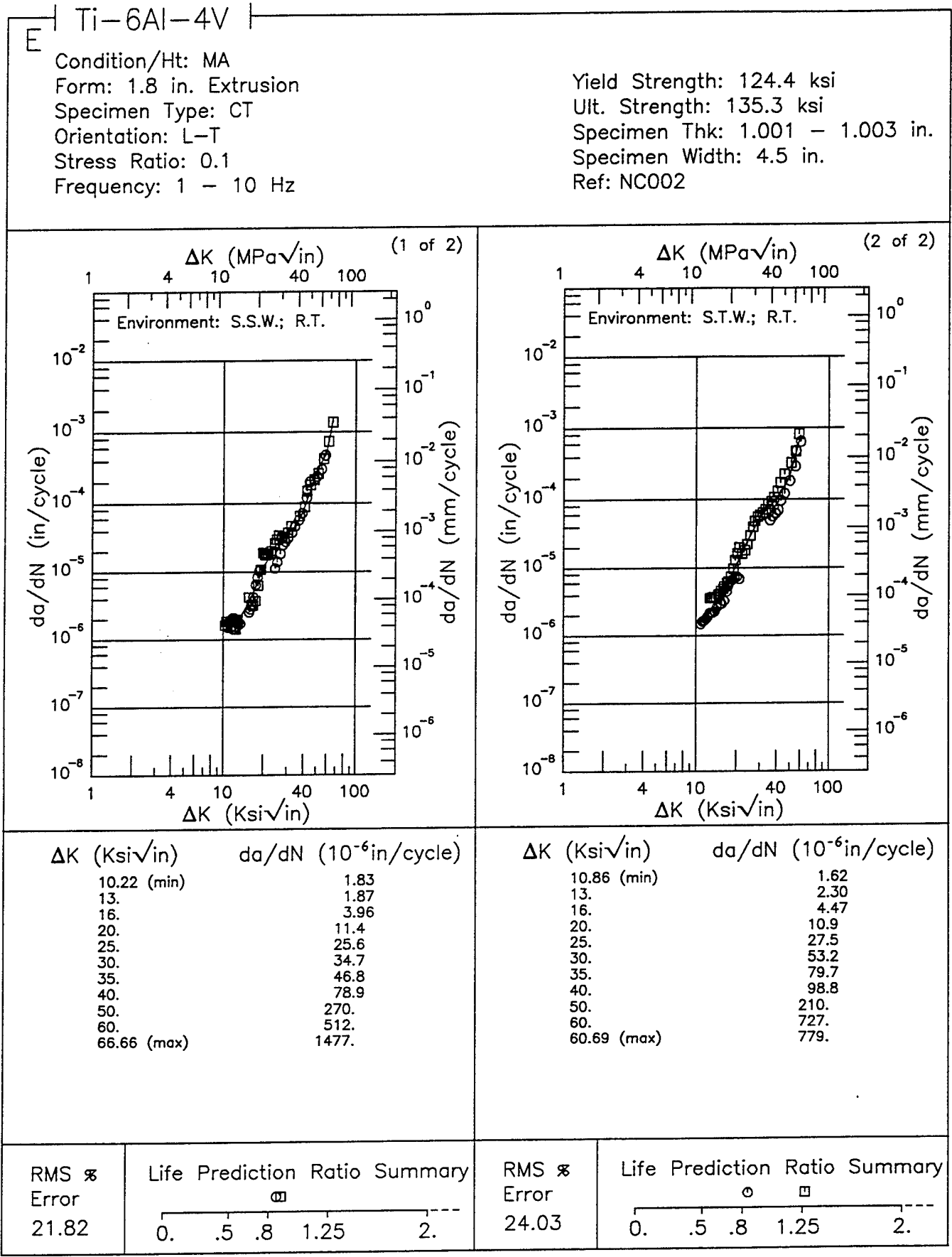


Figure 6.16.3.1.86

Ti-6Al-4V F

Condition/Ht: MA
 Form: 1.8 in. Extrusion
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.1
 Environment: LAB AIR; RT

Yield Strength: 124.4 ksi
 Ult. Strength: 135.3 ksi
 Specimen Thk: 1 - 1.002 in.
 Specimen Width: 4.5 in.
 Ref: NC002

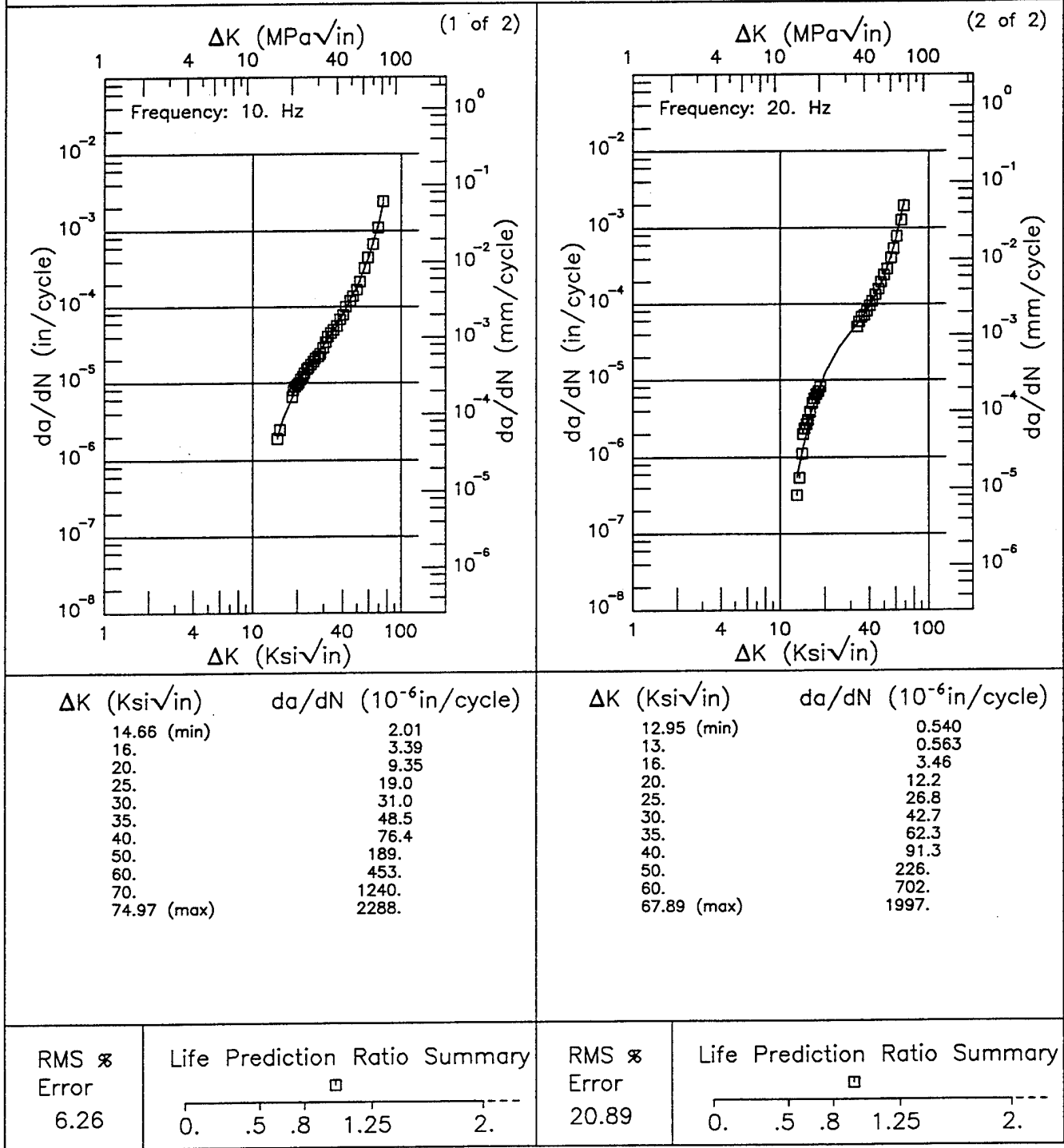
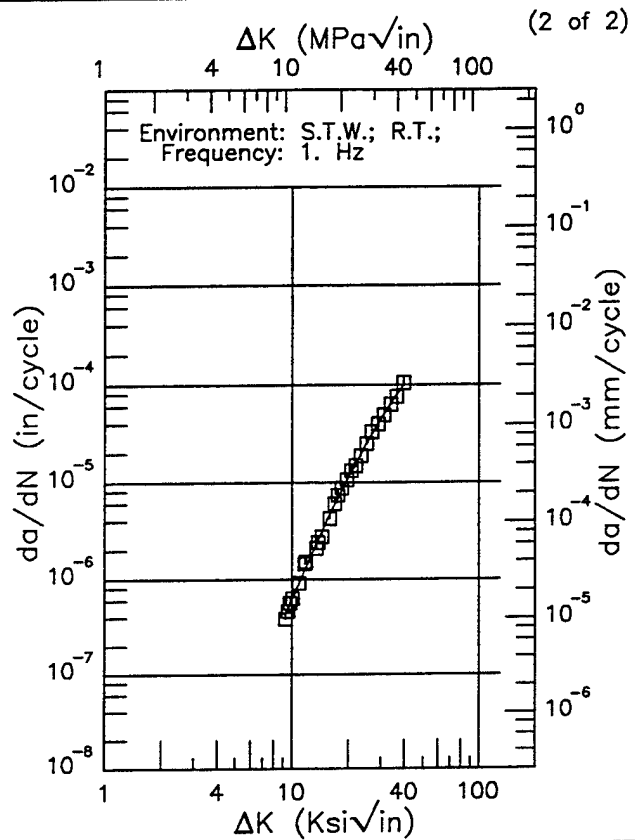
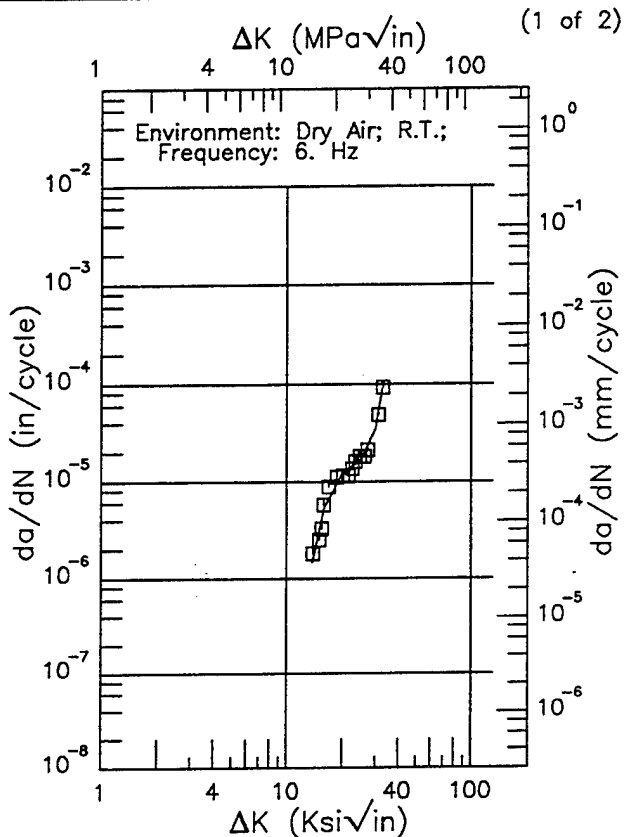


Figure 6.16.3.1.87

EF Ti-6Al-4V

Condition/Ht: MA
 Form: Extrusion
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.08

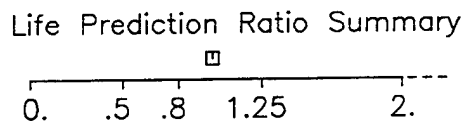
Yield Strength: 125 ksi
 Ult. Strength: 142 ksi
 Specimen Thk: 0.994 in.
 Specimen Width: 3.77 in.
 Ref: 85837



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
13.81 (min)	1.47
16.	5.48
20.	12.1
25.	16.5
30.	35.4
32.77 (max)	89.6

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
9.22 (min)	0.428
10.	0.620
13.	1.97
16.	4.67
20.	11.1
25.	24.8
30.	45.1
35.	71.6
39.27 (max)	98.5

RMS $\%$
 Error
 15.22



RMS $\%$
 Error
 6.87

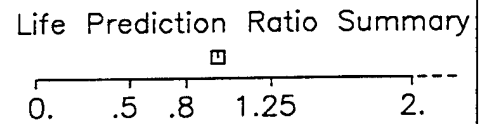


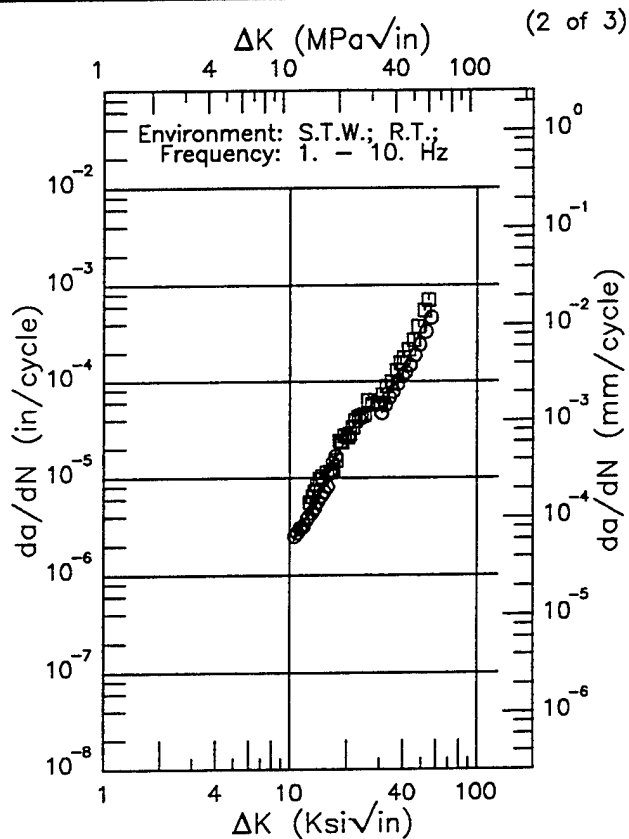
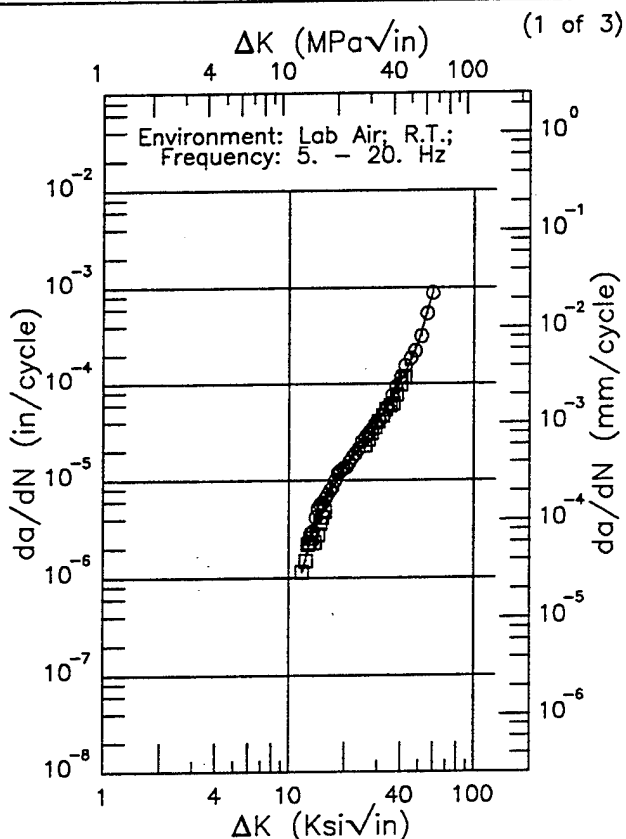
Figure 6.16.3.1.88

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EF | Ti-6Al-4V |

Condition/Ht: MA
 Form: 1.8 in. Extrusion
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1

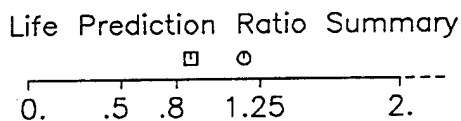
Yield Strength: 126.6 ksi
 Ult. Strength: 138.6 ksi
 Specimen Thk: 1 - 1.003 in.
 Specimen Width: 4.5 in.
 Ref: NC002



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
11.73 (min)	1.14
13.	2.33
16.	6.65
20.	13.7
25.	23.8
30.	38.5
35.	63.0
40.	103.
50.	277.
59.40 (max)	865.

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
10.58 (min)	2.30
13.	4.95
16.	11.3
20.	25.9
25.	48.8
30.	66.6
35.	86.0
40.	124.
50.	354.
56.74 (max)	587.

RMS %
 Error
 11.93



RMS %
 Error
 18.13

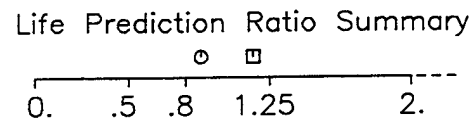
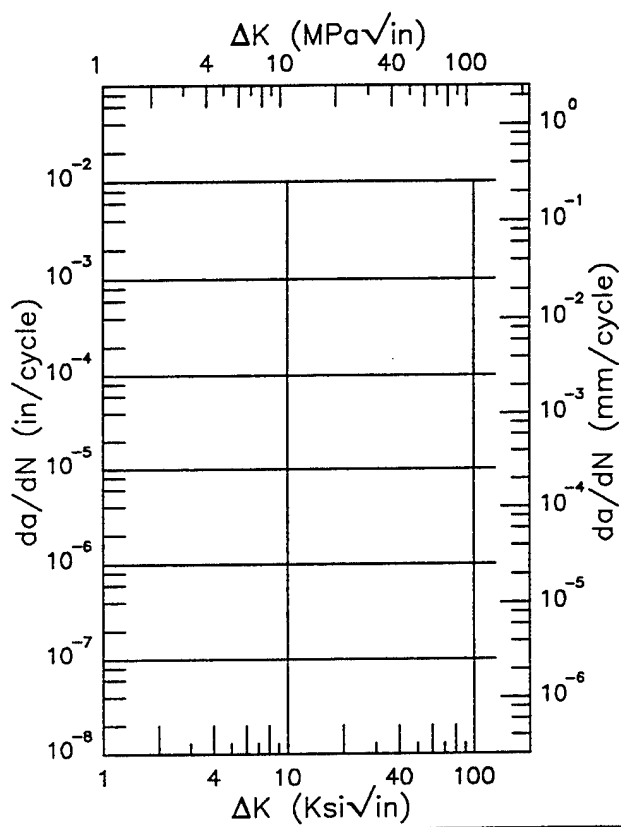
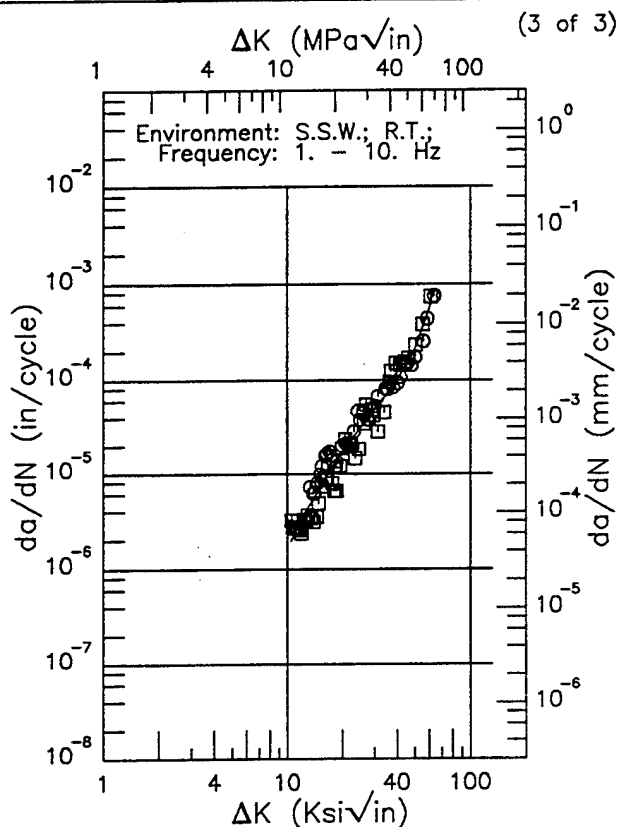


Figure 6.16.3.1.89

Ti-6Al-4V EF

Condition/Ht: MA
 Form: 1.8 in. Extrusion
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1

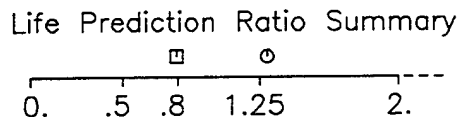
Yield Strength: 126.6 ksi
 Ult. Strength: 138.6 ksi
 Specimen Thk: 1 - 1.003 in.
 Specimen Width: 4.5 in.
 Ref: NC002



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.52 (min)	1.95
13.	4.39
16.	8.83
20.	17.2
25.	31.2
30.	49.1
35.	71.8
40.	102.
50.	209.
60.	560.
63.55 (max)	899.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
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RMS $\%$
 Error
 29.88



RMS $\%$
 Error

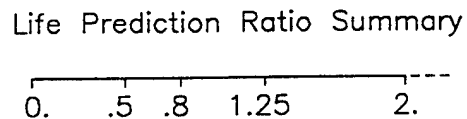


Figure 6.16.3.1.89 (Concluded)

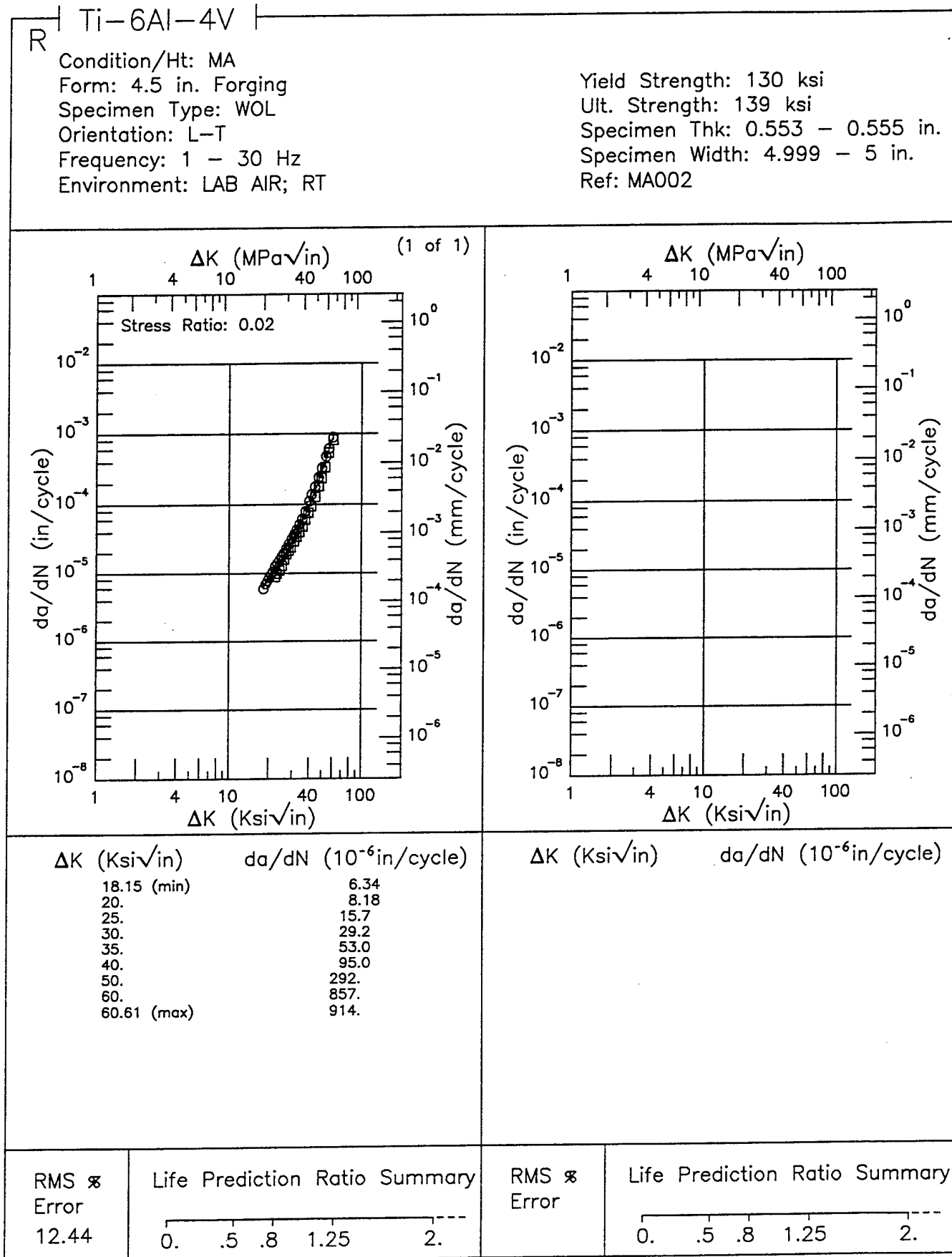


Figure 6.16.3.1.90

Ti-6Al-4V E

Condition/Ht: MA
 Form: 2 in. Forging
 Specimen Type: DCB
 Orientation: L-T
 Stress Ratio: 0.02
 Frequency: 0.1 Hz

Yield Strength: 145 ksi
 Ult. Strength: 156 ksi
 Specimen Thk: 0.75 in.
 Specimen Width: 5.5 in.
 Ref: 84360

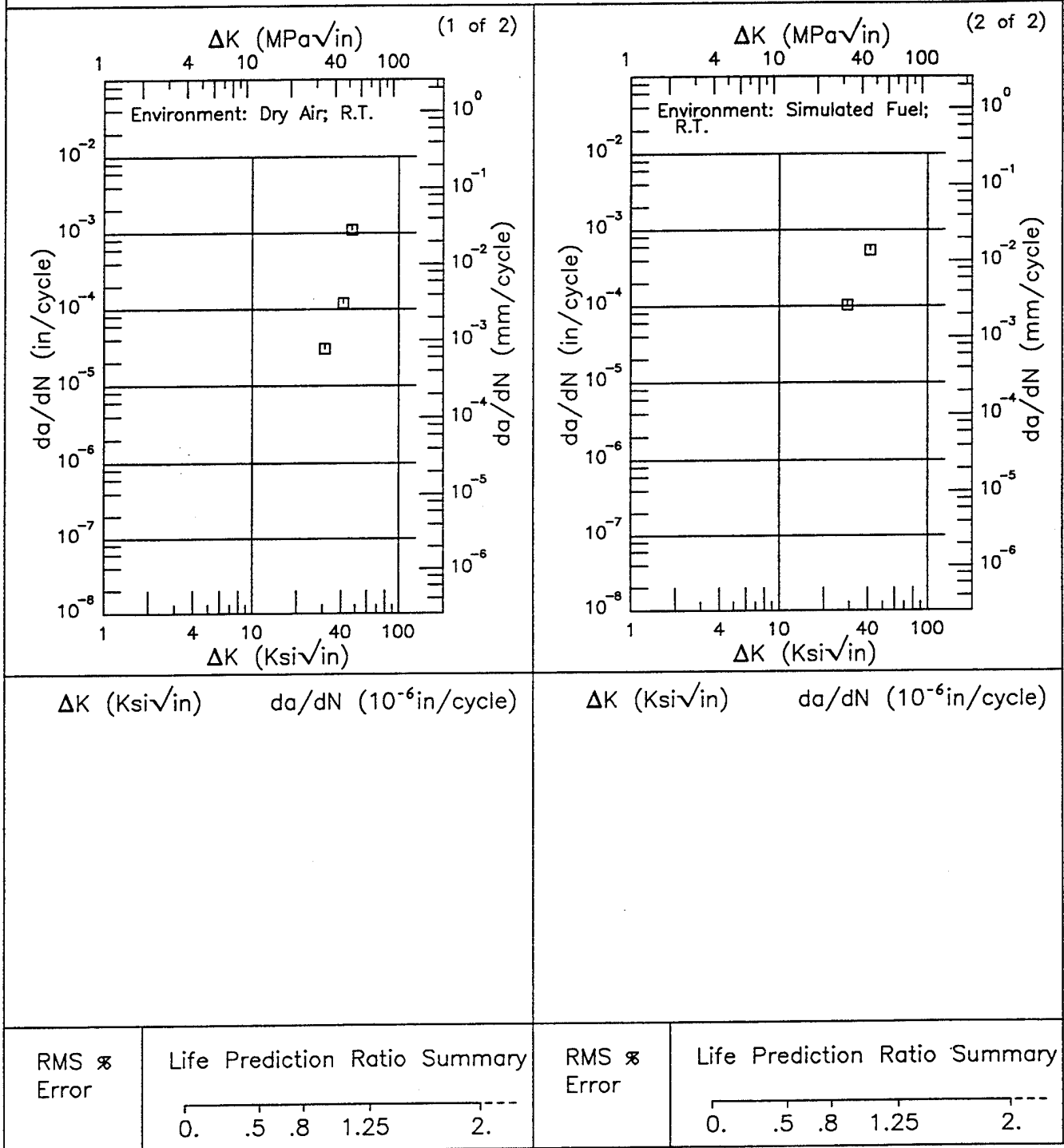


Figure 6.16.3.1.91

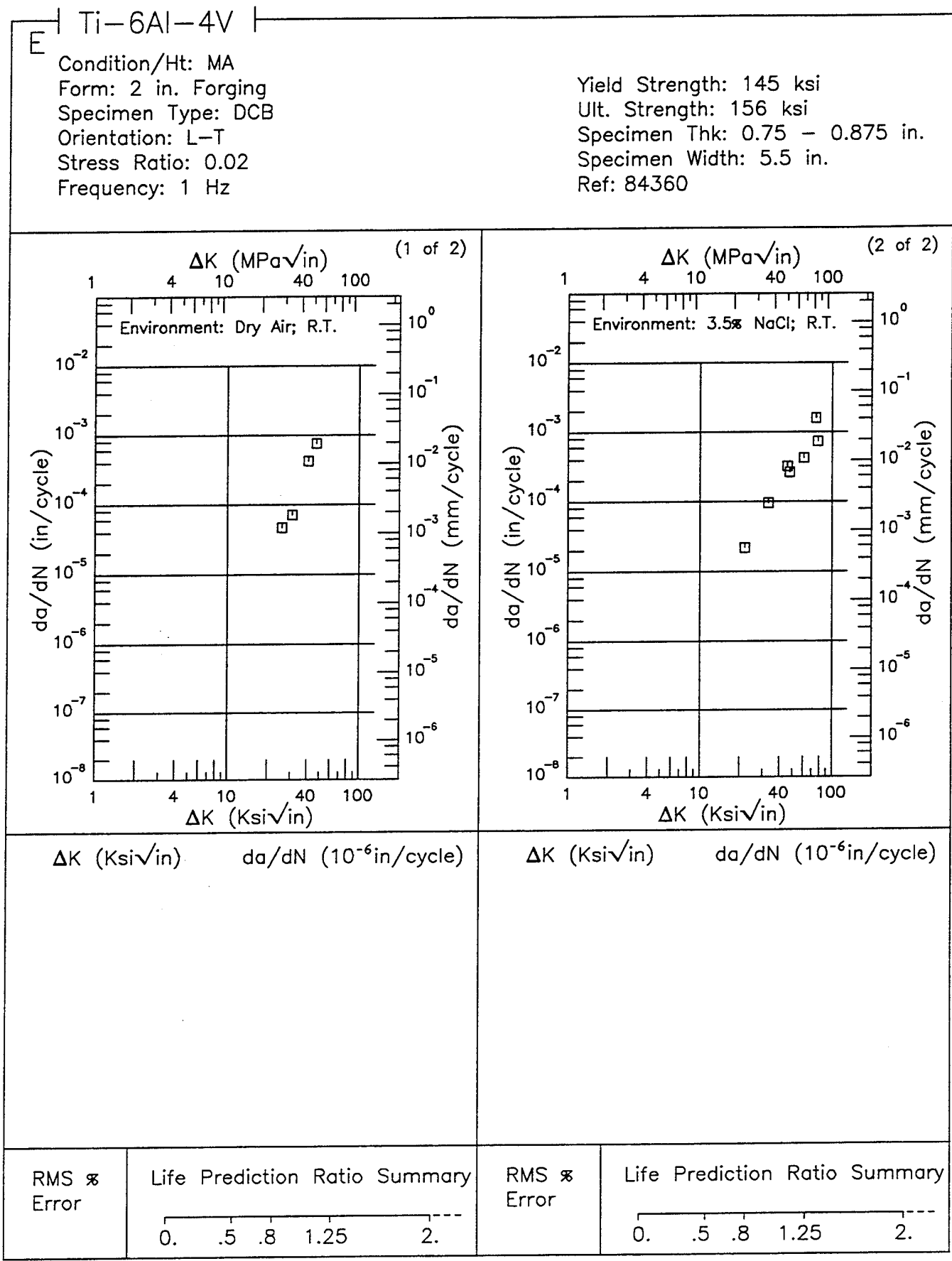


Figure 6.16.3.1.92

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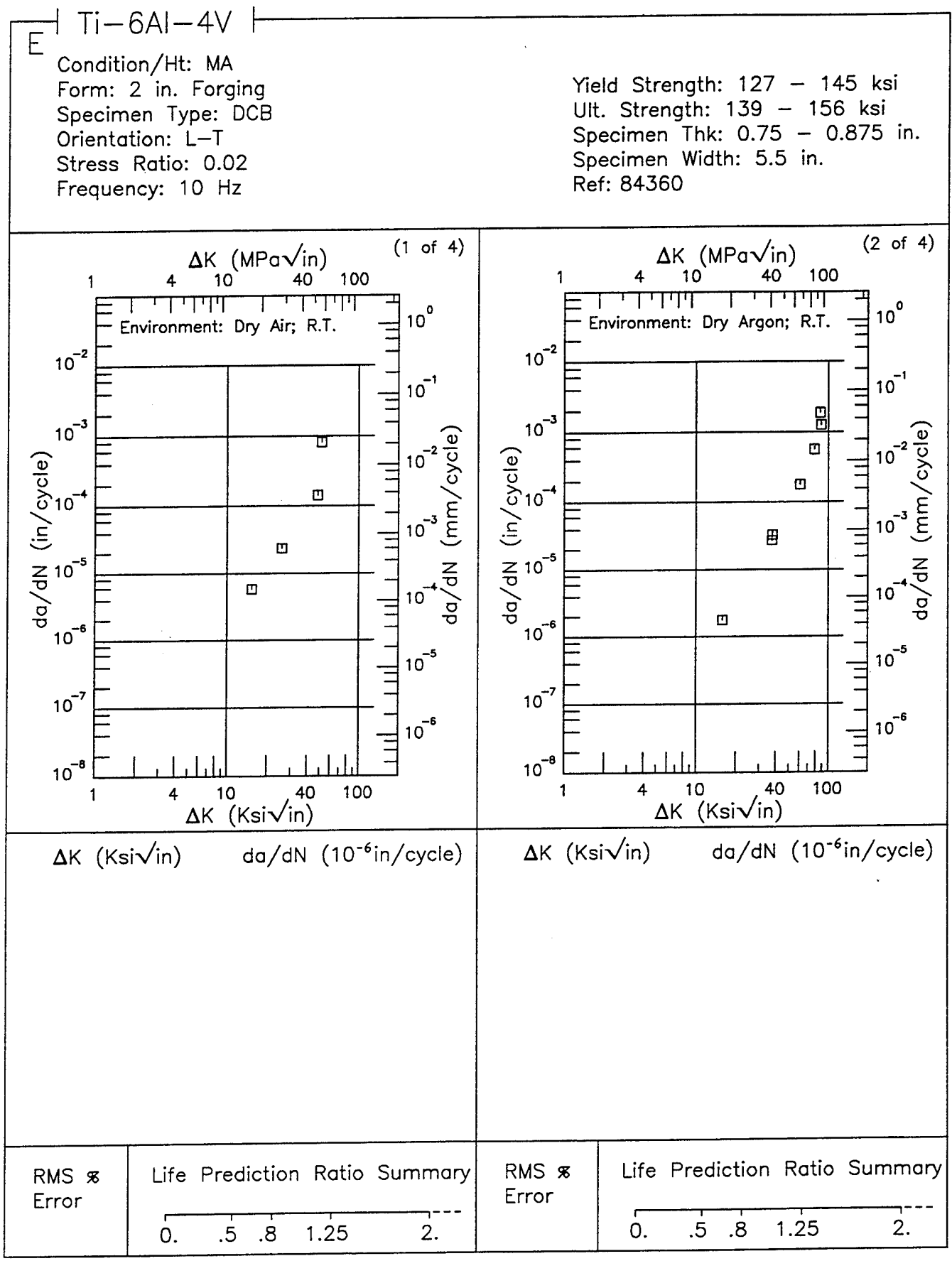


Figure 6.16.3.1.93

Ti-6Al-4V E

Condition/Ht: MA
 Form: 2 in. Forging
 Specimen Type: DCB
 Orientation: L-T
 Stress Ratio: 0.02
 Frequency: 10 Hz

Yield Strength: 127 - 145 ksi
 Ult. Strength: 139 - 156 ksi
 Specimen Thk: 0.75 - 0.875 in.
 Specimen Width: 5.5 in.
 Ref: 84360

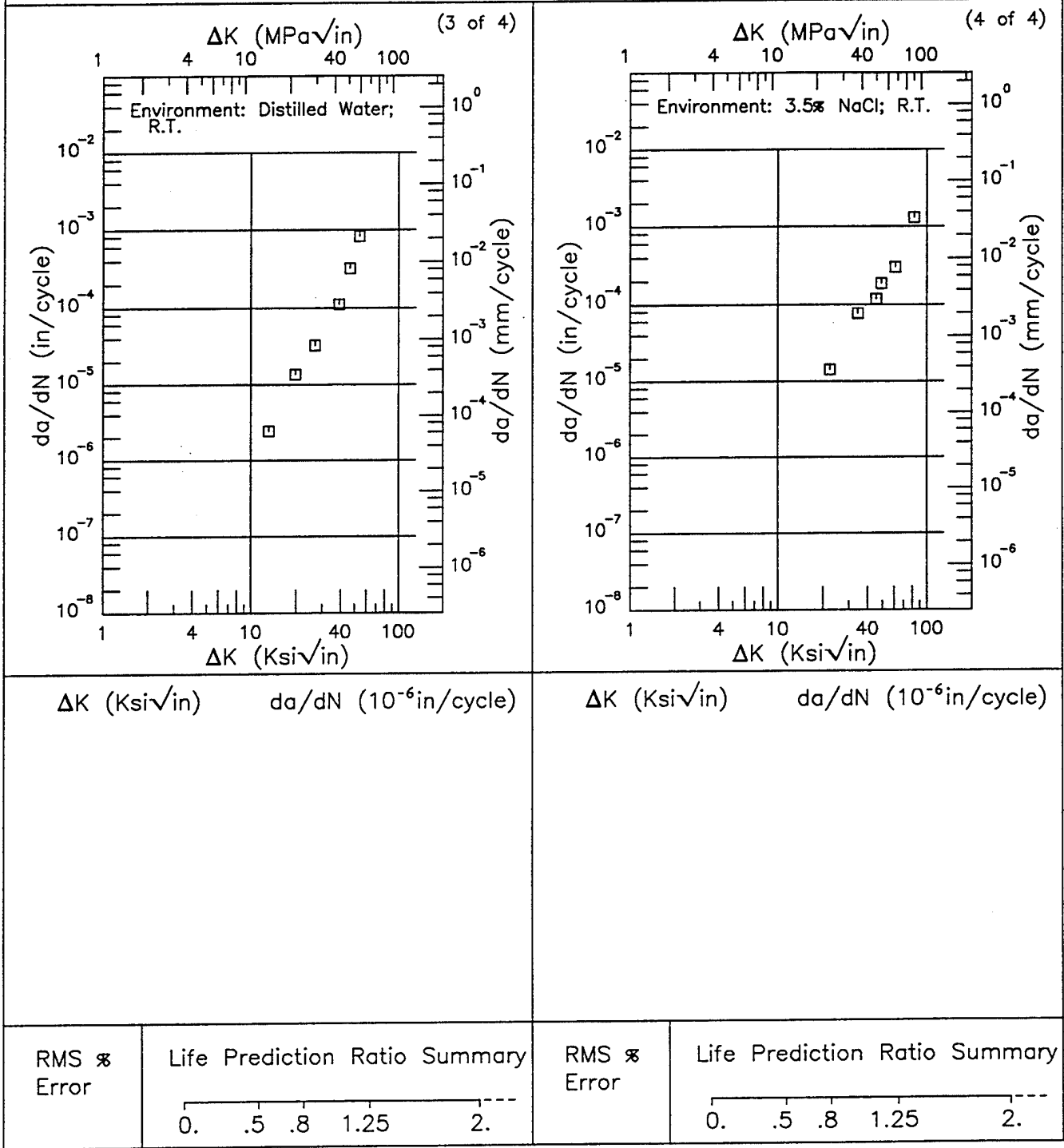


Figure 6.16.3.1.93 (Concluded)

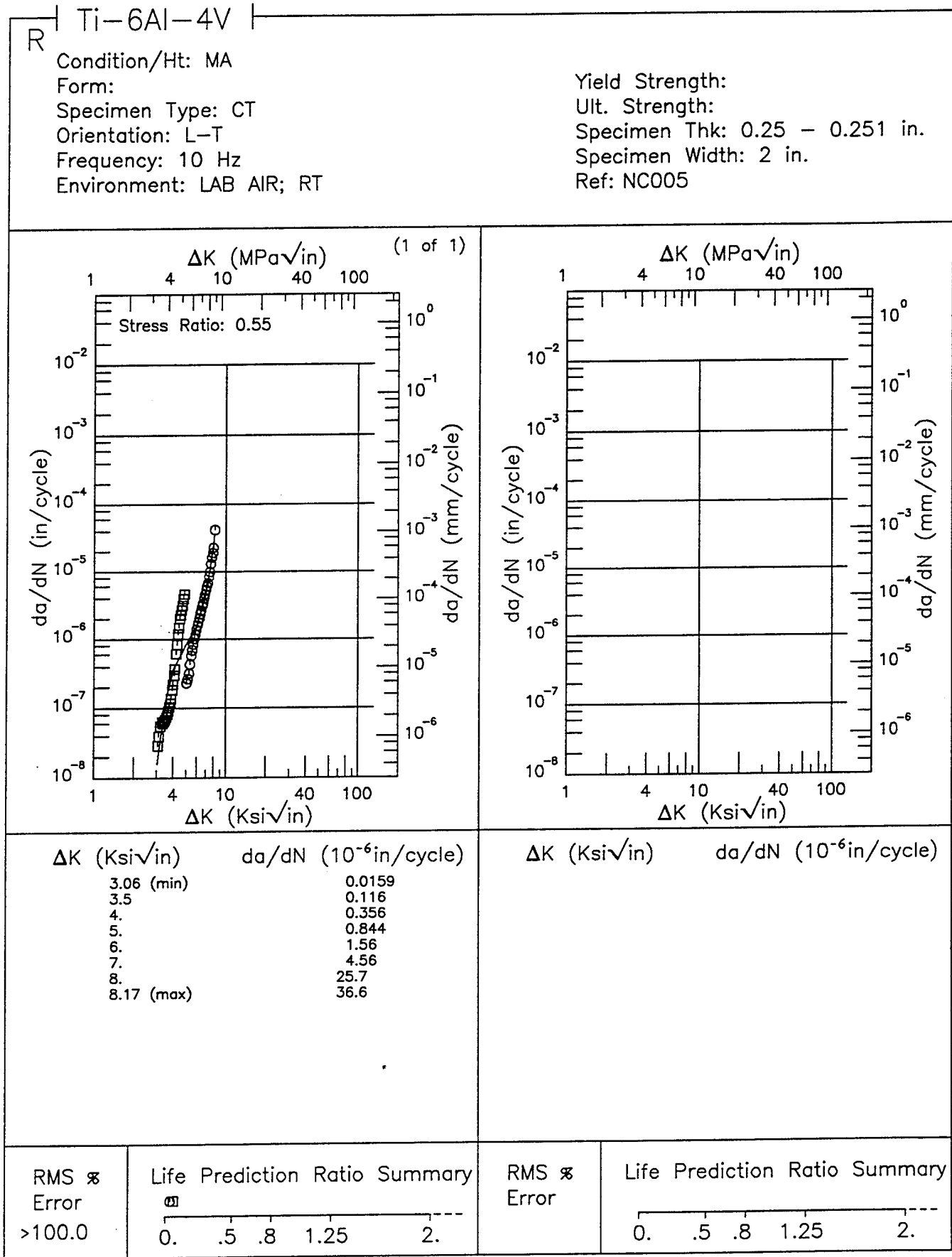


Figure 6.16.3.1.94

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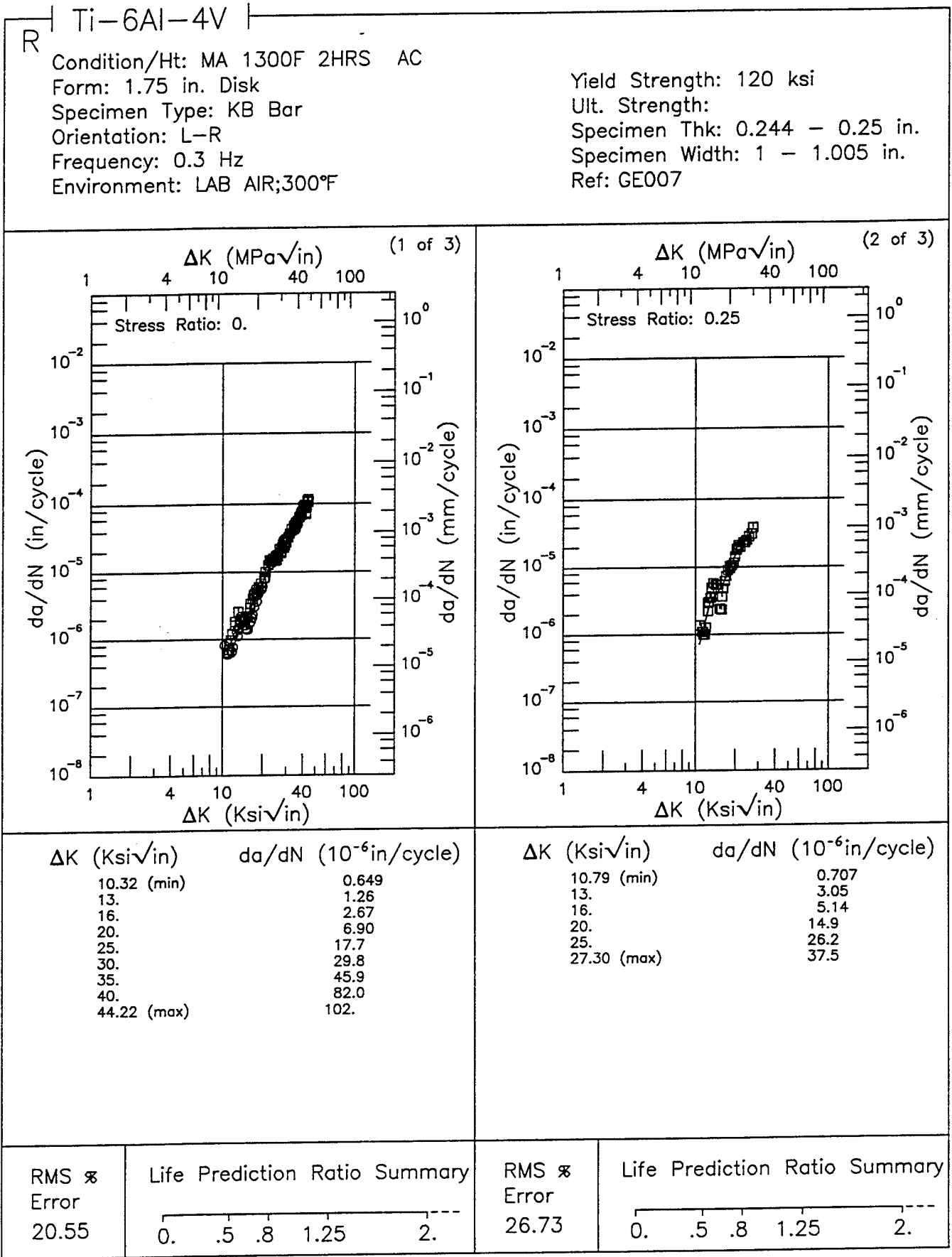
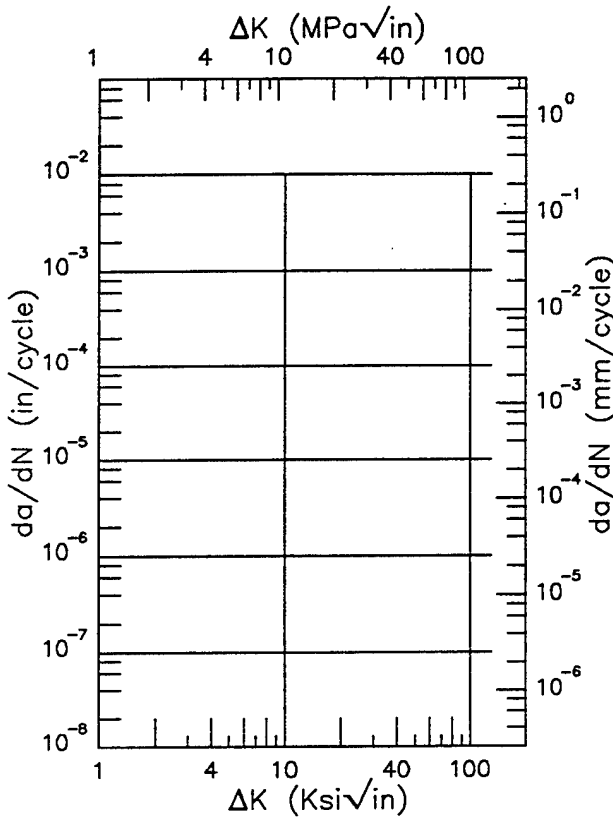
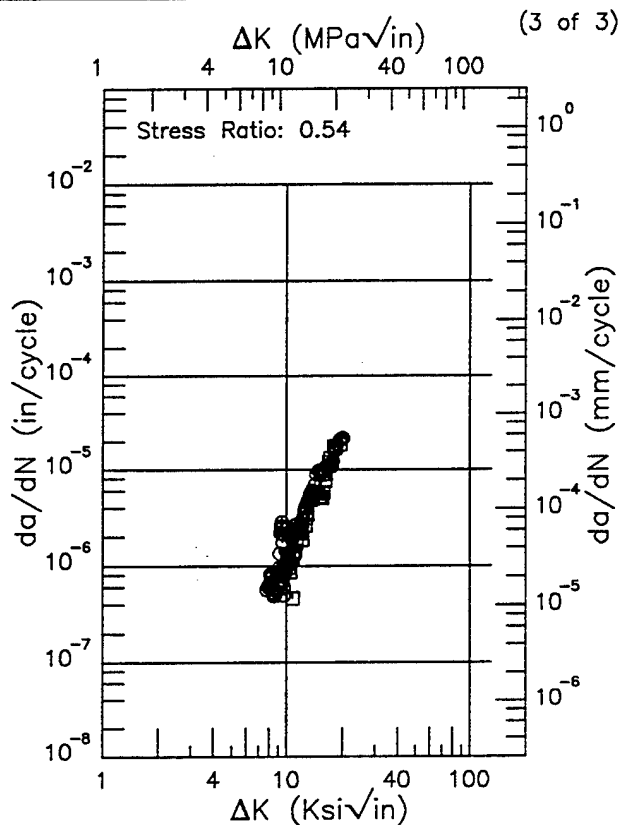


Figure 6.16.3.1.95

Ti-6Al-4V R

Condition/Ht: MA 1300F 2HRS AC
 Form: 1.75 in. Disk
 Specimen Type: KB Bar
 Orientation: L-R
 Frequency: 0.3 Hz
 Environment: LAB AIR;300°F

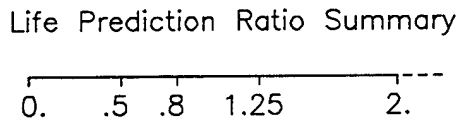
Yield Strength: 120 ksi
 Ult. Strength:
 Specimen Thk: 0.244 - 0.25 in.
 Specimen Width: 1 - 1.005 in.
 Ref: GE007



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
7.74 (min)	0.526
8.	0.612
9.	0.868
10.	1.12
13.	4.23
16.	8.63
19.93 (max)	20.0

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
--------------------------------------	-------------------------------

RMS \times
 Error
 43.65



RMS \times
 Error

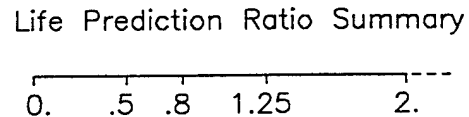


Figure 6.16.3.1.95 (Concluded)

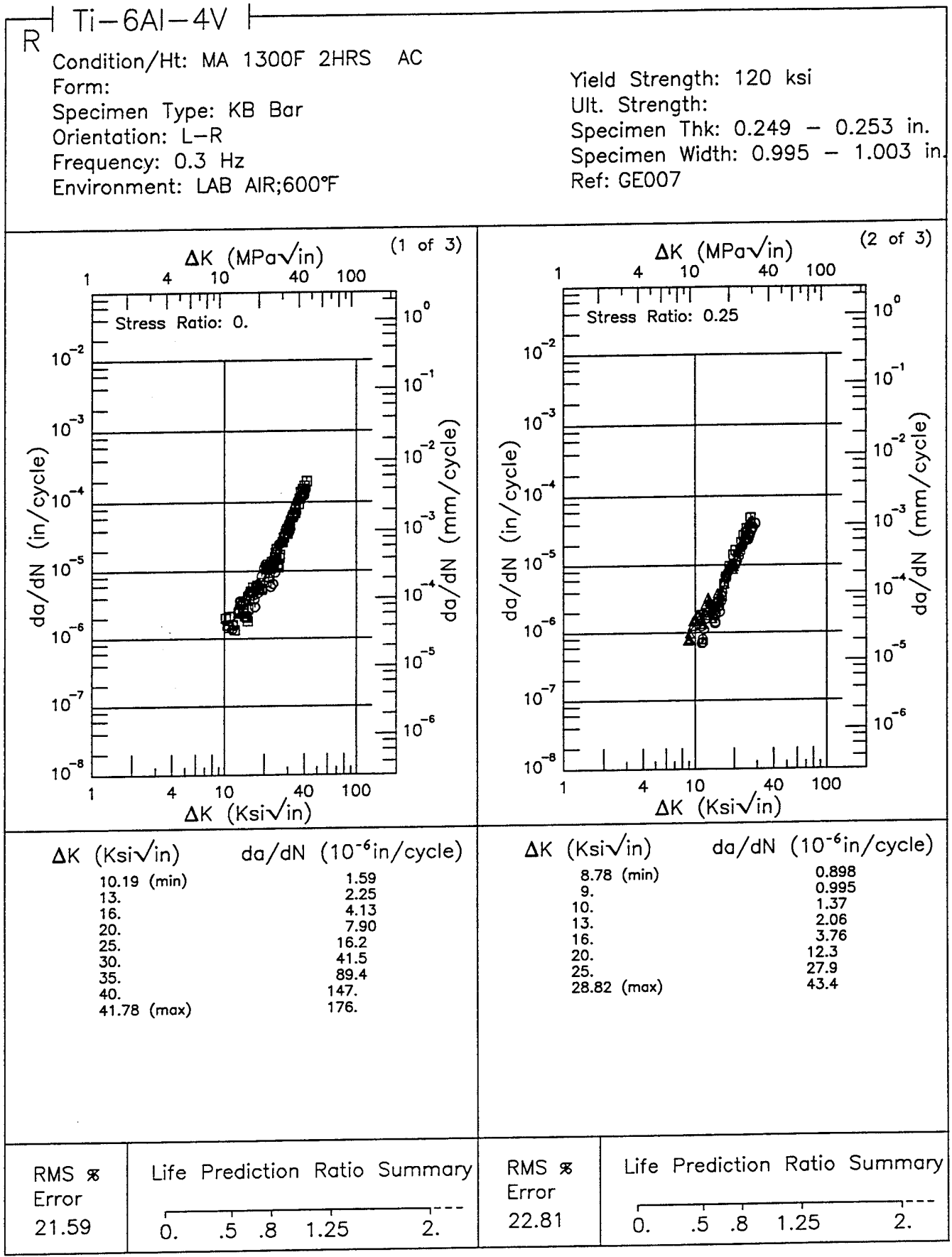
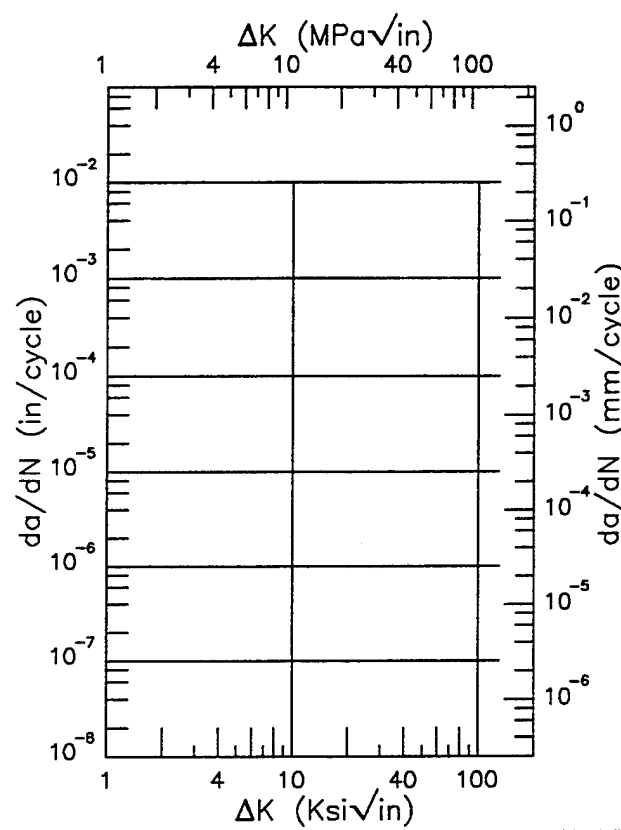
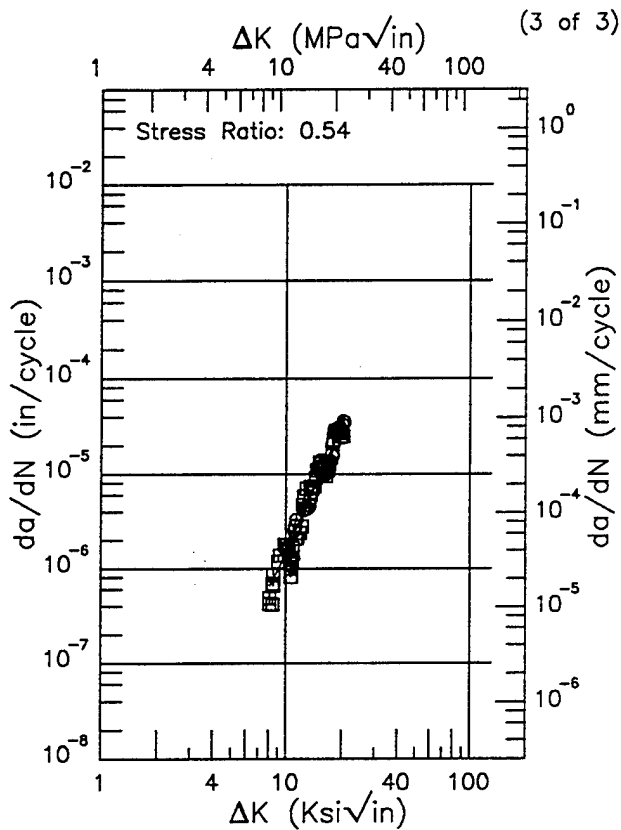


Figure 6.16.3.1.96

Ti-6Al-4V R

Condition/Ht: MA 1300F 2HRS AC
 Form:
 Specimen Type: KB Bar
 Orientation: L-R
 Frequency: 0.3 Hz
 Environment: LAB AIR;600°F

Yield Strength: 120 ksi
 Ult. Strength:
 Specimen Thk: 0.249 - 0.253 in.
 Specimen Width: 0.995 - 1.003 in.
 Ref: GE007



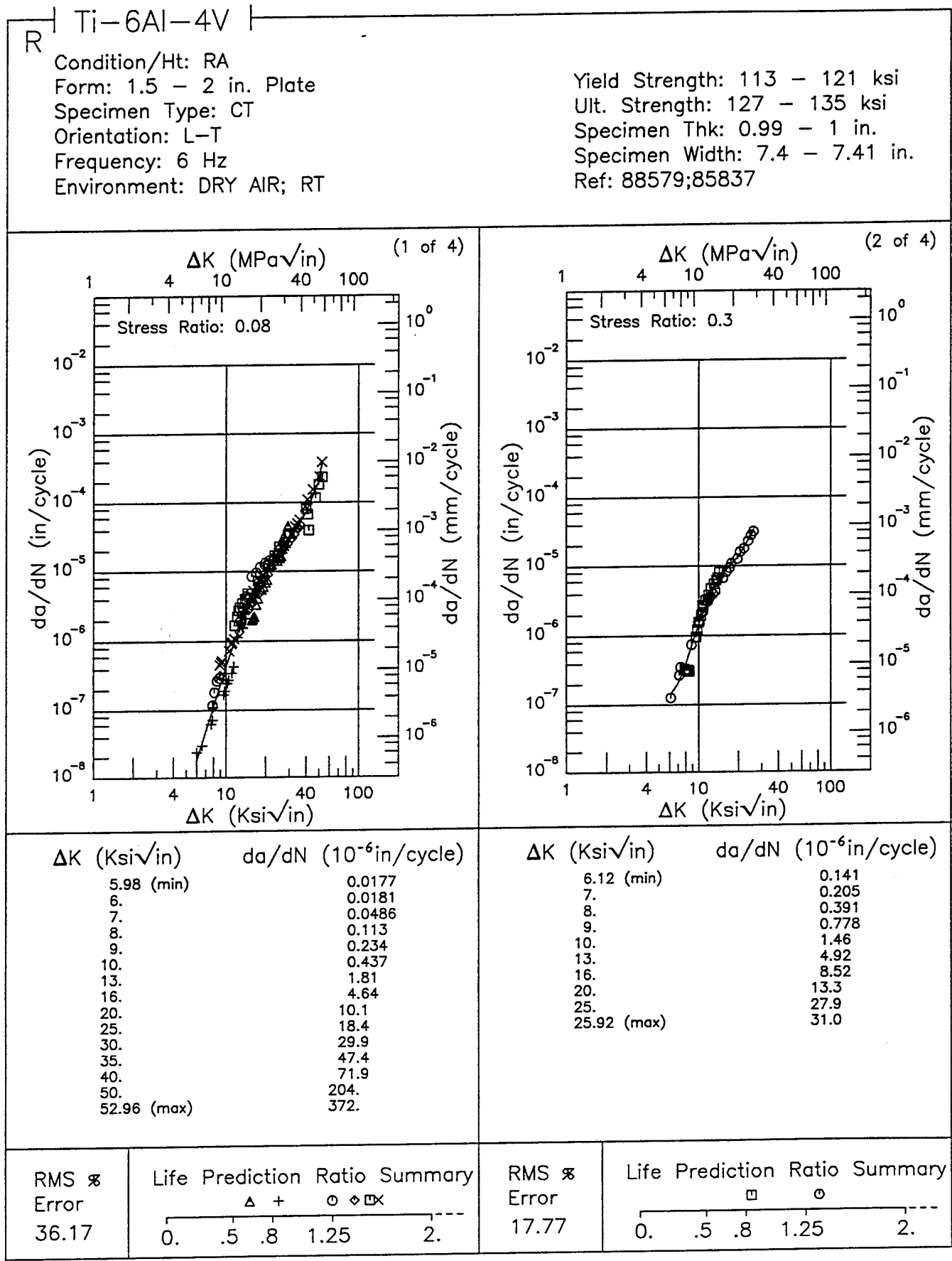
ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.13 (min)	0.658
9.	0.952
10.	1.46
13.	4.76
16.	12.4
20.	31.1
20.36 (max)	33.2

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS % Error	Life Prediction Ratio Summary
25.57	

RMS % Error	Life Prediction Ratio Summary

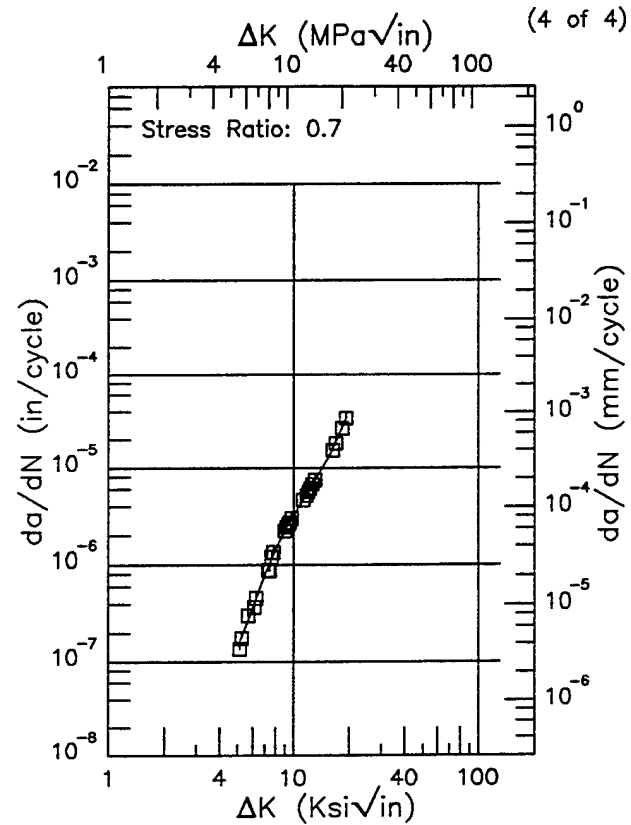
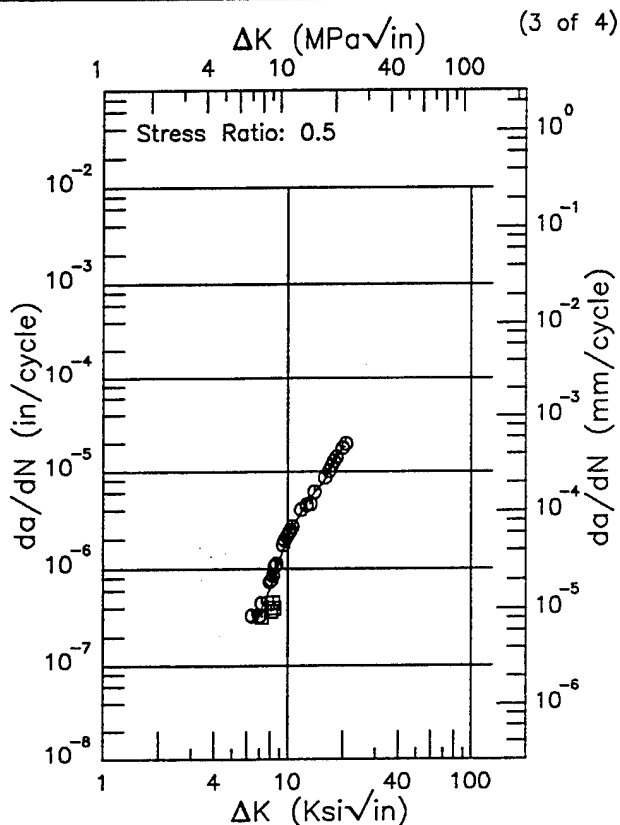
Figure 6.16.3.1.96 (Concluded)



Ti-6Al-4V R

Condition/Ht: RA
 Form: 1.5 - 2 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Frequency: 6 Hz
 Environment: DRY AIR; RT

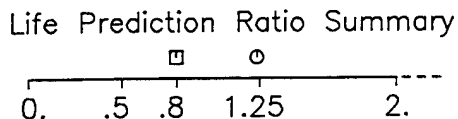
Yield Strength: 113 - 121 ksi
 Ult. Strength: 127 - 135 ksi
 Specimen Thk: 0.99 - 1 in.
 Specimen Width: 7.4 - 7.41 in.
 Ref: 88579;85837



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
6.39 (min)	0.292
7.	0.353
8.	0.628
9.	1.21
10.	2.16
13.	5.09
16.	8.33
20.	18.1
20.65 (max)	18.2

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
5.11 (min)	0.159
6.	0.361
7.	0.775
8.	1.42
9.	2.25
10.	3.21
13.	7.23
16.	14.6
19.21 (max)	33.6

RMS %
 Error
 18.14



RMS %
 Error
 5.80

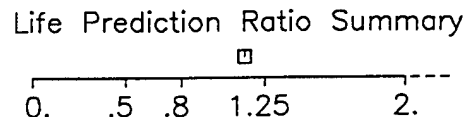


Figure 6.16.3.1.97 (Concluded)

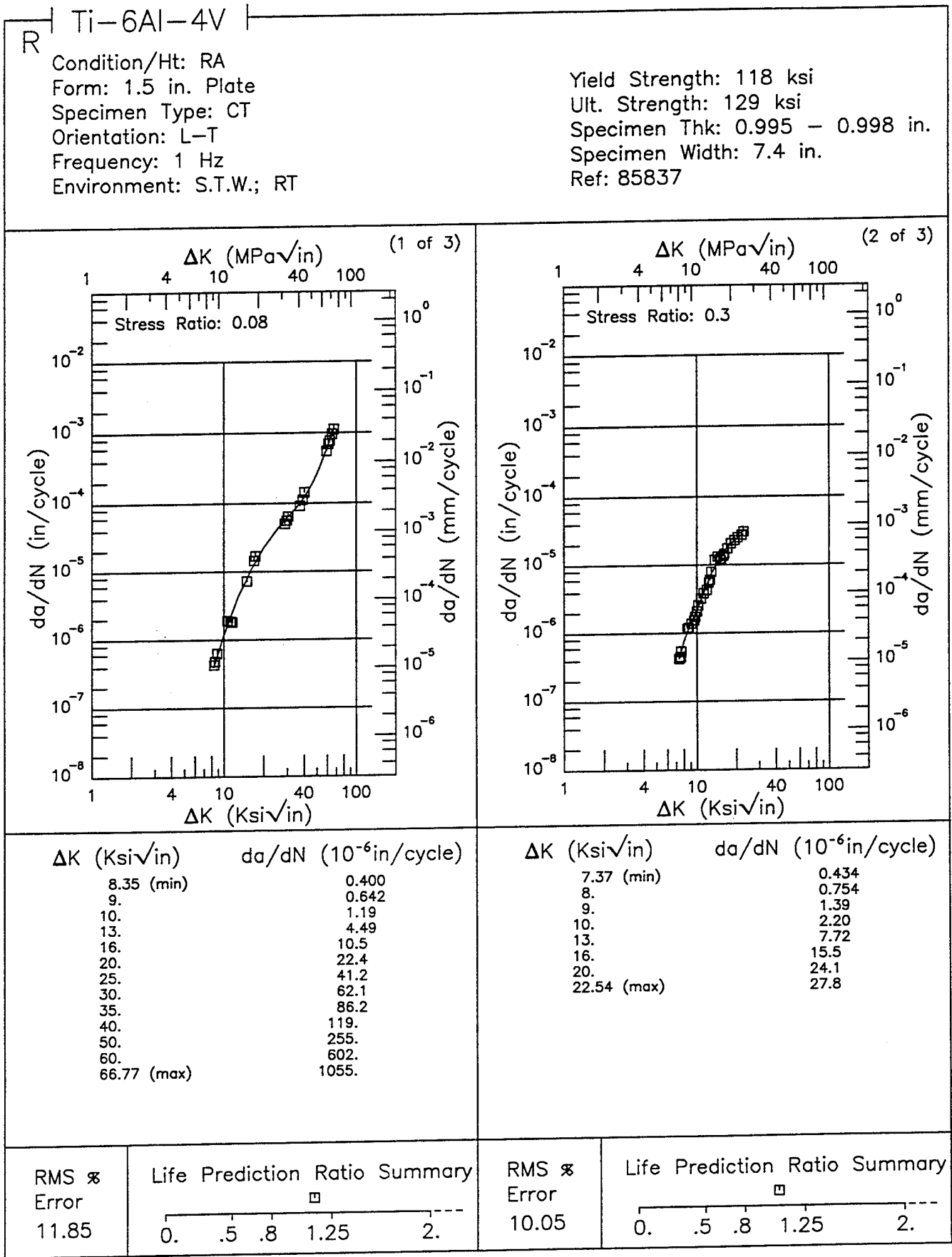
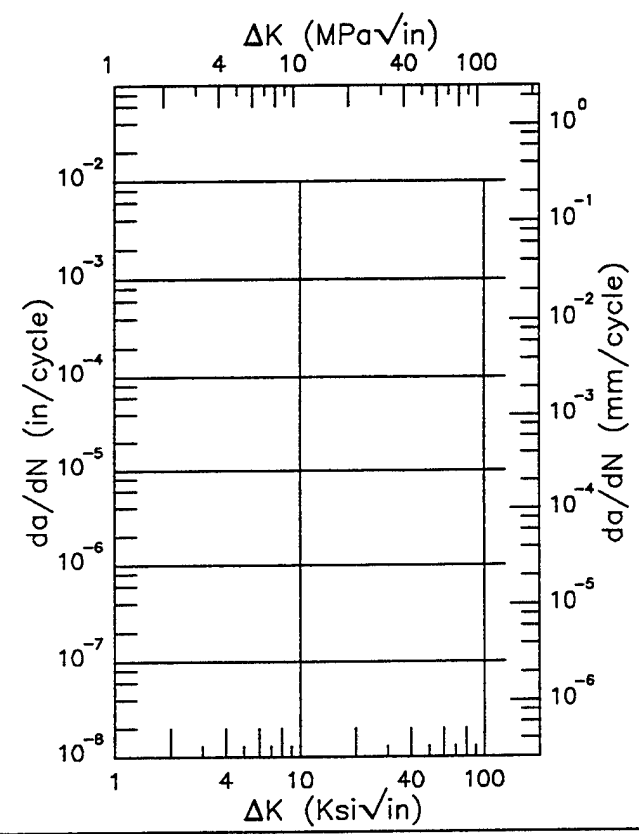
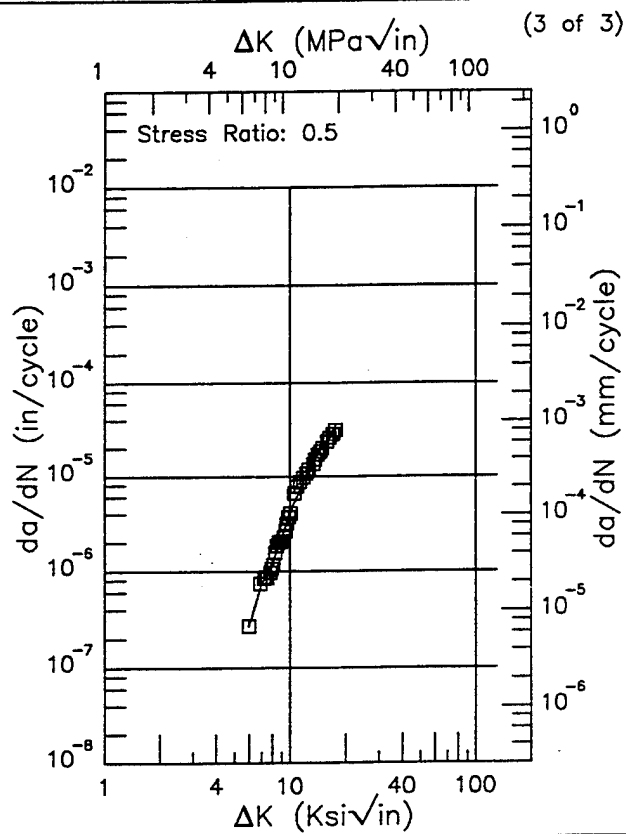


Figure 6.16.3.1.98

Ti-6Al-4V R

Condition/Ht: RA
 Form: 1.5 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Frequency: 1 Hz
 Environment: S.T.W.; RT

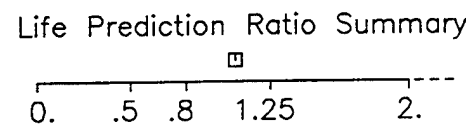
Yield Strength: 118 ksi
 Ult. Strength: 129 ksi
 Specimen Thk: 0.995 - 0.998 in.
 Specimen Width: 7.4 in.
 Ref: 85837



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
5.97 (min)	0.272
6.	0.284
7.	0.731
8.	1.24
9.	2.14
10.	4.18
13.	13.6
16.	24.8
17.45 (max)	29.3

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
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RMS \times
 Error
 10.63



RMS \times
 Error

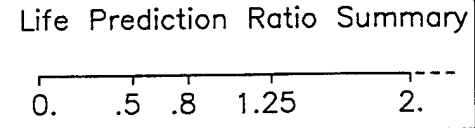


Figure 6.16.3.1.98 (Concluded)

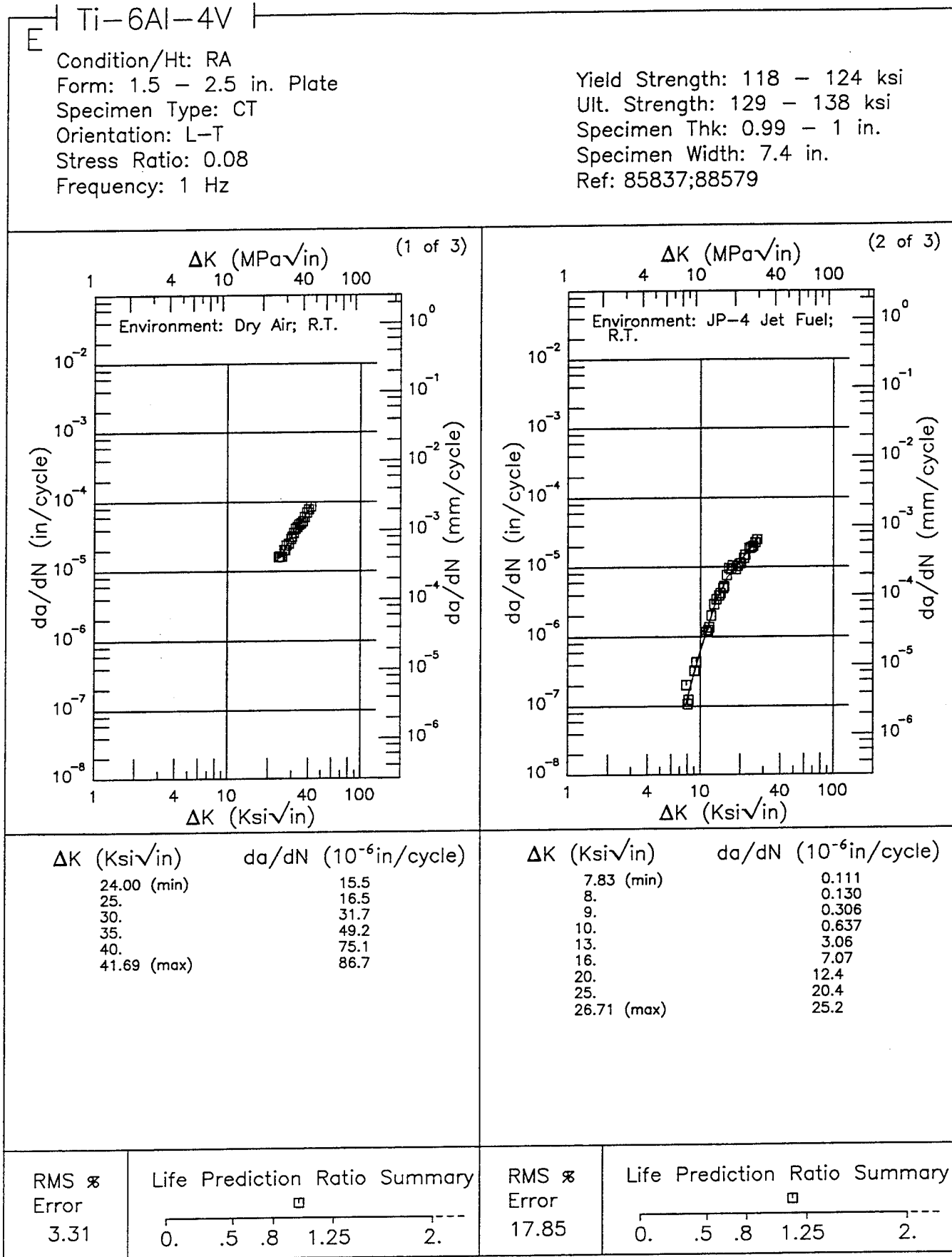


Figure 6.16.3.1.99

Ti-6Al-4V E

Condition/Ht: RA
 Form: 1.5 - 2.5 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.08
 Frequency: 1 Hz

Yield Strength: 118 - 124 ksi
 Ult. Strength: 129 - 138 ksi
 Specimen Thk: 0.99 - 1 in.
 Specimen Width: 7.39 - 7.4 in.
 Ref: 85837;88579

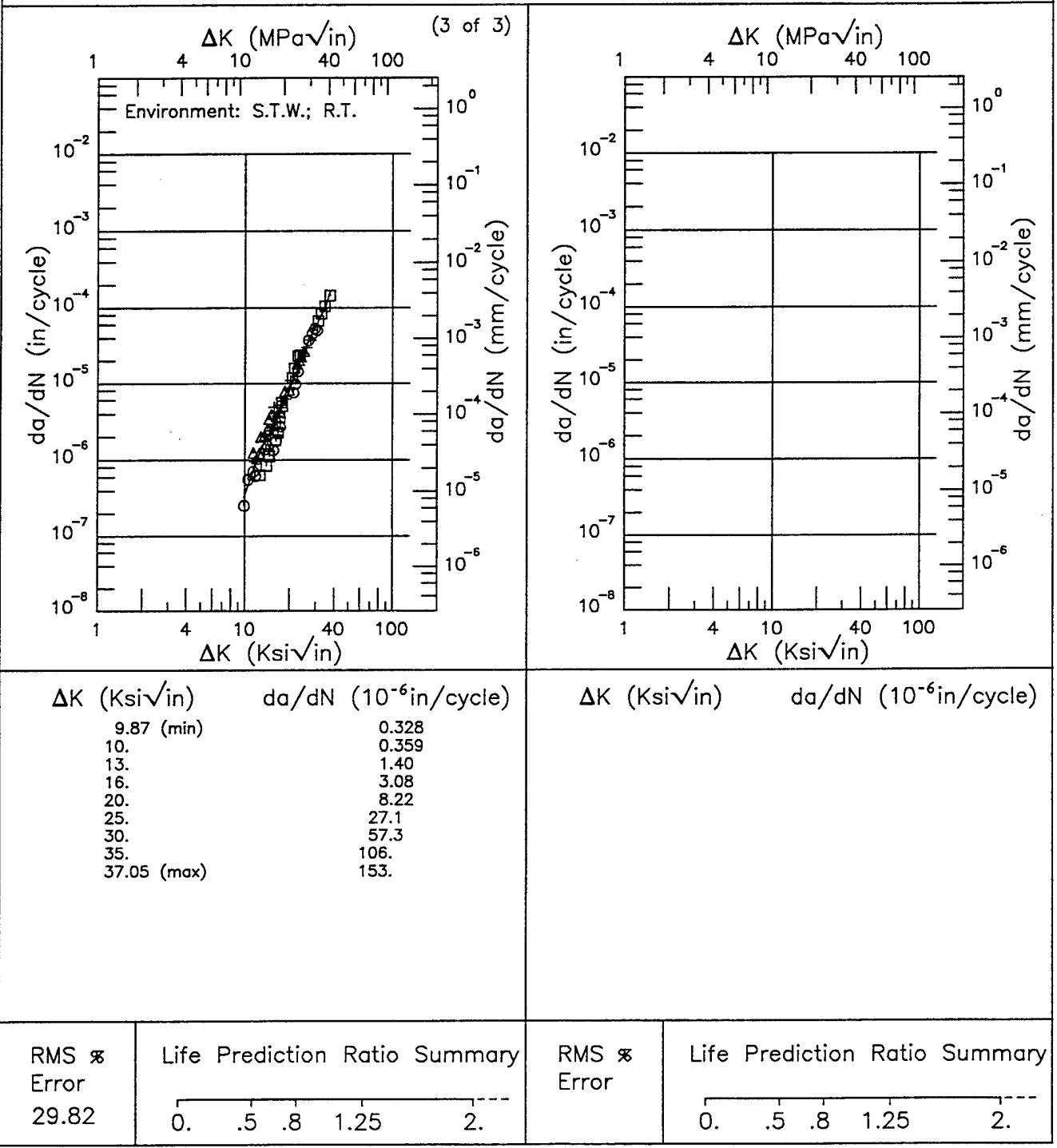


Figure 6.16.3.1.99 (Concluded)

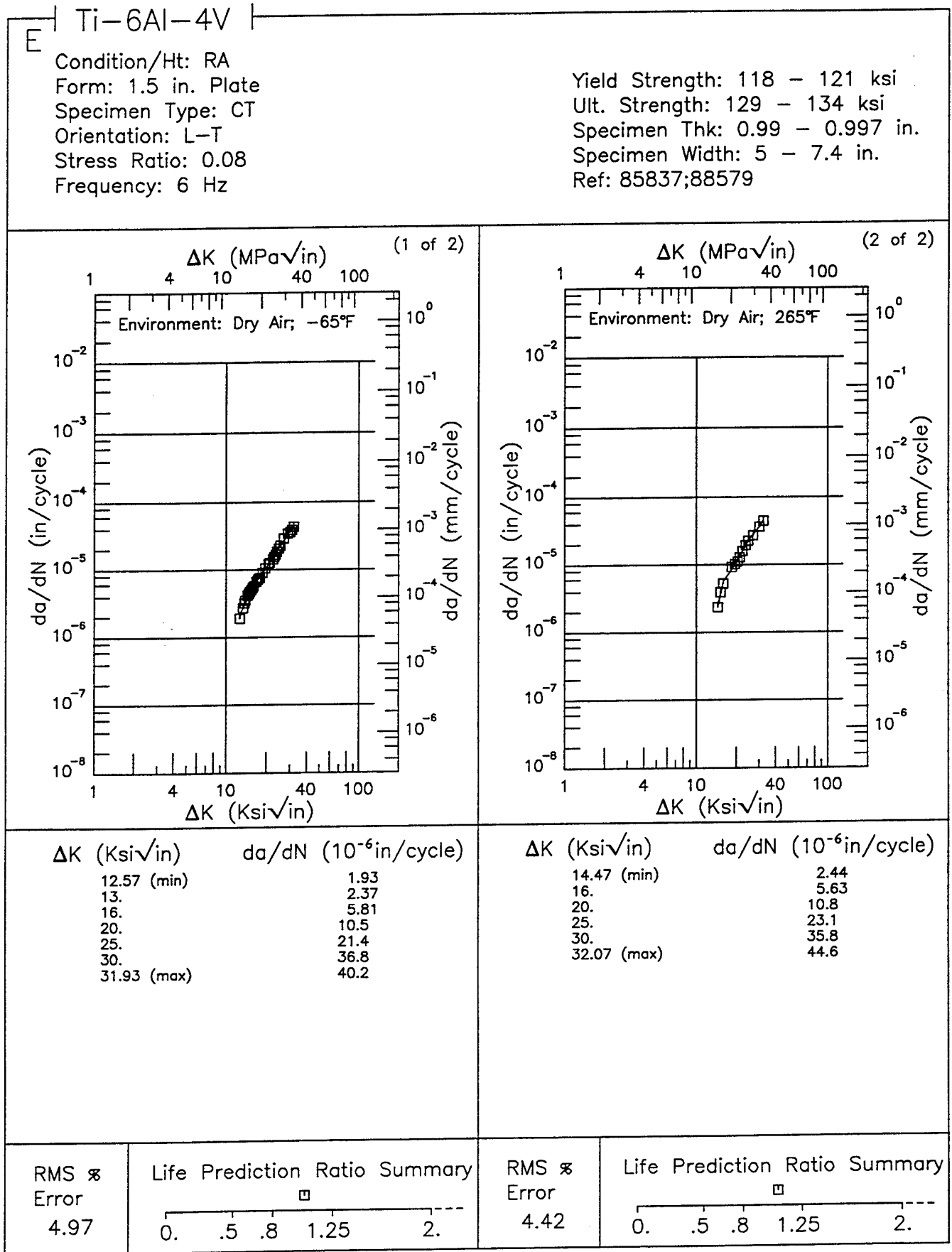


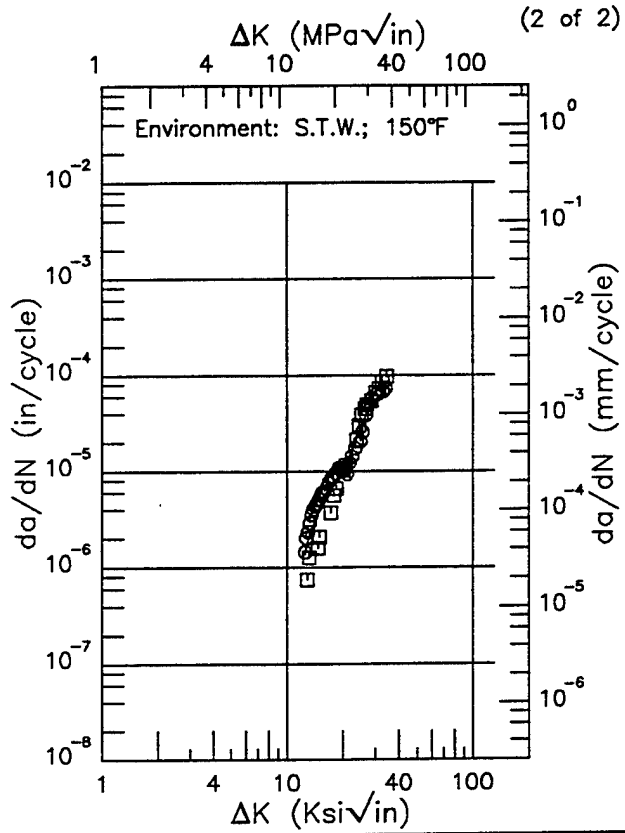
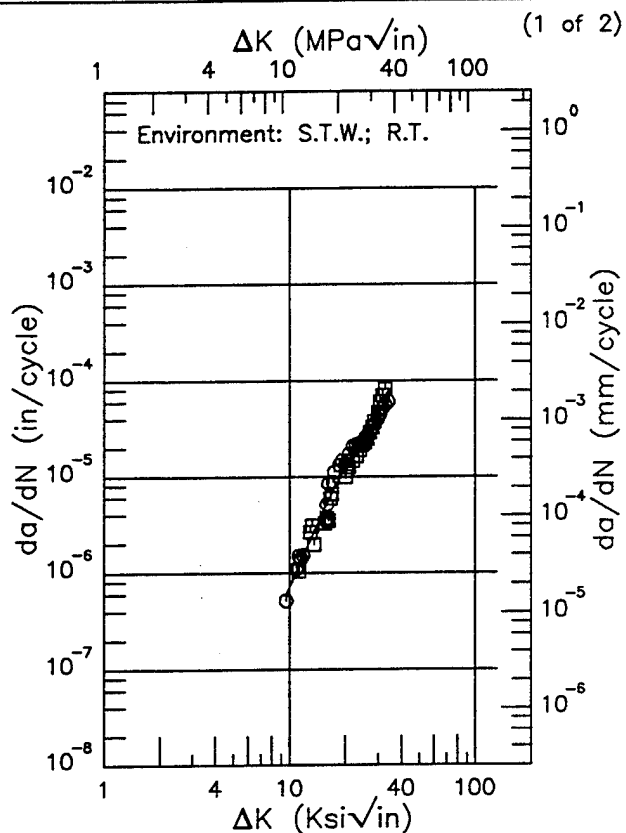
Figure 6.16.3.1.100

Ti-6Al-4V

E

Condition/Ht: RA
 Form: 1.5 - 2 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.08
 Frequency: 0.1 - 1 Hz

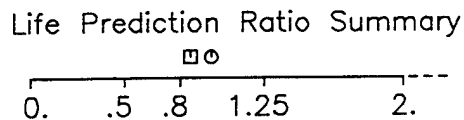
Yield Strength: 117 - 122 ksi
 Ult. Strength: 133 - 135 ksi
 Specimen Thk: 0.988 - 1.01 in.
 Specimen Width: 7.39 - 7.4 in.
 Ref: 88579;85837



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
9.52 (min)	0.585
10.	0.707
13.	2.18
16.	5.44
20.	12.9
25.	24.3
30.	43.0
33.81 (max)	76.9

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
12.49 (min)	1.35
13.	1.93
16.	5.61
20.	9.73
25.	28.1
30.	66.3
34.28 (max)	85.2

RMS %
 Error
 20.11



RMS %
 Error
 26.14

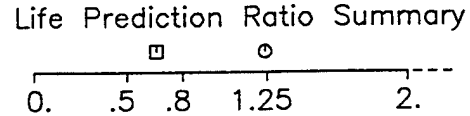


Figure 6.16.3.1.101

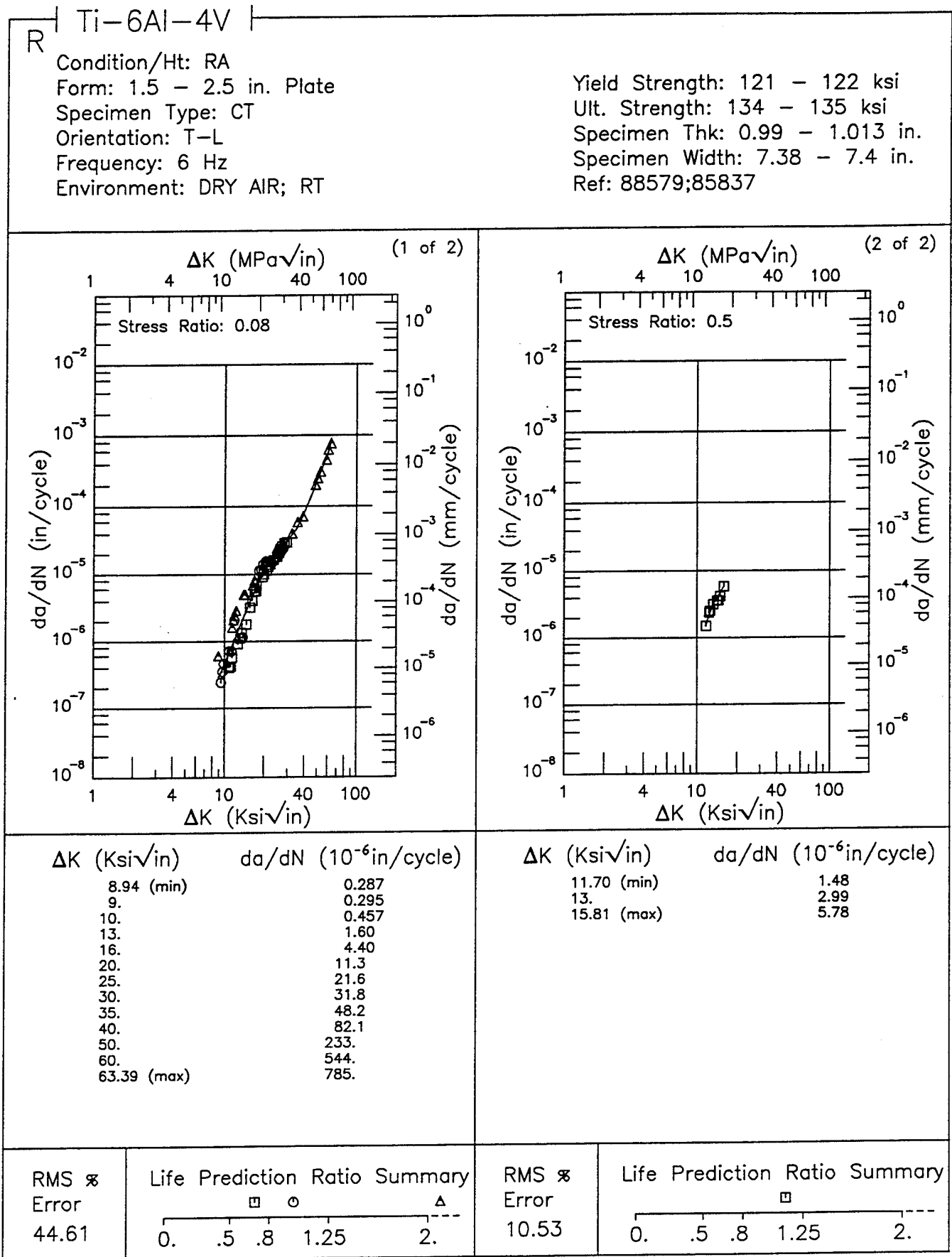
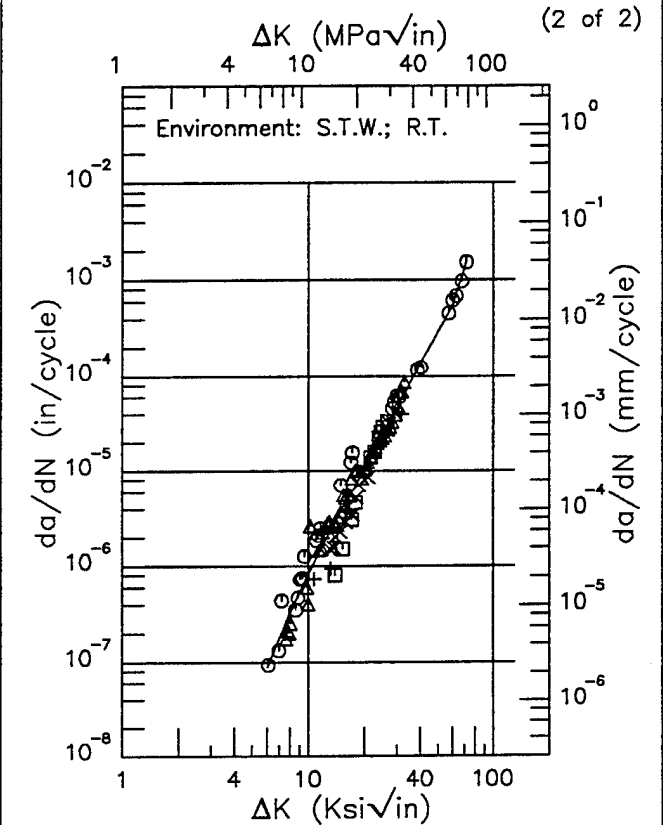
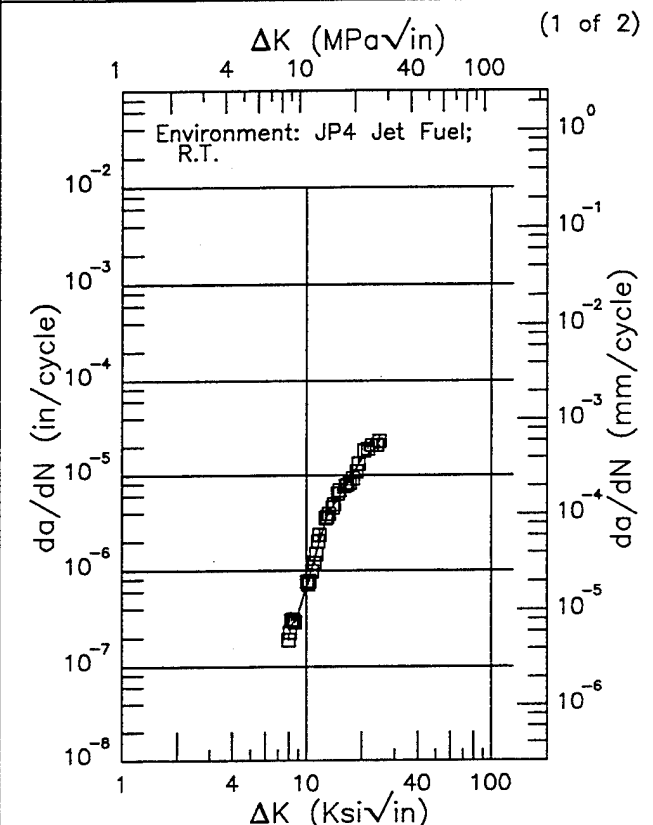


Figure 6.16.3.1.102

Ti-6Al-4V E

Condition/Ht: RA
 Form: 1.5 - 2.5 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.08
 Frequency: 1 Hz

Yield Strength: 122 - 126 ksi
 Ult. Strength: 134 - 135 ksi
 Specimen Thk: 0.994 - 1 in.
 Specimen Width: 7.4 in.
 Ref: 85837;88579



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
7.99 (min)	0.244
8.	0.245
9.	0.356
10.	0.692
13.	3.76
16.	7.32
20.	14.9
24.76 (max)	20.7

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
6.08 (min)	0.0881
7.	0.177
8.	0.326
9.	0.541
10.	0.832
13.	2.26
16.	4.77
20.	10.5
25.	23.2
30.	45.9
35.	83.0
40.	137.
50.	295.
60.	552.
70.	1325.
71.06 (max)	1488.

RMS % Error	Life Prediction Ratio Summary
10.46	

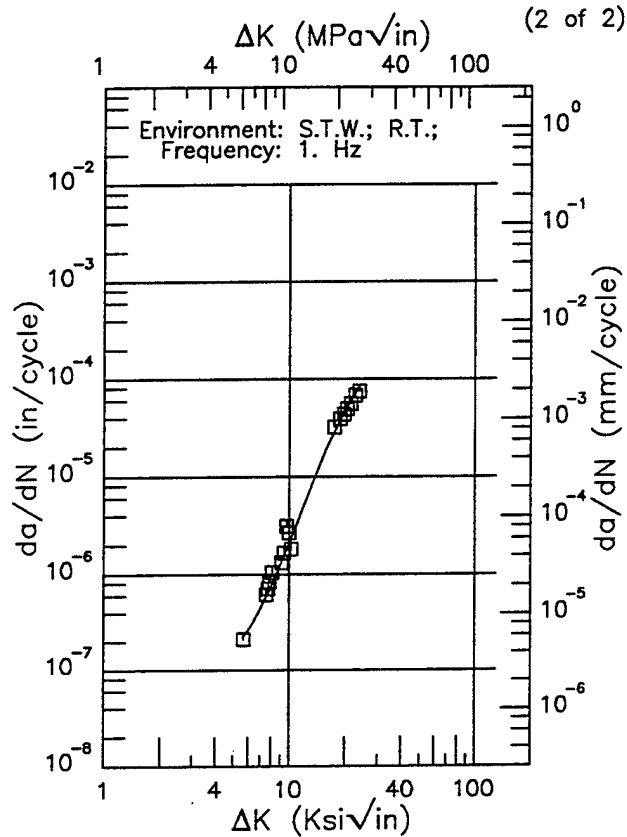
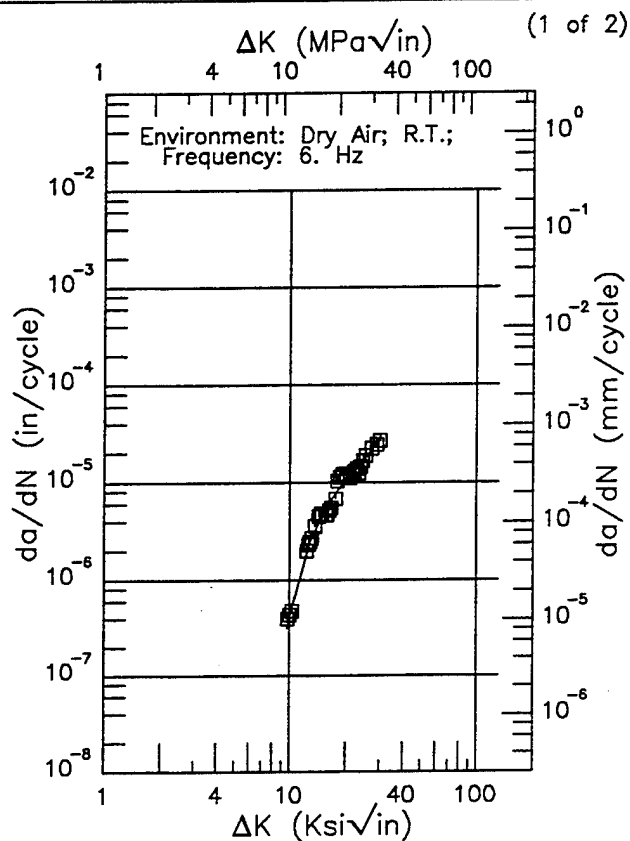
RMS % Error	Life Prediction Ratio Summary
41.07	

Figure 6.16.3.1.103

EF Ti-6Al-4V

Condition/Ht: RA
 Form: 0.38 - 1.5 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.08

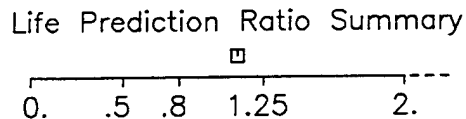
Yield Strength: 125 - 129 ksi
 Ult. Strength: 135 - 140 ksi
 Specimen Thk: 0.41 - 0.501 in.
 Specimen Width: 7.4 in.
 Ref: 85837;88579



ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
9.68 (min)	0.315
10.	0.425
13.	2.58
16.	6.03
20.	10.8
25.	17.2
29.94 (max)	27.4

ΔK (Ksi√in)	da/dN (10 ⁻⁶ in/cycle)
5.67 (min)	0.225
6.	0.262
7.	0.437
8.	0.747
9.	1.26
10.	2.08
13.	7.51
16.	19.9
20.	48.7
23.65 (max)	81.2

RMS %
 Error
 14.04



RMS %
 Error
 23.04

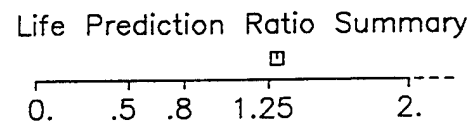
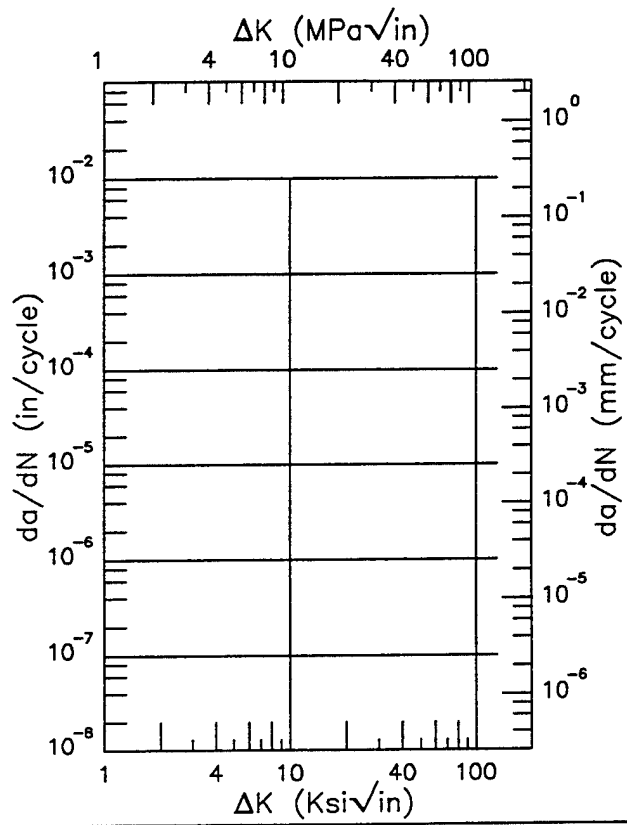
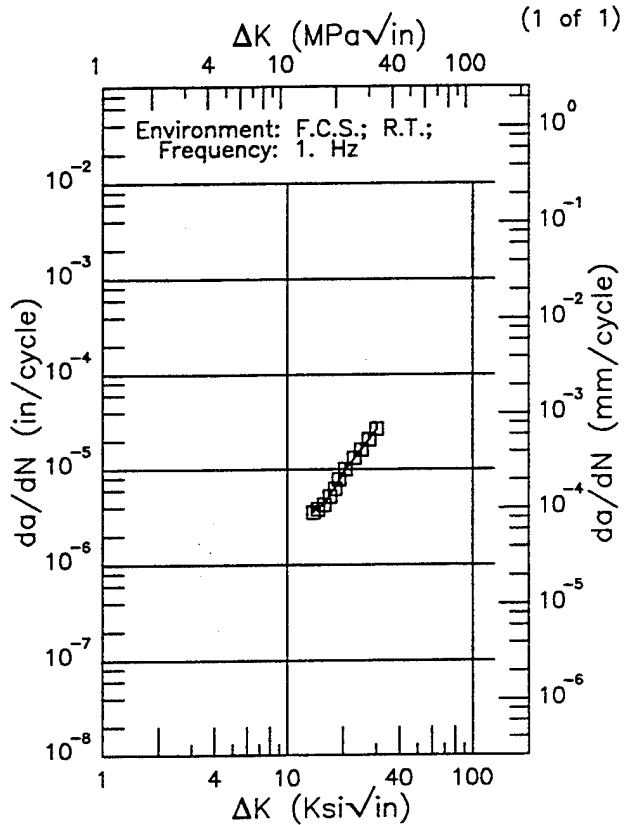


Figure 6.16.3.1.104

Ti-6Al-4V EF

Condition/Ht: RA
 Form: 1.5 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.08

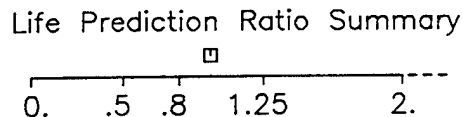
Yield Strength: 117 ksi
 Ult. Strength: 133 ksi
 Specimen Thk: 1.38 in.
 Specimen Width: 4 in.
 Ref: 88579



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.81 (min)	3.62
16.	4.34
20.	9.10
25.	16.4
30.	25.8
30.21 (max)	26.4

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS %
 Error
 2.42



RMS %
 Error

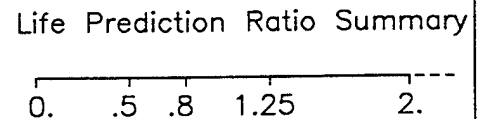
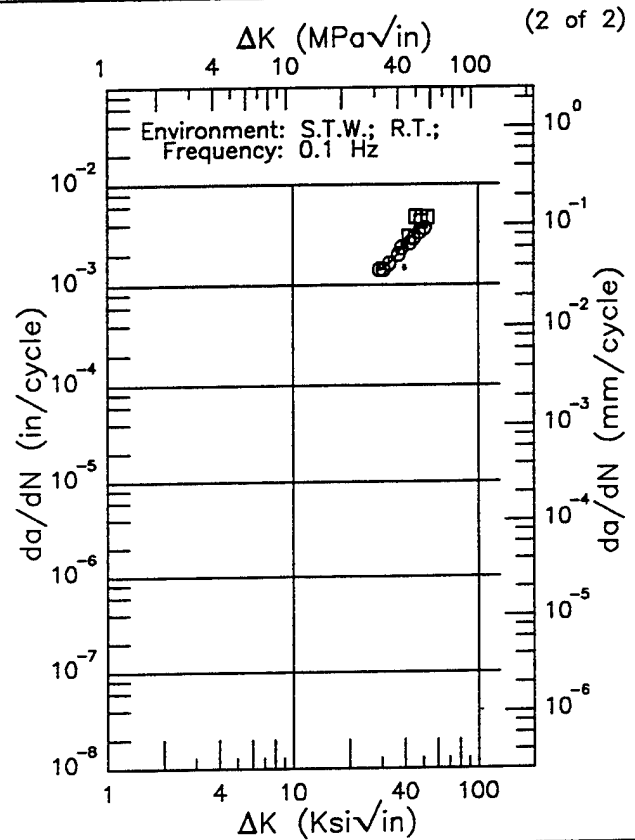
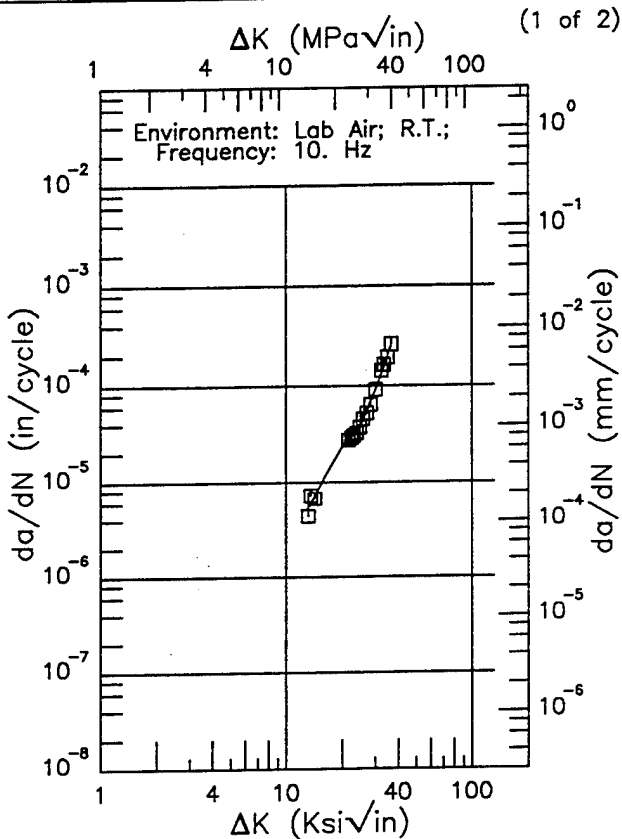


Figure 6.16.3.1.105

EF | Ti-6Al-4V |

Condition/Ht: RA
 Form: 0.38 in. Plate
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1

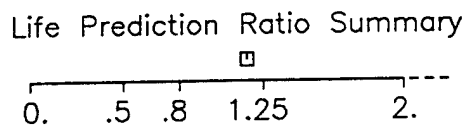
Yield Strength: 141.7 - 149.2 ksi
 Ult. Strength: 151.8 - 156.2 ksi
 Specimen Thk: 0.372 - 0.375 in.
 Specimen Width: 2.549 - 2.55 in.
 Ref: 90981



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
13.05 (min)	5.31
16.	11.3
20.	23.5
25.	39.0
30.	90.6
35.	204.
36.63 (max)	253.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
29.01 (min)	1405.
30.	1418.
35.	1800.
40.	2531.
50.	3951.
52.51 (max)	4043.

RMS %
 Error
 7.37



RMS %
 Error
 14.17

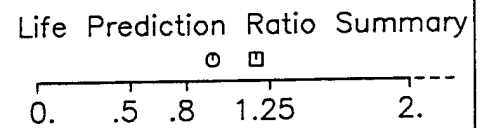
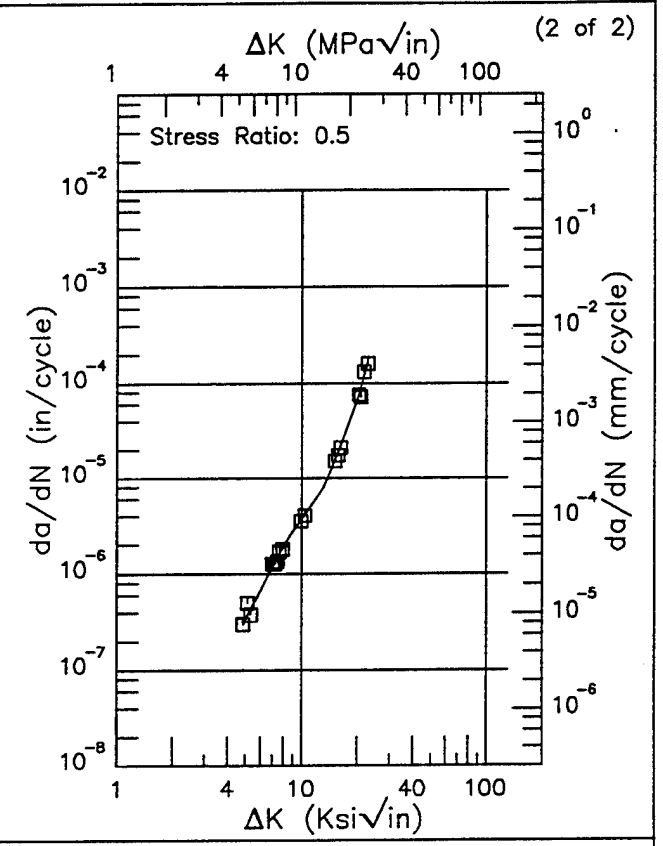
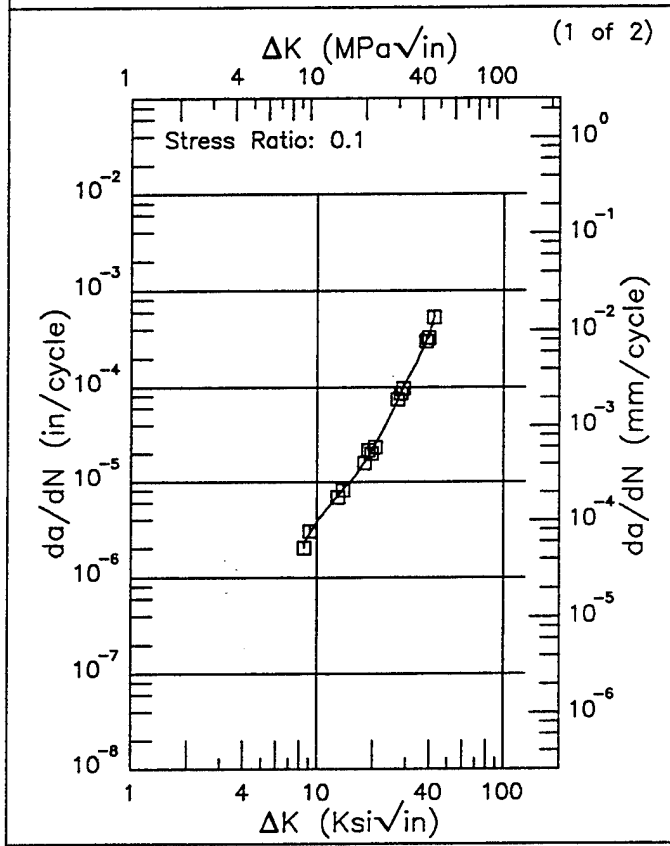


Figure 6.16.3.1.106

Ti-6Al-4V R

Condition/Ht: RA
 Form: 0.13 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: T-L
 Frequency: 10 Hz
 Environment: H.H.A.; RT

Yield Strength: 149.1 - 149.2 ksi
 Ult. Strength: 156.2 ksi
 Specimen Thk: 0.124 - 0.125 in.
 Specimen Width: 5 in.
 Ref: 90981



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
8.50 (min)	2.21
9.	2.75
10.	3.86
13.	7.21
16.	11.4
20.	22.1
25.	52.3
30.	115.
35.	207.
40.	365.
42.13 (max)	513.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
4.86 (min)	0.301
5.	0.335
6.	0.670
7.	1.19
8.	1.90
9.	2.74
10.	3.66
13.	7.71
16.	19.6
20.	65.8
22.81 (max)	171.

RMS % Error	Life Prediction Ratio Summary
6.15	

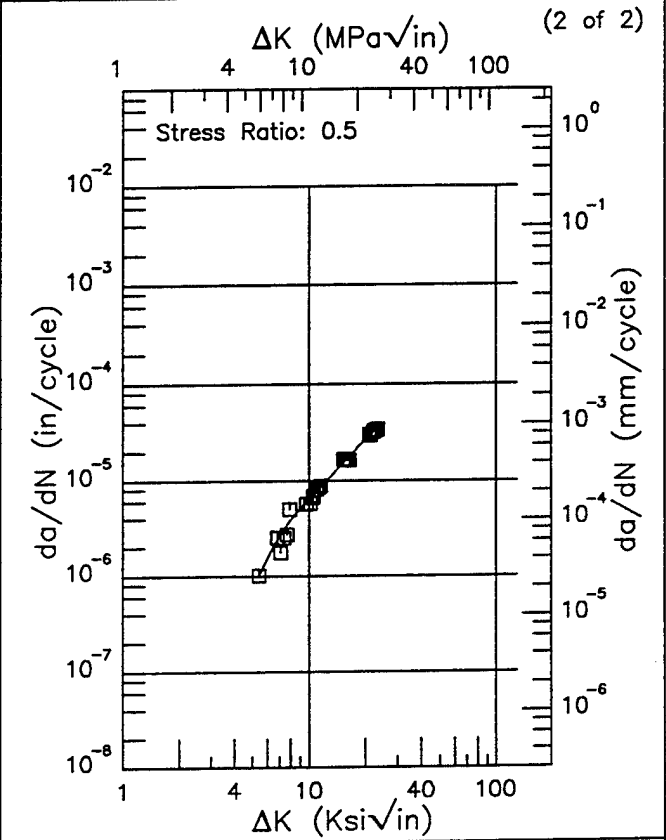
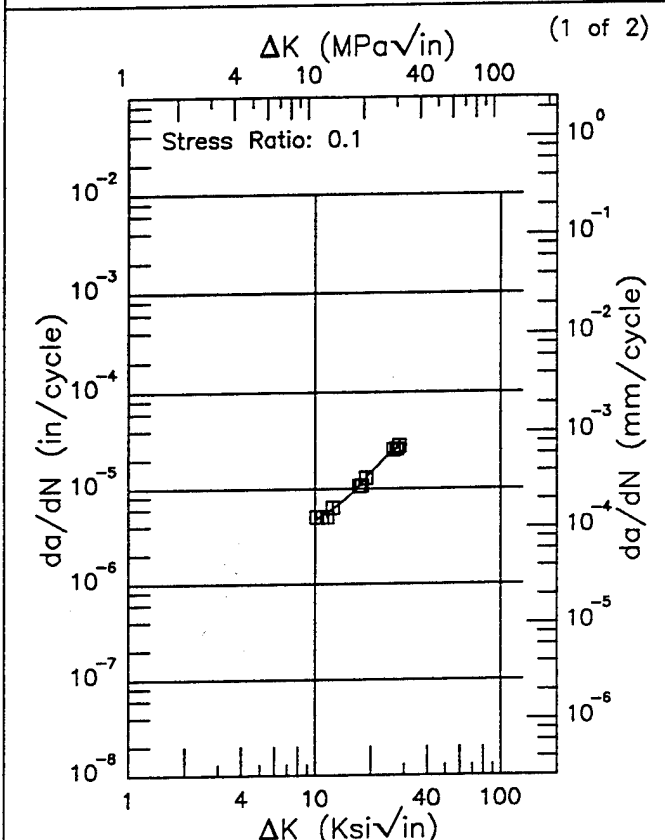
RMS % Error	Life Prediction Ratio Summary
12.10	

Figure 6.16.3.1.107

R Ti-6Al-4V

Condition/Ht: RA
 Form: 0.38 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: T-L
 Frequency: 10 Hz
 Environment: H.H.A.; RT

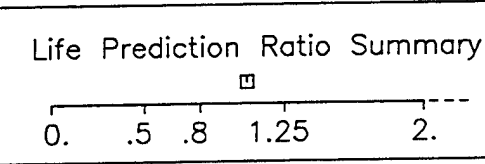
Yield Strength: 141.7 ksi
 Ult. Strength: 151.8 ksi
 Specimen Thk: 0.37 - 0.371 in.
 Specimen Width: 5 in.
 Ref: 90981



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
10.10 (min)	4.73
13.	6.48
16.	9.13
20.	14.6
25.	23.4
28.25 (max)	27.1

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
5.41 (min)	0.990
6.	1.54
7.	2.64
8.	3.85
9.	5.11
10.	6.41
13.	10.9
16.	16.8
20.	26.6
23.27 (max)	34.9

RMS % Error
4.83



RMS % Error
12.82

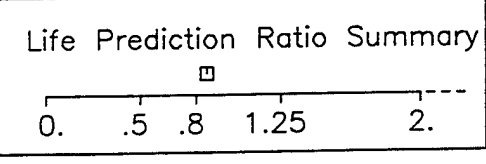
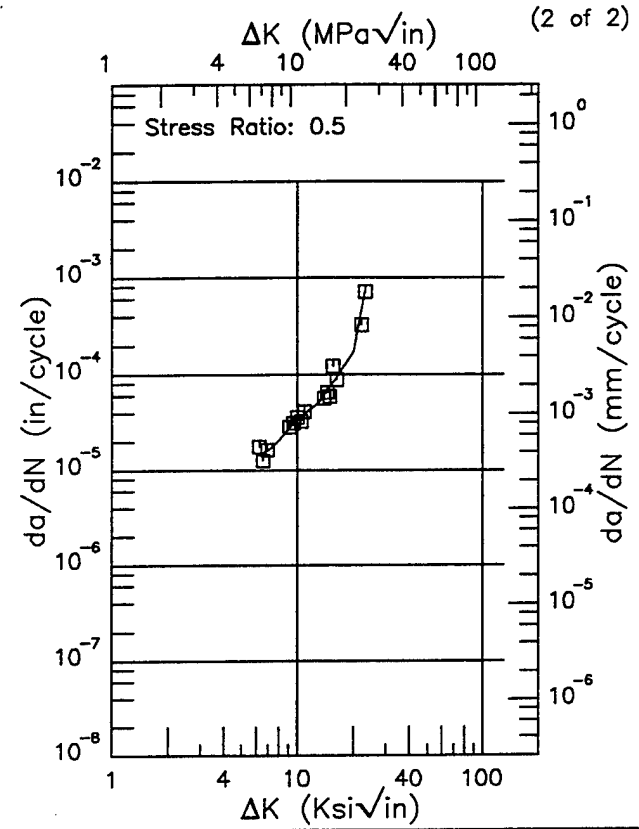
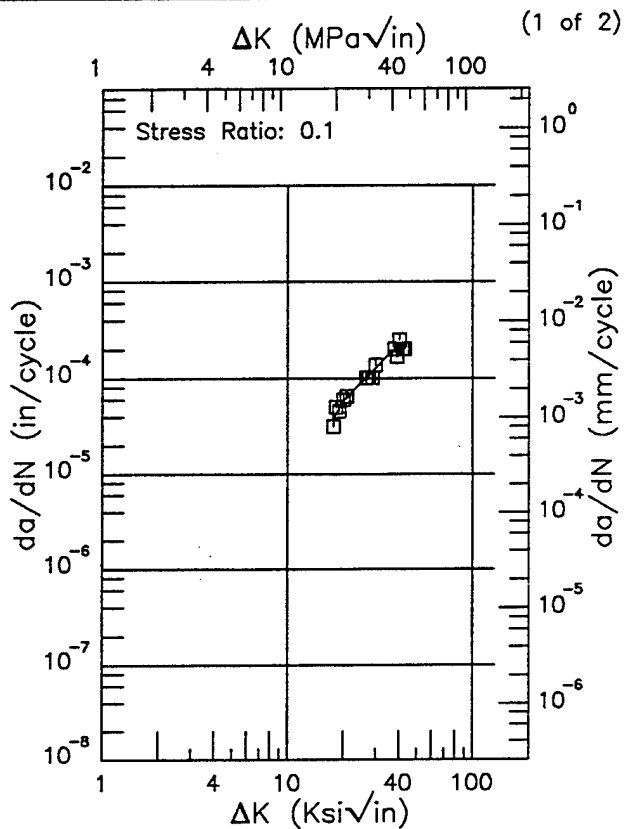


Figure 6.16.3.1.108

Ti-6Al-4V R

Condition/Ht: RA
 Form: 0.38 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: T-L
 Frequency: 10 Hz
 Environment: 3.5% NaCl; RT

Yield Strength: 141.7 ksi
 Ult. Strength: 151.8 ksi
 Specimen Thk: 0.372 in.
 Specimen Width: 5 in.
 Ref: 90981



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
17.57 (min)	37.8
20.	57.5
25.	91.7
30.	128.
35.	177.
40.	208.
42.87 (max)	202.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
6.24 (min)	14.4
7.	16.1
8.	20.7
9.	27.2
10.	34.2
13.	51.0
16.	88.0
20.	171.
23.07 (max)	704.

RMS % Error	Life Prediction Ratio Summary
11.36	

RMS % Error	Life Prediction Ratio Summary
17.38	

Figure 6.16.3.1.109

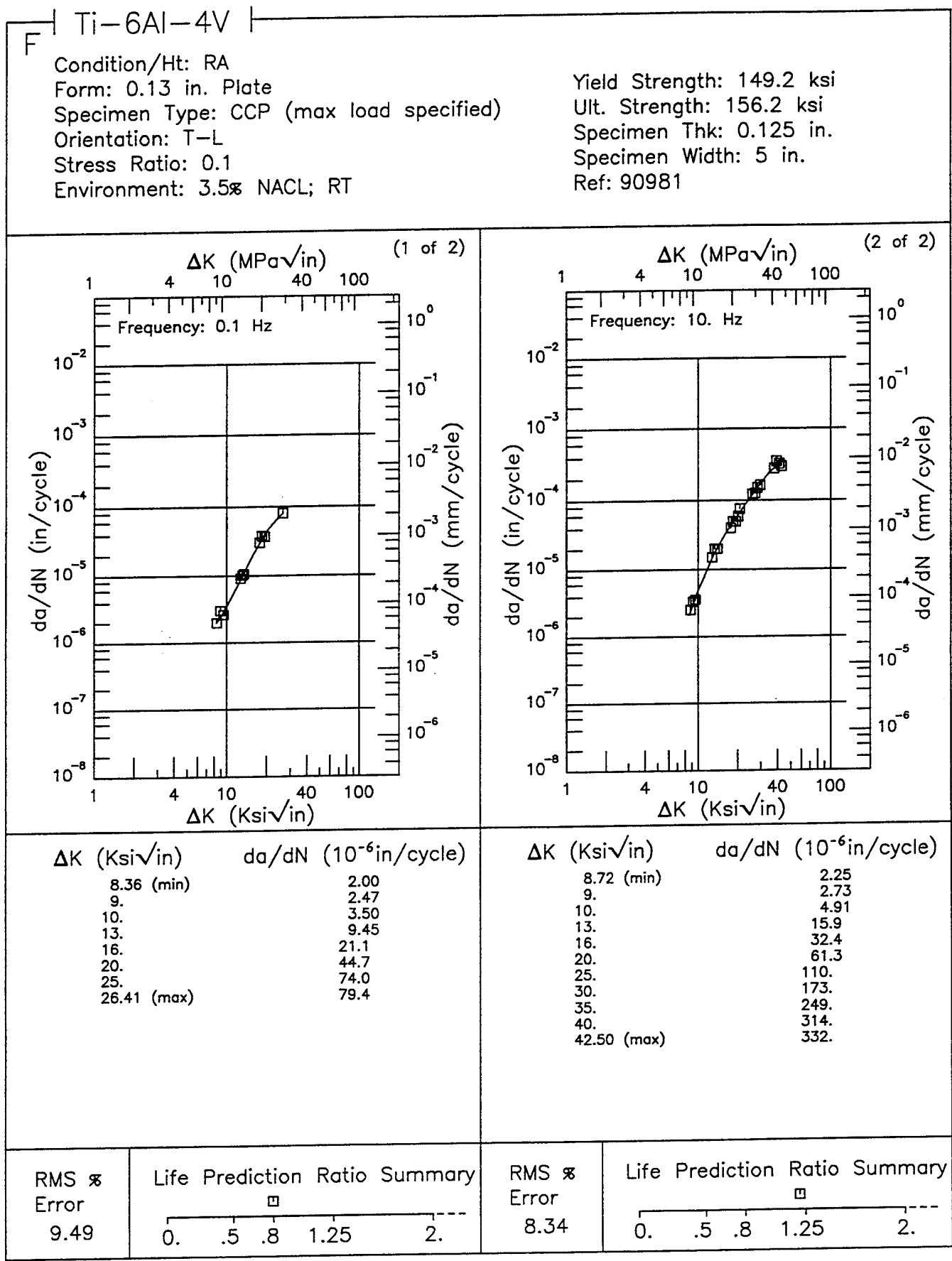


Figure 6.16.3.1.110

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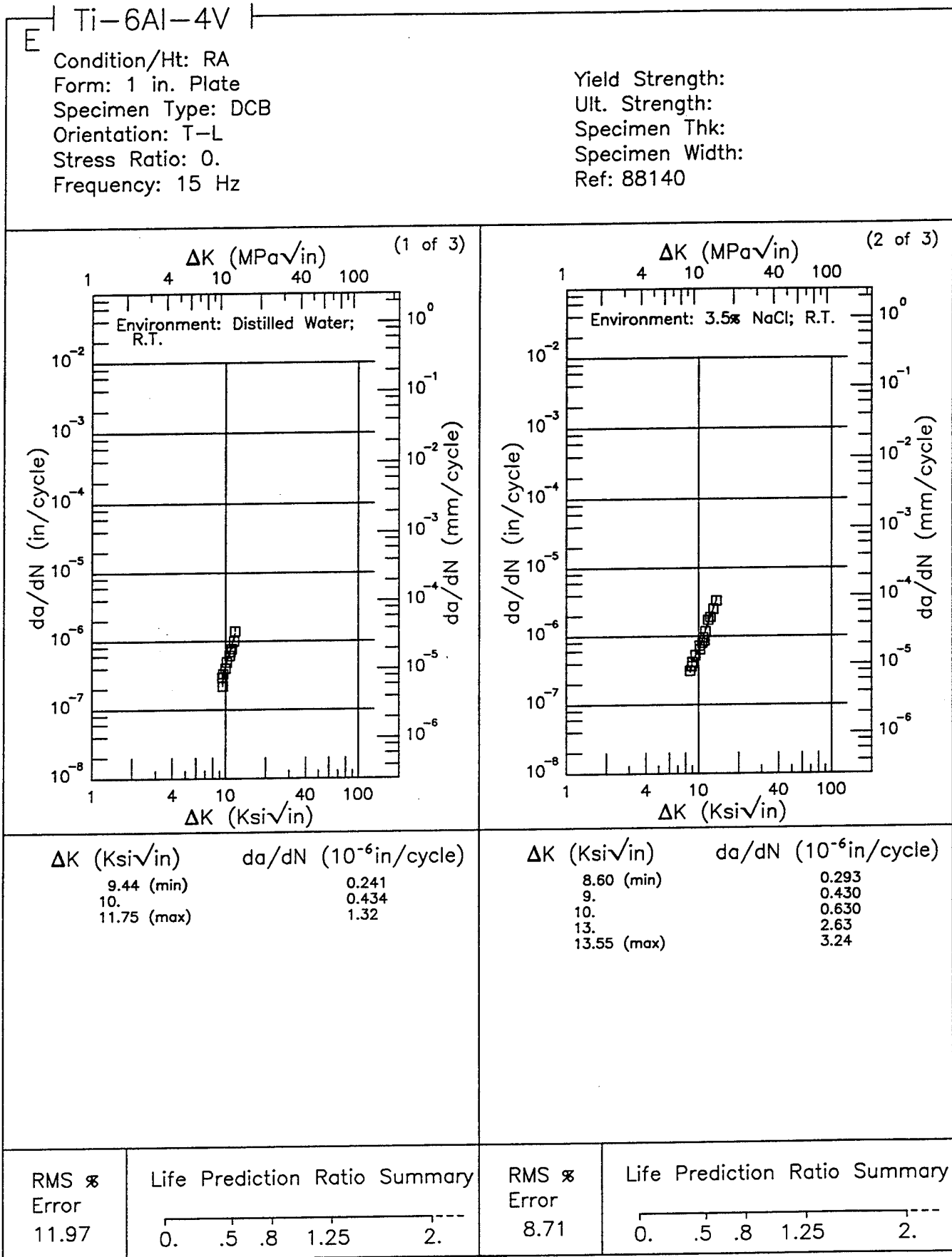
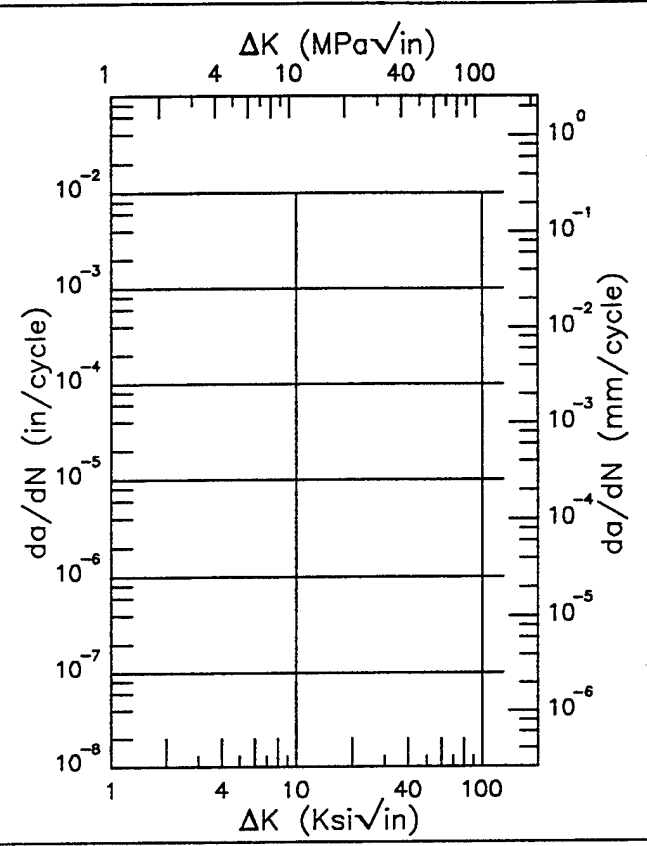
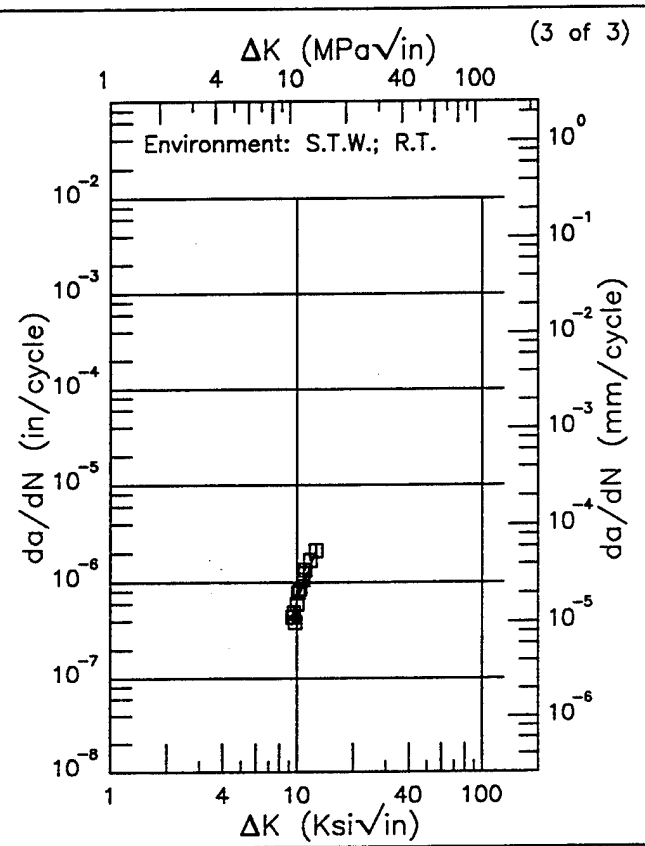


Figure 6.16.3.1.111

Ti-6Al-4V E

Condition/Ht: RA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.
 Frequency: 15 Hz

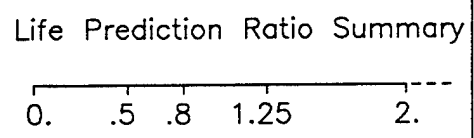
Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
9.40 (min)	0.365
10.	0.646
12.61 (max)	2.09

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS %
 Error
 17.28



RMS %
 Error

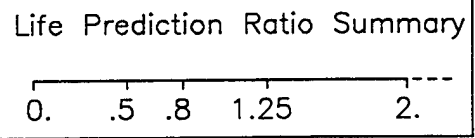


Figure 6.16.3.1.111 (Concluded)

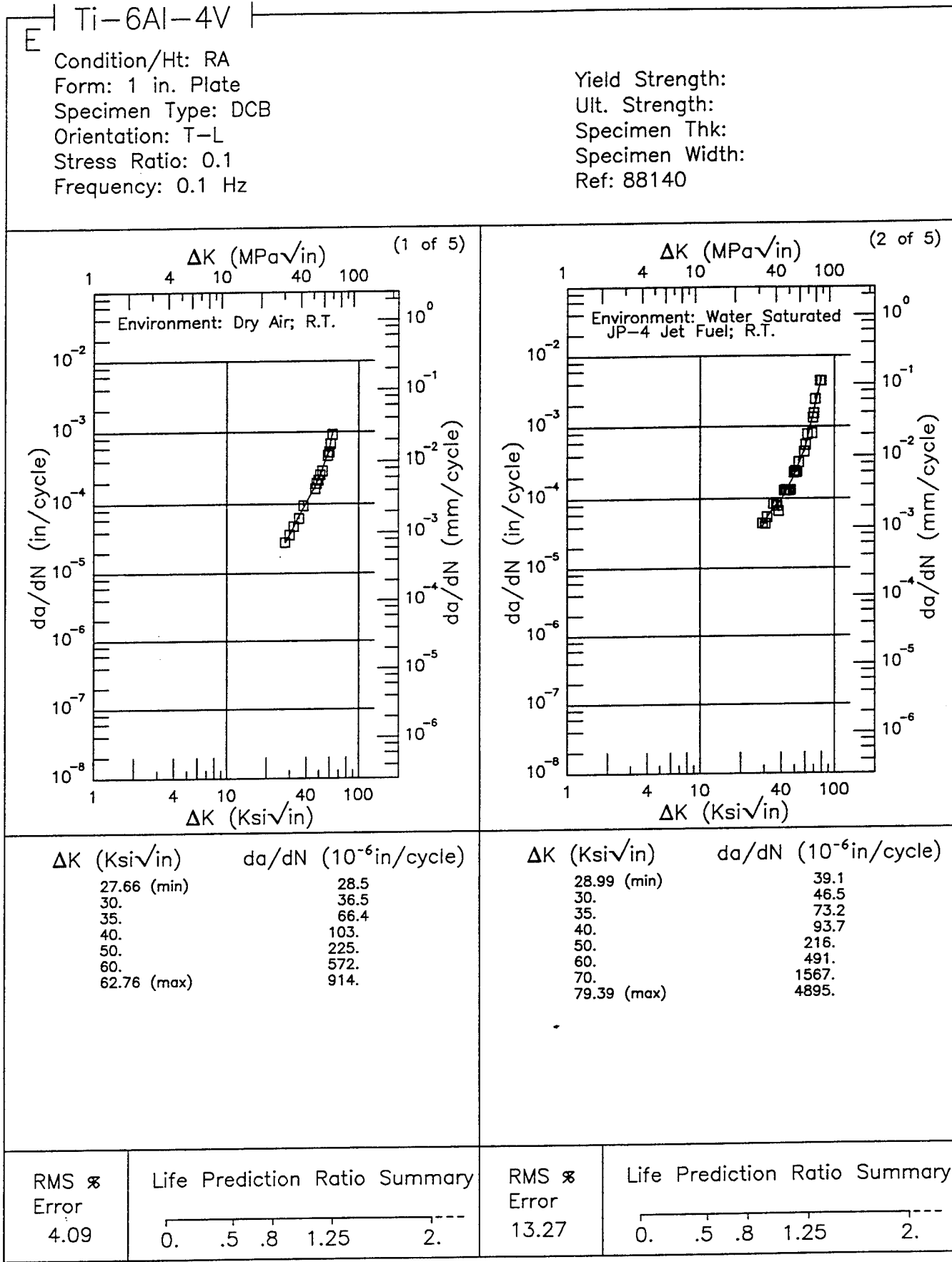
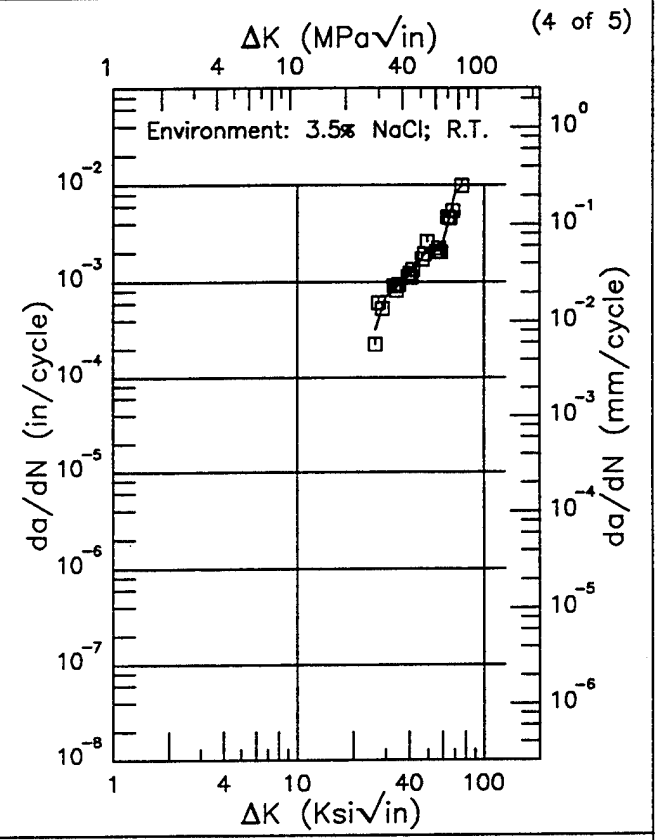
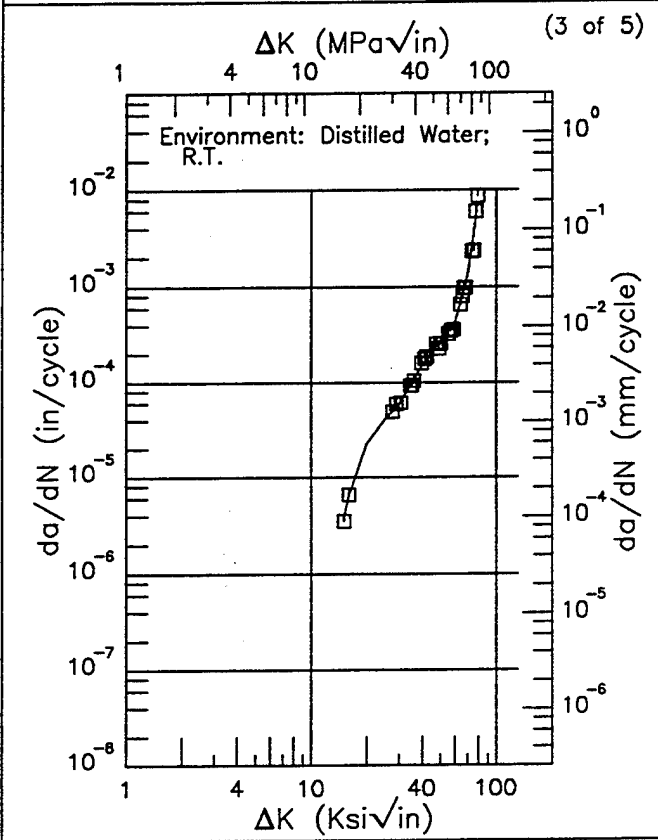


Figure 6.16.3.1.112

Ti-6Al-4V E

Condition/Ht: RA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.1
 Frequency: 0.1 Hz

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
14.90 (min)	3.68
16.	6.74
20.	22.5
25.	40.4
30.	59.8
35.	96.4
40.	154.
50.	270.
60.	428.
70.	1327.
78.93 (max)	7237.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
26.28 (min)	321.
30.	697.
35.	928.
40.	1173.
50.	2144.
60.	2469.
70.	8074.
75.42 (max)	9712.

RMS % Error	Life Prediction Ratio Summary
10.54	

RMS % Error	Life Prediction Ratio Summary
16.28	

Figure 6.16.3.1.112 (Continued)

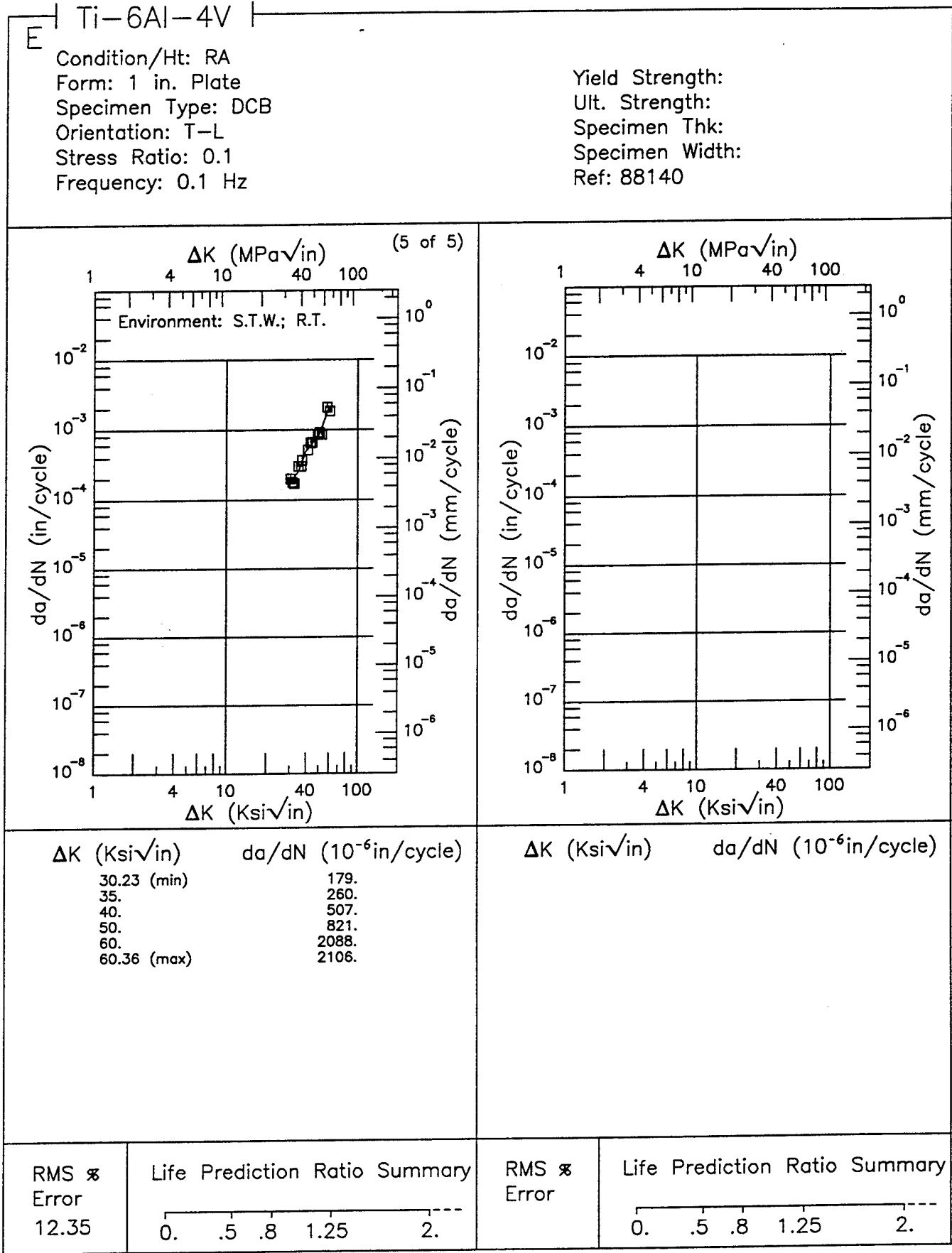


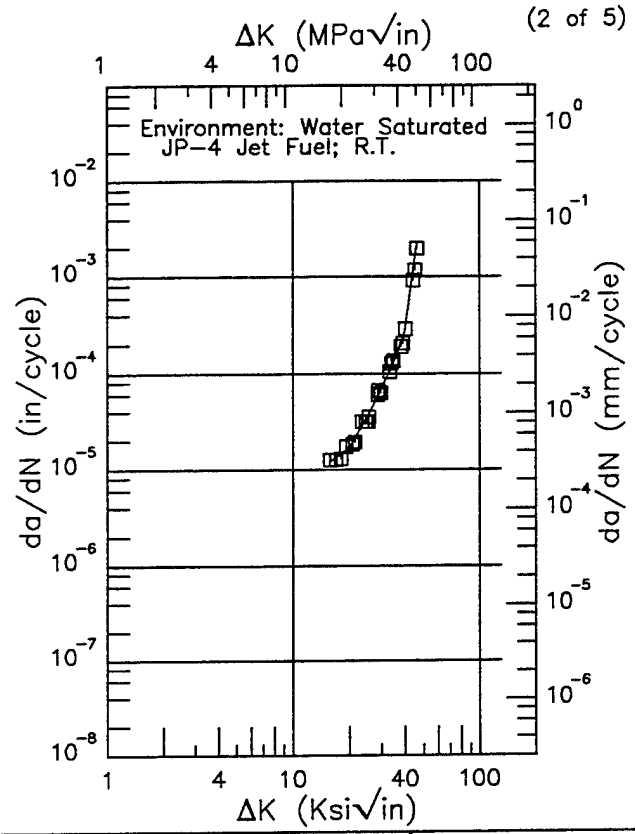
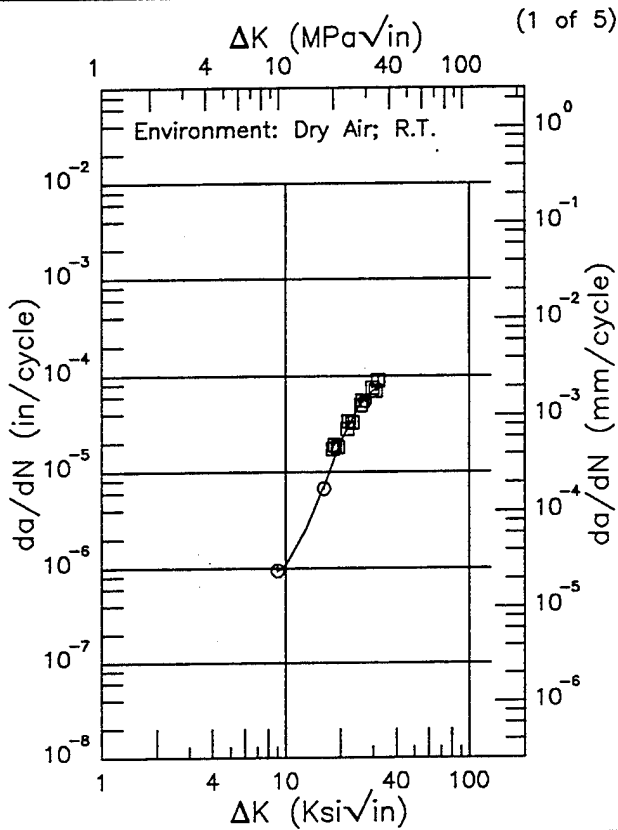
Figure 6.16.3.1.112 (Concluded)

Ti-6Al-4V

E

Condition/Ht: RA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.5
 Frequency: 0.1 Hz

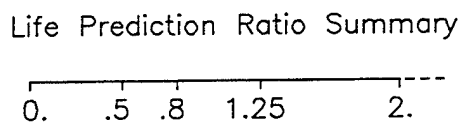
Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.99 (min)	0.930
9.	0.930
10.	1.08
13.	2.57
16.	6.86
20.	20.4
25.	48.6
30.	70.2
31.84 (max)	72.2

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
15.67 (min)	12.1
16.	12.2
20.	17.2
25.	35.1
30.	71.4
35.	136.
40.	274.
46.00 (max)	2017.

RMS %
 Error
 16.54



RMS %
 Error
 9.48

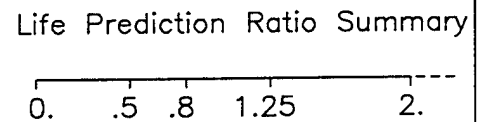
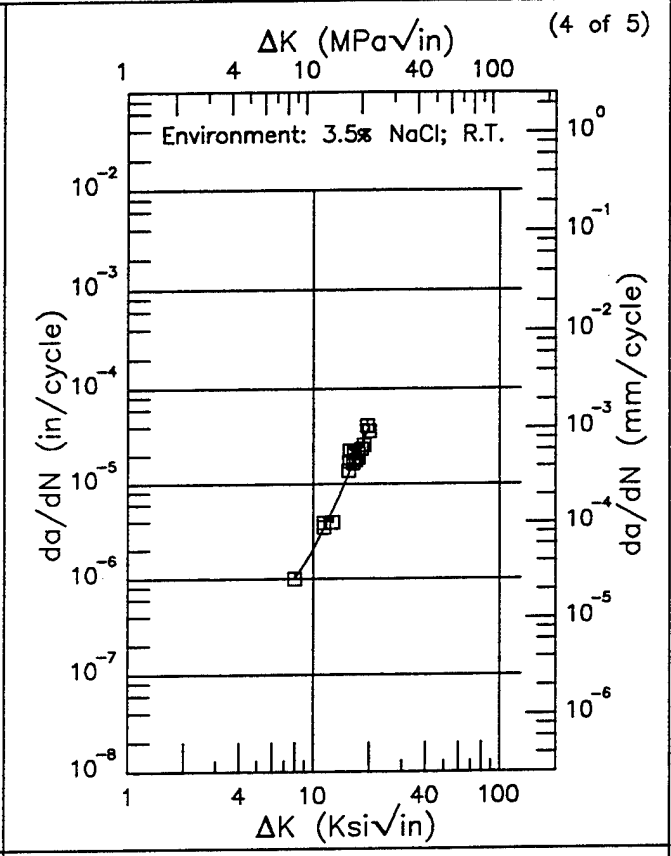
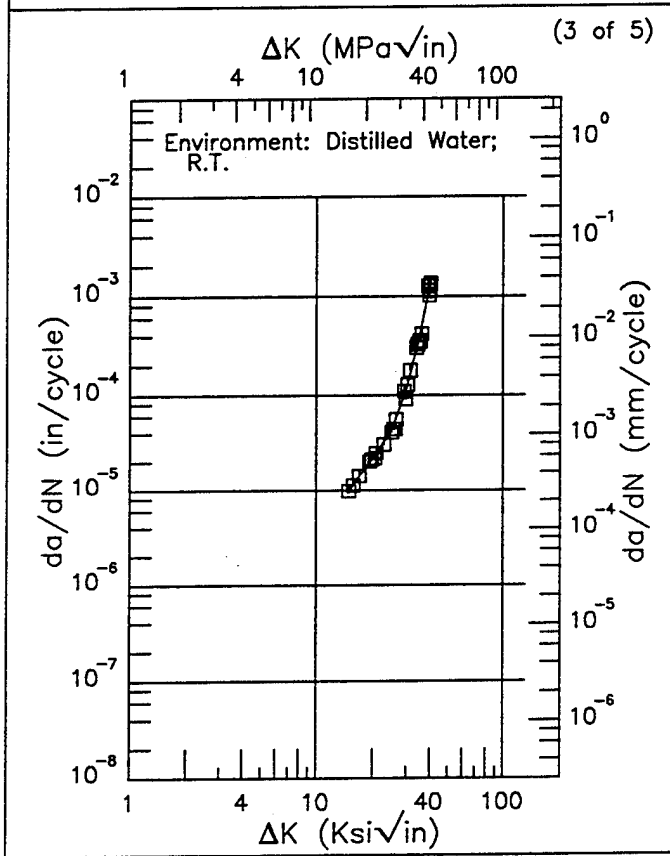


Figure 6.16.3.1.113

E | Ti-6Al-4V |

Condition/Ht: RA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.5
 Frequency: 0.1 Hz

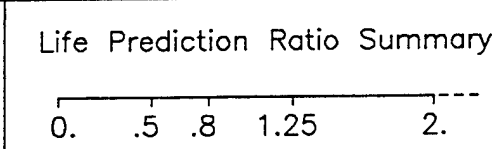
Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
14.81 (min)	9.39
16.	12.2
20.	21.8
25.	43.4
30.	117.
35.	391.
40.	1347.
40.19 (max)	1411.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
7.97 (min)	0.998
8.	1.01
9.	1.42
10.	2.03
13.	5.75
16.	14.5
19.93 (max)	40.2

RMS % Error
11.14



RMS % Error
22.59

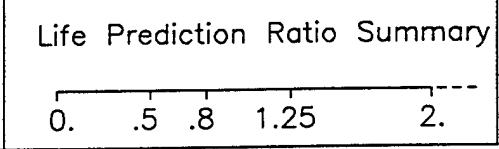
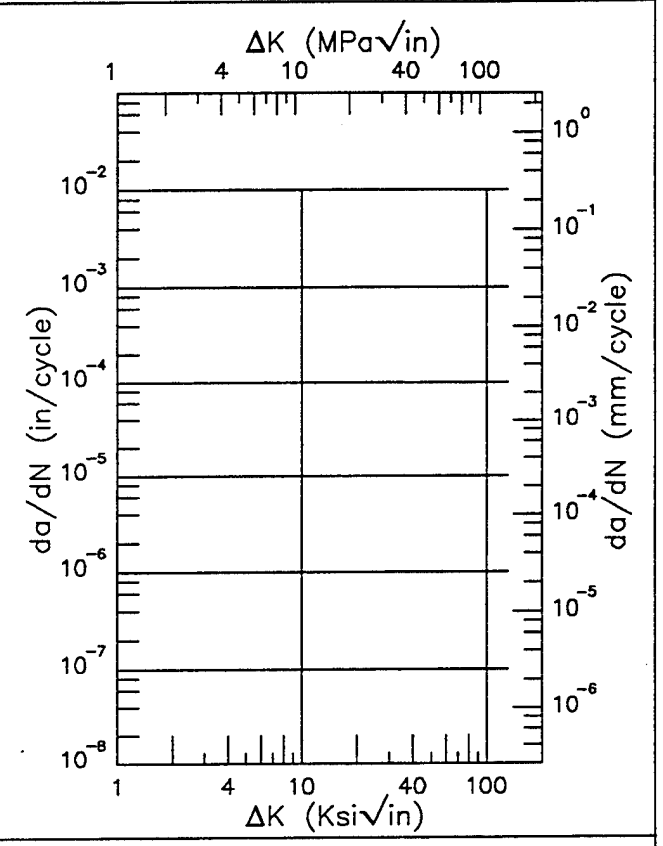
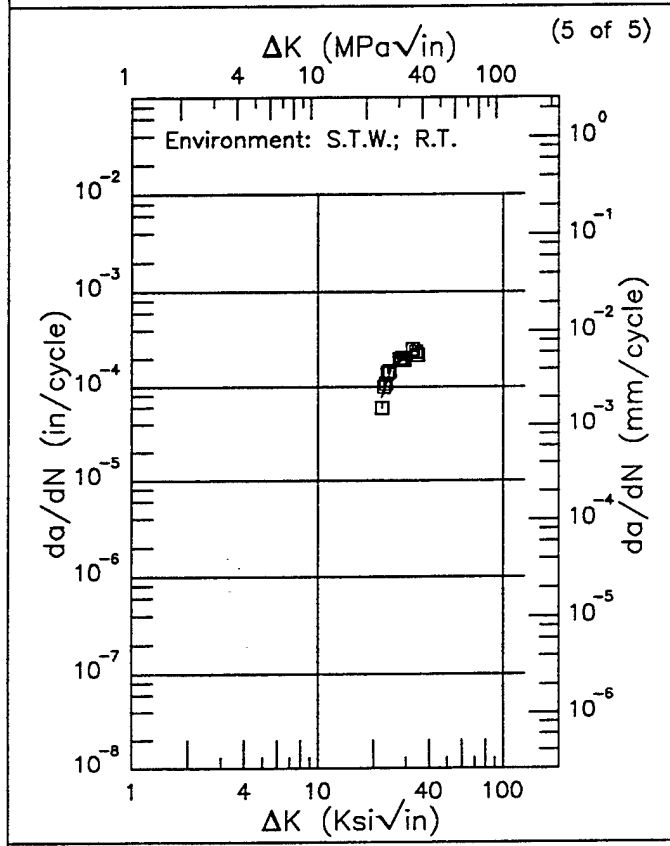


Figure 6.16.3.1.113 (Continued)

Ti-6Al-4V E

Condition/Ht: RA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.5
 Frequency: 0.1 Hz

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
22.18 (min)	77.7
25.	160.
30.	209.
34.79 (max)	262.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
22.18 (min)	77.7
25.	160.
30.	209.
34.79 (max)	262.

RMS % Error	Life Prediction Ratio Summary
12.42	0. .5 .8 1.25 2.

RMS % Error	Life Prediction Ratio Summary
	0. .5 .8 1.25 2.

Figure 6.16.3.1.113 (Concluded)

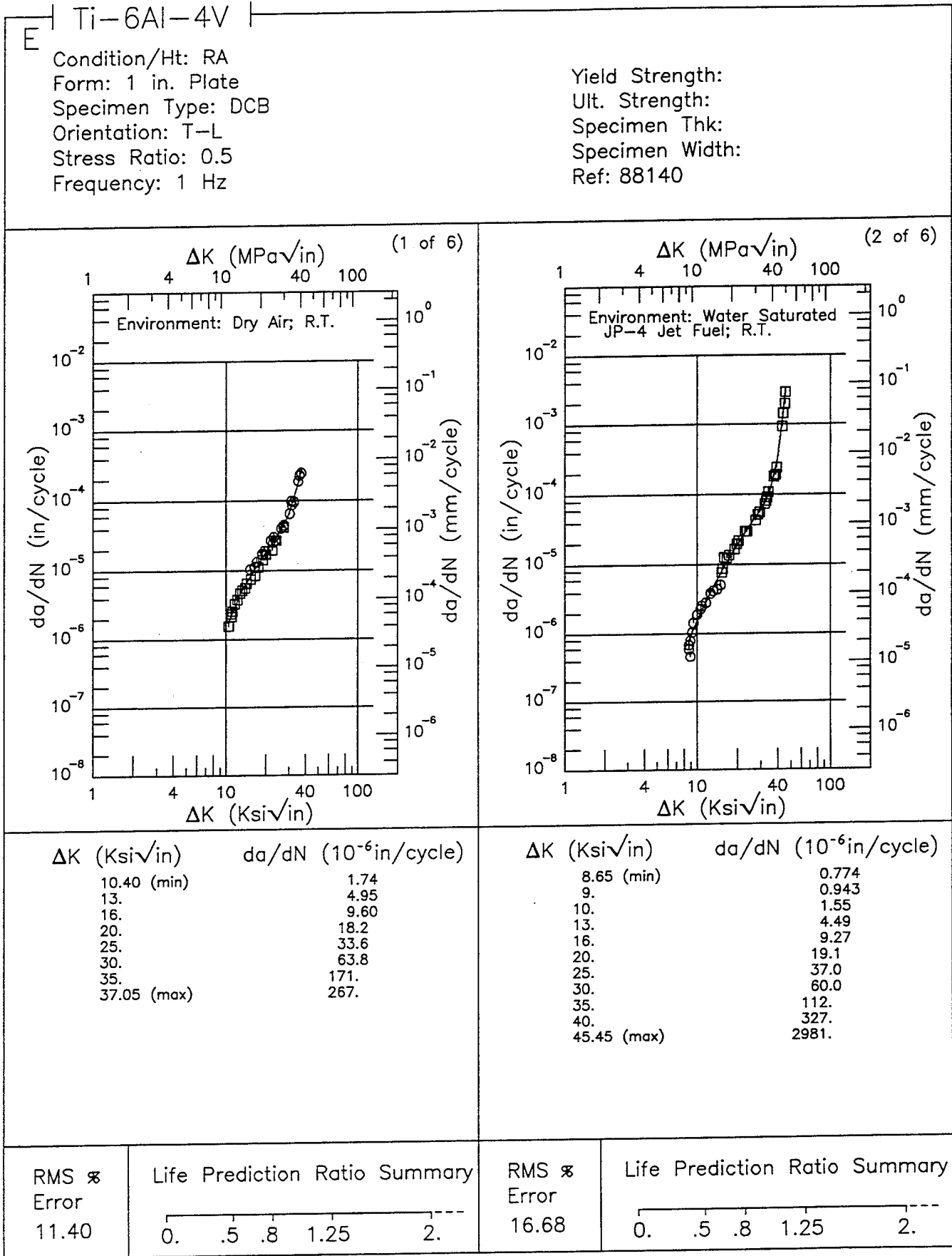
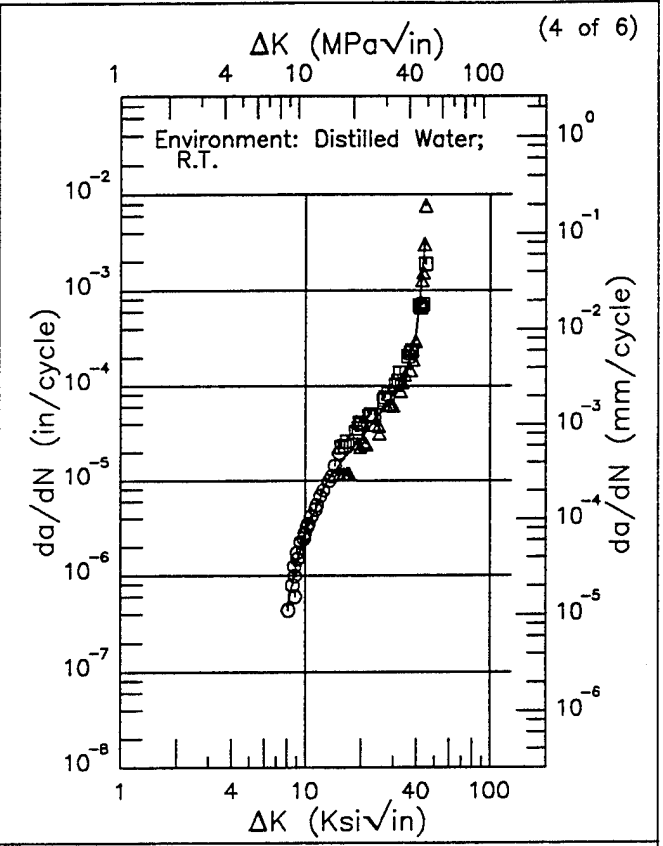
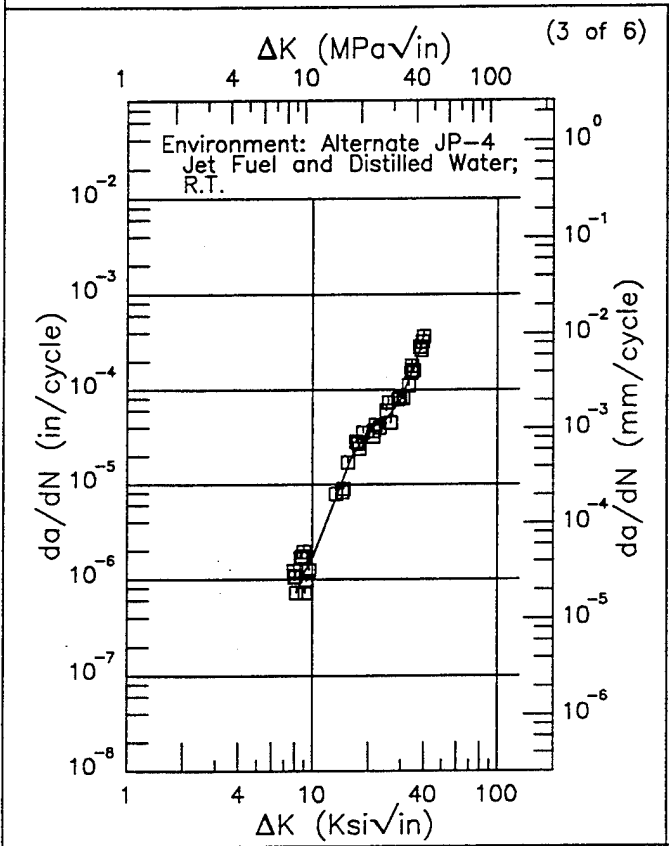


Figure 6.16.3.1.114

Ti-6Al-4V E

Condition/Ht: RA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.5
 Frequency: 1 Hz

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.94 (min)	0.985
8.	0.986
9.	1.17
10.	1.66
13.	6.23
16.	18.4
20.	37.8
25.	49.5
30.	77.9
35.	168.
39.99 (max)	349.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.08 (min)	0.475
9.	1.29
10.	2.77
13.	9.43
16.	16.6
20.	28.0
25.	49.3
30.	82.5
35.	147.
40.	372.
44.64 (max)	3246.

RMS % Error 23.26	Life Prediction Ratio Summary
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RMS % Error 29.23	Life Prediction Ratio Summary
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Figure 6.16.3.1.114 (Continued)

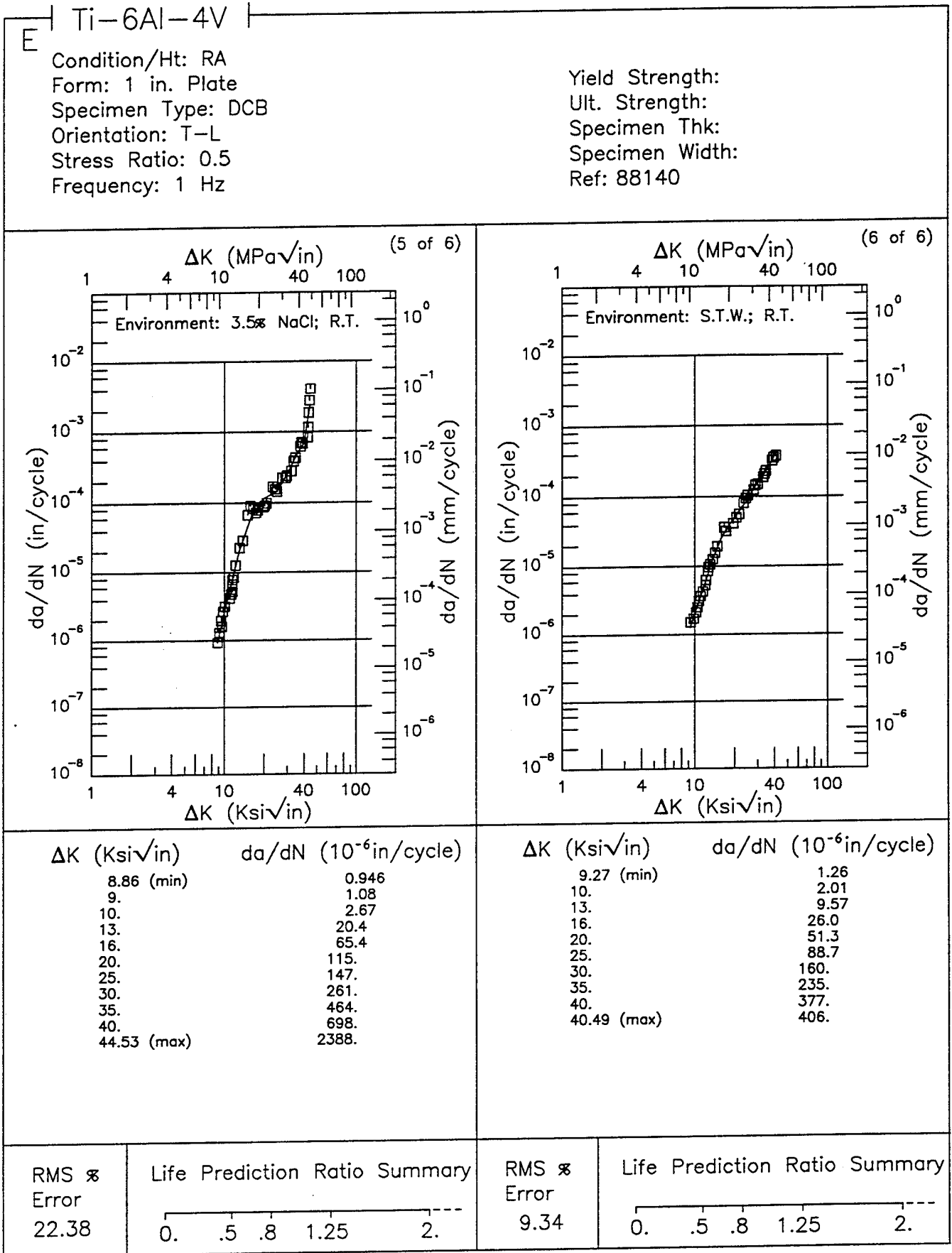


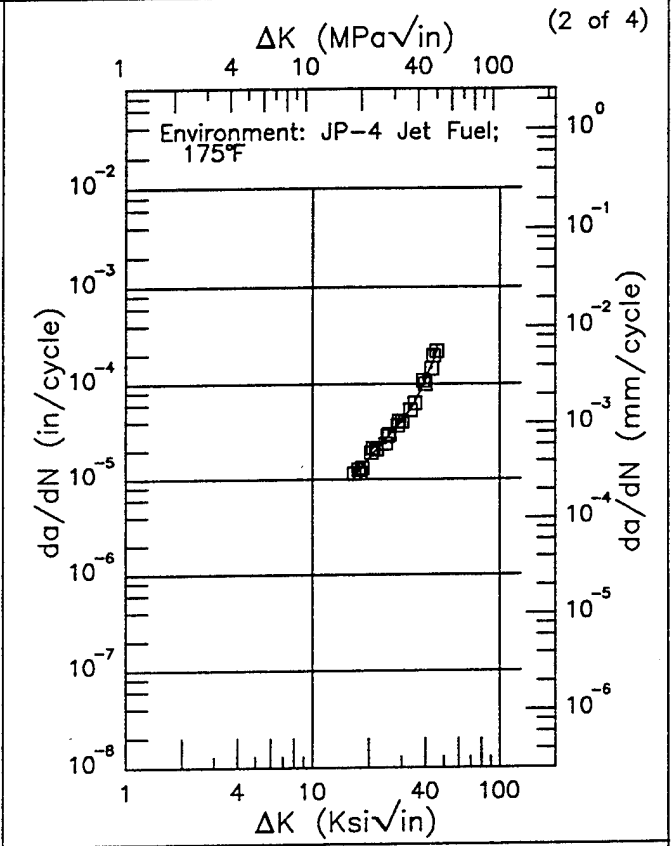
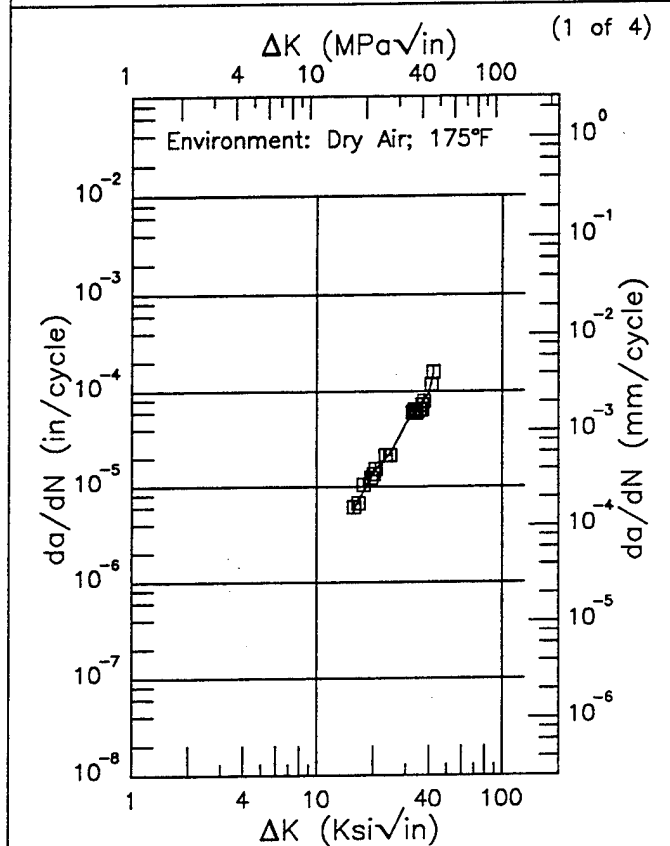
Figure 6.16.3.1.114 (Concluded)

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E | Ti-6Al-4V |

Condition/Ht: RA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.5
 Frequency: 1 Hz

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
15.65 (min)	5.45
16.	6.24
20.	13.9
25.	23.6
30.	46.1
35.	66.0
40.	97.7
42.30 (max)	156.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.60 (min)	10.7
20.	16.7
25.	27.4
30.	40.7
35.	63.2
40.	110.
45.72 (max)	215.

RMS % Error	Life Prediction Ratio Summary
7.82	0. .5 .8 1.25 2.---

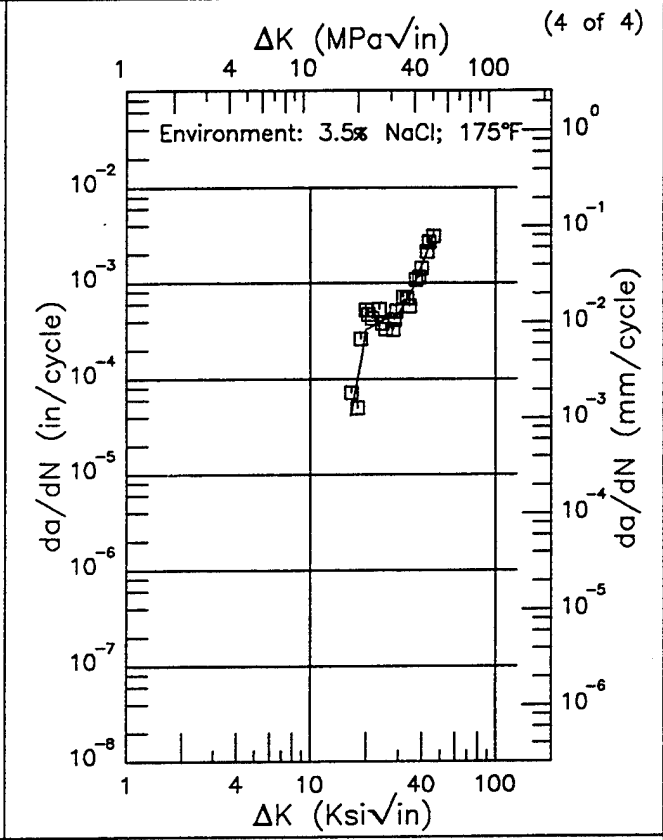
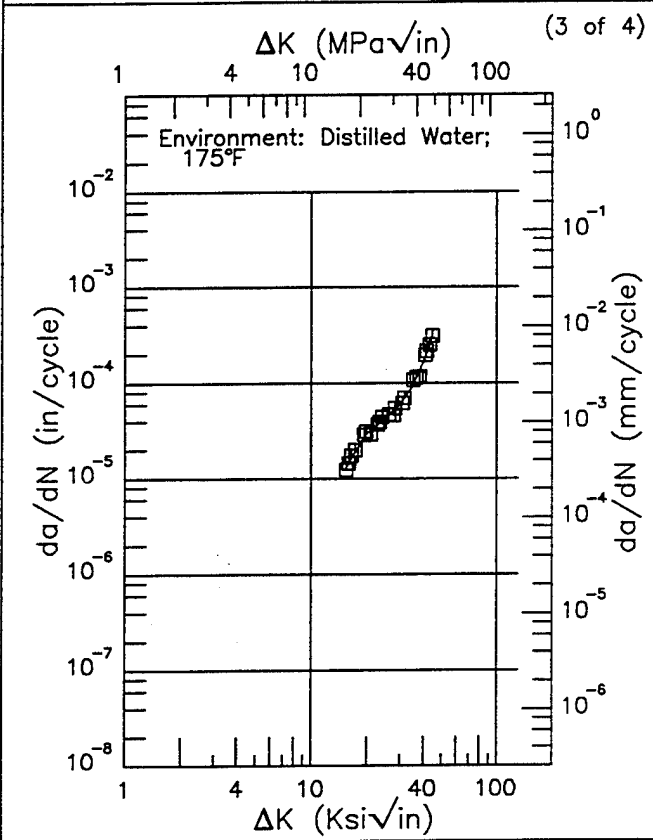
RMS % Error	Life Prediction Ratio Summary
6.58	0. .5 .8 1.25 2.---

Figure 6.16.3.1.115

Ti-6Al-4V E

Condition/Ht: RA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.5
 Frequency: 1 Hz

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
15.44 (min)	12.7
16.	15.0
20.	29.8
25.	43.2
30.	60.3
35.	94.7
40.	168.
44.48 (max)	304.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.62 (min)	41.8
20.	326.
25.	416.
30.	474.
35.	786.
40.	1507.
46.18 (max)	3196.

RMS % Error 6.23	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
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RMS % Error 33.19	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

Figure 6.16.3.1.115 (Concluded)

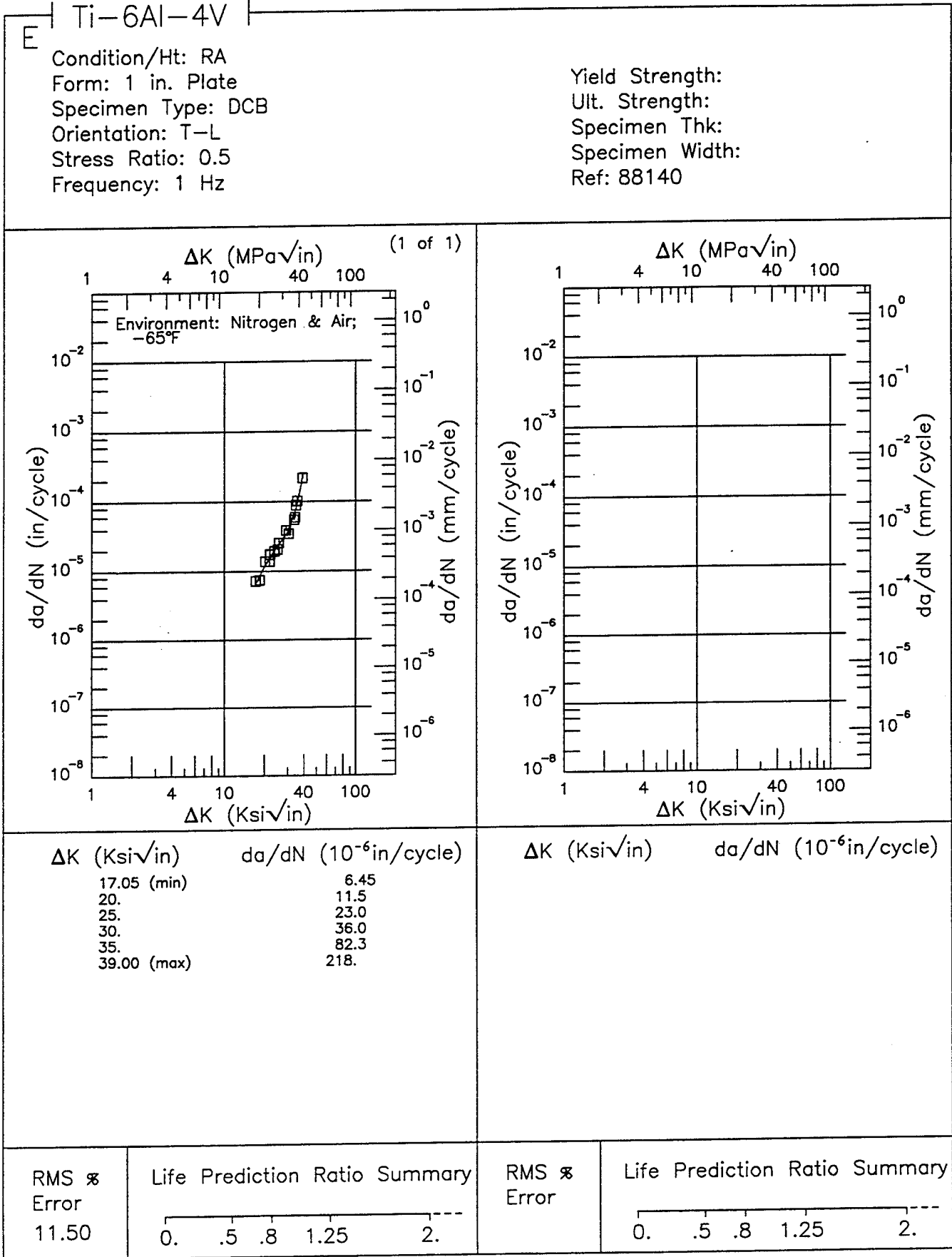


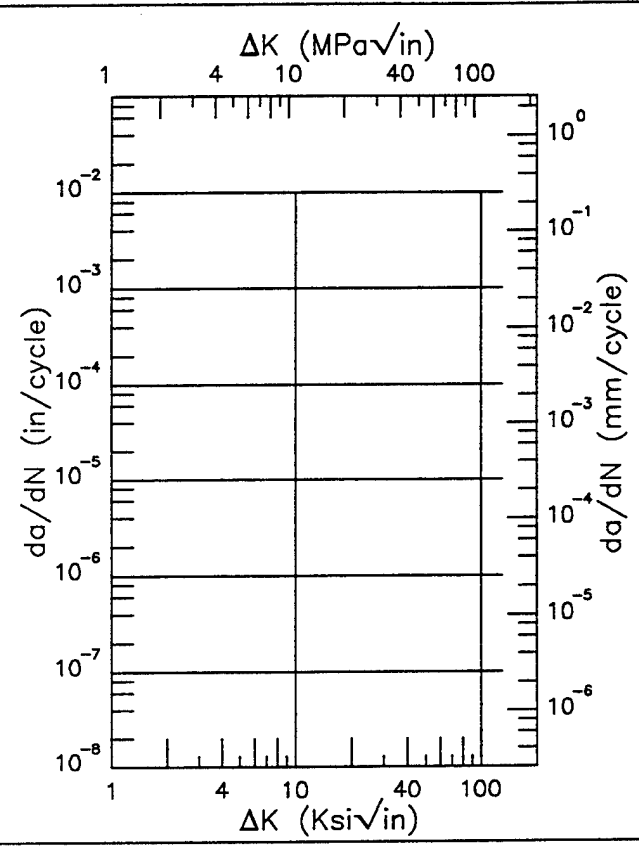
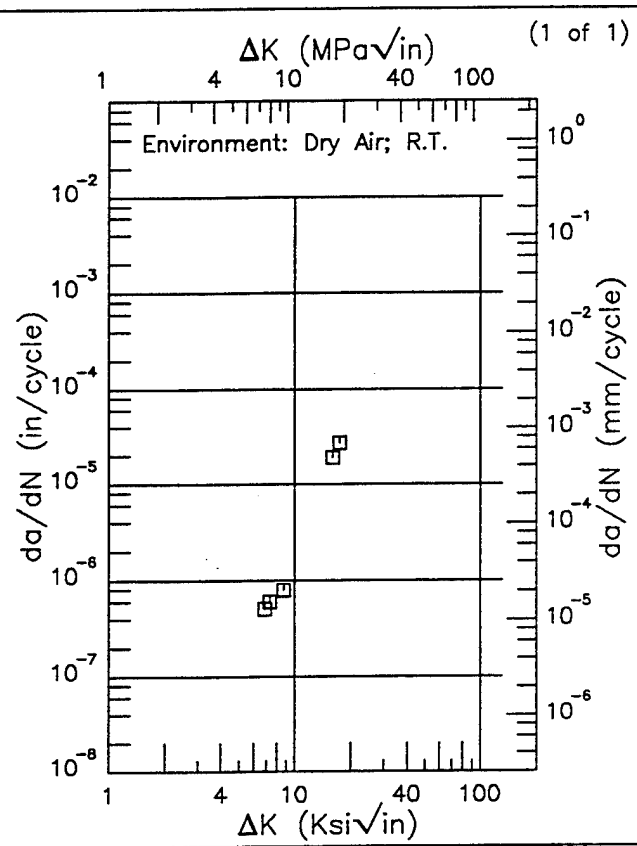
Figure 6.16.3.1.116

Ti-6Al-4V

E

Condition/Ht: RA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.8
 Frequency: 1 Hz

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140

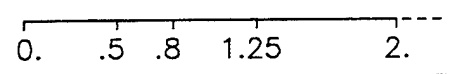


ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS % Error

Life Prediction Ratio Summary



RMS % Error

Life Prediction Ratio Summary

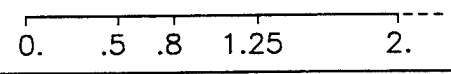


Figure 6.16.3.1.117

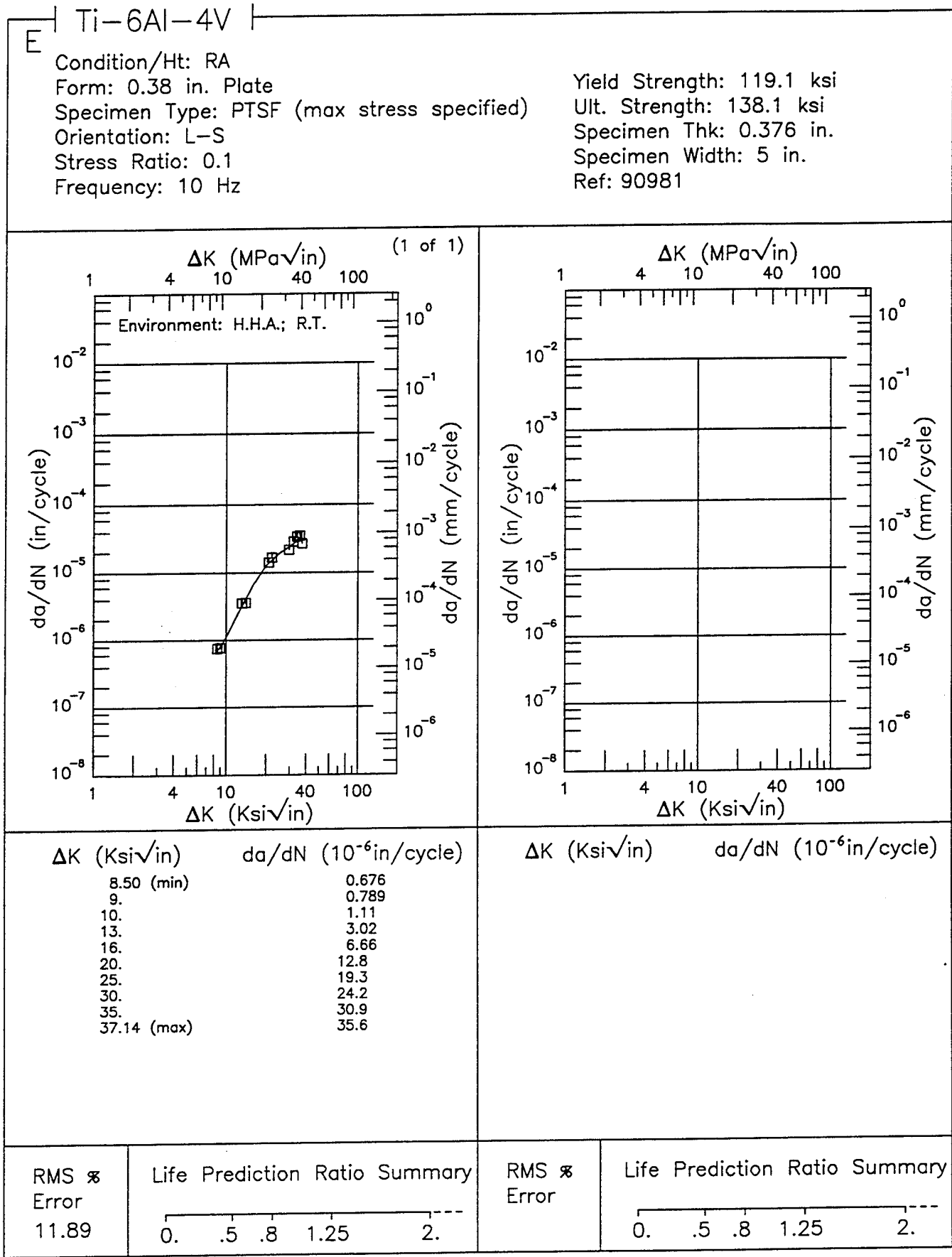


Figure 6.16.3.1.118

Ti-6Al-4V E

Condition/Ht: RA
 Form: 0.38 in. Plate
 Specimen Type: PTSF (max stress specified)
 Orientation: T-S
 Stress Ratio: 0.1
 Frequency: 10 Hz

Yield Strength: 141.7 ksi
 Ult. Strength: 151.8 ksi
 Specimen Thk: 0.377 in.
 Specimen Width: 5 in.
 Ref: 90981

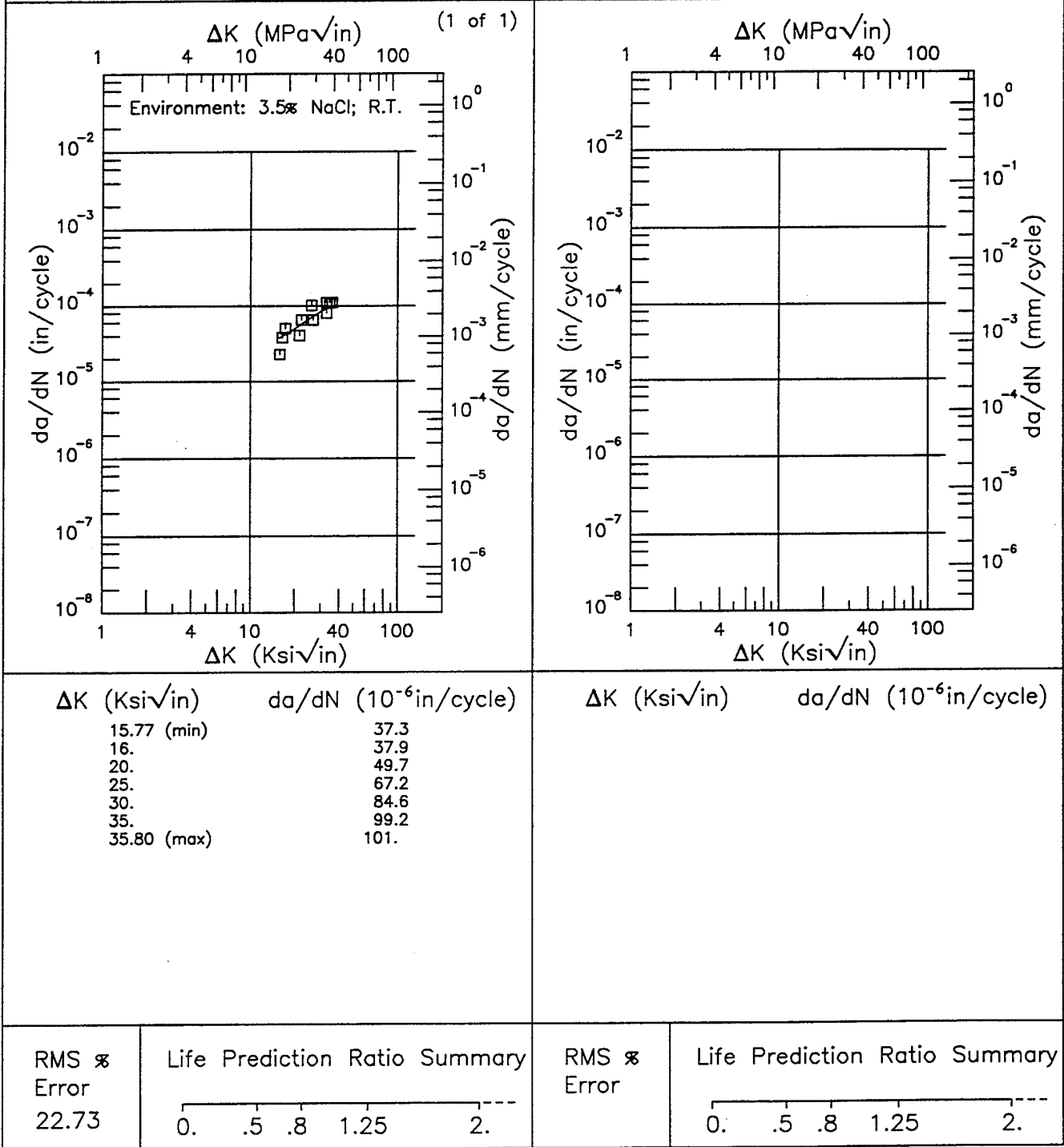


Figure 6.16.3.1.119

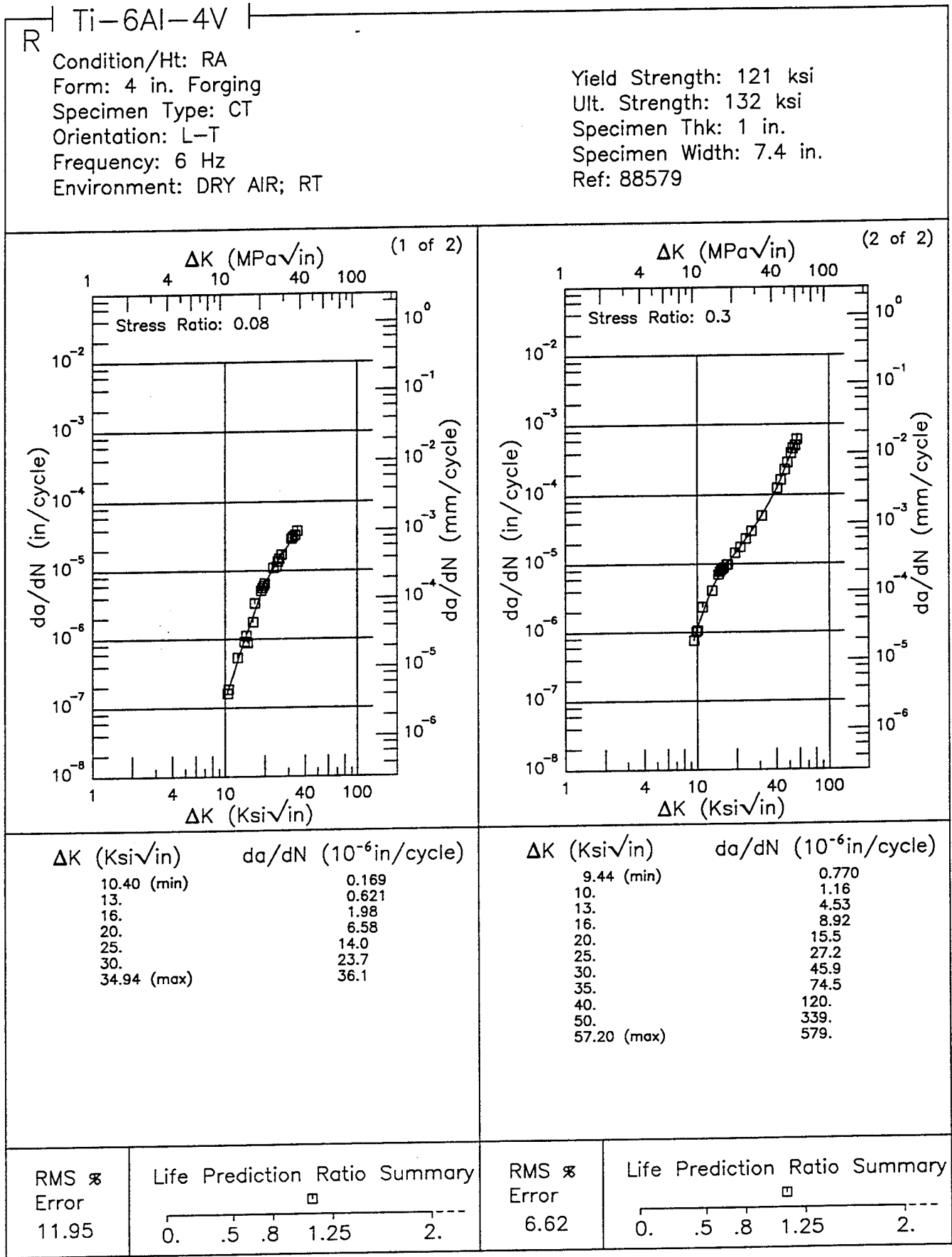


Figure 6.16.3.1.120

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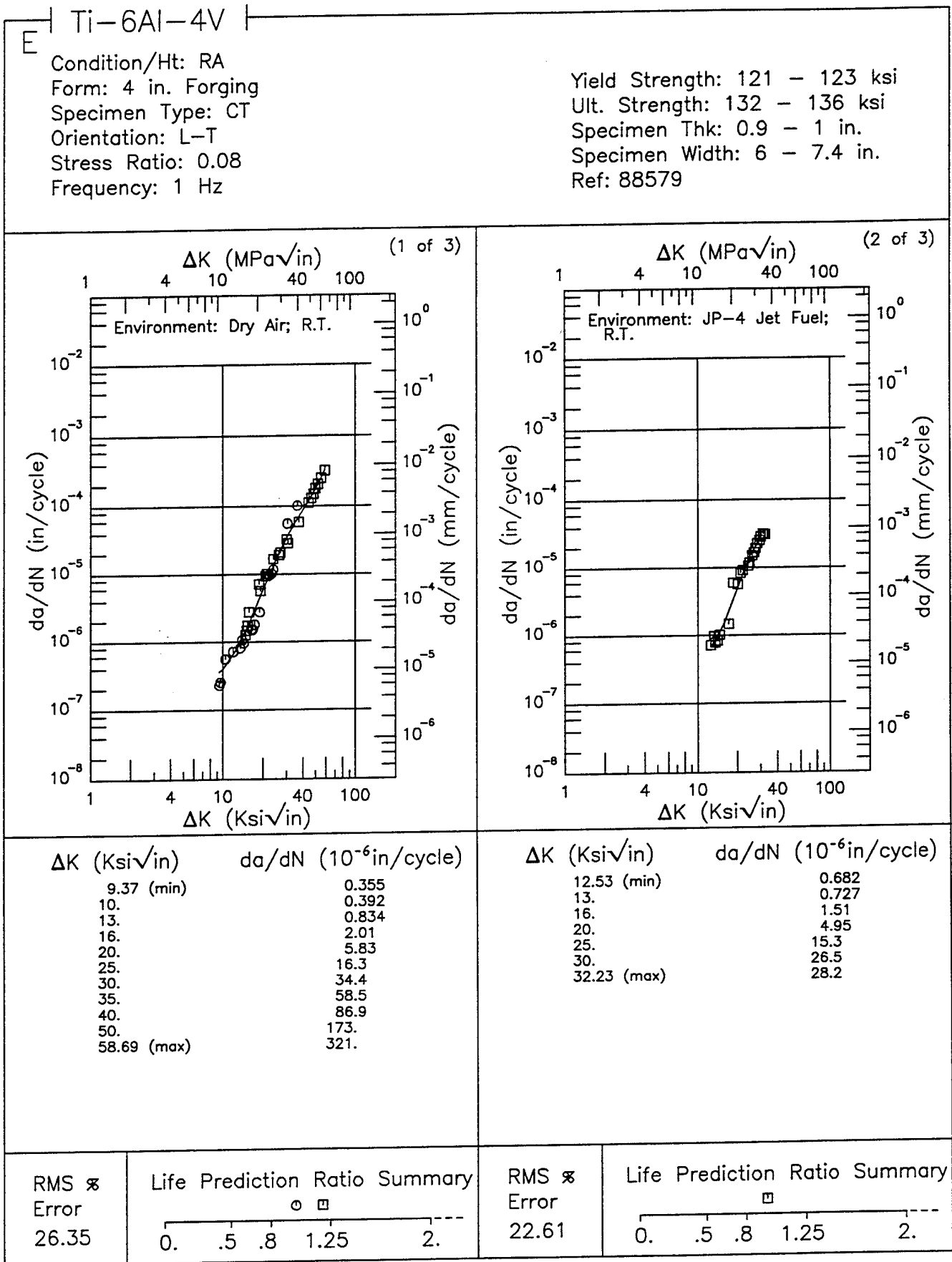


Figure 6.16.3.1.121

Ti-6Al-4V E

Condition/Ht: RA
 Form: 4 in. Forging
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.08
 Frequency: 1 Hz

Yield Strength: 121 - 123 ksi
 Ult. Strength: 132 - 136 ksi
 Specimen Thk: 0.9 - 1 in.
 Specimen Width: 6 - 7.4 in.
 Ref: 88579

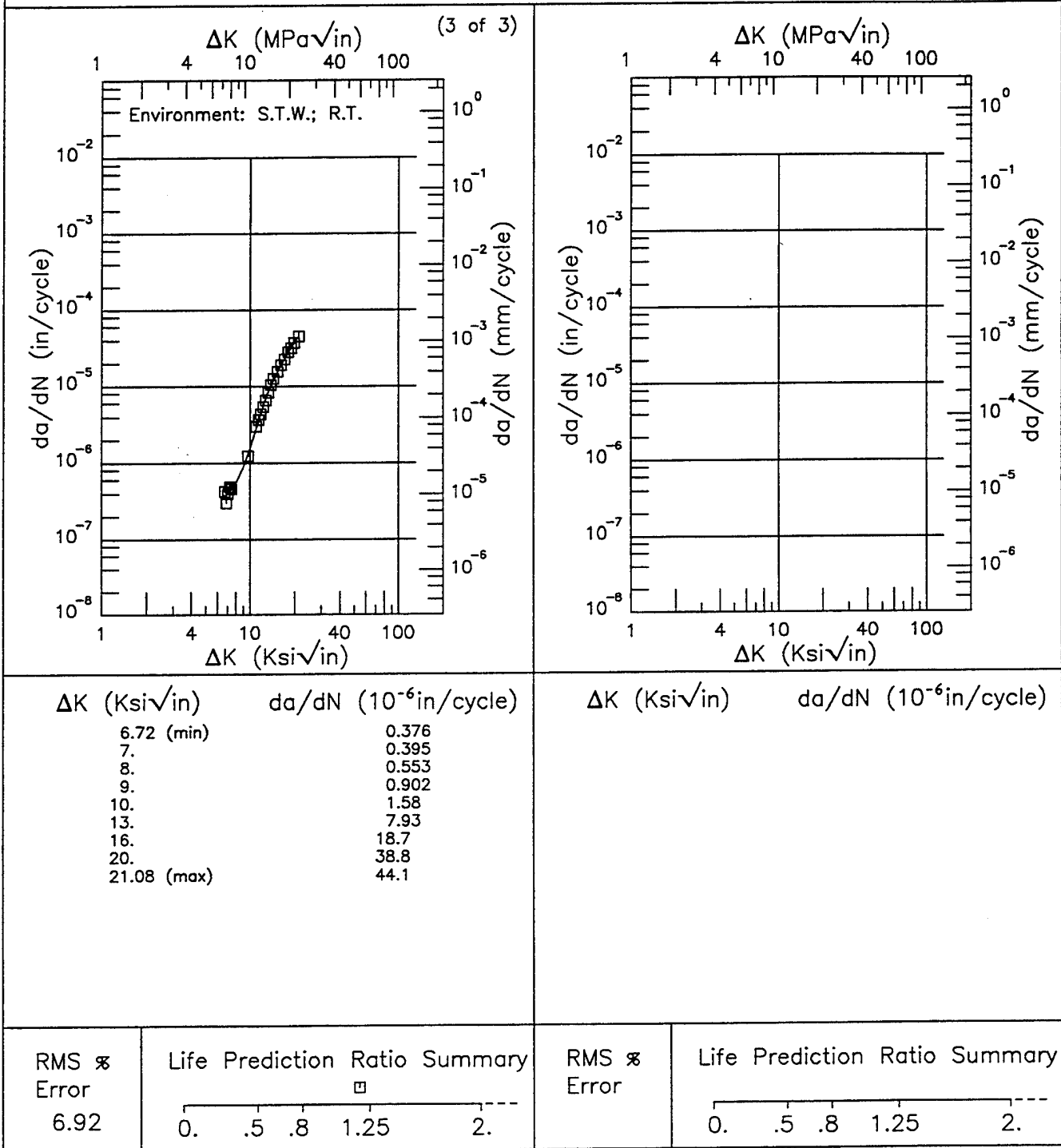


Figure 6.16.3.1.121 (Concluded)

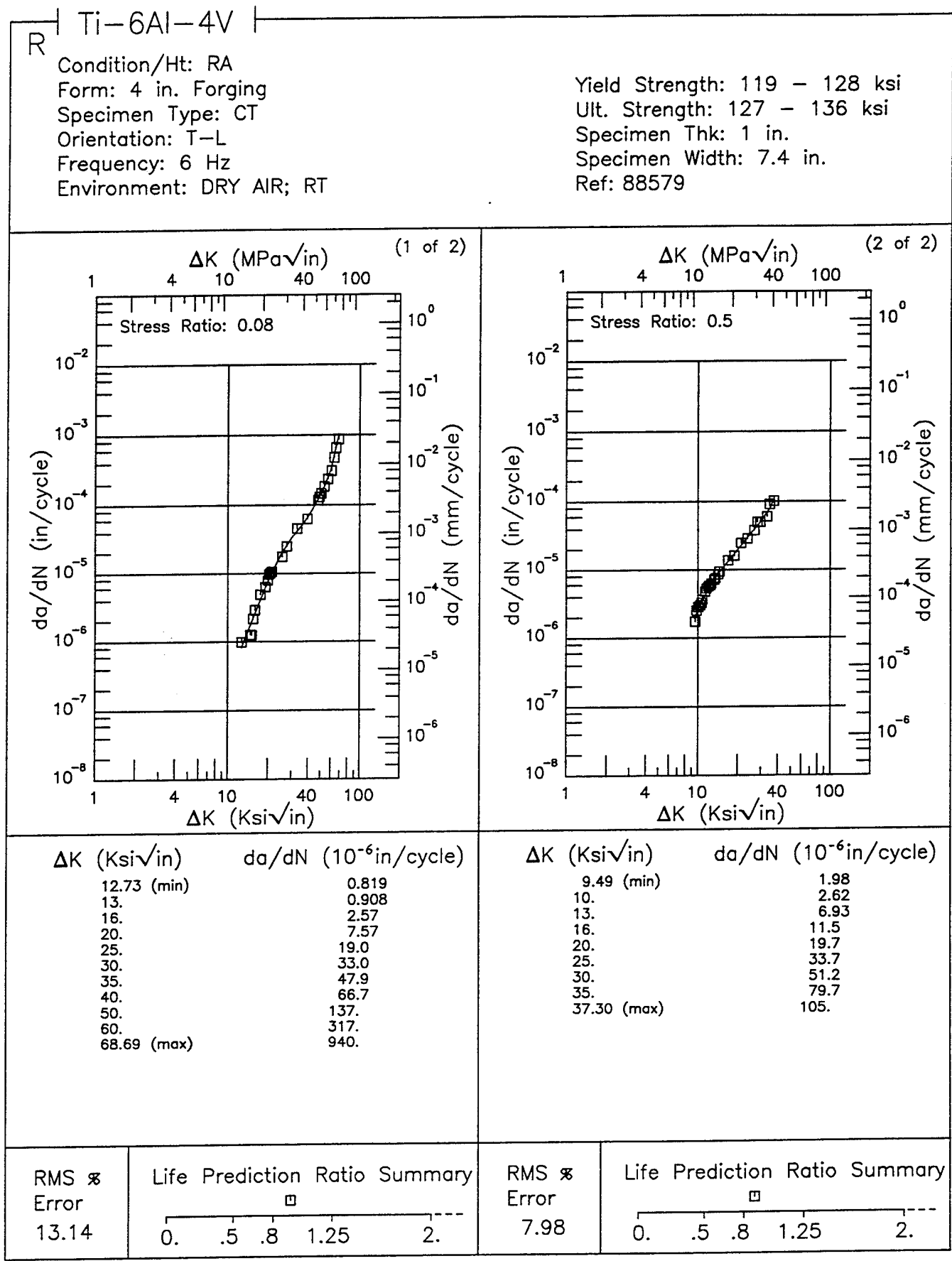


Figure 6.16.3.1.122

Ti-6Al-4V

R

Condition/Ht: RA
 Form: 4 in. Forging
 Specimen Type: CT
 Orientation: T-L
 Frequency: 1 Hz
 Environment: S.T.W.; RT

Yield Strength: 128 ksi
 Ult. Strength: 136 ksi
 Specimen Thk: 0.66 - 0.77 in.
 Specimen Width: 7.4 in.
 Ref: 88579

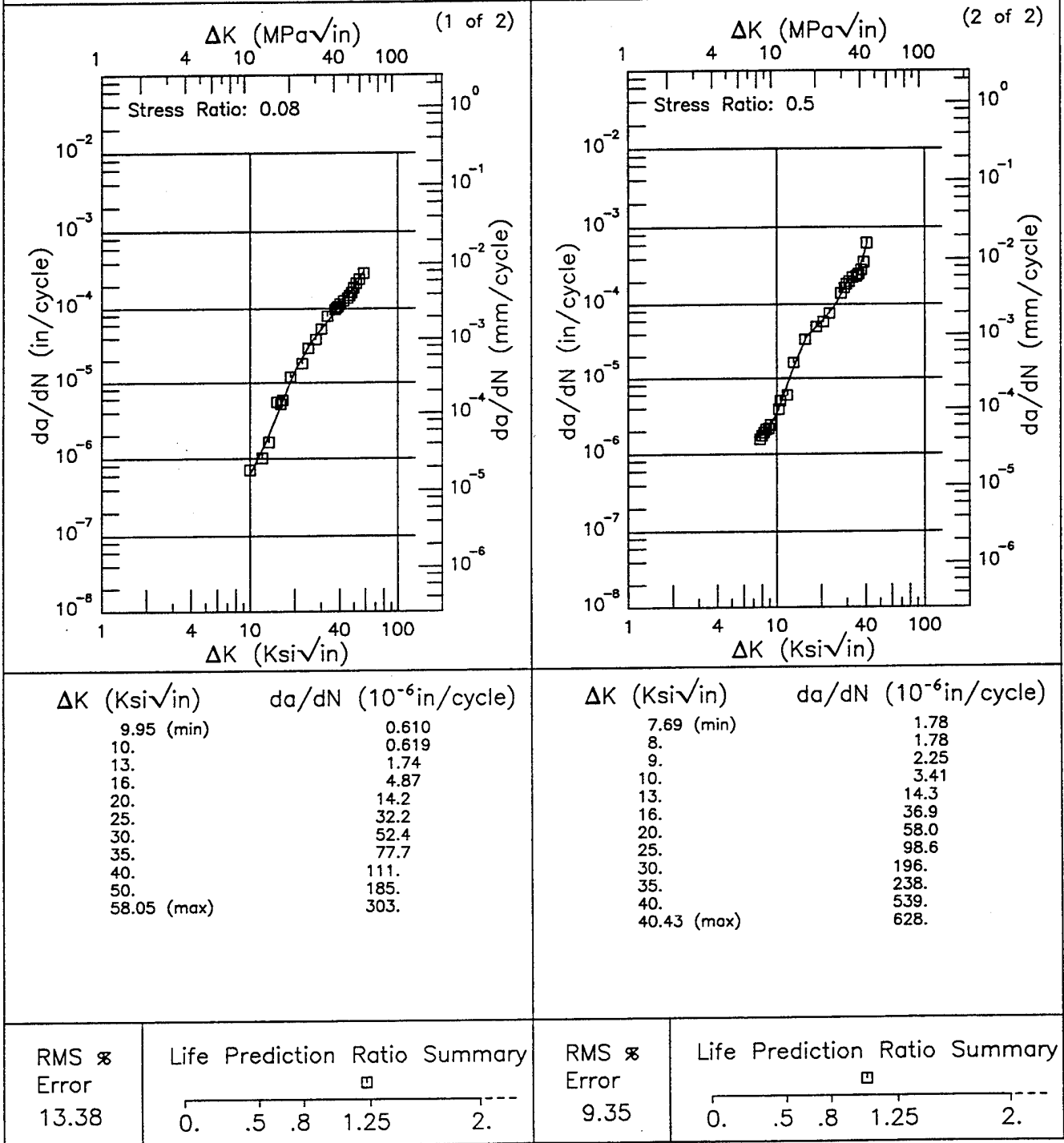


Figure 6.16.3.1.123

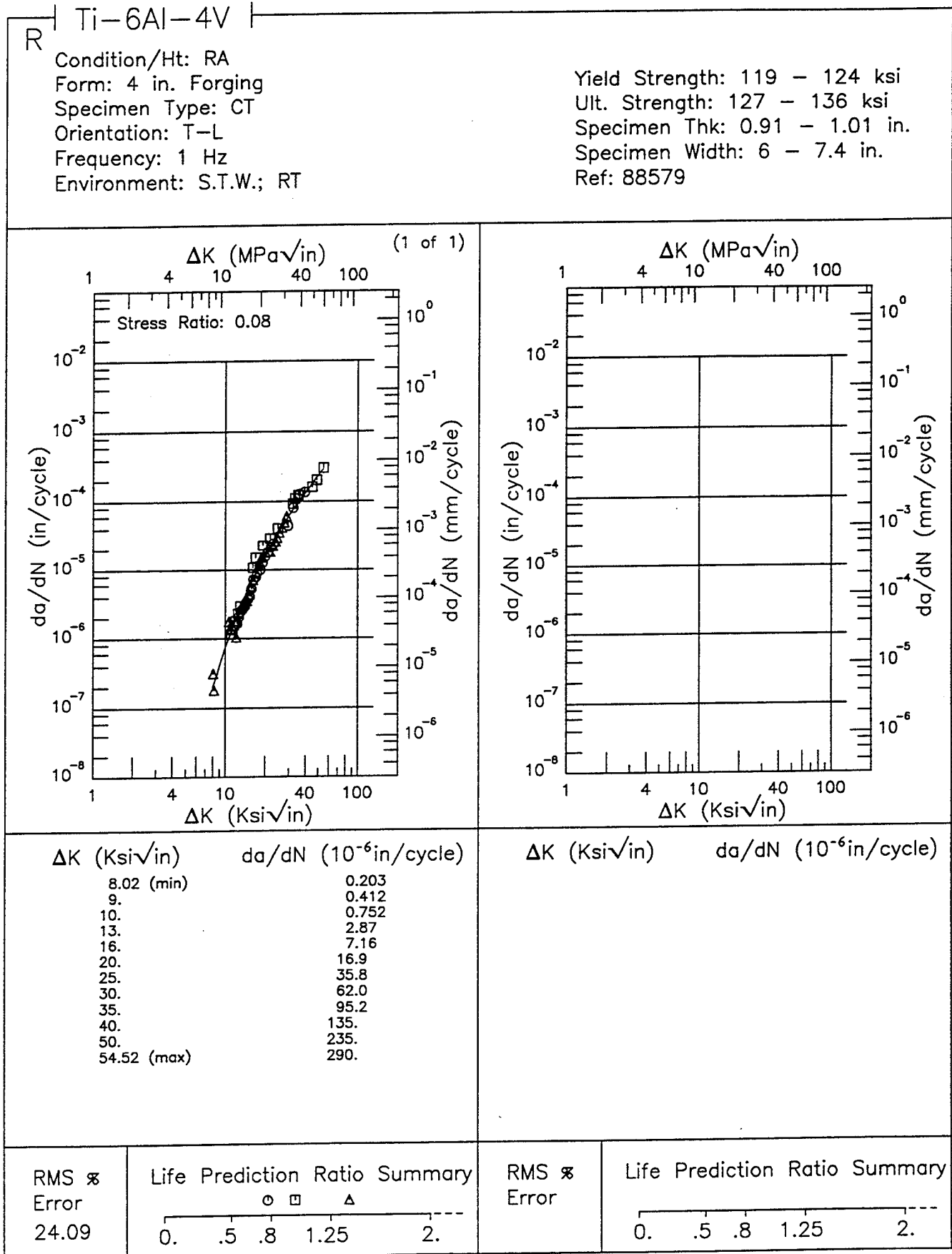


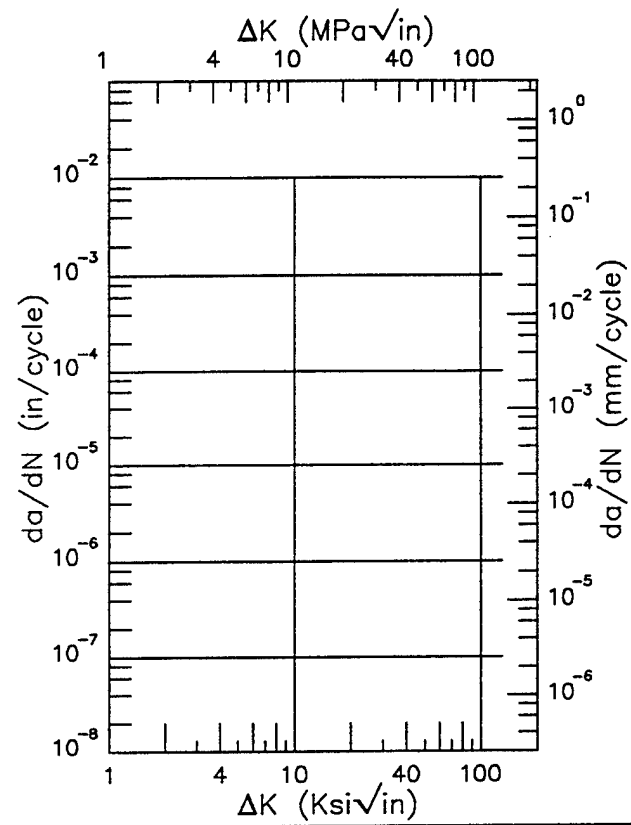
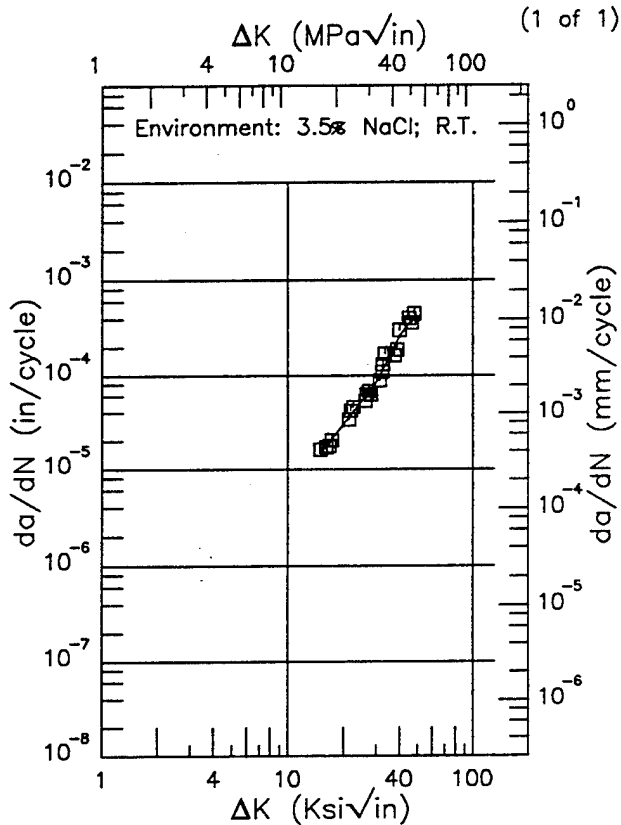
Figure 6.16.3.1.124

Ti-6Al-4V

E

Condition/Ht: RA(FAST COOLED)
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.1
 Frequency: 1 Hz

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
14.96 (min)	15.2
16.	17.2
20.	29.7
25.	52.4
30.	84.0
35.	151.
40.	265.
47.90 (max)	420.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
--------------------------------------	-------------------------------

RMS % Error	Life Prediction Ratio Summary
13.11	

RMS % Error	Life Prediction Ratio Summary

Figure 6.16.3.1.125

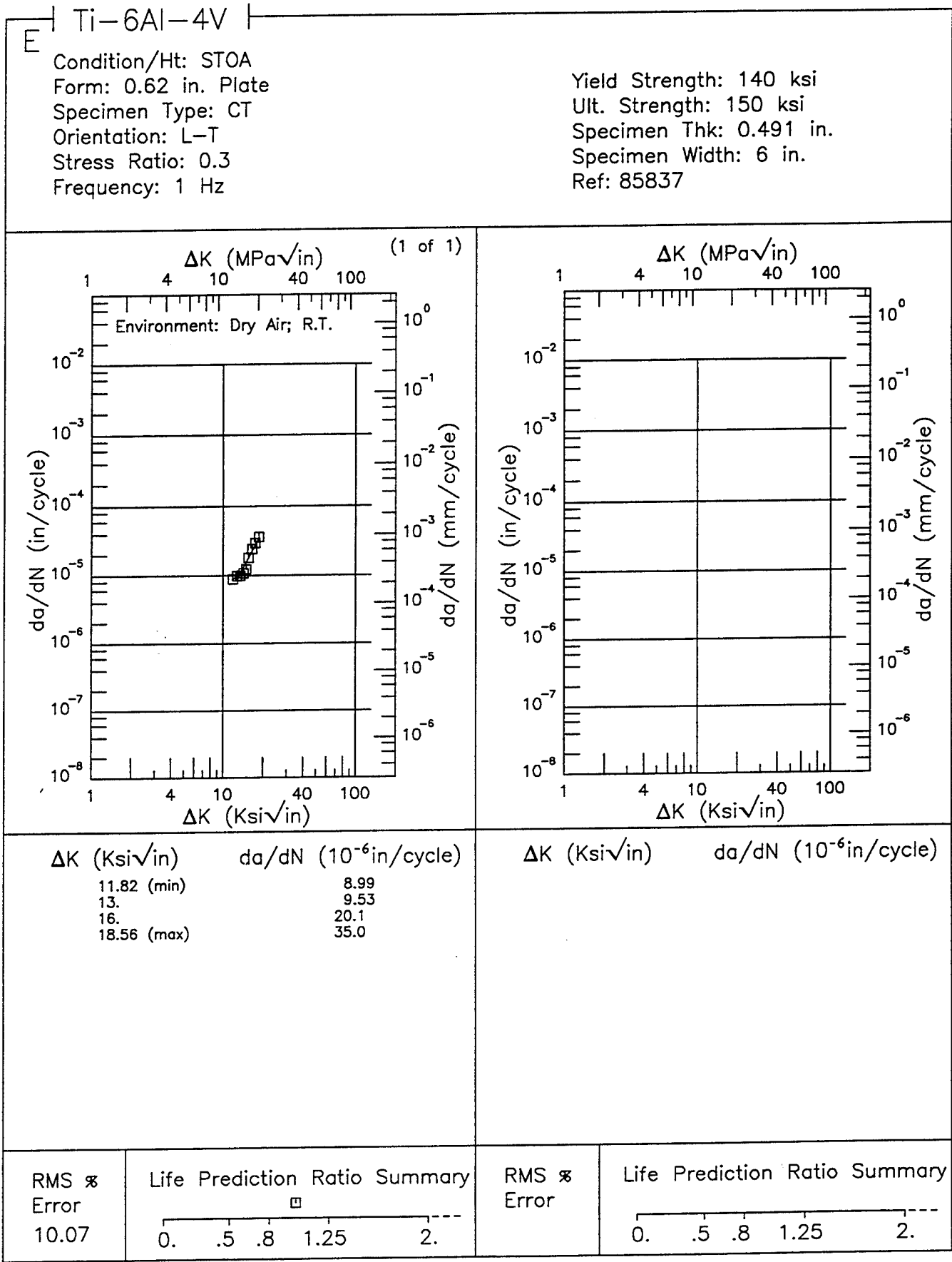
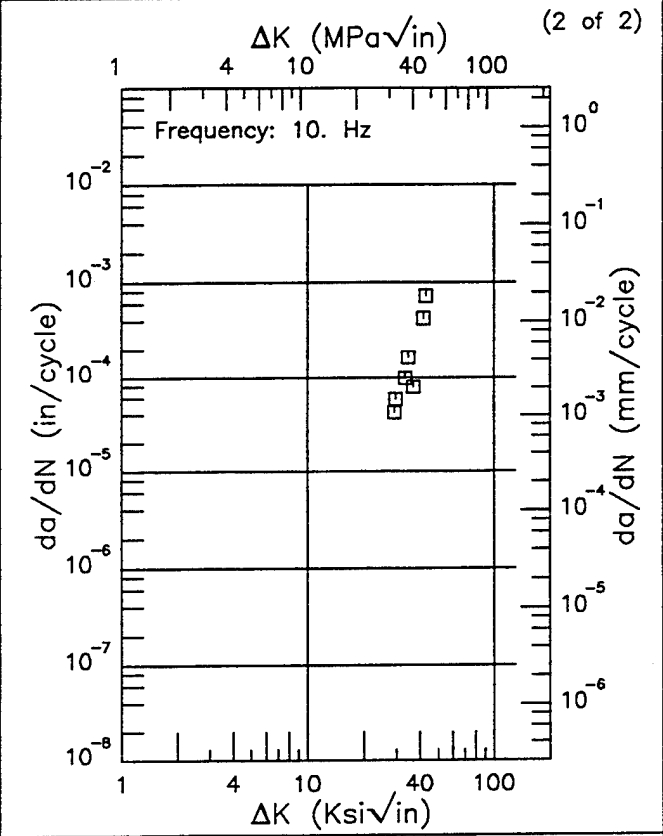
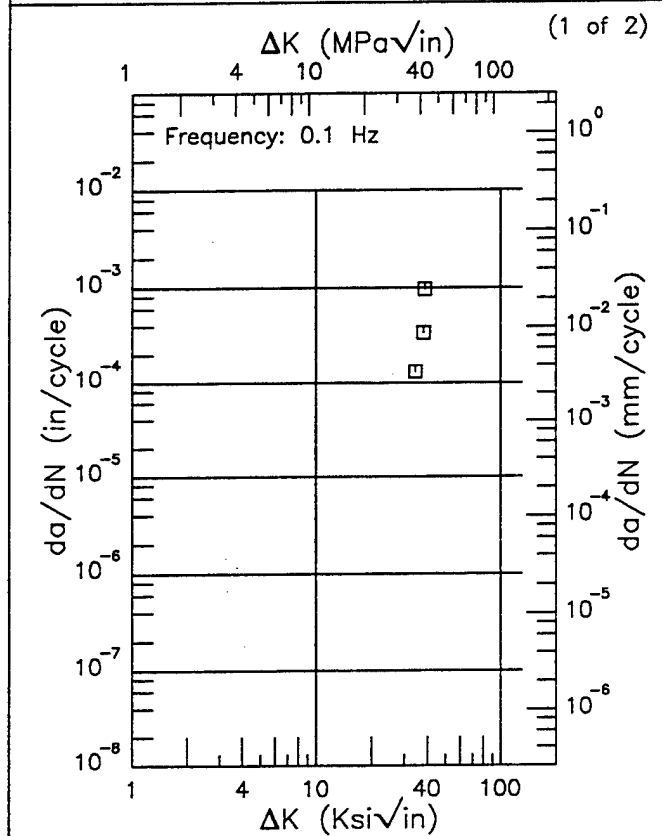


Figure 6.16.3.1.126

Ti-6Al-4V F

Condition/Ht: STRESS RELIEVED E.B. WELDMENT (HAZ)
 Form: Weldment
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1
 Environment: LAB AIR; -65°F

Yield Strength:
 Ult. Strength:
 Specimen Thk: 1 in.
 Specimen Width: 2.55 in.
 Ref: 88144



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
---------------------	-------------------------------

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
---------------------	-------------------------------

RMS % Error	Life Prediction Ratio Summary
	0. .5 .8 1.25 2.

RMS % Error	Life Prediction Ratio Summary
	0. .5 .8 1.25 2.

Figure 6.16.3.1.127

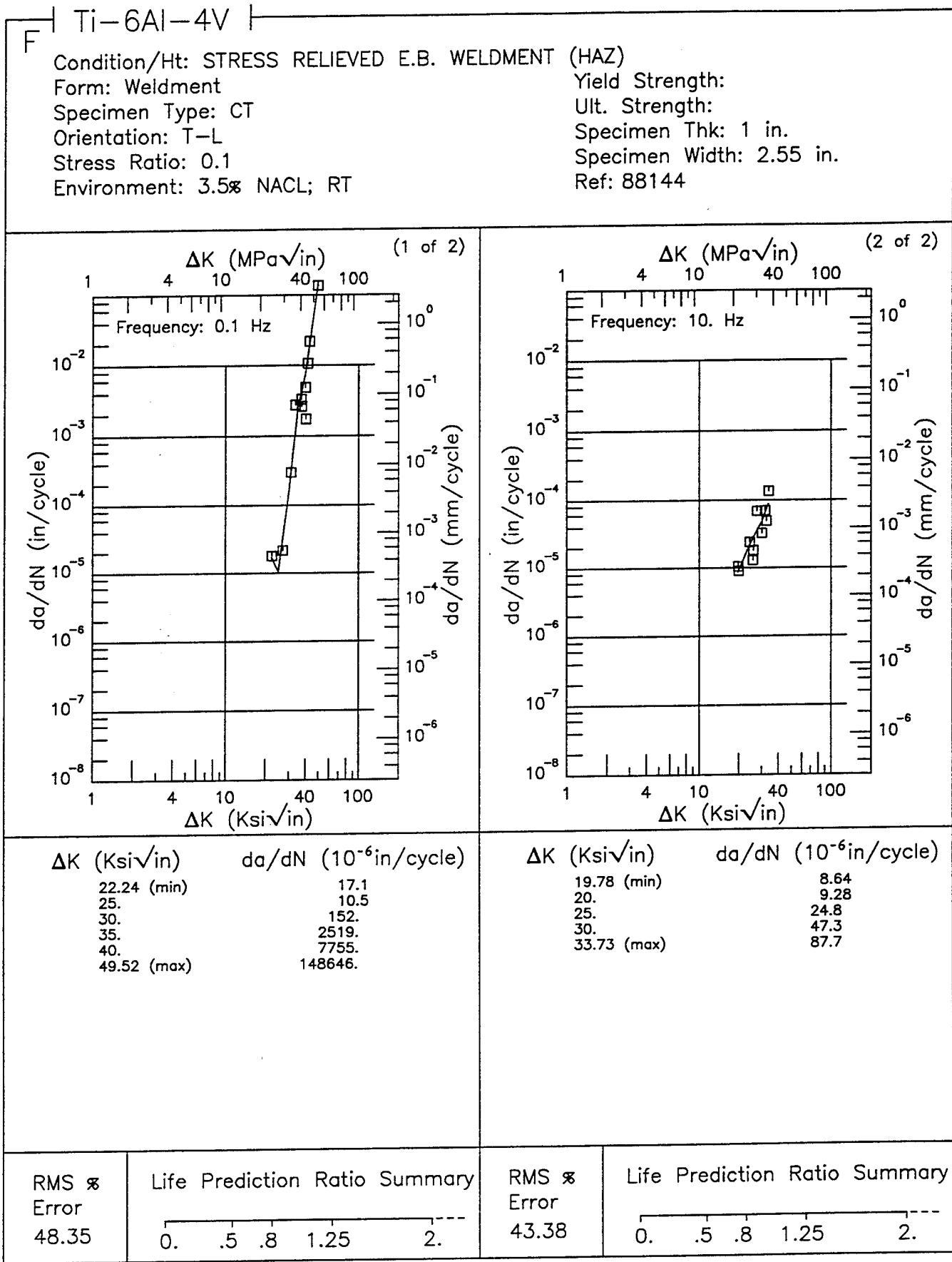


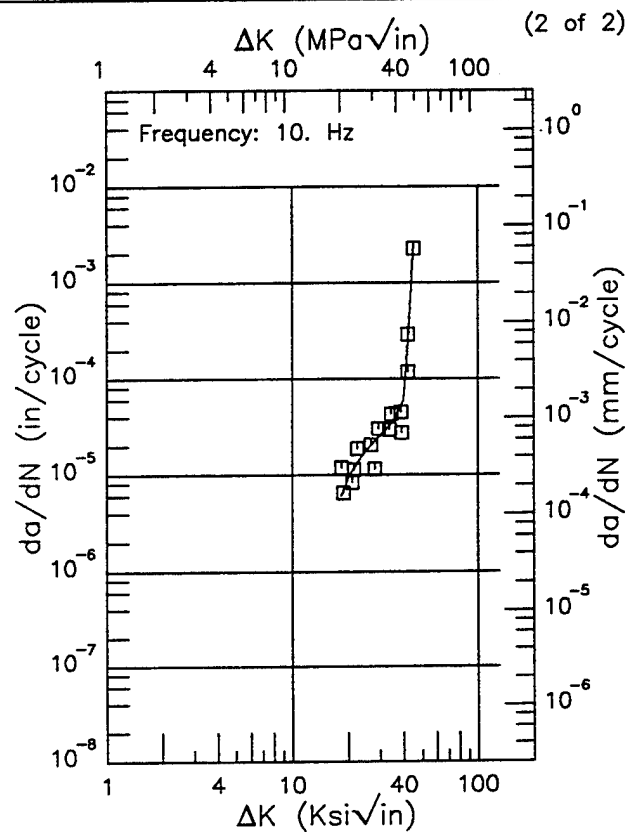
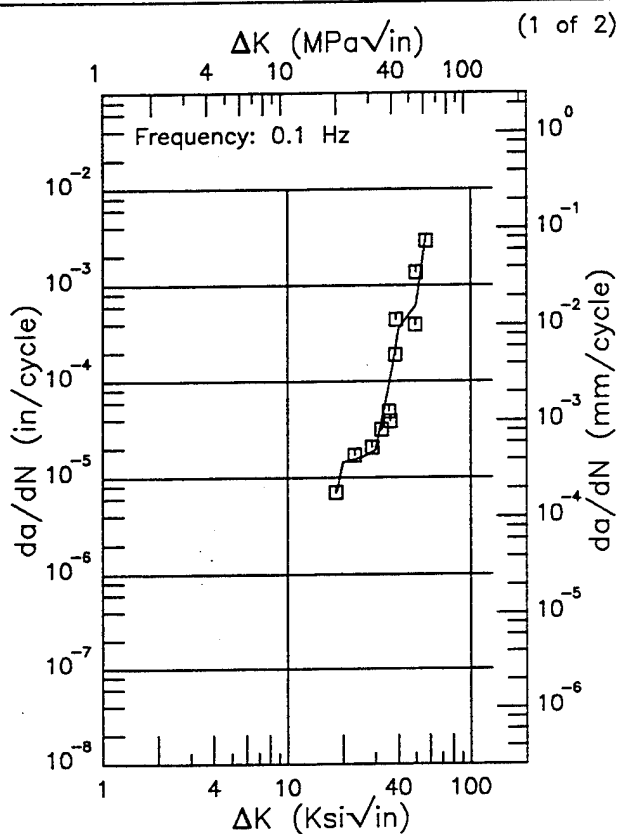
Figure 6.16.3.1.128

Ti-6Al-4V

F

Condition/Ht: STRESS RELIEVED E.B. WELDMENT (HAZ)
 Form: Weldment
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1
 Environment: JP4; RT

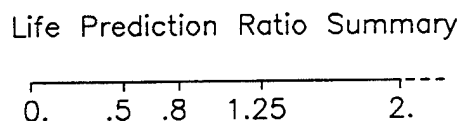
Yield Strength:
 Ult. Strength:
 Specimen Thk: 1 in.
 Specimen Width: 2.55 in.
 Ref: 88144



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
18.23 (min)	6.87
20.	14.6
25.	16.1
30.	19.1
35.	80.1
40.	358.
50.	621.
56.28 (max)	2893.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
18.39 (min)	6.05
20.	9.36
25.	17.8
30.	25.7
35.	35.2
40.	60.6
45.20 (max)	2543.

RMS %
 Error
 51.87



RMS %
 Error
 37.44

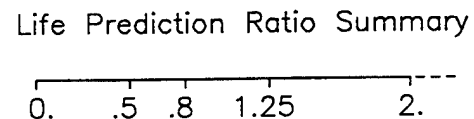


Figure 6.16.3.1.129

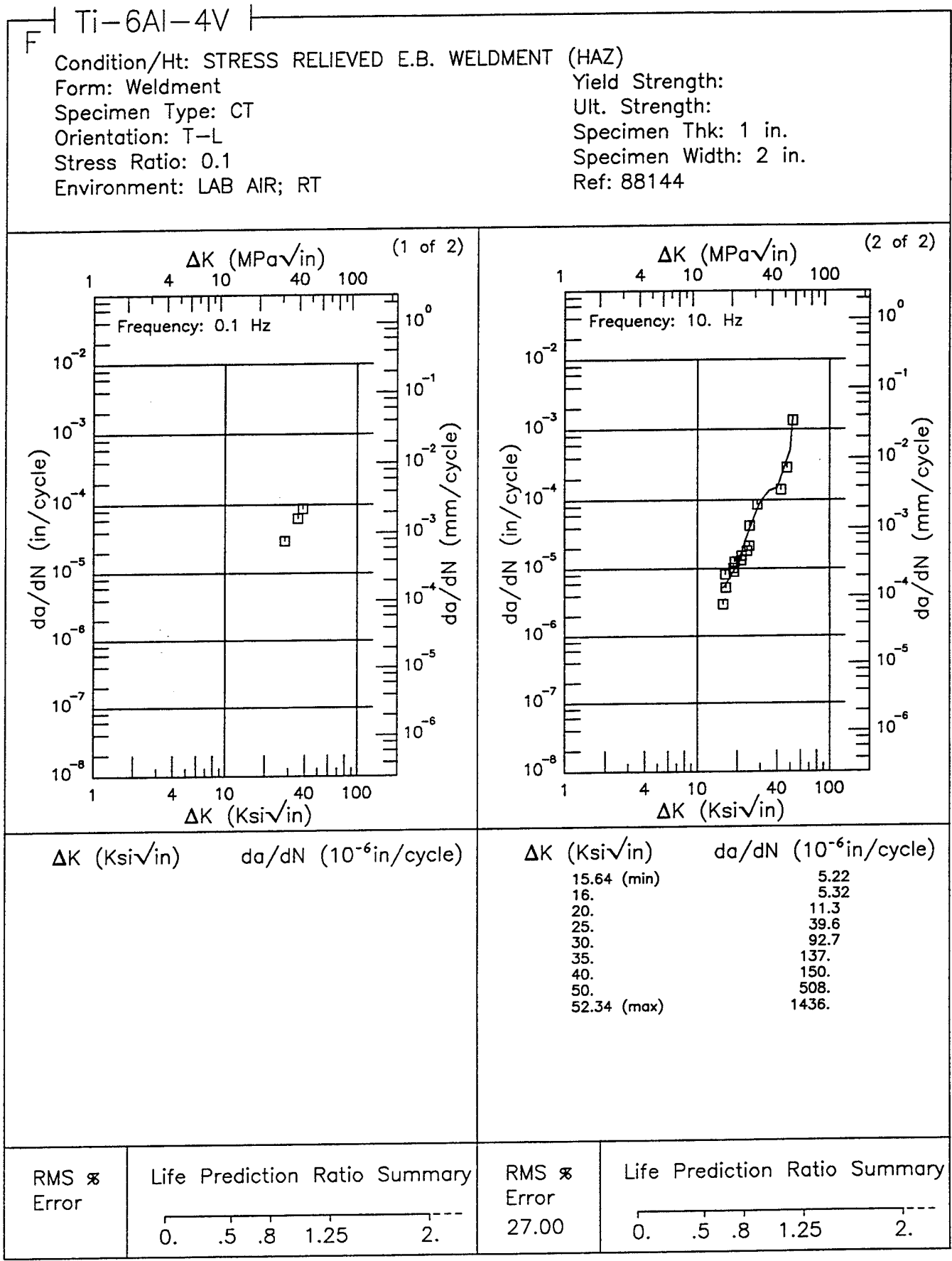


Figure 6.16.3.1.130

Ti-6Al-4V

F

Condition/Ht: STRESS RELIEVED E.B. WELDMENT (HAZ)

Form: Weldment

Specimen Type: CT

Orientation: T-L

Stress Ratio: 0.1

Environment: 3.5% NaCl; 175°F

Yield Strength:

Ult. Strength:

Specimen Thk: 1 in.

Specimen Width: 2.55 in.

Ref: 88144

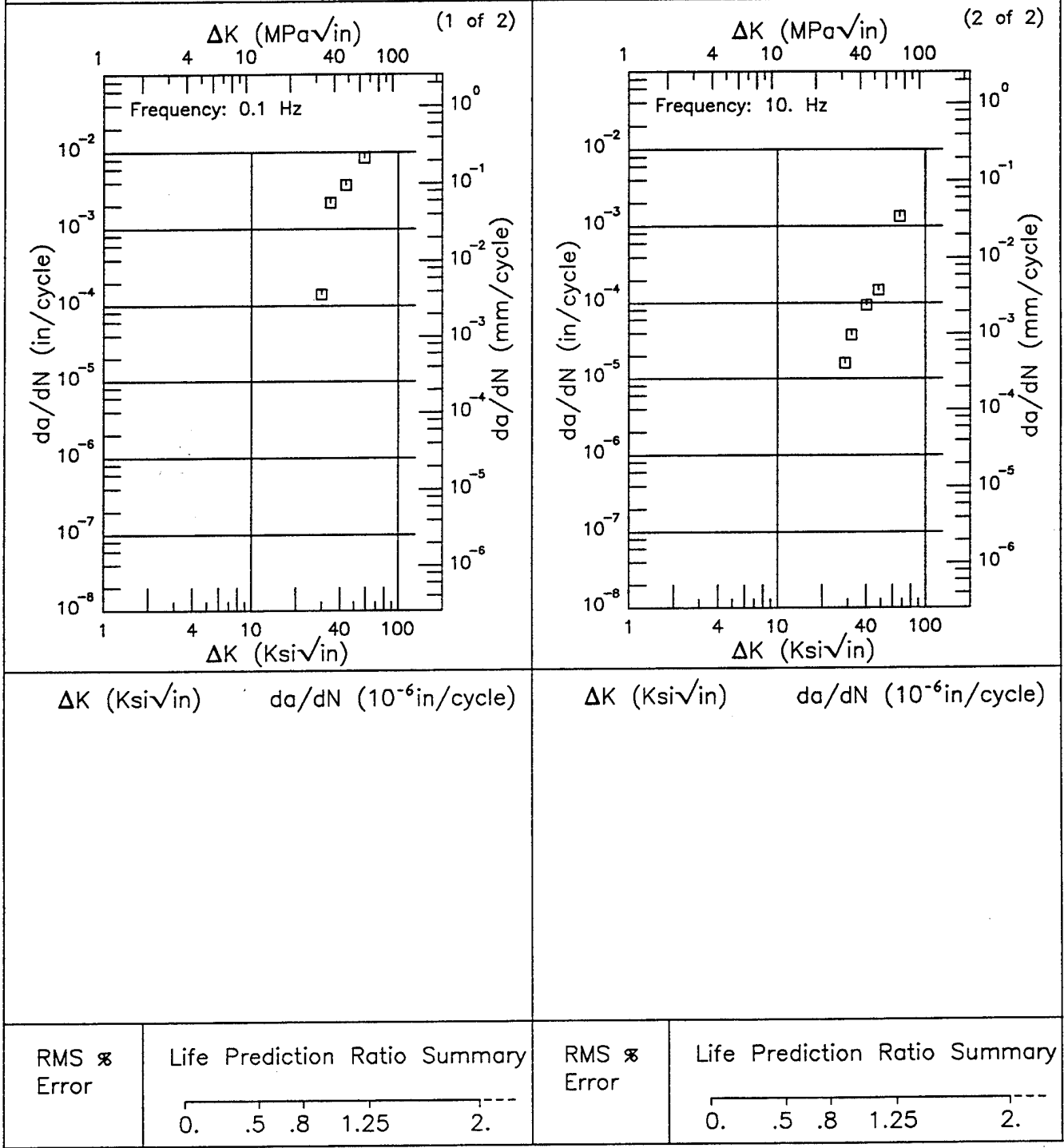


Figure 6.16.3.1.131

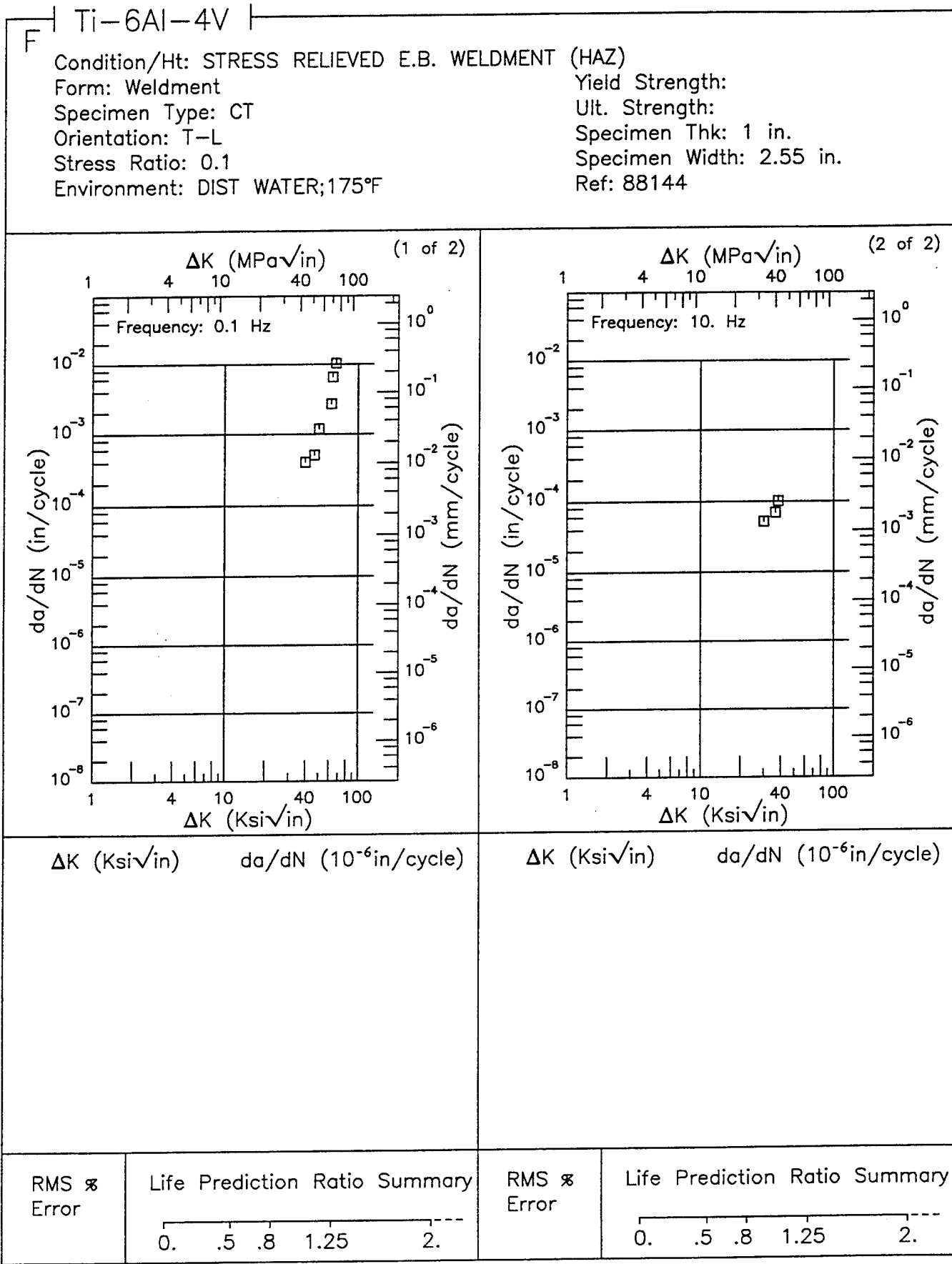
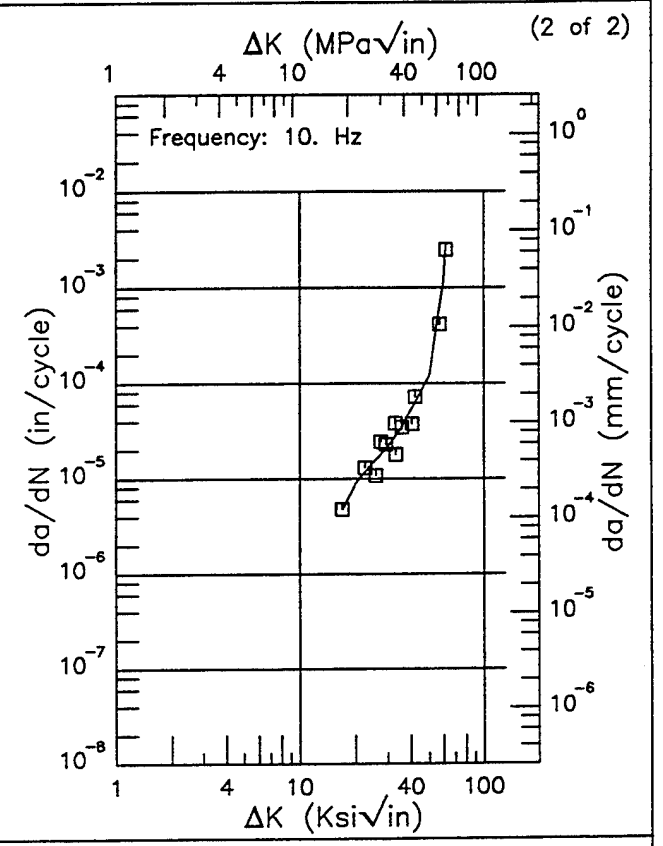
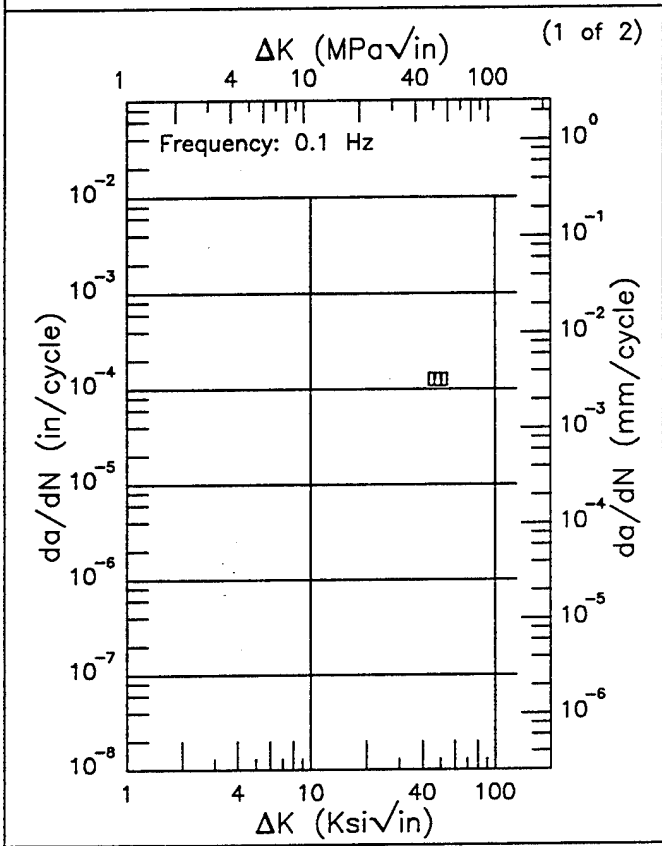


Figure 6.16.3.1.132

Ti-6Al-4V F

Condition/Ht: STRESS RELIEVED E.B. WELDMENT (HAZ)
 Form: Weldment
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1
 Environment: LAB AIR;175°F

Yield Strength:
 Ult. Strength:
 Specimen Thk: 1 in.
 Specimen Width: 2.55 in.
 Ref: 88144



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.93 (min)	4.81
20.	9.03
25.	15.0
30.	21.6
35.	33.1
40.	52.5
50.	119.
60.	1183.
61.62 (max)	2391.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
16.93 (min)	4.81
20.	9.03
25.	15.0
30.	21.6
35.	33.1
40.	52.5
50.	119.
60.	1183.
61.62 (max)	2391.

RMS % Error	Life Prediction Ratio Summary
	0. .5 .8 1.25 2.-----

RMS % Error	Life Prediction Ratio Summary
25.08	0. .5 .8 1.25 2.-----

Figure 6.16.3.1.133

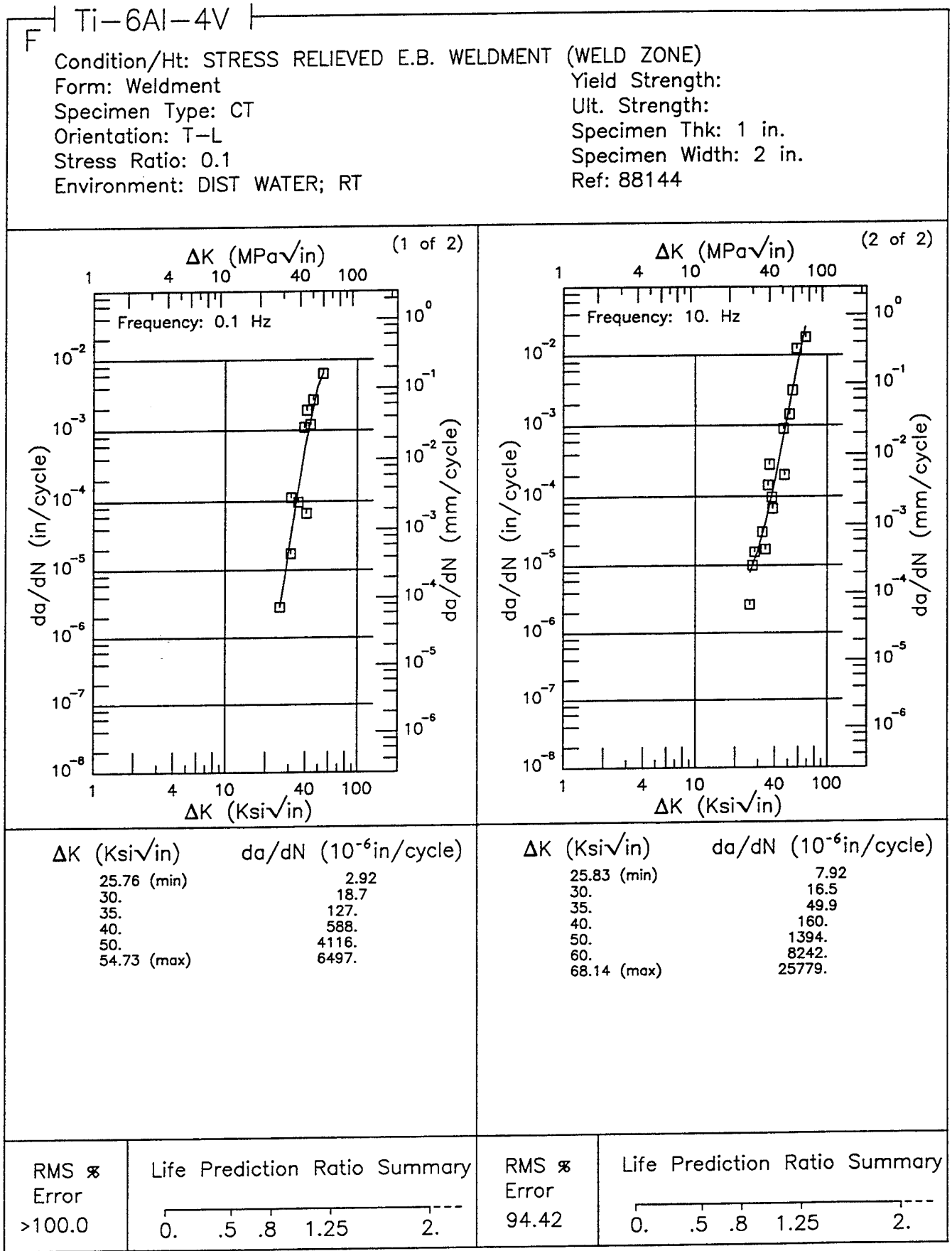


Figure 6.16.3.1.134

Ti-6Al-4V

F

Condition/Ht: STRESS RELIEVED E.B. WELDMENT (WELD ZONE)

Form: Weldment

Specimen Type: CT

Orientation: T-L

Stress Ratio: 0.1

Environment: LAB AIR; RT

Yield Strength:

Ult. Strength:

Specimen Thk: 1 in.

Specimen Width: 2 in.

Ref: 88144

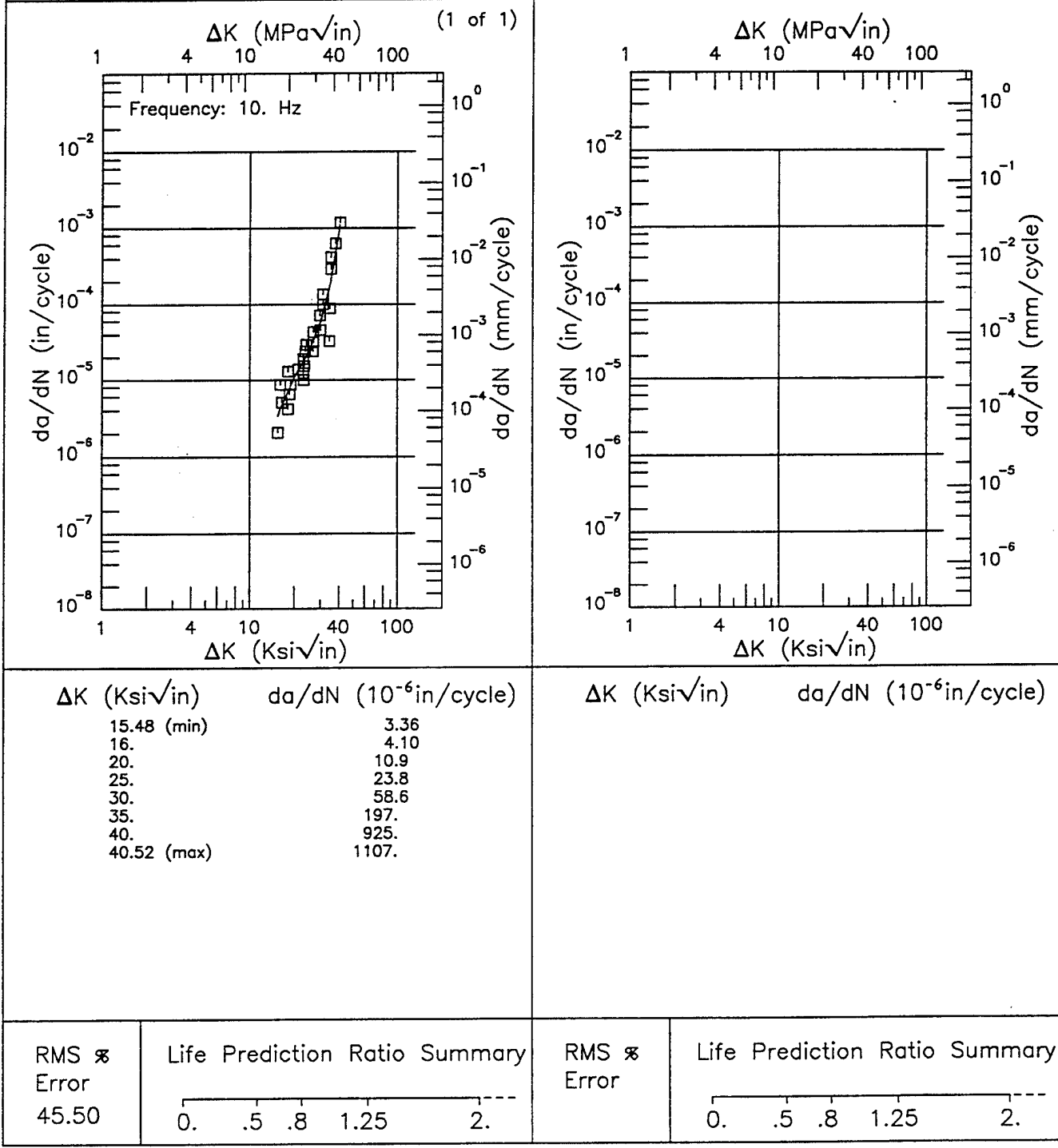


Figure 6.16.3.1.135

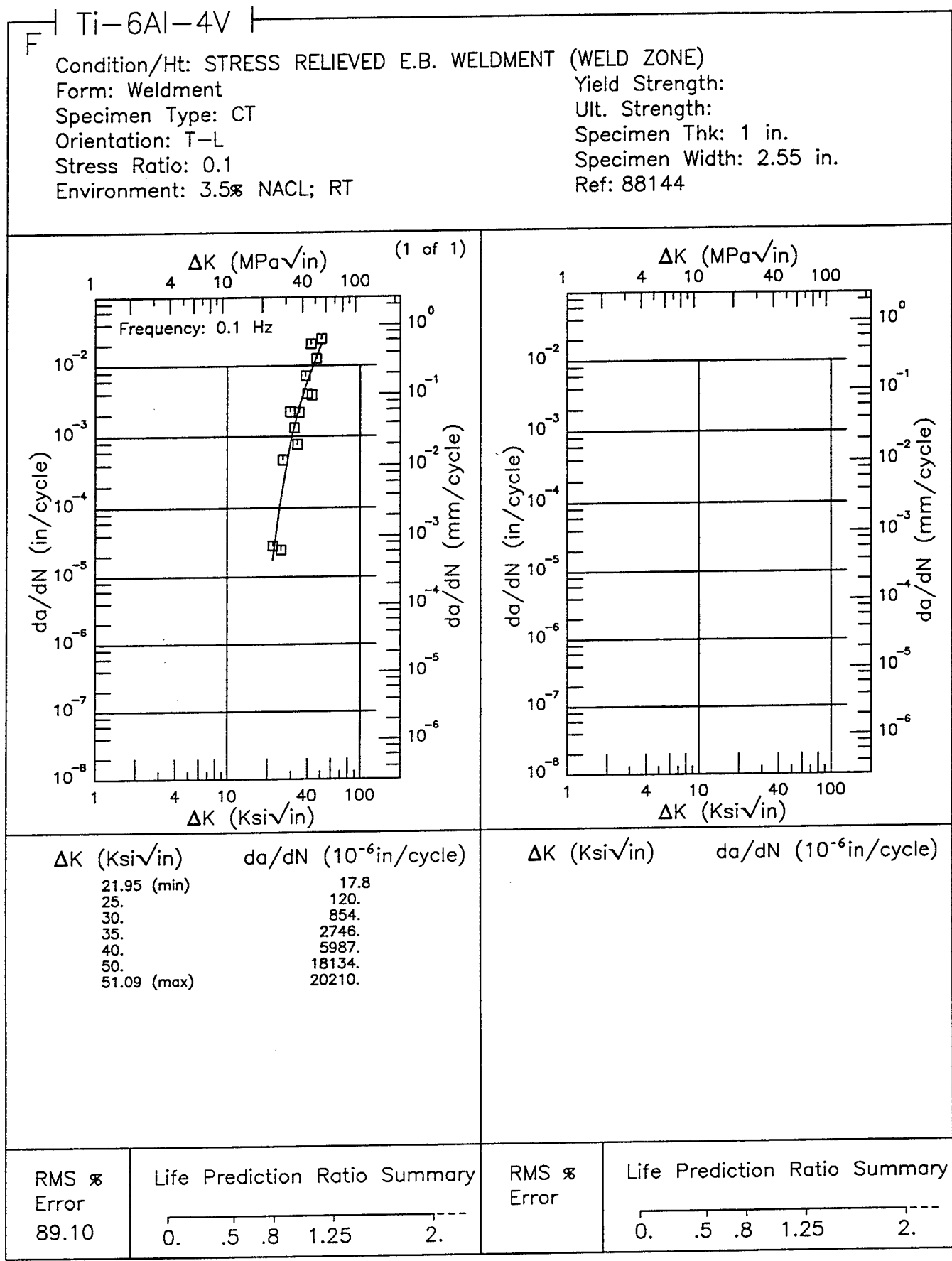


Figure 6.16.3.1.136

Ti-6Al-4V F

Condition/Ht: STRESS RELIEVED E.B. WELDMENT (WELD ZONE)
 Form: Weldment Yield Strength:
 Specimen Type: CT Ult. Strength:
 Orientation: T-L Specimen Thk: 1 in.
 Stress Ratio: 0.1 Specimen Width: 2.55 in.
 Environment: 3.5% NaCl; 175°F Ref: 88144

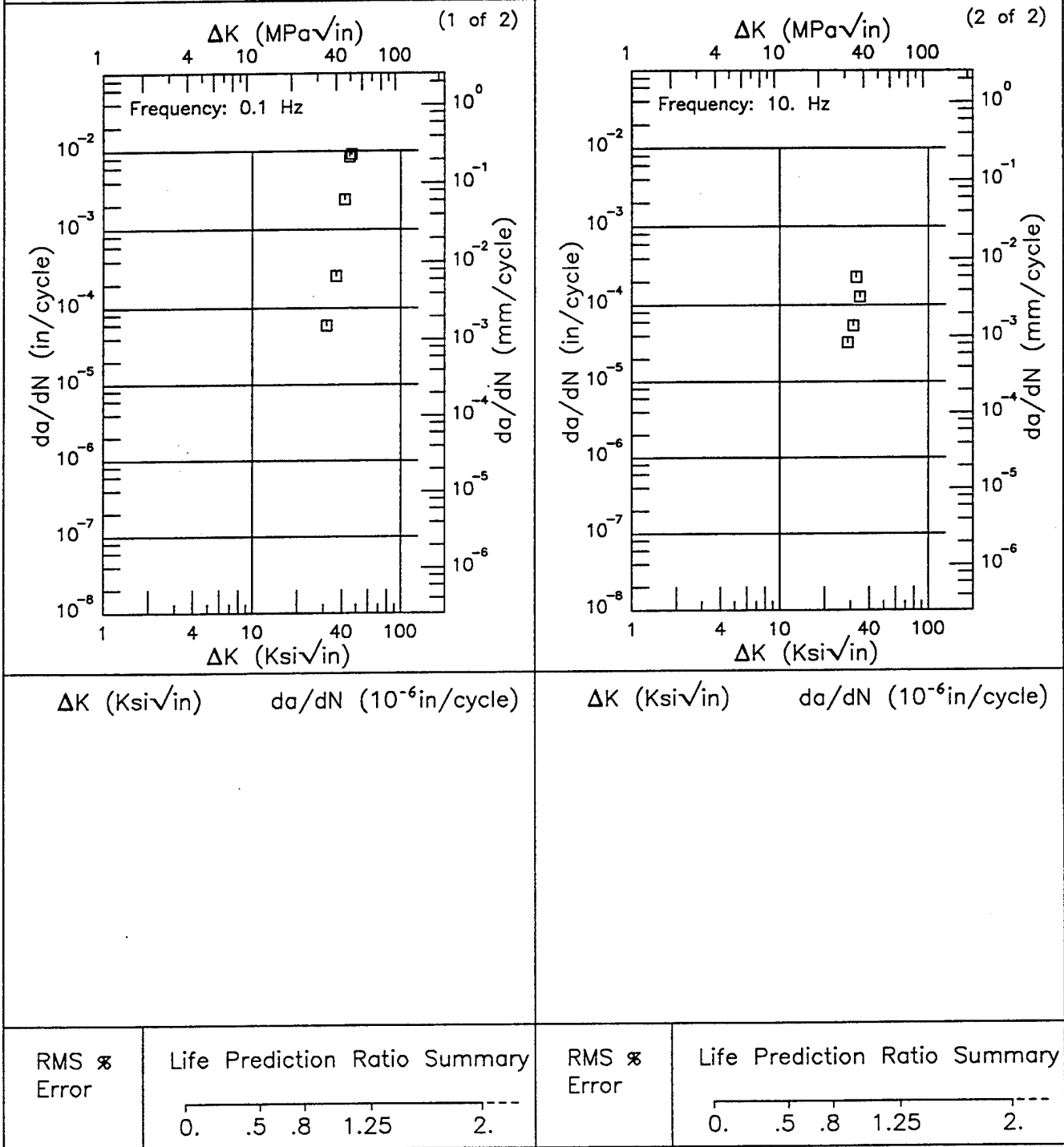
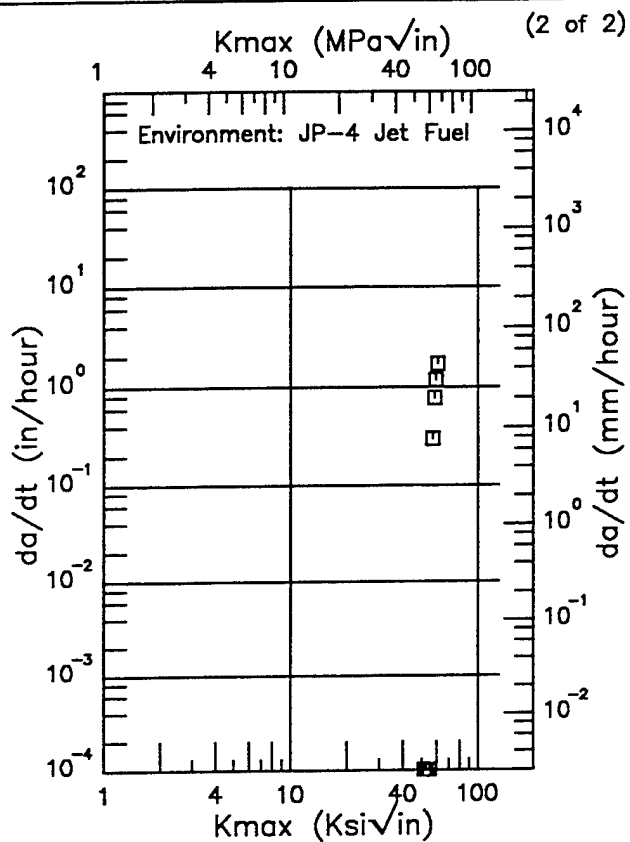
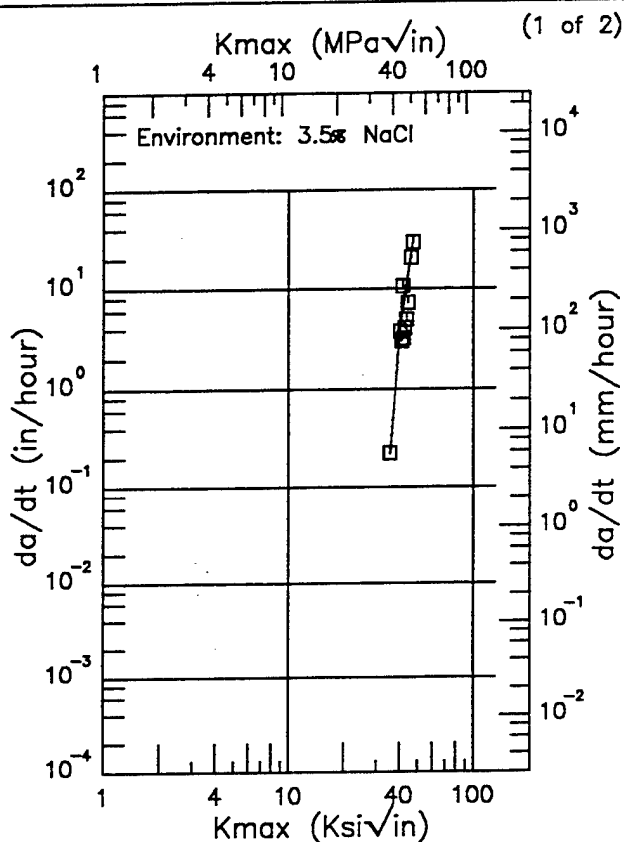


Figure 6.16.3.1.137

Ti-6Al-4V

Condition/Ht: 1000F 2HR
 Form: 2 in. Forging
 Specimen Type: TDCB
 Orientation: L-T
 Yield Strength: 145 ksi
 Ult. Strength:

Specimen Thk: 1.25 in.
 Specimen Width: 5.5 in.
 A₀:
 K_Isc: 31 - 43 ksi
 Ref: 84360



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
35.60 (min)	222.
40.	4001.
47.00 (max)	28004.

Kmax (Ksi√in) da/dt (10⁻³in/hour)

Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
40.	0.001
45.	0.01
50.	0.05
55.	0.1
60.	0.2

RMS %
 Error
 41.5

RMS %
 Error

Figure 6.16.3.2.1

Ti-6Al-4V

Condition/Ht: EB WELD STRESS RELIEVED (HEAT AFFECTED ZONE)

Form: Weldment

Specimen Type: CT

Orientation: T-L

Yield Strength: 132 ksi

Ult. Strength:

Specimen Thk: 1 in.

Specimen Width: 2.55 in.

A₀:

K_{Isc}: 33 ksi

Ref: 88144

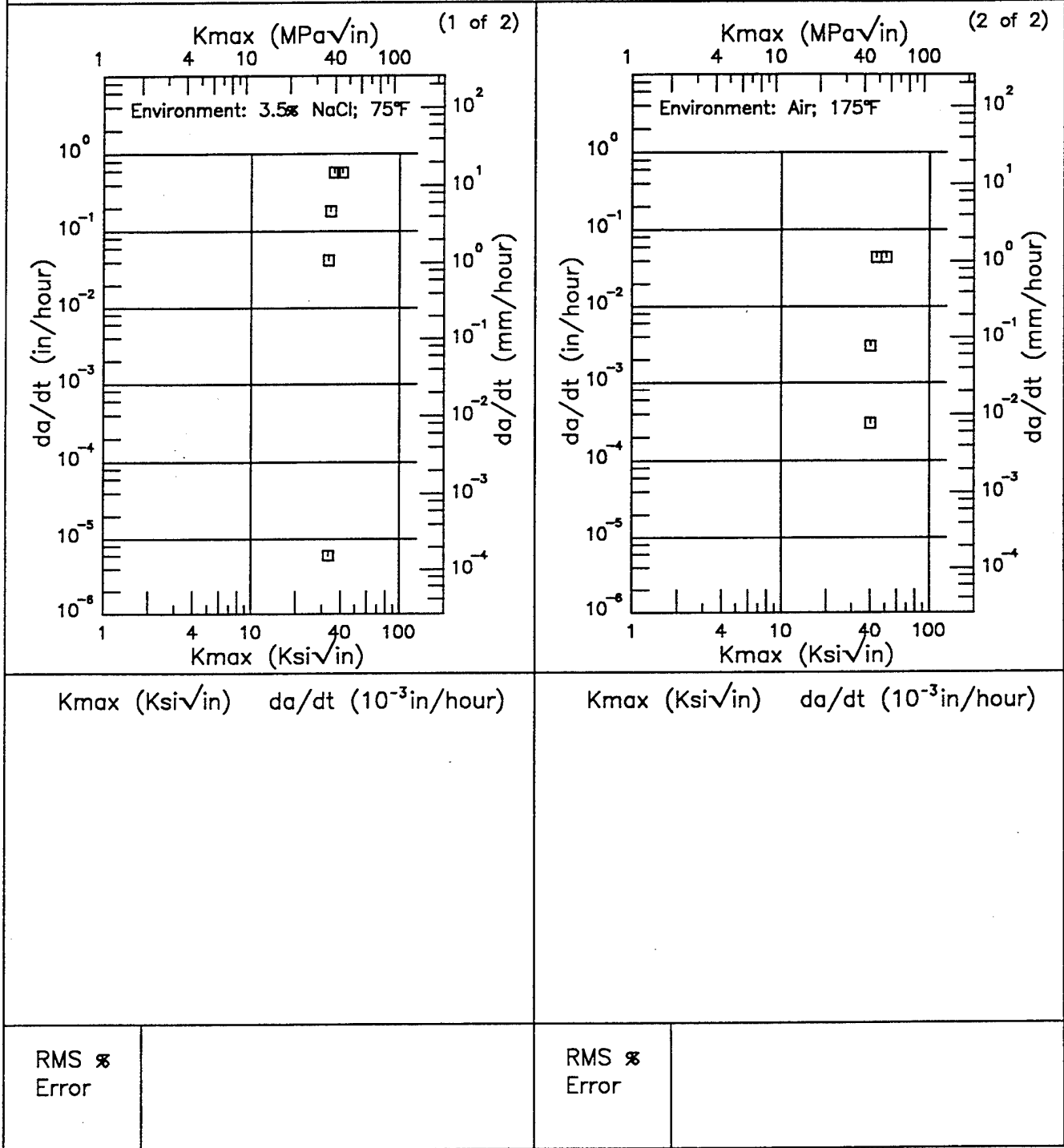


Figure 6.16.3.2.2

Ti-6Al-4V

Condition/Ht: EB WELD STRESS RELIEVED (WELD ZONE)

Form: Weldment

Specimen Type: CT

Orientation: T-L

Yield Strength: 132 ksi

Ult. Strength:

Specimen Thk: 1 in.

Specimen Width: 2.55 in.

A₀:

K_{Isc}: 36 - 51.5 ksi

Ref: 88144

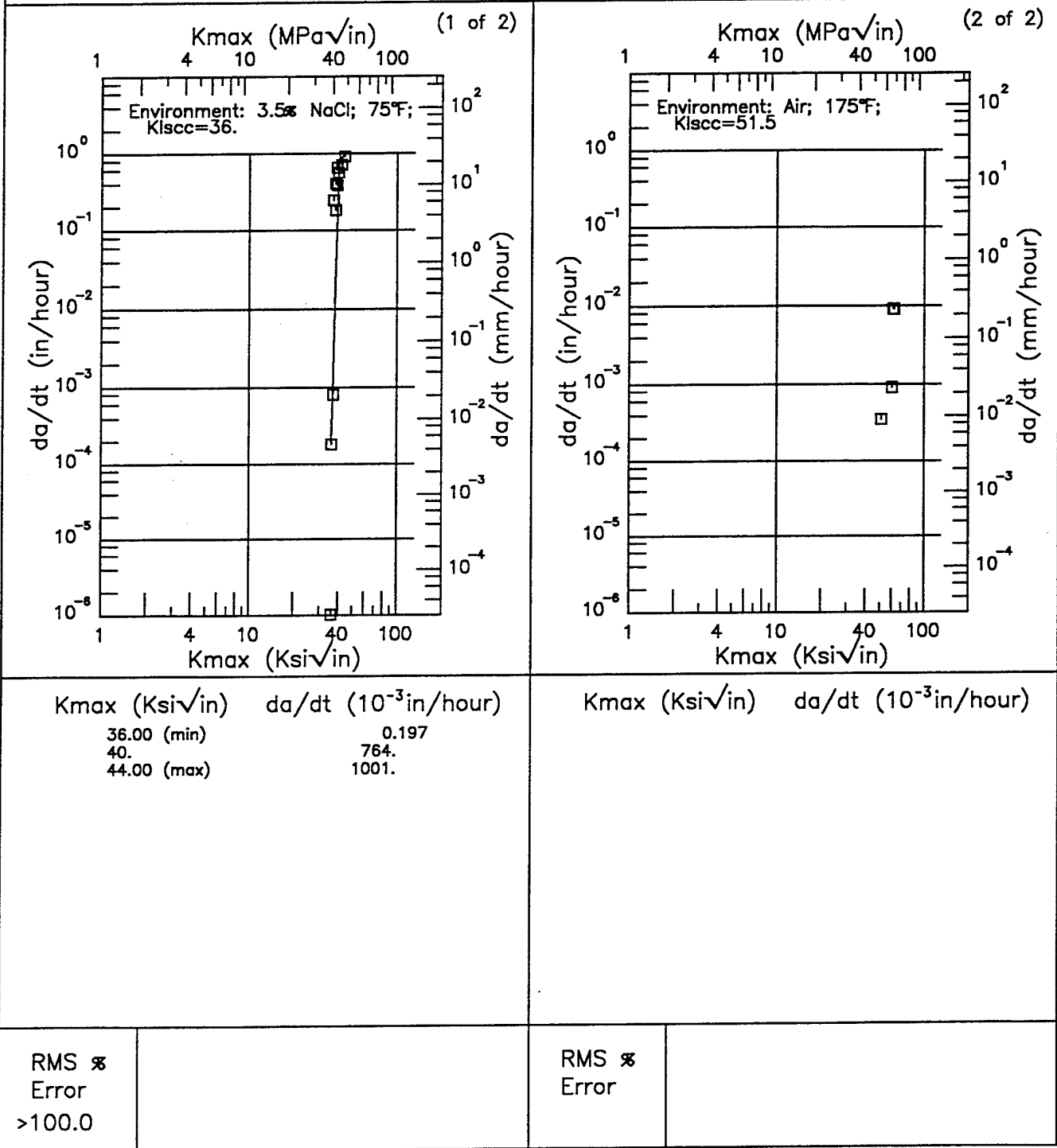


Figure 6.16.3.2.3

Ti-6Al-4V

Condition/Ht: MA
 Form: 0.2 in. Sheet
 Specimen Type: DCB
 Orientation: T-L
 Yield Strength:
 Ult. Strength:

Specimen Thk: 0.2 in.
 Specimen Width:
 A₀:
 K_{Isc}:
 Ref: 81221

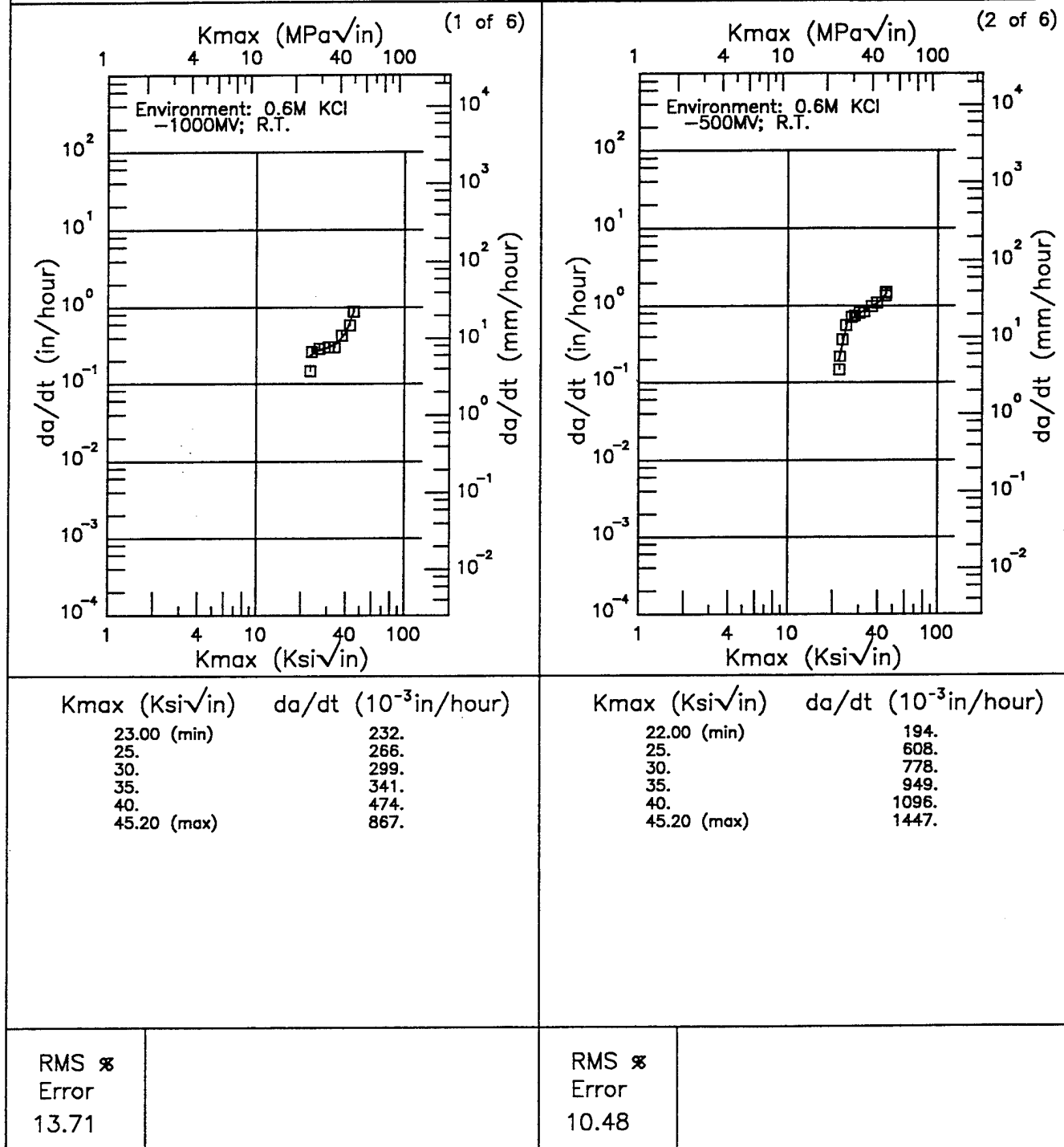
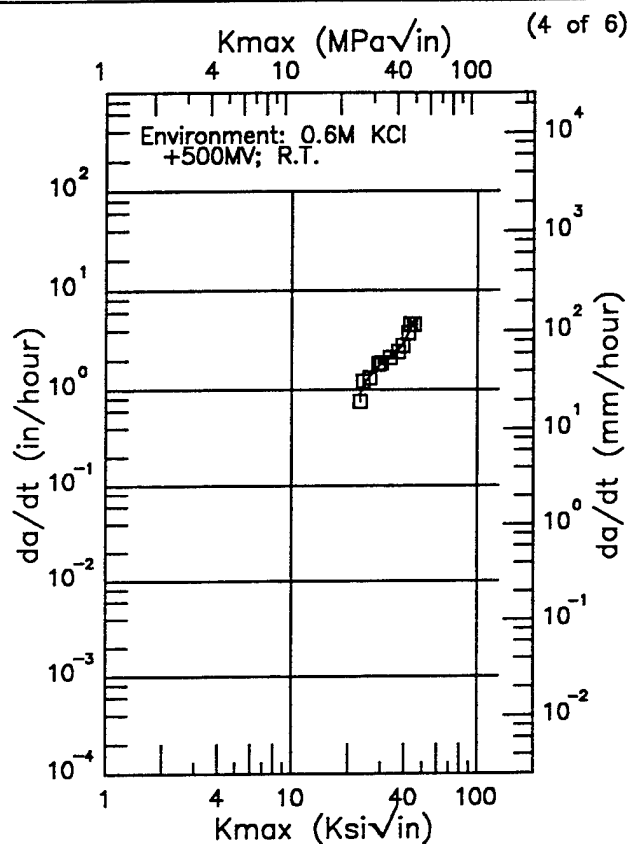
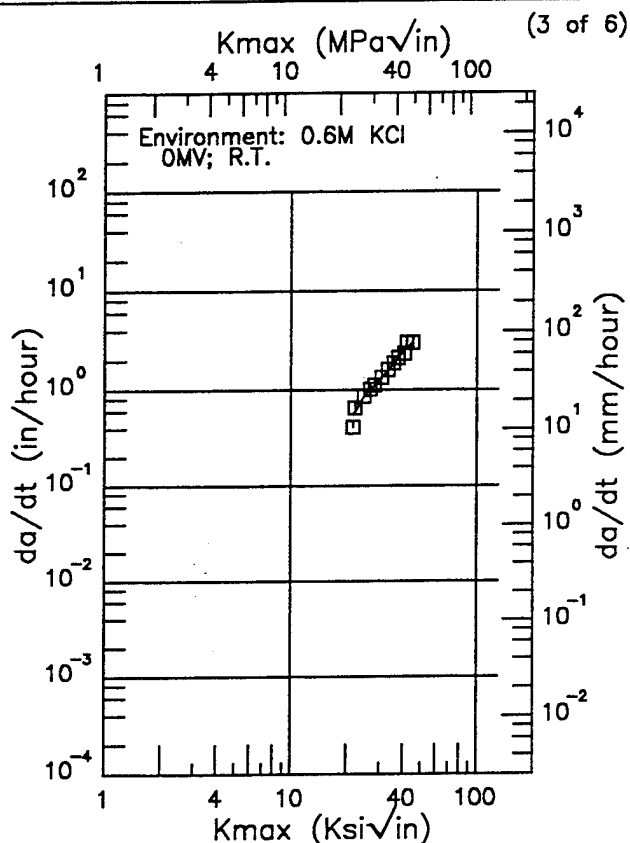


Figure 6.16.3.2.4

Ti-6Al-4V

Condition/Ht: MA
 Form: 0.2 in. Sheet
 Specimen Type: DCB
 Orientation: T-L
 Yield Strength:
 Ult. Strength:

Specimen Thk: 0.2 in.
 Specimen Width:
 A₀:
 K_Isc:cc:
 Ref: 81221



K _{max} (Ksi√in)	da/dt (10 ⁻³ in/hour)
21.50 (min)	549.
25.	898.
30.	1264.
35.	1785.
40.	2470.
45.00 (max)	3072.

K _{max} (Ksi√in)	da/dt (10 ⁻³ in/hour)
23.00 (min)	882.
25.	1236.
30.	1823.
35.	2242.
40.	3027.
45.20 (max)	5110.

RMS %
 Error
 8.74

RMS %
 Error
 8.28

Figure 6.16.3.2.4 (Continued)

Ti-6Al-4V

Condition/Ht: MA
 Form: 0.2 in. Sheet
 Specimen Type: DCB
 Orientation: T-L
 Yield Strength:
 Ult. Strength:

Specimen Thk: 0.2 in.
 Specimen Width:
 A₀:
 K_{Isc}:
 Ref: 81221

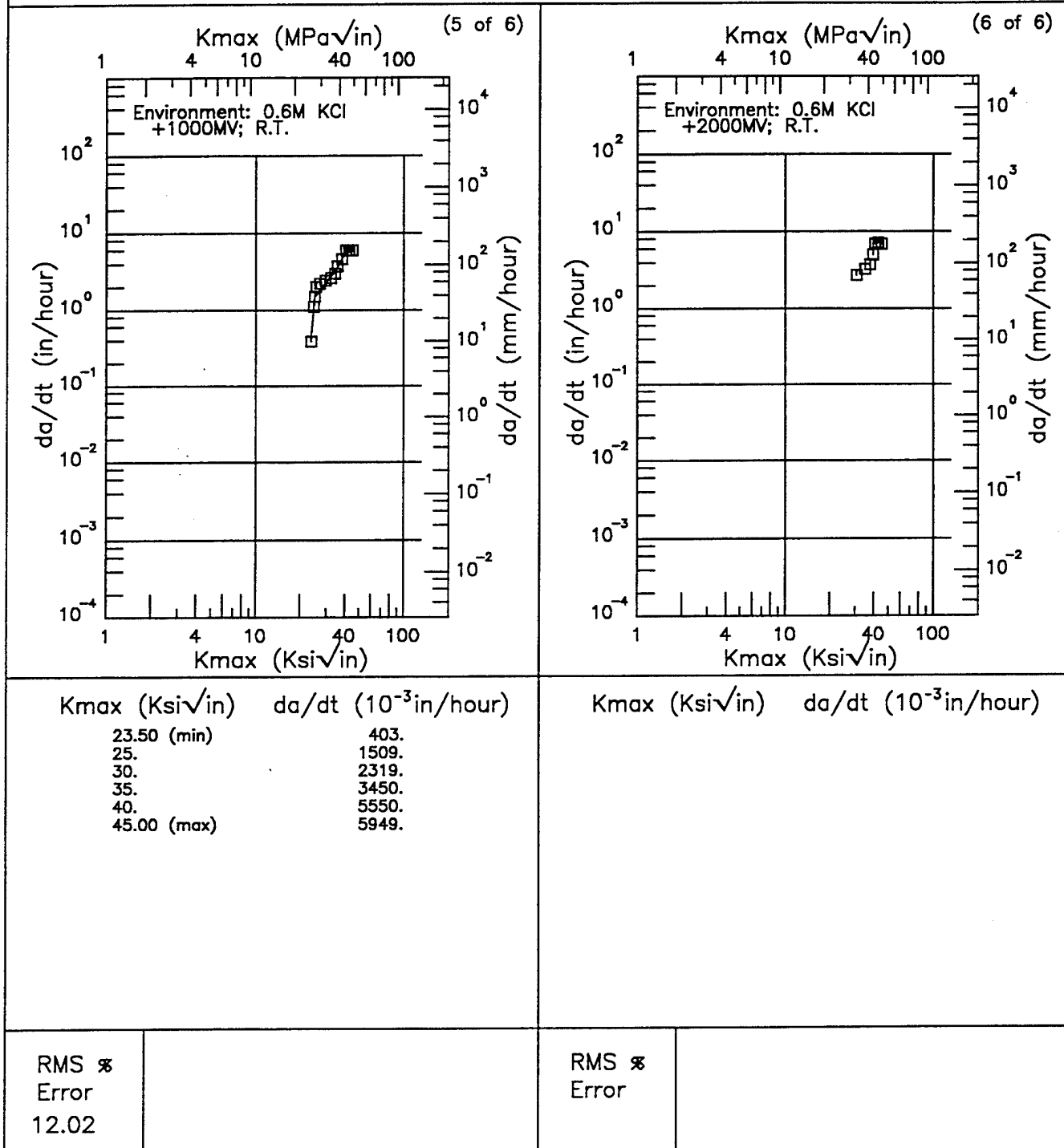


Figure 6.16.3.2.4 (Concluded)

Ti-6Al-4V

Condition/Ht: MA
 Form: 0.2 in. Sheet
 Specimen Type: DCB
 Orientation: T-L
 Yield Strength:
 Ult. Strength:

Specimen Thk: 0.2 in.
 Specimen Width:
 A₀:
 K_{Isc}: 35 ksi
 Ref: 81221

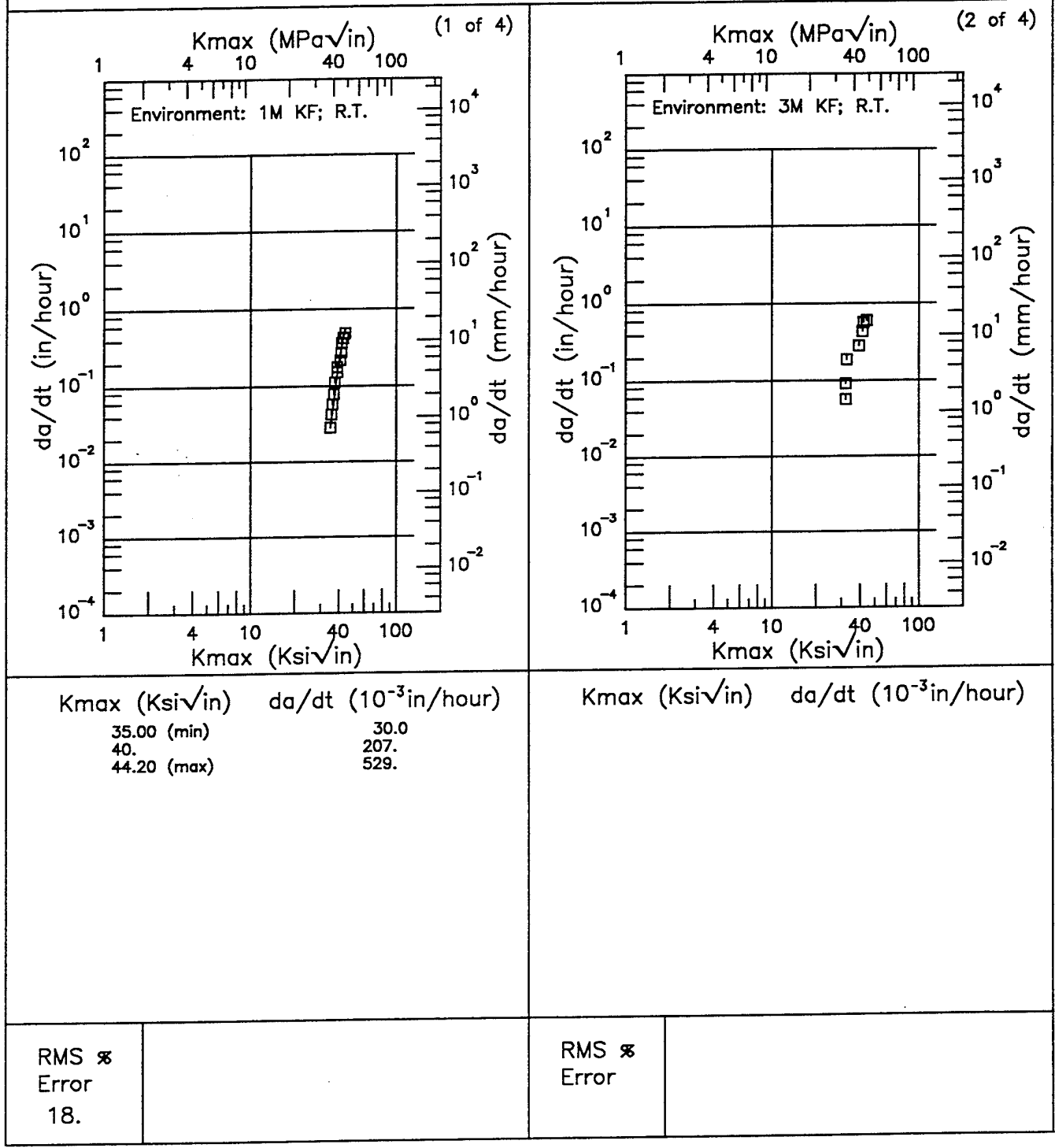
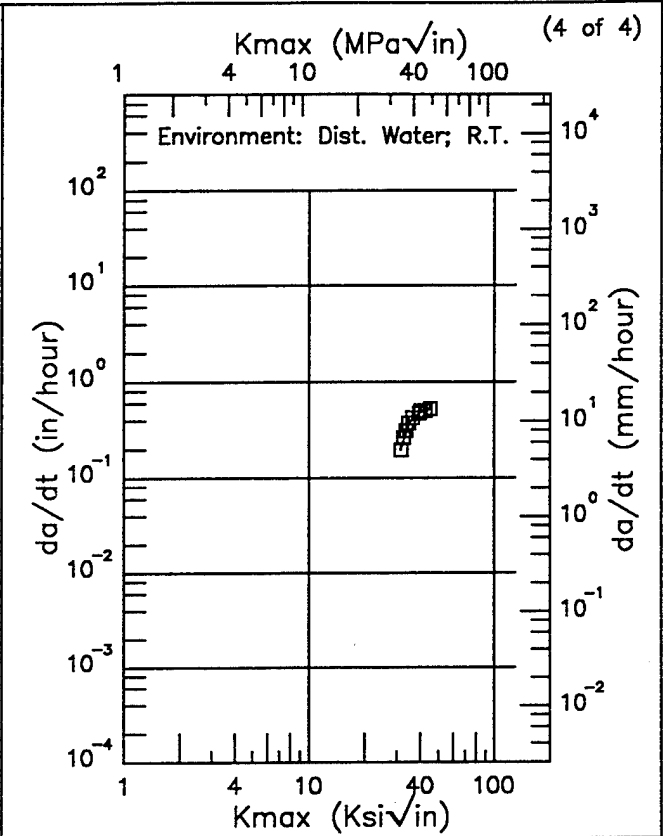
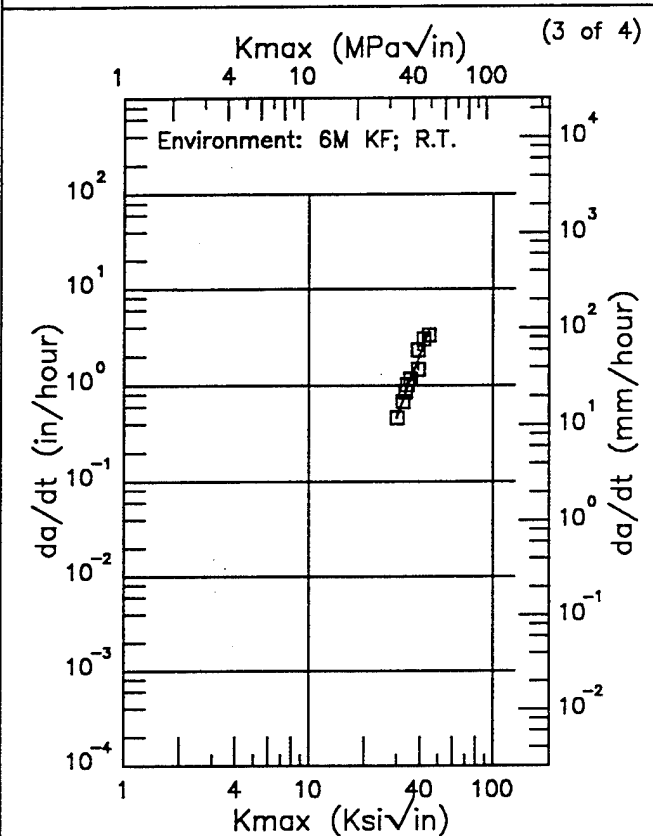


Figure 6.16.3.2.5

Ti-6Al-4V

Condition/Ht: MA
 Form: 0.2 in. Sheet
 Specimen Type: DCB
 Orientation: T-L
 Yield Strength:
 Ult. Strength:

Specimen Thk: 0.2 in.
 Specimen Width:
 A₀:
 K_Isec: 35 ksi
 Ref: 81221



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
30.00 (min)	461.
35.	1075.
40.	2121.
44.50 (max)	3590.

Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
31.40 (min)	196.
35.	381.
40.	485.
45.00 (max)	522.

RMS %
 Error
 11.91

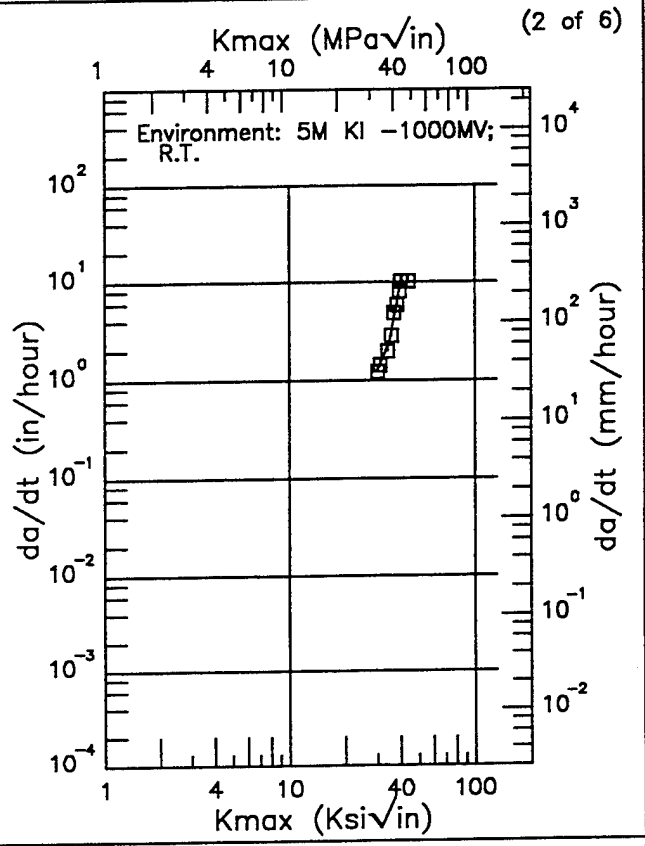
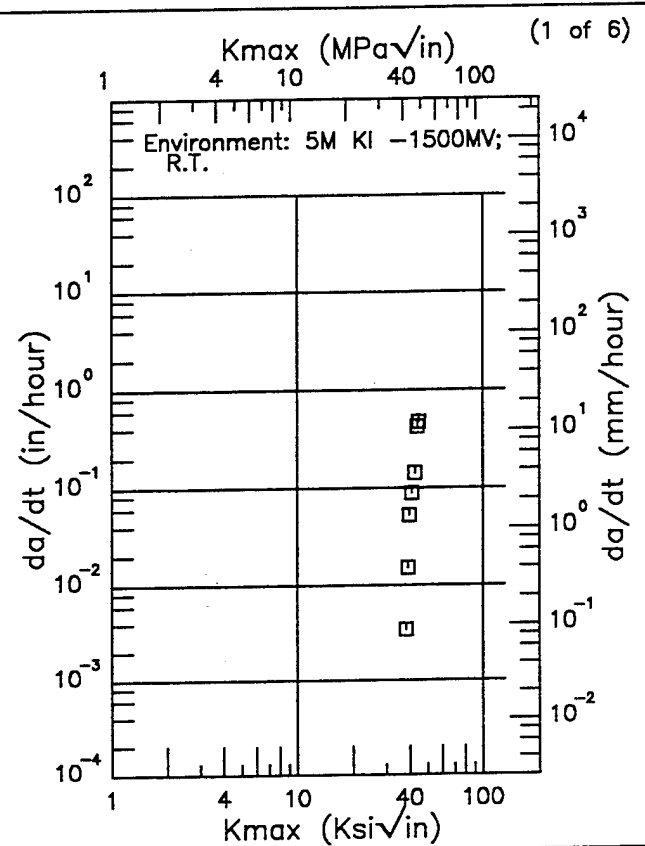
RMS %
 Error
 3.87

Figure 6.16.3.2.5 (Concluded)

Ti-6Al-4V

Condition/Ht: MA
 Form: 0.2 in. Sheet
 Specimen Type: DCB
 Orientation: T-L
 Yield Strength:
 Ult. Strength:

Specimen Thk: 0.2 in.
 Specimen Width:
 A₀:
 K_{Isc}:
 Ref: 81221



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
30.00 (min)	1237.
35.	2496.
40.	9603.
44.00 (max)	10002.

Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
30.00 (min)	1237.
35.	2496.
40.	9603.
44.00 (max)	10002.

RMS ⌘
 Error

RMS ⌘
 Error
 7.12

Figure 6.16.3.2.6

Ti-6Al-4V

Condition/Ht: MA
 Form: 0.2 in. Sheet
 Specimen Type: DCB
 Orientation: T-L
 Yield Strength:
 Ult. Strength:

Specimen Thk: 0.2 in.
 Specimen Width:
 A₀:
 K_{Isc}:
 Ref: 81221

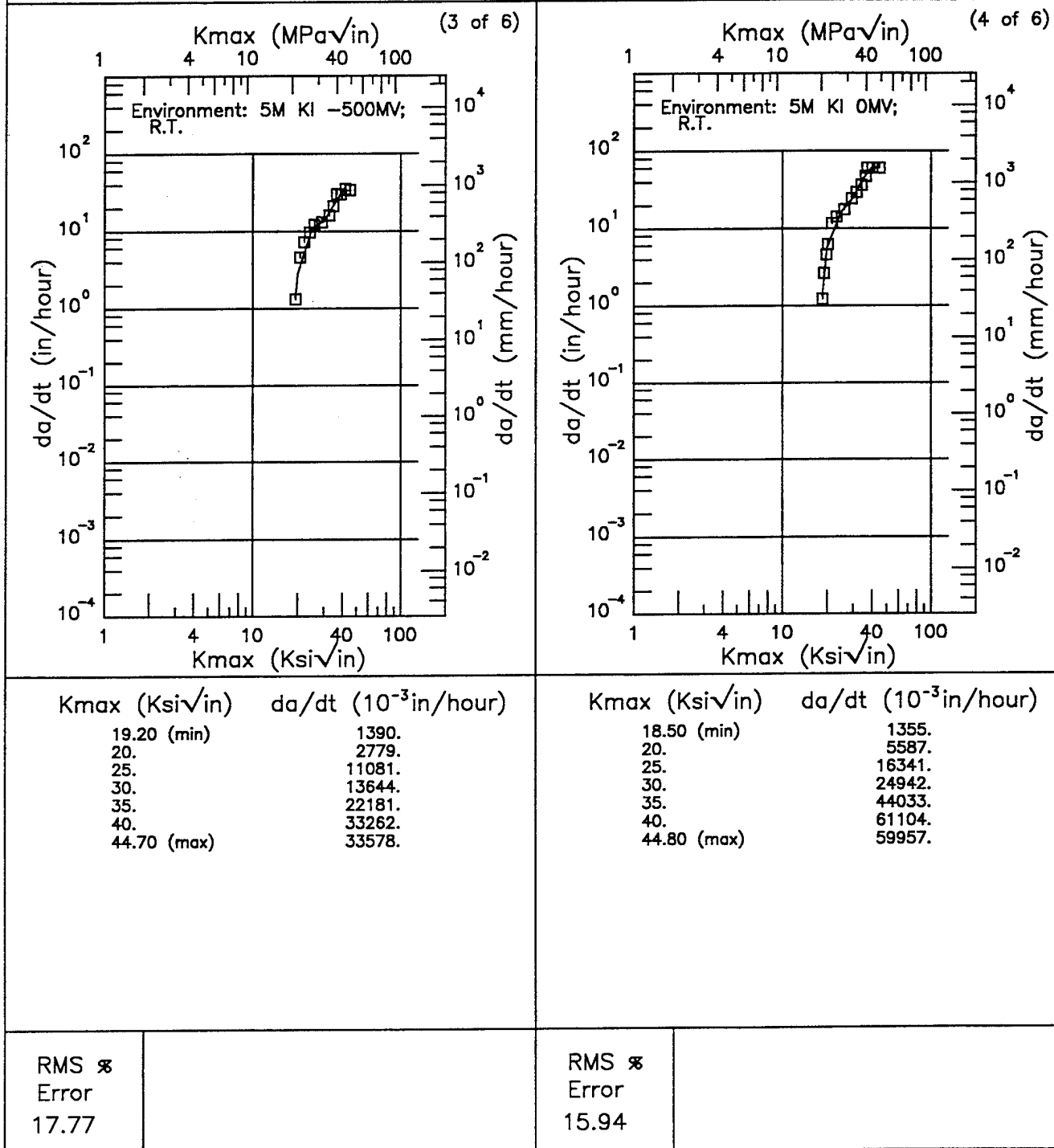
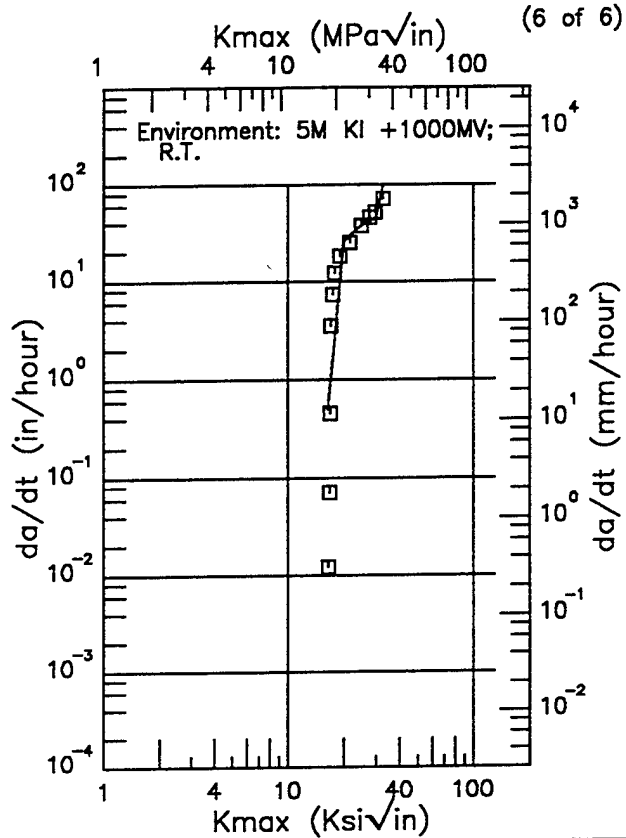
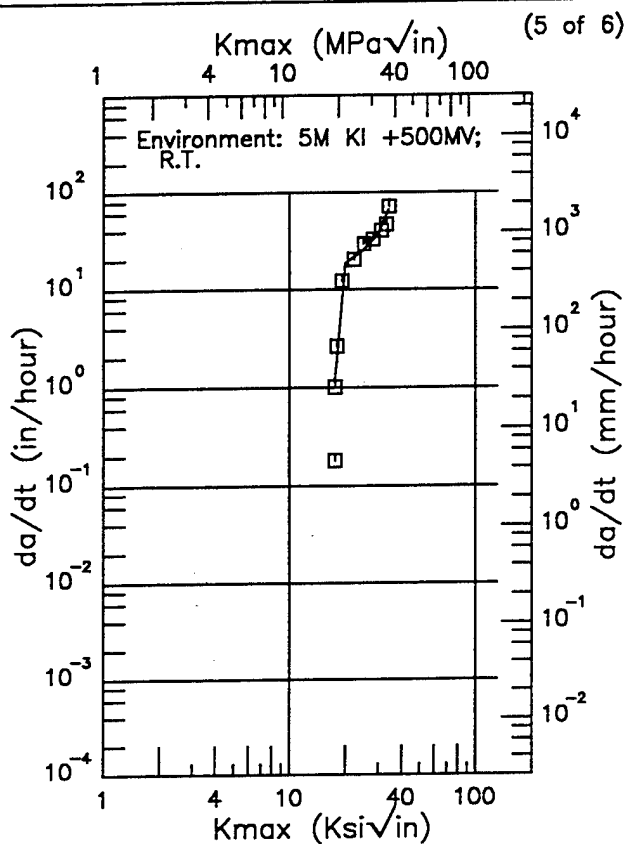


Figure 6.16.3.2.6 (Continued)

Ti-6Al-4V

Condition/Ht: MA
 Form: 0.2 in. Sheet
 Specimen Type: DCB
 Orientation: T-L
 Yield Strength:
 Ult. Strength:

Specimen Thk: 0.2 in.
 Specimen Width:
 A₀:
 K_Isc: Ref: 81221



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
17.50 (min)	1161.
20.	18558.
25.	25355.
30.	36711.
34.00 (max)	66053.

Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
16.50 (min)	503.
20.	25954.
25.	37810.
30.	45074.
32.50 (max)	98475.

RMS %
 Error
 37.84

RMS %
 Error
 >100.0

Figure 6.16.3.2.6 (Concluded)

(1 of 8)

TABLE 6.16.3.3

 K_{Isec} SUMMARY FOR TITANIUM ALLOY Ti-6Al-4V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Crack (in)	K_{Ic} (Ksi√in)	K_{Isec} (Ksi√in)	Test Time (min)	Test Date	Refer				
						Design	Width (in)	Thick (in)							Prod Thk (in)			
1300° F 2hr AC	E	R.T.	L-S	121.2	3.5% NaCl	NB	1.5	0.48	0.5	---	94.7	65*	---	1972	84282			
			L-T	128.9			1.5	0.48	0.5	---	83.3	73*	---	1972	84282			
1700° F 4hr FC to 1400° F AC; Diffusion Bond Thermal Cycle	P	R.T.	L-T	---	S.T.W.	DCB	5.5	1	1.5	---	---	---	---	---	1974	89004		
							5.5	1	1.5	---	---	---	---	---	1974	89004		
							5.5	1	1.5	---	---	---	---	---	1974	89004		
			T-L	---		S.T.W.	DCB	5.5	1	1.5	---	92	44*	---	---	---	1974	89004
								5.5	1	2	---	92	56*	---	---	1974	89004	
								5.5	1	2	---	92	53*	---	---	1974	89004	
1725° F 1hr WQ; 1000° F 1hr AC; (STA)	E	R.T.	L-S	145.7	Shop Cleaning Solvent	DCB	5.5	1	1.5	---	94	69*	---	1974	89004			
							1.5	0.48	0.5	---	67.2	51	---	---	1972	84282		
1725° F 1hr WQ; 1250° F 4hr AC; (STOA)	E	R.T.	L-S	146.9	3.5% NaCl	NB	1.5	0.48	0.5	---	65.8	46	---	1972	84282			
							1.5	0.48	0.5	---	77.8	60	---	---	1972	84282		
				140.3	3.5% NaCl	NB	1.5	0.48	0.5	---	74.9	70*	---	1972	84282			

Ti-6Al-4V

(2 of 8)

TABLE 6.16.3.3 (CONTINUED)

 K_{Isec} SUMMARY FOR TITANIUM ALLOY Ti-6Al-4V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Crack (in)	K_{Ic} (Ksi√in)	Test Time (min)	Test Date	Refer
						Design	Width (in)	Thick (in)					
1750° F 1.5hr WQ; 1050° F- 1100° F 8hr; 950° F 8hr	F	R.T.	---	160.9	Aerazine 50	PTSC	0.8	0.095	---	52.2	---	1969	75528
					Freon TF	PTSC	0.8	0.095	---	52.2	---	1969	75528
					N ₂ O ₄	PTSC	0.8	0.095	---	52.2	---	1969	75528
					Aerazine 50	PTSC	0.8	0.095	---	51.5	---	1969	75528
1750° F; 1000° F 2hr AC	F	R.T.	L-T	161.4	Freon TF	PTSC	0.8	0.095	---	51.5	---	1969	75528
					N ₂ O ₄	PTSC	0.8	0.095	---	51.5	---	1969	75528
					3.5% NaCl	TDCB	5	1.25	2.3	50.9	---	1971	84360
					JP-4 Fuel	TDCB	5	1.25	2.3	50.9	---	1971	84360
1750° F WQ; 1000° F 8hr 1000° F; (Alpha+Beta)	F	R.T.	---	160	Dist Water +500	PTSC	1	0.03	0.5	43.5	---	1968	77290
					Methanol	PTSC	1	0.06	0.5	45.9	---	1968	77290
						PTSC	1	0.03	0.5	43.5	---	1968	77290
					PPM Na ₂ Cr ₂ O ₇	PTSC	1	0.06	0.5	45.9	---	1968	77290
1790° F 1.5hr WQ; 1160° F 8hr + 1025° F 8hr AC	S	R.T.	---	---	Martin- Marietta Refined Grade Hydrazine	WOL	1.3	0.125	1.25	---	---	1974	88700
					Propellant Grade Hydrazine	WOL	1.3	0.125	1.25	---	---	1974	88700

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TABLE 6.16.3.3 (CONTINUED)

 $K_{I_{ecc}}$ SUMMARY FOR TITANIUM ALLOY Ti-6Al-4V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K_q (Ksi/in)	$K_{I_{ecc}}$ (Ksi/in)	Test Time (min)	Test Date	Refer	
						Design	Width (in)	Thick (in)								
Alpha-Beta Forged	F	R.T.	T-L	144.9	3.5% NaCl	CT	2	1	2.25	---	36.5	27	---	1973	86688	
						CT	2	1	2.25	---	36.5	27	---	1973	86688	
						CT	2	1	2.25	---	36.5	18	---	1973	86688	
Beta Forged	F	R.T.	T-L	134.9	3.5% NaCl	CT	2	1	2.25	---	44.2	34	---	1973	86688	
						CT	2	1	2.25	---	44.2	42	---	1973	86688	
						CT	2	1	2.25	---	44.2	42	---	1973	86688	
Finish Rolled 1440° F	P	R.T.	T-S	114	3.5% NaCl	NB*	3	1.5	3	1.05	---	74.5	---	1972	84036	
						NB*	3	1.5	3	1.11	---	79.6	---	1972	84036	
						NB*	2	0.75	3	1.15	---	116	94.6*	---	1972	84036
						NB*	2	0.75	3	1.17	---	116	90.1*	---	1972	84036
						NB*	2	0.75	3	1.22	---	116	95*	---	1972	84036
						NB*	2	0.75	3	0.3	0.3	116	96*	---	1972	84036
						NB*	3	1.5	3	1.04	---	---	78.6	---	1972	84036
						NB*	3	2	3	0.92	---	---	61.5	---	1972	84036
						NB*	3	2	3	1	---	---	78	---	1972	84036
						NB*	3	1.5	3	1.16	---	---	87.2	---	1972	84036
						NB*	3	2	3	0.96	---	---	73.7	---	1972	84036

Ti-6Al-4V

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TABLE 6.16.3.3 (CONTINUED)
K_{I_{sec}} SUMMARY FOR TITANIUM ALLOY Ti-6Al-4V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K _q (Ksi/in)	K _{I_{sec}} (Ksi/in)	Test Time (min)	Test Date	Refer
						Design	Width (in)	Thick (in)							
GTA Weld; Postweld 1100° F 2hr (Heat Affected Zone)	P	R.T.	L-T	---	Field Cleaning Solvent	DCB	5.5	0.5	1.25	---	---	>48*	---	1974	89004
					Shop Cleaning Solvent	DCB	5.5	0.5	1.25	---	---	>64*	---	1974	89004
GTA Weld; Postweld 1100° F 2hr (Weld Zone)	P	R.T.	L-T	---	S.T.W.	DCB	5.5	0.5	1.25	---	---	99*	---	1974	89004
						DCB	5.5	0.75	1.25	---	---	76*	---	1974	89004
GTA Weld; Postweld 1100° F 2hr (Heat Affected Zone)	P	R.T.	L-T	---	S.T.W.	DCB	5.5	0.5	1.25	---	---	>89*	---	1974	89004
						DCB	5.5	1	1.25	---	---	58*	---	1974	89004
GTA Weld; Postweld 1200° F 1hr (Heat Affected Zone)	P	R.T.	L-T	---	S.T.W.	DCB	5.5	0.125	1.25	---	---	>67*	---	1974	89004
						DCB	5.5	0.25	1.25	---	---	>66*	---	1974	89004
GTA Weld; Postweld 1400° F 1hr (Heat Affected Zone)	P	R.T.	L-T	---	S.T.W.	DCB	5.5	0.125	1.25	---	---	>62*	---	1974	89004
						DCB	5.5	0.25	1.25	---	---	>70*	---	1974	89004
As Received; Probably Mill Annealed	P	R.T.	T-S	125	3.5% NaCl	CANT	---	---	---	---	120	105*	---	1968	74355

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TABLE 6.16.3.3 (CONTINUED)

K_{Iacc} SUMMARY FOR TITANIUM ALLOY Ti-6Al-4V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K _Q (Ksi/in)	K _{Iacc} (Ksi/in)	Test Time (min)	Test Date	Refer			
						Design	Width (in)	Thick (in)										
Mill Annealed	S	R.T.	T-L	---	A-50	WOL	1.3	0.125	0.13	---	---	68*	---	1974	88700			
					MCB	WOL	1.3	0.125	0.13	---	---	38.6*	---	---	---	1974	88700	
					MCB/1% CO ₂	WOL	1.3	0.125	0.13	---	---	40.8*	---	---	---	---	1974	88700
					1 M KF	DCB	---	0.2	0.2	---	---	35*	---	---	---	---	1971	81221
					3 M KF	DCB	---	0.2	0.2	---	---	37*	---	---	---	---	1971	81221
					6 M KF	DCB	---	0.2	0.2	---	---	31*	---	---	---	---	1971	81221
					6M KF -1500MV	DCB	---	0.2	0.2	---	---	38*	---	---	---	---	1971	81221
					6M KF -1000MV	DCB	---	0.2	0.2	---	---	31*	---	---	---	---	1971	81221
					6M KF -500MV	DCB	---	0.2	0.2	---	---	20*	---	---	---	---	1971	81221
					6M KF 0 MV	DCB	---	0.2	0.2	---	---	19*	---	---	---	---	1971	81221
					6M KF +1000MV	DCB	---	0.2	0.2	---	---	16*	---	---	---	---	1971	81221
					6M KF +500MV	DCB	---	0.2	0.2	---	---	17*	---	---	---	---	1971	81221
					Dist. Water	DCB	---	0.2	0.2	---	---	33*	---	---	---	---	1971	81221
					P	R.T.	L-S	T-S	---	3.5% NaCl	NB	---	---	0.5	---	61	---	---
3.5% NaCl	CANT*	1	0.75	1						---	88	67*	---	1967	70931			
3.5% NaCl	CANT	0.75	0.1	---						---	64	55*	---	---	1967	70931		
Minuteman Casing	P	R.T.	T-S	165	3.5% NaCl	CANT	0.75	0.1	---	---	---	---	---	---	---			

Ti-6Al-4V

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TABLE 6.16.3.3 (CONTINUED)
 $K_{I_{sec}}$ SUMMARY FOR TITANIUM ALLOY Ti-6Al-4V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K_q (Ksi \sqrt{in})	$K_{I_{sec}}$ (Ksi \sqrt{in})	Test Time (min)	Test Date	Refer
						Design	Width (in)	Thick (in)							
Recrystallize Anneal	P	R.T.	L-T	117	S.T.W.	DCB	5.5	1	1.5	---	74	60	60300	1976	RI006
						DCB	5.5	1	1.5	---	74	60	60300	1976	RI006
						DCB	5.5	1	1.5	---	74	55	54360	1976	RI006
						DCB	5.5	1	1.5	---	74	>69	60300	1976	RI006
					DCB	5.5	1	1.5	---	76	63	54360	1976	RI006	
					DCB	5.5	1	1.5	---	76	>67	54360	1976	RI006	
					DCB	5.5	1	1.5	---	76	>69	54360	1976	RI006	
					DCB	5.5	1	1.5	---	76	62	51720	1976	RI006	
					DCB	5.5	1	2.5	---	77	56	70140	1976	RI006	
					DCB	5.5	1	2.5	---	77	60	54360	1976	RI006	
					DCB	5.5	1	2.5	---	77	60	54360	1976	RI006	
					DCB	5.5	1	2.5	---	77	59	70140	1976	RI006	
F.C.S.	S.C.S.	122	T-L	122	S.T.W.	DCB	5.5	1	1.5	---	77	70	119100	1976	RI006
						DCB	5.5	1	1.5	---	77	63	112200	1976	RI006
						DCB	5.5	1	1.5	---	77	75	59100	1976	RI006
						DCB	5.5	1	1.5	---	77	62	54360	1976	RI006
DCB	5.5	1	1.5	---	77	61	60300	1976	RI006						
DCB	5.5	1	1.5	---	77	62	60300	1976	RI006						

TABLE 6.16.3.3 (CONTINUED)

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 $K_{I_{ecc}}$ SUMMARY FOR TITANIUM ALLOY Ti-6Al-4V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Crack (in)	K_Q (Ksi√in)	$K_{I_{ecc}}$ (Ksi√in)	Test Time (min)	Test Date	Refer									
						Design	Width (in)	Thick (in)							Prod Thk (in)								
	P (cont'd)	R.T. (cont'd)	T-L (cont'd)	122 (cont'd)	S.T.W. (cont'd)	5.5	1	1.5	---	77	58	60300	1976	RI006									
															DCB	5.5	1	2	---	80	55	61980	RI006
															DCB	5.5	1	2	---	80	61	61980	RI006
Recrystallize Anneal (cont'd)	F	R.T.	T-L	119	S.T.W.	5.5	1	4	---	75	51	60360	1976	RI006									
															DCB	5.5	1	4	---	75	55	60360	RI006
															DCB	5.5	1	---	---	75	56*	60180	RI006
	F	R.T.	S-L	---	S.T.W.	5.5	1	---	---	78	>57*	64920	1976	RI006									
															DCB	5.5	1	---	---	78	>71*	60660	RI006
															DCB	5.5	1	---	---	75	56*	60660	RI006
SOL Treated 1050° F 4+4 hr	F	85	---	160	N_2O_4 315PSIG	1	0.052	---	0.02	45	28*	---	1969	78535									
															PTSC	1	0.052	---	0.02	45	30*	---	1969
SOL Treated 1050° F 4hr; Welded 1050° F 4hr	F	85	---	133	N_2O_4 315PSIG	1.5	0.062	---	0.03	47	32*	---	1969	78535									
															PTSC	1.5	0.062	---	0.03	47	31*	---	1969
Unspecified	---	R.T.	L-T	116	3.5% NaCl	---	---	---	---	115	95*	---	1967	70887									

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TABLE 6.16.3.3 (CONCLUDED)

K_{Isec} SUMMARY FOR TITANIUM ALLOY Ti-6Al-4V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K _o (Ksi/in)	K _{Isec} (Ksi/in)	Test Time (min)	Test Date	Refer
						Design	Width (in)	Thick (in)							
Unspecified	F	R.T.	---	105	3.5% NaCl	CANT	1.1	0.75	---	---	98	80*	---	1968	84328
						CANT	1.1	0.75	---	---	98	60*	---	1968	84328
						CANT	1.1	0.75	---	---	98	80*	---	1968	84328
Unspecified	F	R.T.	S-T	106	JP-4 Fuel	BWOL	3.075	1.249	1.25	1.37	---	>66.5	133920	1977	MA005
						BWOL	3.073	1.251	1.25	1.36	---	>66.6	133920	1977	MA005
					Sim. Sea Water	BWOL	3.077	1.249	1.25	1.62	---	>69.6	133920	1977	MA005
						BWOL	3.077	1.25	1.25	1.37	---	>69.3	133920	1977	MA005
					JP-4 Fuel	BWOL	3.078	1.251	1.25	1.38	---	>69	133920	1977	MA005
						BWOL	3.078	1.252	1.25	1.36	---	>68	133920	1977	MA005
Sim. Sea Water	BWOL	3.077	1.251	1.25	1.36	---	>67.7	133920	1977	MA005					
	BWOL	3.073	1.253	1.25	1.36	---	>68.5	133920	1977	MA005					

* specimen thickness does not meet minimum requirements of $2.5 \left(\frac{K_{Isec}^2}{\sigma_{ys}} \right)$

* asterisk in specimen design column indicates that specimens are side-grooved

TABLE 6.17.1.1

MEAN PLANE STRAIN FRACTURE TOUGHNESS
FOR TITANIUM ALLOY Ti-6Al-4V ELI AT ROOM TEMPERATURE

Product Form	Condition/Heat Treatment	K_{Ic} ($ksi\sqrt{in}$)								
		Specimen Orientation						S-L		
		L-T			T-L			Mean K_{Ic}	Std Dev	n
Plate	RA	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n
Forging	ANNEALED	76.1	4.	3	76.8	0.7	3
		83.5	1.3	3	84.3	0.4	3

Ti-6Al-4V ELI

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TABLE 6.17.1.2.1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V ELI AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.6	5.0	10.0	20.0	50.0	100.0
ANNEALED	FORGING	0.1	1-10				13.27	209.7	
		-1	8			3.1			
		-1	10				23.32		
		-0.66	8			2.97	22.01		
		-0.66	10				23.96		
		-0.33	8			2.5	21.42		
		-0.33	10				20.27		
		0	8			1.61	21.49		
		0	10			1.71	18.61		
		0.1	10				11.54		
RA	PLATE	0.55	8	0.01	0.2	5.22			

TABLE 6.17.1.2.2

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V ELI AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: S.S.W.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	6.0	10.0	20.0	50.0	100.0
RA	PLATE	0.1	1-10				16.66	372.39	

Ti-6Al-4V ELI

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TABLE 6.17.1.2.3

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V ELI AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: 3.5% NaCl

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi/ \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
BA	PLATE	0.	15				11.36			
		0.1	0.1					308.61		
		0.1	1						214.39	
		0.5	0.1				17.18			
		0.5	1				32.61			

TABLE 6.17.1.2.4

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V ELI AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: ALT JP4/H₂O (D)

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	6.0	10.0	20.0	60.0	100.0
BA	PLATE	0.1	1						
		0.5	1				12.9		121.9

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TABLE 6.17.1.2.5

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V ELI AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: Distilled Water

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
BA	PLATE	0	16						
		0.1	1				6.45		
		0.5	1			0.25	12.97	140.93	

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TABLE 6.17.1.2.6

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V ELI AT ROOM TEMPERATURE

ORIENTATION: T-L

ENVIRONMENT: Dry Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi $\sqrt{\text{in}}$)						
				2.5	5.0	10.0	20.0	50.0	100.0	
BA	PLATE	0.	16							
		0.1	1					130.8		
		0.5	0.1				11.6			
		0.5	1				11.87			
		0.8	1			1.08				

Ti-6Al-4V ELI

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TABLE 6.17.1.2.7

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V ELI AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
ANNEALED	FORGING	0.1	1-10				12.21	249.29	
		0.1	5-20				8.34	168.75	
RA	PLATE	0.1	1-10				7.61	227.75	

TABLE 6.17.1.2.8

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V ELI AT ROOM TEMPERATURE

ENVIRONMENT: S.S.W.

ORIENTATION: T-L

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
RA	PLATE	0.1	1-10	2.5	5.0	10.0	20.0	50.0	100.0

Ti-6Al-4V ELI

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TABLE 6.17.1.2.9

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-4V ELI AT ROOM TEMPERATURE

ORIENTATION: T-L

ENVIRONMENT: S.T.W.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	6.0	10.0	20.0	50.0	100.0	
BA	PLATE	0.1	1							
		0.5	0.1				12.91			
		0.5	1			1.24	28.27			

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TABLE 6.17.1.2.10

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
Ti-6Al-4V ELI AT ROOM TEMPERATURE**

ORIENTATION: T-L

ENVIRONMENT: Water Sat JP4

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level ($Ksi\sqrt{in}$)						
				2.5	6.0	10.0	20.0	50.0	100.0	
BA	PLATE	0.	16				6.09			
								117.82		
								104.49		
							13.65			
							13.53			

Ti-6Al-4V ELI

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TABLE 6.17.2.1

TITANIUM Ti-6Al-4V ELI K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic} / \sqrt{A})^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Ksi√in.)	K_{Ic} MEAN	STAN DEV		
ANNEALED	Forging	3.00	R.T.	L-T	117.0	3.996	2.001	CT	2.136	1.31	84.91	83.6	1.3	1976	NC001
		3.00				3.996	2.005	CT	2.143	1.26	83.21			1976	NC001
		3.00				4.000	2.005	CT	2.130	1.24	82.42			1976	NC001
ANNEALED	Forging	3.00	R.T.	T-L	117.0	3.997	2.023	CT	2.157	1.31	84.71	84.3	0.4	1976	NC001
		3.00				3.995	2.020	CT	2.146	1.28	83.89			1976	NC001
		3.00				3.991	2.005	CT	2.126	1.29	84.23			1976	NC001
RECRYSTALLIZE ANNEAL	Plate	3.00	R.T.	L-T	119.0	4.000	2.000	CT	2.072	0.94	73.32	76.1	4.0	1976	NC001
		3.00				4.000	2.000	CT	2.088	1.14	80.66			1976	NC001
		3.00				4.000	1.999	CT	2.117	0.97	74.26			1976	NC001
RECRYSTALLIZE ANNEAL	Plate	3.00	R.T.	T-L	122.0	4.000	2.000	CT	2.069	0.97	76.19	76.8	0.7	1976	NC001
		3.00				4.000	2.000	CT	2.102	0.98	76.61			1976	NC001
		3.00				4.000	2.000	CT	2.094	1.01	77.61			1976	NC001

TABLE 6.17.2.2

TI-6AL-4V (ELI) K _C																			
CONDITION HEAT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K _{app}			K _C			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	INT (in.) 2a _i	FINAL (in.) 2a _f	ONSET (KSI) σ _o	MAX (KSI) σ _{max}	K _{app} (KSI/in.)	K _{app} MEAN	STAN DEV	K _C (KSI/in.)	K _C MEAN	STAN DEV		
BUCKLING OF CHAOK EDGES RESTRAINED																			
ANNEALED	Sheet	0.03	R.T.	L-T	136.0	2.000	0.025	0.490	0.780	104.00	94.77*	127.27*	167.42	167.42	167.42	1964	60578		
						2.000	0.025	0.480	0.800	104.00	93.65*	129.62*							
						2.000	0.025	0.470	0.750	107.00	95.20*	127.37*							
						2.000	0.025	0.480	0.750	104.00	93.65*	123.79*							
						2.000	0.025	0.480	0.700	105.00	94.55*	119.24*							
						4.000	0.025	1.270	1.600	71.80	108.22	126.56*							
						4.000	0.025	1.280	1.700	68.50	103.76	126.31*							
						4.000	0.025	1.260	1.650	70.60	105.88	127.29*							
						4.030	0.025	1.270	1.580	73.60	110.82	125.44*							
						4.040	0.025	1.270	1.650	74.90	112.74*	134.72*							
ANNEALED	Sheet	0.03	R.T.	L-T	136.0	17.990	0.024	5.480	7.950	44.10	137.33	107.2	107.2	107.2	1964	60578			
						17.980	0.025	5.490	7.610	42.10	131.26								
						17.980	0.025	5.490	6.190	49.50	154.33								
						18.000	0.025	5.500	6.980	41.60	129.83								
						18.000	0.025	5.490	7.350	41.60	129.68								
						17.990	0.024	5.480	7.950	44.10	137.33								
						17.980	0.025	5.490	7.610	42.10	131.26								
						17.980	0.025	5.490	6.190	49.50	154.33								
						18.000	0.025	5.500	6.980	41.60	129.83								
						18.000	0.025	5.490	7.350	41.60	129.68								
ANNEALED	Sheet	0.03	R.T.	L-T	136.0	17.990	0.024	5.480	7.950	44.10	137.33	136.5	136.5	136.5	1964	60578			
						17.980	0.025	5.490	7.610	42.10	131.26								
						17.980	0.025	5.490	6.190	49.50	154.33								
						18.000	0.025	5.500	6.980	41.60	129.83								
						18.000	0.025	5.490	7.350	41.60	129.68								
						17.990	0.024	5.480	7.950	44.10	137.33								
						17.980	0.025	5.490	7.610	42.10	131.26								
						17.980	0.025	5.490	6.190	49.50	154.33								
						18.000	0.025	5.500	6.980	41.60	129.83								
						18.000	0.025	5.490	7.350	41.60	129.68								

* NOTE: NET SECTION STRESS EXCEEDS 80% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

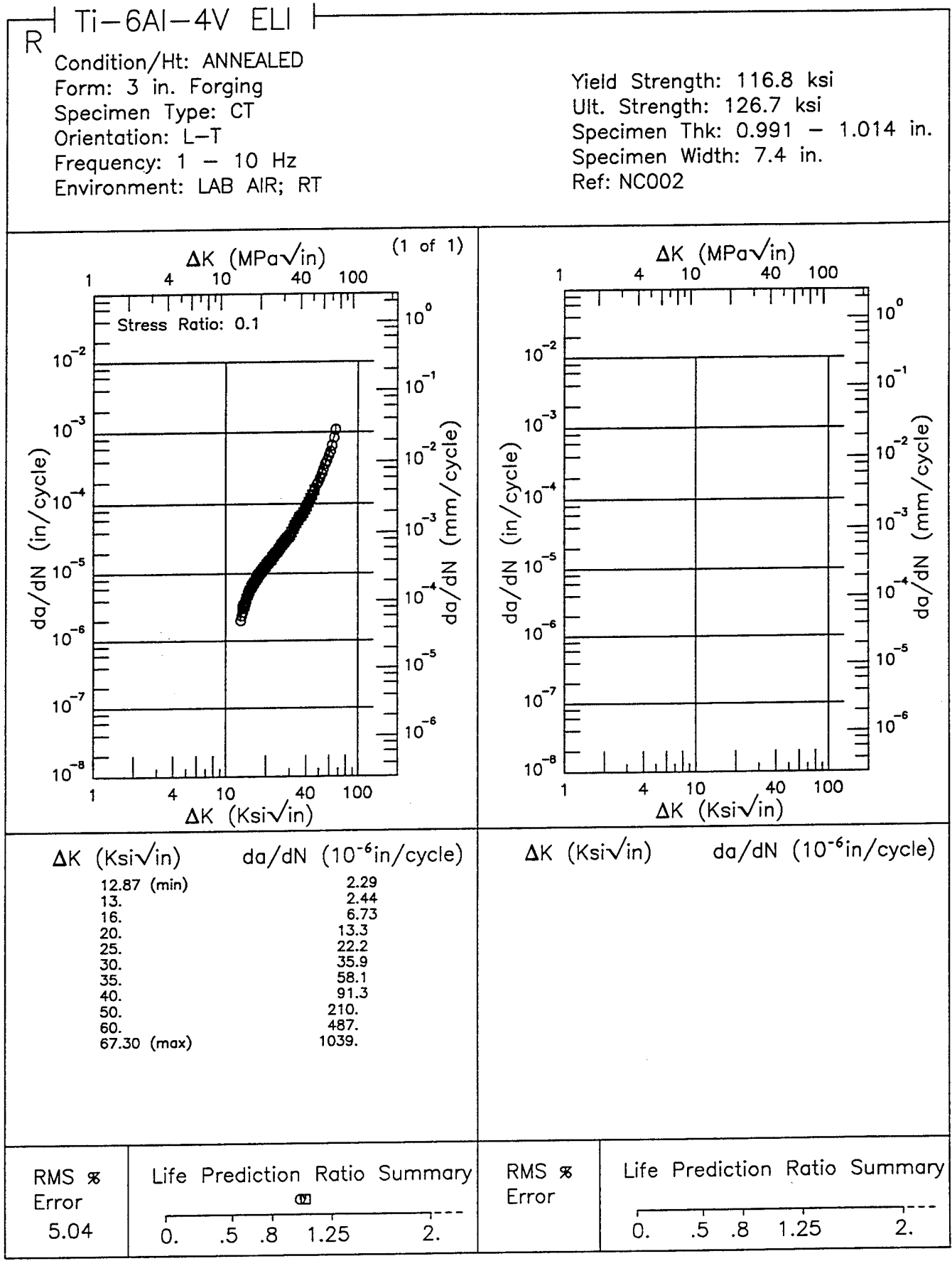
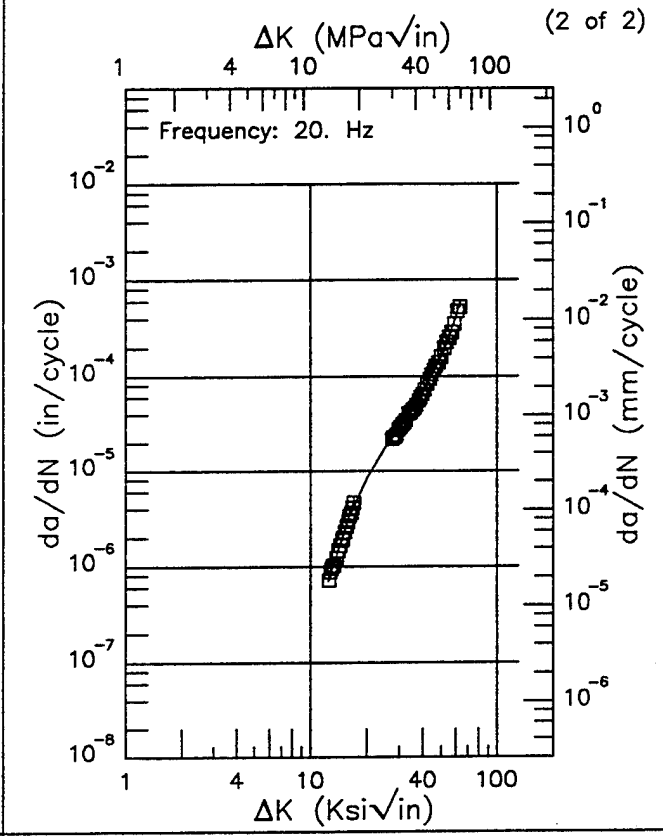
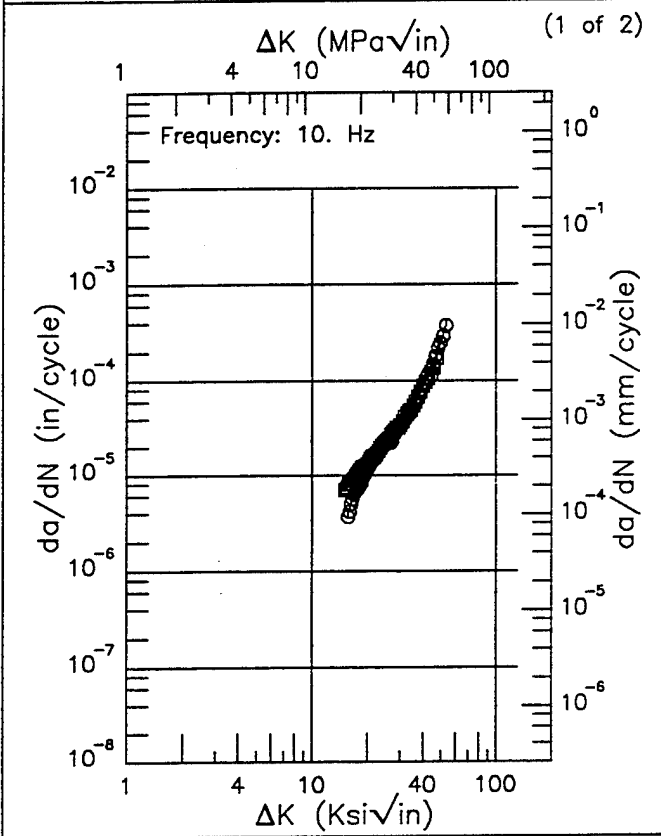


Figure 6.17.3.1.1

Ti-6Al-4V ELI F

Condition/Ht: ANNEALED
 Form: 3 in. Forging
 Specimen Type: CT
 Orientation: T-L
 Stress Ratio: 0.1
 Environment: LAB AIR; RT

Yield Strength: 116.9 ksi
 Ult. Strength: 127 ksi
 Specimen Thk: 0.999 - 1.002 in.
 Specimen Width: 7.4 in.
 Ref: NC002



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
15.12 (min)	5.71
16.	6.57
20.	12.2
25.	22.1
30.	33.4
35.	51.4
40.	85.8
50.	249.
53.22 (max)	377.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
12.55 (min)	0.700
13.	0.903
16.	3.15
20.	8.34
25.	17.3
30.	28.7
35.	44.3
40.	67.1
50.	159.
60.	408.
63.28 (max)	567.

RMS $\%$ Error 12.15	Life Prediction Ratio Summary
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RMS $\%$ Error 4.28	Life Prediction Ratio Summary
------------------------	-----------------------------------

Figure 6.17.3.1.2

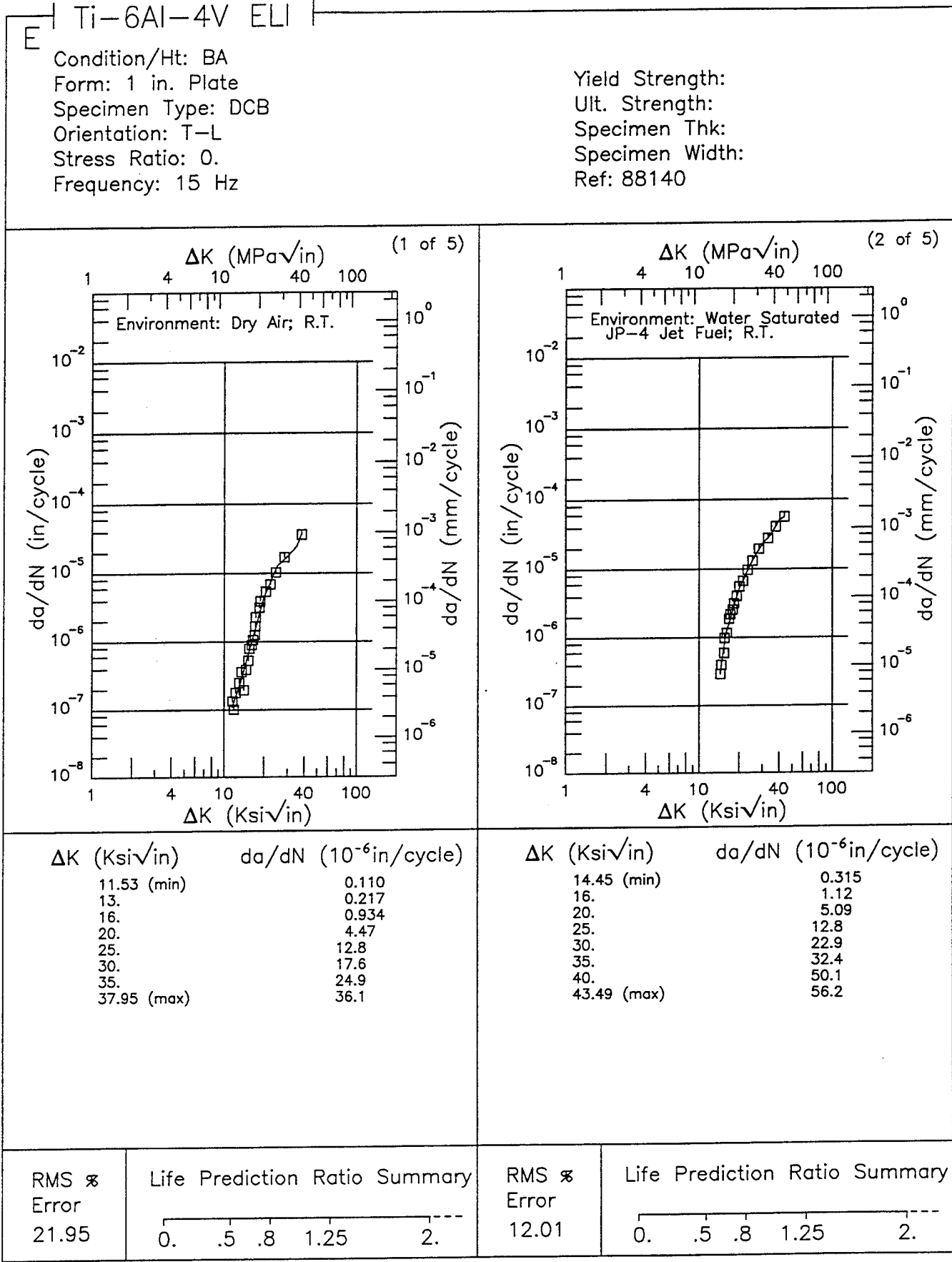
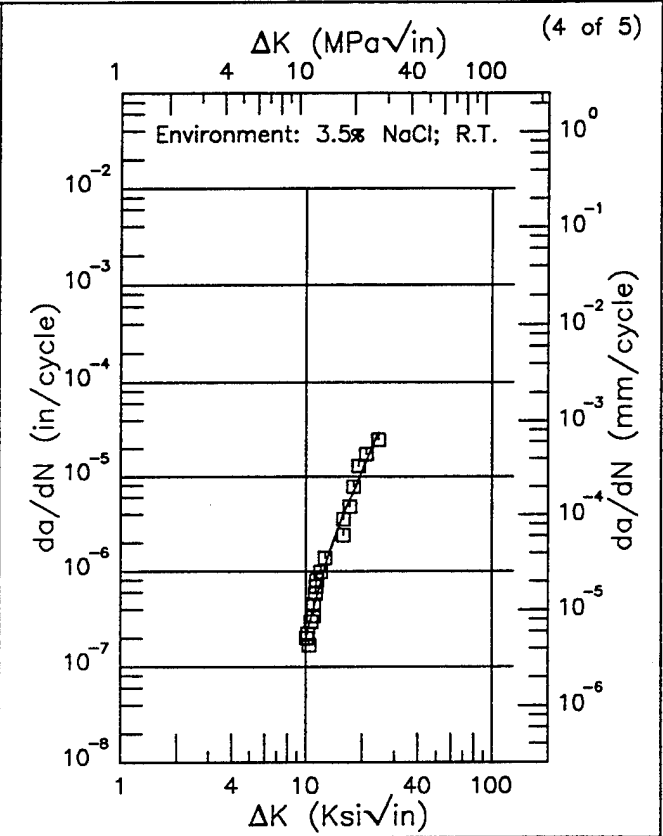
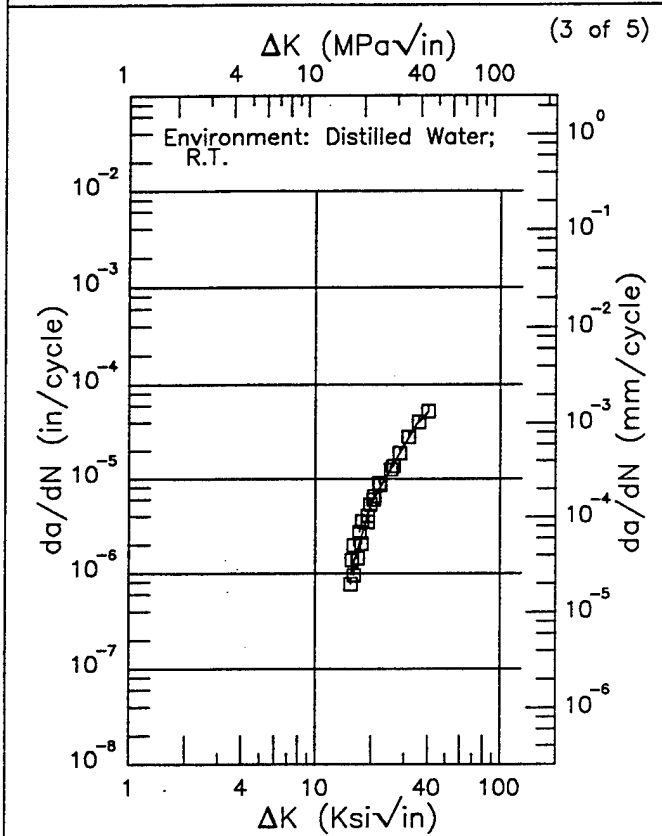


Figure 6.17.3.1.3

Ti-6Al-4V ELI E

Condition/Ht: BA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.
 Frequency: 15 Hz

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
15.47 (min)	0.959
16.	1.24
20.	5.45
25.	12.2
30.	23.0
35.	38.5
40.	51.4
40.10 (max)	51.9

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
10.07 (min)	0.227
13.	1.34
16.	4.02
20.	11.3
24.47 (max)	29.9

RMS % Error 20.50	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
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RMS % Error 26.12	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

Figure 6.17.3.1.3 (Continued)

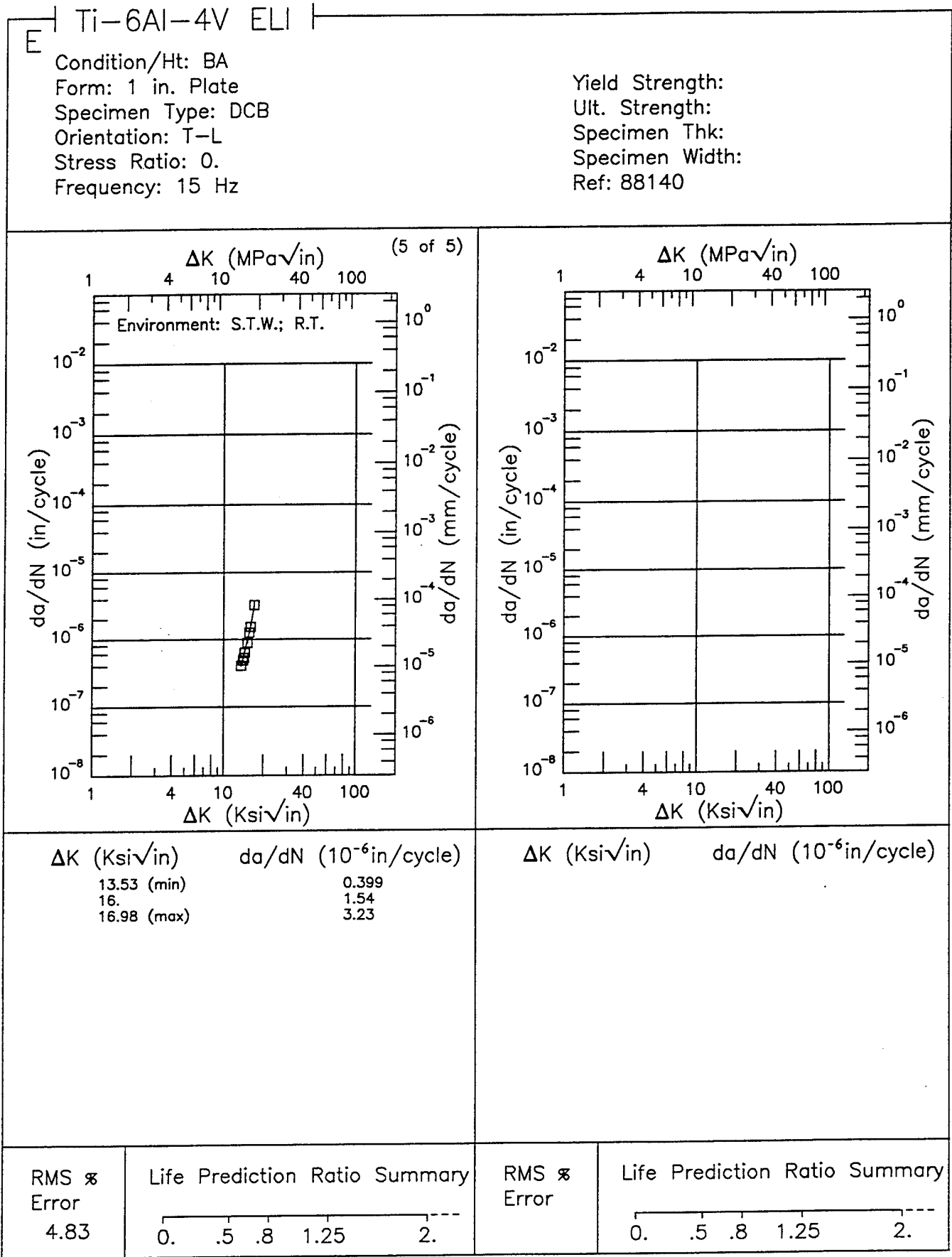
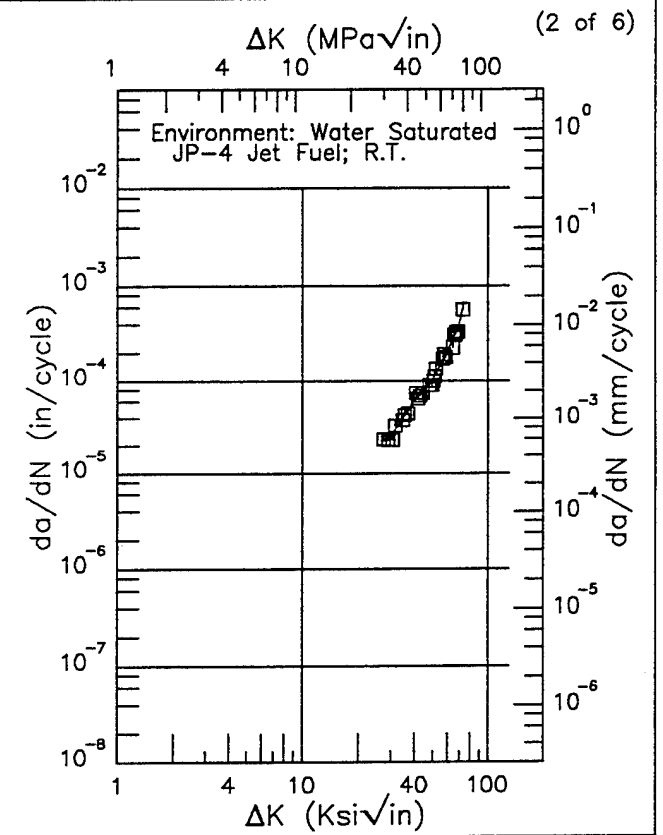
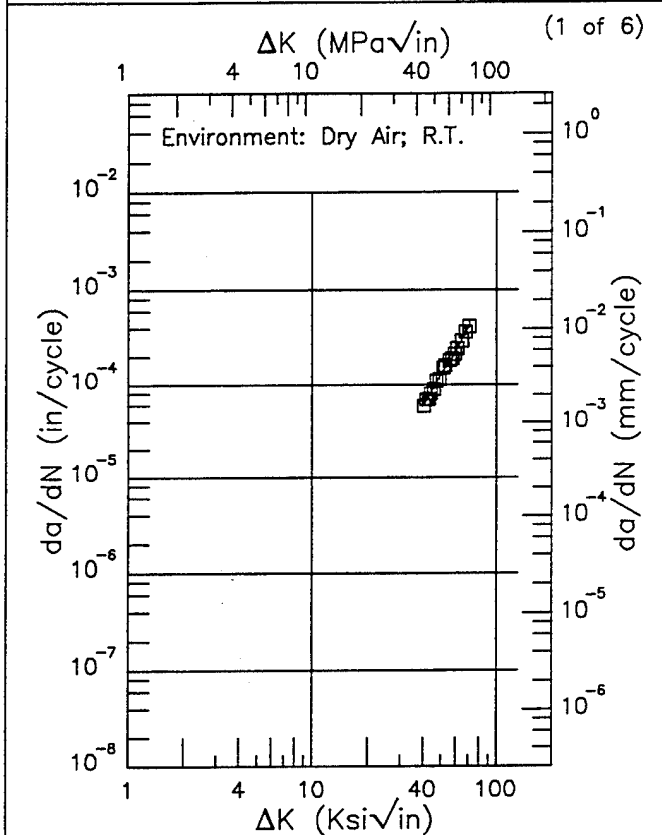


Figure 6.17.3.1.3 (Concluded)

Ti-6Al-4V ELI E

Condition/Ht: BA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.1
 Frequency: 1 Hz

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
40.32 (min)	61.2
50.	131.
60.	221.
70.	407.
71.02 (max)	414.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
27.33 (min)	22.4
30.	25.1
35.	40.1
40.	60.8
50.	104.
60.	203.
70.	399.
72.48 (max)	571.

RMS % Error	Life Prediction Ratio Summary
3.90	0. .5 .8 1.25 2.

RMS % Error	Life Prediction Ratio Summary
8.12	0. .5 .8 1.25 2.

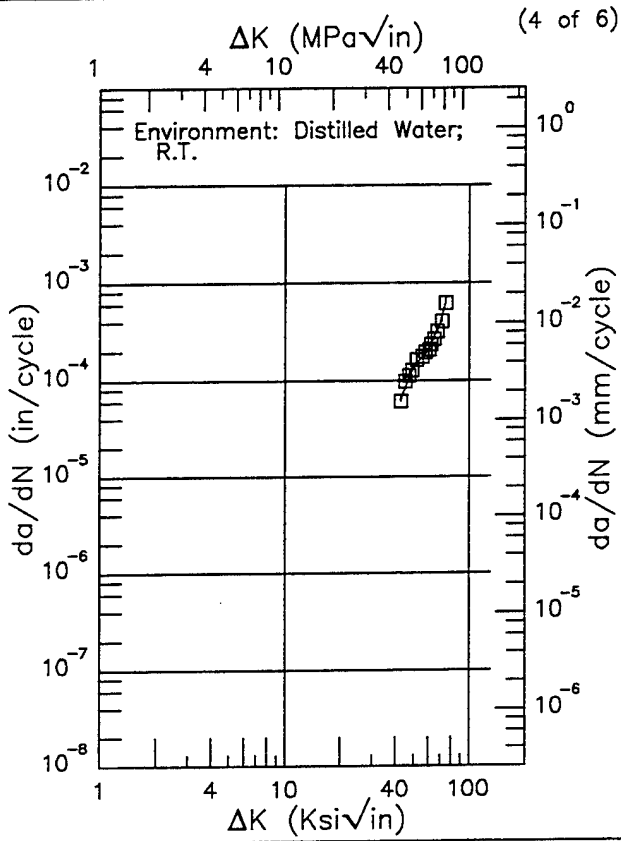
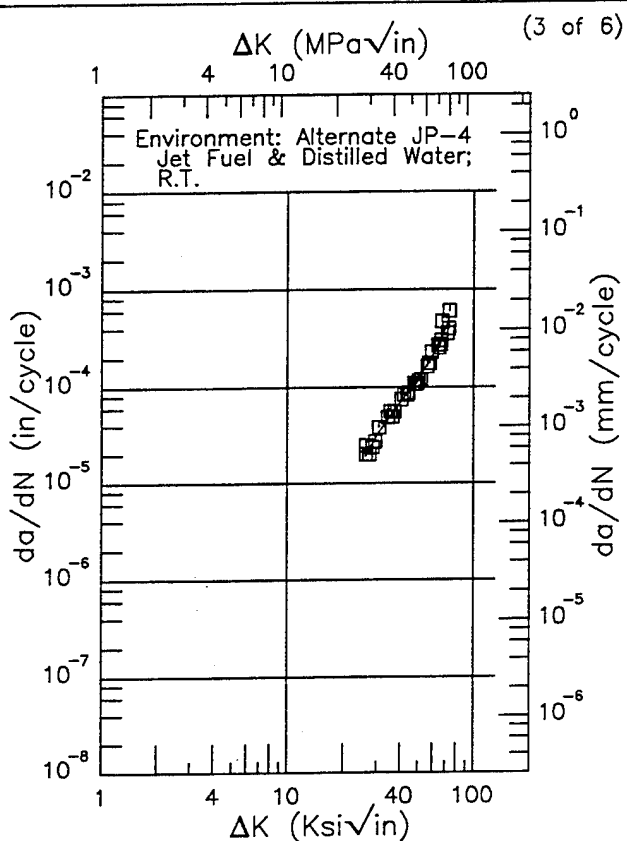
Figure 6.17.3.1.4

Ti-6Al-4V ELI

E

Condition/Ht: BA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.1
 Frequency: 1 Hz

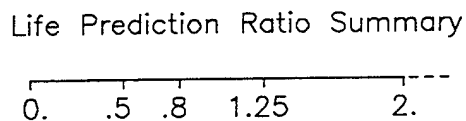
Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
26.24 (min)	19.2
30.	31.1
35.	48.9
40.	69.0
50.	122.
60.	212.
70.	388.
73.62 (max)	489.

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
42.57 (min)	63.1
50.	141.
60.	210.
70.	383.
74.51 (max)	607.

RMS % Error
12.68



RMS % Error
6.05

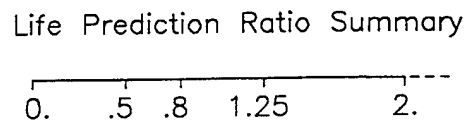
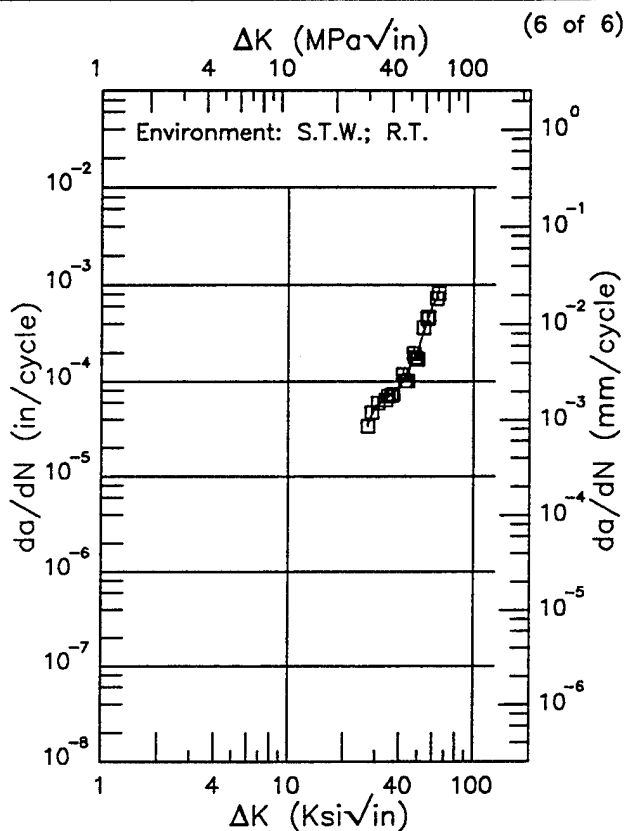
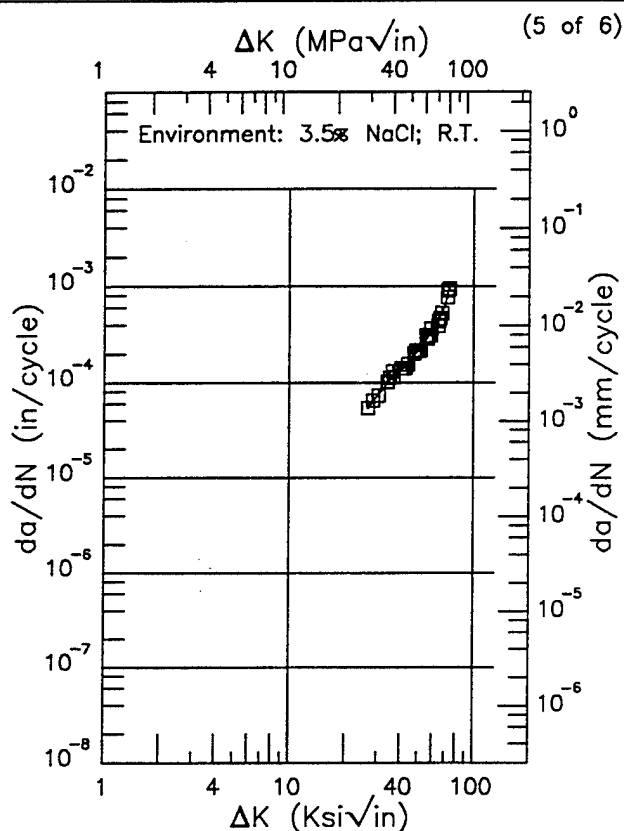


Figure 6.17.3.1.4 (Continued)

Ti-6Al-4V ELI E

Condition/Ht: BA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.1
 Frequency: 1 Hz

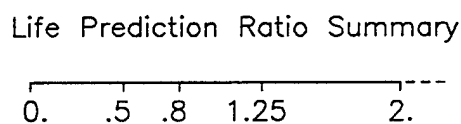
Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
26.53 (min)	53.1
30.	73.9
35.	107.
40.	138.
50.	214.
60.	365.
70.	628.
73.43 (max)	1018.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
27.01 (min)	35.5
30.	55.2
35.	69.5
40.	87.7
50.	217.
60.	620.
64.60 (max)	754.

RMS %
 Error
 6.28



RMS %
 Error
 10.25

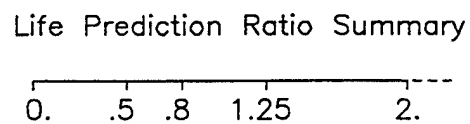


Figure 6.17.3.1.4 (Concluded)

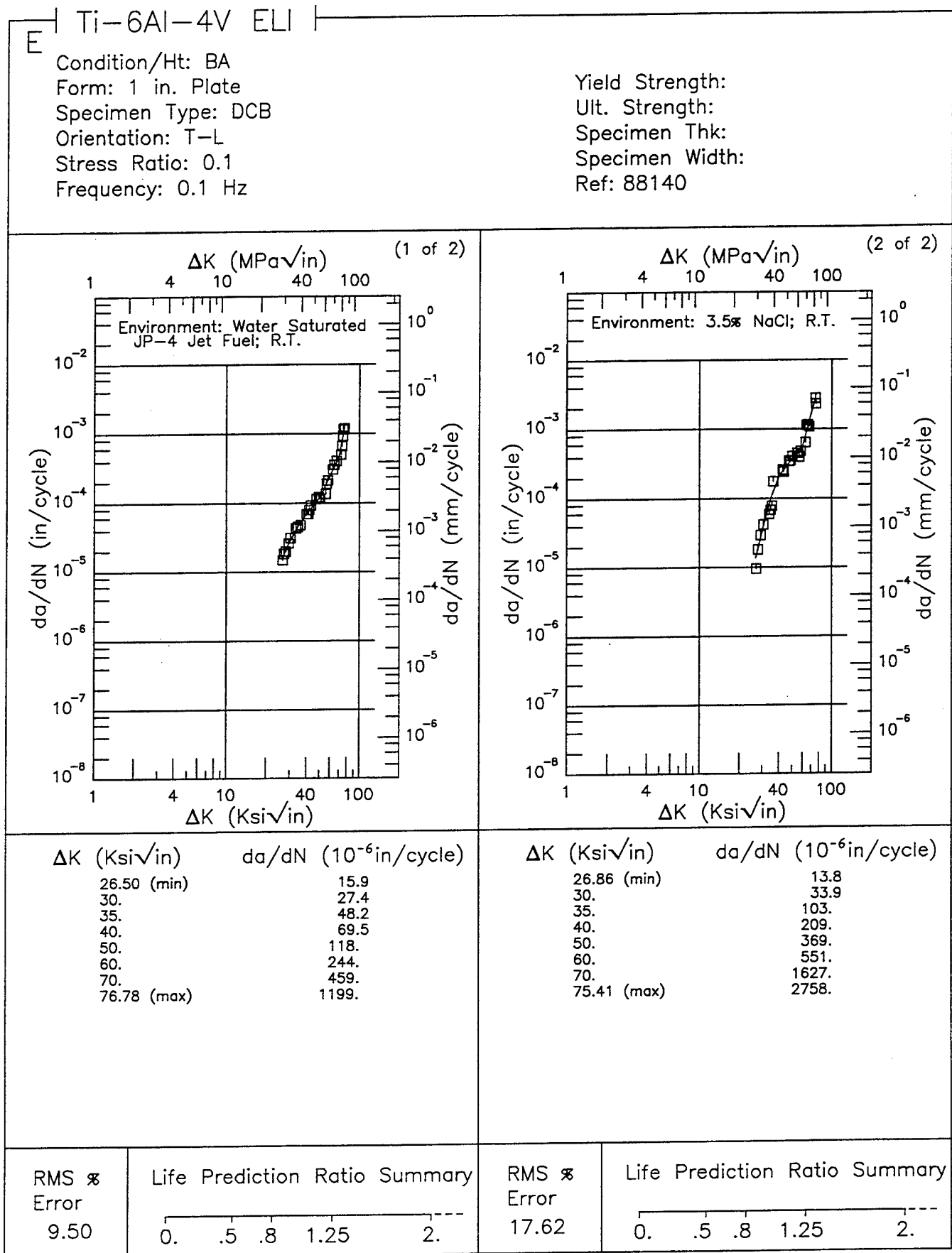


Figure 6.17.3.1.5

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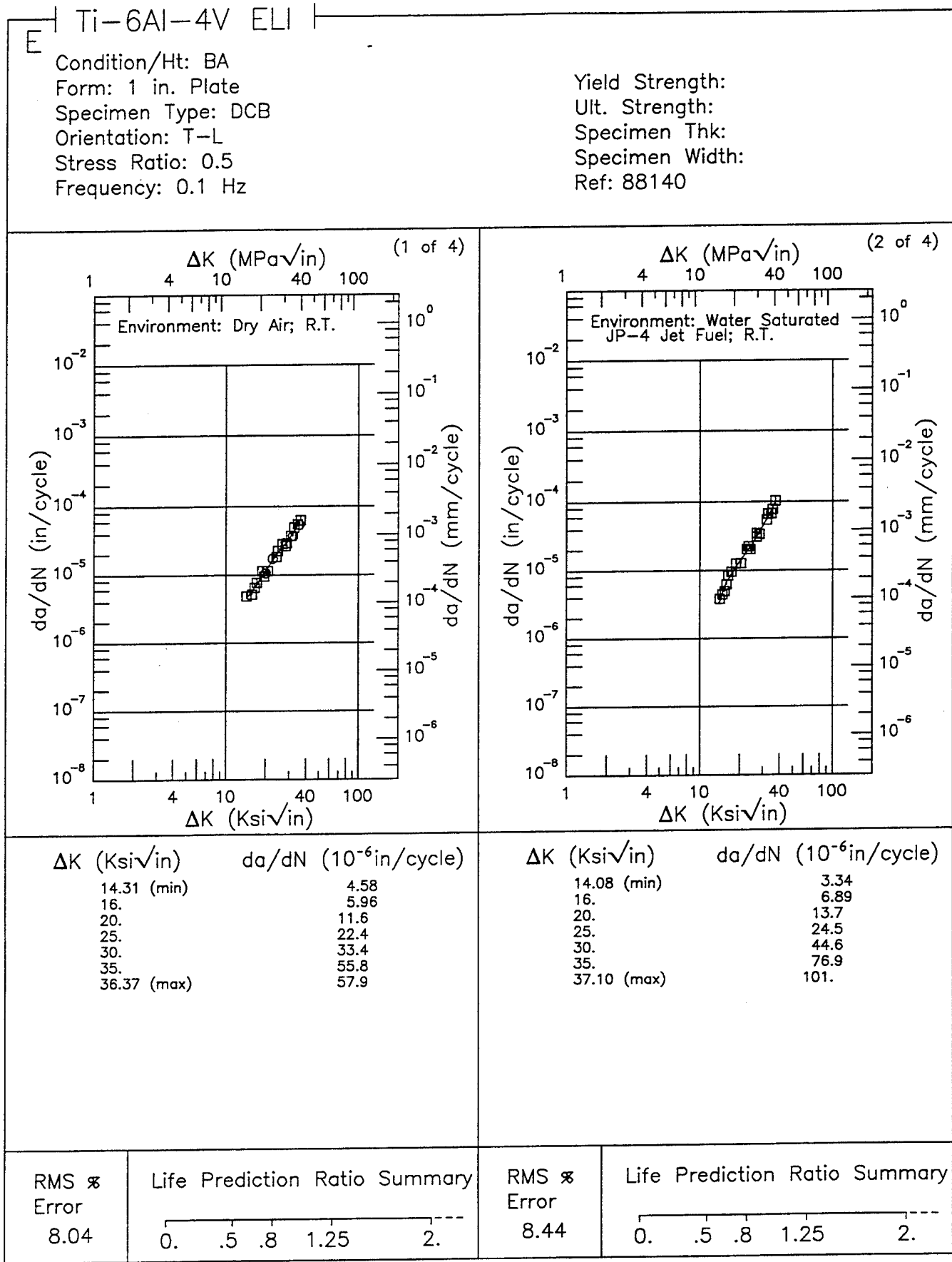
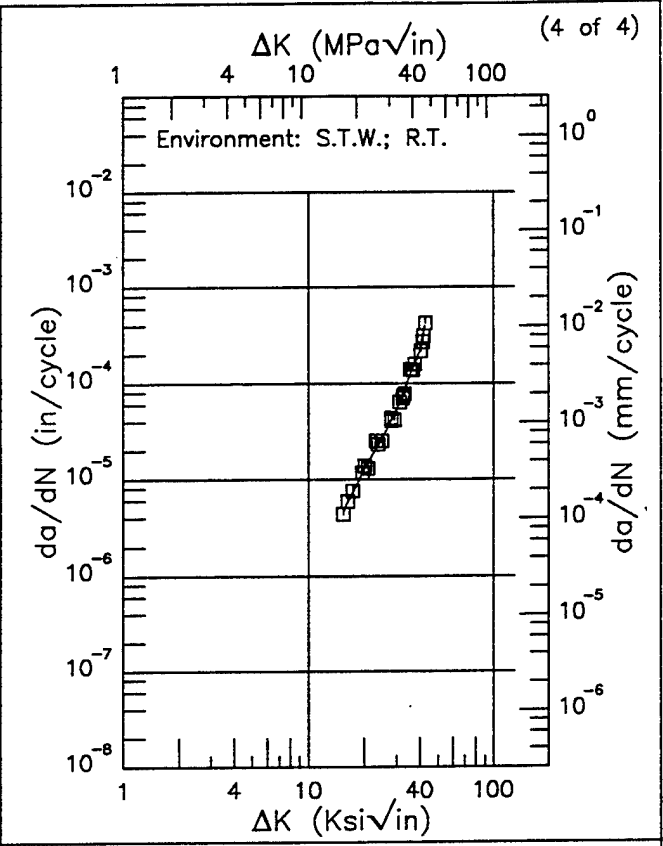
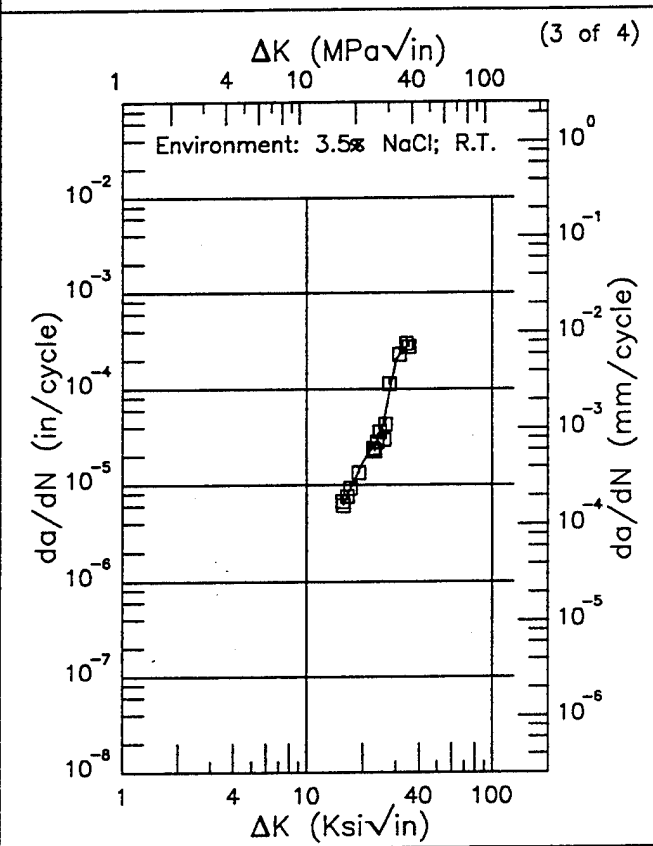


Figure 6.17.3.1.6

Ti-6Al-4V ELI E

Condition/Ht: BA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.5
 Frequency: 0.1 Hz

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
15.54 (min)	6.43
16.	6.87
20.	17.2
25.	34.1
30.	203.
35.	294.
35.28 (max)	305.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
15.38 (min)	4.60
16.	5.38
20.	12.9
25.	26.5
30.	53.2
35.	116.
40.	217.
42.70 (max)	395.

RMS % Error	Life Prediction Ratio Summary
15.51	

RMS % Error	Life Prediction Ratio Summary
7.85	

Figure 6.17.3.1.6 (Concluded)

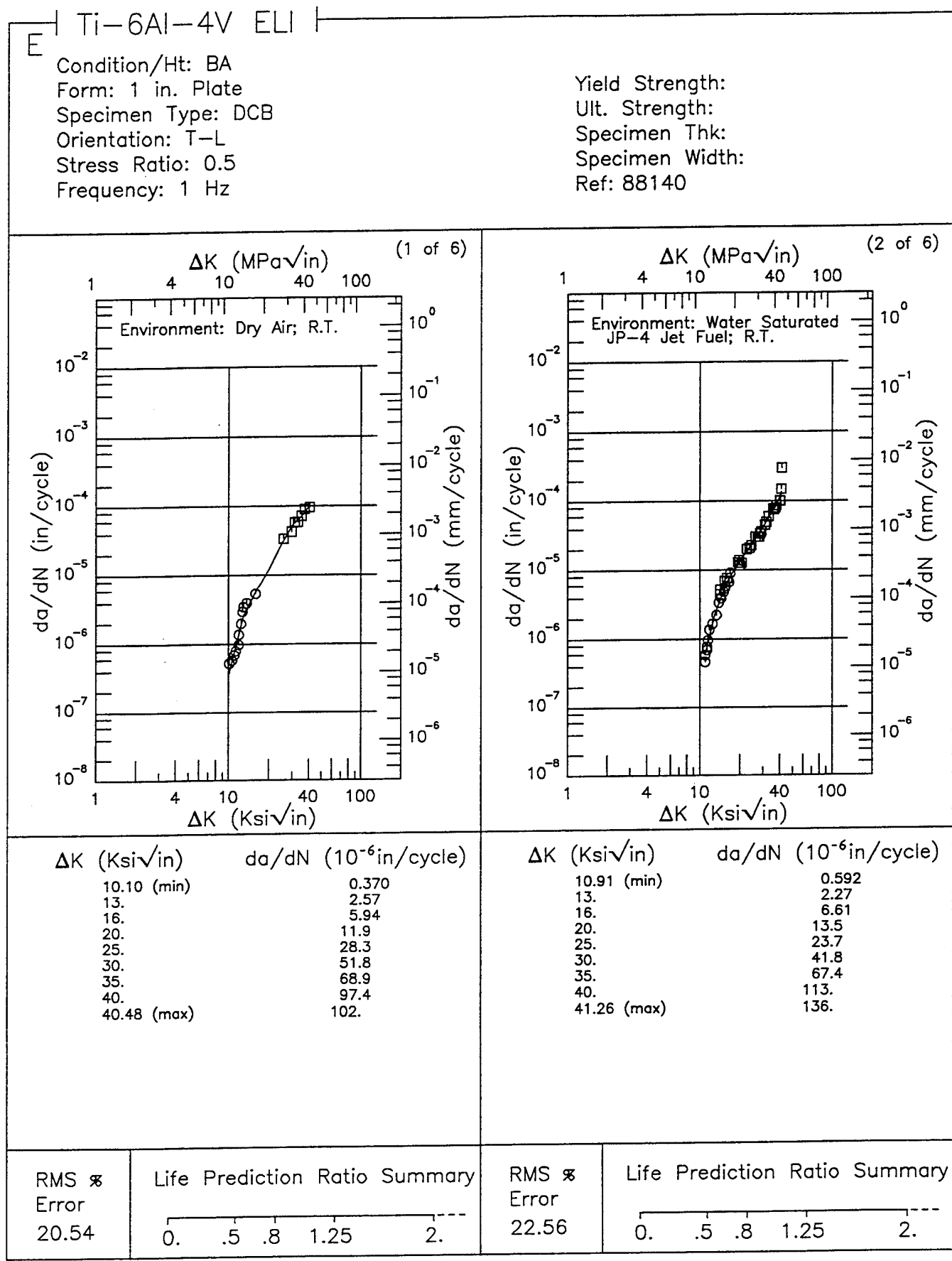
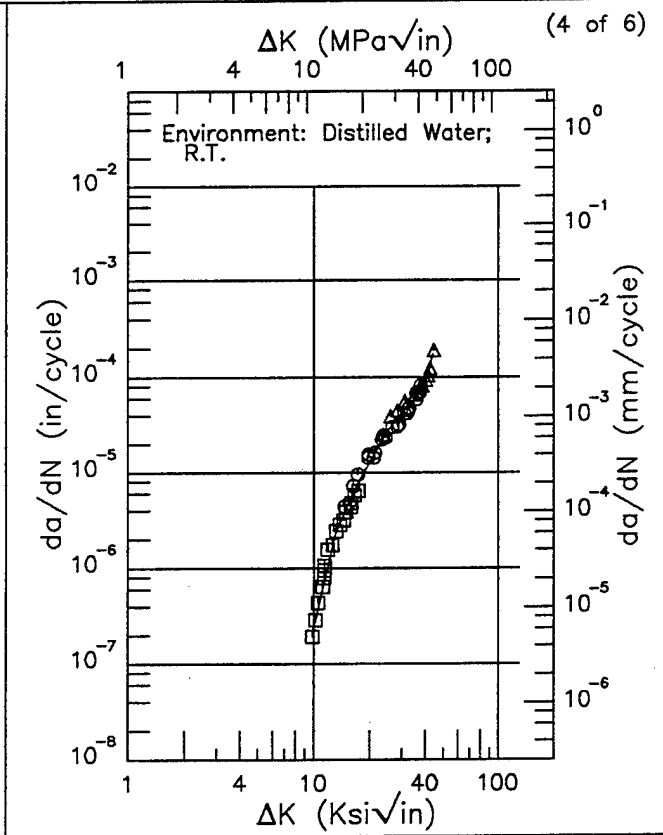
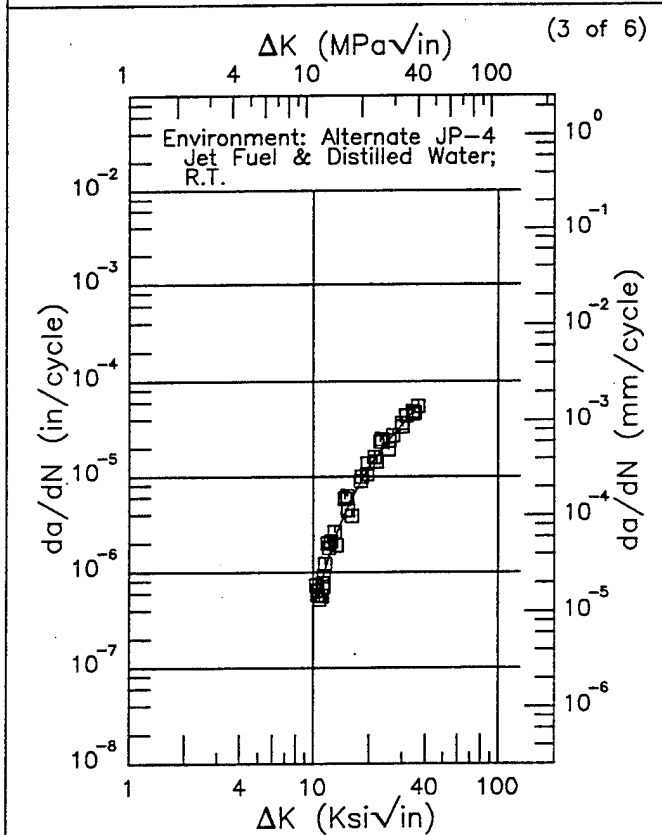


Figure 6.17.3.1.7

Ti-6Al-4V ELI E

Condition/Ht: BA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.5
 Frequency: 1 Hz

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
10.34 (min)	0.502
13.	2.19
16.	5.87
20.	12.9
25.	23.5
30.	35.5
35.	50.1
36.59 (max)	55.4

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
9.85 (min)	0.208
10.	0.247
13.	2.16
16.	5.72
20.	13.0
25.	27.7
30.	42.1
35.	58.5
40.	95.3
44.21 (max)	173.

RMS % Error 22.73	Life Prediction Ratio Summary 0. 0.5 0.8 1.25 2.
----------------------	---

RMS % Error 16.37	Life Prediction Ratio Summary 0. 0.5 0.8 1.25 2.
----------------------	---

Figure 6.17.3.1.7 (Continued)

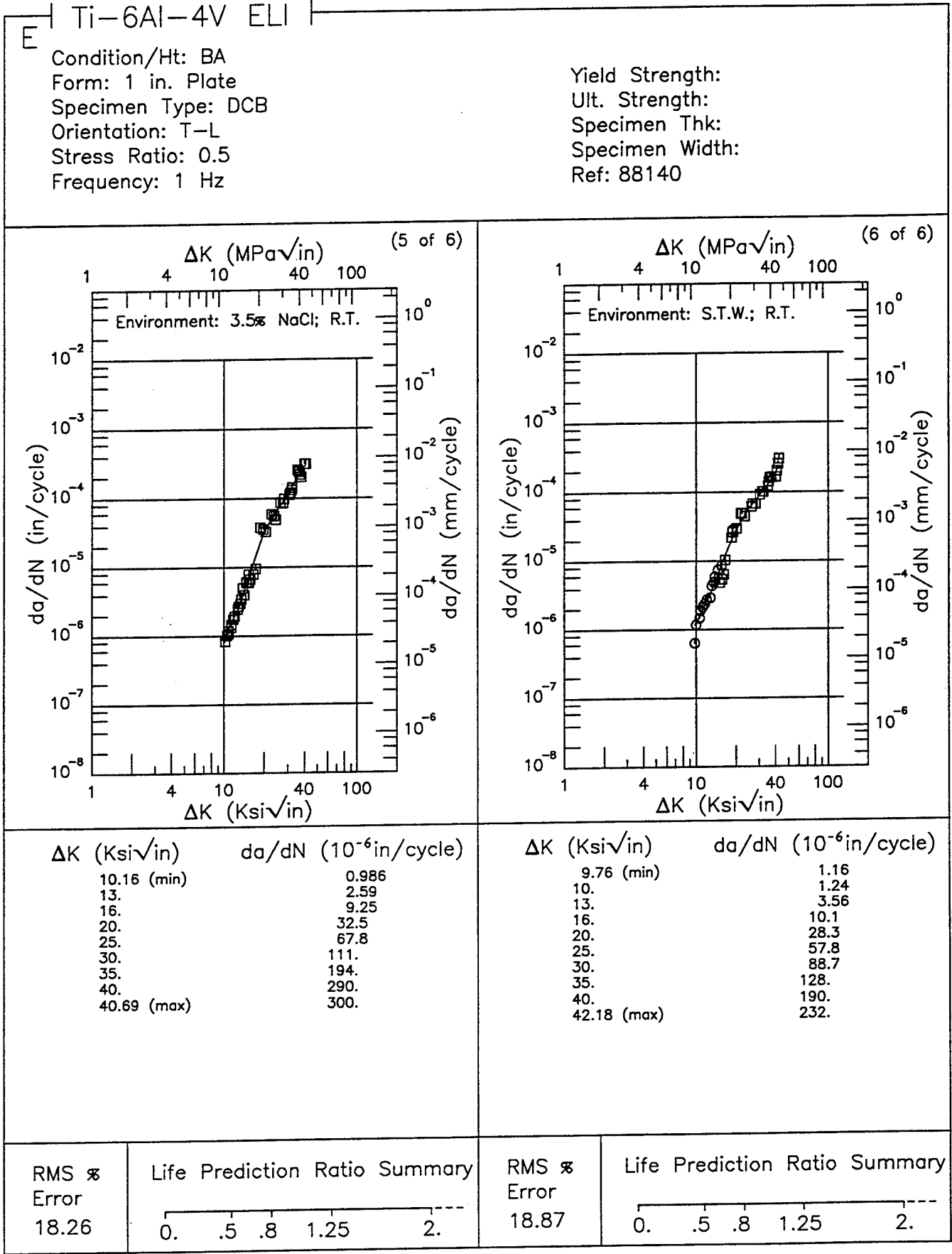


Figure 6.17.3.1.7 (Concluded)

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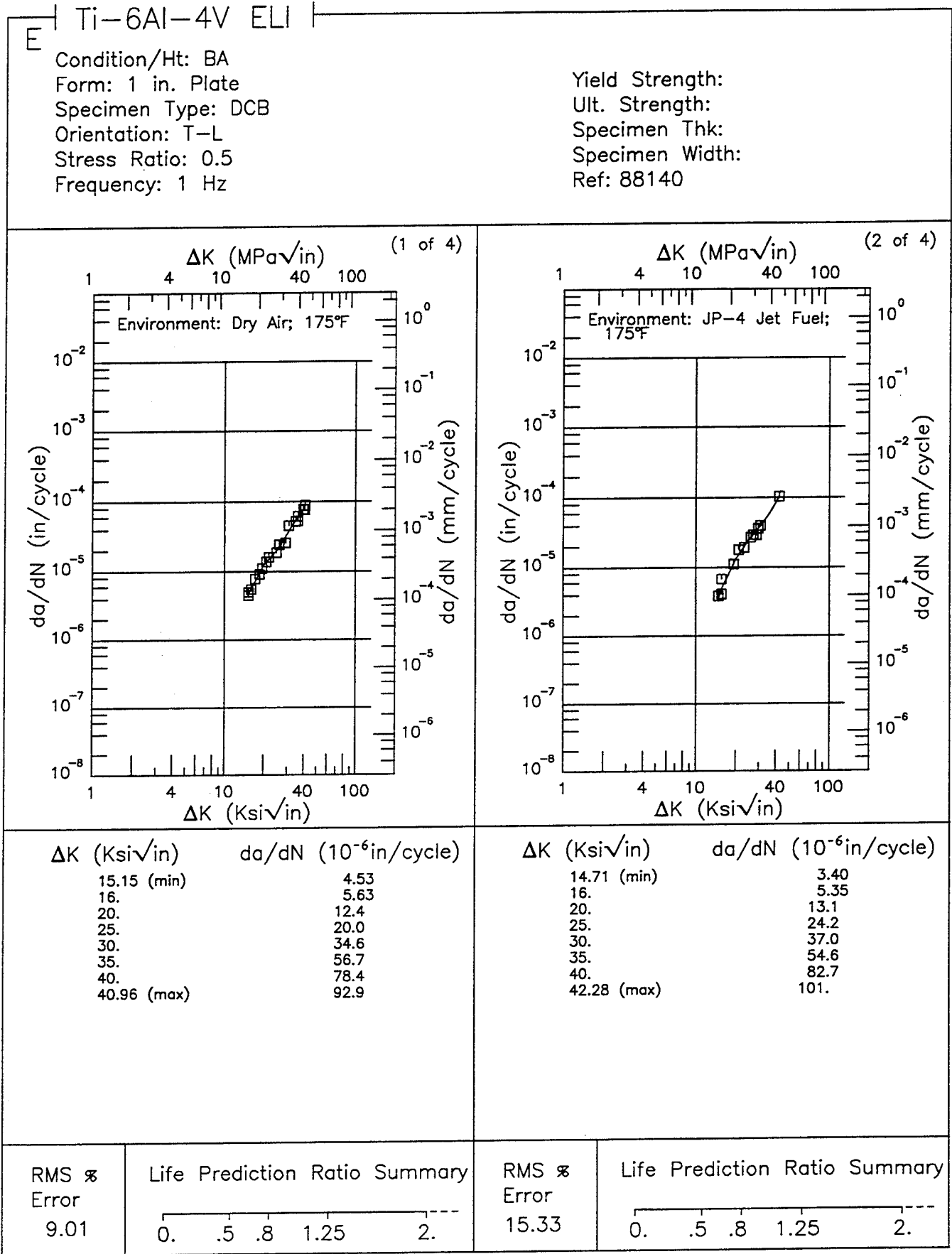
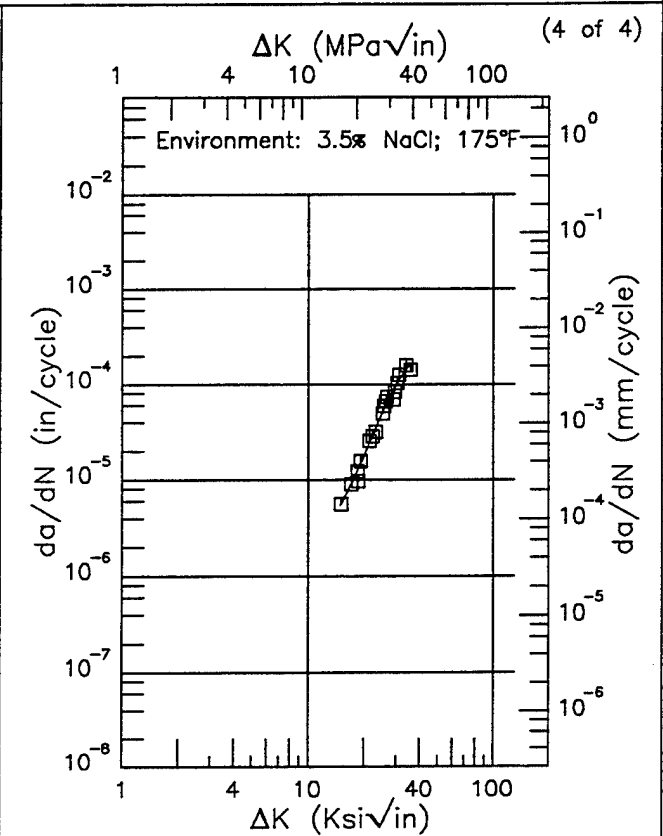
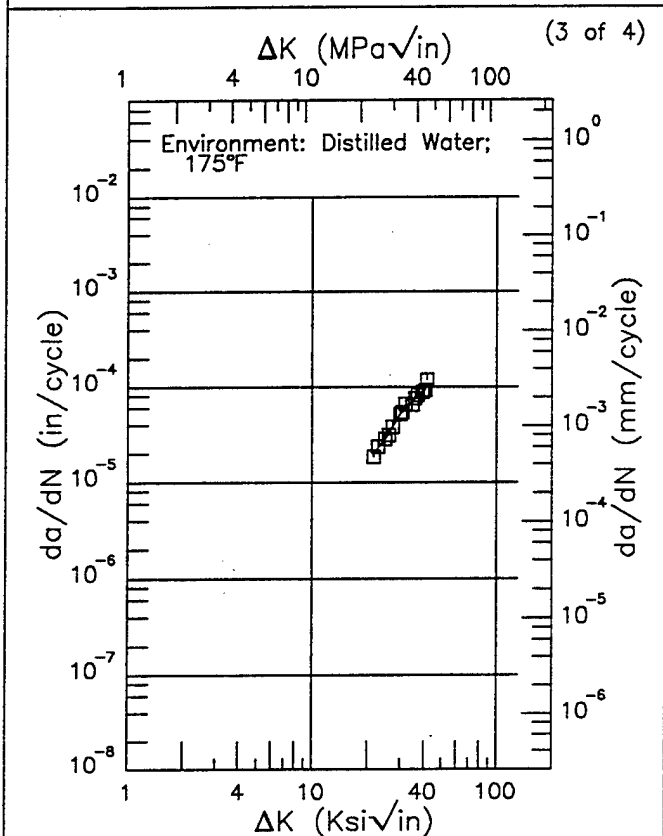


Figure 6.17.3.1.8

Ti-6Al-4V ELI E

Condition/Ht: BA
 Form: 1 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Stress Ratio: 0.5
 Frequency: 1 Hz

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 88140



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
21.49 (min)	20.2
25.	28.9
30.	52.4
35.	71.2
40.	93.3
41.36 (max)	104.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
15.01 (min)	5.51
16.	6.93
20.	17.8
25.	48.6
30.	100.
35.	159.
35.56 (max)	165.

RMS % Error	Life Prediction Ratio Summary
5.93	0. .5 .8 1.25 2.

RMS % Error	Life Prediction Ratio Summary
11.42	0. .5 .8 1.25 2.

Figure 6.17.3.1.8 (Concluded)

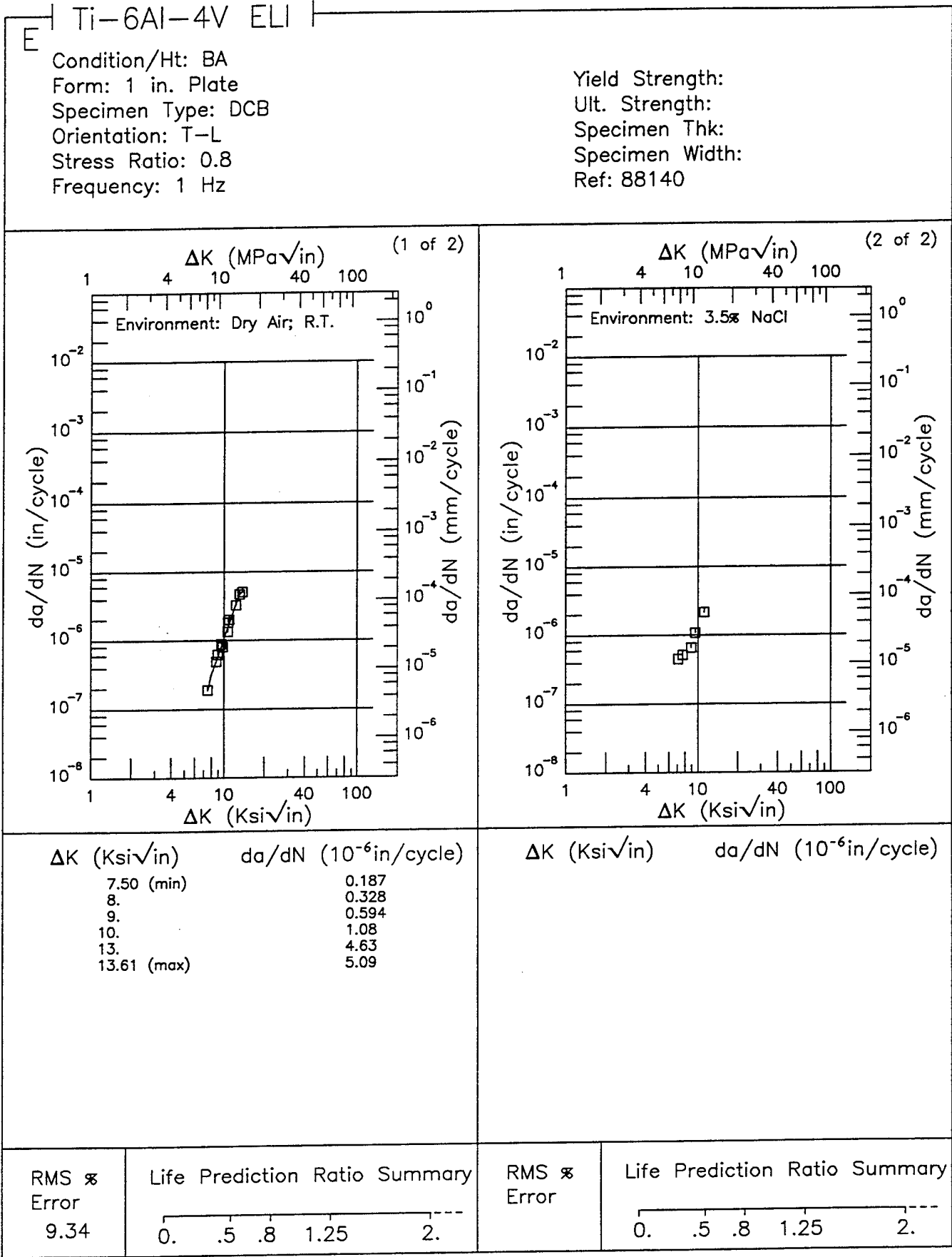


Figure 6.17.3.1.9

Ti-6Al-4V ELI E

Condition/Ht: RA
 Form: 3 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.1
 Frequency: 1 - 10 Hz

Yield Strength: 119.4 ksi
 Ult. Strength: 127.8 ksi
 Specimen Thk: 1.003 - 1.04 in.
 Specimen Width: 7.4 in.
 Ref: NC002

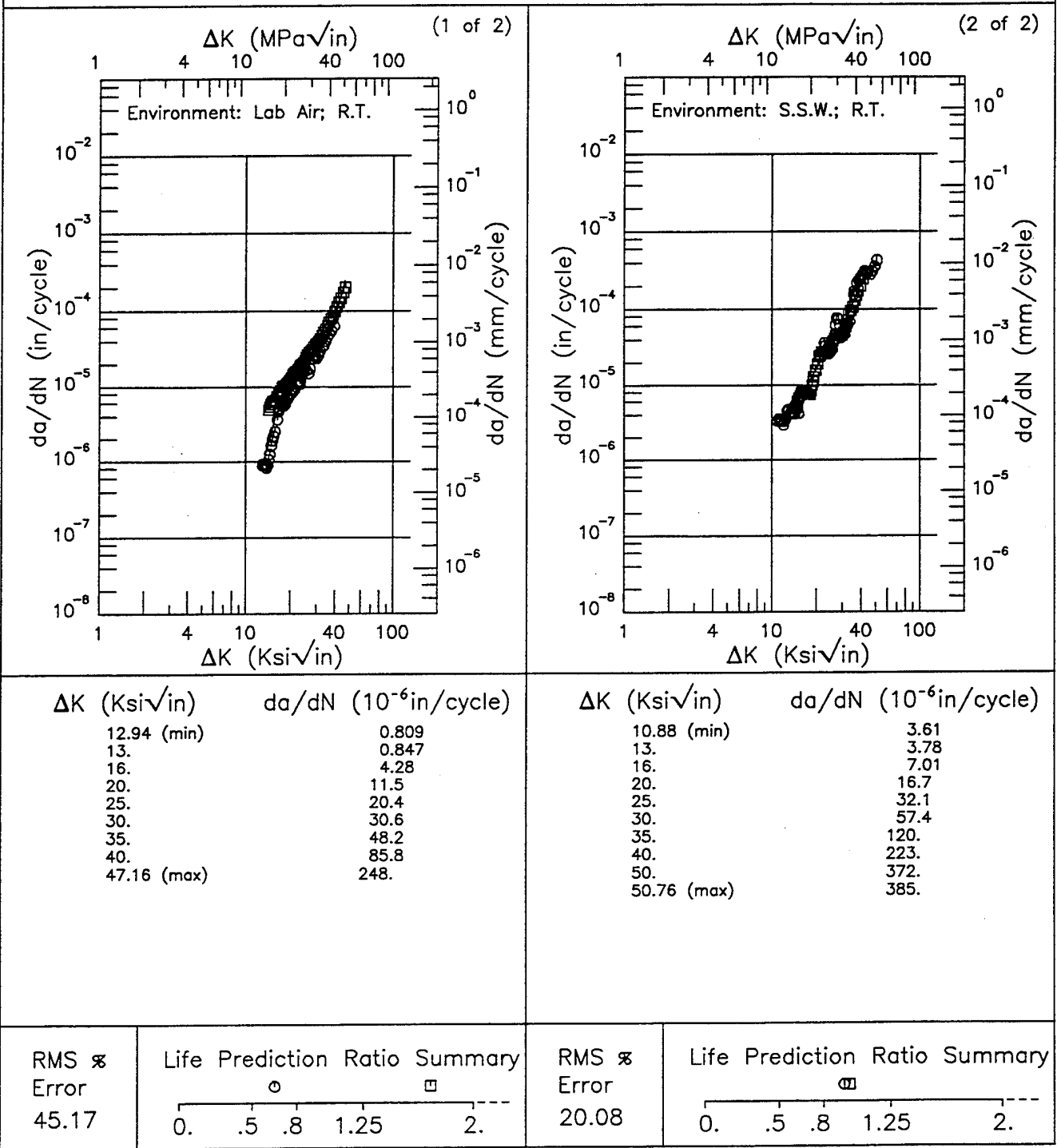
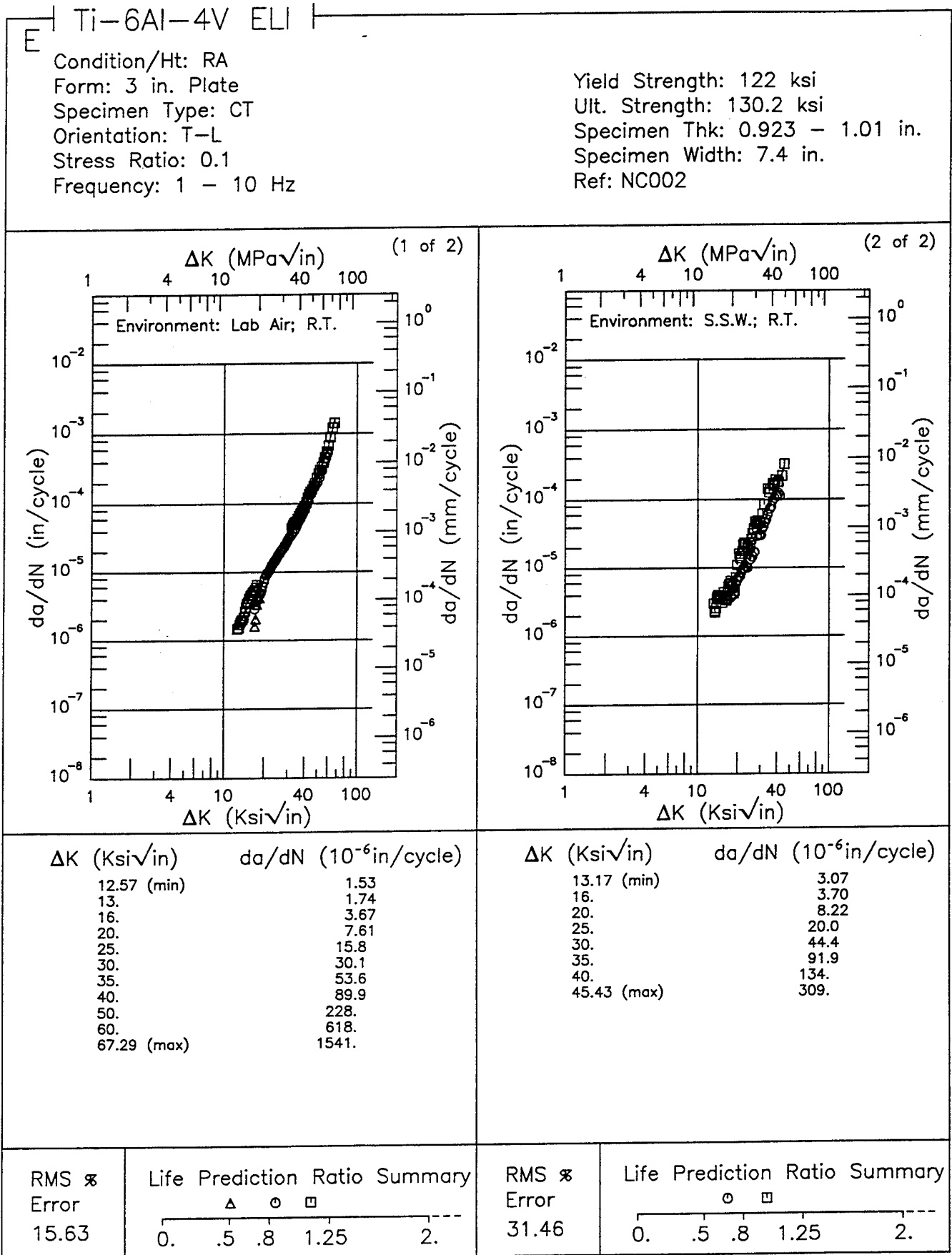


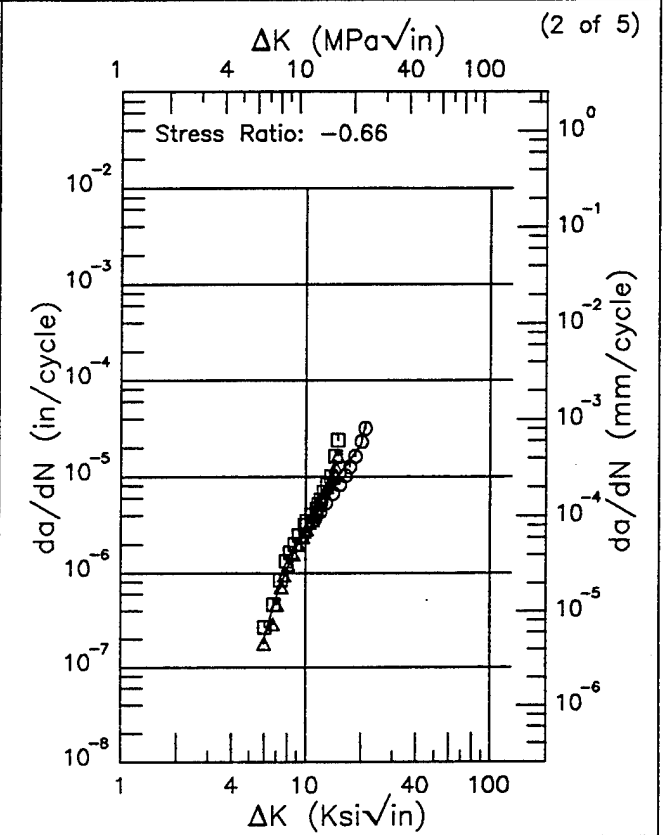
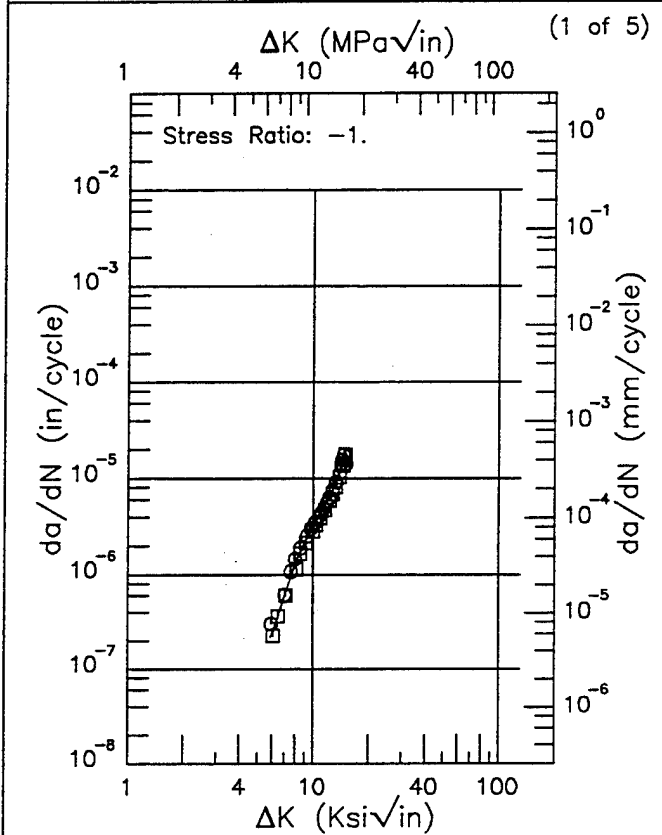
Figure 6.17.3.1.10



Ti-6Al-4V ELI R

Condition/Ht: RA
 Form: 0.5 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 8 Hz
 Environment: LAB AIR; RT

Yield Strength:
 Ult. Strength:
 Specimen Thk: 0.246 - 0.252 in.
 Specimen Width: 3.995 - 4.005 in.
 Ref: NC005



ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
5.92 (min)	0.219
6.	0.234
7.	0.591
8.	1.30
9.	2.19
10.	3.10
13.	8.07
14.97 (max)	18.3

ΔK (Ksi \sqrt{in})	da/dN (10^{-6} in/cycle)
5.97 (min)	0.201
6.	0.208
7.	0.547
8.	1.11
9.	1.92
10.	2.97
13.	7.40
16.	11.7
20.	22.0
21.02 (max)	33.5

RMS $\%$ Error	Life Prediction Ratio Summary
10.56	

RMS $\%$ Error	Life Prediction Ratio Summary
27.47	

Figure 6.17.3.1.12

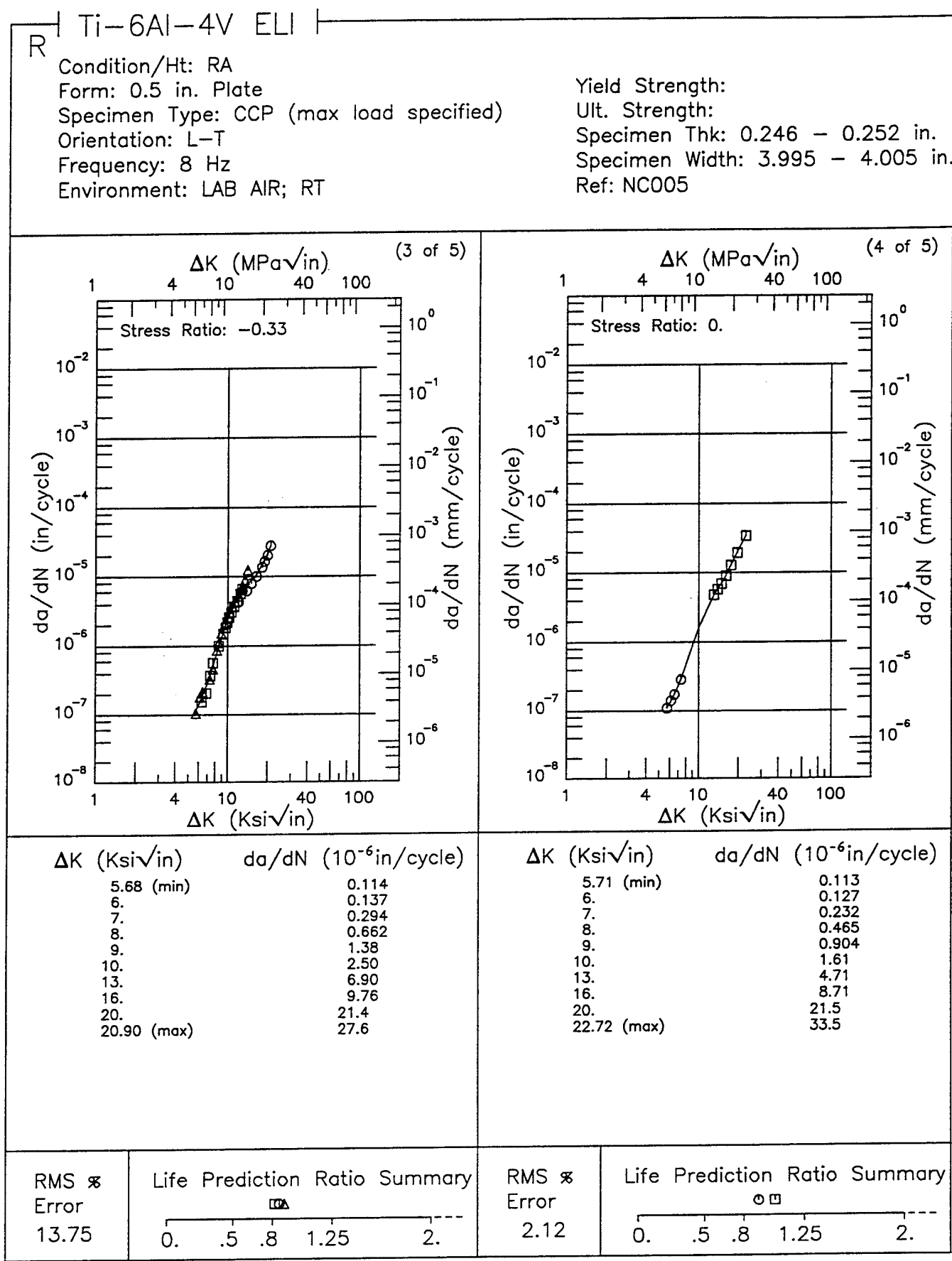


Figure 6.17.3.1.12 (Continued)

Ti-6Al-4V ELI R

Condition/Ht: RA
 Form: 0.5 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 8 Hz
 Environment: LAB AIR; RT

Yield Strength:
 Ult. Strength:
 Specimen Thk: 0.246 - 0.252 in.
 Specimen Width: 3.995 - 4.005 in.
 Ref: NC005

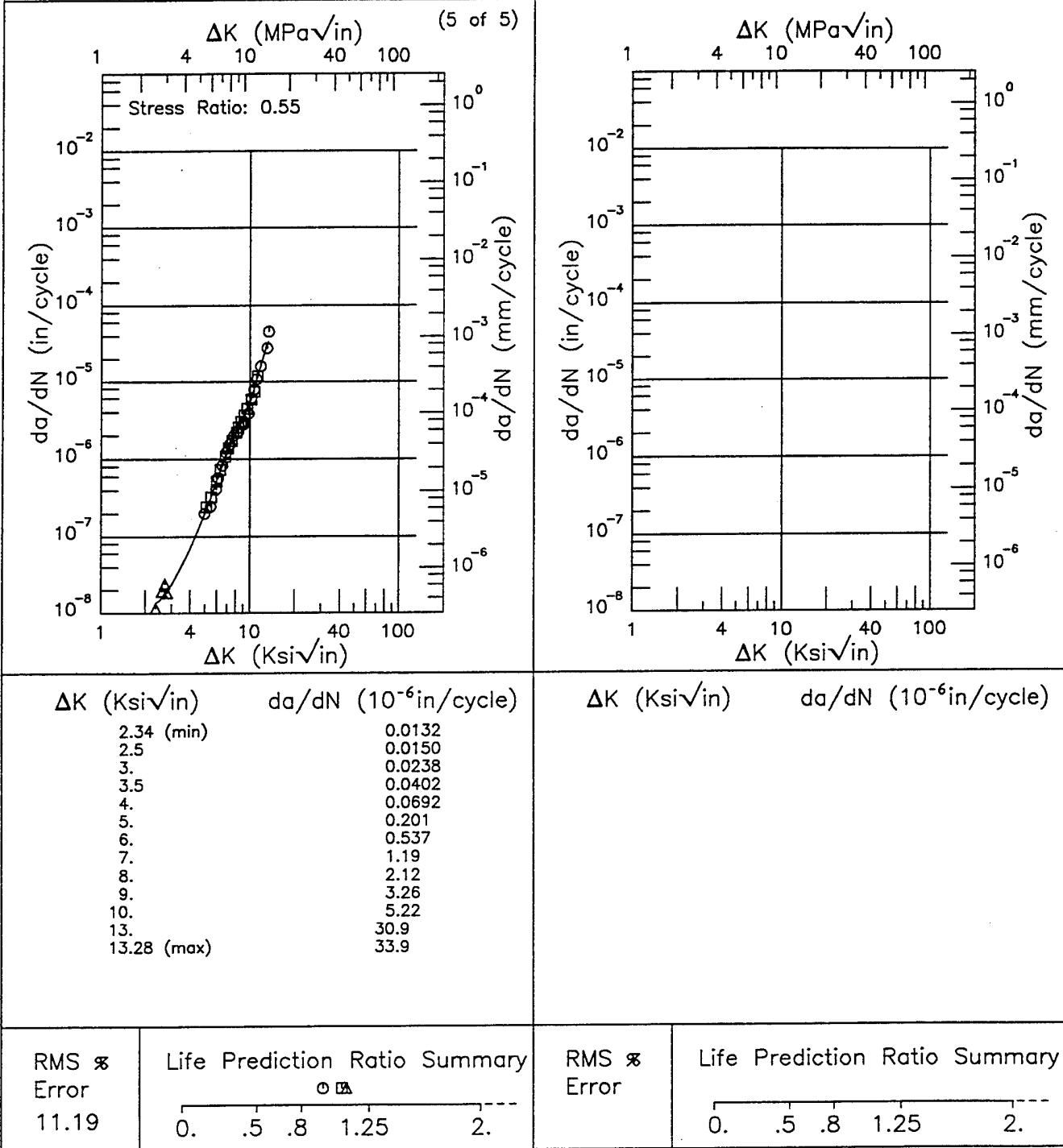
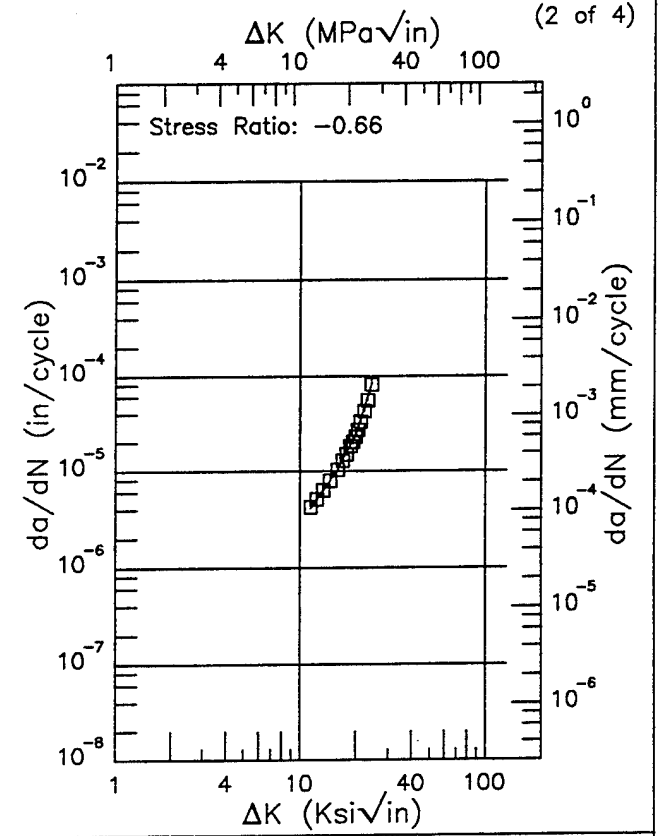
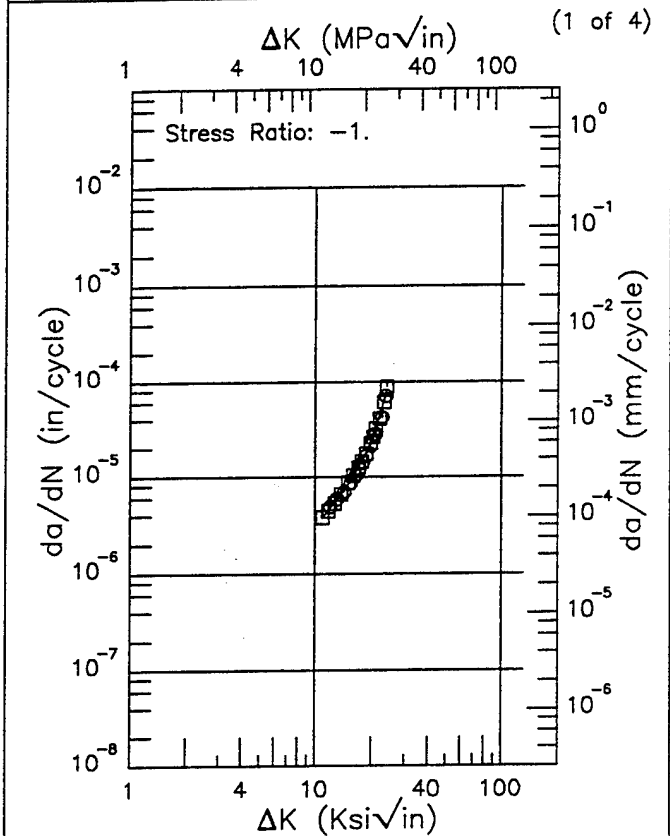


Figure 6.17.3.1.12 (Concluded)

R | Ti-6Al-4V ELI |

Condition/Ht: RA
 Form: 0.5 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 10 Hz
 Environment: LAB AIR; RT

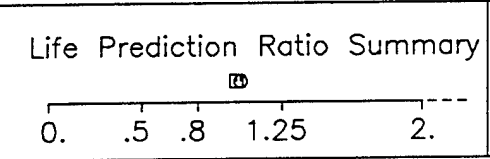
Yield Strength:
 Ult. Strength:
 Specimen Thk: 0.249 - 0.254 in.
 Specimen Width: 3.996 - 4.002 in.
 Ref: NC005



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
10.90 (min)	3.76
13.	5.71
16.	9.95
20.	23.3
23.96 (max)	76.7

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
11.44 (min)	4.11
13.	5.83
16.	10.2
20.	24.0
24.13 (max)	78.1

RMS % Error
6.23



RMS % Error
3.94

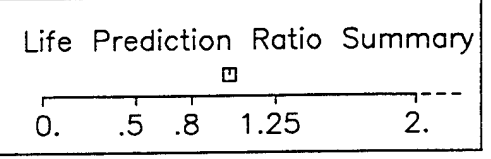
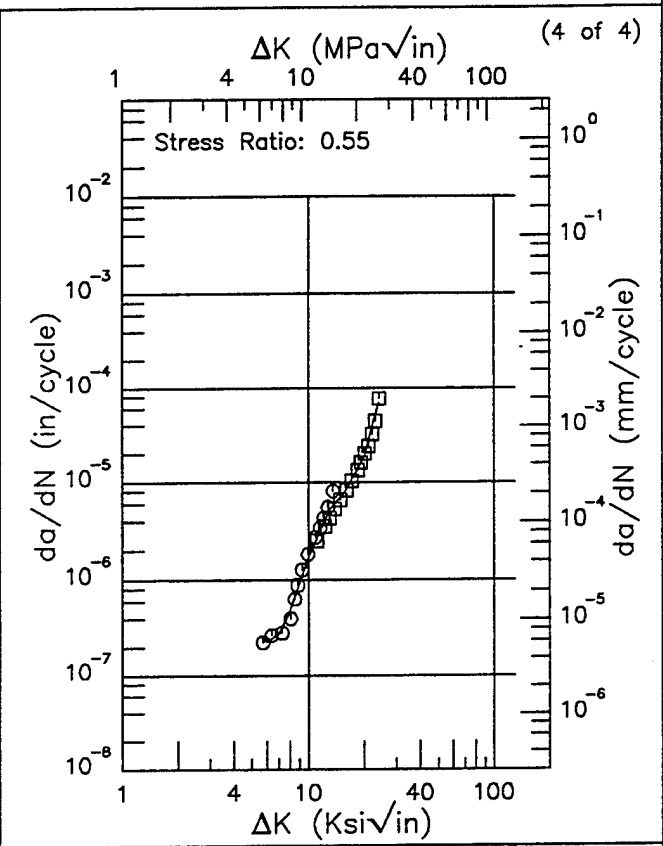
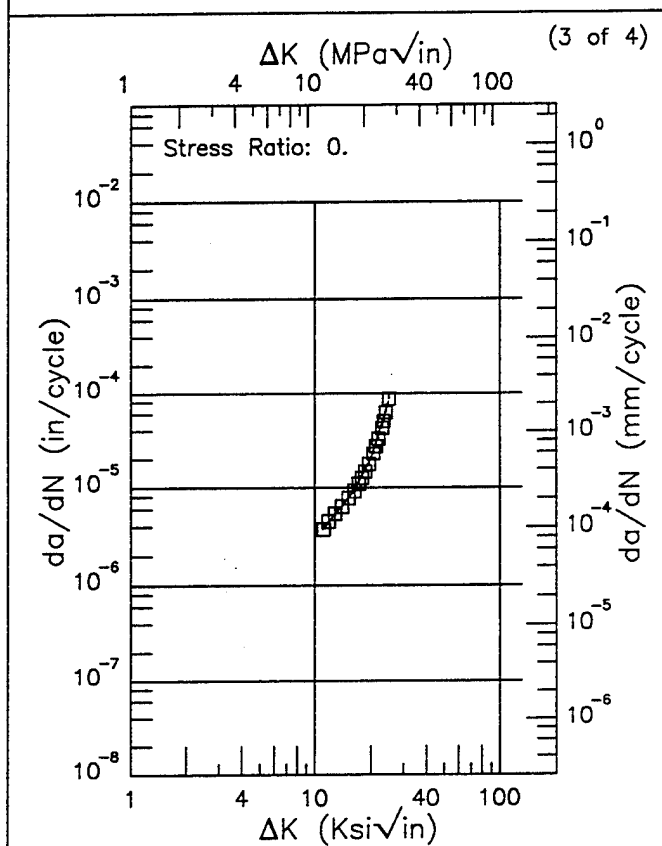


Figure 6.17.3.1.13

Ti-6Al-4V ELI R

Condition/Ht: RA
 Form: 0.5 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 10 Hz
 Environment: LAB AIR; RT

Yield Strength:
 Ult. Strength:
 Specimen Thk: 0.249 - 0.254 in.
 Specimen Width: 3.996 - 4.002 in.
 Ref: NC005



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
11.07 (min)	3.73
13.	5.50
16.	8.98
20.	20.3
24.82 (max)	82.1

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
5.70 (min)	0.249
6.	0.230
7.	0.280
8.	0.488
9.	0.931
10.	1.71
13.	5.25
16.	8.32
20.	18.6
24.00 (max)	69.6

RMS % Error	Life Prediction Ratio Summary
5.19	

RMS % Error	Life Prediction Ratio Summary
13.21	

Figure 6.17.3.1.13 (Concluded)

Ti-6Al-4V ELI

(1 of 1)

TABLE 6.17.3.3
 K_{Isc} SUMMARY FOR TITANIUM ALLOY Ti-6Al-4V (ELI)

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K_Q (Ksi/in)	K_{Isc} (Ksi/in)	Test Time (min)	Test Date	Reference
						Design	Width (in)	Thick (in)							
1800° F 1hr Helium Cool	P	R.T.	T-S	115.2	3.5% NaCl	CANT*	1	0.5	1	---	112	84*	---	1967	70931

* specimen thickness does not meet minimum requirements of $2.5 \left(\frac{K_{Isc}}{\sigma_y} \right)^2$

* asterisk in specimen design column indicates that specimens are side-grooved

TABLE 6.18.1.1

1 of 1

**MEAN PLANE STRAIN FRACTURE TOUGHNESS
FOR TITANIUM ALLOY Ti-6Al-6V-2Sn AT ROOM TEMPERATURE**

Product Form	Condition/Heat Treatment	K_{Ic} ($ksi\sqrt{in}$)											
		Specimen Orientation											
		L-T			T-L			S-L					
Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n		
Plate	BETA ANNEAL 1810F 1HR ARGON COOL	54.3	2.	3		
	BETA ANNEAL & STOA-1800F 0.5HR AC 1575F 0.5HR WQ 1050F 8HR AC	50.1	1.8	2		
	DUPLEX ANNEAL	65.1	2.	3		
	MA	35.	5.2	4		
	STA - 1675F 0.25HR WQ 1100F 4HR AC	34.1	3.8	3		
	STOA - 1700F 1HR WQ 1400F 1HR AC	42.9	1.2	3	46.1	3.1	4		
Forging	MA	58.6	2.7	3		
	STA - 1600F 0.5HR WQ 1000F 6HR AC	30.8	0.7	3		
Billet	MA	52.3	6.4	4		
	MA 1000F 2HR AC	57.1	2.2	2		
	STOA - 1700F 1HR WQ 1400F 1HR AC	62.8	6.9	4	57.	3.7	4		

Ti-6Al-6V-2Sn

1 of 1

TABLE 6.18.1.2.1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-6V-2Sn AT ROOM TEMPERATURE

ORIENTATION: L-S ENVIRONMENT: 3.5% NaCl

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Kst $_{1/2}$ in)					
STOA	PLATE	0.1	10	2.5	5.0	10.0	20.0	50.0	100.0
						2.92	81.72		

1 of 1

TABLE 6.18.1.2.2

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
Ti-6Al-6V-2Sn AT ROOM TEMPERATURE

ENVIRONMENT: H.H.A.

ORIENTATION: L-S

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Kst_i/\bar{m})					
				2.5	6.0	10.0	20.0	50.0	100.0
STOA	PLATE	0.1	0.1			1.94	8.89		
		0.1	10			2.53	7.63		

Ti-6Al-6V-2Sn

1 of 1

TABLE 6.18.1.2.3

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-6V-2Sn AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: 3.5% NaCl

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
STOA	PLATE	0.1	1							
		0.1	20	0.22	3.05	33.78	19.49			

TABLE 6.18.1.2.4

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-6V-2Sn AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: Dry Argon

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)							
				ΔK Level (Ksi \sqrt{in})							
MA	FORGING	0.02	1	2.6	5.0	10.0	20.0	50.0	100.0		
							10	674.14			

Ti-6Al-6V-2Sn

1 of 1

TABLE 6.18.1.2.5

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-6V-2Sn AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: H.H.A.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.6	6.0	10.0	20.0	50.0	100.0
STOA	PLATE	0.1	1			1.64	16.12	394.82	
		0.1	20		0.11	1.7	13.83		

TABLE 6.18.1.2.6

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-6V-2Sn AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	5.0	10.0	20.0	50.0	100.0
MA	EXTRUSION	0.02	0.1-10			0.43	8		
		0.02	20		0.05				

Ti-6Al-6V-2Sn

1 of 1

TABLE 6.18.1.2.7

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-6V-2Sn AT ROOM TEMPERATURE

ENVIRONMENT: 3.5% NaCl

ORIENTATION: T-S

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi \sqrt{in})						
				2.5	5.0	10.0	20.0	50.0	100.0	
BA	PLATE	0.1	0.1							
STOA	PLATE	0.1	1			1.3	100.49			

1 of 1

TABLE 6.18.1.2.8

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
Ti-6Al-6V-2Sn AT ROOM TEMPERATURE

ORIENTATION: T-S

ENVIRONMENT: H.H.A.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	6.0	10.0	20.0	50.0	100.0
STOA	PLATE	0.1	0.1						
		0.5	0.1						
						0.88	39.8		
						1.83	48.78		

Ti-6Al-6V-2Sn

1 of 1

TABLE 6.18.1.2.9

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-6V-2Sn AT ROOM TEMPERATURE

ORIENTATION: T-L ENVIRONMENT: 3.5% NaCl

ORIENTATION: T-L

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)				
				ΔK Level (Ksi \sqrt{in})				
STOA	PLATE	0.1	0.1	2.5	5.0	10.0	20.0	
							50.0	100.0

1 of 1

TABLE 6.18.1.2.10

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-6V-2Sn AT ROOM TEMPERATURE

ORIENTATION: T-L

ENVIRONMENT: H.H.A.

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)						
				ΔK Level (Ksi/ $\sqrt{\text{in}}$)						
				2.6	6.0	10.0	20.0	50.0	100.0	
STOA	PLATE	0.1	10			0.73	6.98	42.31		
		0.1	10			2.16	12.37			
		0.5	10			2.37	44.01			

Ti-6Al-6V-2Sn

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TABLE 6.18.1.2.11

**FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
Ti-6Al-6V-2Sn AT ROOM TEMPERATURE**

ORIENTATION: UNSPECIFIED ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)				
				ΔK Level (Ksn/in)				
RA (Interstitial Oxygen = 0.08%)	UNSPECIFIED			2.5	5.0	10.0	20.0	
								100.0

TABLE 6.18.1.2.12

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-6Al-6V-2Sn AT ROOM TEMPERATURE

ORIENTATION: UNSPECIFIED ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Kksi/in)					
				2.5	5.0	10.0	20.0	50.0	100.0
MA (Interstitial Oxygen = 0.16%)	UNSPECIFIED						12.24		

Ti-6Al-6V-2Sn

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TABLE 6.18.2.1

TITANIUM Ti-6Al-6V-2Sn K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	2.5 • (K _{1c} /√S) ² (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K _{1c} (Ksi • √in.)	K _{1c} MEAN	STAN DEV		
1650F 1 HR WQ 1050F 1 HR AC	Forging	4.50	-320	L-C	270.0	0.501	0.250	NB	0.122	0.02	22.60	23.6	1.3	1965	84316
		4.50													
1650F 1 HR WQ 1050F 1 HR AC	Forging	4.50	R.T.	L-C	184.0	0.501	0.250	NB	0.139	0.09	34.30	31.2	2.2	1965	84316
		4.50													
		4.50													
		4.50													
		4.50													
		4.50													
1675F 2 HR AC 1600F HR FC	Plate	2.00	R.T.	---	150.0	2.500	1.250	CT	1.250	0.34	55.00	53.8	1.8	1973	90584 (1)
		2.00													
1675F 2 HR AC 1600F HR FC	Forging	---	R.T.	---	138.0	2.500	1.250	CT	1.250	0.36	52.70	55.8	4.4	1973	90584

1675F 2 HR AC 1600F HR FC	Forged Bar	1.50	R.T.	---	150.0	2.500	1.250	CT	1.250	0.43	62.20	60.2	2.8	1973	90584
		1.50													
ANNEAL-COARSE GRAIN 1350F 2 HR AC	Forging	1.00	R.T.	---	143.0	2.000	1.000	CT	1.000	0.76	79.80	70.1	6.6	1974	88962
		1.00													
		1.00													
		1.00													
		1.00													
		1.00													

NOTES: (1) COMPOSITION(WT PERCENT) 5.6Al, 5.4V, 2.0 Sn, 0.026C, 0.026C, 0.57Fe, 0.014N, 0.094H, 0.18O, 0.50Cu

TABLE 6.18.2.1 (CONTINUED)

TITANIUM Ti-6Al-6V-2Sn K ₁₀															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	2.0 • (K ₁₀ TYS) ^a (in.)	K ₁₀			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K ₁₀ (Ksi • √in.)	K ₁₀ MEAN	STAN DEV		
ANNEAL-FINE GRAIN - 1350F 2 HR AC	Forging	6.00	2.000	1.000	CT	148.0	2.000	1.000	1.000	0.34	54.60	54.0	2.1	1974	88962
		6.00	2.000	1.000	CT	148.0	2.000	1.000	1.000	0.34	54.40				
		6.00	2.000	1.000	CT	148.0	2.000	1.000	1.000	0.36	56.60				
		6.00	2.000	1.000	CT	148.0	2.000	1.000	1.000	0.28	50.30				
		6.00	2.000	1.000	CT	148.0	2.000	1.000	1.000	0.34	54.40				
		6.00	2.000	1.000	CT	148.0	2.000	1.000	1.000	0.33	53.80				
ANNEALED 10-20% PRIMARY ALPHA ANNEALED 1350F 2 HR AC	Forging	2.50	2.000	1.000	CT	145.0	2.000	1.000	1.000	0.31	51.30	50.9	0.6	1974	88962
		2.50	2.000	1.000	CT	145.0	2.000	1.000	1.000	0.30	50.40				
ANNEALED 40-50% PRIMARY ALPHA ANNEALED 1350F 2 HR AC	Forging	2.50	2.000	1.000	CT	149.0	2.000	1.000	1.000	0.28	49.40	48.8	0.9	1974	88962
		2.50	2.000	1.000	CT	149.0	2.000	1.000	1.000	0.26	48.10				
BB AB FIN-10 BETA BLOCKED ALPHA-BETA FINISHED 10% REDUCTION, SOLUTION TREATED & OVERAGED 1650F 1 HR WQ 1300F 2 HR AC	Forging	2.50	2.000	1.000	CT	148.0	2.000	1.000	1.000	0.67	76.60	72.6	5.6	1974	88962
		2.50	2.000	1.000	CT	148.0	2.000	1.000	1.000	0.54	68.70				
BB AB FIN-10MA BETA BLOCKED ALPHA-BETA FINISHED 10% REDUCTION MILL ANNEALED 1350F 2 HR AC	Forging	2.50	2.000	1.000	CT	147.0	2.000	1.000	1.000	0.60	71.80	77.0	6.8	1974	88962
		2.50	2.000	1.000	CT	147.0	2.000	1.000	1.000	0.83	84.70				
		2.50	2.000	1.000	CT	147.0	2.000	1.000	1.000	0.64	74.60				
BB AB FIN-30 BETA BLOCKED ALPHA-BETA FINISHED 30% REDUCTION SOLUTION	Forging	2.50	2.000	1.000	CT	148.0	2.000	1.000	1.000	0.47	64.30	61.6	2.4	1974	88962
		2.50	2.000	1.000	CT	148.0	2.000	1.000	1.000	0.42	60.80				
		2.50	2.000	1.000	CT	148.0	2.000	1.000	1.000	0.40	59.60				

Ti-6Al-6V-2Sn

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TABLE 6.18.2.1 (CONTINUED)

TITANIUM Ti-6Al-6V-2Sn K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	2.5 * $(K_{Ic}/\sqrt{S})^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Ksi * $\sqrt{in.}$)	K_{Ic} MEAN	STAN DEV		
BB AB FIN-30MA BETA BLOCKED ALPHA-BETA FINISHED 30% REDUCTION MILL ANNEALED 1350F 2 HR AC	Forging	2.50	R.T.	---	145.0	2.000	1.000	CT	1.000	0.51	65.40	64.7	0.6	1974	88962
		2.50									64.20				
		2.50									64.40				
BB B FIN-10 BETA BLOCKED BETA FINISHED 10% REDUCTION SOLUTION TREATED & OVERAGED 1650F 1 HR WQ 1300F 2 HR AC	Forging	2.50	R.T.	---	140.0	2.000	1.000	CT	1.000	0.69	73.60	72.1	2.1	1974	88962
		2.50									70.70				
		2.50									70.70				
BB B FIN-10MA BETA BLOCKED BETA FINISHED 10% REDUCTION MILL ANNEALED 1350F 2 HR AC	Forging	2.50	R.T.	---	136.0	2.000	1.000	CT	1.000	0.69	71.70	70.5	1.0	1974	88962
		2.50									70.20				
		2.50									69.70				
BETA ANNEAL	Plate	1.00	R.T.	T-L	136.4	2.550	1.000	WOL-CT EQ.	---	0.73	74.60	---	---	1977	JEM01
BETA ANNEAL 1810F 1 HR ARGON COOL	Plate	0.50	R.T.	---	136.8	1.000	0.447	CT	0.568	0.36	53.20	64.3	2.0	1971	83222
		0.50									53.00				
		0.50									56.60				
BETA ANNEAL & STOA-1800F 0.5 HR AC 1575F 0.5 HR WQ 1050F 8 HR AC	Plate	0.62	R.T.	L-T	156.0	2.000	0.626	CT	1.000	0.24	48.80	---	---	1974	88186
		0.62			156.0	2.000	0.626	CT	1.000	0.27	51.30	---	---	1974	88186

TABLE 6.18.2.1 (CONTINUED)

TITANIUM Ti-6Al-6V-2Sn K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic} \sqrt{TS})^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			CRACK LENGTH (in.) A	K_{Ic} MEAN	STAN DEV		
BF AB FOR-ANN BETA FLECTED ALPHA-BETA FORGED ANNEALED 1350F 2 HR AC	Forging	2.50	R.T.	---	138.0	2.000	1.000	CT	1.000	0.82	79.30	3.7	1974	88962	
		2.50									74.50				
		2.50									72.10				
BF B FOR-ANN BETA FLECTED BETA FORGED ANNEALED 1350F 2 HR AC	Forging	2.50	R.T.	---	136.0	2.000	1.000	CT	1.000	0.74	73.90	0.6	1974	88962	
		2.50									74.70				
		2.50									73.80				
BF LAB FOR-ANN BETA FLECTED LOW ALPHA-BETA FORGED (1600F) ANNEALED 1350F 2 HR AC	Forging	2.50	R.T.	---	141.0	2.000	1.000	CT	1.000	0.38	55.20	1.6	1974	88962	
		2.50									55.70				
		2.50									56.10				
DUPLEX ANNEAL	Plate	0.50	R.T.	T-L	150.5	2.000	0.485	CT	1.066	0.50	67.40	2.0	1971	83222 (1)	
		0.50									64.50				
		0.50									63.50				
MILL ANNEALED	Plate	2.00	R.T.	---	157.0	2.500	1.250	CT	1.250	0.18	42.30	0.0	1973	90584 (2)	
		2.00									42.30				
		1.00									42.80				
MILL ANNEALED	Plate	0.50	R.T.	T-L	163.3	1.000	0.495	CT	0.522	0.09	32.00	5.2	1971	83222	
		0.50									32.20				
		0.50									32.10				

NOTES: (1) 1700F 1HR ARGON COOL 1400F 1HR ARGON COOL

(2) COMPOSITION(WT PERCENT) 6.6Al, 6.4V, 2.0 Sn, 0.025C, 0.57Fe, 0.014N, 0.084H, 0.180, 0.50Cu

Ti-6Al-6V-2Sn

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TABLE 6.18.2.1 (CONTINUED)

TITANIUM Ti-6Al-6V-2Sn K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	2.5 * $(K_{Ic}/\sqrt{A})^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Ksi * $\sqrt{in.}$)	K_{Ic} MEAN	STAN DEV		
MILL ANNEALED	Forging	3.80	-65	L-T	169.0	2.498	1.008	CT	1.299	0.19	47.20	44.2	5.7	1973	90589
		3.80													
		3.80													
MILL ANNEALED	Forging	---	R.T.	---	144.0	2.500	1.250	CT	1.250	0.37	55.60	55.7	0.1	1973	90584

MILL ANNEALED	Forging	3.80	R.T.	L-T	149.0	2.499	0.999	CT	1.299	0.43	61.70	58.6	2.7	1973	90589
		3.80													
		3.80													
MILL ANNEALED	Forged Bar	1.50	R.T.	---	153.0	2.500	1.250	CT	1.250	0.20	43.10	---	---	1973	90584
		2.20													
		2.20													
MILL ANNEALED	Billet	2.20	R.T.	L-T	144.0	2.500	1.251	CT	1.332	0.37	55.60	52.3	6.4	1971	84360
		2.20													
		2.20													
MILL ANNEALED 1000F 2 HR AC	Billet	2.20	R.T.	L-T	155.0	2.495	1.255	CT	1.295	0.32	55.50	57.1	2.2	1971	84360
		2.20													
		2.20													
RECRYSTALLIZE ANNEAL	Plate	1.00	R.T.	T-L	150.0	2.550	1.000	WOL-CT EQ	---	0.45	63.70	---	---	1977	JEM01
		3.80													
		3.80													
STA-1600F .5HR WQ 1000F 6 HR AC	Forging	3.80	-65	L-T	209.0	2.499	1.005	CT	1.267	0.05	29.80	28.4	0.6	1973	90589
		3.80													
		3.80													

TABLE 6.18.2.1 (CONTINUED)

TITANIUM Ti-6Al-6V-2Sn K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (In.) A	$2.5 \cdot (K_{Ic} / \sqrt{A})^2$ (In.)	K_{Ic}			DATE	REFER
	FORM	THICK (In.)				WIDTH (In.) W	THICK (In.) B	DESIGN			K_{Ic} (Ksi $\cdot \sqrt{\text{In.}}$)	K_{Ic} MEAN	STAN DEV		
STA-1600F .5 HR WQ 1000F 6 HR AC	Forging	3.80	R.T.	L-T	184.0	2.501	1.007	CT	1.278	0.07	30.60	30.8	0.7	1973	90589
		3.80													
		3.80													
STA-1600F .5 HR WQ 1000F 6 HR AC	Forging	3.80	300	L-T	165.0	2.497	1.015	CT	1.278	0.27	54.20	53.0	3.2	1973	90589
		3.80													
		3.80													
STA-1650F 0.5 HR WQ 1050F 24 HR AC	Forging	---	R.T.	L-C	186.0	0.998	0.500	CT	0.508	0.04	24.60	25.1	0.6	1972	86494 (1)

STA-1650F 0.5 HR WQ 1050F 24 HR AC	Forging	---	R.T.	C-L	186.0	1.001	0.500	CT	0.505	0.05	25.30	28.7	2.1	1972	86494 (1)

STA-1675F 0.25 HR WQ 1100F 4 HR	Plate	1.25	R.T.	T-L	173.3	2.000	0.489	CT	1.138	0.07	29.80	34.1	3.8	1971	83222
		1.25													
		1.25													
STOA-1600F 1.5 HR WQ 1250F 6 HR AC	Extrusion	3.00	R.T.	C-R	143.0	1.499	0.749	CT	0.783	0.36	54.50	53.6	1.4	1973	87230 (2)
		3.00													
		3.00													
STOA-1650F 1 HR WQ 1300F 2 HR AC	Forging	2.50	R.T.	---	154.0	2.000	1.000	CT	1.000	0.17	40.40	41.0	0.8	1974	88962
		2.50													

NOTES: (1) ISOTHERMAL FORGING FOR AIRCRAFT NOSE WHEEL

(2) ALPHA PRECIPITATE IN BETA MATRIX
STRAIGHTNESS OF CRACK FRONT MAY NOT MEET ASTM E398-72 REQUIREMENTS

Ti-6Al-6V-2Sn

7 of 7

TABLE 6.18.2.1 (CONCLUDED)

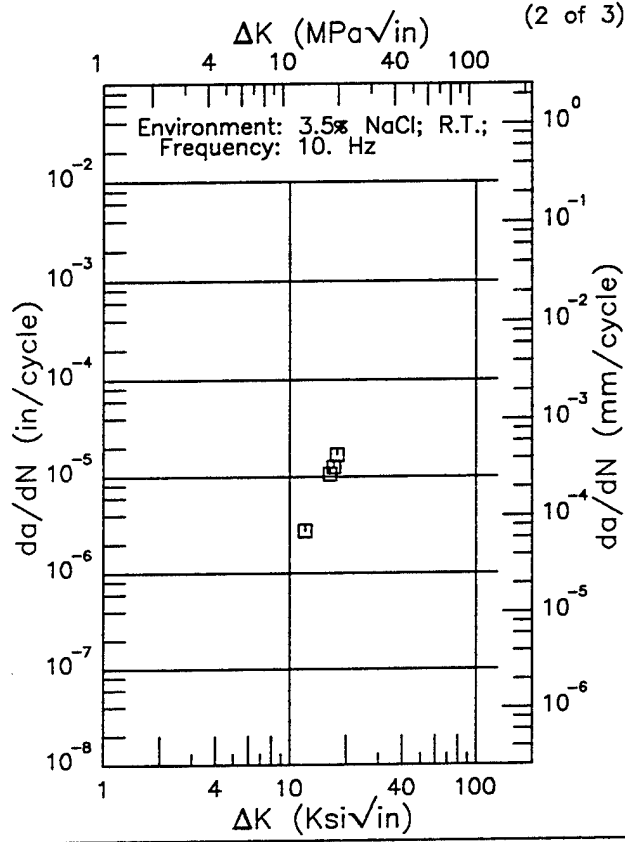
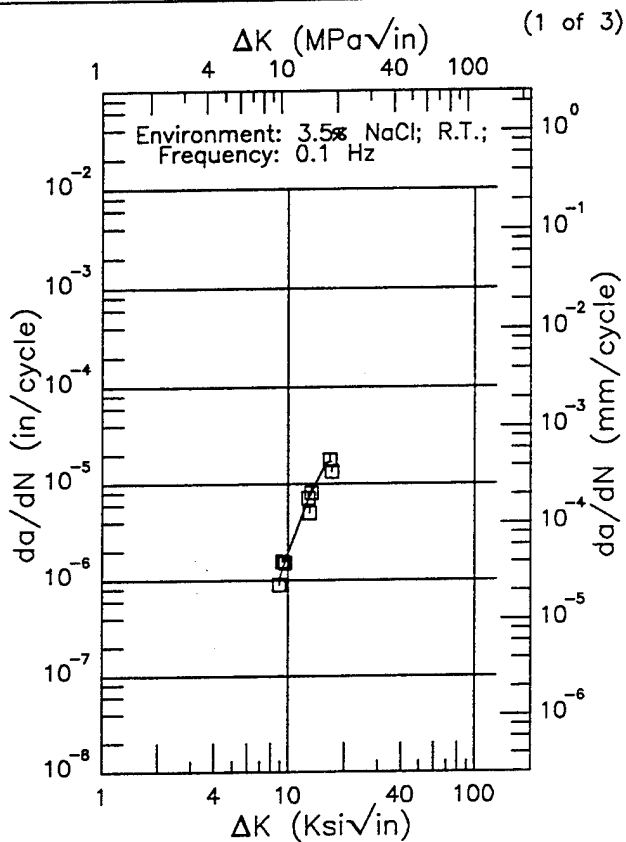
TITANIUM Ti-6Al-6V-2Sn K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (ksi)	SPECIMEN			CRACK LENGTH (in.) Δ	$2.5 \cdot (K_{Ic}/Y)^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (ksi) $\sqrt{\text{in.}}$	K_{Ic} MEAN	STAN DEV		
STOA-1700F 1 HR WQ 1400F 1 HR AC	Plate	0.38	R.T.	L-T	156.0	1.000	0.375	CT	---	0.18	41.90	42.9	1.2	1974	90981
		0.38									42.50				
		0.38									44.30				
STOA-1700F 1 HR WQ 1400F 1 HR AC	Plate	0.38	R.T.	T-L	160.0	1.000	0.375	CT	---	0.18	42.90	46.1	3.1	1974	90981
		0.38									44.00				
		0.38									49.10				
		0.38									48.30				
STOA-1700F 1 HR WQ 1400F 1 HR AC	Billet	12.00	R.T.	L-T	147.0	2.000	1.020	NB	---	0.33	53.60	62.8	6.9	1974	90981
		12.00									70.40				
		12.00									63.70				
		12.00									63.40				
STOA-1700F 1 HR WQ 1400F 1 HR AC	Billet	12.00	R.T.	T-L	144.0	2.000	1.020	NB	---	0.42	58.80	57.0	3.7	1974	90981
		12.00									58.00				
		12.00									51.50				
		12.00									59.60				

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EF | Ti-6Al-6V-2Sn |

Condition/Ht: BA
 Form: 0.38 in. Plate
 Specimen Type: PTSF (max stress specified)
 Orientation: T-S
 Stress Ratio: 0.1

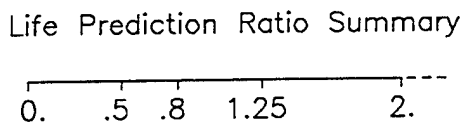
Yield Strength: 130 - 160 ksi
 Ult. Strength: 168 ksi
 Specimen Thk: 0.375 in.
 Specimen Width: 5 in.
 Ref: 90981



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
8.91 (min)	1.01
9.	1.07
10.	1.94
13.	7.24
16.	16.6
17.02 (max)	16.9

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
--------------------------------------	-------------------------------

RMS %
 Error
 16.37



RMS %
 Error

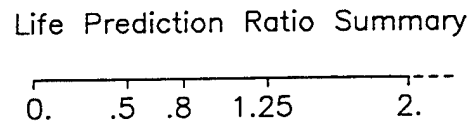


Figure 6.18.3.1.1

Ti-6Al-6V-2Sn EF

Condition/Ht: BA
 Form: 0.38 in. Plate
 Specimen Type: PTSF (max stress specified)
 Orientation: T-S
 Stress Ratio: 0.1

Yield Strength: 130 - 160 ksi
 Ult. Strength: 168 ksi
 Specimen Thk: 0.375 in.
 Specimen Width: 5 in.
 Ref: 90981

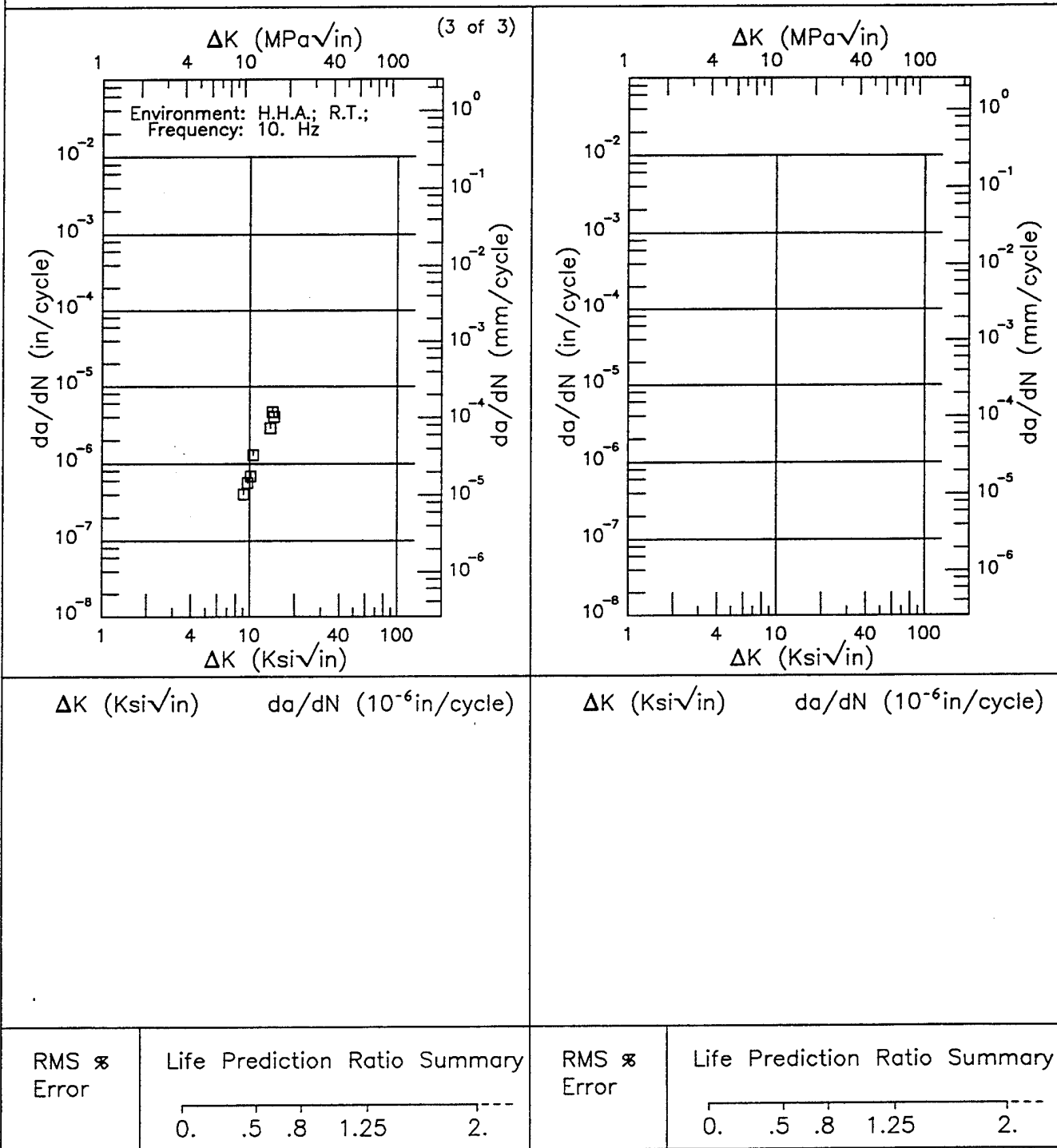
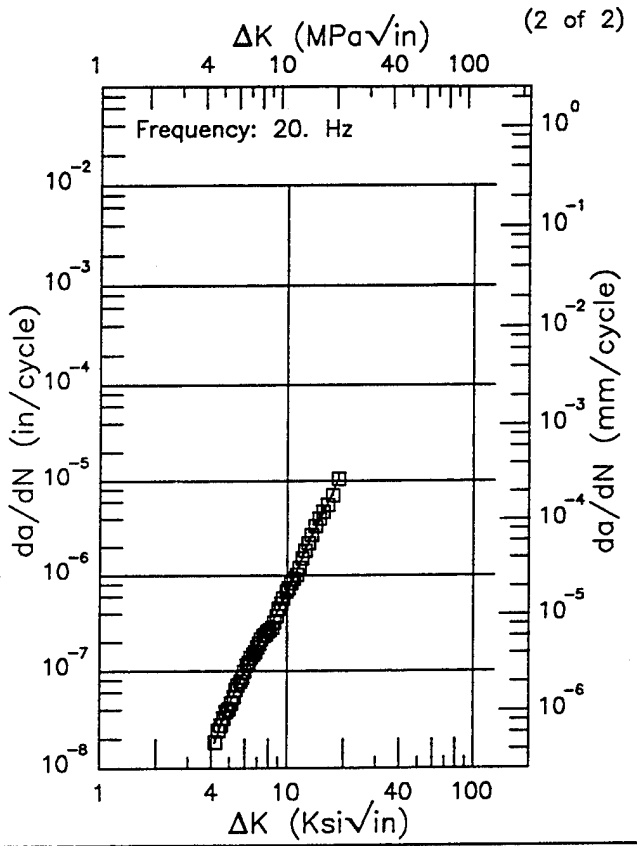
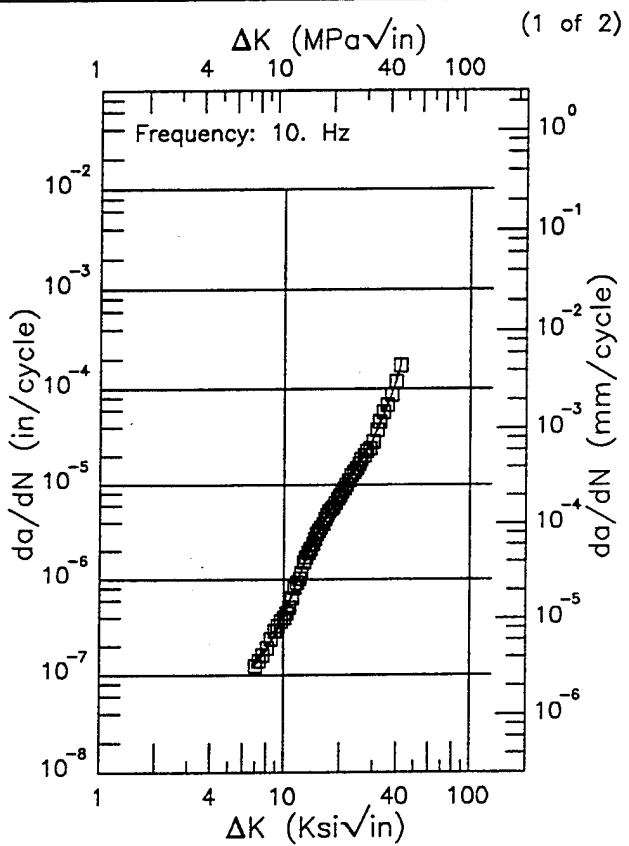


Figure 6.18.3.1.1 (Concluded)

F Ti-6Al-6V-2Sn

Condition/Ht: MA
 Form: Extrusion
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Stress Ratio: 0.02
 Environment: LAB AIR; RT

Yield Strength: 143 ksi
 Ult. Strength: 157 ksi
 Specimen Thk: 0.081 - 0.082 in.
 Specimen Width: 3.509 - 3.519 in.
 Ref: MA002



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
6.99 (min)	0.130
7.	0.130
8.	0.178
9.	0.271
10.	0.426
13.	1.51
16.	3.77
20.	8.00
25.	15.9
30.	30.9
35.	61.9
40.	128.
41.81 (max)	168.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
4.16 (min)	0.0182
5.	0.0464
6.	0.0988
7.	0.174
8.	0.279
9.	0.431
10.	0.653
13.	2.22
16.	5.10
18.61 (max)	10.4

RMS % Error 4.86	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
---------------------	---

RMS % Error 5.05	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
---------------------	---

Figure 6.18.3.1.2

Ti-6Al-6V-2Sn

E

Condition/Ht: MA
 Form: 2 in. Forging
 Specimen Type: DCB
 Orientation: L-T
 Stress Ratio: 0.02
 Frequency: 0.1 Hz

Yield Strength: 144 ksi
 Ult. Strength: 151 ksi
 Specimen Thk: 0.75 in.
 Specimen Width: 5.5 in.
 Ref: 84360

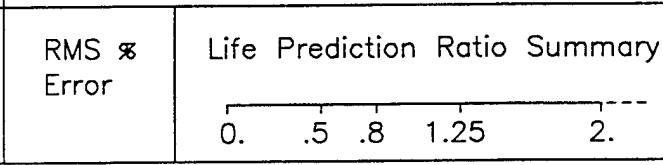
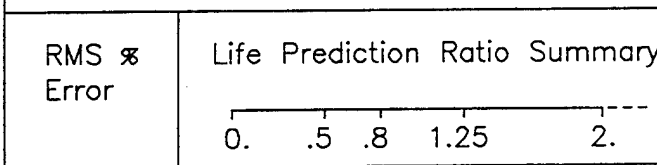
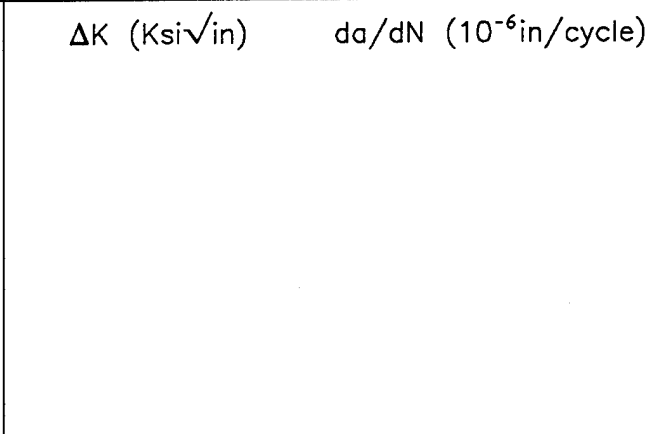
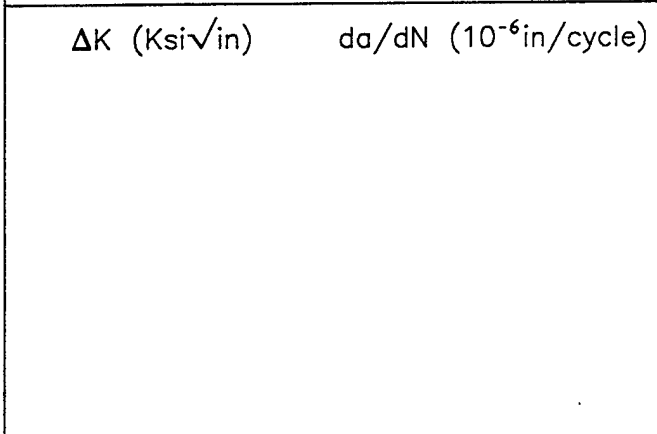
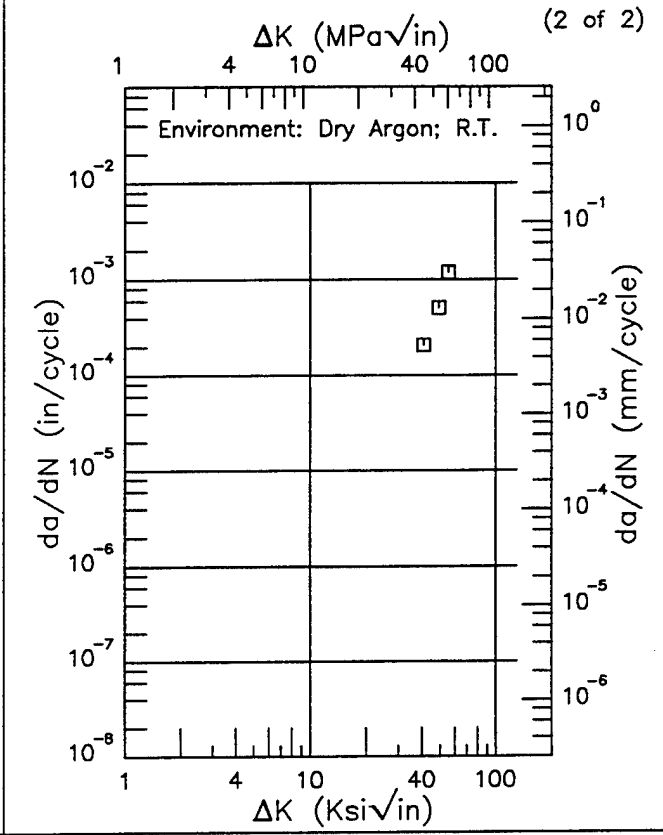
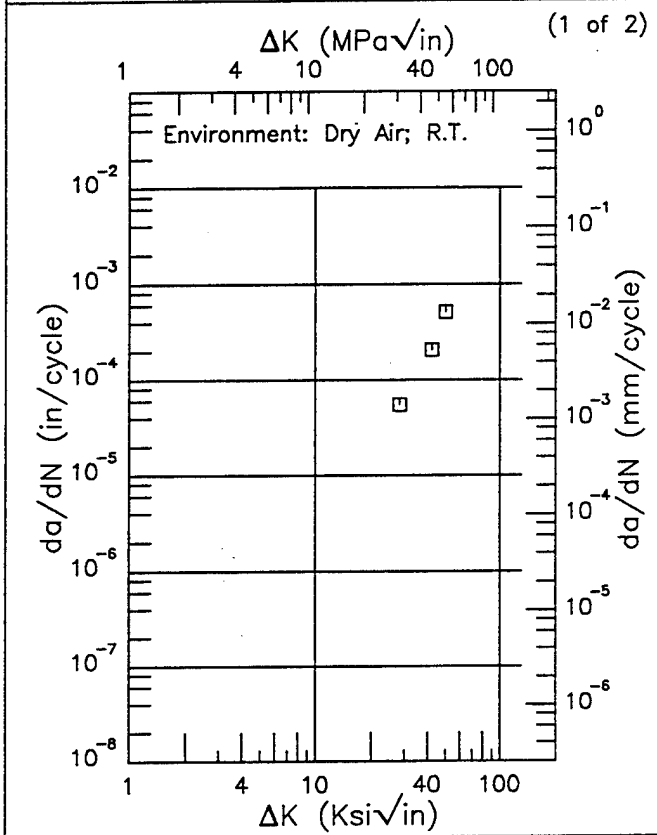


Figure 6.18.3.1.3

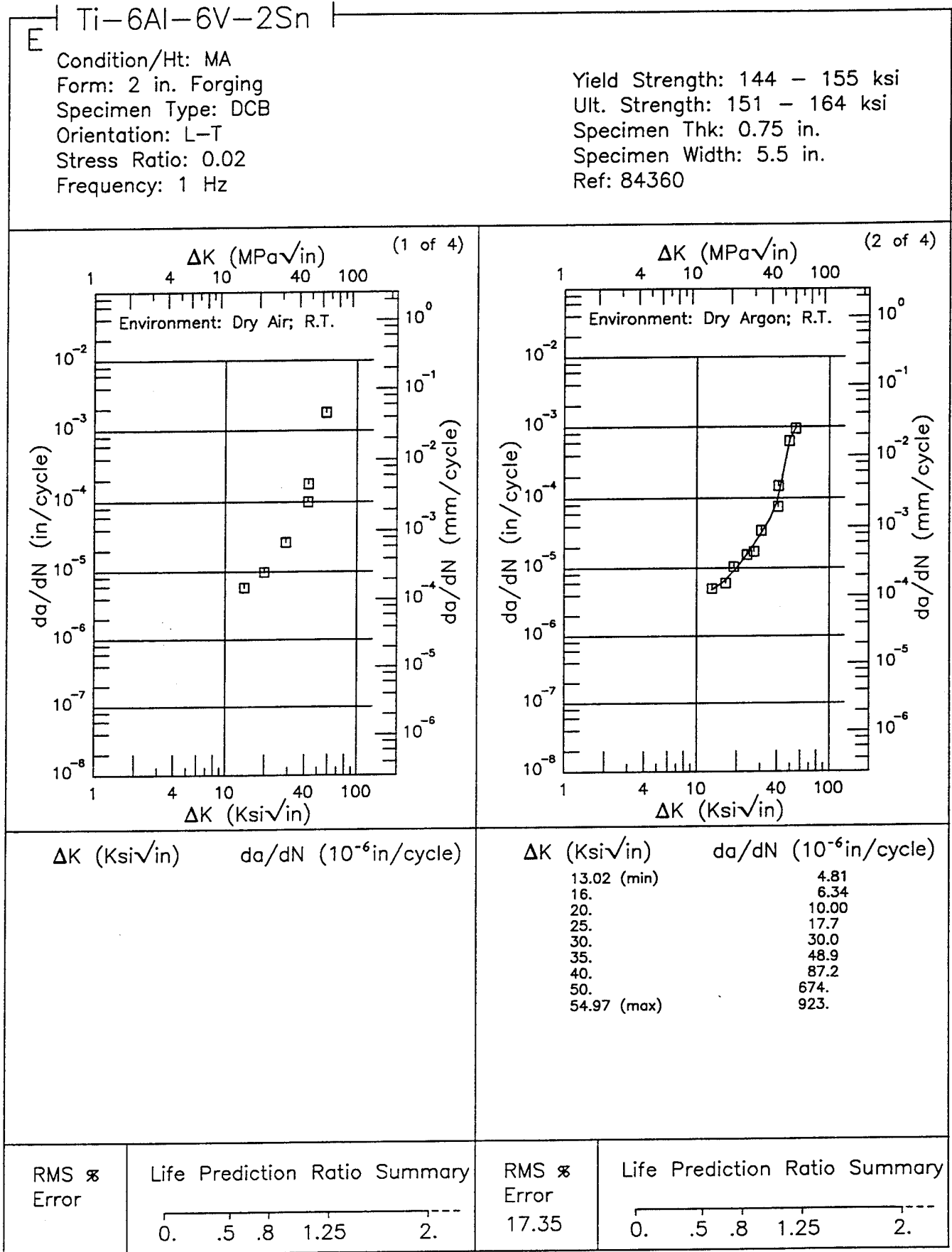


Figure 6.18.3.1.4

Ti-6Al-6V-2Sn E

Condition/Ht: MA
 Form: 2 in. Forging
 Specimen Type: DCB
 Orientation: L-T
 Stress Ratio: 0.02
 Frequency: 1 Hz

Yield Strength: 144 - 155 ksi
 Ult. Strength: 151 - 164 ksi
 Specimen Thk: 0.75 in.
 Specimen Width: 5.5 in.
 Ref: 84360

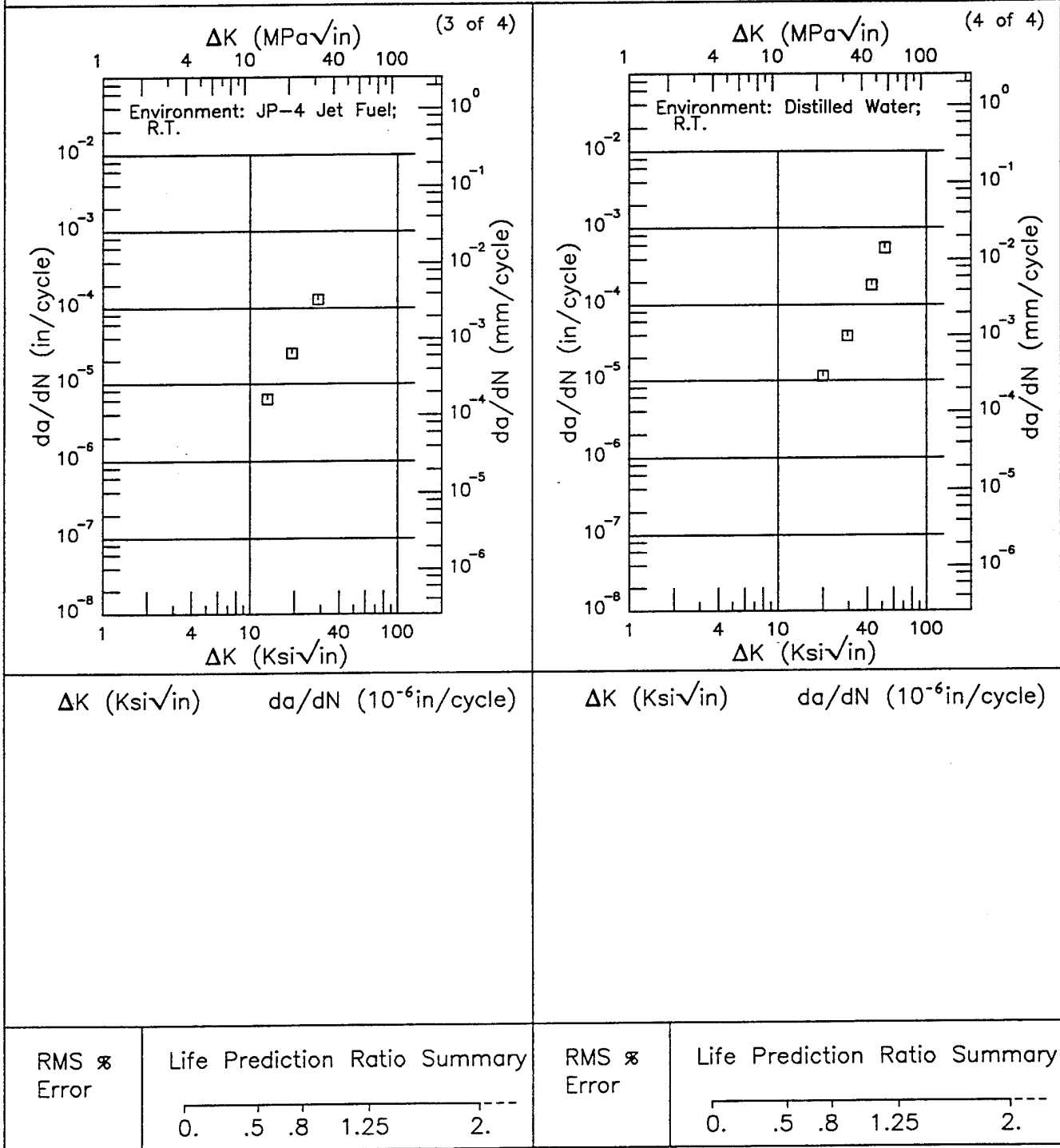


Figure 6.18.3.1.4 (Concluded)

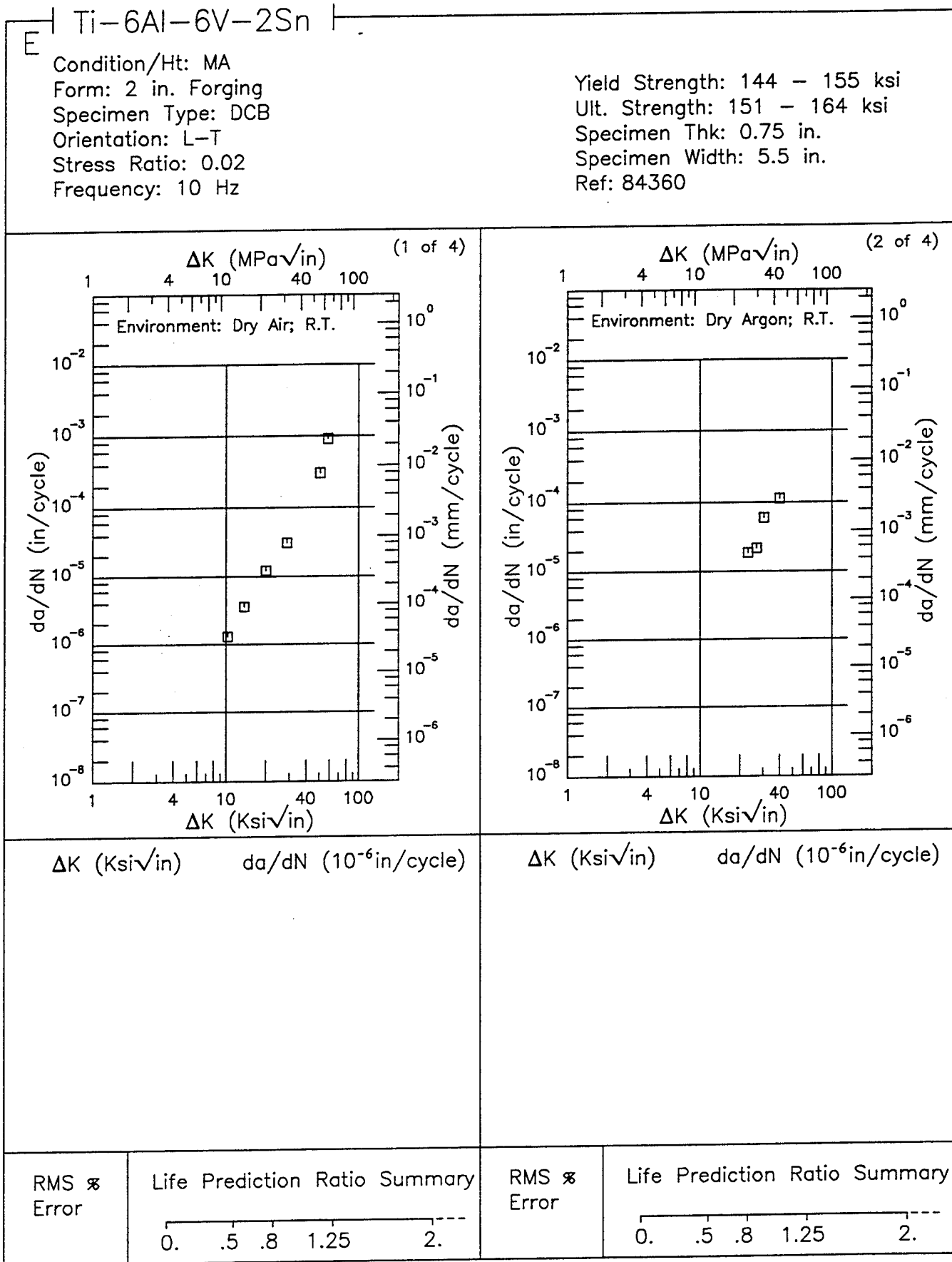


Figure 6.18.3.1.5

Ti-6Al-6V-2Sn

E

Condition/Ht: MA
 Form: 2 in. Forging
 Specimen Type: DCB
 Orientation: L-T
 Stress Ratio: 0.02
 Frequency: 10 Hz

Yield Strength: 144 - 155 ksi
 Ult. Strength: 151 - 164 ksi
 Specimen Thk: 0.75 in.
 Specimen Width: 5.5 in.
 Ref: 84360

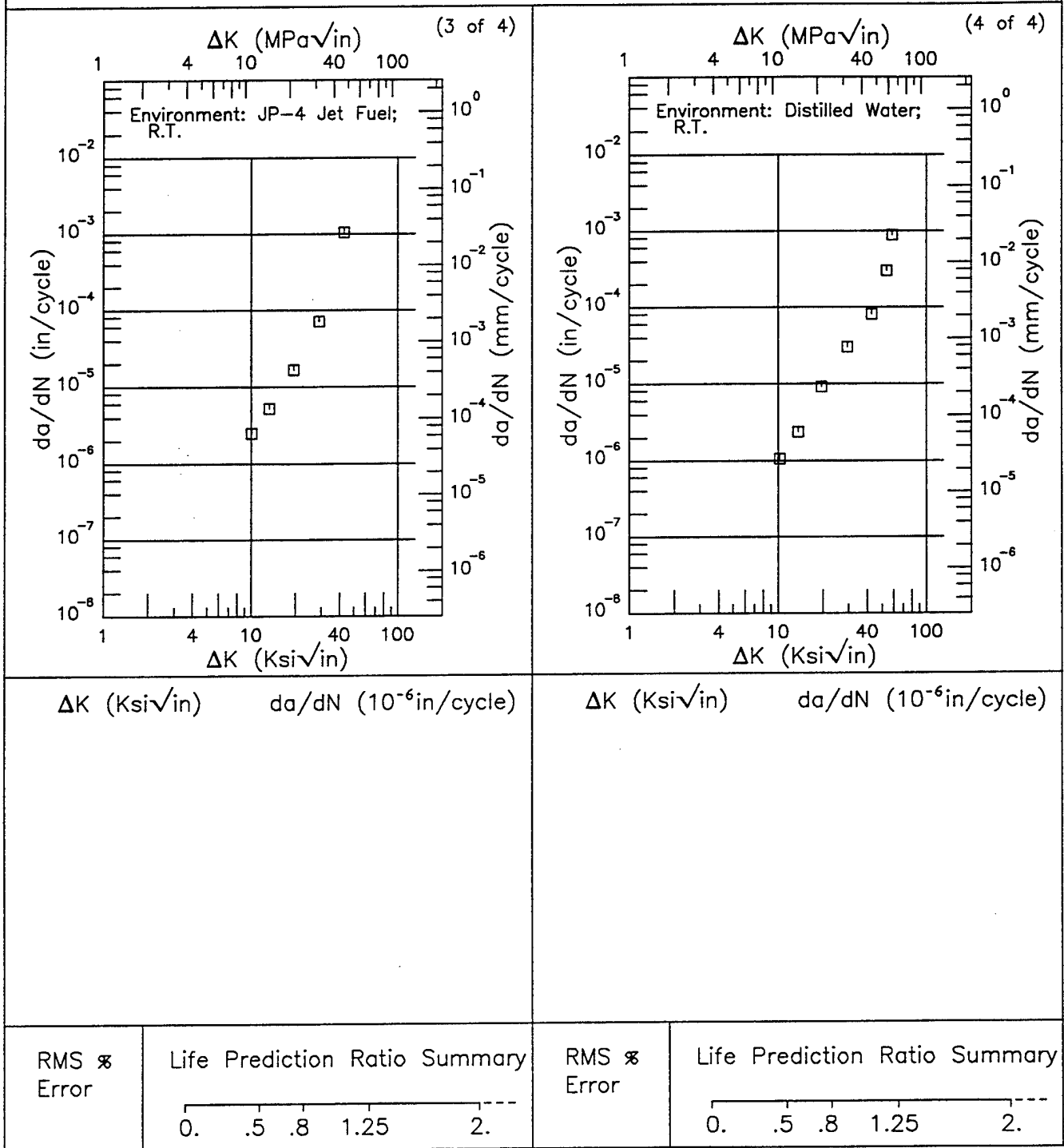


Figure 6.18.3.1.5 (Concluded)

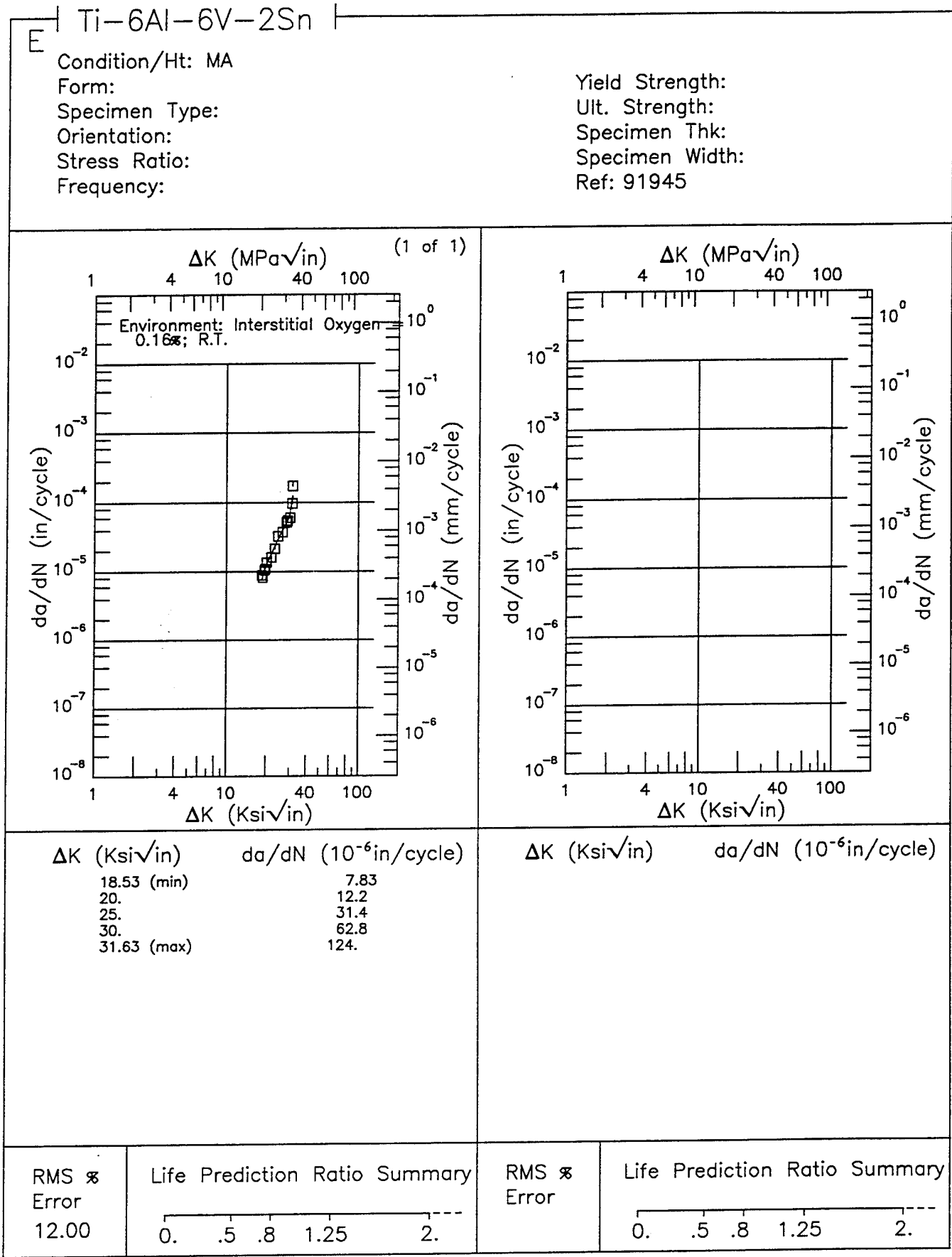


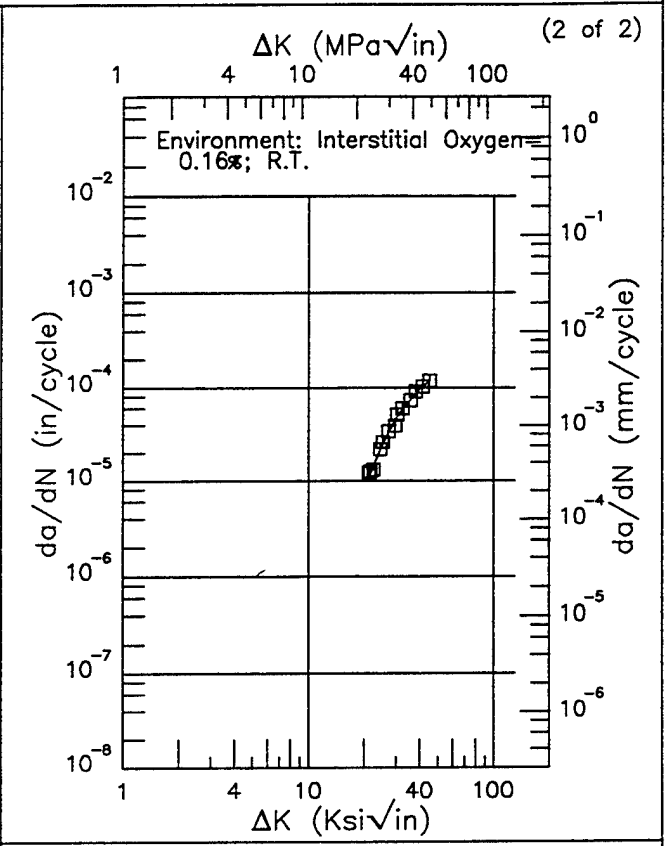
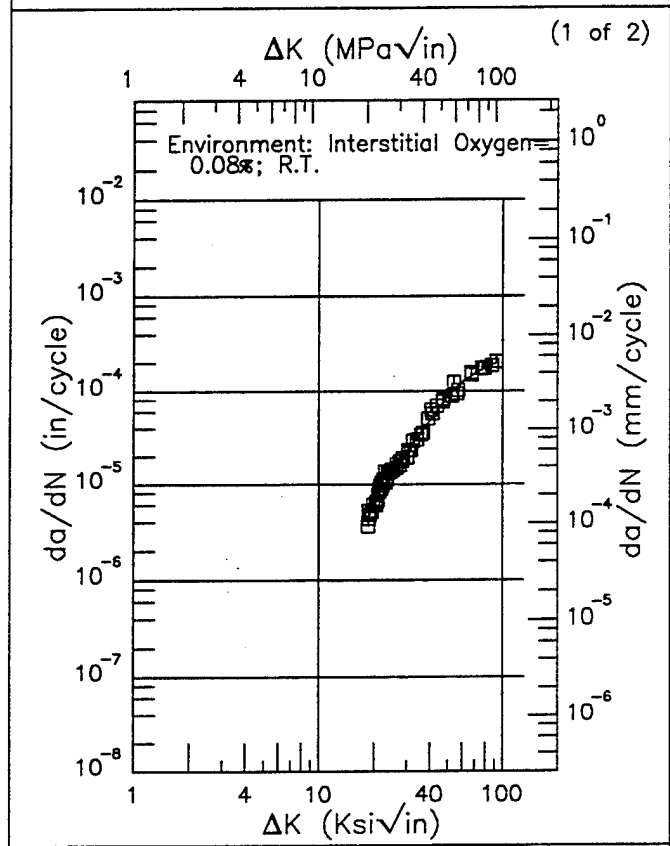
Figure 6.18.3.1.6

Ti-6Al-6V-2Sn

E

Condition/Ht: RA
 Form:
 Specimen Type:
 Orientation:
 Stress Ratio:
 Frequency:

Yield Strength:
 Ult. Strength:
 Specimen Thk:
 Specimen Width:
 Ref: 91945



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
18.44 (min)	4.48
20.	6.34
25.	13.7
30.	22.8
35.	34.1
40.	49.0
50.	87.9
60.	124.
70.	151.
80.	177.
90.	195.
91.66 (max)	197.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
21.24 (min)	10.9
25.	23.7
30.	47.1
35.	71.6
40.	97.7
45.04 (max)	116.

RMS % Error 11.34	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

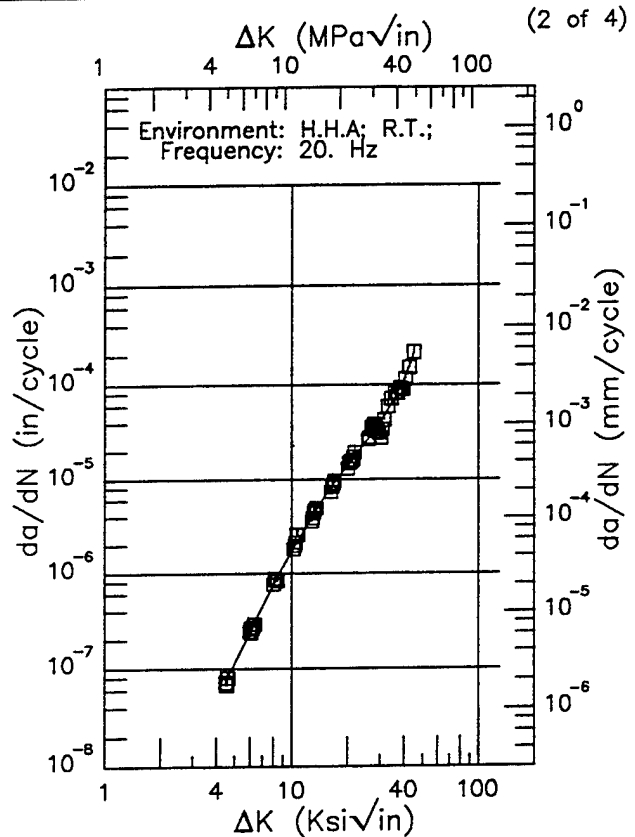
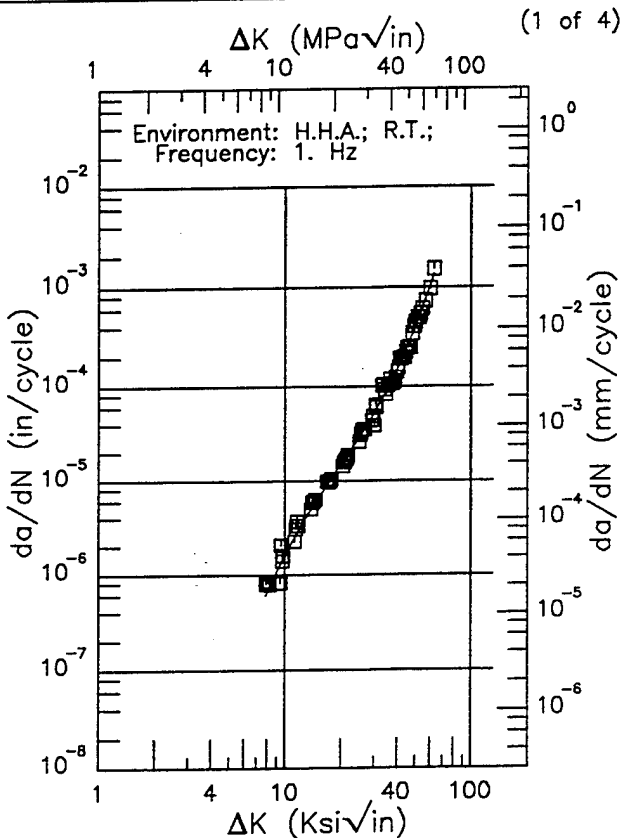
RMS % Error 6.60	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
---------------------	---

Figure 6.18.3.1.7

EF Ti-6Al-6V-2Sn

Condition/Ht: STOA
 Form: 0.63 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.1

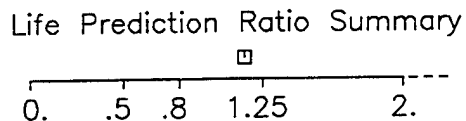
Yield Strength: 160 ksi
 Ult. Strength: 170 ksi
 Specimen Thk: 0.148 - 0.151 in.
 Specimen Width: 3 in.
 Ref: 86844



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.86 (min)	0.603
8.	0.653
9.	1.08
10.	1.64
13.	4.11
16.	7.84
20.	15.1
25.	29.4
30.	52.4
35.	89.5
40.	149.
50.	395.
60.	1014.
63.06 (max)	1347.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
4.47 (min)	0.0703
5.	0.112
6.	0.237
7.	0.440
8.	0.741
9.	1.16
10.	1.70
13.	4.13
16.	7.64
20.	13.8
25.	24.5
30.	40.5
35.	66.0
40.	111.
45.10 (max)	209.

RMS %
 Error
 14.78



RMS %
 Error
 9.62

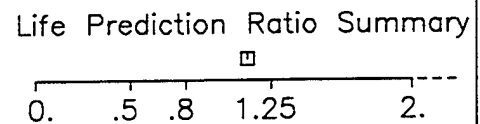
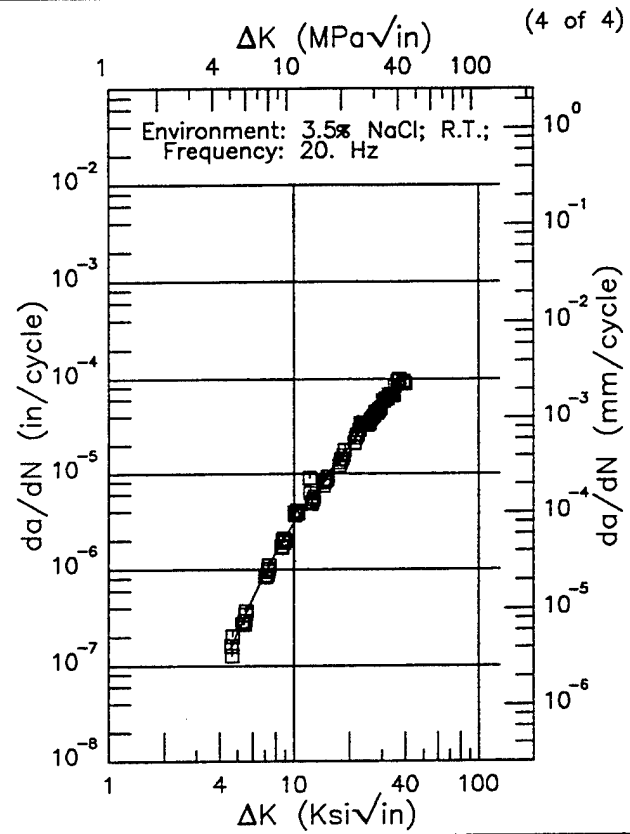
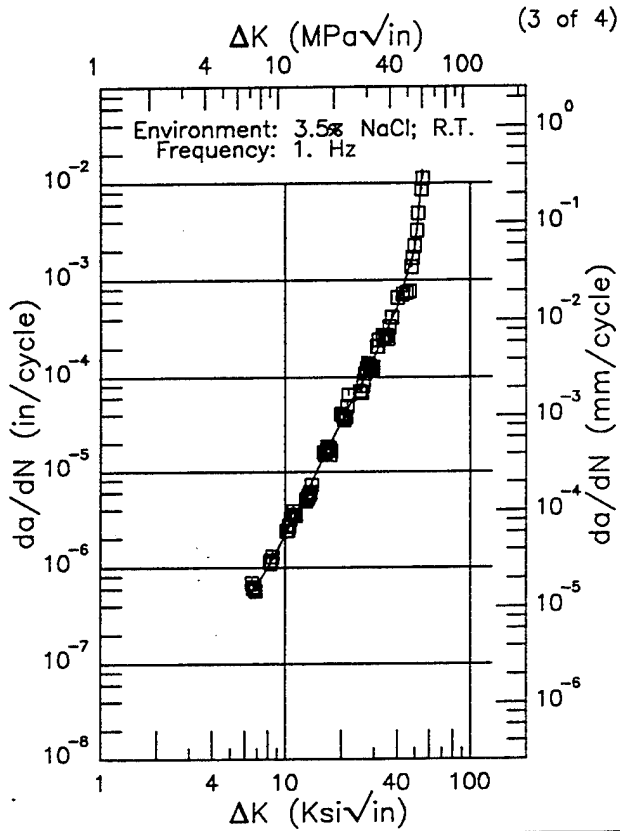


Figure 6.18.3.1.8

Ti-6Al-6V-2Sn EF

Condition/Ht: STOA
 Form: 0.63 in. Plate
 Specimen Type: CT
 Orientation: L-T
 Stress Ratio: 0.1

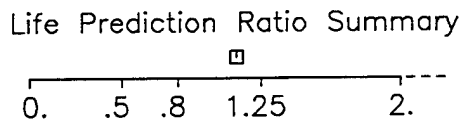
Yield Strength: 160 ksi
 Ult. Strength: 170 ksi
 Specimen Thk: 0.148 - 0.151 in.
 Specimen Width: 3 in.
 Ref: 86844



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
6.57 (min)	0.552
7.	0.660
8.	0.999
9.	1.49
10.	2.19
13.	6.04
16.	13.9
20.	33.8
25.	80.2
30.	164.
35.	305.
40.	528.
50.	2094.
54.97 (max)	13672.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
4.61 (min)	0.151
5.	0.223
6.	0.499
7.	0.916
8.	1.48
9.	2.19
10.	3.05
13.	6.46
16.	11.1
20.	19.5
25.	34.0
30.	53.7
35.	78.2
39.72 (max)	105.

RMS %
 Error
 14.53



RMS %
 Error
 12.57

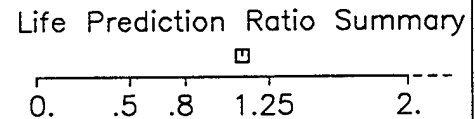
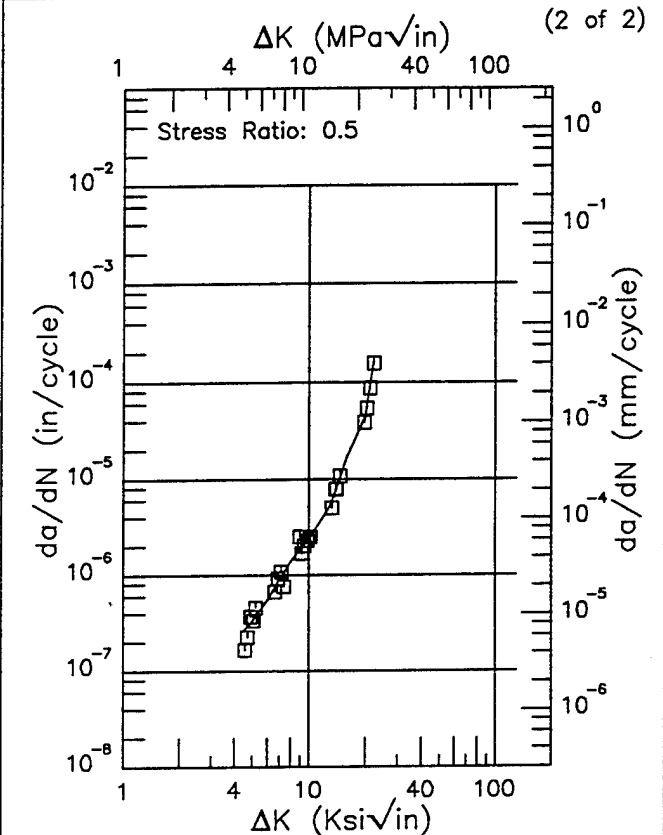
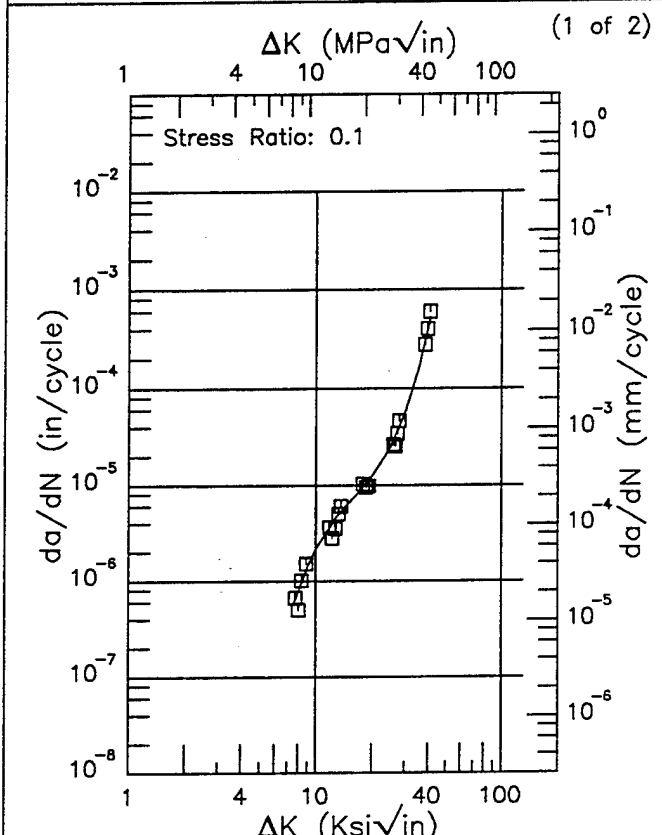


Figure 6.18.3.1.8 (Concluded)

R | Ti-6Al-6V-2Sn |

Condition/Ht: STOA
 Form: 0.38 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: T-L
 Frequency: 10 Hz
 Environment: H.H.A.; RT

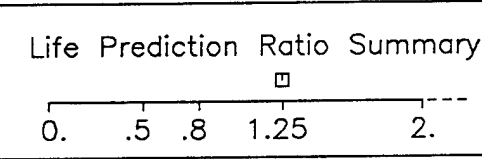
Yield Strength: 160.3 ksi
 Ult. Strength: 167.8 ksi
 Specimen Thk: 0.375 in.
 Specimen Width: 5 in.
 Ref: 90981



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
7.71 (min)	0.615
8.	0.764
9.	1.40
10.	2.16
13.	4.71
16.	7.45
20.	12.4
25.	24.7
30.	56.3
35.	146.
40.	421.
41.04 (max)	531.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
4.54 (min)	0.271
5.	0.332
6.	0.550
7.	0.896
8.	1.35
9.	1.87
10.	2.37
13.	5.27
16.	16.9
20.	44.0
22.27 (max)	172.

RMS % Error
15.21



RMS % Error
16.67

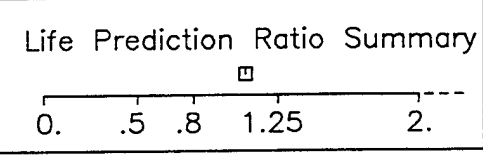


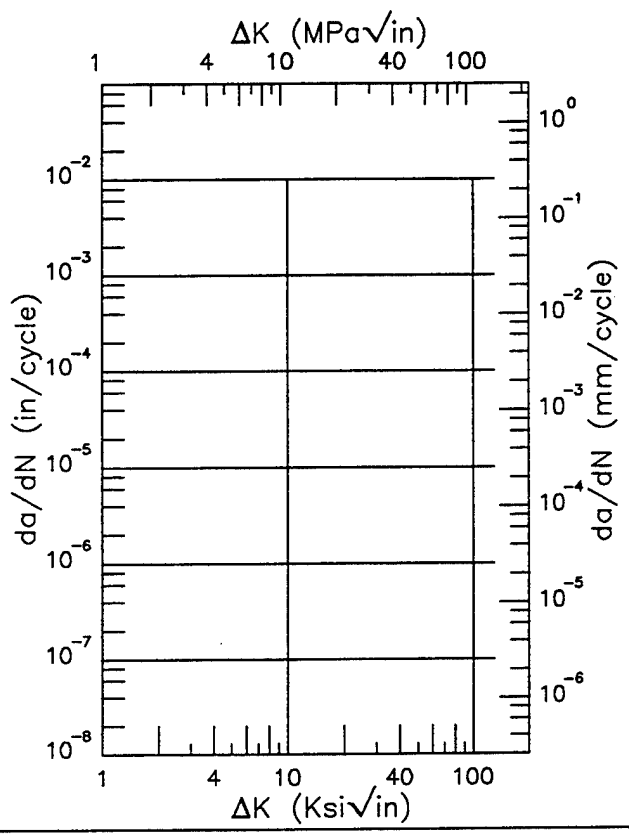
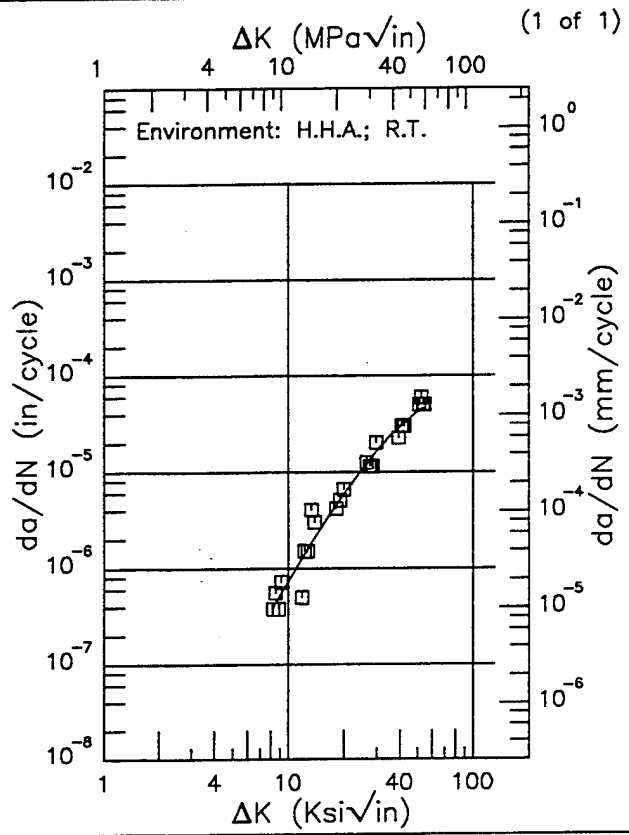
Figure 6.18.3.1.9

Ti-6Al-6V-2Sn

E

Condition/Ht: STOA
 Form: 0.13 in. Plate
 Specimen Type: CCP (max load specified)
 Orientation: T-L
 Stress Ratio: 0.1
 Frequency: 10 Hz

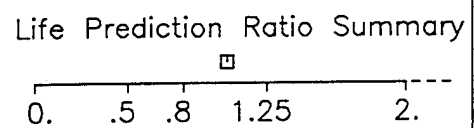
Yield Strength: 137.4 ksi
 Ult. Strength: 143 ksi
 Specimen Thk: 0.125 in.
 Specimen Width: 5 in.
 Ref: 90981



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.20 (min)	0.390
9.	0.525
10.	0.734
13.	1.67
16.	3.15
20.	5.98
25.	10.8
30.	16.7
35.	23.2
40.	29.9
50.	42.3
54.31 (max)	46.9

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
---------------------	-------------------------------

RMS %
 Error
 34.05



RMS %
 Error

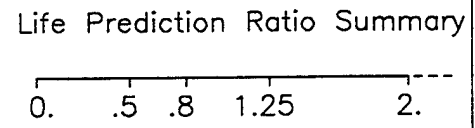


Figure 6.18.3.1.10

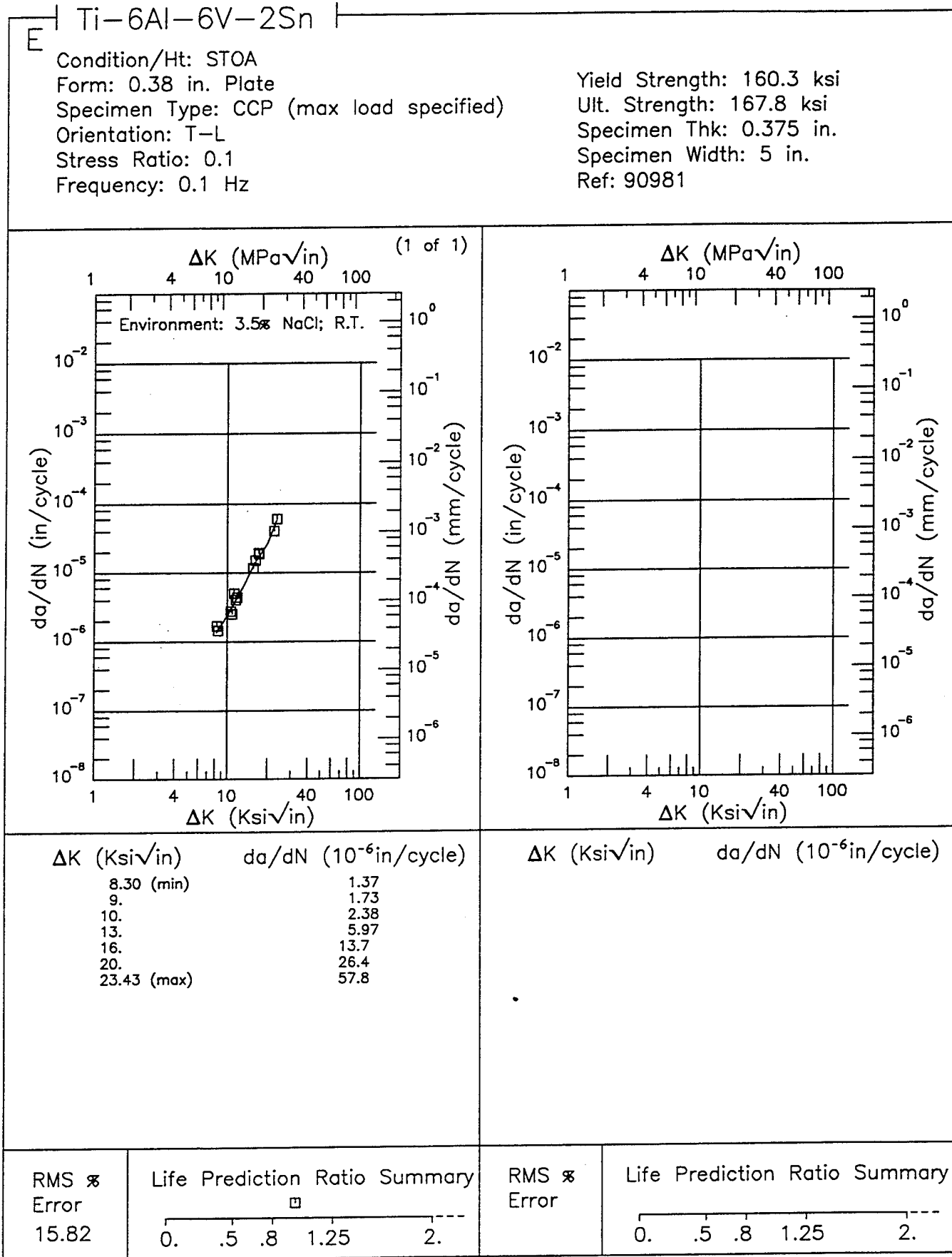


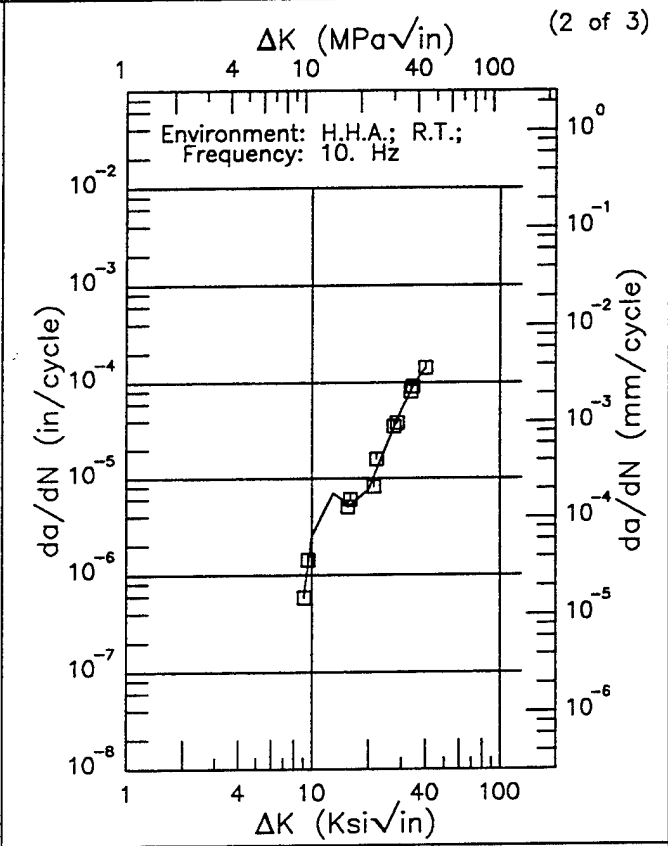
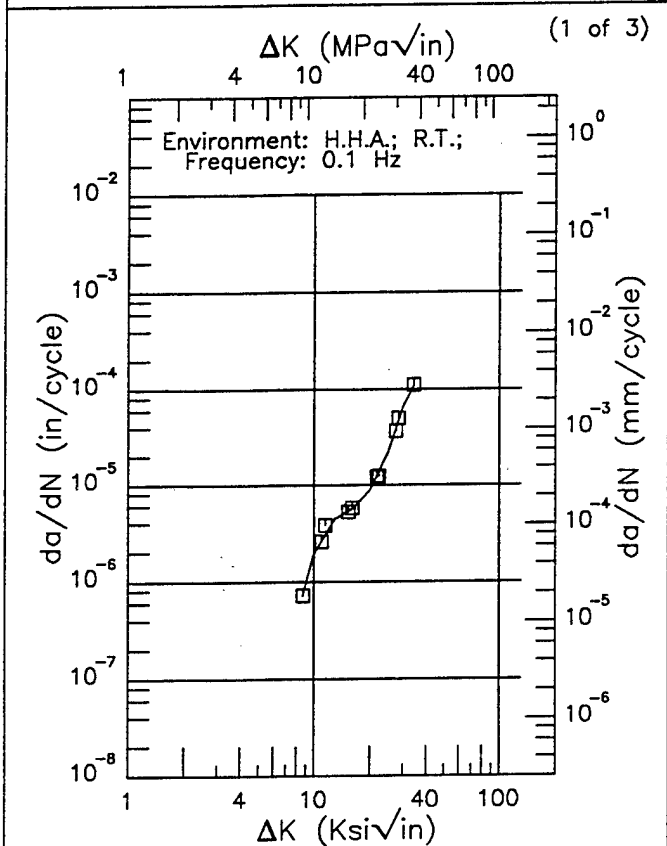
Figure 6.18.3.1.11

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EF | Ti-6Al-6V-2Sn |

Condition/Ht: STOA
 Form: 0.38 in. Plate
 Specimen Type: PTSF (max stress specified)
 Orientation: L-S
 Stress Ratio: 0.1

Yield Strength: 130 ksi
 Ult. Strength: 167.2 ksi
 Specimen Thk: 0.372 - 0.374 in.
 Specimen Width: 5 in.
 Ref: 90981



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
8.70 (min)	0.708
9.	0.949
10.	1.94
13.	4.44
16.	5.64
20.	8.89
25.	22.6
30.	65.9
34.23 (max)	110.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
9.04 (min)	0.592
10.	2.53
13.	7.01
16.	5.18
20.	7.63
25.	22.0
30.	51.7
35.	96.5
39.96 (max)	140.

RMS % Error	Life Prediction Ratio Summary
9.58	

RMS % Error	Life Prediction Ratio Summary
12.37	

Figure 6.18.3.1.12

Ti-6Al-6V-2Sn EF

Condition/Ht: STOA
 Form: 0.38 in. Plate
 Specimen Type: PTSF (max stress specified)
 Orientation: L-S
 Stress Ratio: 0.1

Yield Strength: 130 ksi
 Ult. Strength: 167.2 ksi
 Specimen Thk: 0.372 - 0.374 in.
 Specimen Width: 5 in.
 Ref: 90981

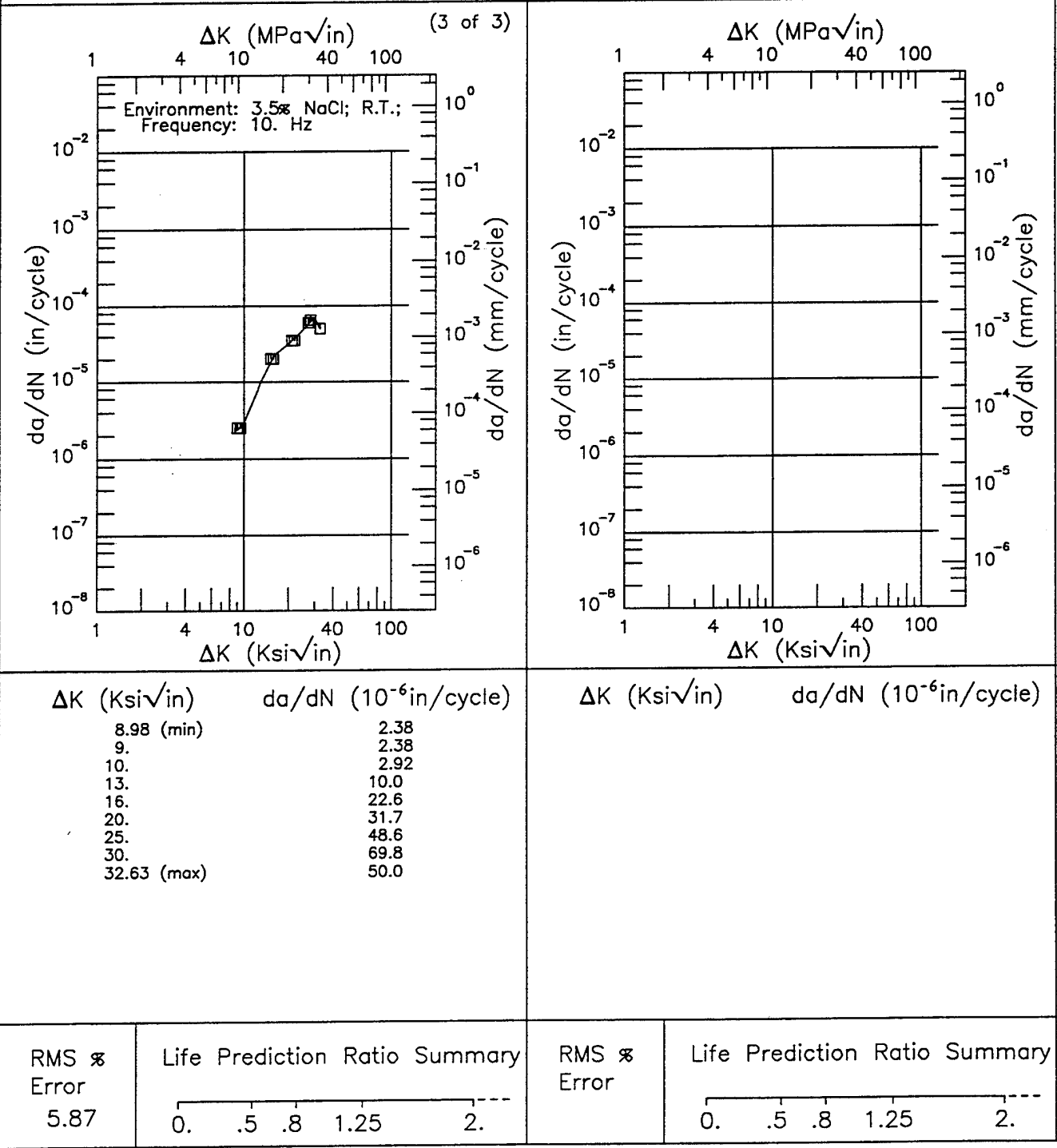
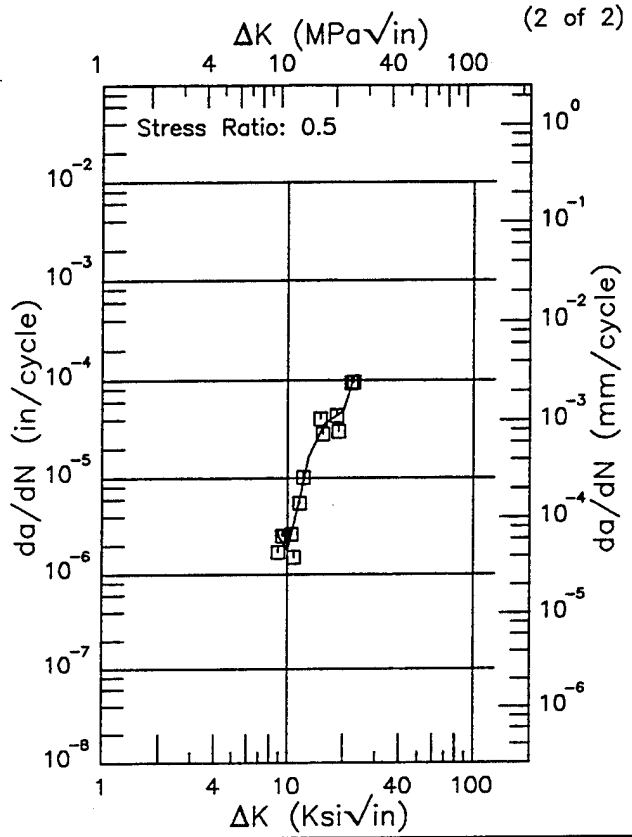
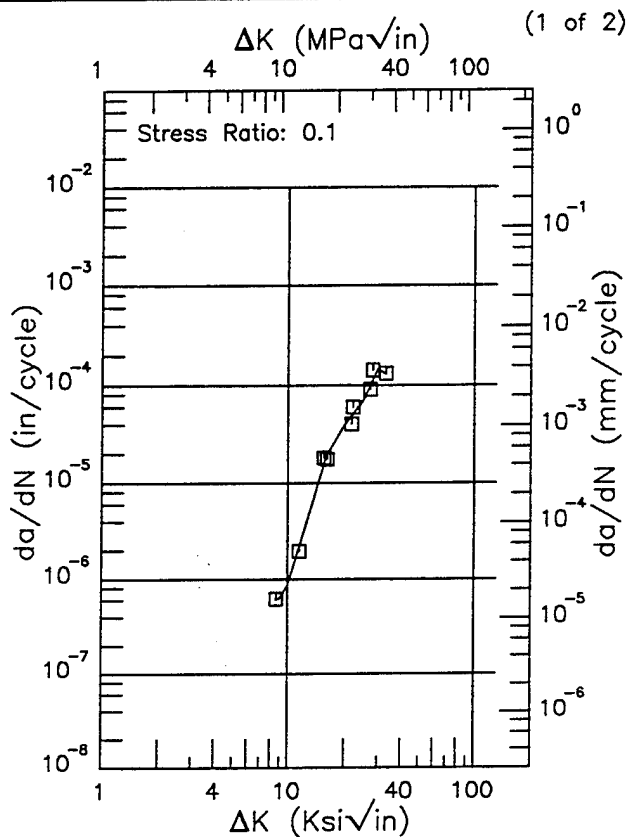


Figure 6.18.3.1.12 (Concluded)

R | Ti-6Al-6V-2Sn |

Condition/Ht: STOA
 Form: 0.38 in. Plate
 Specimen Type: PTSF (max stress specified)
 Orientation: T-S
 Frequency: 0.1 Hz
 Environment: H.H.A.; RT

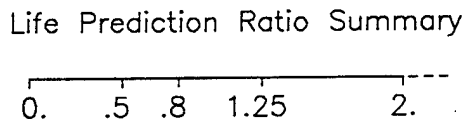
Yield Strength: 130 ksi
 Ult. Strength: 167.2 ksi
 Specimen Thk: 0.373 - 0.374 in.
 Specimen Width: 5 in.
 Ref: 90981



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.60 (min)	0.608
9.	0.630
10.	0.875
13.	4.81
16.	19.4
20.	39.8
25.	70.4
30.	148.
33.13 (max)	131.

ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
8.90 (min)	2.79
9.	2.47
10.	1.83
13.	16.6
16.	36.5
20.	48.8
22.70 (max)	102.

RMS %
 Error
 14.14



RMS %
 Error
 26.12

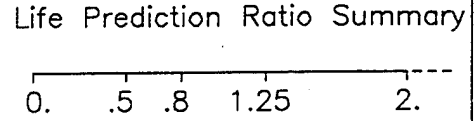
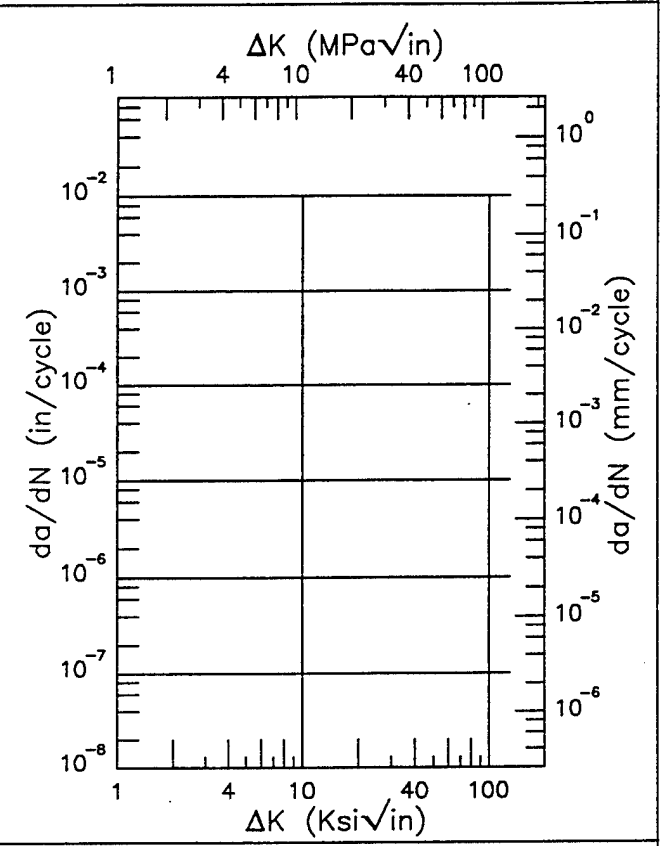
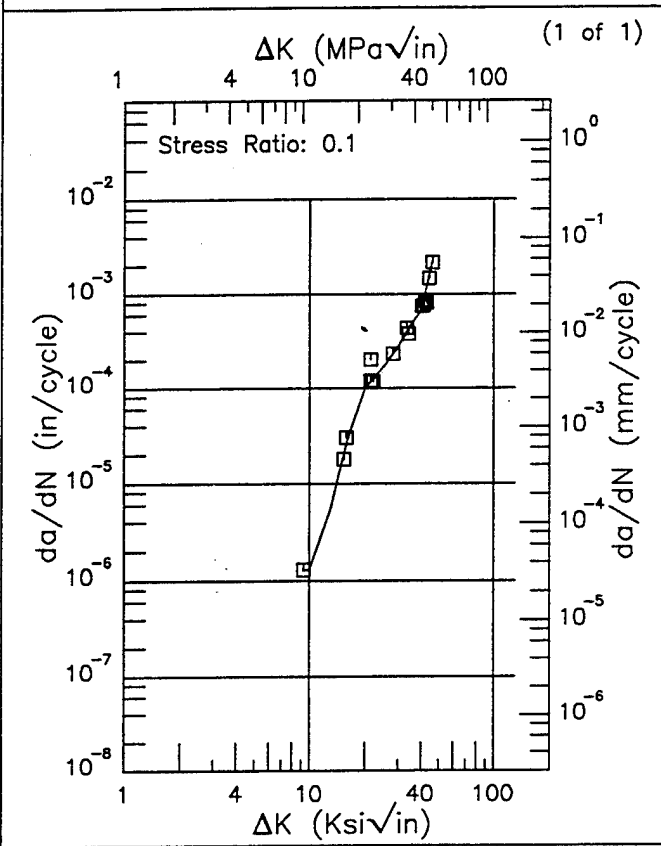


Figure 6.18.3.1.13

Ti-6Al-6V-2Sn R

Condition/Ht: STOA
 Form: 0.38 in. Plate
 Specimen Type: PTSF (max stress specified)
 Orientation: T-S
 Frequency: 1 Hz
 Environment: 3.5% NaCl; RT

Yield Strength: 156.5 ksi
 Ult. Strength: 167.8 ksi
 Specimen Thk: 0.375 in.
 Specimen Width: 5 in.
 Ref: 90981



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
9.26 (min)	1.28
10.	1.30
13.	5.25
16.	28.1
20.	100.1
25.	168.
30.	275.
35.	470.
40.	674.
45.85 (max)	2109.

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
9.26 (min)	1.28
10.	1.30
13.	5.25
16.	28.1
20.	100.1
25.	168.
30.	275.
35.	470.
40.	674.
45.85 (max)	2109.

RMS % Error	Life Prediction Ratio Summary
22.11	0. .5 .8 1.25 2.

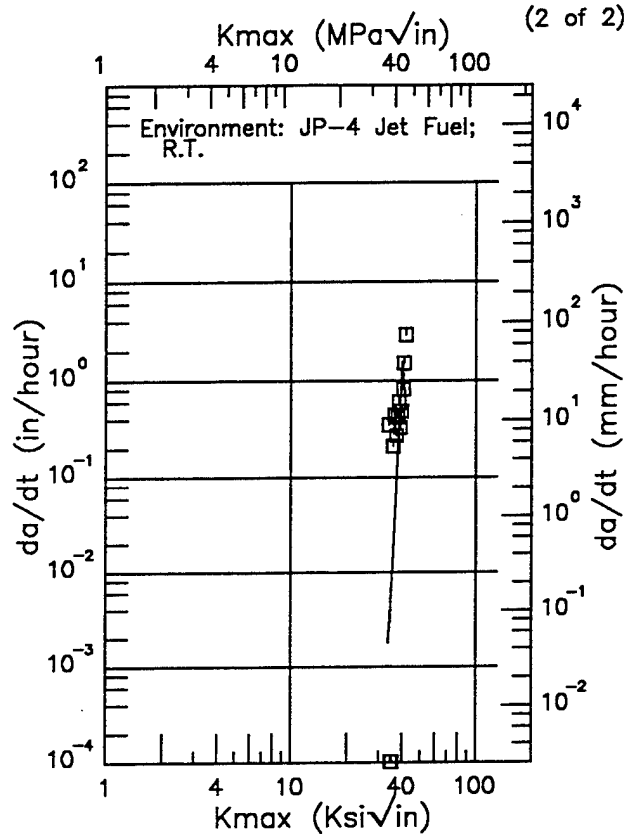
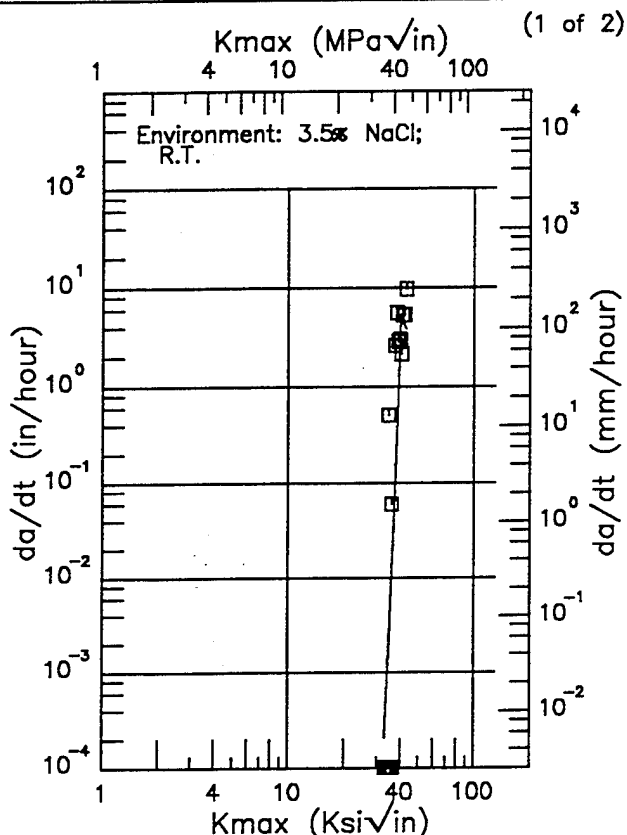
RMS % Error	Life Prediction Ratio Summary
	0. .5 .8 1.25 2.

Figure 6.18.3.1.14

Ti-6Al-6V-2Sn

Condition/Ht: 1300F 2HR
 Form: 2 in. Forging
 Specimen Type: TDCB
 Orientation: L-T
 Yield Strength: 146 ksi
 Ult. Strength:

Specimen Thk: 1.25 in.
 Specimen Width: 5.5 in.
 A₀:
 K_{Isc}: 31 - 32 ksi
 Ref: 84360



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
33.10 (min)	0.210
35.	3.68
40.	5322.
42.70 (max)	3848.

Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
33.70 (min)	1.79
35.	6.54
40.	1553.
41.50 (max)	631.

RMS %
 Error
 >100.0

RMS %
 Error
 >100.0

Figure 6.18.3.2

(1 of 1)

TABLE 6.19.3.3

 K_{Isec} SUMMARY FOR TITANIUM ALLOY Ti-6Al-6V-2.5Sn

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K_Q (Ksi√in)	K_{Isec} (Ksi√in)	Test Time (min)	Test Date	Reference
						Design	Width (in)	Thick (in)							
Unspecified	P	R.T.	---	186	3.5% NaCl	CANT	---	---	---	55	21	---	1967	70887	
1000° F 2hr AC	F	R.T.	L-T	155.1	JP-4 Fuel	TDCB	5	1.25	2.2	57.1	30.5	---	---	1971	84360
1300° F 2hr AC	F	R.T.	L-T	146.2	3.5% NaCl	TDCB	5	1.25	2.2	47.3	32.4	---	---	1971	84360
1550° F 1hr WQ 900° F 4hr AC	P	R.T.	T-S	179.6	3.5% NaCl	CANT*	1	0.75	1	55	21	---	---	1967	70931

* asterisk in specimen design column indicates that specimens are side-grooved

TABLE 6.20.2.1

TITANIUM Ti-6Al-2Sn-4Zr-6Mo K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic} \cdot TYS)^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (KSI $\sqrt{in.}$)	K_{Ic} MEAN	STAN DEV		
60% PRIMARY ALPHA	Forging	2.00	R.T.	...	153.0	...	2.000	0.21	43.96	R1005	
		2.00													
BETA PROCESSED	Forging	2.75	R.T.	...	156.6	...	2.000	0.52	71.44	R1005	
		2.75													
BU B FIN-10MA BETA UPSET BETA FINISHED 10% PRIMARY ALPHA MILL ANNEALED 1300 F 1 HR AC	Forging	2.75	R.T.	...	148.0	2.000	1.000	CT	1.000	0.09	26.50	...	1.0	88962 (1)	
		2.75													
		2.75													
BU B FIN-10STA BETA UPSET BETA FINISHED 10% PRIMARY ALPHA SOLUTION TREATED	Forging	2.75	R.T.	...	162.0	2.000	1.000	CT	1.000	0.10	31.30	...	0.6	88962 (1)	
		2.75													
		2.75													
BU B FIN-10STO BETA UPSET BETA FINISHED 10% PRIMARY ALPHA SOLUTION TREATED & OVERAGED 1625F 1 HR AC 1300F 1 HR AC	Forging	2.75	R.T.	...	148.0	2.000	1.000	CT	1.000	0.08	27.10	...	1.8	88962 (1)	
		2.75													
		2.75													
BU B FIN-60MA BETA UPSET BETA FINISHED 60% PRIMARY ALPHA MILL ANNEALED 1300F 1 HR AC	Forging	2.75	R.T.	...	154.0	2.000	1.000	CT	1.000	0.05	22.70	...	1.6	88962 (1)	
		2.75													
		2.75													
BU B FIN-60STA BETA UPSET BETA FINISHED 60% PRIMARY ALPHA SOLUTION TREATED & AGED 1625F 1 HR AC 1100F 8 HR AC	Forging	2.75	R.T.	...	167.0	2.000	1.000	CT	1.000	0.05	24.00	...	0.6	88962 (1)	
		2.75													
		2.75													

NOTES: (1) COMP. DISK

TABLE 6.20.2.1 (CONCLUDED)

TITANIUM Ti-6Al-2Sn-4Zr-6Mo K _{1c}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	2.5 * (K _{1c} /TYS) ² (in.)	K _{1c}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K _{1c} (KSI/in.)	K _{1c} MEAN	STAN DEV		
BU HABFIN10STA BETA UPSET HI ALPHA-BETA FINISHED 10% REDUCTION SOLUTION	Forging	2.75	R.T.	...	159.0	2.000	1.000	CT	1.000	0.16	40.50	41.4	0.9	1974	88962 (1)
		2.75									42.90				
		2.75									41.50				
BU HABFIN30STA BETA UPSET HI ALPHA-BETA FINISHED 30% REDUCTION SOLUTION TREATED	Forging	2.75	R.T.	...	161.0	2.000	1.000	CT	1.000	0.13	36.40	37.4	1.3	1974	88962 (1)
		2.75									38.90				
		2.75									36.90				
BU LABFIN10STA BETA UPSET LO ALPHA-BETA FINISHED 10% REDUCTION SOLUTION TREATED	Forging	2.75	R.T.	...	155.0	2.000	1.000	CT	1.000	0.21	44.80	44.7	1.1	1974	88962 (1)
		2.75									43.60				
		2.75									45.80				
STA-1625F 2 HR AC 1100F 8 HR AC	Forging	...	R.T.	...	163.0	2.000	1.000	CT	1.000	0.10	32.40	31.2	1.3	1974	88962 (2)
		...									32.70				
		...									31.60				
		...									29.20				
		...									30.20				
		...									31.20				

NOTES: (1) COMP. DISK
 (2) COMP. DISK COARSE GRAIN SIZE
 (3) COMP. DISK FINE GRAIN SIZE

Ti-8Al-1Mo-1V

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TABLE 6.21.1.2.1

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-8Al-1Mo-1V AT ROOM TEMPERATURE

ORIENTATION: L-T ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Ksi \sqrt{in})					
				2.5	6.0	10.0	20.0	60.0	100.0
1825F 1HR AC 1350F 2HRS AC	UNSPECIFIED	0.	0.33				10.22		
				0.04		0.51			
DA	SHEET	0.1	43				7.92	238.6	
				0.25				160.57	
MA	SHEET	0.1	43			2.54			
				0.02					
UNSPECIFIED	SHEET	0.02	0.1-12			2.28	13.54	144.35	

TABLE 6.21.1.2.2

FATIGUE CRACK GROWTH RATE AT DEFINED LEVELS OF STRESS INTENSITY FACTOR ΔK
 Ti-8Al-1Mo-1V AT ROOM TEMPERATURE

ORIENTATION: C-R ENVIRONMENT: Lab Air

CONDITION/ HEAT TREATMENT	PRODUCT FORM	R	FREQ (Hz)	FCGR (10^{-6} in/cycle)					
				ΔK Level (Kst \sqrt{in})					
				2.6	5.0	10.0	20.0	50.0	100.0
1830F 1HR WQ 1100F 8HRS AC	FORGING	0.1	30		0.13	1.14			

Ti-8Al-1Mo-1V

1 of 1

TABLE 6.21.2.1

TITANIUM Ti-8Al-1Mo-1V K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Kgf)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 * (K_{Ic}/\sqrt{in.})^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Kgf $\sqrt{in.}$)	K_{Ic} MEAN	STAN DEV		
1830F 1 HR WQ 1100F 8 HRS AC	Forging	...	R.T.	C-R	142.0	2.500	0.500	0.18	38.22	1977	PW002

TABLE 6.21.2.2

TI-8AL-1MO-1V K _C																			
CONDITION HEAT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K _{app}			K _C			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	INIT (in.) 2a _i	FINAL (in.) 2a _f	ONSET (Ksi) σ _s	MAX (Ksi) σ _{max}	K _{app} (Ksi√in.)	K _{app} MEAN	STAN DEV	K _C (Ksi√in.)	K _C MEAN	STAN DEV		
BUCKLING OF CRACK EDGES RESTRAINED																			
DA	Sheet	0.05	R.T.	L-T	133.6	8.000	0.060	2.100	2.100	75.20	105.00	134.26*	199.24*	1968	71709				
		8.000				0.060	3.990	3.990	52.90	79.20	145.62	206.08*	1968	71709					
		8.000				0.060	5.010	5.700	19.00	33.70	127.02	162.64*	13.2	1968	71709				
		8.000				0.060	4.790	4.790	27.00	79.90	213.08*	285.48*	1968	71709					
		8.000				0.060	1.500	3.060	62.00	87.50	137.30*	210.73*	1968	71709					
DA	Sheet	0.02	R.T.	L-T	135.5	9.000	0.020	2.110	2.630	62.20	73.30	138.16	167.35	1966	67821				
DA	Sheet	0.04	R.T.	L-T	132.6	9.000	0.045	2.100	2.970	45.90	66.10	124.25	153.19	1966	67821				
DA	Sheet	0.02	R.T.	L-T	135.5	12.000	0.020	2.020	2.360	41.60	47.90	86.95	94.49	1966	67821				
		12.000				0.020	0.960	1.240	75.00	85.00	104.24	110.42	1966	67821					
		12.000				0.020	0.580	0.700	95.00	110.20	105.34*	115.80*	10.0	111.7	15.0	1966	67821		
		12.000				0.020	2.120	2.800	42.50	55.90	104.02	121.33	1966	67821					
		20.000				0.050	2.060	5.000	62.40	85.90	155.54	250.46*	1968	71709					
DA	Sheet	0.05	R.T.	L-T	133.6	20.000	0.060	2.020	4.000	29.10	77.40	138.75	199.94	1968	71709				
		20.000				0.060	8.020	14.500	26.50	42.70	188.60	314.96*	11.8	220.5	15.8	1968	71709		
		20.000				0.050	9.940	15.000	22.80	36.40	170.65	265.62*	1968	71709					

* NOTE: NET SECTION STRESS EXCEEDS 80% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

TI-8Al-1Mo-1V

2 of 2

TABLE 6.21.2.2 (CONCLUDED)

TI-8Al-1Mo-1V K_C																			
CONDITION HEAT TREAT	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN		CRACK LENGTH		GROSS STRESS		K_{app}			K_C			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	INIT (in.) $2a_i$	FINAL (in.) $2a_f$	ONSET (KSI) σ_o	MAX (KSI) σ_{max}	K_{app} (KSI/in.)	K_{app} MEAN	STAN DEV	K_C (KSI/in.)	K_C MEAN	STAN DEV		
DA Cont'd	Sheet Cont'd	0.06	20.000	0.050	6.040	10.000	26.80	47.60	155.45	224.35	1968	71709	Cont'd	Cont'd	Cont'd	Cont'd	Cont'd	1968	71709
		0.05	20.000	0.050	9.980	15.500	19.30	30.50	143.50	259.41*	1968	71709							
		0.05	20.000	0.050	6.800	11.080	36.40	45.60	160.64	236.94	1968	71709							
		0.05	20.000	0.050	4.260	7.500	36.90	58.90	156.77	221.71	1968	71709							
		0.05	20.000	0.050	3.760	7.520	48.60	69.80	173.43	263.23*	1968	71709							
		0.05	8.000	0.050	4.910	5.700	26.90	37.80	139.02	171.22*	1968	71709							
		0.05	8.000	0.050	1.000	2.250	65.00	105.00	132.88*	207.62*	1968	71709							
		0.05	8.000	0.050	2.940	4.250	39.40	62.00	145.55	195.48*	1968	71709							
		0.05	8.000	0.050	2.020	3.650	42.90	80.10	148.56	220.88*	1968	71709							
		0.05	8.000	0.050	1.820	3.100	51.70	91.80	145.09*	223.65*	1968	71709							
BUCKLING OF CRACK SIDGES RESTRAINED																			
DA	Sheet	0.05	8.000	0.050	2.940	4.250	39.40	62.00	145.55	144.4	4.9	1968	71709	1968	71709
		0.05	8.000	0.050	2.020	3.650	42.90	80.10	148.56	220.88*	1968	71709	1968	71709	
		0.05	8.000	0.050	1.820	3.100	51.70	91.80	145.09*	223.65*	1968	71709	1968	71709	
		0.05	8.000	0.050	1.820	3.100	51.70	91.80	145.09*	223.65*	1968	71709	1968	71709	

* NOTE: NET SECTION STRESS EXCEEDS 86% OF YIELD STRENGTH. VALUE NOT INCLUDED IN MEAN OR STANDARD DEVIATION.

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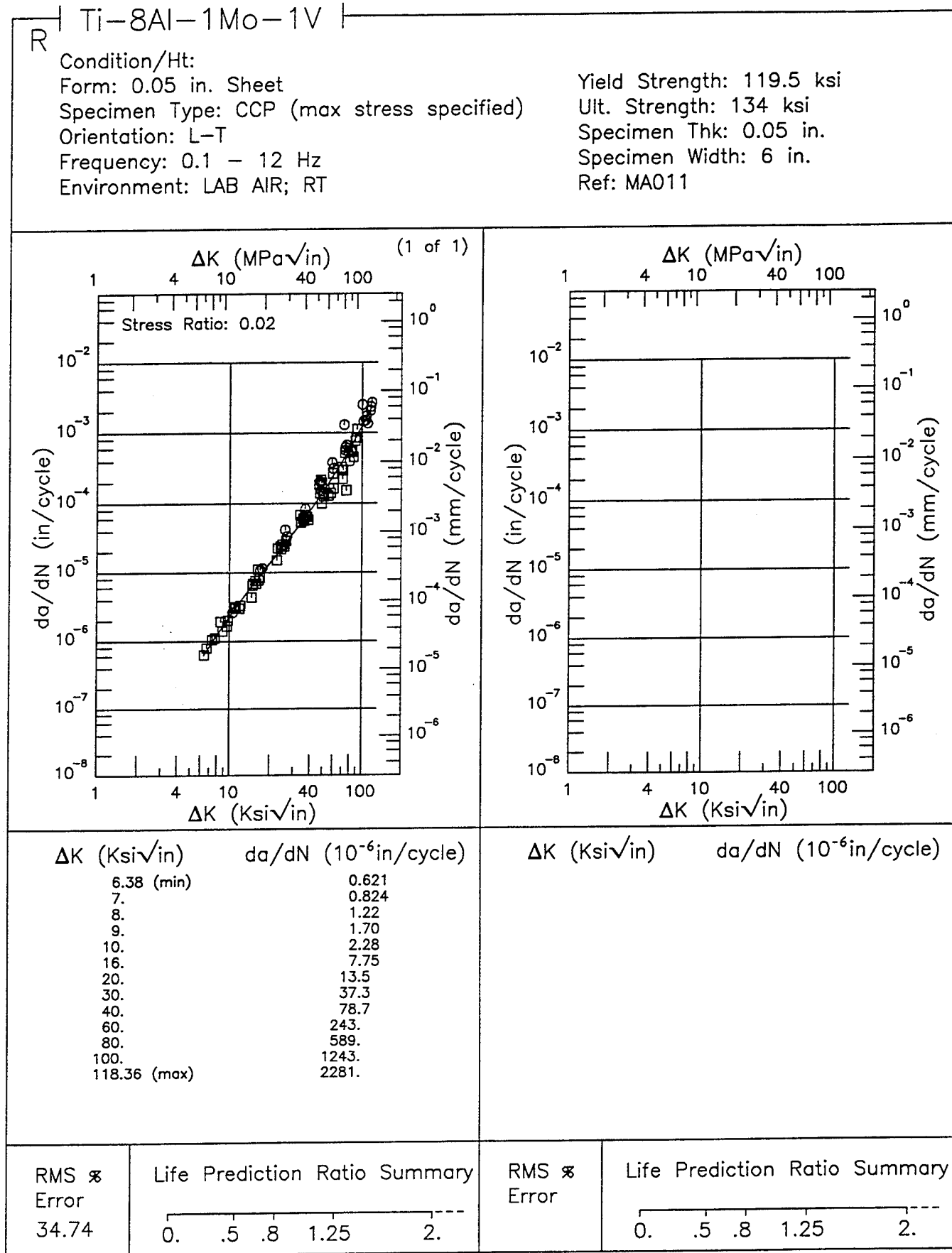


Figure 6.21.3.1.1

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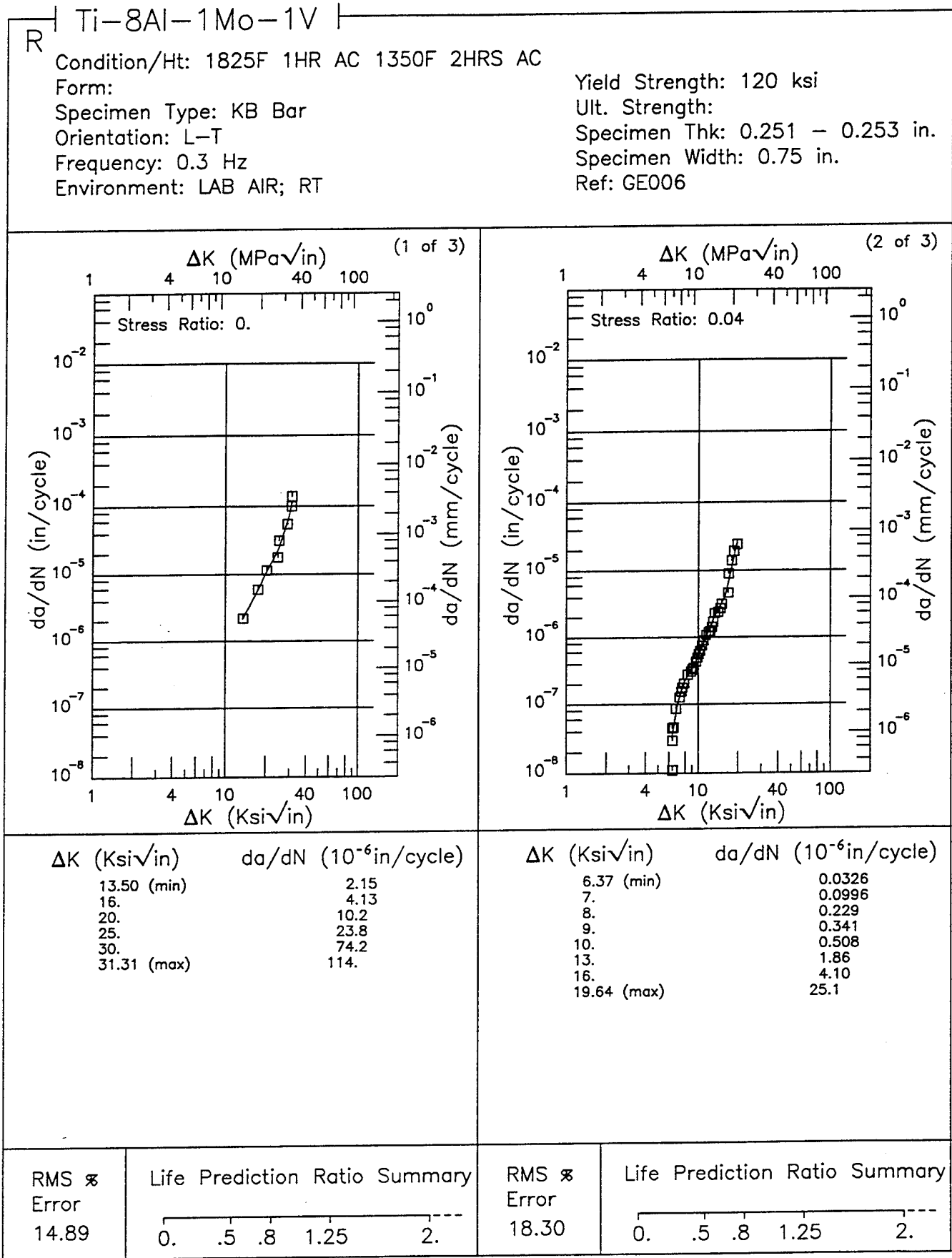


Figure 6.21.3.1.2

Ti-8Al-1Mo-1V R

Condition/Ht: 1825F 1HR AC 1350F 2HRS AC
 Form:
 Specimen Type: KB Bar
 Orientation: L-T
 Frequency: 0.3 Hz
 Environment: LAB AIR; RT

Yield Strength: 120 ksi
 Ult. Strength:
 Specimen Thk: 0.251 - 0.253 in.
 Specimen Width: 0.75 in.
 Ref: GE006

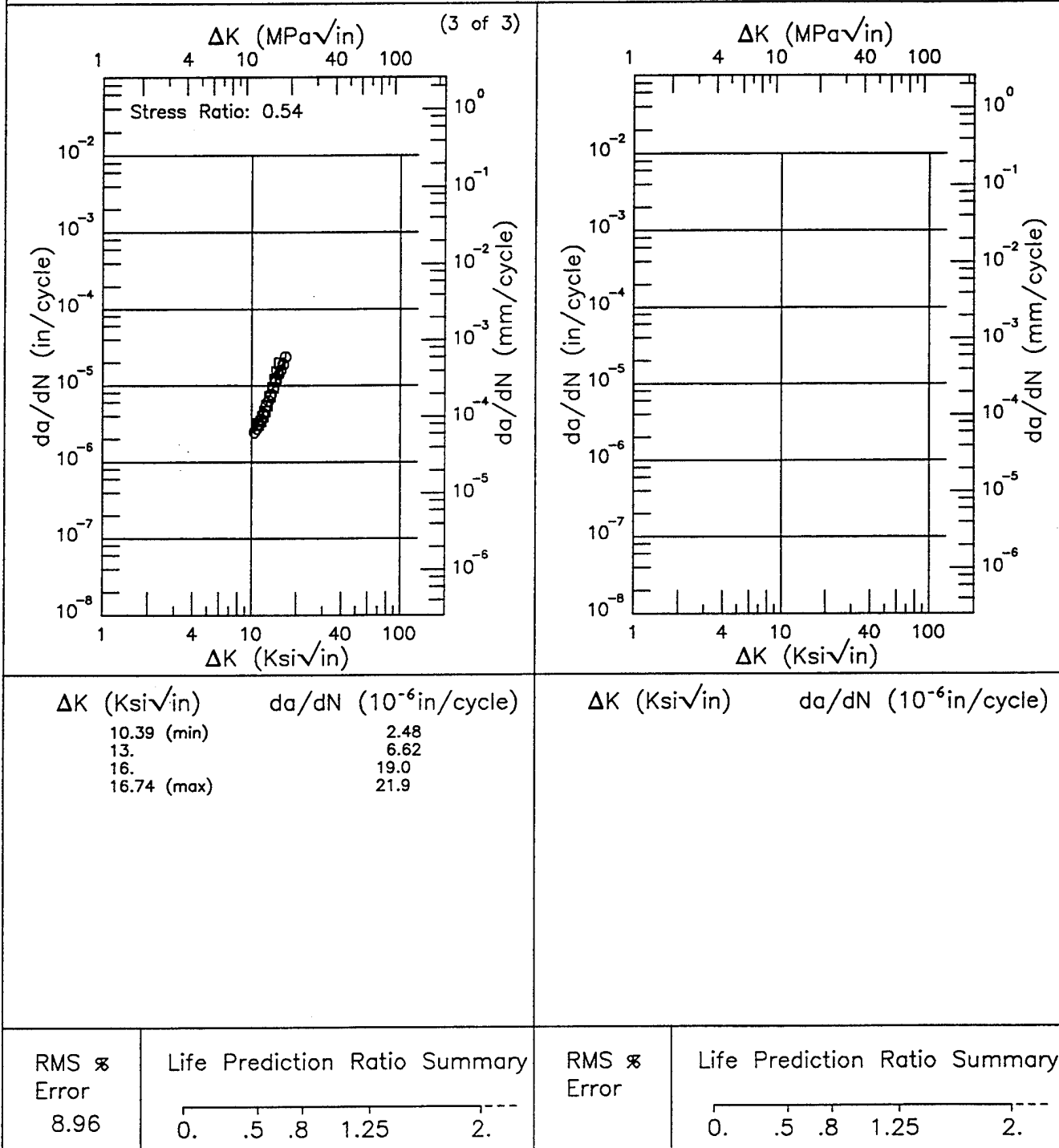
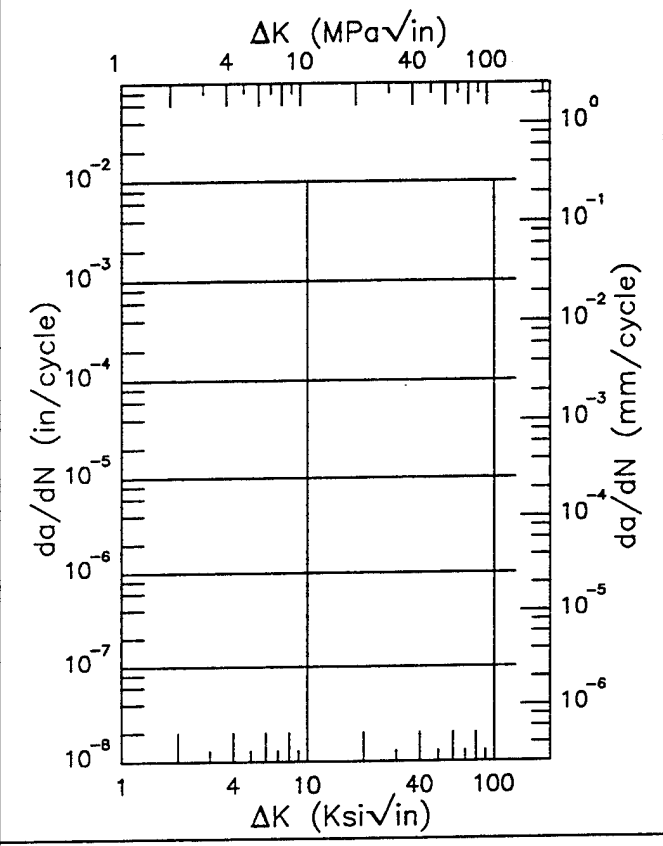
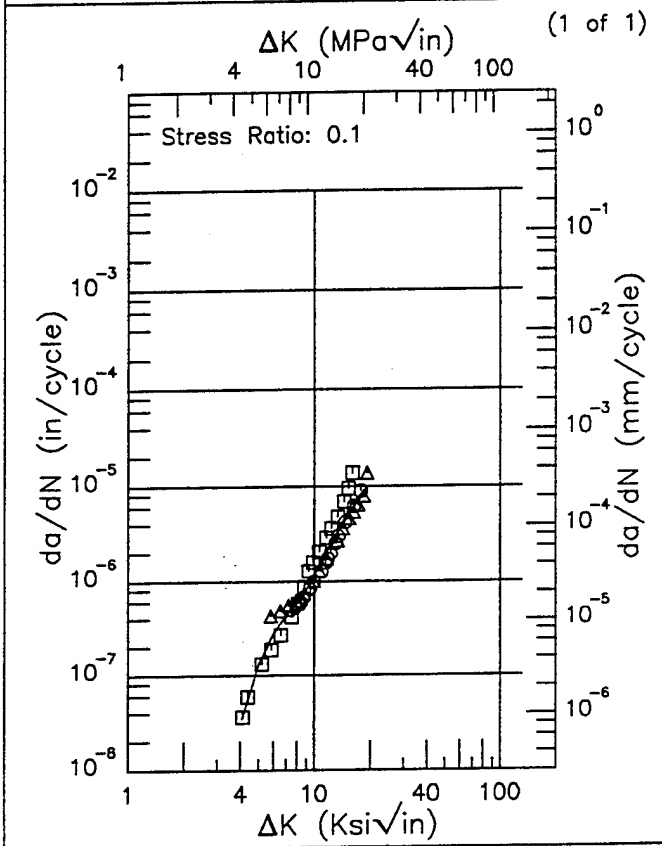


Figure 6.21.3.1.2 (Concluded)

R | **Ti-8Al-1Mo-1V** |

Condition/Ht: 1830F 1HR WQ 1100F 8HRS AC
 Form: 1 in. Forging
 Specimen Type: CCP (max load specified)
 Orientation: C-R
 Frequency: 30 Hz
 Environment: LAB AIR; RT

Yield Strength: 135.5 - 145.8 ksi
 Ult. Strength: 148.5 - 159.3 ksi
 Specimen Thk: 0.078 - 0.08 in.
 Specimen Width: 1.75 in.
 Ref: PW002



ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
4.09 (min)	0.0347
5.	0.130
6.	0.277
7.	0.433
8.	0.604
9.	0.824
10.	1.14
13.	3.04
16.	6.83
19.11 (max)	9.55

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
4.09 (min)	0.0347
5.	0.130
6.	0.277
7.	0.433
8.	0.604
9.	0.824
10.	1.14
13.	3.04
16.	6.83
19.11 (max)	9.55

RMS % Error 31.78	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
----------------------	---

RMS % Error	Life Prediction Ratio Summary 0. .5 .8 1.25 2.
-------------	---

Figure 6.21.3.1.3

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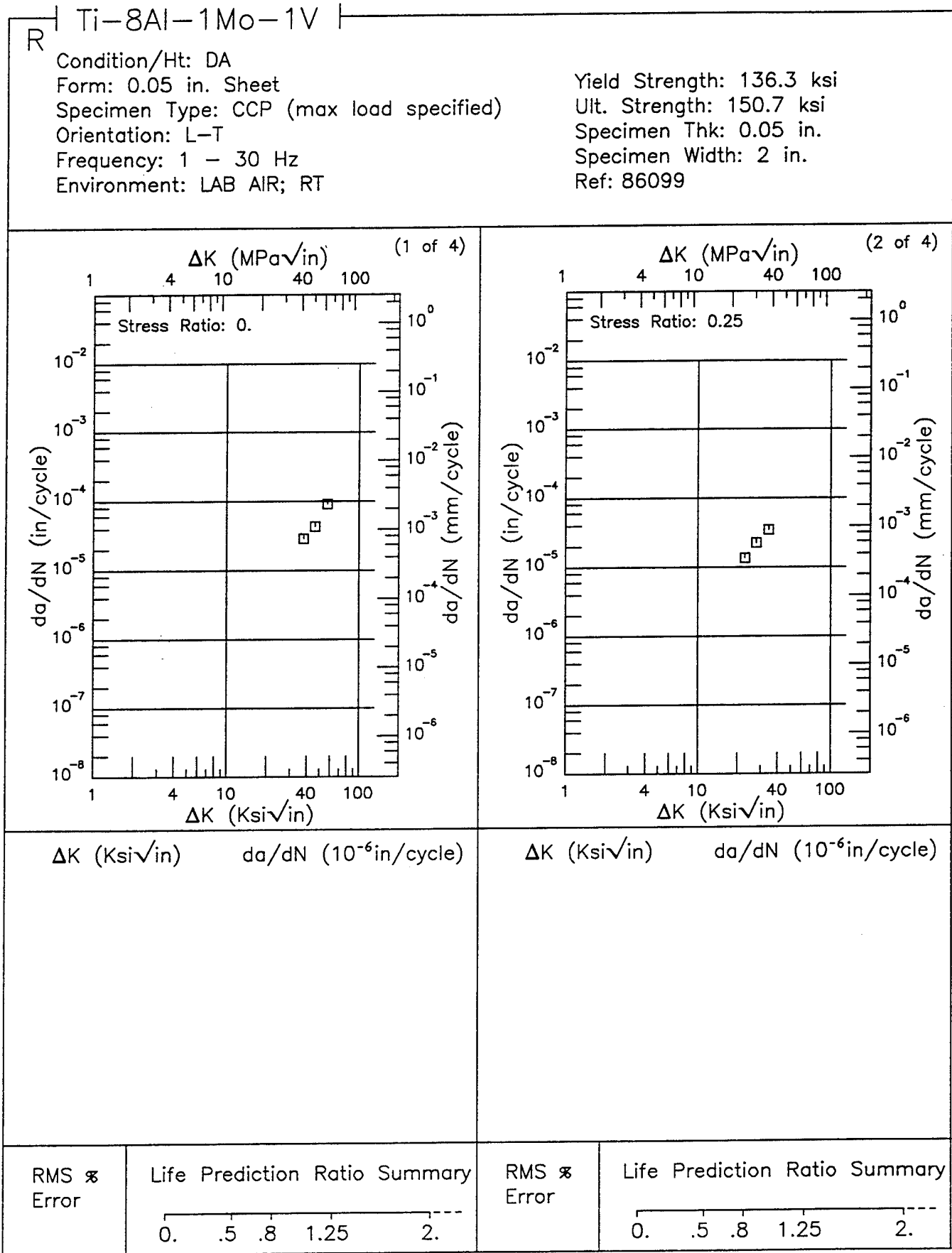


Figure 6.21.3.1.4

Ti-8Al-1Mo-1V R

Condition/Ht: DA
 Form: 0.05 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 1 - 30 Hz
 Environment: LAB AIR; RT

Yield Strength: 136.3 ksi
 Ult. Strength: 150.7 ksi
 Specimen Thk: 0.05 in.
 Specimen Width: 2 in.
 Ref: 86099

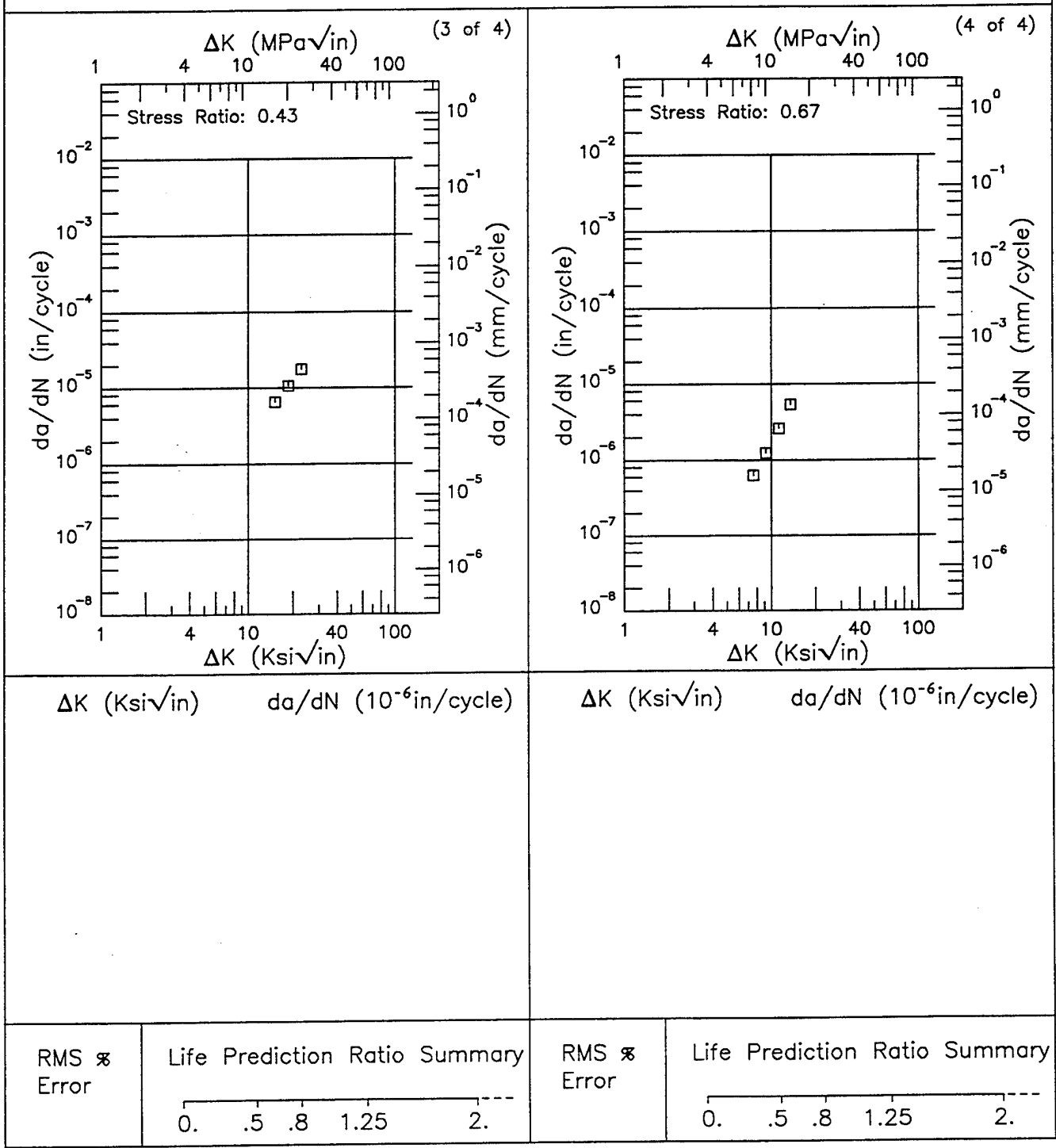
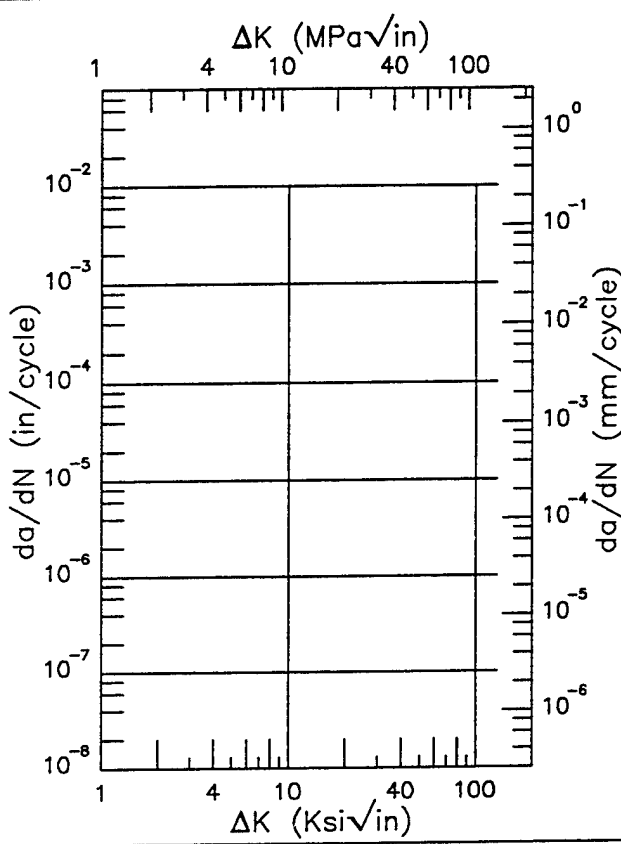
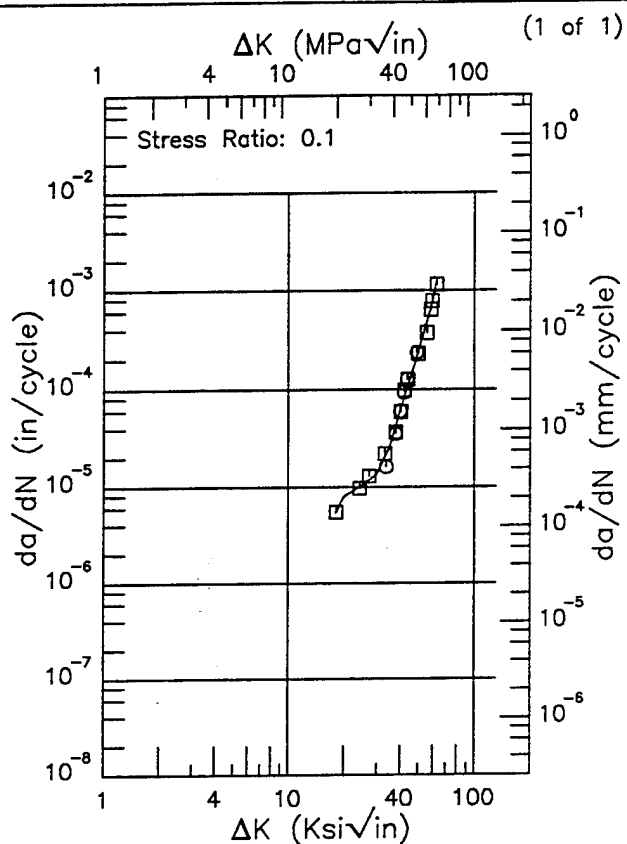


Figure 6.21.3.1.4 (Concluded)

R | Ti-8Al-1Mo-1V |

Condition/Ht: DA
 Form: 0.05 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 43 Hz
 Environment: LAB AIR; RT

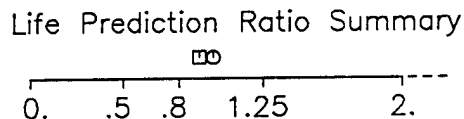
Yield Strength: 140 ksi
 Ult. Strength: 149.2 ksi
 Specimen Thk: 0.064 in.
 Specimen Width: 3 in.
 Ref: 86099



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
17.97 (min)	5.41
20.	7.92
25.	10.5
30.	13.6
35.	25.8
40.	61.9
50.	239.
60.	792.
62.14 (max)	1190.

ΔK (Ksi√in) da/dN (10^{-6} in/cycle)

RMS %
 Error
 13.38



RMS %
 Error

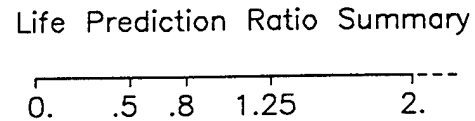


Figure 6.21.3.1.5

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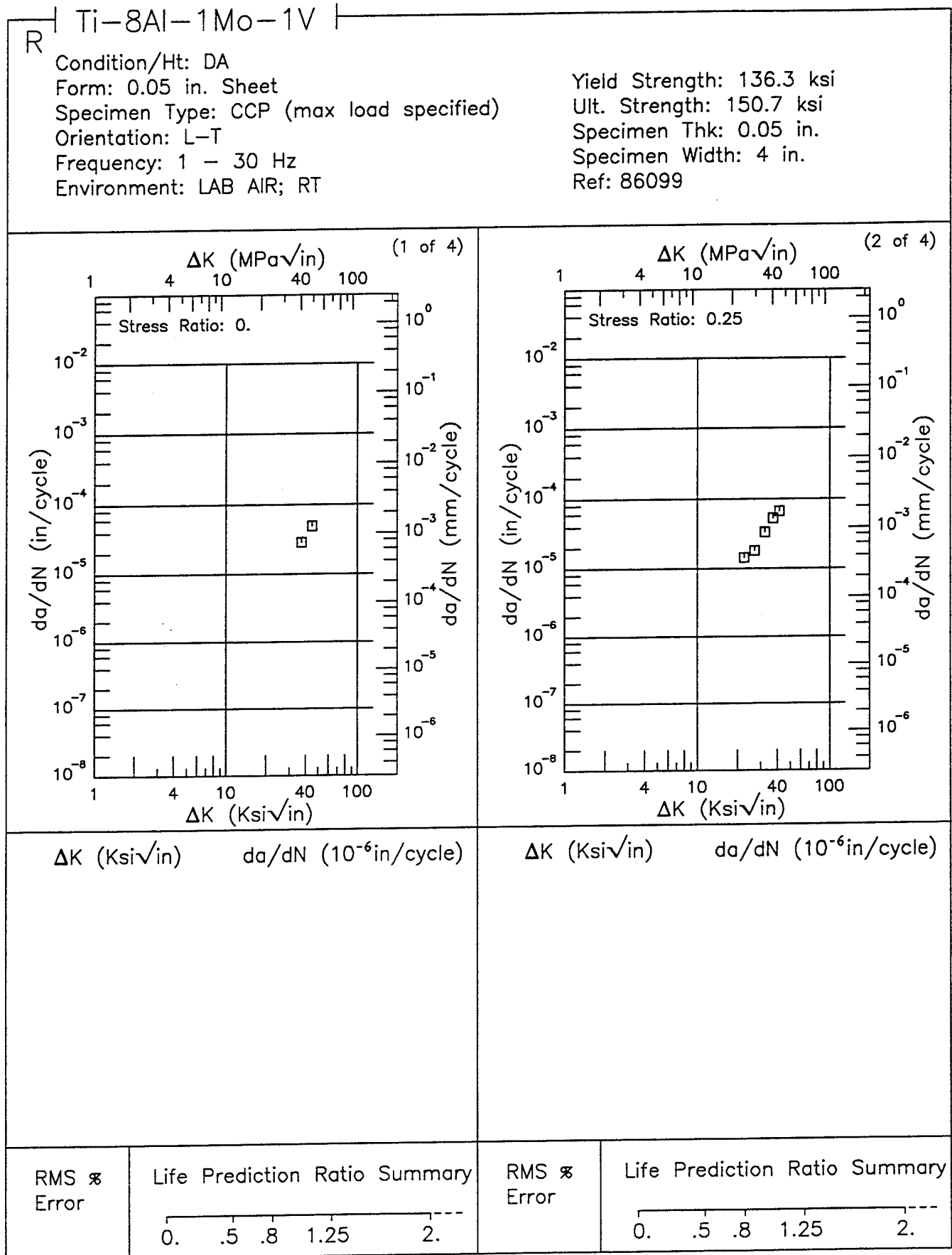


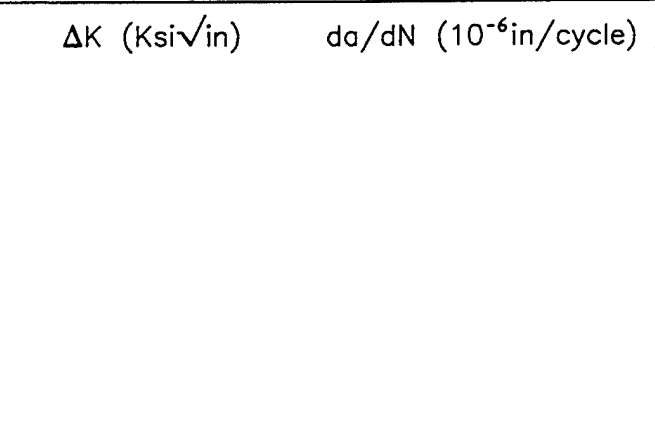
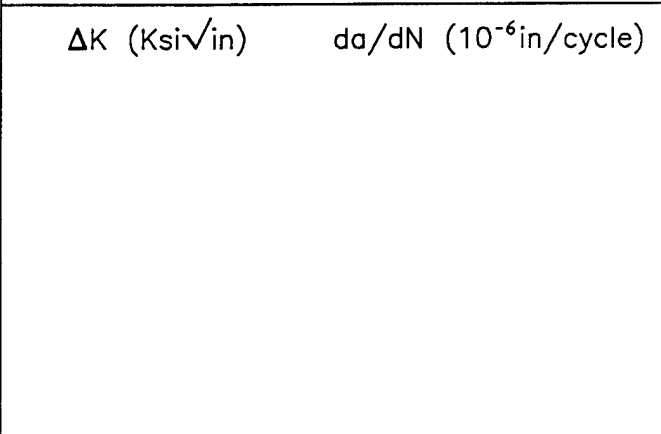
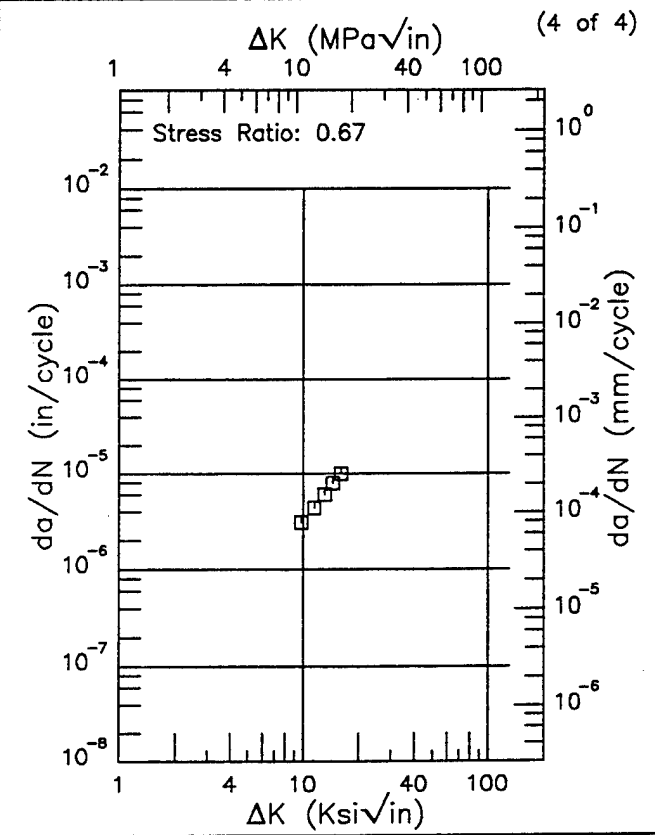
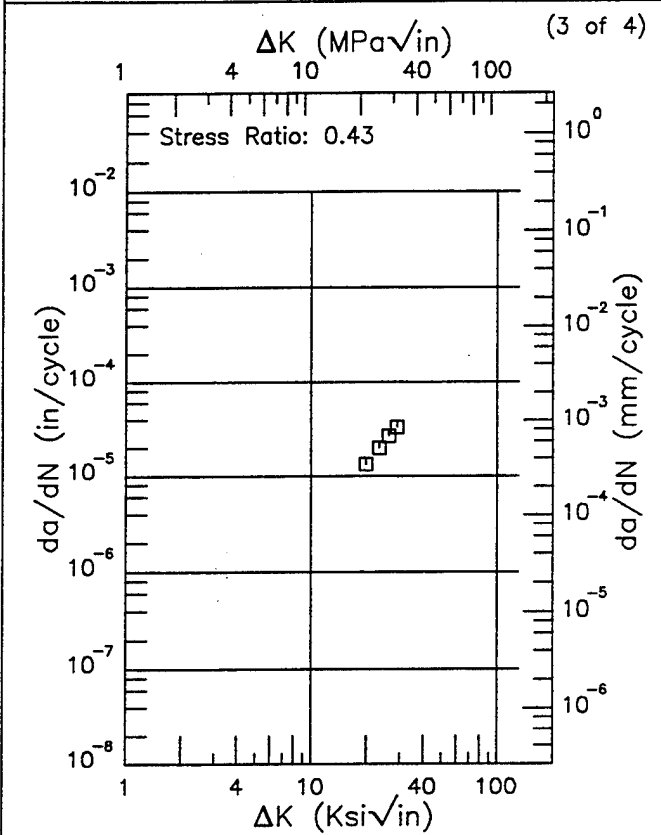
Figure 6.21.3.1.6

Ti-8Al-1Mo-1V

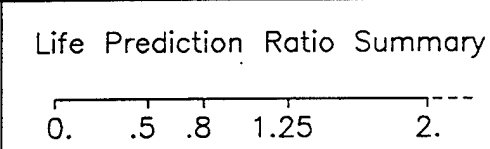
R

Condition/Ht: DA
 Form: 0.05 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 1 - 30 Hz
 Environment: LAB AIR; RT

Yield Strength: 136.3 ksi
 Ult. Strength: 150.7 ksi
 Specimen Thk: 0.05 in.
 Specimen Width: 4 in.
 Ref: 86099



RMS % Error



RMS % Error

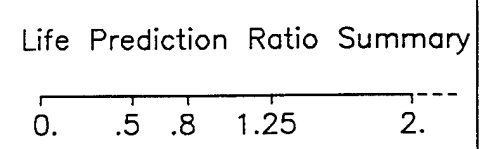
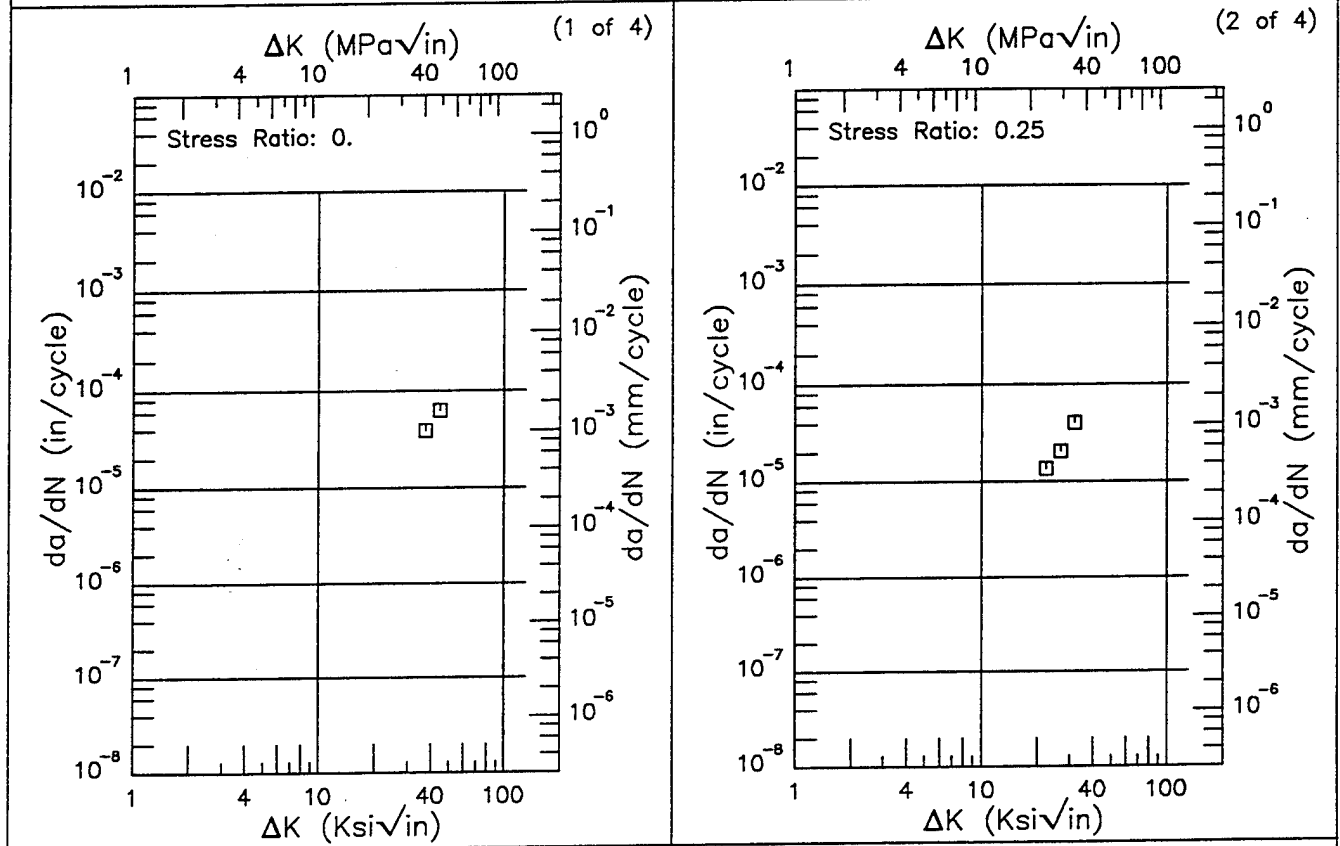


Figure 6.21.3.1.6 (Concluded)

R | Ti-8Al-1Mo-1V |

Condition/Ht: DA	Yield Strength: 136.3 ksi
Form: 0.05 in. Sheet	Ult. Strength: 150.7 ksi
Specimen Type: CCP (max load specified)	Specimen Thk: 0.05 in.
Orientation: L-T	Specimen Width: 7.992 in.
Frequency: 1 - 30 Hz	Ref: 86099
Environment: LAB AIR; RT	



ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)	ΔK (Ksi√in)	da/dN (10^{-6} in/cycle)
---------------------	-------------------------------	---------------------	-------------------------------

RMS % Error	Life Prediction Ratio Summary	RMS % Error	Life Prediction Ratio Summary
	0. .5 .8 1.25 2.		0. .5 .8 1.25 2.

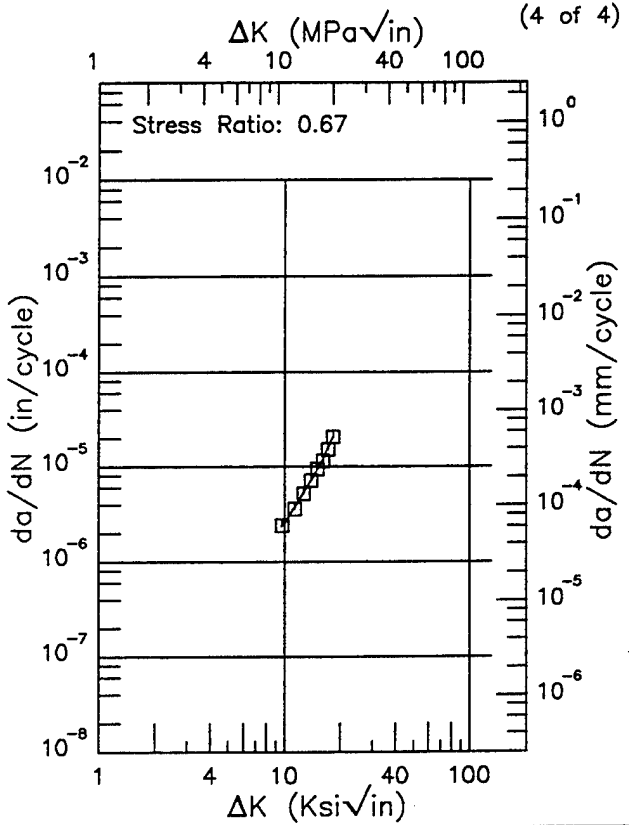
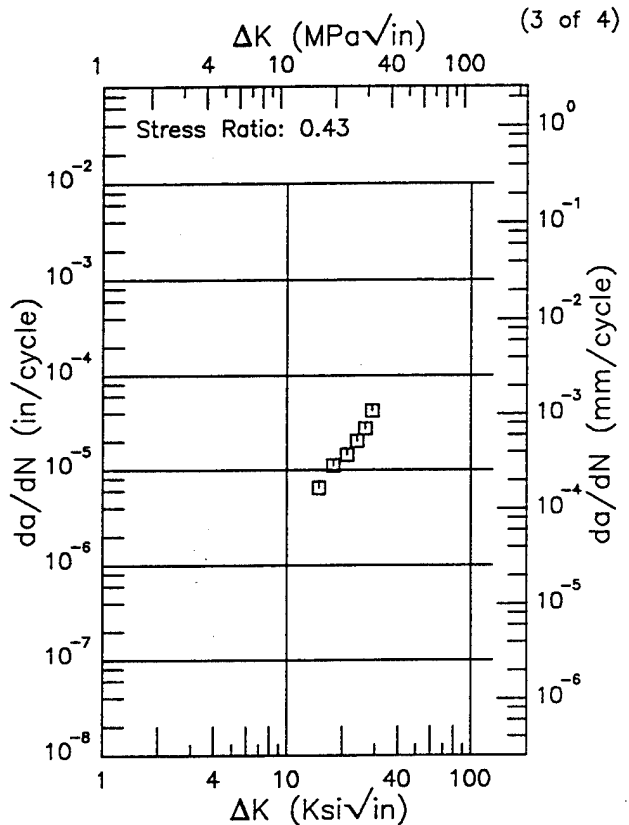
Figure 6.21.3.1.7

Ti-8Al-1Mo-1V

R

Condition/Ht: DA
 Form: 0.05 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 1 - 30 Hz
 Environment: LAB AIR; RT

Yield Strength: 136.3 ksi
 Ult. Strength: 150.7 ksi
 Specimen Thk: 0.05 in.
 Specimen Width: 7.992 in.
 Ref: 86099

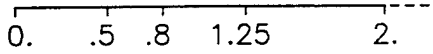


ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
9.65 (min)	2.38
10.	2.54
13.	5.76
16.	11.3
18.26 (max)	20.3

ΔK (Ksi $\sqrt{\text{in}}$)	da/dN (10^{-6} in/cycle)
9.65 (min)	2.38
10.	2.54
13.	5.76
16.	11.3
18.26 (max)	20.3

RMS % Error

Life Prediction Ratio Summary



RMS % Error

2.13

Life Prediction Ratio Summary

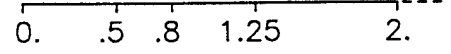


Figure 6.21.3.1.7 (Concluded)

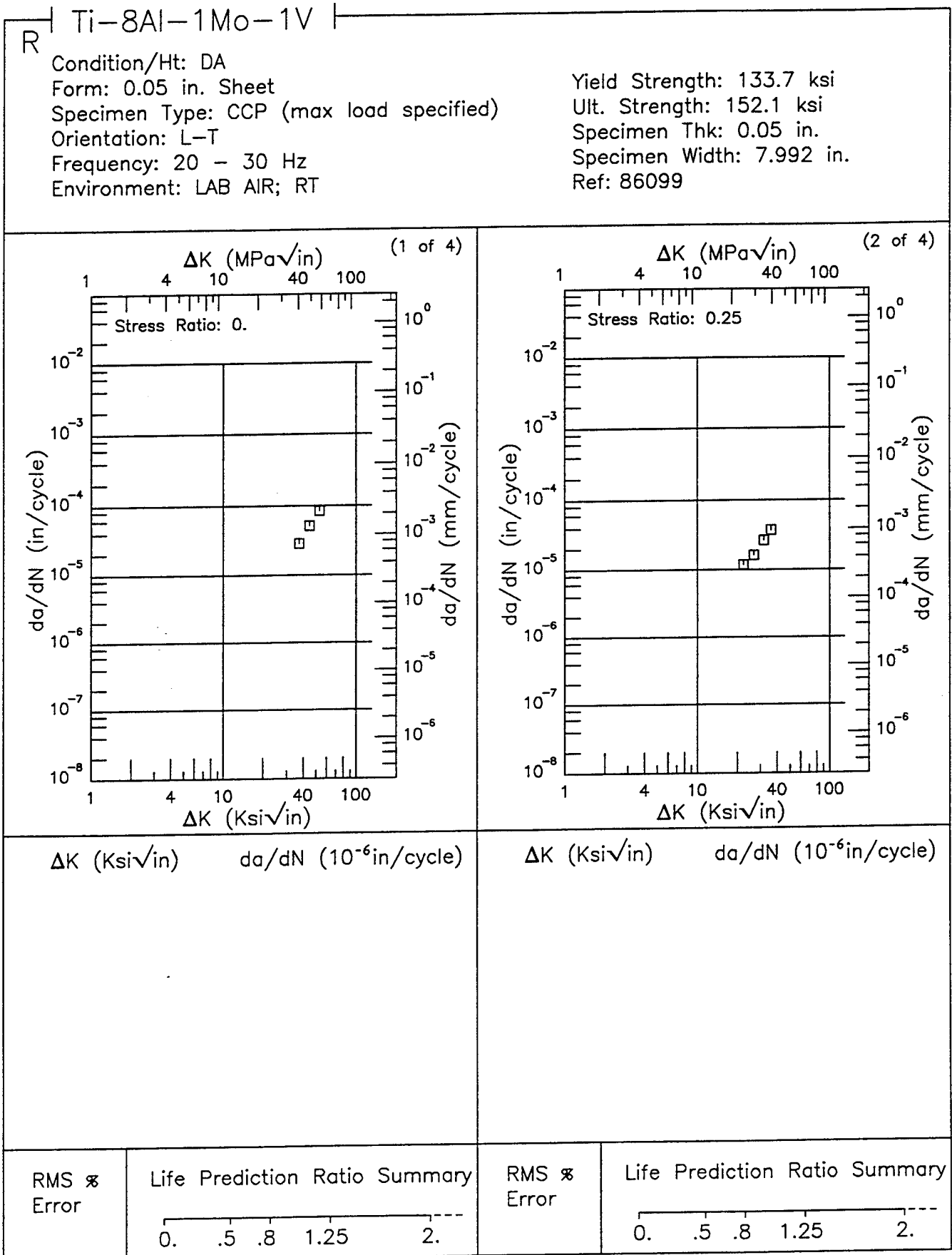


Figure 6.21.3.1.8

Ti-8Al-1Mo-1V

R

Condition/Ht: DA
 Form: 0.05 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 20 - 30 Hz
 Environment: LAB AIR; RT

Yield Strength: 133.7 ksi
 Ult. Strength: 152.1 ksi
 Specimen Thk: 0.05 in.
 Specimen Width: 7.992 in.
 Ref: 86099

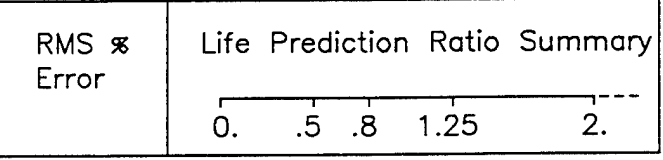
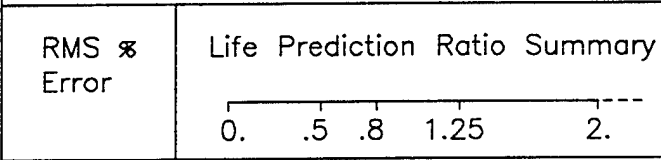
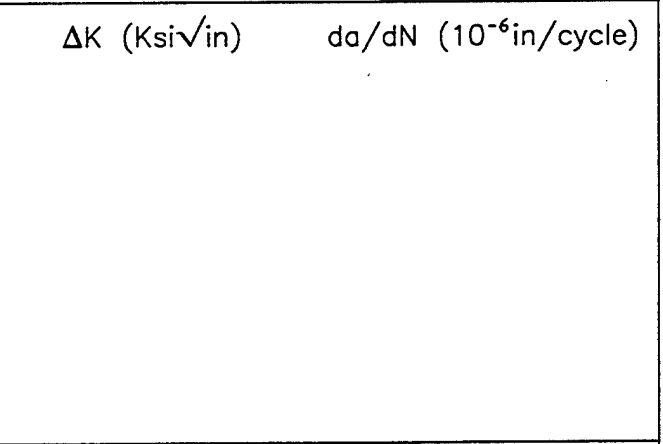
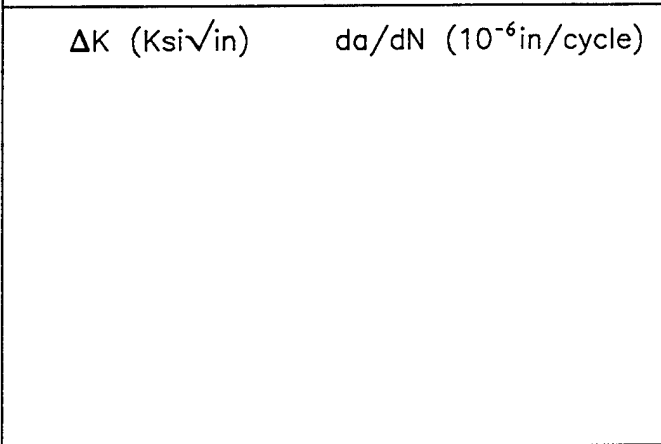
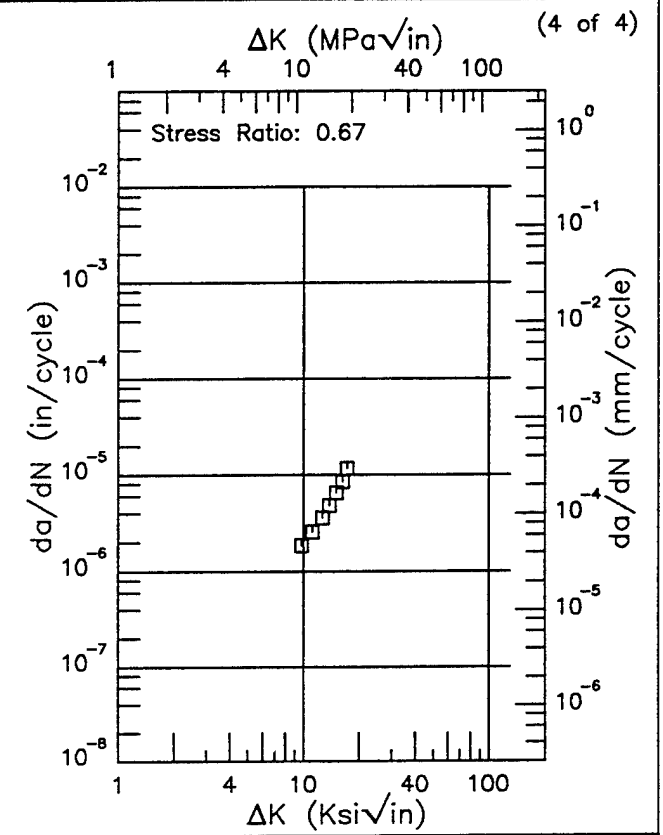
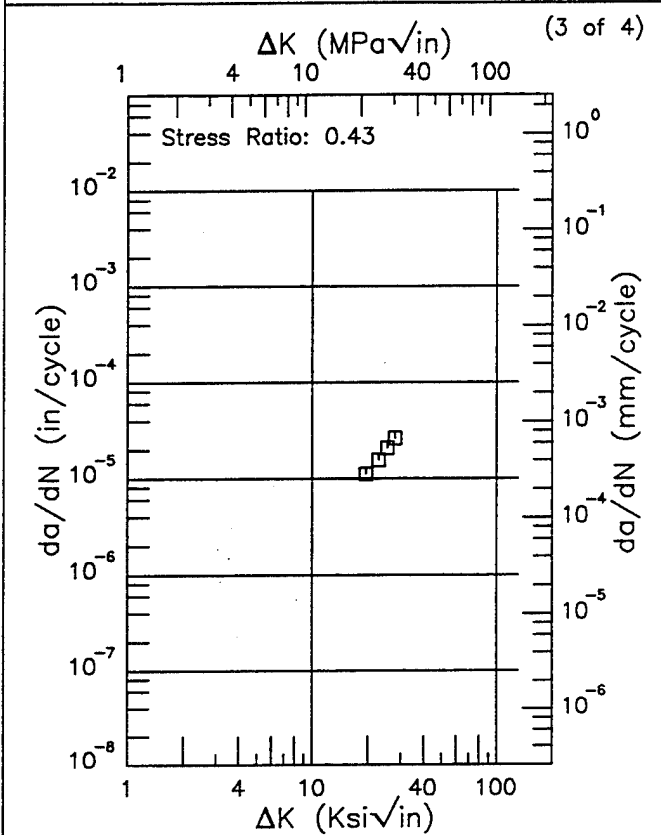


Figure 6.21.3.1.8 (Concluded)

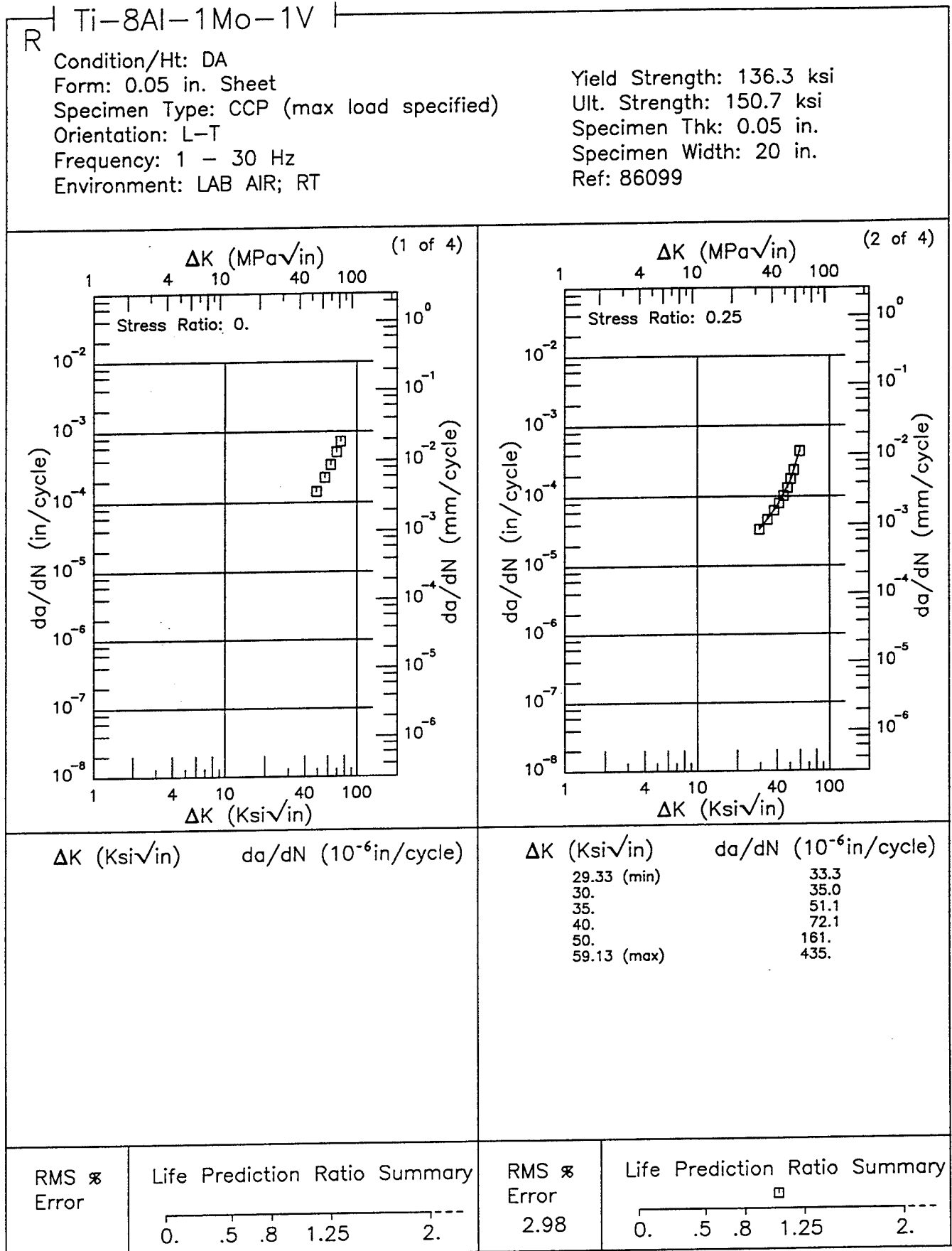


Figure 6.21.3.1.9

Ti-8Al-1Mo-1V

R

Condition/Ht: DA
 Form: 0.05 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation: L-T
 Frequency: 1 - 30 Hz
 Environment: LAB AIR; RT

Yield Strength: 136.3 ksi
 Ult. Strength: 150.7 ksi
 Specimen Thk: 0.05 in.
 Specimen Width: 20 in.
 Ref: 86099

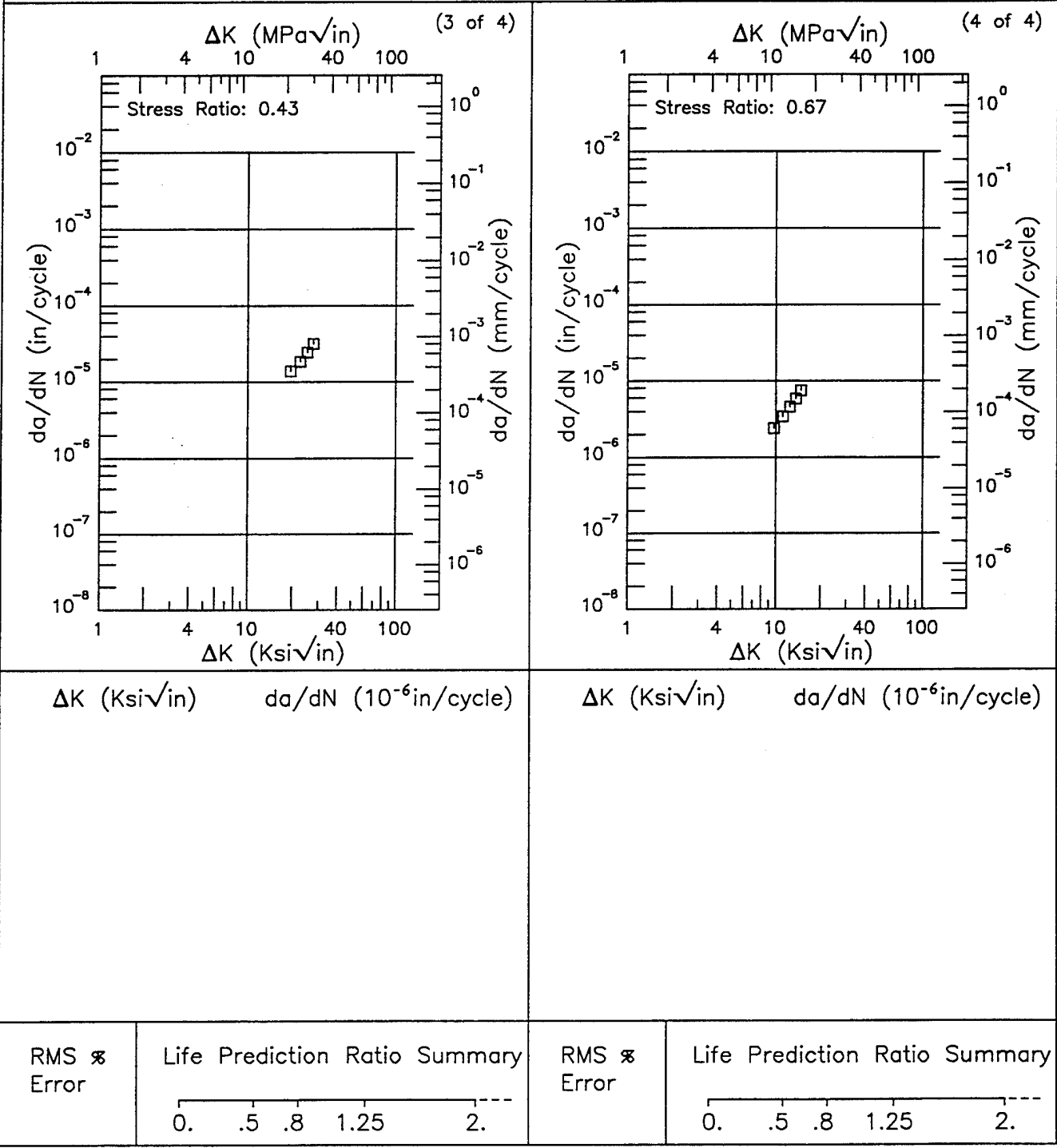


Figure 6.21.3.1.9 (Concluded)

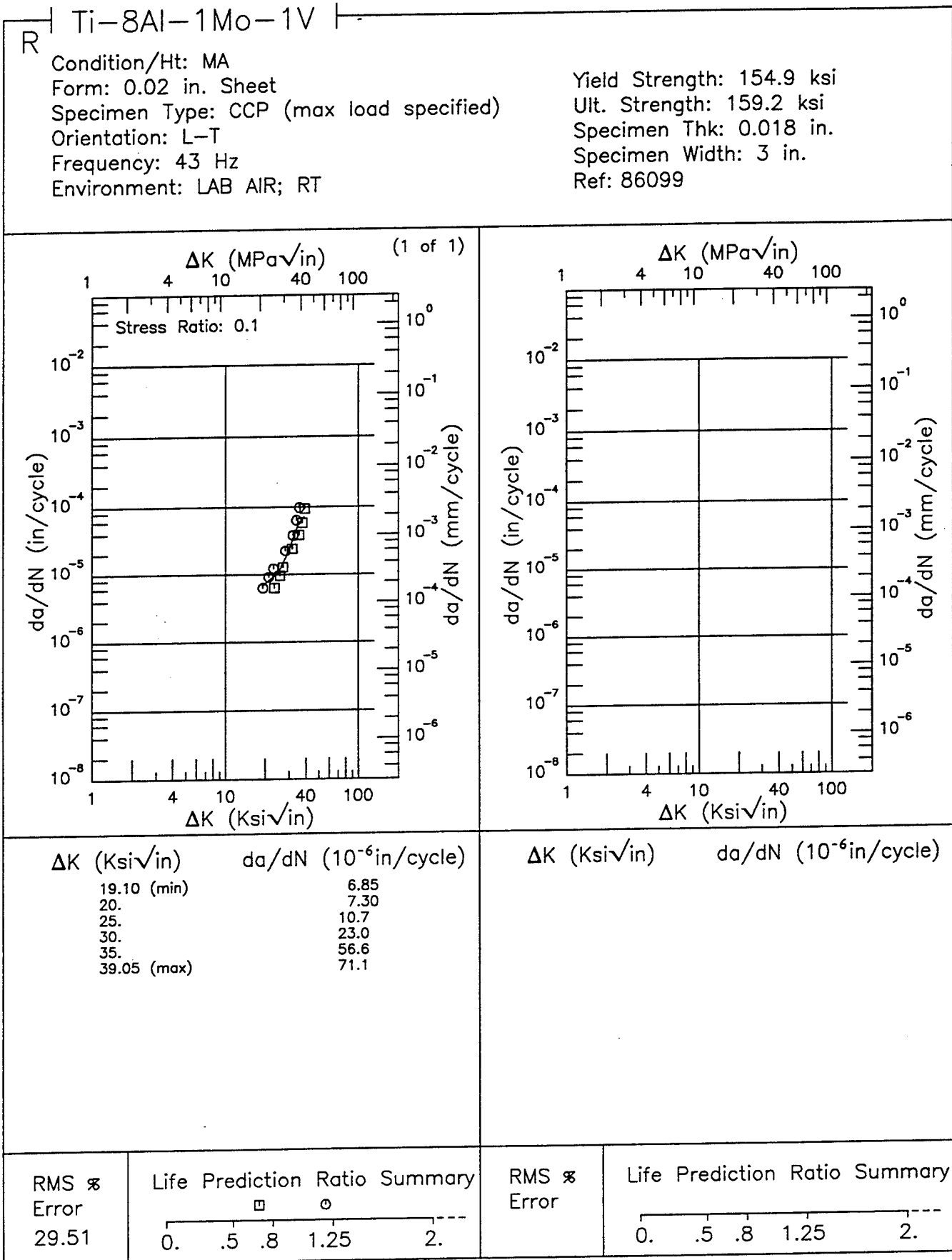


Figure 6.21.3.1.10

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Ti-8Al-1Mo-1V

Condition/Ht:
 Form:
 Specimen Type: SENT
 Orientation:
 Yield Strength:
 Ult. Strength:

Specimen Thk:
 Specimen Width:
 A₀:
 K_Isc:
 Ref: 82651

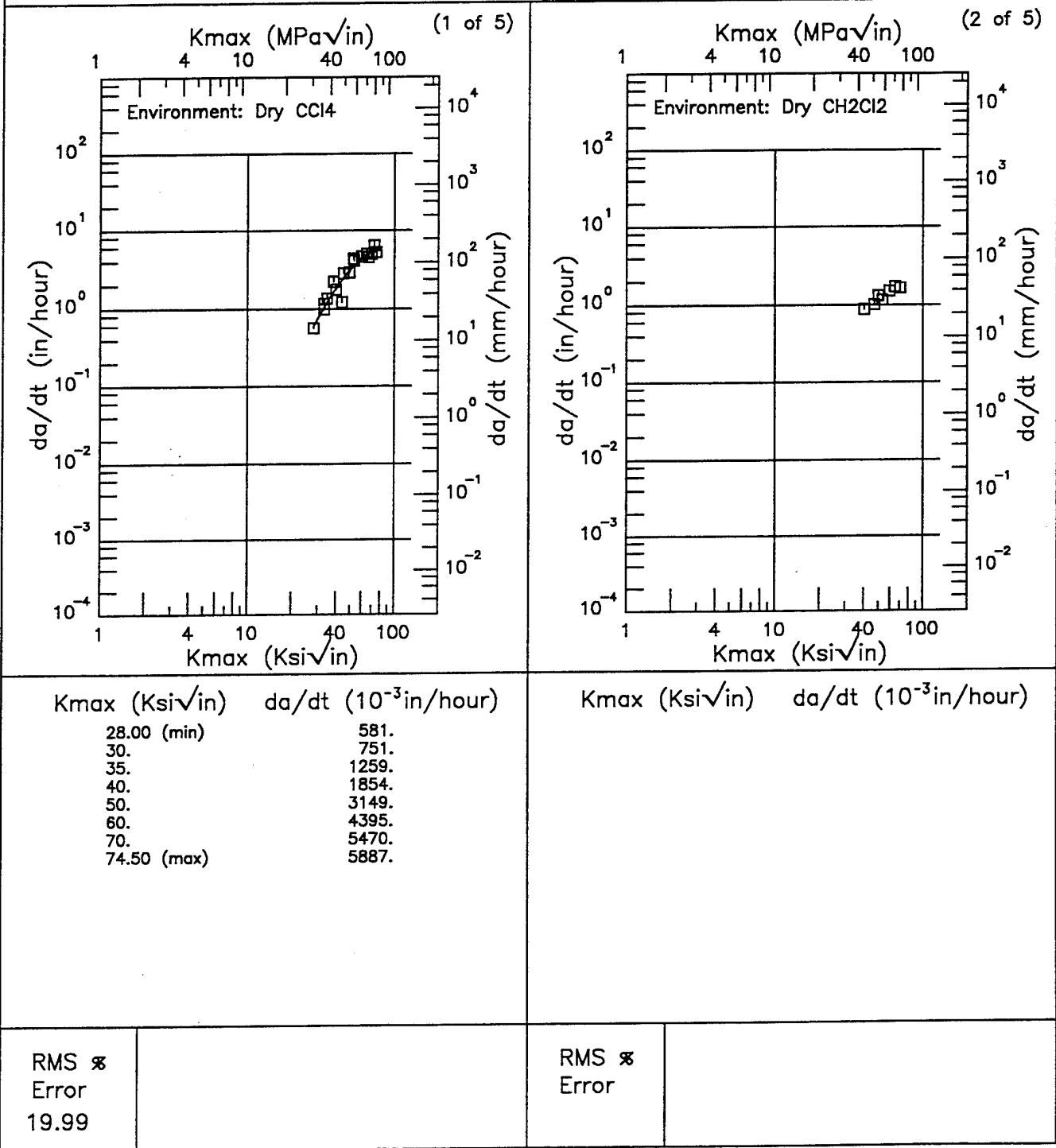
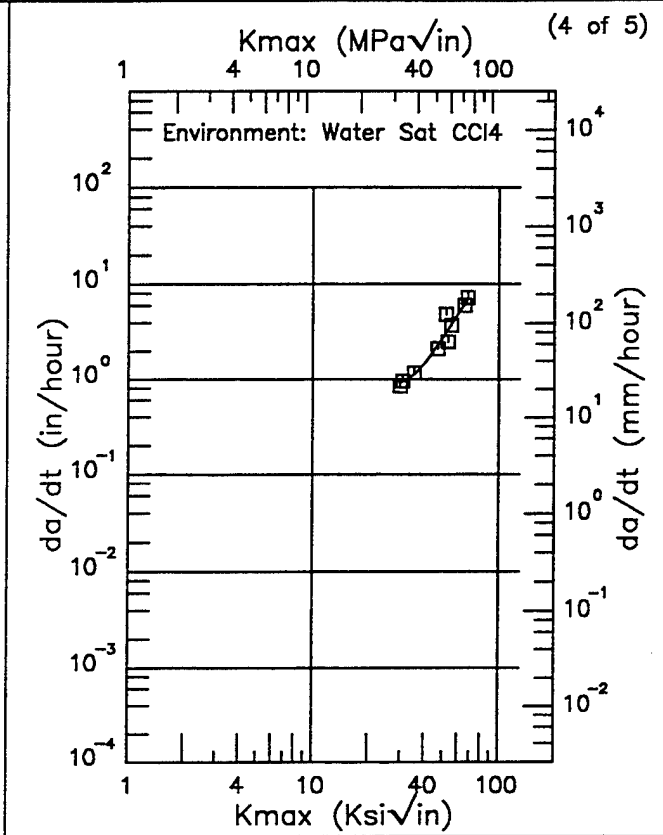
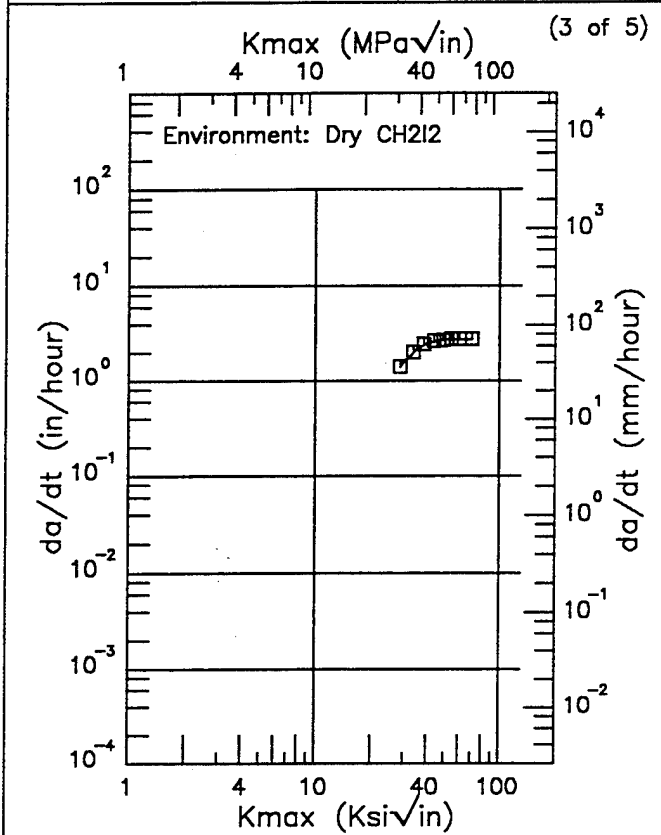


Figure 6.21.3.2.1

Ti-8Al-1Mo-1V

Condition/Ht:
 Form:
 Specimen Type: SENT
 Orientation:
 Yield Strength:
 Ult. Strength:

Specimen Thk:
 Specimen Width:
 A₀:
 K_{Isc}:
 Ref: 82651



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
28.50 (min)	1394.
30.	1601.
35.	2163.
40.	2507.
50.	2739.
60.	2740.
70.00 (max)	2753.

Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
29.50 (min)	926.
30.	933.
35.	1118.
40.	1482.
50.	2779.
60.	4880.
67.50 (max)	6868.

RMS %
 Error
 1.13

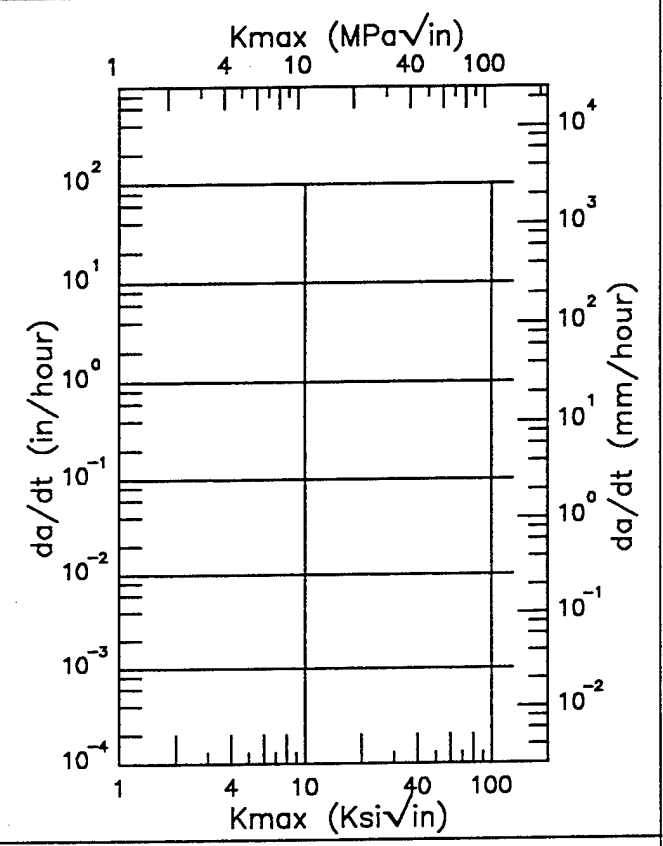
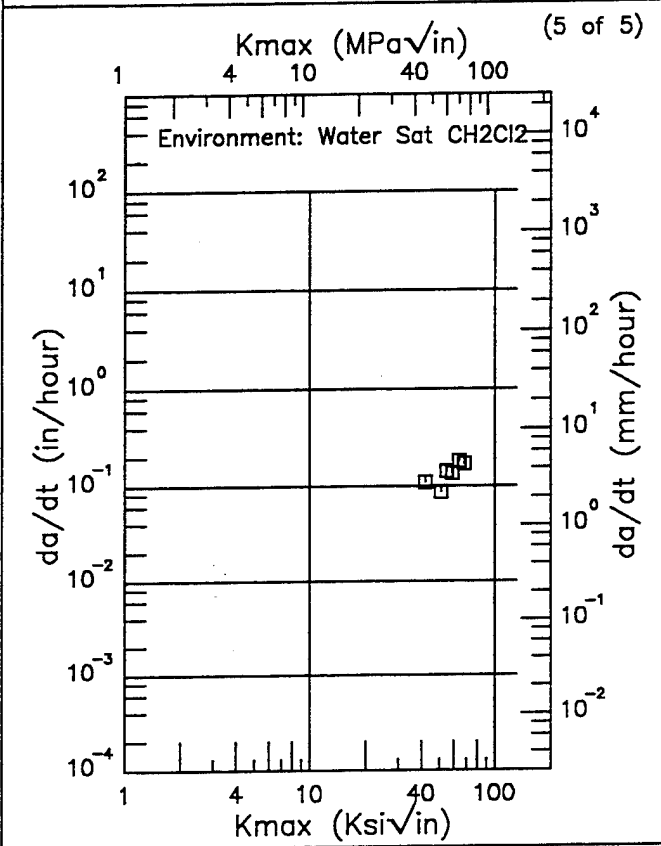
RMS %
 Error
 20.21

Figure 6.21.3.2.1 (Continued)

Ti-8Al-1Mo-1V

Condition/Ht:
 Form:
 Specimen Type: SENT
 Orientation:
 Yield Strength:
 Ult. Strength:

Specimen Thk:
 Specimen Width:
 A_o:
 K_{Isc}:
 Ref: 82651



K_{max} (Ksi√in) da/dt (10⁻³in/hour)

K_{max} (Ksi√in) da/dt (10⁻³in/hour)

RMS %
 Error

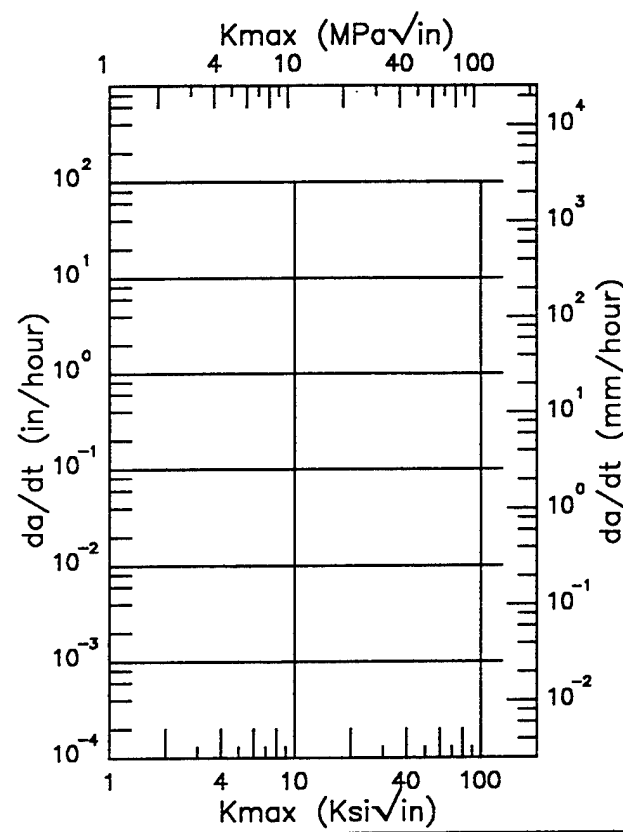
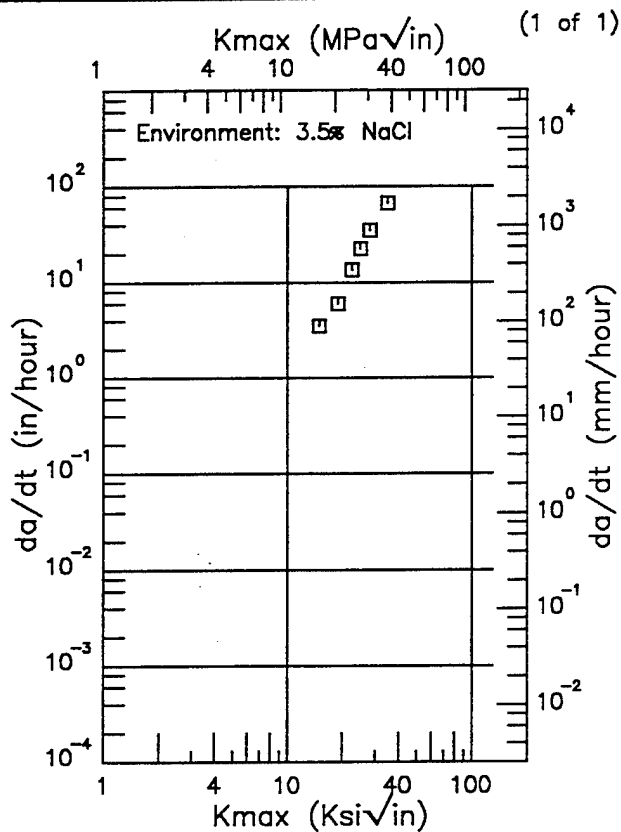
RMS %
 Error

Figure 6.21.3.2.1 (Concluded)

Ti-8Al-1Mo-1V

Condition/Ht:
 Form: 0.13 in. Sheet
 Specimen Type: SENT
 Orientation:
 Yield Strength: 150 ksi
 Ult. Strength:

Specimen Thk: 0.125 in.
 Specimen Width: 3 in.
 A₀:
 K_{Isc}: 22 ksi
 Ref: 77456



Kmax (Ksi√in) da/dt (10⁻³in/hour)

Kmax (Ksi√in) da/dt (10⁻³in/hour)

RMS %
Error

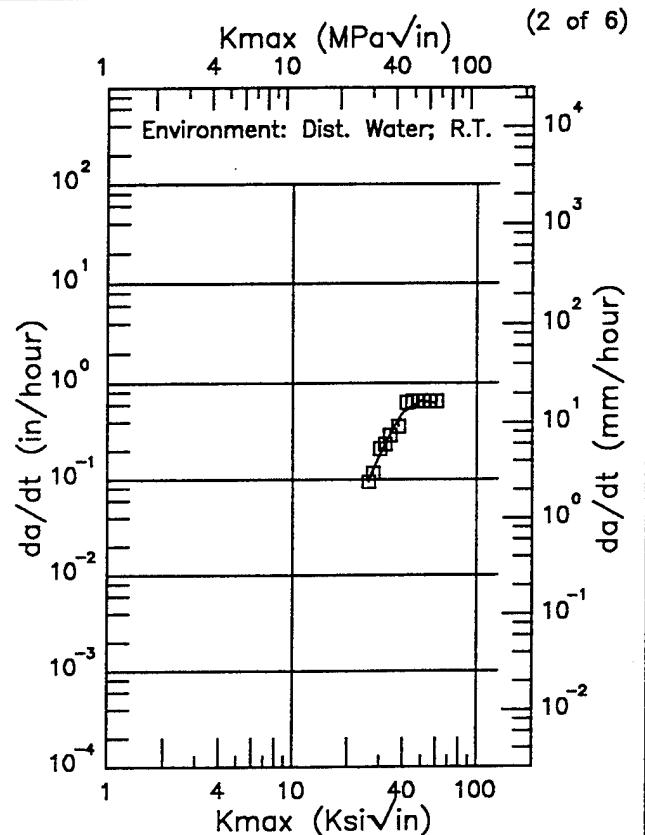
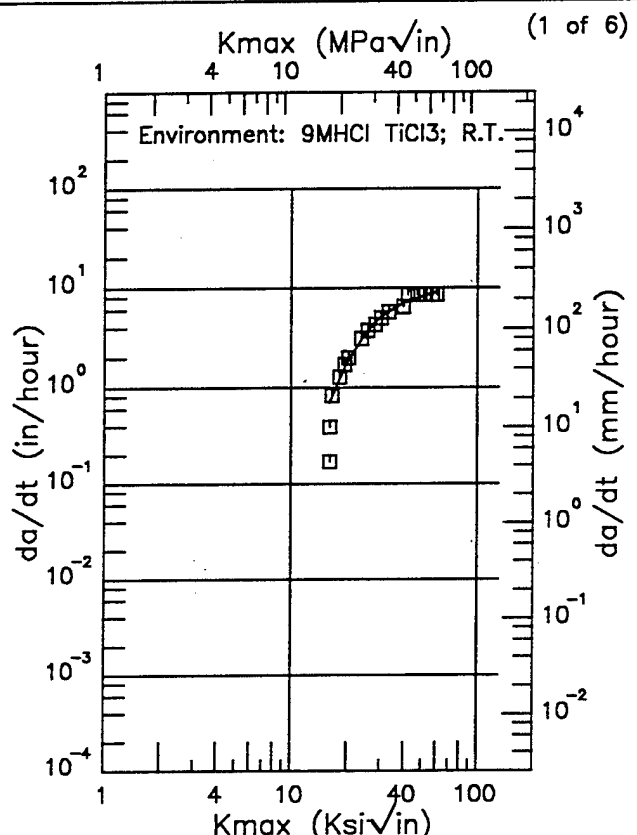
RMS %
Error

Figure 6.21.3.2.2

Ti-8Al-1Mo-1V

Condition/Ht: 1520F 1HR WQ
 Form: 0.25 in. Plate
 Specimen Type: DCB
 Orientation:
 Yield Strength: 124 ksi
 Ult. Strength:

Specimen Thk: 0.25 in.
 Specimen Width:
 A_o:
 K_Isc_c:
 Ref: 83689



K _{max} (Ksi√in)	da/dt (10 ⁻³ in/hour)
16.00 (min)	700.
20.	1848.
25.	3569.
30.	5098.
35.	6260.
40.	7104.
50.	8238.
60.00 (max)	9154.

K _{max} (Ksi√in)	da/dt (10 ⁻³ in/hour)
26.00 (min)	96.2
30.	179.
35.	330.
40.	498.
50.	659.
60.00 (max)	626.

RMS %
 Error
 22.24

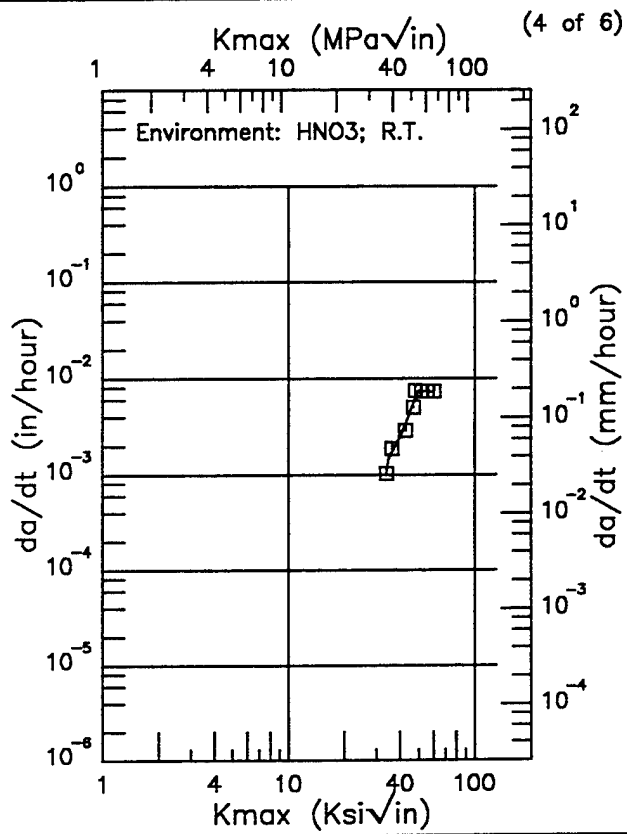
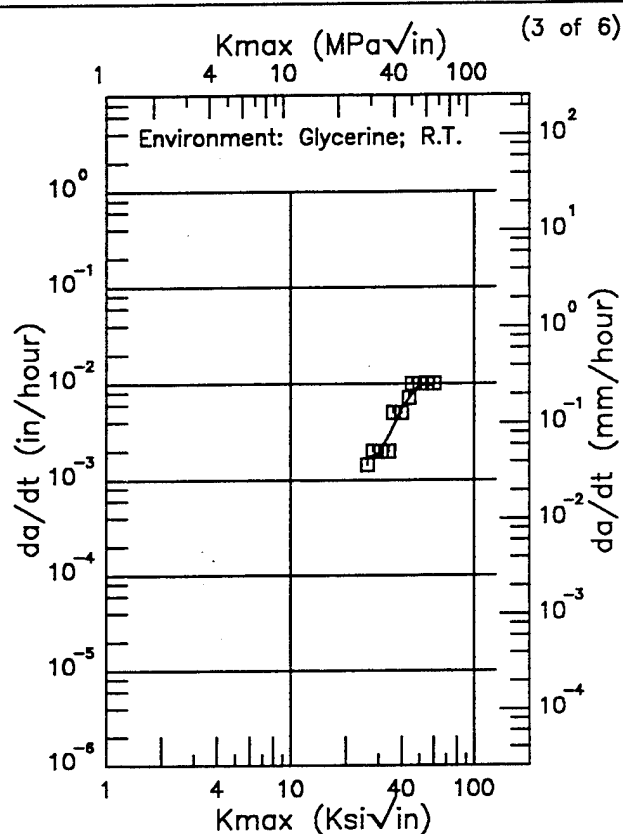
RMS %
 Error
 9.79

Figure 6.21.3.2.3

Ti-8Al-1Mo-1V

Condition/Ht: 1520F 1HR WQ
 Form: 0.25 in. Plate
 Specimen Type: DCB
 Orientation:
 Yield Strength: 124 ksi
 Ult. Strength:

Specimen Thk: 0.25 in.
 Specimen Width:
 A₀:
 K_Isc:cc:
 Ref: 83689



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
26.00 (min)	1.70
30.	1.87
35.	3.14
40.	5.46
50.	9.93
60.00 (max)	9.53

Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
33.80 (min)	1.02
35.	1.51
40.	2.45
50.	7.50
60.00 (max)	7.29

RMS %
 Error
 18.77

RMS %
 Error
 10.49

Figure 6.21.3.2.3 (Continued)

Ti-8Al-1Mo-1V

Condition/Ht: 1520F 1HR WQ
 Form: 0.25 in. Plate
 Specimen Type: DCB
 Orientation:
 Yield Strength: 124 ksi
 Ult. Strength:

Specimen Thk: 0.25 in.
 Specimen Width:
 A₀:
 K_Isc:
 Ref: 83689

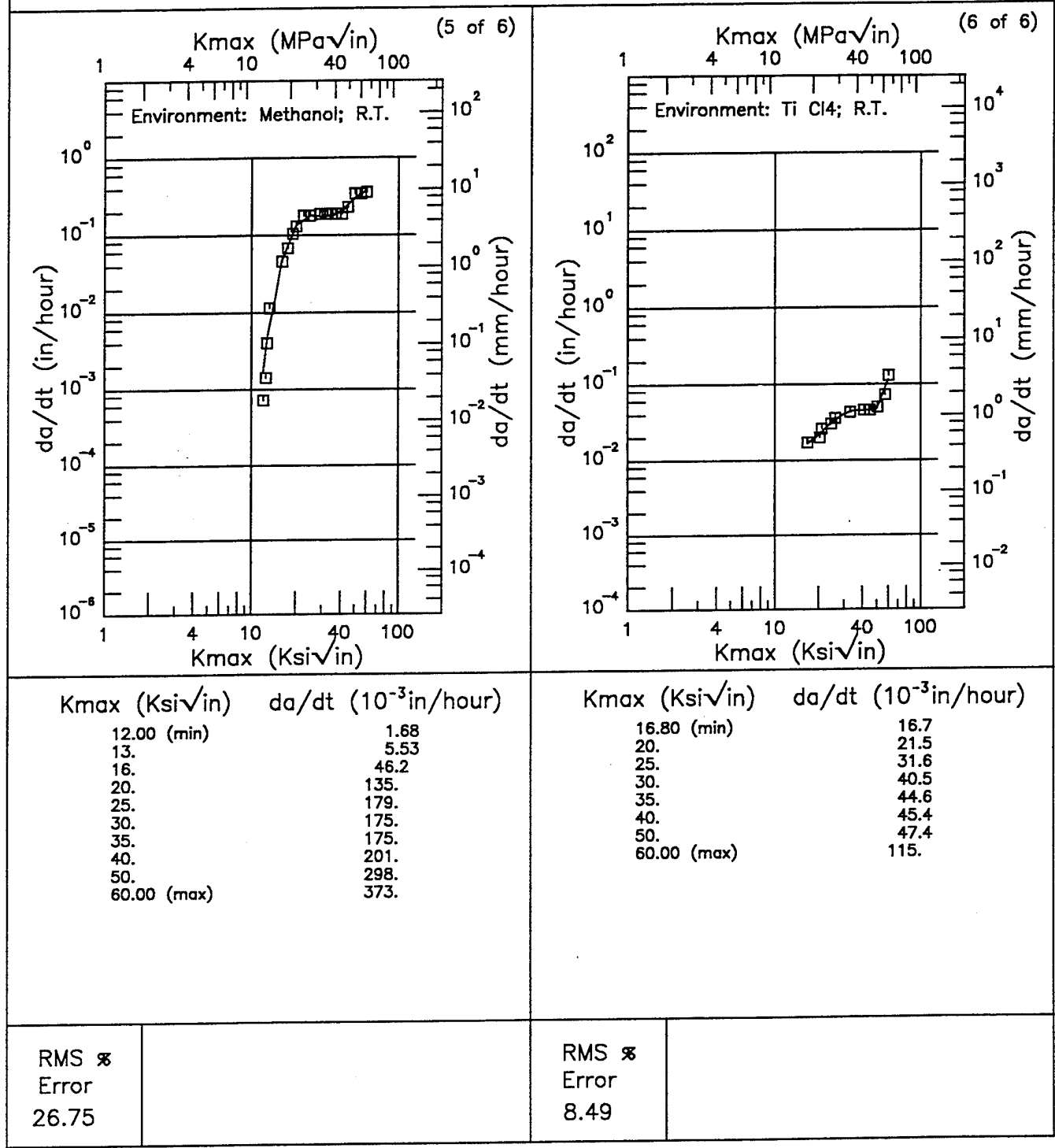
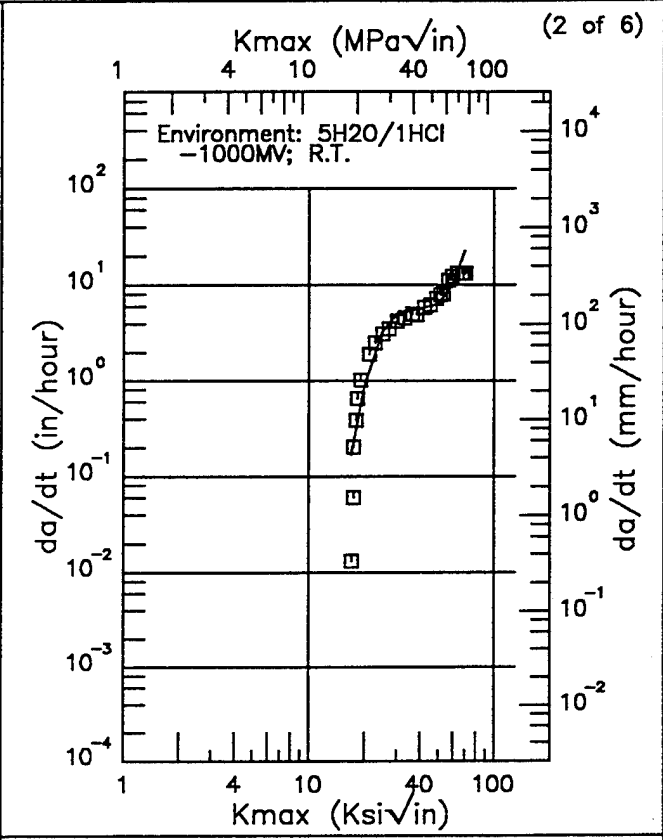
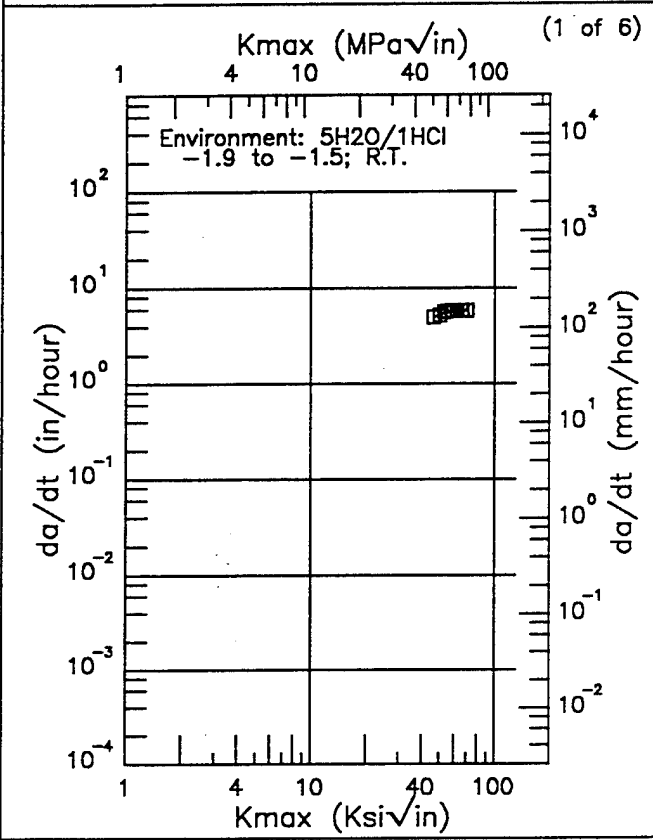


Figure 6.21.3.2.3 (Concluded)

Ti-8Al-1Mo-1V

Condition/Ht: 1520F 1HR WQ
 Form: 0.25 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Yield Strength: 124 ksi
 Ult. Strength:

Specimen Thk: 0.25 in.
 Specimen Width: 1 in.
 A₀: 0.985 in.
 K_{Isc}:
 Ref: 83689



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
46.50 (min)	4980.
50.	5251.
60.	5848.
70.00 (max)	5819.

Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
17.00 (min)	168.
20.	903.
25.	3256.
30.	4993.
35.	5561.
40.	5665.
50.	6453.
60.	10102.
70.00 (max)	22488.

RMS % Error	1.56
-------------	------

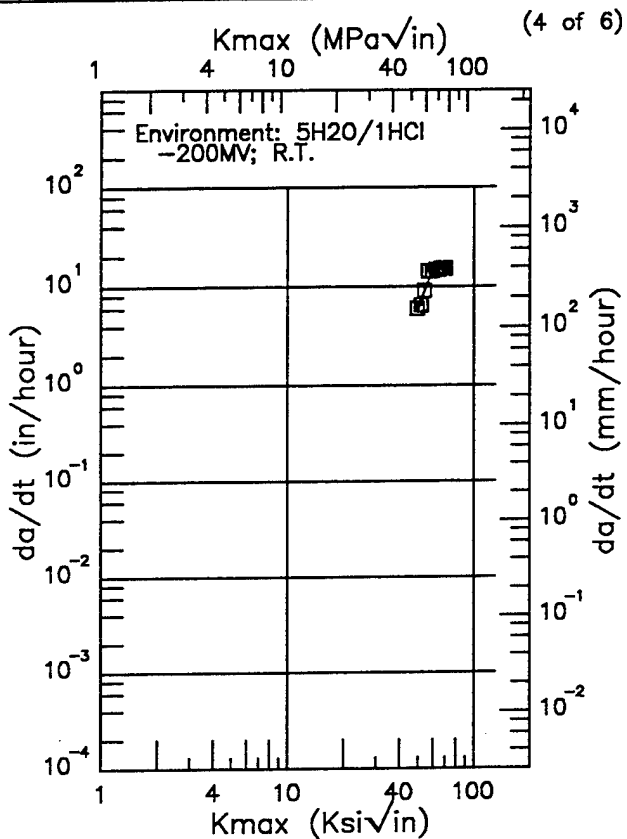
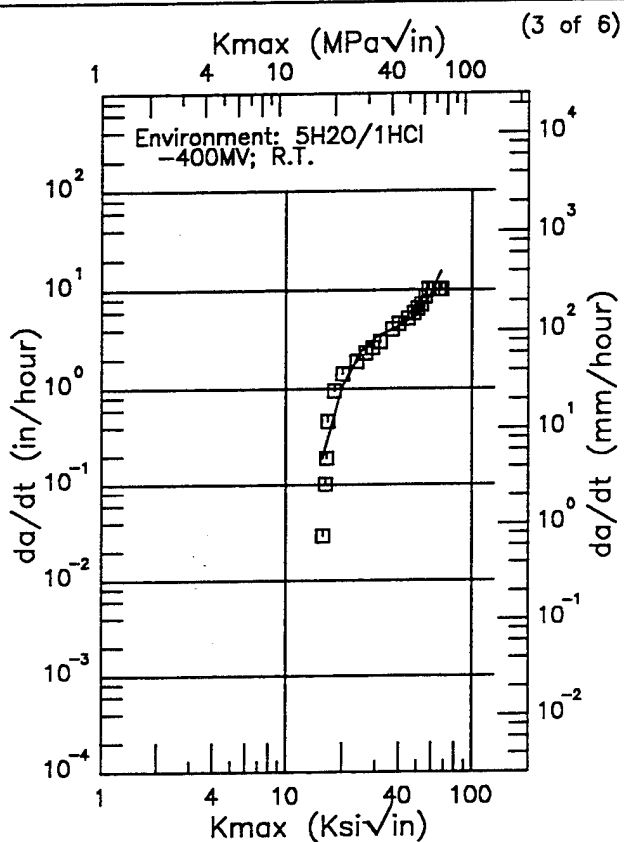
RMS % Error	40.62
-------------	-------

Figure 6.21.3.2.4

Ti-8Al-1Mo-1V

Condition/Ht: 1520F 1HR WQ
 Form: 0.25 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Yield Strength: 124 ksi
 Ult. Strength:

Specimen Thk: 0.25 in.
 Specimen Width: 1 in.
 A₀: 0.985 in.
 K_{Isc}:
 Ref: 83689



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
15.70 (min)	190.
16.	225.
20.	1047.
25.	2394.
30.	3328.
35.	3861.
40.	4278.
50.	5633.
60.	9067.
68.00 (max)	15513.

Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
49.50 (min)	5721.
50.	5869.
60.	15346.
70.00 (max)	16018.

RMS %
 Error
 35.81

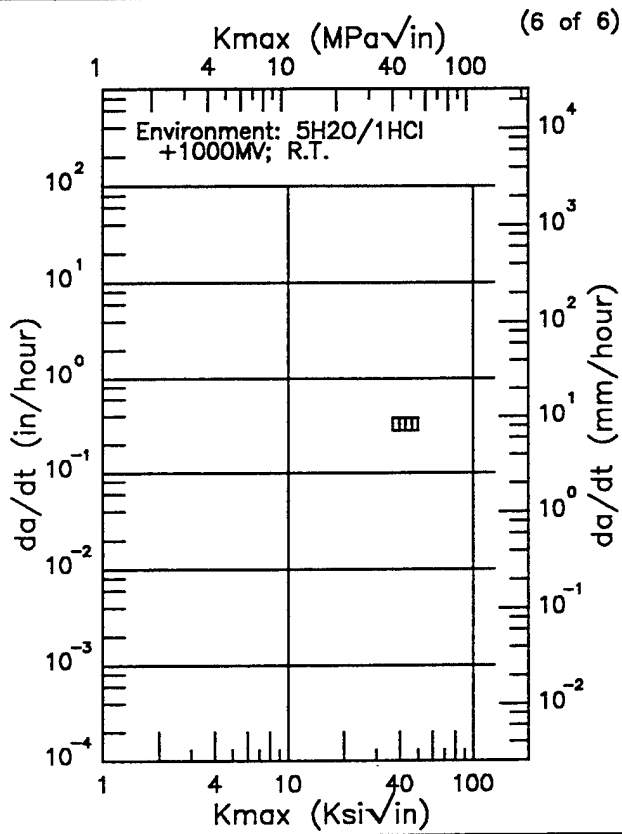
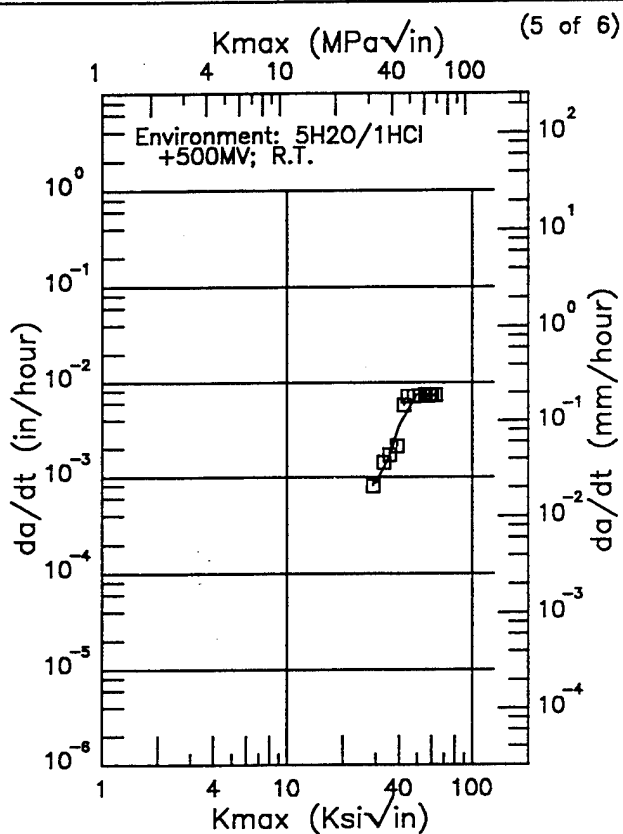
RMS %
 Error
 8.89

Figure 6.21.3.2.4 (Continued)

Ti-8Al-1Mo-1V

Condition/Ht: 1520F 1HR WQ
 Form: 0.25 in. Plate
 Specimen Type: DCB
 Orientation: T-L
 Yield Strength: 124 ksi
 Ult. Strength:

Specimen Thk: 0.25 in.
 Specimen Width: 1 in.
 A₀: 0.985 in.
 K_{Isc}:
 Ref: 83689



K _{max} (Ksi√in)	da/dt (10 ⁻³ in/hour)
29.00 (min)	0.850
30.	0.871
35.	1.57
40.	3.46
50.	7.48
60.	7.20
63.00 (max)	7.30

K_{max} (Ksi√in) da/dt (10⁻³in/hour)

RMS %
 Error
 17.47

RMS %
 Error

Figure 6.21.3.2.4 (Concluded)

Ti-8Al-1Mo-1V

Condition/Ht: 1725F FC 1200F 3HR WQ
 Form: 0.25 in. Plate
 Specimen Type: CB
 Orientation: T-L
 Yield Strength:
 Ult. Strength:

Specimen Thk: 0.25 in.
 Specimen Width: 2 in.
 A_0 : 0.25 in.
 K_{Jsc} :
 Ref: 85855

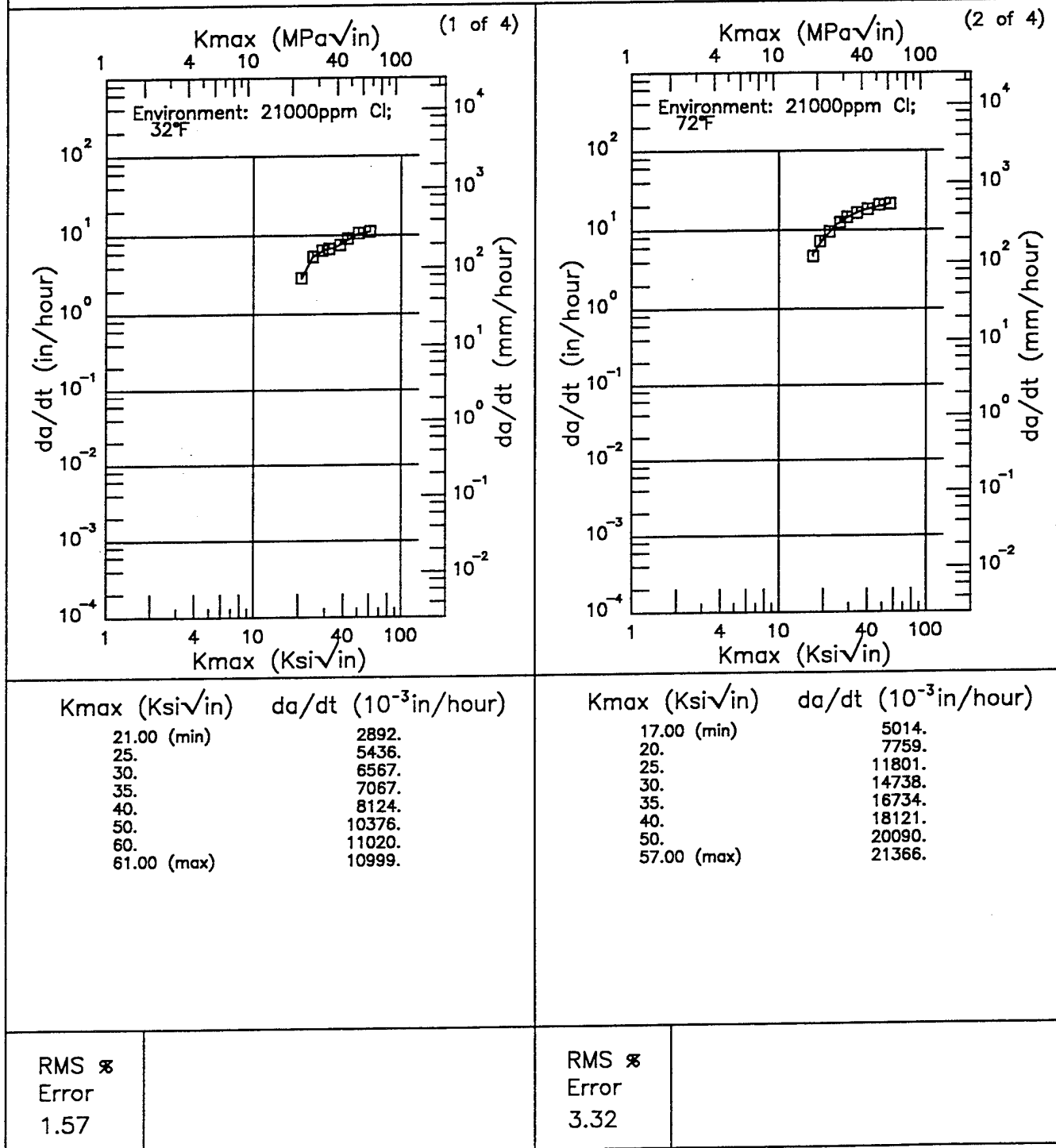
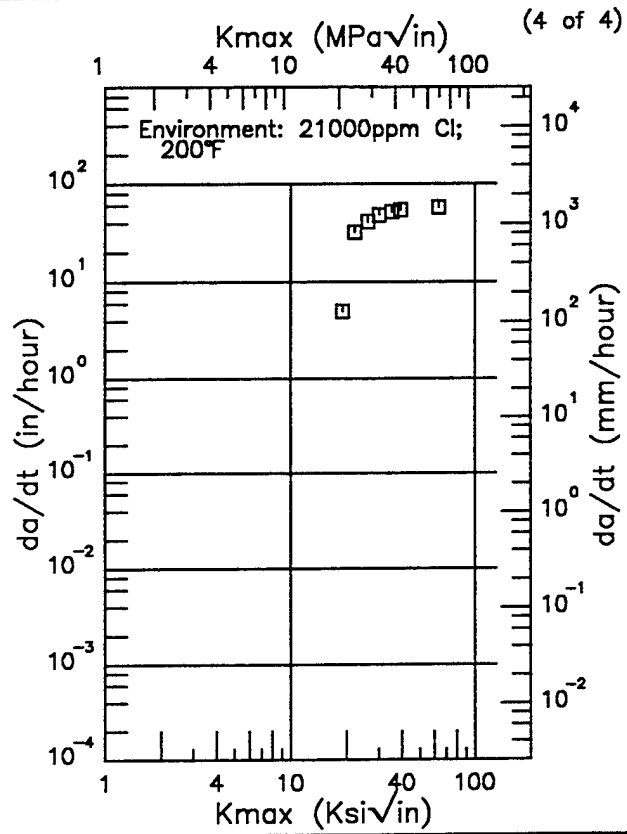
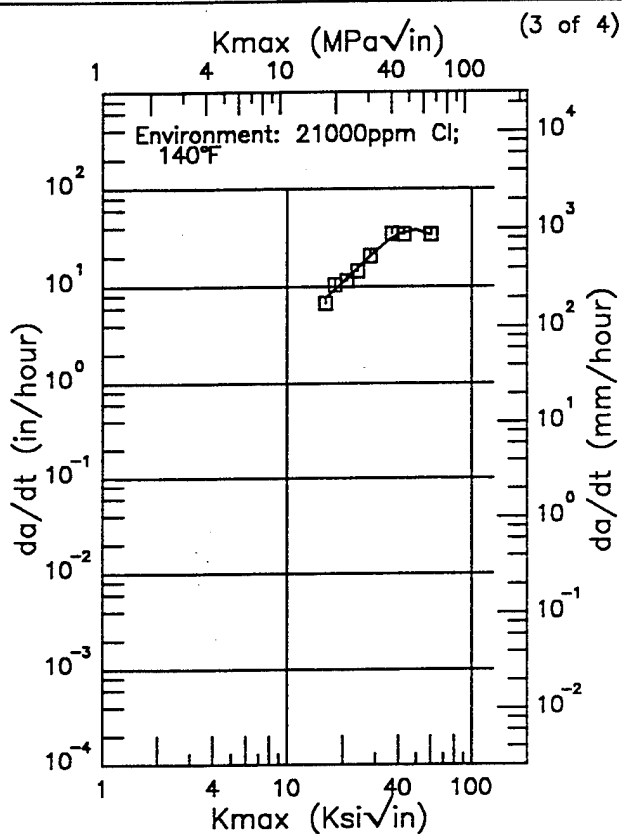


Figure 6.21.3.2.5

Ti-8Al-1Mo-1V

Condition/Ht: 1725F FC 1200F 3HR WQ
 Form: 0.25 in. Plate
 Specimen Type: CB
 Orientation: T-L
 Yield Strength:
 Ult. Strength:

Specimen Thk: 0.25 in.
 Specimen Width: 2 in.
 A₀: 0.25 in.
 K_{Isc}:
 Ref: 85855



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
16.00 (min)	7709.
20.	10915.
25.	16530.
30.	23037.
35.	29281.
40.	34200.
50.	37827.
60.00 (max)	33748.

Kmax (Ksi√in) da/dt (10⁻³in/hour)

RMS %
 Error
 7.58

RMS %
 Error

Figure 6.21.3.2.5 (Concluded)

Ti-8Al-1Mo-1V

Condition/Ht: 1725F FC 1200F 3HR WQ
 Form: 0.25 in. Plate
 Specimen Type: CB
 Orientation: T-L
 Yield Strength: 150 ksi
 Ult. Strength:

Specimen Thk: 0.25 in.
 Specimen Width: 2 in.
 A₀: 0.25 in.
 K_{Isc}:
 Ref: 85855

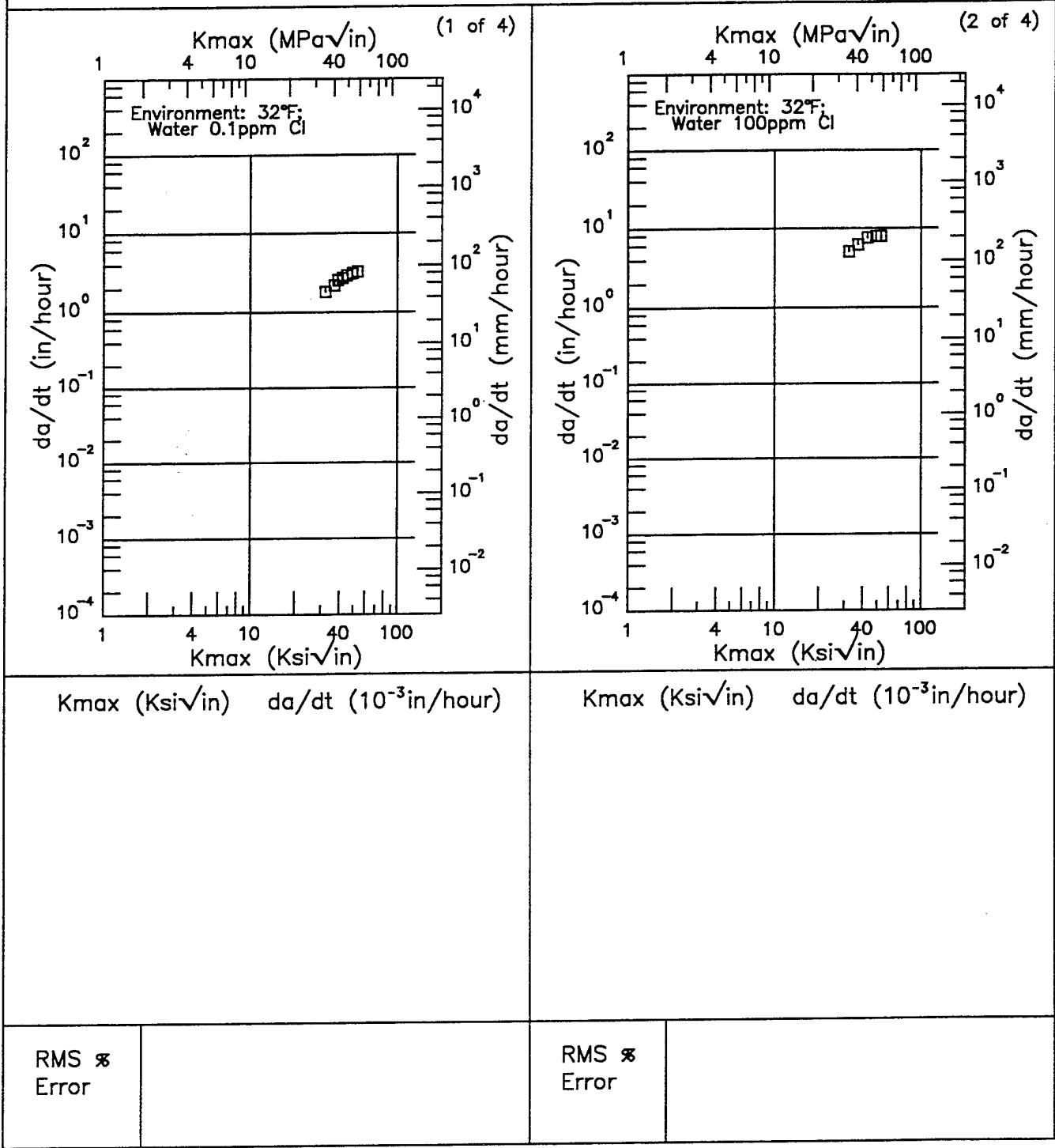


Figure 6.21.3.2.6

Ti-8Al-1Mo-1V

Condition/Ht: 1725F FC 1200F 3HR WQ
 Form: 0.25 in. Plate
 Specimen Type: CB
 Orientation: T-L
 Yield Strength: 150 ksi
 Ult. Strength:

Specimen Thk: 0.25 in.
 Specimen Width: 2 in.
 A₀: 0.25 in.
 K_Isc:
 Ref: 85855

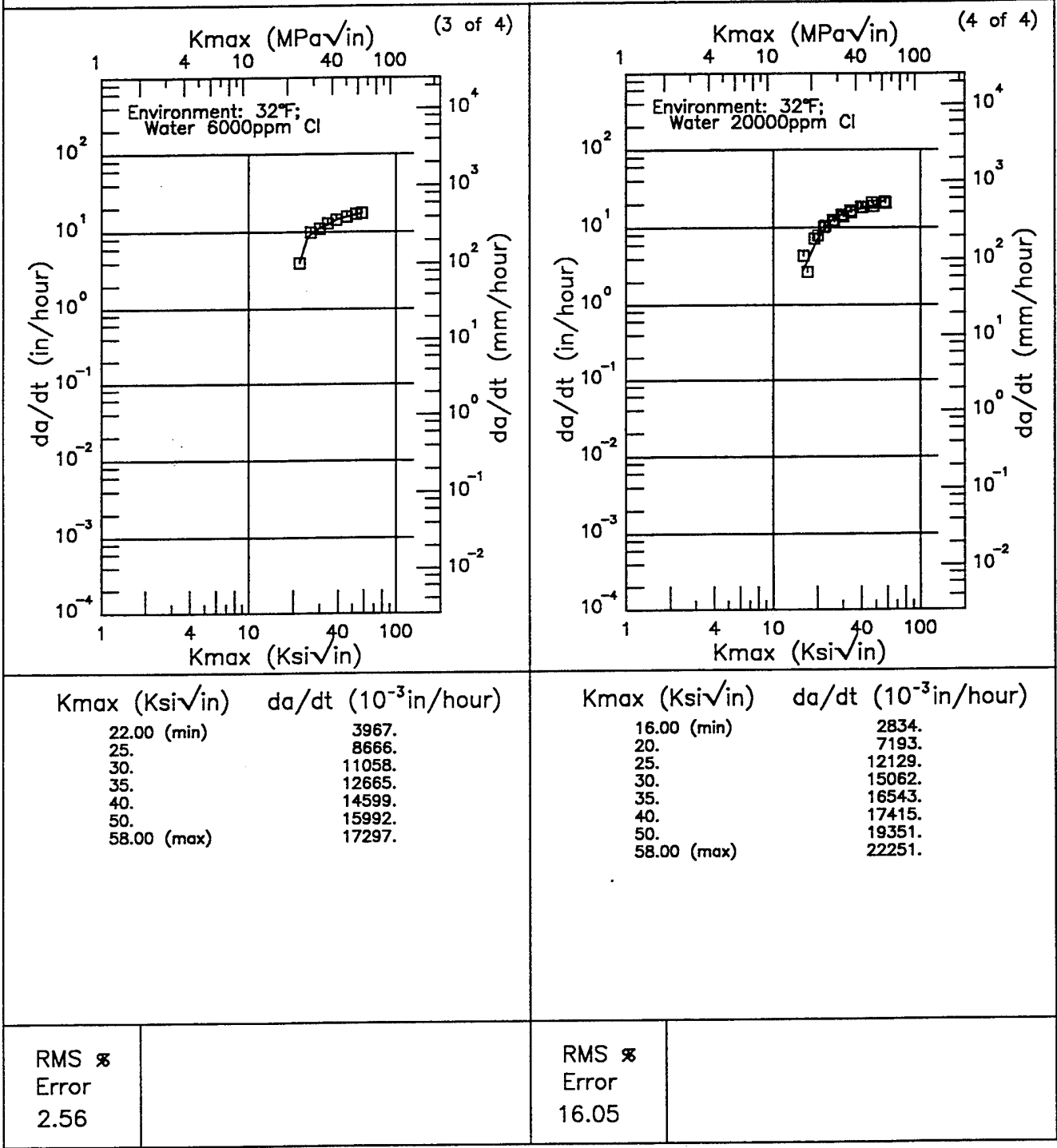


Figure 6.21.3.2.6 (Concluded)

Ti-8Al-1Mo-1V

Condition/Ht: MA
 Form: 0.16 in. Sheet
 Specimen Type: CCP (max load specified)
 Orientation:
 Yield Strength: 145 ksi
 Ult. Strength:

Specimen Thk: 0.16 in.
 Specimen Width: 8 in.
 A₀:
 K_{Isc}: 21 ksi
 Ref: 84290

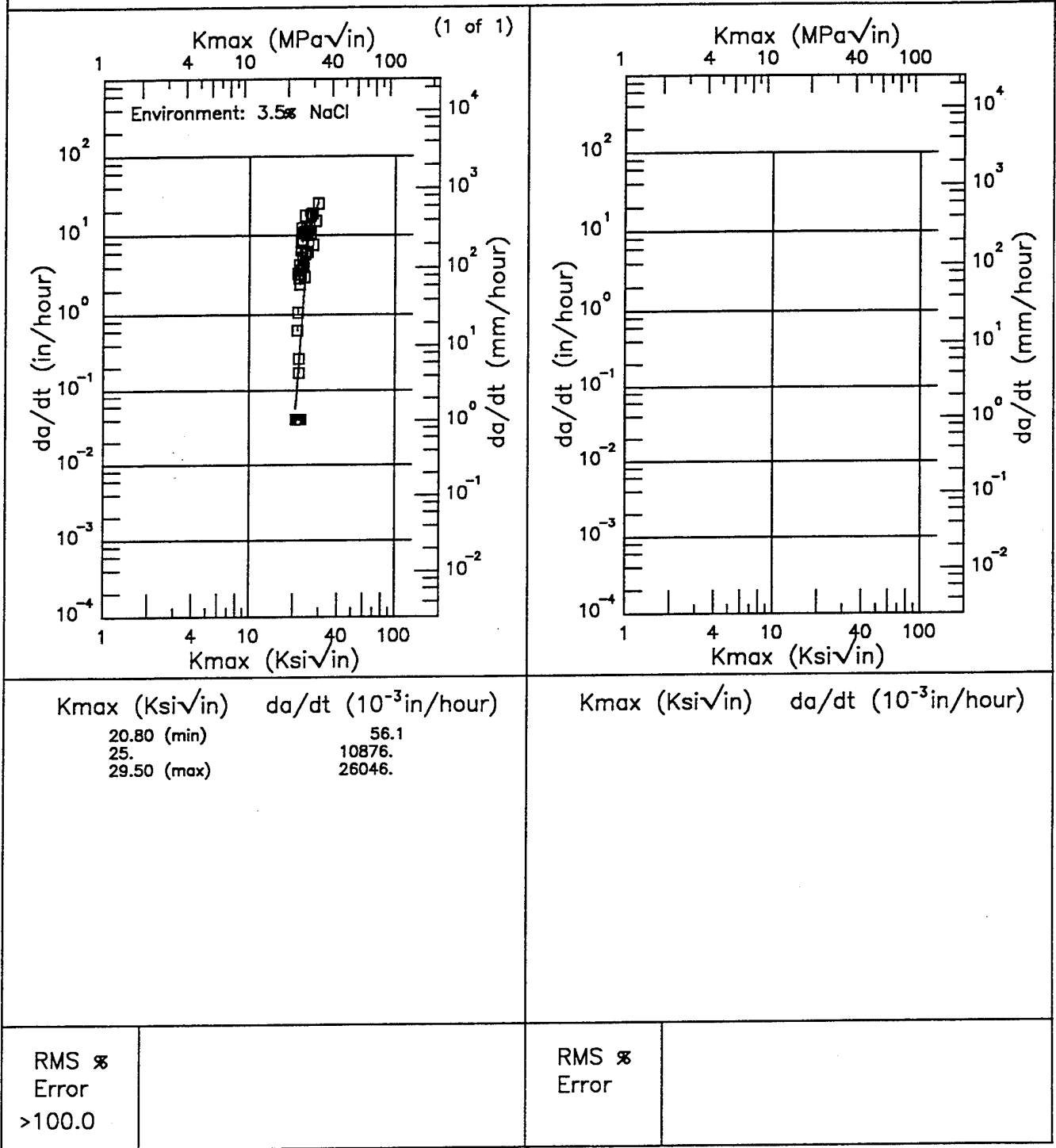
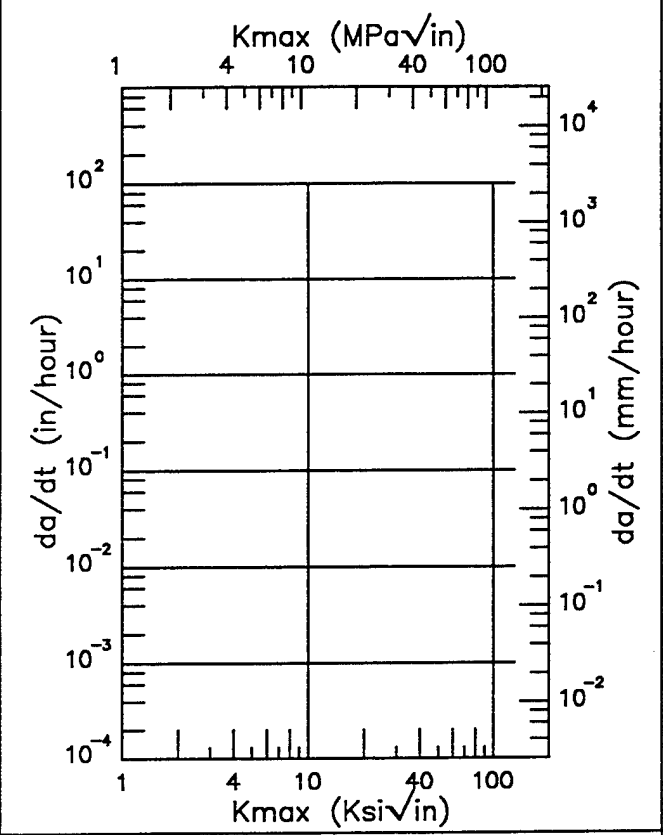
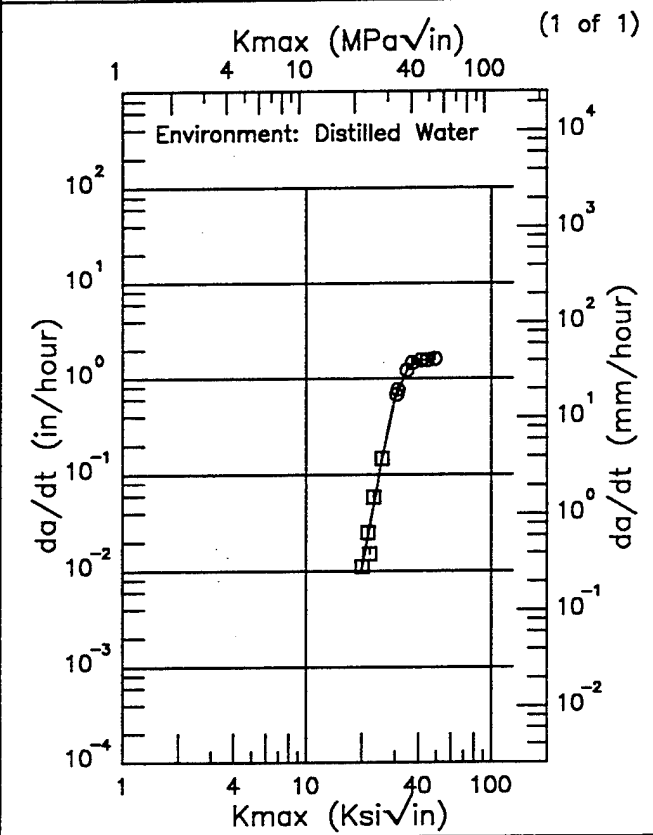


Figure 6.21.3.2.7
 6-558

Ti-8Al-1Mo-1V

Condition/Ht: MA
 Form: 0.27 in. Plate
 Specimen Type: SENT
 Orientation:
 Yield Strength:
 Ult. Strength:

Specimen Thk: 0.26 in.
 Specimen Width: 2 in.
 A₀:
 K_{Isc}:
 Ref: 81741



Kmax (Ksi√in)	da/dt (10 ⁻³ in/hour)
20.00 (min)	10.3
25.	118.
30.	614.
35.	1261.
40.	1543.
49.00 (max)	1600.

Kmax (Ksi√in) da/dt (10⁻³in/hour)

RMS %
 Error
 16.05

RMS %
 Error

Figure 6.21.3.2.8

Ti-8Al-1Mo-1V

(1 of 7)

TABLE 6.21.3.3

K_{Isec} SUMMARY FOR TITANIUM ALLOY Ti-8Al-1Mo-1V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen		Prod Thk (in)	Crack (in)	K _q (Ksi/in)	K _{Isec} (Ksi/in)	Test Time (min)	Test Date	Refer	
						Design	Width (in)								Thick (in)
Unspecified	S	R.T.	T-S	130	3.5% NaCl	CANT	1	0.125	---	54	18	---	1967	70931	
	P	R.T.	---	---	3.5% NaCl	CANT	---	---	---	100	36*	---	1967	70887	
1520° F 1hr WQ	P	-51	T-L	124	5 to 1 Solvent to Solute Ratio	DCB	1	0.25	0.99	95	20	---	1962	83689	
		-15	T-L	124	Water LiCl	DCB	1	0.25	0.99	95	22.7	---	1962	83689	
		12	T-L	124	CCl ₄	DCB	1	0.25	0.99	95	19.4	---	1962	83689	
		27	T-L	124	Water LiCl	DCB	1	0.25	0.99	95	20.3	---	1962	83689	
		36	T-L	124	Water LiCl	DCB	1	0.25	0.99	95	17	---	1962	83689	
						Water LiCl 0.1M	DCB	1	0.25	0.99	95	23.1	---	1962	83689
						5 to 1 Solvent to Solute Ratio Water LiCl -1000MV	DCB	1	0.25	0.99	95	23.2	---	1962	83689
						Air	DCB	1	0.25	0.99	95	36	---	1962	83689
						CCl ₄	DCB	1	0.25	0.99	95	20	---	1962	83689
				R.T.	T-L	124	Chloroform	DCB	1	0.25	0.99	95	21	---	1962
					DMSO LiCl	DCB	1	0.25	0.99	95	19	---	1962	83689	
					Glycerine	DCB	1	0.25	0.99	95	26	---	1962	83689	
					Glycerine LiCl	DCB	1	0.25	0.99	95	19	---	1962	83689	
					Hexamethyl-Phos- phoric Triamide	DCB	1	0.25	0.99	95	27.8	---	1962	83689	

TABLE 6.21.3.3 (CONTINUED)

(2 of 7)

K_{sec} SUMMARY FOR TITANIUM ALLOY Ti-8Al-1Mo-1V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K _q (Ksi√in)	K _{sec} (Ksi√in)	Test Time (min)	Test Date	Refer
						Design	Width (in)	Thick (in)							
					Methanol	DCB	1	0.25	0.25	0.99	95	13.8	---	1962	83689
					Methanol LiCl	DCB	1	0.25	0.25	0.99	95	13	---	1962	83689
					Mixed Solvents 5 to 1 Solvent to Solute Ratio Formic Acid	DCB	1	0.25	0.25	0.99	95	25	---	1962	83689
					N n-Dimethyl- Acetamide	DCB	1	0.25	0.25	0.99	95	30.6	---	1962	83689
					Water	DCB	1	0.25	0.25	0.99	95	26	---	1962	83689
					Water 0.1M LiCl	DCB	1	0.25	0.25	0.99	95	24.8	---	1962	83689
WQ	P	R.T.	T-L	124	Water 9M LiCl 9M LiNO ₃	DCB	1	0.25	0.25	0.99	95	23.2	---	1962	83689
					Water HCl 1M TiCl ₃	DCB	1	0.25	0.25	0.99	95	16.3	---	1962	83689
					Water LiBr	DCB	1	0.25	0.25	0.99	95	19.5	---	1962	83689
					Water LiCl	DCB	1	0.25	0.25	0.99	95	14.7	---	1962	83689
					Water LiCl 0.6M NaCl -500MV	DCB	1	0.25	0.25	0.99	95	20	---	1962	83689
					Water LiCl -1000MV	DCB	1	0.25	0.25	0.99	95	17	---	1962	83689
					Water LiCl -200MV	DCB	1	0.25	0.25	0.99	95	15.5	---	1962	83689

Ti-8Al-1Mo-1V

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TABLE 6.21.3.3 (CONTINUED)
K_{Iscc} SUMMARY FOR TITANIUM ALLOY Ti-8Al-1Mo-1V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Crack (in)	K _Q (Ksi√in)	K _{Iscc} (Ksi√in)	Test Time (min)	Test Date	Refer	
						Design	Width (in)	Thick (in)							
WQ (cont'd)	P (cont'd)	R.T. (cont'd)	T-L (cont'd)	124 (cont'd)	Water LiCl - 200MV	DCB	1	0.25	0.25	0.99	95	19.9	---	1962	83689
					Water LiCl +500MV	DCB	1	0.25	0.25	0.99	95	27.7	---	1962	83689
					Water LiCl +2000MV	DCB	1	0.25	0.25	0.99	95	20.8	---	1962	83689
					Water LiCl +2000MV	DCB	1	0.25	0.25	0.99	95	20	---	1962	83689
					Water(1) Glycerine(4) LiCl	DCB	1	0.25	0.25	0.99	95	24.5	---	1962	83689
					Water(2,5) Glycerine (2,5) LiCl	DCB	1	0.25	0.25	0.99	95	18	---	1962	83689
					Water(3,4) DMSO(1.6) LiCl	DCB	1	0.25	0.25	0.99	95	19.5	---	1962	83689
					Water(4) Glycerine(1) LiCl	DCB	1	0.25	0.25	0.99	95	19	---	1962	83689
					Water 3.5% NaCl	DCB	1	0.25	0.25	0.99	95	21.4	---	1962	83689
					CCl ₄	DCB	1	0.25	0.25	0.99	95	19.5	---	1962	83689
					DMSO LiCl	DCB	1	0.25	0.25	0.99	95	25	---	1962	83689
					Methanol LiCl	DCB	1	0.25	0.25	0.99	95	19	---	1962	83689
					Water LiCl	DCB	1	0.25	0.25	0.99	95	21.8	---	1962	83689
					Acetone LiBr	DCB	1	0.25	0.25	0.99	95	30	---	1962	83689

TABLE 6.21.3.3 (CONTINUED)

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K_{Isc} SUMMARY FOR TITANIUM ALLOY Ti-8Al-1Mo-1V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K _Q (Ksi√in)	K _{Isc} (Ksi√in)	Test Time (min)	Test Date	Refer
						Design	Width (in)	Thick (in)							
		151 (cont'd)	T-L (cont'd)	124 (cont'd)	Water Ammonium Chloride	DCB	1	0.25	0.25	0.99	95	26	---	1962	83689
		194	T-L	124	Water 0.1M LiCl	DCB	1	0.25	0.25	0.99	95	30.2	---	1962	83689
					Water LiCl -1000MV	DCB	1	0.25	0.25	0.99	95	29	---	1962	83689
		203	T-L	124	Water LiCl +1000MV	DCB	1	0.25	0.25	0.99	95	21.3	---	1962	83689
					Water LiCl +2000MV	DCB	1	0.25	0.25	0.99	95	23.9	---	1962	83689
					Water LiCl 0MV	DCB	1	0.25	0.25	0.99	95	22.2	---	1962	83689
					Glycerine LiCl	DCB	1	0.25	0.25	0.99	95	26	---	1962	83689
		205	T-L	124	Water	DCB	1	0.25	0.25	0.99	95	42*	---	1962	83689
					Water LiCl	DCB	1	0.25	0.25	0.99	95	20.8	---	1962	83689
		212	T-L	124	Silicone Oil	DCB	1	0.25	0.25	0.99	95	65*	---	1962	83689
		235	T-L	124	Water LiCl	DCB	1	0.25	0.25	0.99	95	22.7	---	1962	83689
		412	T-L	124	Glycerine LiCl	DCB	1	0.25	0.25	0.99	95	26	---	1962	83689
1675°F 1hr AC 1075°F 8hr AC 1000°F 2hr AC	P	R.T.	T-L	138.9	3.5% NaCl	CANT	---	0.866	1	---	---	26.4	---	1981	NR001

Ti-8Al-1Mo-1V

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TABLE 6.21.3.3 (CONTINUED)
 K_{Isc} SUMMARY FOR TITANIUM ALLOY Ti-8Al-1Mo-1V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K_{Isc} (Ksi/in)	Test Time (min)	Test Date	Refer
						Design	Width (in)	Thick (in)						
1700° F 1hr AC 1200° F 2hr WQ	P	R.T.	T-S	107.9	3.5% NaCl	CANT	1	0.5	1	---	28	---	1967	70931
1775° F 0.5hr FC to 1200° F 1200° F 0.5hr AC 1200° F 3hr Argon Quench	P	32	T-L	150	Water 21000PPM Chloride	CANT	2	0.25	0.25	0.2	65.4	---	1973	85855
		R.T.	T-L	150	Water 0.1PPM Chloride	CANT	2	0.25	0.25	0.2	59.5	---	1973	85855
		R.T.	T-L	150	Water 100PPM Chloride	CANT	2	0.25	0.25	0.2	58.7	---	1973	85855
		R.T.	T-L	150	Water 21000PPM Chloride	CANT	2	0.25	0.25	0.2	67.4	---	1973	85855
		140	T-L	150	Water 6000PPM Chloride	CANT	2	0.25	0.25	0.2	60.3	---	1973	85855
		200	T-L	150	Water 21000PPM Chloride	CANT	2	0.25	0.25	0.2	62.1	---	1973	85855
1825° F 1hr AC	P	R.T.	T-S	120.4	3.5% NaCl	CANT	1	0.5	1	---	23	---	1967	70931
2000° F 0.5hr AC	P	R.T.	T-L	115.1	3.5% NaCl	CANT	---	0.866	1	---	47.3	---	1981	NR001

TABLE 6.21.3.3 (CONTINUED)

(6 of 7)

 K_{Isec} SUMMARY FOR TITANIUM ALLOY Ti-8Al-1Mo-IV

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Crack (in)	K_Q (Ksi√in)	K_{Isec} (Ksi√in)	Test Time (min)	Test Date	Refer	
						Design	Width (in)	Thick (in)							
Mill Annealed	P	R.T.	T-S	123	3.5% NaCl	NB	0.5	...	45	20*	...	1969	75386
						CANT*	1	0.1	1	...	75	46*	...	1967	84327
						CANT*	1	0.25	1	...	75	38	...	1967	84327
						CANT	1	0.25	1	...	75	61*	...	1967	84327
						CANT	1	0.5	1	...	83	41	...	1967	84327
						CANT*	1	0.5	1	...	83	41	...	1967	84327
						CANT*	1	1	1	...	83	38	...	1967	84327
						CANT	1	1	1	...	83	39	...	1967	84327
Mill Annealed 1435° F 8hr FC	S	R.T.	T-S	145	3.5% NaCl	CNT	8	0.16	0.16	35	21	...	1967	70733	
						SENT	3	0.125	0.13	...	55.5	22	...	1969	77456
						CANT	1	0.27	44	29	...	1968	84326
VAC Annealed	P	R.T.	T-L	135	3.5% NaCl	CANT	1	0.27	...	102	27	...	1968	84326	
						CANT	1	0.27	65	29	...	1968	84326
						CANT*	1	0.27	65	34	...	1968	84326
						CANT*	1	0.27	44	34	...	1968	84326
						CANT	1	0.27	65	23	...	1968	84326
Methanol						CANT	1	0.27	...	44	29	...	1968	84326	
						CANT	1	0.27	44	29	...	1968	84326

Ti-8Al-1Mo-1V

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TABLE 6.21.3.3 (CONCLUDED)

K_{Isec} SUMMARY FOR TITANIUM ALLOY Ti-8Al-1Mo-1V

Condition/ Heat Treat	Prod Form	Test Temp (°F)	Spec Or.	Yield Str (Ksi)	Envir.	Specimen			Prod Thk (in)	Crack (in)	K_Q (Ksi√in)	K_{Isec} (Ksi√in)	Test Time (min)	Test Date	Refer
						Design	Width (in)	Thick (in)							
VAC Annealed (cont'd)	P (cont'd)	R.T. (cont'd)	T-L (cont'd)	135 (cont'd)	Methanol (cont'd)	CANT*	1	0.27	---	---	102	22	---	1968	84326

* specimen thickness does not meet minimum requirements of $2.5 \left(\frac{K_{Isec}}{\sigma_y} \right)^2$

* asterisk in specimen design column indicates that specimens are side-grooved

TABLE 6.22.1.1
MEAN PLANE STRAIN FRACTURE TOUGHNESS
FOR TITANIUM ALLOY Ti-Mo8V2Fe3Al AT ROOM TEMPERATURE

Product Form	Condition/Heat Treatment	K_{Ic} (ksi \sqrt{in})								
		Specimen Orientation								
		L-T			T-L			S-L		
		Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n
Plate	STA REAGED AT 1100F 6HR	54	1.	3	53.9	1.	3

Ti-Mo8V2Fe3Al

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TABLE 6.22.2.1

TITANIUM Ti-8Mo-8V-2Fe-3Al K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (KSI)	SPECIMEN			CRACK LENGTH (in.) A	3.5 * (K_{Ic} , TYS) (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (KSI*in.)	K_{Ic} MEAN	STAN DEV		
1475F 1.5 HR WQ 1000F 8 HR AC	Extrusion	3.00	155.0	...	1.498	0.750	CT	0.766	0.12	34.30	34.9	1.2	1973	87230	
		3.00	155.0	1.498	0.750	CT	0.747	0.12	34.10						
		3.00	155.0	1.500	0.750	CT	0.776	0.14	36.30						
STA REAGED AT 1100F 6 HR	Plate	1.00	170.0	L-T	1.989	0.998	CT	1.001	0.26	55.00	54.0	1.0	---	86429	
		1.00	170.0	1.990	1.002	CT	0.994	0.24	53.10						
		1.00	170.0	1.995	1.005	CT	0.995	0.25	53.90						
STA REAGED AT 1100F 6HR	Plate	1.00	177.0	T-L	1.991	0.991	CT	0.987	0.23	53.40	53.9	1.0	---	86429 (1)	
		1.00	177.0	1.995	0.998	CT	1.008	0.24	55.00						
		1.00	177.0	1.993	0.996	CT	1.014	0.23	53.20						

NOTES: (1) COMPOSITION(WT PERCENT) 2.26Al, 7.99V, 8.17Mo, 0.022C, 0.018N, 0.0070H, 0.160, 0.006Cu
ALPHA PRECIPITATE IN BETA MATRIX
STRAIGHTNESS OF CRACK FRONT MAY NOT MEET ASTM E399-72 REQUIREMENTS

TABLE 6.23.2.1

TITANIUM Ti-5Al2.5Sn ELI K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (ksi)	SPECIMEN			CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic}/\sqrt{A})^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (ksi√in.)	K_{Ic} MEAN	STAN DEV		
ANNEALED	Forging	---	-423	---	186.0	2.000	1.005	CT	1.060	0.32	66.60	67.5	4.4	1970	88439
		---									61.10				
		---									66.30				
		---									74.50				
		---									69.30				
		---									67.00				
		---									67.90				
		---									74.50				
ANNEALED	Forging	17.00	-423	R-L	189.0	2.000	1.000	CT	0.910	0.32	70.8	2.9	1970	88439	
		17.00													74.50
		17.00													71.50
		17.00													69.30
		17.00													52.40
		17.00													49.80
		17.00													57.30
		17.00													70.80
ANNEALED	Forging	17.00	-423	R-C	189.0	2.000	1.000	CT	1.040	0.17	53.2	3.8	1970	88439	
		17.00													57.30
		---													70.80
		---													70.00
		---													69.80
		---													71.20
		---													69.80
		---													70.00
ANNEALED EQUIAXED STRUCTURE	Forging	---	-423	---	186.0	2.000	1.001	CT	1.010	0.35	70.3	0.6	1970	88439	
		---													69.80
		---													71.20
		---													69.80
		---													70.00
		---													71.20
		---													69.80
		---													70.00

Ti-5Al2.5Sn ELI

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TABLE 6.23.2.1 (CONCLUDED)

TITANIUM Ti-5Al2.5Sn ELI K_{Ic}															
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN			CRACK LENGTH (in.) A	$2.0 \cdot (K_{Ic} / YS)^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B	DESIGN			K_{Ic} (Ksi/in.)	K_{Ic} MEAN	STAN DEV		
ANNEALED INTERMEDIATE STRUCTURE BETWEEN PLATELET ALPHA AND FINE EQUIAXED GRAINS	Forging	---	-423	---	186.0	2.000	1.000	CT	1.010	0.56	88.20	79.2	6.6	1970	88439
		---			186.0	2.000	1.001	CT	1.020	0.50	83.30			1970	88439
		---			186.0	2.000	1.002	CT	1.020	0.37	71.70			1970	88439
		---			186.0	2.000	1.001	CT	0.990	0.41	75.20			1970	88439
		---			186.0	2.000	1.002	CT	1.060	0.44	77.60			1970	88439
		---			---	---	---	---	---	---	---			---	---

TABLE 6.24.1.1

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**MEAN PLANE STRAIN FRACTURE TOUGHNESS
FOR TITANIUM ALLOY Ti-6Al6V2Sn ELI AT ROOM TEMPERATURE**

Product Form	Condition/Heat Treatment	K_{Ic} (ksi \sqrt{in})										
		Specimen Orientation						S-L				
		L-T		T-L		S-L		T-L		S-L		
Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	Mean K_{Ic}	Std Dev	n	
Plate	1600F 1 HR WQ 1050F 4HR AC	29.8	0.5	3
	1650F 1 HR WQ 1050F 4HR AC	34	3.5	2

TABLE 6.24.2.1

1 of 1

TITANIUM Ti-6Al6V-2Sn ELI K_{Ic}														
CONDITION	PRODUCT		TEST TEMP (°F)	SPEC OR	YIELD STR (Ksi)	SPECIMEN		CRACK LENGTH (in.) A	$2.5 \cdot (K_{Ic}/TYS)^2$ (in.)	K_{Ic}			DATE	REFER
	FORM	THICK (in.)				WIDTH (in.) W	THICK (in.) B			DESIGN	K_{Ic} (Ksi√in.)	K_{Ic} MEAN		
1600F 1 HR WQ 1050F 4 HR AC	Plate	1.00	R.T.	L-S	179.0	0.499	0.250	NB	0.184	0.07	30.10	32.4	1965	84316
		1.00									32.10			
		1.00									35.10			
1600F 1 HR WQ 1050F 4 HR AC	Plate	1.00	R.T.	L-T	179.0	0.499	0.255	NB	0.177	0.07	30.20	29.8	1965	84316
		1.00									30.00			
		1.00									29.20			
1650F 1 HR WQ 1125F 4 HR AC	Plate	1.00	-320	L-S	258.0	0.499	0.250	NB	0.206	0.02	22.60	23.7	1965	84316
		1.00									24.70			
		1.00									24.70			
1650F 1 HR WQ 1125F 4 HR AC	Plate	1.00	R.T.	L-S	170.0	0.499	0.250	NB	0.191	0.12	37.50	38.1	1965	84316
		1.00									38.60			
		1.00									36.50			
1650F 1 HR WQ 1125F 4 HR AC	Plate	1.00	R.T.	L-T	170.0	0.499	0.251	NB	0.219	0.11	31.50	34.0	1965	84316
		1.00									31.50			

TABLE 6.25

REFERENCES FOR THE TITANIUM ALLOY DATA

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|-------|---|----------------|
| 54304 | Ti-6Al-4V | K _c |
| | Figge, I. E., "Residual Static Strength of Several Titanium and Stainless Steel Alloys and One Superalloy at - 109 F, 70 F, and 550F," NASA TN D-2045, Langley Research Center (December 1963). | |
| 57573 | Ti-6Al-4V | K _c |
| | Anon., "Fracture Toughness and Tear Tests," Air Force Materials Laboratory, Research and Technology Division, Report No. ML-TDR-64-238, October 1964. | |
| 58782 | Ti-6Al-4V | K _c |
| | Anon., "Thick Section Fracture Toughness," ML-TDR-64-236, Boeing-North American (October 1964). | |
| 60578 | Ti-6Al-4V ELI | K _c |
| | Christian, J. L., Yang, C. T., and Witzell, W. E., "Physical and Mechanical Properties of Pressure Vessel Materials for Application in a Cryogenic Environment," ASD-TDR-62-258, Part III, General Dynamics/Astronautics (December 1964). | |
| 66103 | Ti-5Al-2.5Sn | K _c |
| | Ferguson, C. W., "Hypervelocity Impact Effects on Liquid Hydrogen Tanks," NASA CR-54852, Douglas Aircraft Co., Inc. (March 1966). | |
| 66218 | Ti-5Al-2.5Sn | K _c |
| | Tiffany, C. F., Lorenz, P. M., and Hall, L. R., "Investigation of Plane-Strain Flaw Growth in Thick-Walled Tanks," NASA CR-54837, The Boeing Company, (February 1966). | |
| 67821 | Ti-8Al-1Mo-1V | K _c |
| | Walker, E. K., "A Study of the Influence of Geometry on the Strength of Fatigue Cracked Panels," AFFDL-TR-66-92, Northrop Norair (June 1966). | |

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

- | | | | |
|-------|--|--|---|
| 68968 | Ti-5Al-2.5Sn | K_I | Sullivan, T. L., "Uniaxial and Biaxial Fracture Toughness of Extra-Low-Interstitial 5Al-2.5Sn Titanium Alloy Sheet at 20 K," NASA TN D-4016, Lewis Research Center, (June 1967). |
| 70733 | Ti-8Al-1Mo-1V | K_{Isc} | Smith, H. R., et al., "A Study of Stress Corrosion Cracking by Wedge-Force Loading," Report D6-19768, The Boeing Company, Renton, Wash., Contract N00014-66-C-0365, (June 1967). |
| 70887 | Ti-6Al-4V
Ti-6Al-6V-2.5Sn
Ti-8Al-1Mo-1V | K_{Isc}
K_{Isc}
K_{Isc} | Peterson, M. H., Brown, B. F., Newbegin, R. L., and Groover, R. E., "Stress Corrosion Cracking of High Strength Steels and Titanium Alloys in Chloride Solutions at Ambient Temperature," Corrosion, <u>23</u> (5), 142-148 (May 1967). |
| 70931 | Ti-6Al-4V
Ti-6Al-4V ELI
Ti-6Al-6V-2.5Sn
Ti-8Al-1Mo-1V | K_{Isc}
K_{Isc}
K_{Isc}
K_{Isc} | Judy, Jr., R. W., and Goode, R. J., "Stress-Corrosion Cracking Characteristics of Alloys of Titanium in Salt Water," Interim Report 6564, Naval Research Laboratory, Washington, D. C., Contracts NONR-610(09), NONR-760(31) and N00014-66-C0365 (July 21, 1967). |
| 71709 | Ti-8Al-1Mo-1V | K_I | Figge, I. E., "Residual-Static-Strength and Slow-Crack-Growth Behavior of Duplex-Annealed Ti-8Al-1Mo-1V Sheet," NASA TN D-4358, Langley Research Center, (March 1968). |
| 74355 | Ti-6Al-4V | K_{Isc} | Stanley, J. K., "Solutions to Some Stress Corrosion Cracking Problems in Aerospace Applications," Technical Report TR-0200 (4112.22)-1, Aerospace Corporation, El Segundo, Calif., Contract F04701-68-C-0200 (November 1968). |

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

- | | | |
|-------|--|--|
| 75386 | Ti-4Al-3Mo-1V
Ti-6Al-4V
Ti-8Al-1Mo-1V | K_{Isc}
K_{Isc}
K_{Isc} |
| | Curtis, R. E., et al., "Relationship Between Composition, Micro-structure, and Stress Corrosion Cracking (In Salt Solution) In Titanium Alloys," ASM Transactions Quarterly, 62 (2), 457-469 (June 1969). | |
| 75528 | Ti-6Al-4V | K_{Isc} |
| | Bixler, W. D., "Fracture Characteristics of 6Al-4V Titanium Alloy Forgings Containing Alpha Stringer Microstructure," NASA Final Report CR-99512, The Boeing Company, Seattle, Wash., Contract NAS-9-8809 (January 1969). | |
| 76411 | Ti-6Al-4V | K_{Ic} |
| | Wessel, E. T., et al., "Engineering Methods for the Design and Selection of Materials Against Fracture," Final Technical Report, Westinghouse Research Laboratories, Pittsburgh, PA., Contract DA-30-069-AMC-602 (T) (June 24, 1966). | |
| 77290 | Ti-6Al-4V | K_{Isc} |
| | Reuter, W. G., et al., "Monitoring of Crack Growth on Ti-6Al-4V Alloy by the Stress Wave Analysis Technique," NASA Report CR-101888, Aerojet General Corporation, Sacramento, Calif. (December 1968). | |
| 77456 | BETA
BETA Ti
Ti-8Al-1Mo-1V | K_{Isc}
da/dt
$da/dt; K_{Isc}$ |
| | Katz, Y., "Micro-Mechanical Approach to Stress Corrosion Cracking in Titanium Alloys," Thesis Report No. UCRL-19046, University of California, Lawrence Radiation Laboratory, Berkeley, Calif., AEC Contract W-7405-eng-48 (September 1969). | |
| 78535 | Ti-6Al-4V | K_{Isc} |
| | Masters, J. N., et al., "Fracture and Nitrogen Tetroxide/Sustained Load Flaw Growth in 6Al-4V Titanium," Final Report D2-121397-1, NASA CR-109366, The Boeing Company, Seattle, Wash., Contract NAS7-100 (October 1969). | |

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

- | | | | |
|-------|-----------------------------------|--|--|
| 80104 | Ti-5Al-2.5Sn | K_{Ic} | Orange, T. W., Sullivan, T. L., and Calfo, F. D., "Fracture of Thin Sections Containing Through and Part-Through Cracks," NASA TN D-6305, Lewis Research Center (April 1971). |
| 80538 | Ti-6Al-4V | K_{Ic} | Petрак, G. J., "Mechanical Property Evaluation of Beta Forged Ti-6Al-4V," Report AFML-TR-70-291, University of Dayton Research Institute, Dayton, Ohio, Contract F33615-69-C-1471 (January 1971). |
| 81221 | Ti-6Al-4V | $da/dt; K_{Isc}$ | Beck, T. R., et al., "Stress Corrosion Cracking of Titanium Alloys: Studies of Cracks in Thin Specimens; SCC of Ti-6Al-4V in Chloride, Iodide and Fluoride Solutions; Stress Corrosion Cracking in Molten Salts; Electrochemistry of Freshly Generated Titanium Surfaces," Report 20, The Boeing Company, Seattle, Wash., Contract NAS 7-489, (June 1971). |
| 81741 | Ti-8Al-1Mo-1V | da/dt | Bucci, R. J., and Paris, P. C., "Observations on Sustained Load Environmental Crack Growth of Titanium 8Al-1Mo-1V Alloy," Corrosion, <u>27</u> (12) 525-530 (December 1971). |
| 82651 | BETA
BETA III
Ti-8Al-1Mo-1V | K_{Isc}
$da/dt; K_{Isc}$
da/dt | Beck, T. R., and Blackburn, M. J., "Stress Corrosion Cracking of Titanium Alloys; SCC Velocity; Concentration of $TiCl_3$," Report 18, The Boeing Company, Seattle, Wash., Contract NAS 7-489 (December 1970). |
| 83222 | Ti-6Al-6V-2Sn | K_{Ic} | Amateau, M. F., et al., "F-15 Program Final Report Ti-6Al-6V-2Sn and Ti-6Al-4V Fatigue Crack Propagation," Report No. ATR-72(9990)-3, The Aerospace Corporation, El Segundo, Calif. (September 29, 1971). |

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

- | | | | |
|-------|---------------------------------|----------------------|--|
| 83689 | Ti-8Al-1Mo-1V | $da/dt; K_{Isc}$ | Beck, T. R., et al., "Fundamental Investigation of Stress Corrosion Cracking," Report D180-15006-1, The Boeing Company, Aerospace Group, Seattle, Washington, Contract NASW-2245 (April 1962). |
| 83984 | Ti-6Al-4V | K_{Ic} | Feddersen, C. E., and Hyler, W. S., "Fracture and Fatigue-Crack-Propagation Characteristics of 1/4-Inch Mill-Annealed Ti-6Al-4V Titanium Alloy Plate," Report G-9706, Battelle, Columbus Laboratories, Columbus, Ohio, Contract N00156-70-C-1336 (November 1, 1971). |
| 84036 | Ti-6Al-4V | K_{Isc} | Beachem, C. D., and Meyn, D. A., "The Effect of Thickness Upon Sustained Load Crack Propagation in Ti-6Al-4V Alloy Tested in 3.5% NaCl Solution," NRL Report 7449, Naval Research Laboratory, Washington, D. C. (August 8, 1972). |
| 84282 | Ti-6Al-4V | K_{Isc} | Spurr, W. F., "SST Technology Follow-On Program, Phase I, Titanium Alloy 6Al-4V Extrusion," Report No. FAA-SS-72-06, Boeing Report D6-60206, The Boeing Company, Commercial Airplane Group, Seattle, Wash., Contract No. DOT-FA-SS-71-12, (July 1972). |
| 84290 | Ti-8Al-1Mo-1V | da/dt | Smith, H. R., Piper, D. E., and Downey, F. K., "A Study of Stress-Corrosion Cracking by Wedge Force Loading," Engineering Fracture Mechanics, <u>1</u> , p 123-128 (1968), Pergamon Press. |
| 84306 | Ti-6Al-4V | K_{Ic} | Harrigan, M. J., "B-1 Fracture Mechanics Data for Air Force Handbook Usage," Report TFD-72-501, North American Rockwell, Los Angeles Division, Los Angeles, CA (April 21, 1972). |
| 84316 | Ti-6Al-6V-2Sn
Ti6Al6V2Sn ELI | K_{Ic}
K_{Ic} | DeSisto, T. S., and Hickey, Jr., C. F., "Low-Temperature Mechanical Properties and Fracture Toughness of Ti-6Al-6V-2Sn," ASTM Proceedings, <u>65</u> , 641-653 (1965). |

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

- | | | | |
|-------|---|--|---|
| 84326 | Ti-8Al-1Mo-1V | K_{Isc} | Report of NRL Progress, Naval Research Laboratory, Washington, D. C., (May 1968). |
| 84327 | Ti-8Al-1Mo-1V | K_{Isc} | Report of NRL Progress, Naval Research Laboratory, Washington, D. C., (May 1967). |
| 84328 | Ti-6Al-4V | K_{Isc} | Report of NRL Progress, Naval Research Laboratory, Washington, D. C., (November 1968). |
| 84360 | Ti-6Al-4V
Ti-6Al-6V-2.5Sn
Ti-6Al-6V-2Sn | K_{Ic} ; da/dN; da/dt; K_{Isc}
K_{Isc}
K_{Ic} ; da/dN; da/dt | McDonnell Aircraft Company, McDonnell Douglas Corp., St. Louis, Mo., Phase B Test Program, Report MDC A0913 (May 18, 1971). |
| 85034 | Ti-6Al-4V | K_{Ic} | Mitchell, John, "Laboratory Reports on Fracture Toughness Tests," per memo from Ed Cawthorne of February 5, 1973; data sheets from Schultz Steel Co., South Gate, CA. |
| 85064 | Ti-6Al-4V | K_{Ic} | Bjeletich, J. G., "Development of Engineering Data on Thick-Section Electron-Beam-Welded Titanium," Report N-LJ-71-18, Lockheed Aircraft Corporation, Lockheed Palo Alto Research Laboratory, Palo Alto, Calif., Contract F33615-71-C-1338, (October 12, 1972). |
| 85634 | Ti-6Al-4V | K_{Ic} | "Fracture Toughness and Tensile Properties Data for Ti-6Al-4V Forgings," Schultz Steel Company, South Gate, Calif., Attached to memo from Ed Cawthorne dated March 5, 1973. |
| 85836 | Ti-6Al-4V | K_{Ic} | "B-1 Fracture Toughness Data (K_{Ic}) - Rockwell International," Rockwell International Corporation, Los Angeles, CA (April 24, 1973). |

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

- 85837 Ti-6Al-4V a-vs-N; da/dN
 "Fracture Toughness Data Collection, Rockwell International Corporation, from B-1 Program," Rockwell International Corporation, Los Angeles, CA., April 1973.
- 85855 Ti-8Al-1Mo-1V da/dt; K_{Isc}
 Boyd, J. D., "Stress-Corrosion Cracking of Ti-8Al-1Mo-1V in Aqueous Environments: 1. The Kinetics of Subcritical Crack Propagation," Metallurgical Transactions, 4 (4) 1029-1035 (April 1973).
- 85857 Ti-6Al-4V K_{Ic}
 "Shultz Steel Company - Fracture Toughness Data - May 10, 1973," per memo from Ed Cawthorne of May 10, 1973.
- 86099 Ti-8Al-1Mo-1V a-vs-N; da/dN
 Wanhill, R. J. H., "A Review and Analysis of Fatigue Crack Propagation in Titanium Alloys at Room Temperature," Report No. NRL-TR-71035U, National Aerospace Laboratory, The Netherlands (February 1971).
- 86429 Ti-Mo8V2Fe3Al K_{Ic}
 "Fracture Toughness Data," Progress Report on Materials Test Program, General Dynamics Corporation, Fort Worth Division, Fort Worth, Texas, Contract F33615-72-C-2149 (Received July 6, 1973).
- 86494 Ti-6Al-6V-2Sn K_{Ic}
 Harmsworth, C. L., "Evaluation Report - Fracture Toughness Evaluation of Isothermally Forged Ti-6Al-6V-2Sn STA," Report MXE 72-64, Air Force Materials Laboratory, WPAFB, Ohio (November 15, 1972).
- 86575 Ti-6Al-4V a-vs-N; da/dN
 "Rockwell International, B-1 Program, da/dN Data, Center-Cracked Tension Specimens," Lockheed California Company, Burbank, CA, Report LR25152 (Received July 1973) (Memo from Ed Cawthorne dated July 10, 1973).

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

86688	Ti-6Al-4V	K_{Ic} ; K_{Isc}	Sprowls, D. O., et al., "Evaluation of Stress Corrosion Cracking Susceptibility Using Fracture Mechanics Techniques," Final Report Part I, Aluminum Co. of America, Alcoa Technical Center, Alcoa, Pa., Contract NASA-21487, May 31, 1973.
86844	Ti-6-2-2-2-2 Ti-6Al-6V-2Sn	a-vs-N; da/dN a-vs-N; da/dN	"Crack Growth Rate Data Generated Under USAF Contract F33615-72-C-2165," Lockheed Aircraft Corporation, Lockheed-Georgia Company, Marietta, GA, Contract F33615-72-C-2165, Data Sheets received from AFFDL August 13, 1973.
87230	BETA III Ti-6Al-6V-2Sn Ti-Mo8V2Fe3Al	K_{Ic} K_{Ic} K_{Ic}	DeSisto, T. S., "Fracture Toughness Measurements of Three Titanium Alloy Extrusions," Report AMMRC-TR-73-31, Army Materials and Mechanics Research Center, Watertown, MA (July 1973).
88140	Ti-6Al-4V Ti-6Al-4V ELI	da/dN da/dN	Hall, L. R., Finger, R. W., and Spurr, W. F., "Corrosion Fatigue Crack Growth in Aircraft Structural Materials," Report AFML-TR-73-204, Boeing Aerospace Company, Seattle, WA. Contract AF33615-71-C-1687 (September 1973).
88144	BETA III Ti-6Al-4V	K_{Ic} ; da/dN K_{Ic} ; da/dN; da/dt	Bjeletich, J. G., "Development of Engineering Data in Thick-Section Electron Beam Welded Titanium," Report AFML-TR-73-197, Lockheed Missiles and Space Company, Inc., Palo Alto, CA, Contract F33615-71-C-1338 (August 1973).
88186	TI-* Ti-6Al-6V-2Sn	K_{Ic} K_{Ic}	"Selected Pages from Materials Section of Final Report on Cargo/Tanker Phase IA (AFFDL-TR-73-51) Lockheed Report SMN 378," Summary Report, Lockheed-Georgia Company, Marietta, GA, Contract F33615-72-C-2165 (February 8, 1974).

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

- | | | | |
|-------|---------------------|--|---|
| 88439 | Ti5Al2.5Sn ELI | K_{Ic} | Reuter, W. G., "Fracture Toughness of Ti-5Al-2.5Sn ELI Forgings at -423 F," Summary Report, Aerojet-General Corporation, Sacramento, CA (September 9, 1970). |
| 88440 | Ti-6Al-4V | K_{Ic} | "Titanium-6Al-4V Fracture Toughness and Tensile Test Data of December 19, 1973," Memo E. W. Cawthorne with fracture toughness data from Schultz Steel Company (December 19, 1973). |
| 88468 | Ti-6Al-4V | a-vs-N; da/dN | Bell, P. D., "Data Sheets for Constant Amplitude Crack Growth Data Generated by Grumman Aerospace Corporation for 2219-T851 Aluminum and Mill Annealed 6Al-4V Titanium Alloy Plate," letter to J. E. Campbell from Grumman Aerospace Corporation, Bethpage, NY, Contract F33615-72-C-1744 (March 15, 1974). |
| 88575 | BETA C
Ti-6Al-4V | K_{Ic} ; a-vs-N; da/dN
K_{Ic} ; a-vs-N; da/dN | "Advanced Metallic Air Vehicle Structure Program," Material Property Data Test Report Phase II, Report FZM-6148A, General Dynamics, Convair Aerospace Division, Fort Worth, TX, Contract AF33615-73-C-3001 (January 1974). |
| 88579 | Ti-6Al-4V | a-vs-N; da/dN | "B-1 Program da/dN Data for Aluminum Alloys," Rockwell International Corporation, memorandum to H. D. Moran from E. W. Cawthorne, Battelle's Columbus Laboratories (April 3, 1974). |
| 88700 | Ti-6Al-4V | K_{Isc} | Gilbreath, W. P., and Adamson, M. J., "The Stress Corrosion Susceptibility of Several Alloys in Hydrazine Fuels," NASA Technical Note, Report NASA TN D-7604, Ames Research Center, Moffett Field, CA (February 1974). |
| 88911 | Ti-5Al-2.5Sn | a-vs-N; da/dN | Wanhill, R. J. H., et al., "Fatigue Crack Propagation Data for Titanium Sheet Alloys - Interim Report No. 3:Ti-5Al-2.5Sn," Report NLR TR 72093 U, National Aerospace Laboratory, the Netherlands (July 1972). |

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

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|-------|--|----------------------------------|
| 88962 | Ti-6Al-4V
Ti-6Al-6V-2Sn
Ti-6Al2Sn4Zr6Mo | K_{Ic}
K_{Ic}
K_{Ic} |
| | Sparks, R. B., and Long, J. R., "Improvement Manufacturing Methods for Producing High Integrity More Reliable Titanium Forgings," Report AFML-TR-73-301, Wyman-Gordon Company, Worchester, MA, Contract AF33615-71-C-1560 (February 1974). | |
| 89004 | Ti-6Al-4V | K_{Ic} ; K_{Isc} |
| | "Rockwell International, B-1 Program Titanium K_{Ic} , K_{Ic} , and K_{Isc} Data for HB-01 Revision," with data attached to memorandum from E. W. Cawthorne to H. D. Moran (May 1, 1974). | |
| 89504 | Ti-6Al-4V | K_{Ic} |
| | Cervay, R. R., "Mechanical Properties of Ti-6Al-4V Annealed Forgings," Report AFML-TR-74-49, University of Dayton Research Institute, Dayton, OH, Contract F33615-72-C-1282 (March 1974). | |
| 90012 | Ti-6Al-4V | K_{Ic} |
| | "Ti-6Al-4V Fracture Toughness Data - Shultz Steel Company, South Gate, CA, of August 8, 1974," with memorandum from E. W. Cawthorne to H. D. Moran of Battelle's Columbus Laboratories (August 8, 1974). | |
| 90584 | Ti-6Al-4V
Ti-6Al-6V-2Sn | K_{Ic}
K_{Ic} |
| | DeMay, S., "Improved Fracture Toughness of Titanium," Final Report, Grumman Aerospace Corporation, Bethpage, NY, Contract N62269-73-C-0127 (June 1973) (AD 778 652). | |
| 90589 | Ti-6Al-6V-2Sn | K_{Ic} |
| | Fiftal, C. F., and Beck, E. J., "Development of Fracture Mechanics Data for 6Al-6V-2Sn Titanium Alloy," Report MCR-74-43, Martin Marietta Corp., Denver Division, Denver, CO, Contract NAS 9-13599 (January 1974). | |

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

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|-------|---|--|
| 90981 | Ti-6Al-4V
Ti-6Al-6V-2Sn | a-vs-N; da/dN
K _{ic} ; a-vs-N; da/dN |
| | Krupp, W. E., Wimmer, F. T., Pettit, D. E., and Hoepfner, D. W., Data Sheets for Final Report on "Investigation of the Effects of Stress and Chemical Environments on the Prediction of Fracture in Aircraft Structural Materials," Rye Canyon Research Laboratory, Lockheed-California Company, Burbank, CA, Contract F33615-71-C-1688, data sheets received October 21, 1974. | |
| 91332 | Ti-6Al-4V | da/dN |
| | Wells, R. R., "New Alloys for Advanced Metallic Fighter-Wing Structures," Northrop Corporation, Aircraft Division, Hawthorne, CA, AIAA/ASME/SAE 15th Structures, Structural Dynamics and Materials Conference, Las Vegas, NV (April 17-19, 1974). | |
| 91793 | BETA III | K _{ic} |
| | Van Stone, R. H., Low, J. R. Jr., and Shannon, J. L., Jr., "The Effect of Microstructure on the Fracture Toughness of Titanium Alloys," NASA Technical Report No. 2-Ti, Carnegie-Mellon University, Pittsburgh, Pa., and NASA, Lewis Research Center, Cleveland, Ohio, Research Grant NGR 39-087-047 (December 1974). | |
| 91945 | Ti-6Al-6V-2Sn | da/dN |
| | Kondas, K. R., et al., "Cyclic-Crack-Growth and Fracture Resistance of Ti-6Al-6V-2Sn as Influenced by Recrystallization Anneal and Interstitial Oxygen Content," NRL Report 7844, Naval Research Laboratory, Washington, D. C. (January 8, 1975). | |
| GD007 | Ti-6Al-4V | a-vs-N; da/dN |
| | Margolis, W. S., "F-16 Material Allowables Evaluating Beta Annealed 6Al-4V Titanium Alloy," General Dynamics, Fort Worth Division, Report No. 16PR944, September 1978. | |
| GE006 | Ti-8Al-1Mo-1V | da/dN |
| | "Fatigue Crack Growth Rate Data on Titanium Ti-8-1-1 Alloy Using K _{ic} Bar Specimens from the TF34 DTA Effort". Data Sent from M. S. Gilbert, General Electric Co., Evendale, Ohio, October 1982. | |

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

GE007	Ti-6Al-4V	da/dN	"Fatigue Crack Growth Rate Data on Titanium Ti-6-4 Alloy Using Kb Bar Specimens from the TF34 DTA Effort," Data sent from M. S. Gilbert, General Electric Co., Evendale, Ohio, October 1982.
JEM01	Ti-6Al-4V Ti-6Al-6V-2Sn	K_{Ic} K_{Ic}	Yoder, G. R., Cooley, L. A., and Crooker, T. W., "Enhancement of Fatigue Crack Growth and Fracture Resistance in Ti-6Al-4V and Ti-6Al-6V-2Sn Through Micro-Structural Modification," Journal of Engineering Materials and Technology, Vol. 99, pp. 315-318, October 1977.
MA002	Ti-6Al-4V Ti-6Al-6V-2Sn	K_{Ic} ; a-vs-N; da/dN a-vs-N; da/dN	Fracture Toughness of Ti-6Al-4V Plate and Forging, Aluminum 2124-T851 Plate and 7175-T37652 Forging and Fatigue Crack Growth Rate for Ti-6Al-4V Plate and Forging, Ti-6Al-6V-2Sn Extrusion, Aluminum 2124-T851 Plate and Aluminum 7175-T73652 Forging, Data Submitted by D. L. Rich of McDonnell Aircraft Co., St. Louis, MO, Attachment #2, Received March 12, 1982.
MA003	Ti-6Al-4V	K_{Ic} ; a-vs-N; da/dN	Seay, S., and Krieg, J. F., "Wing Carry Through Lug Joint Crack Growth Tests," McDonnell Aircraft Co., St. Louis, MO, Report MDC A3449, Attachment #3, Submitted by D. L. Rich, March 12, 1982.
MA005	Ti-6Al-4V	da/dN; K_{Isc}	Garland, K., and Krieg, J. F., "Final Report - Basic Fracture Data for F-18 Material," McDonnell Aircraft Company, St. Louis, MO, Report No. 3 NA-66-7KW, Attachment #5, March 1977.
MA006	Ti-6Al-4V	da/dN	Garland, K., and Krieg, J. F., "Evaluation of the Effect of Material Cyclic Softening and Hardening on Crack Initiation Life and Crack Growth, with and without Overloads as a Function of Stress Ratio," McDonnell Aircraft Company, St. Louis, MO, April 1978.

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

MA011	Ti-8Al-1Mo-1V	da/dN	
	"Final Report, F/RF-4C/D Damage Tolerance and Life Assessment Study - Vol. II," McDonnell Aircraft Company, St. Louis, MO, Contract No. AFSC F33657-73-A-0062, Report No. MDC A2883, February 1975.		
NC001	Ti-6Al-4V Ti-6Al-4V ELI	K_{Ic} K_{Ic}	
	"Plane Strain Fracture Toughness Data Sets on Aluminum, Steel, and Titanium Alloys," Data sent from P. G. Porter of Northrop Corp., Hawthorne, CA, March 1, 1982.		
NC002	Ti-6Al-4V Ti-6Al-4V ELI	a-vs-N; da/dN a-vs-N; da/dN	
	"Fatigue Crack Growth Rate Data on Aluminum, Steel, and Titanium Alloy," Data sent from P. G. Porter, Northrop Corp., Hawthorne, CA, March 1, 1982.		
NL001	Ti-6Al-4V	a-vs-N; da/dN	
	Data Sheets on Ti-6Al-4V - Fatigue Crack Growth Rate Tests - submitted by W. S. Johnson, NASA Langley Research Center, Hampton, VA, March 1982.		
NR001	Ti-6Al-4V Ti-8Al-1Mo-1V	K_{Ic} K_{Isc}	
	Yoder, G. R., Cooley, L. A., and Crooker, T. W., "Effects of Microstructure and Frequency on Corrosion-Fatigue Crack Growth in Ti-8Al-1Mo-1V and Ti-6Al-4V," Naval Research Laboratory, Washington, D. C., Report 4678, December 1981.		
PW002	Ti-6-2-4-2 Ti-6-2-4-6 Ti-8Al-1Mo-1V	da/dN da/dN K_{Ic} ; da/dN	
	Beyer, J. R., Sims, D. L., and Wallace, R. M., "Titanium Damage Tolerant Design Data for Propulsion Systems," United Technologies Corp., Pratt and Whitney Aircraft Group, West Palm Beach, FL, Report AFML-TR-77-101, Contract No. F33615-75-C-5130, June 1977.		

TABLE 6.25 (CONTINUED)

REFERENCES FOR THE TITANIUM ALLOY DATA

PW003	Ti-6Al-4V	da/dN
	Fatigue Crack Growth Rate Data on Titanium and Nickel Base Alloy from B. S. Schwartz, Pratt and Whitney Aircraft Group, Government Products Division, West Palm Beach, FL, July 1982.	
RI005	CORONA 5 Ti-6Al-4V Ti-6Al2Sn4Zr6Mo	K_{Ic} K_{Ic} K_{Ic}
	Chesnutt, J. C., et al., "Fatigue Crack Propagation and Fracture of Titanium Alloys," Rockwell International Science Center, Thousand Oaks, CA, and Department of Metallurgy and Material Science, Carnegie-Mellon University, Pittsburgh, PA.	
RI006	Ti-6Al-4V	K_{Isc}
	Ferguson, R. R., and Berryman, R. C., "Fracture Mechanics Evaluation of B-1 Materials," Rockwell International, B-1 Division, Los Angeles, CA, Contract No. F33657-70-C-0800, Report No. AFML-TR-76-137, October 1976.	
SA001	Ti-6Al-4V	a-vs-N; da/dN
	Rungta, R., Deel, O., and Frey, N., "Fatigue-Crack-Propagation Behavior of Materials of Several H-53 Helicopter Materials," Sikorsky Aircraft Company, Stratford, CT, Data Submitted by Battelle, Columbus, OH, 43201, Received October 20, 1992.	
UD001	Ti-6-2-4-6	da/dN
	Cervay, R. R., "Ti-6-2-4-6 Elevated Temperature Modeling of Fatigue Crack Growth Rate Data," University of Dayton Research Institute, Dayton, Ohio, Contract No. F33615-80-C-5011 Technical Memorandum UDR-TM-81-48, December 1981.	
UD008	Ti-6Al-4V	K_{Ic} ; da/dN
	Cervay, R. R., "Beta Processed Titanium 6Al-4V Fracture Properties," University of Dayton Research Institute, Dayton, Ohio, Contract No. F33615-74-C-5024, Technical Memorandum UDRI-TM-75-08, July 1975.	

TABLE 6.25 (CONCLUDED)

REFERENCES FOR THE TITANIUM ALLOY DATA

UM001

Ti-6Al-4V

da/dN

Wilson, D. A., "A Statistically Based Investigation of Microstructural Effect on the Fatigue Properties of Titanium and Titanium Alloys," Dissertation to University of Missouri, in Partial Fulfillment of PhD Requirement, December 1978.