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National Aeronautics and  
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**George C. Marshall Space Flight Center**  
Marshall Space Flight Center, Alabama 35812

TEST CONTROL DOCUMENT FOR  
ASSESSMENT OF FLEXIBLE LINES  
FOR FLOW INDUCED VIBRATION

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TEST CONTROL DOCUMENT FOR  
ASSESSMENT OF FLEXIBLE LINES  
FOR FLOW INDUCED VIBRATION



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## REVISION LOG

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NOTE: The portion of this specification affected by the changes is indicated by a vertical line in the outer margins of the page.

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## 1.0 GENERAL

1.1 SCOPE. This specification outlines the methods and criteria to employ in flow testing bellows and flex hose assemblies.

1.2 PURPOSE. To establish a standard flow test procedure for detecting flow induced vibrations in a given bellows and flex hose assembly. The MSFC Drawing #20M02540 provides the analytical assessment for predicting cyclic stress, excitation frequencies, and resonant flow ranges.

## 2.0 TEST SPECIMENS

2.1 SELECTION. Specimens shall be randomly selected from a production run. Bellows and flex hose assemblies selected as representative of several assemblies for qualification purposes should represent equal or worst case conditions.

## 3.0 TEST REQUIREMENTS

3.1 GENERAL. All testing must be performed simulating actual operating life conditions (i.e., medium, temperature, internal and external pressure, etc.,) unless otherwise approved by MSFC.

3.2 NON-VACUUM JACKETED BELLOWS. When testing non-vacuum jacketed bellows with both a cryogenic flow media and an external ambient environment, provisions must be made to avoid the dampening effect of liquid air, frost, or slush on the convolutes. A helium purge environment may be used if heat transfer problems will not occur; otherwise, a vacuum environment ( $10^{-5}$  to  $10^{-6}$  torr) must be used.

3.3 FLOW MEDIA. The bellows and flex hose design fluid media shall be used for flow testing and conform to all requirements for its intended operating condition unless otherwise approved by MSFC. If a substitute media is to be used, an additional analysis must be performed to verify the bellows or flex hose design integrity.

3.4 FACILITIES. The flow facilities shall be capable of flowing the service media under the bellows or flex hose actual operating conditions for the duration of four (4) times the operational life and be capable of flowing +/- 10 percent of the expected flow range.

3.5 OPERATIONAL LIFE. Operational life shall be defined as the time accrued when the number of missions is multiplied by the mission system operation time plus any ground checkout operations.

#### 4.0 TEST SET-UP

4.1 GENERAL. The flow test set-up shall include the actual operating configuration assemblies and supports. The bellows and flex hose elements of the line assembly shall be deflected for worst case static loading. Installation of flow affecting hardware; such as orifices, transitions, valves, or bends, within ten (10) diameters upstream of the bellows is required.

4.2 INSTRUMENTATION. Each bellows and flex hose assembly shall be instrumented to measure cyclic stress levels and excitation frequencies without impairing freedom of movement or the integrity of the bellows assembly. Flowrate through the bellows and flex hose assembly shall also be accurately measured. Instrumentation shall conform to the state-of-the-art hardware, where practical and feasible. Due to the variety of methods that can be used to gather strain gage and accelerometer data, it will be left to the discretion of the testing organization, with MSFC approval, as to the method employed. The method chosen must adhere to the local codes and abide by current approved practices.

4.2.1 STRAIN GAGE ATTACHMENT. If strain gages are used, they may be attached using an adhesive such as epoxy (Eastman 910 or equivalent). Surface preparation generally requires an agent and light sandblasting. Location of the gages shall be as follows:

Free Bellows: Locate gages on the second convolute crown from the upstream end and on the second convolute crown from the downstream end. Also, locate gages at half the length of the convoluted section. See figure 1 for strain gage locations.

Flex hose: A wire braid cut-out is required to install strain gage (figure 2). Locate near upstream end for liquid flow and near downstream end for gas flow.

#### 5.0 TEST PROCEDURES

5.1 INSTALLATION. With strain gages installed, deflect the bellows assembly (i.e. axial deflection, lateral off-set, and angulation) to worst case static loading seen during actual operating conditions.

5.2 RESONANCE SEARCH. With the bellows or flex hose installed to simulate worst case loading possible during operating conditions, conduct a resonance search by slowly varying the flow rate from zero flow to maximum operating flow plus 10%. Caution shall be exercised to vary the flow at a rate which will not allow resonance to go undetected. Substantiate the search by a downsweep through the same range of flows. Record frequencies, strain levels, and corresponding flow rates. The data from this search is then studied to determine the flow rate at which to perform the resonant dwell testing. If a flow resonance is not detected, continue the flow scan until four (4) times the operational life requirements are met, or a total scan time of 3 hours has been obtained.

5.3 RESONANCE DWELL. This test will consist of dwelling at the most severe resonant conditions until the number of cycles equivalent to four (4) times the operational life have been accumulated at each resonance.

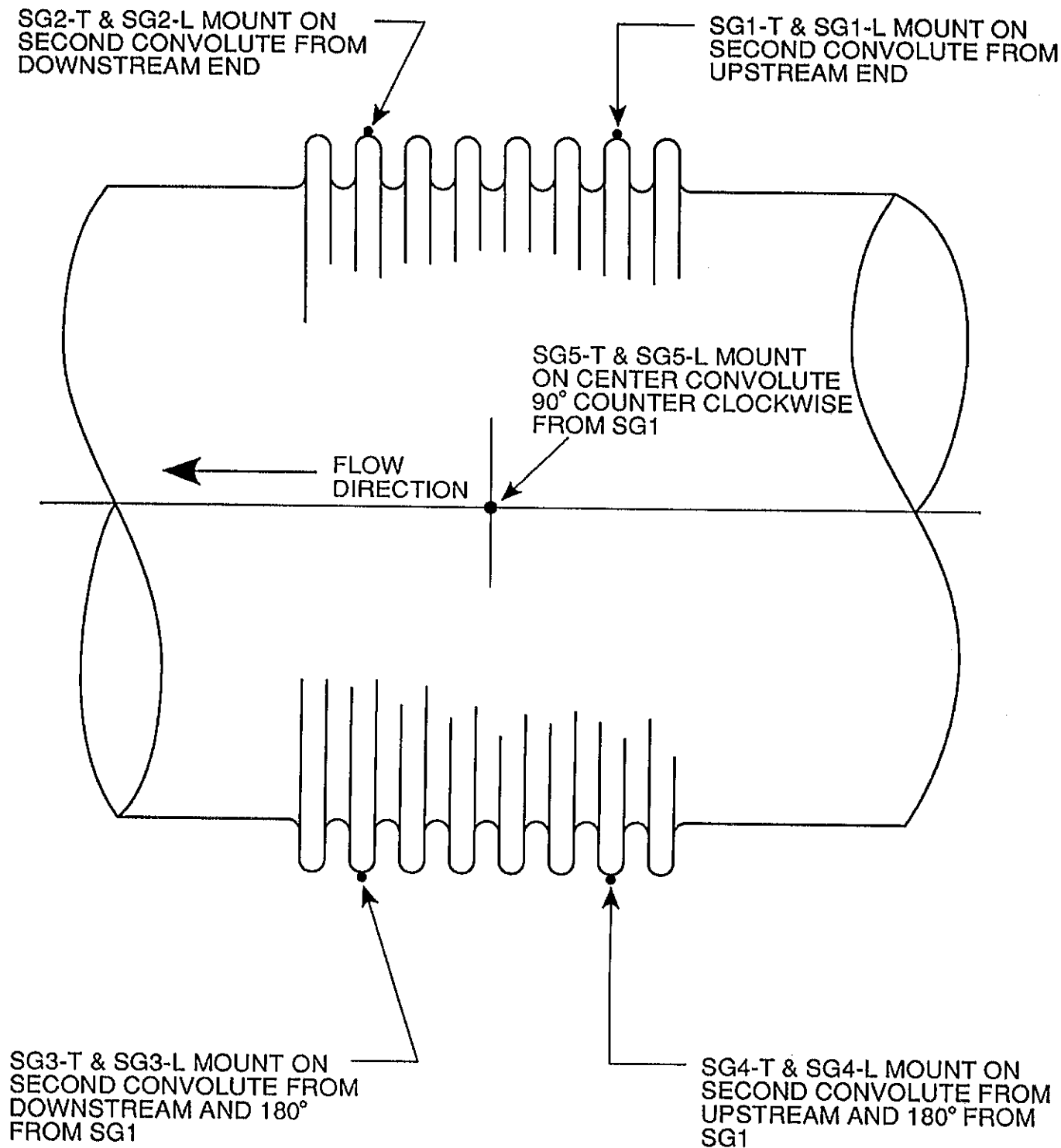
5.4 FAILURE MONITORING. A means to accurately determine the time to failure during the flow testing is required.

## 6.0 REPORTS

6.1 CONTENTS. Data shall include, but not be limited to, the following:

- a. Bellows/flex hose identification
- b. Measurement of as-built dimensions and a comparison to as-designed dimensions
- c. Measurement of actual spring rate
- d. Static conditions - pressure/deflection strains
- e. Dynamic conditions - resonance search data
  - test flow rates and pressures
  - time to failure or test duration
- f. Operational life requirements
- g. Sketch of test set-up or photograph
- h. Instrumentation utilized
- i. Temperature measurements

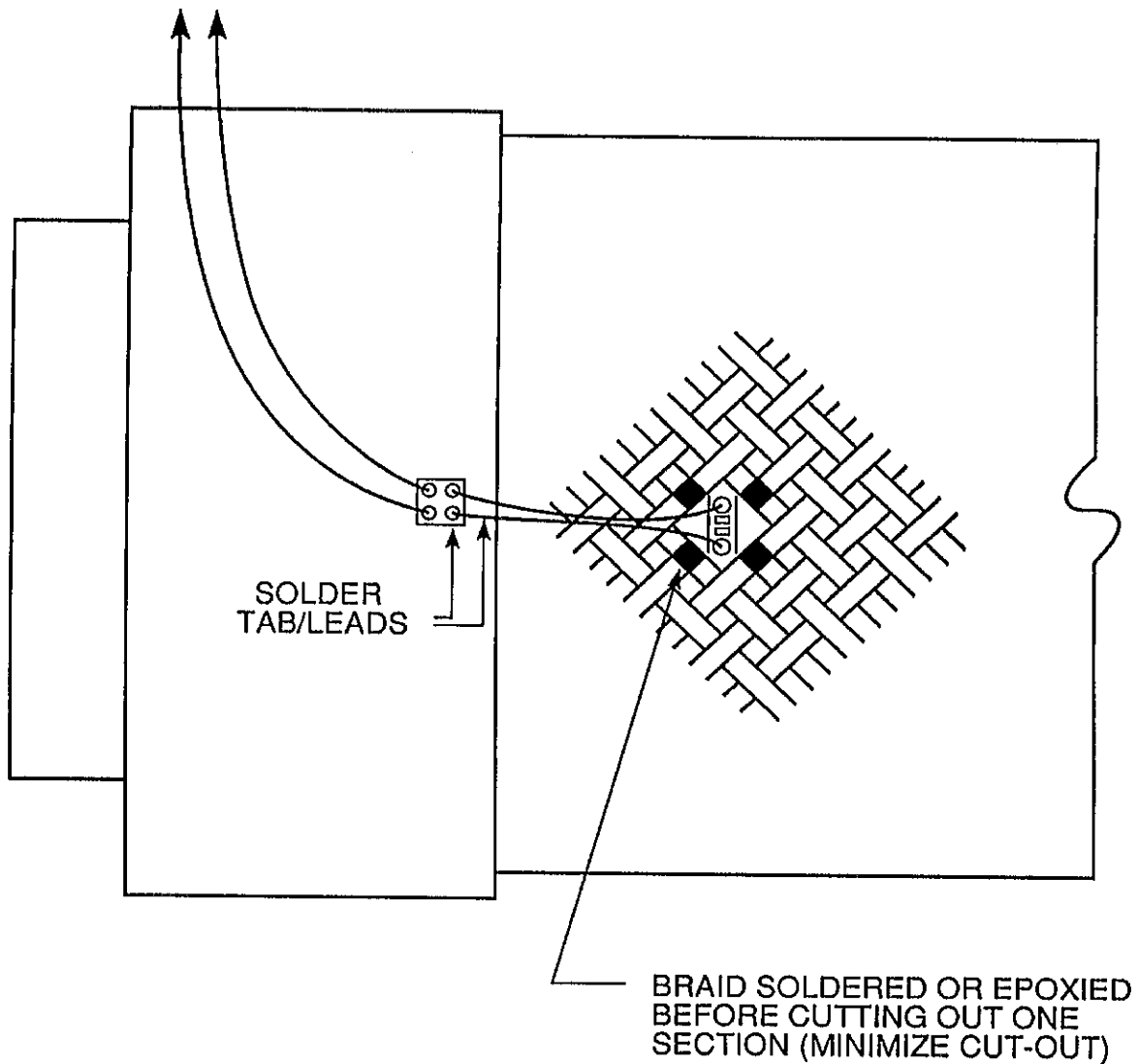


NOTES:

- T TANGENTIAL STRAIN GAGE
- L LONGITUDINAL STRAIN GAGE

**FIGURE 1. STRAIN GAGE LOCATIONS FOR A FREE BELLOWS**

TO RECORDING EQUIPMENT



**FIGURE 2. STRAIN GAGE INSTALLATION ON A FLEX HOSE**

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202 -

DR060PRO

PACKAGE NO. 10443R

DOCUMENTATION RELEASE LIST  
GEORGE C. MARSHALL SPACE FLIGHT CENTER

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PAGE 1

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C H	DOCUMENT NUMBER	DRL DRL DSH REV	TITLE	CCBD NO.	PCN	PC	EFFECTIVITY
*	MSFC-SPEC-626	202 -	TEST CONTROL DOCUMENT FOR ASSESSMENT OF FLEXIBLE LINES FOR FLOW INDUCED VIBRATION	SB3-00-0000	0000000	A	NONE

CHG NO.	CHG REV	CHG NOTICE	RESPONSIBLE ENGINEER	RESPONSIBLE ORGANIZATION	ACTION DATE	DESCRIPTION
			P. TYGIELSKI	EP64	04/13/94	BASELINE RELEASE
*	1	SCN000	EUGENA GOGGANS	EO03	02/22/07	DOCUMENT RELEASED THRU PDS. NO LONGER TRACKED IN ICMS.

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			MSFC-HDBK-1453		202	-			
			MSFC-HDBK-1674		202	-			
			MSFC-HDBK-2221		203	-			
			MSFC-HDBK-505		202	-			
			MSFC-HDBK-670		202	-			
			MSFC-MNL-1951		209	-			
			MSFC-PROC-1301		202	-			
			MSFC-PROC-1721		202	-			
			MSFC-PROC-1831		202	-			
			MSFC-PROC-1832		202	-			
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			MSFC-PROC-547		202	-			
			MSFC-QPL-1918		204	-			
			MSFC-RQMT-1282		202	-			
			MSFC-SPEC-1198		202	-			
			MSFC-SPEC-1238		202	-			
			MSFC-SPEC-1443		202	-			
			MSFC-SPEC-164		202	-			
			MSFC-SPEC-1870		202	-			
			MSFC-SPEC-1918		203	-			
			MSFC-SPEC-1919		206	-			
			MSFC-SPEC-2083		202	-			
			MSFC-SPEC-2223		202	-			
			MSFC-SPEC-2489		206	-			
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			MSFC-SPEC-2492		203	-			
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			MSFC-STD-2904		202	-			
			MSFC-STD-2905		202	-			
			MSFC-STD-2906		202	-			
			MSFC-STD-2907		202	-			
			MSFC-STD-366		202	-			
			MSFC-STD-383		202	-			
			MSFC-STD-486		202	-			
			MSFC-STD-506		203	-			
			MSFC-STD-531		202	-			
			MSFC-STD-557		202	-			
			MSFC-STD-561		203	-			
			MSFC-STD-781		202	-			

SUBMITTED BY ENGINEERING AREA:	BASIC	CHANGE	PARTIAL	COMPLETE	CLOSES	ACTION
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12/19/06

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## III. REPORTS, SPECIFICATIONS, ETC.

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