

"New/ear information is current as of the date of this document. For future coordination of changes in this document, draft circulation should be based on the information in the current 0001SS."

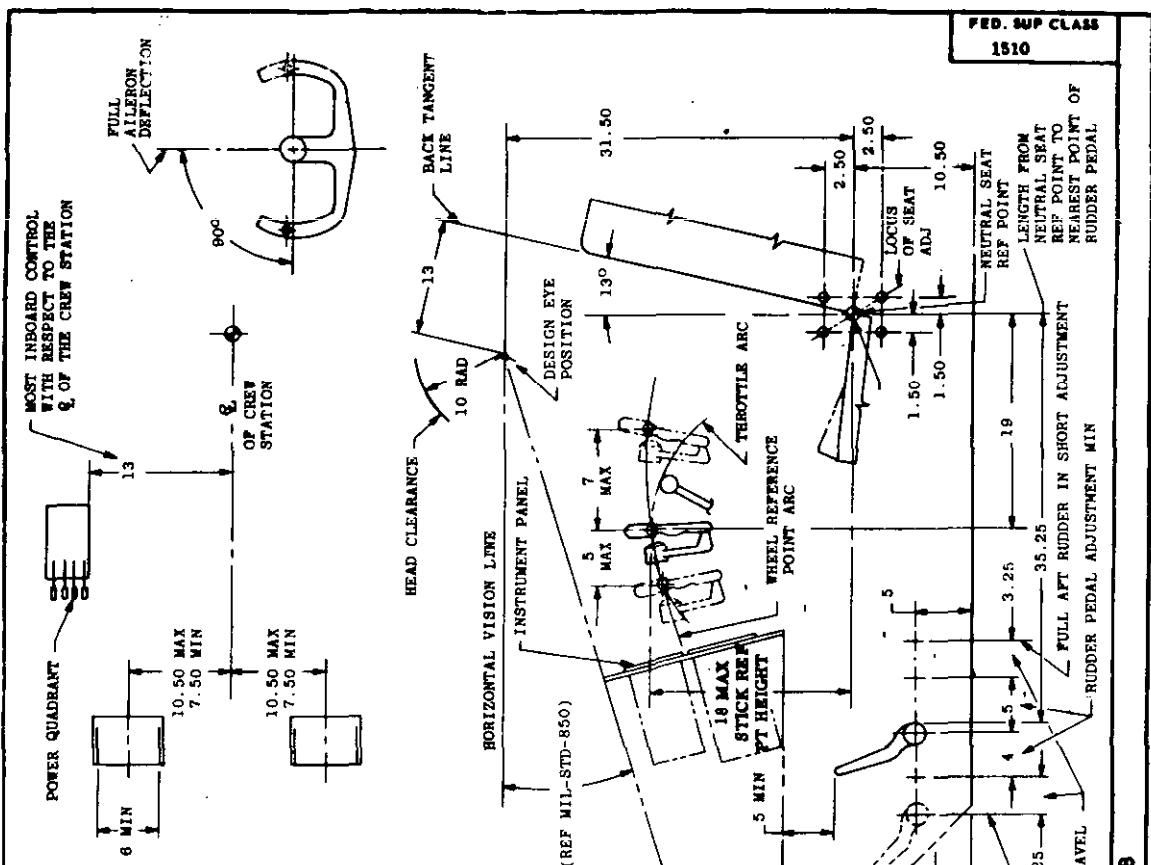
REVIEWER SYMBOLS: USER SYMBOLS:

This military standard is mandatory for use by all Departments & Agencies of the Department of Defense. Selection for all new engineering and design applications and for repetitive use shall be made from this document.

1. ALL MEASUREMENTS ARE BASED UPON THE SEAT REFERENCE POINT AT THE CENTERLINE OF THE SEAT IN THE NEUTRAL POSITION.
2. REFERENCE SPECIFICATION MIL-B-8584 FOR BRAKE PEDAL ANGLES AND DIMENSIONS.
3. THE WHEEL AND THROTTLE REFERENCE POINT IS DEFINED AS THE POINT AT WHICH THE PILOT'S SECOND FINGER IS IN CONTACT WITH THE FORWARD FACE OF THE CONTROL.
4. RUDDER PEDAL ADJUSTMENT SHALL BE IN INCREMENTS OF 1 INCH OR LESS.
5. BECAUSE OF CATAPULT GRIP REQUIREMENT, MAX. FORWARD THROTTLE MOVEMENT FOR AIRCRAFT EQUIPPED FOR CATAPULTING SHALL BE 20 INCHES FROM NEUTRAL SEAT REFERENCE POINT. MAX. FORWARD THROTTLE MOVEMENT FOR NONCATAPULT AIRCRAFT SHALL CORRESPOND TO THAT OF STICK COCKPITS.
7. DIMENSIONS IN INCHES.

THIS IS A DESIGN STANDARD, NOT USED AS A PART NUMBER.

**(B) CANCELED AFTER 27 JANUARY 1978.**  
 USE MIL-STD-1333 FOR ARMY AND NAVY,  
 AND DESIGN HANDBOOK DH2-2 FOR AIR FORCE.



FED. SUP CLASS  
1510

P.A. AIR FORCE - 11 Other Code ARMY - AV	TITLE DIMENSIONS, BASIC, COCKPIT, WHEEL CONTROLLED, FIXED WING AIRCRAFT	MILITARY STANDARD MS33576
PROCUREMENT SPECIFICATION NONE	SUPERSEDES:	SHEET 1 OF 1

APPROVED 27 AUG 56 REVISED - (A) 2 JUN 69 (B) 27 JAN 78