

INCH-POUND

MIL-DTL-7034E
5 May 1998
SUPERSEDING
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DETAIL SPECIFICATION

PULLEYS, GROOVE, ANTIFRICTION BEARING, GREASE LUBRICATED, AIRCRAFT, GENERAL SPECIFICATION FOR

This specification is approved for use by all Departments and Agencies of the Department of Defense.

1. SCOPE

1.1 Scope. This specification covers requirements for single groove pulleys with antifriction bearings.

1.2 Classification. Pulleys are classified as follows:

Type I - Nonmetallic sheave

Type II - Metallic sheave

2. APPLICABLE DOCUMENTS

2.1 General. The documents listed in this section are specified in sections 3 and 4 of this specification. This section does not include documents cited in other sections of this specification or recommended for additional information or as examples. While every effort has been made to ensure the completeness of this list, document users are cautioned that they must meet all specified requirements documents cited in sections 3 and 4 of this specification, whether or not they are listed.

Beneficial comments (recommendations, additions, deletions) and any pertinent data which may be of use in improving this document should be addressed to: Commander, Defense Supply Center, Columbus, DSCC-VAI, 3990 East Broad Street, Columbus, OH 43216-5000, by using the Standardization Document Improvement Proposal (DD Form 1426) appearing at the end of this document or by letter.

AMSC N/A

FSC 3020

DISTRIBUTION STATEMENT A. Approved for public release; distribution is unlimited.

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2.2 Government documents.

2.2.1 Specifications, standards, and handbooks. The following specifications, standards, and handbooks form a part of this document to the extent specified herein. Unless otherwise specified, the issues of these documents are those listed in the issue of the Department of Defense Index of Specifications and Standards (DoDISS) and supplement thereto, cited in the solicitation (see 6.2).

SPECIFICATIONS

FEDERAL

TT-P-1757 - Primer Coating, Alkyd Base, One Component

DEPARTMENT OF DEFENSE

MIL-DTL-7034/1 - Pulley, Groove, Secondary Control, Aircraft
 MIL-DTL-7034/2 - Pulley, Groove, Flight Control, Aircraft
 MIL-DTL-7034/3 - Pulley, Groove, Heavy Duty, Control, Aircraft
 MIL-DTL-7034/4 - Pulley, Control, Anti-friction Bearing
 MIL-B-7949 - Bearings, Ball, Airframe, Anti-friction
 MIL-A-8625 - Anodic Coatings for Aluminum and Aluminum Alloys
 MIL-G-23827 - Grease, Aircraft and Instrument, Gear and Actuator Screw, NATO Code Number G- 354, Metric
 MIL-G-81322 - Grease, Aircraft, General Purpose, Wide Temperature Range
 MIL-W-83420 - Wire Rope, Flexible, for Aircraft Control

STANDARDS

DEPARTMENT OF DEFENSE

MIL-STD-130 - Identification Marking of U.S. Military Property
 MIL-STD-810 - Environmental Test Methods and Engineering Guidelines
 MS21443 - Bearing, Ball, Airframe, Antifriction, Pulley
 MS27640 - Bearing, Ball, Airframe, Antifriction, Heavy Duty

(Unless otherwise indicated, copies of the above specifications and standards are available from the Standardization Document Order Desk, 700 Robbins Avenue, Building 4D, Philadelphia, PA 19111-5094.)

2.3 Non-Government publications. The following documents form a part of this document to the extent specified herein. Unless otherwise specified, the issues of the documents which are DoD adopted are those listed in the issue of the DoDISS cited in the solicitation. Unless otherwise specified, the issues of documents not listed in the DoDISS are the issues of the documents cited in the solicitation (see 6.2).

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AMERICAN SOCIETY OF MECHANICAL ENGINEERS (ASME)

- ASME B46.1 - Surface Texture (Surface Roughness, Waviness and Lay)
(DoD adopted)

(Application for copies of ASME publications should be addressed to American Society of Mechanical Engineers, 345 E. 47th Street, New York, New York 10017)

SOCIETY OF AUTOMOTIVE ENGINEERS (SAE)

- AMS-QQ-A-225/6 - Aluminum alloy 2024, Bar, Rod and Wire, Rolled, Drawn or Cold Finished
SAE AMS 4037 - Aluminum Alloy Sheet and Plate 4.4Cu -1.5Mg - 0.60Mn (2024; -T3 Flat Sheet, -T351 Plate)Solution Heat Treated

(Application for copies of SAE publications should be addressed to the Society of Automotive Engineers, 400 Commonwealth Drive, Warrendale, PA 15086)

AMERICAN NATIONAL STANDARDS INSTITUTE (ANSI)

- ANSI/ASQC-Z1.4 - Sampling Procedures and Tables for Inspection by Attributes (DoD adopted)
ANSI/EIA-557 - Statistical Process Control (DoD adopted)

(Application for copies of ANSI publications should be addressed to the American National Standards Institute, 11 West 42nd Street, New York, NY 10036.)

(Non-Government standards and other publications are normally available from the organizations that prepare or distribute the documents. These documents also may be available in or through libraries or other informational services.)

2.4 Order of precedence. In the event of a conflict between this document and the references cited herein (except for related associated specifications or specification sheets), the text of this document takes precedence. Nothing in this document, however, supersedes applicable laws and regulations unless a specific exemption has been obtained.

3. REQUIREMENTS

3.1 Specification sheets. The individual item requirements shall be as specified herein and in accordance with the applicable specification sheet. In the event of any conflict between the requirements of this specification and the specification sheet, the latter shall govern (see 6.2).

3.2 Qualification. Pulleys furnished under this specification shall be products which are authorized by the qualifying activity for listing on the applicable qualified products list before contract award (see 4.3 and 6.2).

3.3 Materials

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3.3.1 Sheave material. Sheave material requires qualifying activity approval.

3.3.1.1 Nonmetallic sheave, type I. Nonmetallic pulley sheaves shall be manufactured from fabric, or equivalent reinforcing material impregnated with a phenolic condensation substance, which will produce a finished product with uniform strength.

3.3.1.2 Metallic sheave, type II. Metallic pulley sheaves shall be manufactured from aluminum alloy conforming to AMS-QQ-A-225/6 temper T351, or AMS 4037 Temper T351.

3.3.2 Bearings. Bearings shall conform to MIL-B-7949 and MS21443 or MS27640. Only factory new ball bearings shall be used.

3.3.2.1 Finish, metallic sheaves, type II. Metallic pulley sheaves shall be anodized in accordance with MIL-A-8625 type II, before installing the bearing.

3.3.2.2 Aluminum bore surface, types I and II. Aluminum bore surfaces of types I and II sheaves shall be coated with zinc-chromate primer conforming to TT-P-1757 before installing the bearing. The bearing shall be installed while the primer is still wet.

3.4 Design.

3.4.1 Lubrication. The bearings shall be packed with grease conforming to MIL-G-81322 or MIL-G-23827 as specified in MIL-B-7949, so that at least 80% of the bearing's voids will be filled. Pulley bearings shall not be lubricated more than 24 months prior to the day of delivery.

3.4.2 Bearing closures. The bearing shall contain seals as specified in MIL-B-7949 and MS21443 or MS27640.

3.4.3 Non-afterglow, type I. When tested in accordance with 4.5.11, the pulley shall show no afterglow after ignition and flame extinction. The non-afterglow additive shall not bleed from the sheave material.

3.4.4 Temperature. When tested in accordance with 4.5.2, the pulley assembly shall show no cracks, splits, or warpage.

3.4.5 Corrosion. When tested in accordance with 4.5.3, the type I sheave material shall show no evidence that it will corrode tin or zinc coated carbon steel cable.

3.4.6 Fungus. When tested in accordance with 4.5.4, the type I sheave material shall not be adversely affected by fungus growth.

3.4.7 Physical characteristics. Physical characteristics shall be as specified on MIL-DTL-7034/1, MIL-DTL-7034/2, MIL-DTL-7034/3, MIL-DTL-7034/4.

3.4.8 Surface roughness. The surface roughness shall not exceed a 160 roughness height rating for metallic parts and a 500 roughness height rating for phenolic parts, in accordance with ASME B46.1.

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3.5 Performance.

3.5.1 Eccentricity. When tested as specified in 4.5.5, eccentricity shall not exceed the values specified. The eccentricity of the periphery of the groove and groove flanges with respect to the longitudinal axis of the bearing shall be defined as the difference between the maximum and minimum radii. Eccentricity measurements shall be taken in such a manner that the bearing clearance is not included as eccentricity.

3.5.2 Wobble clearance. When the pulley is tested in accordance with 4.5.7, the clearance between the pulley and a plane which is perpendicular to the bearing axis, and which contacts the end of the bearing inner race, shall be not less than the amount specified.

3.5.3 Strength under static loads.

3.5.3.1 Bonding strength. When tested as specified in 4.5.8, the pulleys shall be capable of withstanding the bonding strength proof loads without failure of the bond between the pulley sheave and bearing. Loosening, separating, or shifting of the bearing, shall be evidence of bond failure.

3.5.3.2 Flange strength. When tested as specified in 4.5.9, the pulleys shall be capable of withstanding the specified flange strength loads without failure. Checking or shearing of the flange, shall be evidence of failure.

3.5.3.3 Sheave strength. When tested as specified in 4.5.10, the pulley shall be capable of withstanding the sheave strength loads specified without buckling, splitting, or other types of failure of the sheave.

3.5.4 Endurance. When tested in accordance with 4.5.6, types I and II pulleys shall be capable of enduring the conditions specified without reducing the groove diameter by an amount greater than the wear factor specified.

3.5.5 Pulley structure. The pulley structure shall not have spokes, or holes in the sides.

3.6 Identification.

3.6.1 Pulley. The part number shall be legibly and permanently marked on the pulley in accordance with MIL-STD-130. Type I nonmetallic pulleys shall also be marked with the sheave material lot designation, adequate for traceability. Metal stamping is prohibited. The identification shall be:

Manufacturer's name or trademark
Specification sheet part number

3.6.2 Bearing. The bearing manufacturer's name and the type designation of the bearing shall be legibly and permanently marked on each bearing so it is discernible when installed in the pulley sheave.

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3.7 Workmanship. Pulleys shall be free from cracks, splits, burrs, machine tool marks, and other defects which would affect their performance.

4. VERIFICATION

4.1 Classification of inspection. The inspection requirements specified herein are classified as follows:

- a. Qualification inspection (see 4.3).
- b. Conformance inspection (see 4.4).

4.2 Inspection and tests. The inspection and tests specified herein are intended to verify that the items produced meet or exceed the performance requirements specified.

4.3 Qualification inspection.

4.3.1 Samples. Twenty-five samples of each size of types I and II pulleys offered for qualification shall be submitted to the qualifying activity (see 6.3). Complete information on dimensions, construction, materials, and heat treatment, shall be furnished for each size and type pulley submitted.

4.3.1.1 Number of samples subjected to qualification tests. The number of samples to be subjected to the qualification tests, and the order in which the tests are to be performed, shall be in accordance with table I. To optimize the use of test samples, non-destructive testing shall precede destructive testing.

TABLE I. Number of samples subjected to qualification tests.

Inspection	Test paragraph	Number of samples subjected to qualification tests
Examination of product	4.5.1	25
Eccentricity	4.5.5	10
Wobble clearance and bearing strength	4.5.7	10
Endurance	4.5.6	5
Bonding strength	4.5.8	5
Flange strength	4.5.9	5
Sheave strength	4.5.10	5
Temperature	4.5.2	5
Corrosion	4.5.3	5
Fungus growth	4.5.4	5
Non-afterglow, type I	4.5.11	3

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TABLE II. Qualification inspection and conformance inspection.

Inspection	Requirement paragraph	Qualification inspection paragraph	Conformance inspection paragraph
Examination of product	3.6 and 3.7	4.5.1	4.5.1
Eccentricity	3.5.1	4.5.5	-
Wobble clearance and bearing strength	3.5.2	4.5.7	-
Endurance	3.5.4	4.5.6	-
Bonding strength	3.5.3.1	4.5.8	4.5.8
Flange strength	3.5.3.2	4.5.9	4.5.9
Sheave strength	3.5.3.3	4.5.10	4.5.10
Temperature	3.4.4	4.5.2	-
Corrosion	3.4.5	4.5.3	-
Fungus growth	3.4.6	4.5.4	-
Non-afterglow, type I	3.4.3	4.5.11	4.5.11

4.3.2 Inspection. The qualification inspections shall consist of all the tests listed in table II in the order shown, on the samples submitted in accordance with 4.3.1.

4.4 Conformance inspection. Conformance inspection for types I and II pulleys shall consist of an individual test and a sampling plan.

4.4.1 Individual test. The individual test shall consist of the following:

Examination of product (4.5.1)

4.4.2 Sampling plan.

4.4.2.1 Samples. Sample pulleys shall be selected in accordance with ANSI/ASQC-Z1.4, inspection level S-3, acceptance number zero, for all tests specified in 4.4.3, except examination of product. The number of allowable defects shall be as specified in the contract.

4.4.2.2 Lot. The lot definition, formation, and size shall be in accordance with ANSI/ASQC-Z1.4.

4.4.3 Tests performed on samples. The following tests shall be performed on the sample pulleys:

Bonding strength (4.5.8)
 Flange strength (4.5.9)
 Sheave strength (4.5.10)
 Non-afterglow - type I (4.5.11)

4.5 Test methods.

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4.5.1 Examination of product. Each sample pulley shall be carefully examined to determine conformance to this specification and the applicable specification sheet with respect to dimensions, tolerance, weight, finish, identification of product, and workmanship.

4.5.2 Temperature. The pulley assembly shall be subjected to a temperature of -65 °F for 48 hours and to a temperature of 250 °F for 48 hours. Any pulley assembly showing cracks, splits, or warpage as a result of this test shall be cause for rejection.

4.5.3 Corrosion. Type I nonmetallic pulleys shall be exposed to 95% relative humidity at 100 °F for 48 hours. The pulleys shall be removed from the humidity chamber and the pulley surfaces tested for the presence of efflorescent chloride salts by placing a drop of 20% nitric acid and a drop of 3.0% silver nitrate solution on the same spot of the humidified pulley. Formation of insoluble white silver chloride salts is indication that the sheave material is corrosive to tin or zinc coated carbon steel cable, and shall constitute failure of the pulley.

4.5.4 Fungus growth. Type I nonmetallic pulleys shall be tested in accordance with MIL-STD-810, Method 508.1. Samples shall be considered to have failed if they support fungus in an amount that prevents or impairs flange strength or sheave strength testing.

4.5.5 Eccentricity. The inner bearing race shall be securely held in a fixture as shown in figure 1. When the pulley is rotated through one revolution, the eccentricity tolerance measured shall not exceed the values specified in table III. Eccentricity measurements shall be taken in such a manner that the bearing clearance is not included as eccentricity.

4.5.6 Endurance. The general arrangement of the testing machine shall be as shown in figure 2. The wire rope employed shall conform to MIL-W-83420, and shall be of the size specified in table IV for the pulley being tested. The machine shall be operated under the applicable conditions of table IV. Atmospheric temperature shall be not less than 50 °F. Cable temperature shall not exceed 212 °F.

4.5.6.1 Failure under endurance test. A pulley shall be considered to have failed when any of the following conditions develop, either during the test or at the conclusion of the test:

- a. The root diameter is reduced more than 1/32-inch as specified in table IV.
- b. The pulley fails when subjected to the wobble clearance and bearing strength test as specified in 4.5.7.
- c. The pulley fails when subjected to the bonding strength test as specified in 4.5.8.

4.5.7 Wobble clearance and bearing strength. The inner bearing race shall be secured to a plane surface and the limit load applied with 180° cable wrap and 2° pulloff as shown on figure 3. The minimum clearance between the plane surface and the pulley flange, measured at the point of load application, shall be determined by placing the load at various positions about the complete circumference of the pulley as shown on table V. The test shall be repeated with the pulley inverted.

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4.5.8 Bonding strength. The pulley shall be supported in a jig as shown on figure 4. The bearing shall be unsupported and free to be pushed out of the sheave in case of failure. The load specified in table VI for the particular size pulley under test shall be applied gradually to the inner race of the bearing perpendicular to the plane of the pulley. The maximum load shall be maintained for not less than 10 seconds. The test shall be repeated with the pulley inverted.

4.5.9 Flange strength. The pulley shall be clamped in a jig as shown on figure 5, in such a manner as to hold the sheave rigidly. The diameter of the gage disk shall be as specified on figure 5. The load specified in table VII for the particular size pulley under test shall be applied to the inside of the flange perpendicular to the plane of the pulley, through the loading block in line with the root diameter. Both flanges shall be tested with the load being applied 90° apart. The maximum load shall be maintained for not less than 10 seconds.

4.5.10 Sheave strength. The pulley shall be mounted in the jig as shown on figure 6 in such a manner as to permit rotation. The load specified in table VIII for the particular size pulley under test shall be applied gradually by means of a pressure plate of the size specified in table VIII. The maximum load shall be maintained for not less than 10 seconds. During the test, the pulley shall receive no side support from the jig. Failure of bearing under this test shall not be cause for rejection of the pulley.

4.5.11 Non-afterglow, type I. Type I nonmetallic pulleys shall be clamped in a support, and a Bunsen burner or an alcohol lamp with blue flame 0.5 to 0.75-inch high, shall be placed under the sheave so the flame tip contacts the edge of one flange. After 30 seconds, the flame shall be removed. After removal of the flame, sufficient airblast to extinguish the blaze in from 1 to 2 seconds shall be applied. Eight to 10 seconds later the airblast shall again be applied. There shall be no afterglow or bleeding of the non-afterglow additive. The test shall be repeated on the same area. There shall be no afterglow or bleeding of the non-afterglow additive.

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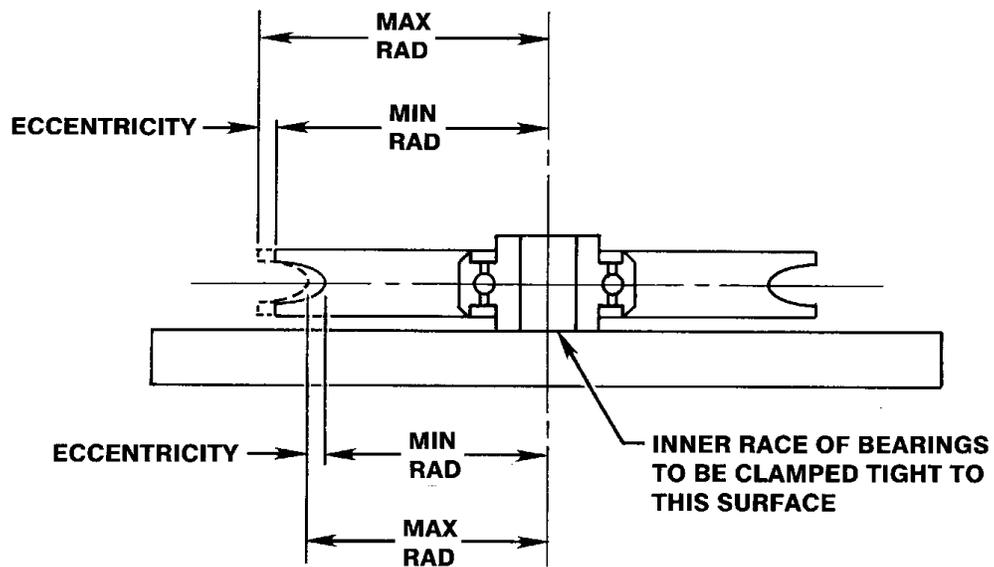


FIGURE 1. Eccentricity test setup.

TABLE III. Eccentricity tolerances.

Part number	Eccentricity Max (inches)
MS20219-1, -A1	.006
-2, -A2	.006
-3, -A3	.006
-4, -A4	.007
-5, -A5	.007
MS20220-1, -A1	.007
-2, -A2	.007
-3, -A3	.010
-4, -A4	.010
MS20221 -A1	.007
-2, -A2	.007
-3, -A3	.010
MS24566-1B	.006
-2B	.006
-3B	.006
-4B	.010
-5B	.010
-6B	.010

Part number MS20221-1 Deleted

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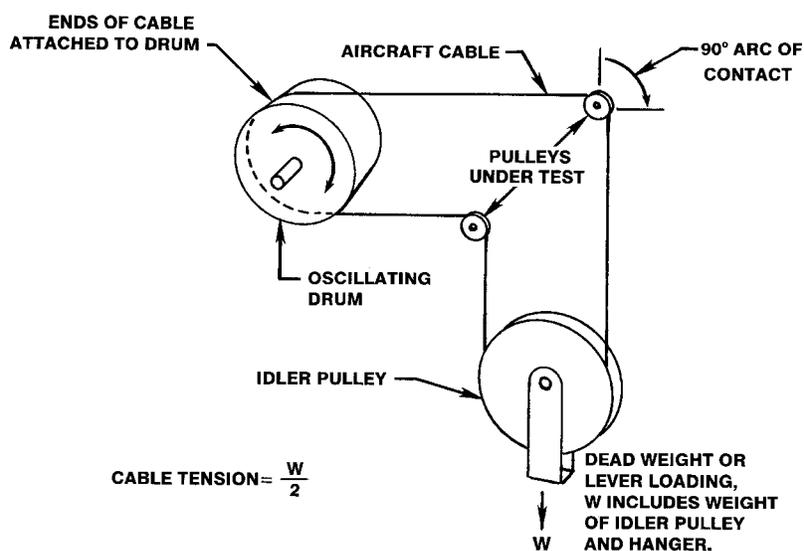


FIGURE 2. Endurance test setup.

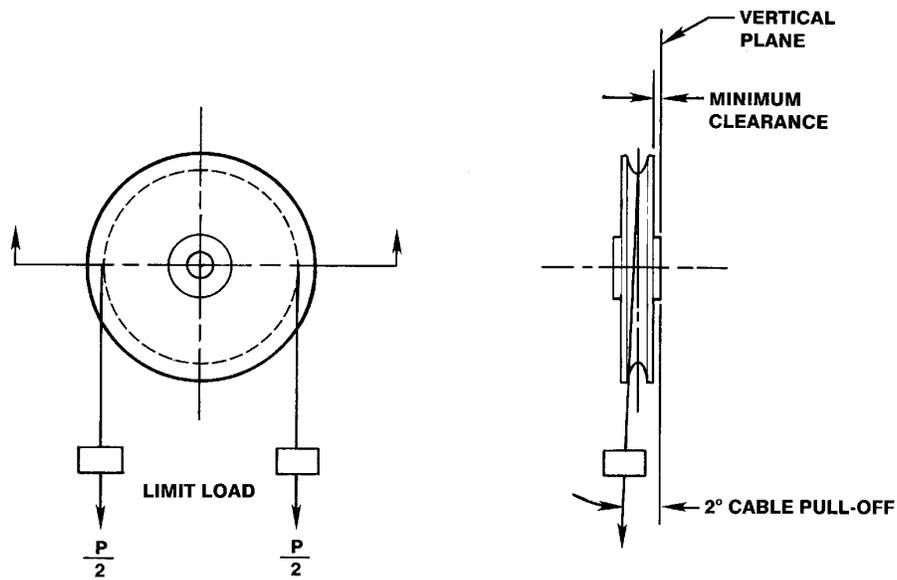
TABLE IV. Endurance test tolerances.

Part number	Cable dia (inch)	Cable travel (inch)	Cable tension (lbs)	Total reversals ^{1/}	Cable wrap (°)	Wear max (inch)
MS20219-1, -A1	3/32	20	120	7,850	90	1/32
-2, -A2	3/32	20	120	11,250	90	1/32
-3, -A3	3/32	20	120	11,250	90	1/32
-4, -A4	3/32	20	230	18,100	90	1/32
-5, -A5	3/32	20	230	18,100	90	1/32
MS20220-1, -A1	3/16	20	125	9,800	90	1/32
-2, -A2	3/16	20	420	20,000	90	1/32
-3, -A3	3/16	20	625	29,200	90	1/32
-4, -A4	3/16	20	625	40,000	90	1/32
MS20221 -A1	1/4	20	700	15,650	90	1/32
-2, -A2	1/4	20	1225	27,400	90	1/32
-3, -A3	1/4	20	1750	40,000	90	1/32
MS24566-1B	3/32	20	75	7,650	90	1/32
-2B	3/32	20	125	17,400	90	1/32
-3B	3/16	20	150	11,850	90	1/32
-4B	3/16	20	300	23,600	90	1/32
-5B	1/4	20	750	34,300	90	1/32
-6B	1/4	20	1000	42,100	90	1/32

Part number MS20221-1 Deleted

^{1/} The number of reversals indicated is equivalent to 50,000 revolutions.

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FIGURE 3. Wobble clearance test setup.TABLE V. Wobble clearance tolerance.

Part number	P pulley limit load (lbs)	Wobble clearance (inch)
MS20219-1, -A1	480	.005
-2, -A2	480	.005
-3, -A3	480	.005
-4, -A4	920	.005
-5, -A5	920	.005
MS20220-1, -A1	500	.005
-2, -A2	1680	.005
-3, -A3	2500	.005
-4, -A4	2500	.005
MS20221 -A1	2800	.005
-2, -A2	4900	.005
-3, -A3	7000	.005
MS24566-1B	300	.002
-2B	500	.002
-3B	600	.002
-4B	1200	.002
-5B	3000	.002
-6B	4000	.002

Part number MS20221-1 Deleted

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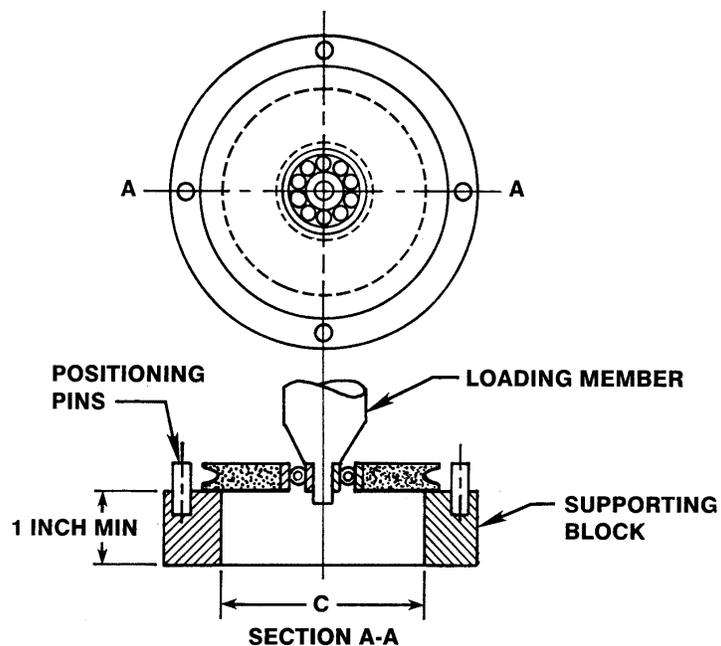


FIGURE 4. Bonding strength test setup.

TABLE VI. Bonding strength test tolerances.

Part number	C Groove dia (inches)	Bonding strength proof load (lbs) min
MS20219-1, -A1	1.000	300
-2, -A2	1.437	300
-3, -A3	1.437	300
-4, -A4	2.312	300
-5, -A5	2.312	300
MS20220-1, -A1	1.250	300
-2, -A2	2.500	500
-3, -A3	3.750	500
-4, -A4	5.000	500
MS20221 -A1	2.000	600
-2, -A2	3.500	800
-3, -A3	5.000	800
MS24566-1B	.972	200
-2B	2.222	200
-3B	1.510	280
-4B	3.010	280
-5B	4.374	800
-6B	5.374	800

Part number MS20221-1 Deleted

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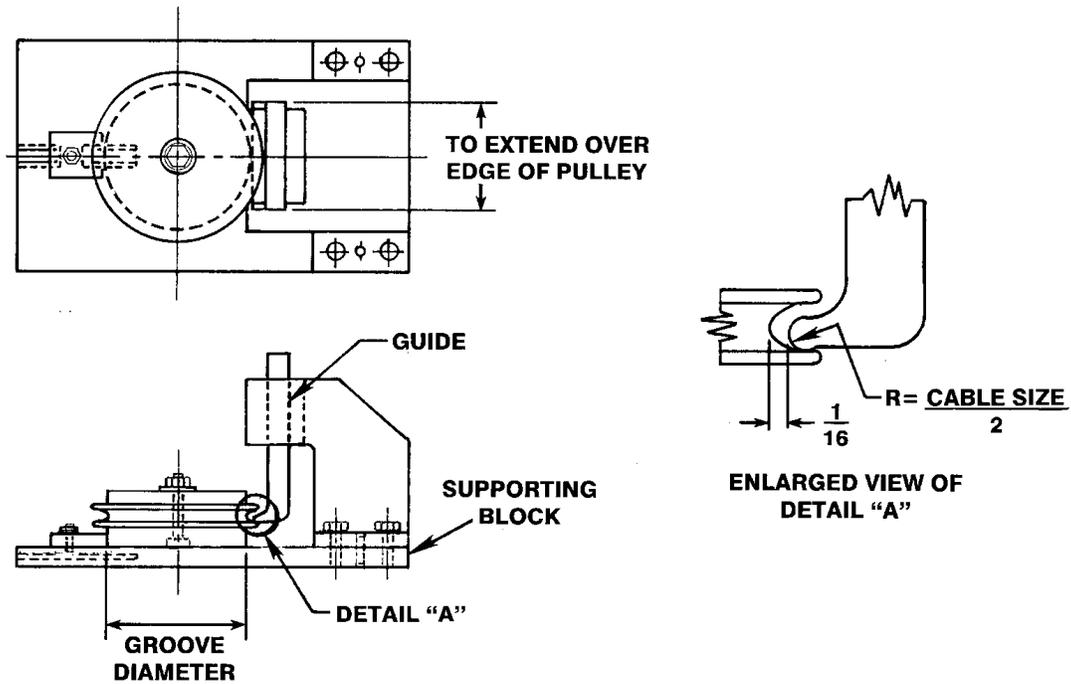


FIGURE 5. Flange strength test setup.

TABLE VII. Flange strength test tolerances.

Part number	Flange strength load (lbs) min
MS20219 -1, -A1	75
-2, -A2	75
-3, -A3	75
-4, -A4	75
-5, -A5	75
MS20220 -1, -A1	125
-2, -A2	175
-3, -A3	175
-4, -A4	175
MS20221 -A1	250
-2, -A2	250
-3, -A3	250
MS24566 -1B	50
-2B	50
-3B	175
-4B	175
-5B	250
-6B	250

Part number MS20221-1 Deleted

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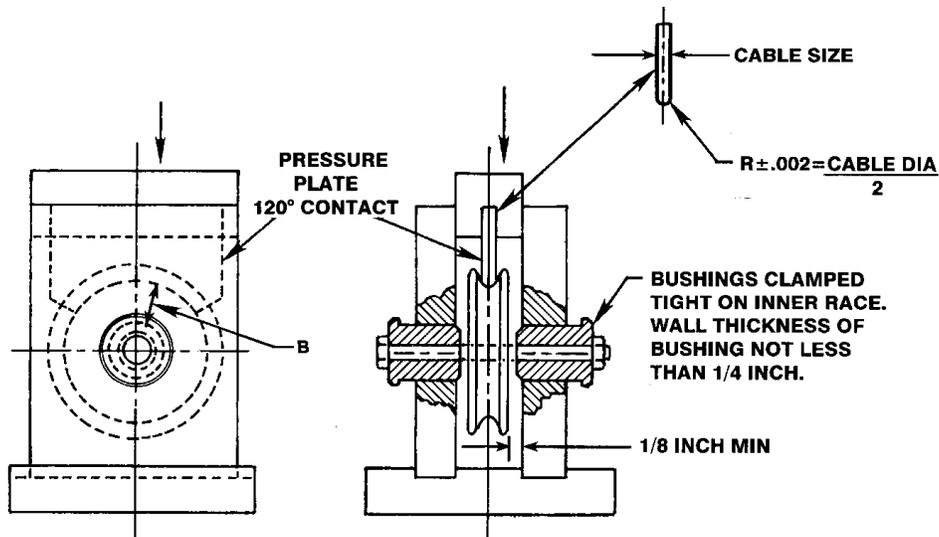


FIGURE 6. Sheave strength test setup.

TABLE VIII. Sheave strength test tolerances.

Part number	B Groove rad ±.004	Cable dia (inch)	Sheave strength load (lbs) min	Pressure plate wrap angle (°)
MS20219-1, -A1	.500	3/32	960	120
-2, -A2	.724	3/32	960	120
-3, -A3	.724	3/32	960	120
-4, -A4	1.161	3/32	1840	120
-5, -A5	1.161	3/32	1840	120
MS20220 -1, -A1	.630	3/16	1000	120
-2, -A2	1.255	3/16	3360	120
-3, -A3	1.880	3/16	5000	120
-4, -A4	2.505	3/16	5000	120
MS20221 -A1	1.005	1/4	5600	120
-2, -A2	1.755	1/4	9800	120
-3, -A3	2.505	1/4	14000	120
MS24566 -1B	.486	3/32	600	120
-2B	1.111	3/32	1000	120
-3B	.755	3/16	1200	120
-4B	1.505	3/16	2400	120
-5B	2.187	1/4	6000	120
-6B	2.687	1/4	8000	120

Part number MS20221-1 Deleted

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5. PACKAGING

5.1 Packaging. For acquisition purposes, the packaging requirements shall be as specified in the contract or order (see 6.2). When actual packaging of material is to be performed by DoD personnel, these personnel need to contact the responsible packaging activity to ascertain requisite packaging requirements. Packaging requirements are maintained by the Inventory Control Point's packaging activity within the Military Department or Defense Agency, or within the Military Department's System Command. Packaging data retrieval is available from the managing Military Department's or Defense Agency's automated packaging files, CD-ROM products, or by contacting the responsible packaging activity.

6. NOTES

(This section contains information of a general or explanatory nature which may be helpful, but is not mandatory.)

6.1 Intended use. These pulleys are used in aircraft applications, such as, control systems and brake installations. These military unique parts of aircraft control and brake systems require interoperability and compatibility with associated components and equipment. This interoperability and compatibility has been assured through the strict adherence to military requirements which contain substantial detail. The detail assures item standardization and compatibility through specific dimensions and tolerances in the "MIL-DTL" 7034 specification sheets and material control requirements in this specification. The qualification process ensures these items will meet the flame resistance, and -65 °F to 250 °F, temperature requirements, and perform as specified in the environments to which they may be subjected. Users and prime equipment manufacturers depend on these detailed technical requirements to ensure delivered product meets the interoperability and compatibility requirements.

6.2 Acquisition requirements. Acquisition documents must specify the following:

- a. Title, number, and date of this specification and applicable specification sheet.
- b. Issue of DoDISS to be cited in the solicitation, and if required, the specific issue of individual documents referenced (see 2.2.1 and 2.3).
- c. Type pulley and quantities required.
- d. Number of allowable defects (see 4.4.2.1)
- e. Packaging requirements (see 5.1).

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6.3 Qualification. With respect to products requiring qualification, awards will be made only for products which are, at the time of award of contract, qualified for inclusion in Qualified Products List QPL-7034, whether or not such products have actually been so listed by that date. The attention of the contractors is called to these requirements, and manufacturers are urged to arrange to have the products that they propose to offer to the Federal Government tested for qualification in order that they may be eligible to be awarded contracts or orders for the products covered by this specification. Information pertaining to qualification of products may be obtained from the Commander, Defense Supply Center, Columbus, DSCC-VAI, 3990 East Broad Street, Columbus, OH 43216-5000.

6.3.1 Retention of qualification. To retain qualification, the manufacturer shall forward certification at two year intervals to the qualifying activity stating that the company still has the capabilities and facilities necessary to produce the item and that the product has not been changed in any way. The qualifying activity will establish the initial reporting date.

6.4 MS cross reference. Former military standard (MS) number cross reference to MIL-DTL-7034 specification sheet number:

MIL-DTL-7034/1	MS20219
MIL-DTL-7034/2	MS20220
MIL-DTL-7034/3	MS20221
MIL-DTL-7034/4	MS24566

The military standard (MS) part numbers have been retained in the specification sheets.

6.5 Subject term (key word) listing.

Eccentricity
 Non-afterglow
 Phenolic
 Sheave
 Wobble clearance

6.6 Changes from the previous issue. Marginal notations are not used in this revision to identify changes with respect to the previous issue due to the extent of the changes.

CONCLUDING MATERIAL

Custodians:
 Air Force - 99
 Army - AV
 Navy - AS

Preparing activity:
 DLA - CC
 (Project 3020-0001)

Review activities:
 Air Force-82

STANDARDIZATION DOCUMENT IMPROVEMENT PROPOSAL

INSTRUCTIONS

1. The preparing activity must complete blocks 1, 2, 3, and 8. In block 1, both the document number and revision letter should be given.
2. The submitter of this form must complete blocks 4, 5, 6, and 7.
3. The preparing activity must provide a reply within 30 days from receipt of the form.
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I RECOMMEND A CHANGE:

1. DOCUMENT NUMBER
MIL-DTL-7034E

2. DOCUMENT DATE (YYMMDD)

3. DOCUMENT TITLE

PULLEYS, GROOVE, ANTIFRICTION BEARING, GREASE LUBRICATED, AIRCRAFT, GENERAL SPECIFICATION FOR

4. NATURE OF CHANGE (*Identify paragraph number and include proposed rewrite, if possible. Attach extra sheets as needed.*)

5. REASON FOR RECOMMENDATION

6. SUBMITTER

a. NAME (*Last, First, Middle Initial*)

b. ORGANIZATION

c. ADDRESS (*Include Zip Code*)

d. TELEPHONE
(*Include Area Code*)
(1) Commercial:

(2) DSN:
(*If Applicable*)

7. DATE SUBMITTED
(YYMMDD)

8. PREPARING ACTIVITY

a. NAME
DLA-CC
COMMANDER
DEFENSE SUPPLY CENTER, COLUMBUS

b. TELEPHONE NUMBER (*Include Area Code*)
(1) Commercial (614) 692-0538 FAX 614-692-6939 (2) DSN 850-6939

c. ADDRESS (*Include Zip Code*)
DSCC-VAI
3990 EAST BROAD STREET
COLUMBUS, OH 43216-5000

IF YOU DO NOT RECEIVE A REPLY WITHIN 45 DAYS, CONTACT:
Defense Quality and Standardization Office, 5203 Leesburg Pike,
Suite 1403, Falls Church, VA 22041-3466
Telephone (703) 756-2340 DSN 289-2340