11 July 1966

Superseding MIL-A-8806(ASG) 25 October 1956

## MILITARY SPECIFICATION

# ACOUSTICAL NOISE LEVEL IN AIRCRAFT, GENERAL SPECIFICATION FOR

This specification has been approved by the Department of Defense and is mandatory for use by the Departments of the Army, the Navy, and the Air Force.

## 1. SCOPE

1.1 This specification covers the general requirements for the control of accustical noise in occupied spaces of aircraft, including the acceptable noise levels and the testing requirements for determining conformance to these levels.

## 2. APPLICABLE DOCUMENTS

2.1 The following documents of the issue in effect on date of invitation for bids or request for proposal, form a part of this specification to the extent specified herein.

### **SPECIFICATIONS**

### Military

MIL-S-3151 Sound-Level-Measuring Equipment

MIL-S-6144 Soundproofing for Aircraft; General Specification for Installation of

MIL-I-7171 Insulation Blanket, Thermal-Acoustical

(When requesting specifications, refer to both title and symbol. Copies of specifications may be obtained upon application to the Commanding Officer, Naval Supply Depot, 5801 Tabor Avenue, Philadelphia, Pennsylvania, 19120, Attention: Code 105).

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#### REQUIREMENTS 3.

#### Acoustical noise levels --3. I

Maximum continuous power - The acoustical noise level in 3.1.1 any part of the aircraft (see 6. 2. 2) intended for occupancy by the crew or other personnel shall mot exceed the values specified in Table IA (preferred) or Table 1B during conditions of MAXIMUM CONTINUOUS POWER.

TABLE L - Maximum acceptable noise level at maximum continuous power

I	A		
Frequency (cps)  Band   Cente	Max. accept- able noise level (db)	Frequency bands (cps)	Max. accept- able noise level (db)
Overall  22.4 - 45 31  45 - 90 63  90 - 180 125  180 - 355 250  355 - 710 500  710 - 1400 1000  1400 - 2800 2000  2800 - 5600 4000  5600 - 11200 8000	113	Overall  37.5 - 75 75 - 150 150 - 300 300 - 600 600 - 1200 1200 - 2400 2400 - 4800 4800 - 9600	113 111 111 111 105 99 93 87 87

Short duration conditions - For takeoff, afterburner operation 3, 1, 2 and other conditions normally not exceeding 5 minutes continuous duration the acoustical noise level in any part of the aircraft (see 6.2.2) intended for occupancy by the crew or other personnel shall not exceed the values specified in Table IIA (preferred) or Table II B.

TABLE II. - Maximum acceptable noise level under short duration conditions

TABLE II Maximum acceptable noise leve					l under short duration conditions		
II A,					AA 43		
Frequency (cps)				Max. accept- able noise level (db)	Frequency bands (cps)	Max. accept- able noise level (db)	
Bar	erall			120	Overall	120	
22.4	-	45 90	31.5 63	118 118	37.5 - 75	118	
45 90	_	180	125	118	75 - 150 150 - 300	118 118	
180 355	-	355 710	· 250 500	118 112	300 - 600	112 106	
710	-	1400	1000	106 100	600 - 1200 1200 - 2400	100	
1400 2800	-	2800 5600	2000 4000	94	2400 - 4800	94 94	
5800	-	11200	1	94	4800 - 9600	34	

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3.1.3. Protective helmets - In aircraft in which personnel must necessarily wear helmets at all times and communicate by electronic means (e.g., single place fighter aircraft), the acoustical noise level (see 6.2.2) shall not exceed the values specified in Table III A (preferred) or Table III B during conditions of MAXIMUM CONTINUOUS POWER.

TABLE III. - Maximum acceptable noise level with protective helmets or devices

III A.					ддв.		
Frequency (cps)		Max. accept- able noise	Frequency, bands	Max. accept- able ncise			
B	Band		Center	level (Pb)	(cps)	. leael (qp)	
Ov	eral	1		113	Overall	113	
22.4	_	45	31.5	111			
45	_	90	63	111	37.5 - 75	111	
90	_	180	125	111	75 ~ 150	111	
180	_	355	250	111	150 - 300	111	
355	_	710	500	109	300 - 600	109	
710	_	1400	1000	106	600 - 1200	106	
1400	_	2800	2000	100	1200 - 2400	100	
2800	-	5600	4000	94	2400 - 4800	94	
5600	-	11200	8000	94	4800 - 9600	94	

3. 1.4 Normal cruise power - The acoustical noise level in any part of the aircraft (see 6. 2. 2) intended for occupancy by the crew or other personnel shall not exceed the values specified in Table IV A (preferred) or Table IV B, during conditions of NORMAL CRUISE POWER. Tables IV A and IV B are applicable to all Naval aircraft procurement; and to Air Force and Army aircraft procurement when so stated in the aircraft detail specification.

TABLE IV. - Maximum acceptable noise level at normal cruise power

IV A.					IV B.		
Frequency (cps)  Band Center		Max. accept- able noise level (db)	Frequency bands (cps)	Max. accept- able noise level (db)			
Ove	erall	<u> </u>		106	Overall	106	
22.4	-	45	31.5	104			
45	_	90	63	104	37.5 - 75	104	
90	-	180	125	104	75 - 150	104	
180	_	355	250	104	150 - 300	104	
355	_	710	500	96	300 - 600	96	
.710	_	1400	1000	90	600 - 1200	90	
1400	_	2800	2000	86	1200 - 2400	86	
2800	_	5600	4000	75	2400 - 4800	75	
5600	-	11200	8000	75	4800 - 9600	75	

- 3.1.5 <u>Auxiliary systems</u> The auxiliary systems which normally operate for longer than 5 minutes shall not produce an increase in noise levels in occupied compartments above the tables herein. Short duration noise levels shall not exceed levels in Table IIA or Table IIB unless specifically approved.
- 3.1.6 Special missions For special missions such as Anti-Submarine Warfare (ASW), Aircraft Early Warning (AEW), and Electronic Counter Measures (ECM) which may require noise levels lower than those required by this specification, the requirements will be so stated in the detail specification.

## 3.2 Noise control methods -

- 3.2.1 During the design and development stages of the aircraft, the contractor shall include in the design and, where pertinent, demonstrate the performance of, those design features necessary to insure compliance with required noise levels. Such design features may include but are not limited to prop phasing, jet noise suppressors, separation of occupied areas from noise sources, relative placement of engines and fuselage, and related items.
- 3.2.2 Acoustical treatment Where applicable, soundproofing conforming to Specification MIL-I-7171 shall be installed in accordance with Specification MIL-S-6144 to effect the specified levels. Acoustic treatments including other than flexible blankets shall be of approved types.

## 3.3 Reports required -

- 3.3.1 Engineering report on noise control measures The contractor shall furnish an engineering report for approval by the procuring activity prior to fabrication of the prototype or major modification of an aircraft which shall include the following material:
  - (a) Engineering estimates of the noise to be developed inside the aircraft and the engineering basis (pertinent structural data, tests, calculations, etc) for such estimates.
  - (b) Sketches and tables showing the type, total and unit weight, total area, construction, location, and method of fastening of all soundproofing to be installed to reduce noise to the levels heretofore required.
  - (c) Estimates or measurements of noise levels generated by auxiliary systems.

- 3.3.2 <u>Flight test noise level report</u> The contractor shall furnish a noise level measurement report prior to acceptance of the aircraft, which shall include the following material:
  - (a) A brief description of the noise level measuring equipment used, (including microphones).
  - (b) Location of the microphones.
  - (c) The recorded data in tabular or graphical form.
  - (d) Test conditions under which the recording was made.
  - (e) A brief description of the soundproofing installed in the test aircraft.

## 4. QUALITY ASSURANCE PROVISIONS

- 4.1 Sampling Acoustic noise level measurements shall be made on each experimental aircraft, and on an early model of each production aircraft or modification thereof.
- 4.2 <u>Test conditions</u> Acoustical noise level measurements shall be made under the following conditions:
- 1.2.1 Test Equipment Measurements shall be obtained with noise level measuring equipment demonstrated to be substantially in accordance with the performance requirements of Specification MIL-S-3151. The test equipment including the microphone, shall be calibrated throughout the frequency range of use. The calibration shall be applied in reporting the results in each octave band.
- 4.2.2 Stations Measurements shall be made near the head levels of all crew stations and of a representative number of passenger stations. The stations selected for measurements shall be subject to the approval of the procuring activity; additional stations may be required if considered necessary. Measurements shall be made for a sufficient period of time to permit a sampling of minimum and maximum levels by octave band.
- 4.2.3 Altitude Except as specified in 3.1.2, measurements shall be obtained with the aircraft in level forward flight at the lowest altitude sufficient to insure the maximum pressure differential between the cabin and the external atmospheric pressure.

- (a) For combat aircraft, measurements shall be made at both normal and combat pressure differentials.
- (b) If the aircraft is not pressurized, the altitude shall be the highest at which maximum control power can be obtained, provided this altitude does not exceed 12,000 ft.

## 4.2.4 Power conditions -

- (a) In order to determine con.pliance with 3.1.1 and 3.1.3, the aircraft shall be operated at maximum continuous power with all auxiliary systems, which normally operate for more than 5 minutes, in full operation.
- (b) In order to determine compliance with 3.1.4, the aircraft shall be operated at normal cruise power with all auxiliary systems, which normally operate for more than 5 minutes, in full operation.
- (c) In order to determine compliance with 3.1.2, the aircraft shall be operated under the appropriate flight conditions which maximize the resulting acoustical noise.

#### 5. PREPARATION FOR DELIVERY

5.1 This section is not applicable to this specification.

## 6. NOTES

6.1 Intended use - This specification defines the general requirements for the control of acoustical noise in occupied spaces of aircraft, including the acceptable noise levels, and the testing requirements for determining conformance to these levels.

### 6.2 Definitions -

- 6.2.1 <u>Acoustic reference level</u> The reference sound pressure level for measurements made in accordance with this specification will be the level produced by a sound pressure of 0.0002 dyne/cm<sup>2</sup>.
- 6.2.2 <u>Acoustical noise level</u> The acoustical noise level of the aircraft shall be considered to be the numerical average of the measured minimum and

maximum levels, provided this average is not less than 3 db below the maximum. In the latter case the reported level shall be the maximum less 3 db.

- Overall accustical noise level The term overall accustical noise level will be interpreted as including all noise within the frequency range from 22.4 to 11200 cycles per second.
- Maximum continuous power Maximum continuous power is 6.2.4 the maximum power that the engine can develop for continuous operation in level flight at the altitude where measurements are to be taken.
- Normal cruise power Normal cruise power is the power the engine can develop for maximum range in level flight at the altitude where measurements are to be taken.
- Auxiliary systems An auxiliary system is any mechanism 6.2.6 or structure other than the airframe or power plant which performs a function at some time during the operation of the aircraft, e.g., heat and vent, pressurization, defrost and defog, inverters, pumps, Auxiliary Power Unit (APU), etc.
- The noise level tables were developed from consideration of 6.3 damage to hearing, speech communication requirements, and effects on crew performance. It is recognized that these levels represent a compromise between those desired and those considered attainable within the state of the art of noise control in aircraft. This compromise considers the noise characteristics of present turbojet, turboprop, and piston-driven aircraft.

Custodians:

Army - MO

Navy - AS

Air Force - 11

Preparing Activity:

Navy · AS

(Project 1500 - 0054)

Reviewers:

Army - MO

Navy - AS

vir Force - 11

Users:

Navy - MC, CG

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