

INCH-POUND

**MIL-PRF-87260B(USAF)
2 November 2006**

**SUPERSEDING
MIL-PRF-87260A(USAF)
13 February 1998**

PERFORMANCE SPECIFICATION

**FOAM MATERIAL, EXPLOSION SUPPRESSION,
INHERENTLY ELECTROSTATICALLY CONDUCTIVE,
FOR AIRCRAFT FUEL TANKS**



AMSC: N/A

FSC: 9330

DISTRIBUTION STATEMENT A. Approved for public release; distribution is unlimited.

MIL-PRF-87260B(USAF)

FOREWORD

This specification is approved for use by the Department of the Air Force and is available for use by all Departments and Agencies of the Department of Defense.

Comments, suggestions, or questions on this document should be addressed to ASC/ENOI, 2530 Loop Road West, Wright-Patterson AFB OH 45433-7101 or emailed to Engineering.Standards@wpafb.af.mil. Since contact information can change, you may want to verify the currency of this address information using the ASSIST Online database at <http://assist.daps.dla.mil/>.

MIL-PRF-87260B(USAF)

PARAGRAPH	CONTENTS	PAGE
1.	SCOPE	1
1.1	Scope	1
1.2	Classification	1
2.	APPLICABLE DOCUMENTS	1
2.1	General	1
2.2	Government documents	1
2.2.1	Specifications, standards, and handbooks	1
2.2.2	Other Government documents, drawings, and publications	2
2.3	Non-Government publications	2
2.4	Order of precedence	3
3.	REQUIREMENTS	4
3.1	Qualification	4
3.1.1	Re-qualification	4
3.2	Materials	4
3.2.1	Product toxicological data and disposal	4
3.2.2	Human exposure toxicology	4
3.2.3	Tracer elements	4
3.2.4	Manufacturer internal specification and production quality control documents	5
3.3	Age	5
3.3.1	Storage life	5
3.3.2	Storage limit	5
3.4	Coloring pigments	5
3.5	Marker legibility	5
3.6	Physical properties and characteristics	6
3.7	Performance requirements	6
3.7.1	Fuel immersion	6
3.7.2	Hydrolytic stability	6
3.7.3	Explosion suppression and flame arrestor characteristics	6
3.7.4	Corrosion and adhesion	8
3.7.5	Electrical resistivity permanence	8
3.7.6	Fuel compatibility	8
3.7.7	Electrical resistivity and electrostatic fuel impingement test	8
3.7.8	Infrared spectrum analysis	9
3.7.9	Static charge dissipation test	9
3.7.10	Electrochemical corrosion	9
3.7.11	Aircraft service test	9
3.7.12	Manufacturing demonstration details	9
3.8	Dimensions and tolerances	9
3.8.1	Optional bun sizes	10
3.9	Final product identification	10
3.10	Marking	10
3.10.1	Additional marking	10
4.	VERIFICATION	11
4.1	Classification of tests	11
4.2	Test conditions	11
4.2.1	Temperature and humidity	11
4.2.2	Test fluids	11
4.2.3	Basic property testing	11
4.2.4	Specimen cutting	11

MIL-PRF-87260B(USAF)

PARAGRAPH	CONTENTS	PAGE
4.3	Qualification testing.	11
4.3.1	Test sample(s).	11
4.3.1.1	Test specimens.	12
4.3.1.1.1	Test section location.	12
4.3.1.1.2	Quantity of specimens.	12
4.3.2	Test report, disposition of test specimens, and data for the qualifying activity.	12
4.3.3	Qualification approval.	13
4.3.4	Qualification tests.	13
4.4	Conformance inspections.	13
4.4.1	Production tests.	13
4.4.2	Lot tests.	14
4.4.2.1	Rejection and retest.	14
4.4.3	Process control tests.	14
4.4.3.1	Rejection and test.	15
4.5	Test methods.	15
4.5.1	Examination of product.	15
4.5.2	Color test.	15
4.5.3	Density test.	15
4.5.3.1	Density uniformity.	15
4.5.4	Porosity (air pressure drop) test.	15
4.5.5	Tensile strength and ultimate elongation tests.	18
4.5.6	Tear resistance test.	18
4.5.7	Constant deflection compression set test.	18
4.5.8	Compression load deflection (CLD) test.	18
4.5.9	Fuel displacement test.	19
4.5.9.1	Theoretical fuel displacement.	19
4.5.10	Fuel and water retention test.	19
4.5.10.1	Fuel retention tests.	19
4.5.10.2	Water retention test.	20
4.5.10.3	Test data and sample requirements.	20
4.5.11	Extractable material test.	21
4.5.12	Volume swell test.	21
4.5.13	Low temperature flexibility test.	21
4.5.14	Entrained solid contamination tests.	22
4.5.15	Steam autoclave exposure test.	22
4.5.16	Fluid immersion tests.	22
4.5.16.1	Tension tests.	22
4.5.16.2	Electrical resistivity test.	23
4.5.16.3	Immersion tests.	23
4.5.17	Hydrolytic stability tests.	23
4.5.17.1	Humidity exposure.	23
4.5.17.1.1	Tension tests.	23
4.5.17.2	Water immersion.	24
4.5.17.2.1	Testing frequency.	24
4.5.17.3	Dry heat.	24
4.5.17.3.1	Testing frequency.	24
4.5.17.4	Data requirements.	24
4.5.18	Explosion suppression and flame arrestor characteristics.	24
4.5.18.1	Material sizing.	25

MIL-PRF-87260B(USAF)

PARAGRAPH	CONTENTS	PAGE
4.5.18.2	Additional tests.....	25
4.5.18.2.1	Single void ignitions.	26
4.5.18.2.2	For Grade II (fine pore material only).	26
4.5.19	Corrosion and adhesion test.....	27
4.5.19.1	Roughness test.....	27
4.5.19.1.1	Surface finish.	27
4.5.19.2	Test report.....	27
4.5.20	Electrical resistivity permanence.	28
4.5.20.1	Water immersion.....	28
4.5.20.1.1	Electrical resistivity specimens.	28
4.5.21	Fuel compatibility test.	28
4.5.22	Electrical resistivity test.....	29
4.5.22.1	Test equipment and preconditioning environment.	29
4.5.22.2	Test procedure and calculation.	29
4.5.22.3	Correlation test data requirements.	31
4.5.23	Electrostatic fuel impingement test.....	31
4.5.23.1	Test data requirements.....	32
4.5.24	Infrared spectrum analysis test.....	32
4.5.25	Electrochemical corrosion test.....	32
4.5.25.1	Testing requirements.	32
4.5.25.2	Test metal panels.	34
4.5.25.3	Test exposure.	34
4.5.25.4	Test intervals.....	34
4.5.25.5	Qualification report.....	34
4.5.26	Marker legibility test.	34
4.5.26.1	Test report.....	35
4.5.27	Electrostatic charge dissipation test.	35
4.5.27.1	Test equipment.	35
4.5.27.2	Specimen preparation, conditioning and mounting.	35
4.5.27.3	Specimen test procedure.....	35
4.5.27.4	Report.	36
4.5.28	Aircraft service test evaluation.....	36
4.6	Inspection of preparation for delivery.	36
5.	PACKAGING.....	37
5.1	Packaging.	37
6.	NOTES.....	37
6.1	Intended use.	37
6.2	Acquisition requirements.	37
6.3	Qualification.	37
6.3.1	Qualifying activity.	38
6.3.2	Qualification tests.	38
6.4	Supersession data.	38
6.5	Definitions.	38
6.5.1	Run of material.....	38
6.5.2	Test section width.	38
6.5.3	Direction of rise.....	38
6.5.4	Machine direction.....	38
6.5.5	Lot.	39
6.6	Classes.	39

MIL-PRF-87260B(USAF)

PARAGRAPH	CONTENTS	PAGE
6.7	Installation guidelines for reticulated foam in aircraft fuel tanks and dry bays.	39
6.8	Service life.	39
6.9	Fabricated parts.	39
6.10	Air deionizer.	39
6.11	Marker identifier.	39
6.12	Product identification label card.	40
6.13	Subject term (key word) listing.	40
6.14	Changes from previous issues.	40

APPENDICES

A.1.	SCOPE	41
A.1.1	Scope.....	41
A.2.	APPLICABLE DOCUMENTS.....	41
A.3.	TEST EQUIPMENT AND MATERIALS	42
A.4.	TEST PARAMETERS AND DETAILS	42
A.5.	TEST FOAM TYPES AND CONFIGURATIONS	43
A.6.	TEST DATA AND DOCUMENTATION	43

FIGURES

FIGURE 1.	Typical porosity (air flow) test apparatus.....	16
FIGURE 2.	Porosity curve.....	17
FIGURE 3.	Flame arrestor apparatus.....	26
FIGURE 4.	Dimensional and electrical diagrams for measuring volume resistivity of conductive materials.	30
FIGURE 5.	Electromechanical test set-up.	33
FIGURE A-1.	Fuel impingement test set-up schematic.	44
FIGURE A-2.	Schematic of two different impingement scenarios.	45
FIGURE A-3.	Tank lid for the uppermost tank of the fuel impingement test rig.	46
FIGURE A-4.	Fuel impingement summary test data sheet.	47
FIGURE A-5.	Fuel impingement test data sheet.	48

TABLES

TABLE I.	Physical properties and characteristics.....	7
Concluding Material	49

MIL-PRF-87260B(USAF)

1. SCOPE**1.1 Scope.**

This document covers the requirements for an electrically conductive open-celled foam material that will be used for explosion suppression in aircraft fuel.

1.2 Classification.

The conductive explosion suppression material (ESM) will be of the following classes and grades (see 6.6):

Class 1 – ESM to be used for single point and over-wing refueling throughout the temperature range of +10 to +160°F

Grade I – Coarse pore

Grade II – Fine pore

Class 2 – ESM to be used for single point and over-wing refueling throughout the temperature range of -25 to +160°F

Grade I – Coarse pore

Grade II – Fine pore

2. APPLICABLE DOCUMENTS**2.1 General.**

The documents listed in this section are specified in *sections 3, 4, or 5* of this specification. This section does not include documents cited in other sections of this specification or recommended for additional information or as examples. While every effort has been made to ensure the completeness of this list, document users are cautioned that they must meet all specified requirements of documents cited in *Sections 3, 4, or 5* of this specification, whether or not they are listed.

2.2 Government documents.**2.2.1 Specifications, standards, and handbooks.**

The following specifications, standards, and handbooks form a part of this document to the extent specified herein. Unless otherwise specified, the issues of these documents are those cited in the solicitation or contract.

FEDERAL SPECIFICATIONS

A-A-208	Ink, Marking, Stencil, Opaque (Porous and Non-Porous Surfaces)
A-A-3174	Plastic Sheet, Polyolefin

MIL-PRF-87260B(USAF)

DEPARTMENT OF DEFENSE SPECIFICATIONS

MIL-DTL-5624	Turbine Fuel, Aviation, Grades JP-4 AND JP-5
MIL-DTL-83054	Baffle and Inerting Material, Aircraft Fuel Tank
MIL-DTL-83133	Turbine Fuels, Aviation, Kerosene Types, NATO F-34(JP-8), NATO F-35, AND JP-8 + 100
MIL-PRF-87937	Cleaning Compound, Aerospace Equipment

DEPARTMENT OF DEFENSE STANDARDS

MIL-STD-129	Military Marking for Shipment and Storage
-------------	---

DEPARTMENT OF DEFENSE HANDBOOKS

MIL-HDBK-831	Preparation of Test Reports
--------------	-----------------------------

(Copies of these documents are available online at <http://assist.daps.dla.mil/quicksearch/> or from the Standardization Document Order Desk, 700 Robbins Avenue, Building 4D, Philadelphia, PA 19111-5094.)

2.2.2 Other Government documents, drawings, and publications.

The following other Government documents, drawings, and publications form a part of this document to the extent specified herein. Unless otherwise specified, the issues of these documents are those cited in the solicitation or contract.

AIR FORCE ENGINEERING AND SERVICE CENTER TEST REPORTS

ESC-TR-84-63	Effective Disposal of Fuel Cell Polyurethane Foam
--------------	---

(Copies of this report are available online at <http://www.ntis.gov> or from the National Technical Information Service (NTIS), 5285 Port Royal Road, Springfield VA 22161.)

AIR FORCE TECHNICAL ORDER

T.O. 42B-1-1	Quality Control of Fuels and Lubricants
--------------	---

(Copies of Air Force Technical Orders are available through your contracting officer or from Oklahoma City Air Logistics Center (OC-ALC/LGLU), Tinker Air Force Base, OK 73145-9147.)

2.3 Non-Government publications.

The following documents form a part of this document to the extent specified herein. Unless otherwise specified, the issues of these documents are those cited in the solicitation or contract.

MIL-PRF-87260B(USAF)

AMERICAN SOCIETY FOR TESTING AND MATERIALS (ASTM)

ASTM-A967	Drive-Studded, Accessory, (254) 10.000 BC Round, Design Standard For
ASTM-B209	Aluminum and Aluminum-Alloy Sheet and Plate
ASTM D257	Standard Test Methods for DC Resistance or Conductance of Insulating Materials
ASTM D1655	Standard Specification for Aviation Turbine Fuels
ASTM D2276	Standard Test Method for Particulate Contaminant in Aviation Fuel by Line Sampling
ASTM D3574	Standard Test Methods for Flexible Cellular Materials Slab, Bonded, and Molded Urethane Foams
ASTM F1110	Standard Test Method for Sandwich Corrosion Test

(Copies of ASTMs are available from <http://www.astm.org> or from the American Society for Testing and Materials, 100 Barr Harbor Drive, West Conshohocken, PA 19428-2959.

SAE INTERNATIONAL (SAE)

SAE-AMS-QQ-P-35	Passivation Treatments for Corrosion-Resistant Steel
SAE-AMS-QQ-A-250	Aluminum and Aluminum Alloy, Plate and Sheet General Specification for
SAE AMS-2629	Fluid, Jet Reference
SAE AIR-4170	Reticulated Polyurethane Foam Explosion Suppression Material for Fuel Systems and Dry Bays

(Copies of SAEs are available from <http://www.sae.org> or from the SAE World Headquarters, 400 Commonwealth Drive, Warrendale, PA 15096-0001.)

2.4 Order of precedence.

In the event of a conflict between the text of this document and the references cited herein, the text of this document takes precedence. Nothing in this document, however, supersedes applicable laws and regulations unless a specific exemption has been obtained.

MIL-PRF-87260B(USAF)

3. REQUIREMENTS

3.1 Qualification.

Material furnished under this specification shall be products that are authorized by the qualifying activity for listing on the applicable qualified products list before contract award (see 4.3 and 6.3).

3.1.1 Re-qualification.

Before any change is made in the quality, composition, source of ingredients, or source of supply of the final product, the manufacturer must contact the qualifying activity to determine if re-qualification or partial re-qualification is required (see 6.3.1).

3.2 Materials.

The raw materials used in processing the material shall be of the highest quality, and standards for commercially available products of this type shall be of the same formulation as that used in the qualification test sample. The end product shall be flexible urethane foam which is suitable for use in aircraft fuel tanks. The final product shall be a material that is inherently electrostatically conductive. It shall be compatible with all materials found inside aircraft fuel tanks and systems.

3.2.1 Product toxicological data and disposal.

Product toxicological data relating to the kit processing and end use, particularly by hot wire thermal decomposition, shall be submitted in the qualification test report for review by an appropriate military medical authority who will determine if any use restrictions exist relative to the safety of personnel. In addition, available data relating to possible methods of disposal for the foam material shall be provided. This includes both new material and material that has been exposed to fuel in an aircraft fuel tank. For additional guidelines on the disposal of fuel cell reticulated foams, consult *ESC-TR-84-63*.

3.2.2 Human exposure toxicology.

The material shall not pose a toxicity hazard to personnel who come in contact with the product. Toxicological data substantiating its safety shall be provided along with the conductive material composition for review by military medical authority to verify personnel safety. All proprietary information shall be protected from disclosure in accordance with appropriate Federal regulations.

3.2.3 Tracer elements.

A tracer element shall be incorporated into the material for identification purposes. The tracer element shall be submitted to the qualifying activity for approval and shall be unique to each vendor. The manufacturer shall also supply the analytical test procedure that can be used to identify the tracer element.

MIL-PRF-87260B(USAF)

3.2.4 Manufacturer internal specification and production quality control documents.

The conductive material manufacturer shall provide adequate internal documents (specifications/manuals) to define and control the production, testing, and quality control of the material. The qualification documentation shall include, as a minimum, the manufacturer's quality control manual and internal specification that describe such items as: manufacturer's product, product part numbers, available bun sizes, production testing and frequency of the production tests, testing requirements (limits), quality control procedures, and any other information relative to the materials' certification or usage. This document(s) shall be made available for use by the Government personnel during production facility inspections and during material procurement certification. The vendor's quality control document(s) should also be referenced in the product qualification report and a copy provided to the qualifying agency as part of the qualification package.

3.3 Age.

The maximum time of delivery from the manufacturer shall not exceed one year. If the time since manufacture exceeds six months, the material shall be visually inspected, and there shall be no evidence of discoloration resulting in surface deterioration that results in a loss of tension properties. Discoloration of urethane foams with age and exposure to ultraviolet light is a normal occurrence and does not necessarily indicate deterioration.

3.3.1 Storage life.

The storage life of the material covered by this specification shall be three years from the date of manufacture, provided it is maintained in the original sealed polyethylene bag plus opaque overwrap at temperatures below 90 degrees Fahrenheit (°F). Storage should be in an area out of direct sunlight and outside weather, including high humidity and temperature. The material should be inspected for evidence of discoloration or surface deterioration (loss in tensile properties) prior to use.

3.3.2 Storage limit.

Material that exceeds the three-year storage limit shall be re-certified by the manufacturer or other competent test group that the material has not been deteriorated or contaminated.

3.4 Coloring pigments.

Coloring pigments shall not be readily extractable when the conductive material is used in contact with fuels conforming to *MIL-DTL-5624*, *MIL-DTL-83133*, and *ASTM D1655*. Product color is not limited except that it shall not be blue, orange, yellow, or red. The preferred color is charcoal grey.

3.5 Marker legibility.

The manufacturer shall identify a suitable marker for use on the conductive material kit pieces (individual identification numbers) to be installed in the aircraft fuel system. The marker shall be compatible with both fuel and the conductive material. A possible source for a fuel compatible marking ink is *A-A-208*.

MIL-PRF-87260B(USAF)

3.6 Physical properties and characteristics.

The physical properties and characteristics of the conductive material at the time of manufacture shall be suitable for the purpose intended and in accordance with [TABLE I](#) and the applicable paragraphs of [section 3](#).

3.7 Performance requirements.

The conductive material shall meet the following performance requirements (in addition to [TABLE I](#)):

3.7.1 Fuel immersion.

The following fuel immersion requirements shall be met:

a. JP-8 immersion at 160°F. The material shall not undergo more than 40.0 percent loss in “dry-to-dry” and “wet-to-wet” tension properties after 4, 8, and 12 weeks exposure. In addition, the dry electrical resistivity property of the material after fluid exposure shall not rise above 5.0×10^{12} ohm-centimeter (ohm-cm) after exposures of 4, 8, and 12 weeks.

b. Wet property assessment at 75°F. The material shall not undergo more than 60.0 percent loss in tension, compression load deflection, and tear resistance properties from dry-to-wet after a 4-week exposure to JP-8 fuel.

3.7.2 Hydrolytic stability.

The conductive material shall meet the following hydrolytic stability requirements:

a. Humidity exposure at 200°F/95 percent relative humidity. The material shall not undergo more than 65 percent loss in tensile strength after 6 weeks exposure. The electrical resistivity property of the material after exposure for 6 and 10 weeks shall not exceed 5.0×10^{12} ohm-cm.

b. Water immersion at 160°F. The material shall not undergo more than 40.0 percent loss in tensile strength after 12 weeks exposure.

c. Dry bay-dry heat tests at 250°F. The material shall not undergo more than 65 percent loss in tensile strength after 4 weeks exposure. The electrical resistivity property of the material shall also be reported on a minimum weekly basis for the duration of exposure and shall not exceed 5.0×10^{12} ohm-cm after 4 weeks exposure.

3.7.3 Explosion suppression and flame arrestor characteristics.

The conductive materials shall meet the following minimum requirements:

a. Coarse pore material. The coarse pore material shall suppress the combustion overpressure for a single void ignition of 20.0 volume percent to a value equal to or below 15.0 psid (pounds per square inch differential).

MIL-PRF-87260B(USAF)

TABLE I. Physical properties and characteristics.

Property	Requirement		
	Grade I	Grade II	Test Para. Ref.
Color	See 4.5.2	See 4.5.2	4.5.2
Density Range (lbs/ft ³)	1.20-1.50	1.20-1.50	4.5.3
Density Uniformity (lbs/ft ³)	Report	Report	4.5.3
Porosity (Air Pressure Drop)*	7.5-21.0	21.5-33.0	4.5.4
Air Pressure Drop (in. water)	0.150-0.250	0.260-0.360	4.5.4
Tensile Strength (psi)	10.0 min	15.0 min	4.5.5
Ultimate Elongation (percent)	100.0 min	100.0 min	4.5.5
Tear Resistance (psi)	3.0 min	3.0 min	4.5.6
Constant Deflection Compression Set (percent)	45.0 max	45.0 max	4.5.7
Compression Load Deflection at 25 percent (psi)	0.35 min	0.35 min	4.5.8
Compression Load Deflection at 65 percent (psi)	0.60 min	0.60 min	4.5.8
Fuel Displacement (vol percent)	2.50 max	2.50 max	4.5.9
Fuel Retention (vol percent) **	2.50 max	5.00 max	4.5.10
Water Retention (vol percent)	Report	Report	4.5.10
Extractable Materials (wt. percent)	3.0 max	3.0 max	4.5.11
Volume Increase (vol. percent)			
Type I Fluid	0-15.0	0-15.0	4.5.12
Type III Fluid	0-40.0	0-40.0	4.5.12
JP-8 Turbine Fuel	0-25.0	0-25.0	4.5.12
Low Temperature Flexibility	No Cracking or Breaking of Strands		4.5.13
Entrained Solid Contamination (mg/ft ³)	11.0 max	11.0 max	4.5.14
Steam Autoclave Exposure (percent tensile loss)	30.0 max	30.0 max	4.5.15
Electrical Resistivity (ohm-cm) at 72°F	1.0x10 ⁷ to 5.0x10 ¹¹		4.5.22
Electrical Resistivity Uniformity at 72°F	2 Orders of Magnitude from Top to Bottom	maximum	4.5.22

*For information only (not for procurement certification)

** To be determined at the maximum air pressure drop limit

MIL-PRF-87260B(USAF)

b. Fine pore material. The fine pore material shall suppress the combustion overpressure for a single void ignition of 35.0 percent volume to a value equal to or below 15.0 psid. In addition, the fine pore material shall prevent flame propagation for the following conditions:

(1) At 0.0 psig (pounds per square inch gage) initial pressure and a combustion volume of 16.7 percent volume, the required material thickness shall be 3.0 inches or less.

(2) At 3.0 psig initial pressure and a combustion volume of 16.7 percent volume, the required material thickness shall be 5.0 inches or less.

3.7.4 Corrosion and adhesion.

The conductive material shall neither adhere to nor cause any pitting, erosion, or corrosion to aluminum alloy plates when in contact for 14 days.

3.7.5 Electrical resistivity permanence.

The conductive material's electrical resistivity property shall not rise above 5.0×10^{12} ohm-cm after exposure to the following conditions:

- a. Water immersion at $120 \pm 5^\circ\text{F}$ for 4 weeks
- b. Water immersion at $160 \pm 5^\circ\text{F}$ for 3 weeks may be substituted

3.7.6 Fuel compatibility.

Properties of the JP-8 fuels exposed to the conductive material shall meet the following requirements listed in *MIL-DTL-83133*:

- a. Color shall be equal to or greater than 15.0 saybolt color units.
- b. Existent gum shall not exceed 14.0 milligrams (mg)/100 milliliters (ml).
- c. Particulates shall not exceed 1.0 mg/liter.
- d. Filtration time shall not exceed 10.0 minutes.
- e. Total acid number shall not exceed 0.015 milligram/KOH/gram.
- f. JFTOT ΔP (260) shall not exceed 25.0 millimeters (mm) of mercury (Hg).
- g. No visible extraction of material coloring pigment.

3.7.7 Electrical resistivity and electrostatic fuel impingement test.

The electrical resistivity and electrostatic compatibility characteristics of the conductive material shall meet the following requirements:

a. The dry electrical resistivity of both classes of conductive material shall be between 1.0×10^7 and 5.0×10^{11} ohm-cm when tested at $72 \pm 5^\circ\text{F}$ and 10, 50, 95 ± 5 percent relative humidity. In addition, the electrical resistivity shall be measured and reported for the following temperatures: 140, 60, 32, 10, 0, -15, -25, -30, -40, $\pm 2^\circ\text{F}$.

b. Class I Materials. The Class I conductive materials shall not produce an incendiary vapor ignition, electrical discharges, or excessive static charge build-up (electrical activity) when impinged by JP-8 fuel that has a conductivity less than 10 conductivity units (cu) as well as in

MIL-PRF-87260B(USAF)

the range from 100 to 800 cu, while in the range of temperatures from 160°F down to +10°F. The range of conductivity shall be achieved through the use of fuel system conductivity additives defined in *MIL-DTL-83133*.

c. **Class II Materials.** The Class II conductive materials shall not produce an incendiary vapor ignition, electrical discharges, or excessive static charge build-up (electrical activity) when impinged by JP-8 fuel that has a conductivity less than 10 cu as well as in the range from 100 to 800 cu, while in the range of temperatures from 160°F down to -25°F. The range of conductivity shall be achieved through the use of fuel system conductivity additives defined in *MIL-DTL-83133*.

3.7.8 Infrared spectrum analysis.

Infrared spectrum data shall be provided for the conductive material and reference standard polyether foam.

3.7.9 Static charge dissipation test.

The conductive material(s) shall be evaluated to determine the time it takes for an induced electrostatic charge to be dissipated at various temperatures including the following: 72, 60, 32, 10, 0, -15, -25, -30, -40 ±2°F. The test procedure defined in 4.5.27 or 4.5.28 shall be used to determine the static charge dissipation times. For Class I the charge shall be dissipated at a maximum of 10 minutes at 10°F. For Class II the charge shall be dissipated at a maximum of 10 minutes at -25°F. This charge dissipation test may be substituted for the 3.7.7 fuel impingement test during qualification.

3.7.10 Electrochemical corrosion.

The conductive material shall neither adhere to nor cause any pitting, erosion, or discoloration of metal plates. In addition, the material shall not produce a current flow greater than that for blue polyether foam produced in accordance with *MIL-DTL-83054*.

3.7.11 Aircraft service test.

The service tested materials' critical properties shall not degrade below the minimum specification requirements. The electrical resistivity shall not increase by more than two orders of magnitude from the original baseline values nor shall it exceed 5.0×10^{12} ohm-cm. There shall be no evidence of electrical buildup, discharge, vapor ignitions, or singeing (burning) of the foam during the service test evaluations.

3.7.12 Manufacturing demonstration details.

The manufacturer shall successfully demonstrate that the material can be fabricated in the various standard bun sizes defined in 3.8. Hot wire is acceptable for use in final fabrication trimming of the materials provided it does not discolor the bun surfaces non-uniformly.

3.8 Dimensions and tolerances.

Material shall be produced in the following standard size buns:

a. Grade I – 40 inches x 80 inches x 8 inches; 44 inches x 110 inches x 8 inches; 44 inches x 110 inches x 12 inches; and 44 inches x 110 inches x 4 inches

MIL-PRF-87260B(USAF)

b. Grade II – 44 inches x 110 inches x 8 inches; 44 inches x 110 inches x 12 inches; and 44 inches x 110 inches x 4 inches

3.8.1 Optional bun sizes.

Optional bun sizes of the material may also be produced by the manufacturer provided the following sizes are offered: 44 inches x 110 inches x 12 inches for Grade I and 40 inches x 80 inches x 8 inches for Grade II. Manufacturing tolerance limits on bun sizes shall be as follows unless otherwise agreed to by the procuring activity and manufacturer:

- a. Width +1, -0 inch
- b. Length +1, -0 inch
- c. Height +1/8, -1/8 inch

3.9 Final product identification.

On the end of each bun, a durable label card shall be attached using 3-inch loop style Swiftachment fasteners. The label card shall clearly identify the manufacturer's part number, date of manufacture, manufacturer's run number, lot number, and bun number. Where applicable, the Government contract or order number shall be included. There shall be no marking or color coding on the outer surface of the bun. Each bun shall be sealed in a clean 4-mil black polyethylene bag, per *A-A-3174 Type I, Class I, Grade C*, as it comes off the manufacturing line to protect it from contamination.

3.10 Marking.

Marking shall be in accordance with *MIL-STD-129*. The nomenclature shall be as follows:

**INHERENTLY ELECTRICALLY CONDUCTIVE EXPLOSION
SUPPRESSION MATERIAL, AIRCRAFT FUEL TANK**

3.10.1 Additional marking.

In addition to the nomenclature, each unit package or container shall contain the following information:

- Specification number
- Class and grade of material
- Manufacturer's part number
- Date of manufacture

MIL-PRF-87260B(USAF)

4. VERIFICATION

4.1 Classification of tests.

The inspection requirements specified herein are classified as follows:

- a. Qualification tests ([4.3](#))
- b. Conformance inspection ([4.4](#))

4.2 Test conditions.

4.2.1 Temperature and humidity.

Unless otherwise specified herein, all tests shall be conducted under known conditions of temperature and relative humidity. Prior to physical property testing, specimens shall be preconditioned in the test environment a minimum of 30 minutes.

4.2.2 Test fluids.

Unless otherwise specified herein, the test fluids shall be of known properties and certified in accordance with the referenced military specification. The turbine fuels conforming to *MIL-DTL-83133* may be obtained from the qualifying activity along with a certified test report defining, as a minimum, the specific gravity, distillation, aromatic content, existent gum, sulfur content, fuel system icing inhibitor level, and fuel electrical conductivity level.

4.2.3 Basic property testing.

Unless otherwise specified herein, all basic property tests shall be in accordance with the applicable sections specified in *ASTM D3574*. In the case where more than one specimen is tested, the average shall be determined. However, all values shall be reported for all but production testing. Unless otherwise specified, all sample specimens shall be tested in the dry condition. In the case where fuel-wet testing is required (special tension, tear resistance, and compression load deflection tests) the specimen should be removed from the test fluid immediately prior to property testing, drained of excess fuel, and then tested.

4.2.4 Specimen cutting.

Unless otherwise specified herein, test specimen cutting shall be by die, saw cutting, or hot wire cutting.

4.3 Qualification testing.

See [6.3](#).

4.3.1 Test sample(s).

The specific bun(s) of material chosen for the qualification tests shall be typical of future production buns in terms of density and porosity (air pressure drop). Unless otherwise specified herein, this bun shall be selected from near the midrange in allowable porosity properties.

MIL-PRF-87260B(USAF)

Additional quantities of material will be required at the upper and lower areas of the porosity range for fluid retention, explosion suppression and flame arrestor testing.

4.3.1.1 Test specimens.**4.3.1.1.1 Test section location.**

All test specimens shall be prepared from production material within the test section locations specified herein.

4.3.1.1.1.1 Qualification and process control tests.

For qualification and process control tests, the test section shall consist of a full-size bun that has been sectioned to provide for all the qualification test samples and test specimens. All qualification test specimens used shall be from the same machine run of production material and from the specified area defined in 4.5. Where practicable, the material used shall be representative of the mid range in density and pore size (air pressure drop) for the given product.

4.3.1.1.1.2 For production and lot testing.

For production and lot testing, the test section shall consist of a section approximately 15 inches long by the normal bun height and width which has been processed along with normal production material or taken from a production bun. Location of the specific test samples within the test section shall be in accordance with the guidelines specified in 4.5. Specimen measurements shall be in accordance with *ASTM D3574*.

4.3.1.1.2 Quantity of specimens.

Unless otherwise specified herein, three specimens per sample shall be tested. The value reported shall be the average of those observed. If any value deviates more than 20.0 percent from the average value, two additional specimens shall be tested and the average for all five values shall be reported.

4.3.2 Test report, disposition of test specimens, and data for the qualifying activity.

The following shall be furnished to the qualifying activity as a qualification package:

a. Test report. A qualification test report shall be prepared in accordance with *MIL-HDBK-831* and shall include, as a minimum, the following:

(1) A tabulation of all qualification test data, including production test data on the qualification foam run. All values obtained shall be included as well as sample calculations.

(2) Detailed discussion of any failures, and retesting data.

b. Disposition of test specimens. All test specimens used in the qualification tests shall be submitted to the qualifying activity (see 6.3.1), except those subjected to the following tests:

(1) Tension test (4.5.5)

(2) Tear resistance test (4.5.6)

MIL-PRF-87260B(USAF)

(3) Steam autoclave exposure test ([4.5.15](#))

(4) Infrared spectrum analysis test ([4.5.24](#))

c. Test material. In addition, the following material shall be submitted to the qualifying activity:

(1) A sample from the qualification test bun(s) or adjacent bun: Size 20 inches x 20 inches x the bun thickness.

(2) Unused retention samples (6 inches x 6 inches x 6 inches) near the bottom, middle, and top of the porosity range and corresponding porosity (air pressure drop) specimens which have been air pressure drop tested.

(3) Sufficient material (approximately four buns of Grade I or two buns of Grade II) for the flame arrestor tests specified in [4.5.18](#).

(4) Corrosion and adhesion metal test specimens (see [4.5.19](#)).

d. Other data. The manufacturer shall include, as a part of the report, any information defined in [6.3](#) that may not have been available at the time of the submittal of the letter of request for testing. Also, any applicable data or information that may relate to the qualification testing or future procurements of the material shall be included. Information referenced in [3.2.1](#), [3.2.2](#), [3.2.3](#), [3.2.4](#), and [3.5](#) shall also be included in the qualification test report.

4.3.3 Qualification approval.

Qualification test reports shall be signed or approved by a responsible representative of the manufacturer.

4.3.4 Qualification tests.

The qualification tests shall consist of all applicable tests described under [4.5](#).

4.4 Conformance inspections.

Conformance inspections shall consist of the following tests:

- a. Production tests ([4.4.1](#))
- b. Lot tests ([4.4.2](#))
- c. Process control tests ([4.4.3](#))
- d. Examination of product ([4.5.1](#))

4.4.1 Production tests.

Production tests shall be conducted on each run of material (see [6.3.1](#)) produced in accordance with the following schedule:

- a. All furnished buns of material shall be visually inspected for examination of product per [4.5.1](#).

MIL-PRF-87260B(USAF)

b. At the front, middle and end of each production run, the following tests shall be conducted once on all products:

- (1) Color test (4.5.2)
- (2) Density test (4.5.3)
- (3) Porosity (air pressure drop) test (4.5.4)
- (4) Tensile strength and elongation tests (4.5.5)
- (5) Entrained solid contamination tests (4.5.14)
- (6) Steam autoclave exposure test (4.5.15) (See NOTE below)
- (7) Electrical resistivity (4.5.22)

NOTE: The steam autoclave exposure test specified in 4.5.15 shall be conducted only once for each machine run to verify the hydrolytic stability characteristics of the material. If the bun thickness is less than 8 inches, then the entrained solid contamination test specimen should be stacked to get an 8-inch thickness.

4.4.2 Lot tests.

In addition to the production tests specified in 4.4.1, the compression load deflection test specified in 4.5.8 and the fuel retention test specified in 4.5.10.1 shall be conducted on each lot (see 6.5.5) or 6-month interval, whichever occurs first. The results shall be maintained on file for future reference.

4.4.2.1 Rejection and retest.

Failure of any of the test specimens to conform to the applicable requirements of 3.7 and TABLE I shall require a retest of the property which failed on an additional set of test specimens from the same test section or from the top of the adjacent bun. Additional testing will be authorized by the qualifying activity in order to isolate the extent of defective material. In the event of failure of any of the retested specimens, the material represented by those specimens shall be rejected.

4.4.3 Process control tests.

In addition to the production and lot tests, the following tests shall be conducted on production material at 12-month intervals, and the results maintained on file for future reference:

- a. Tear resistance test (4.5.6)
- b. Volume swell in JP-5 (4.5.12)
- c. Fluid immersion (wet property tests) (4.5.16.3)
- d. Electrical resistivity permanence (4.5.20)

MIL-PRF-87260B(USAF)

4.4.3.1 Rejection and test.

Failure of any of the test specimens to conform to the process control requirements specified herein shall require a retest of one additional set of test specimens for the property that failed from the same test section. In the event of failure of any of the retested specimens, production shall be halted and no additional material accepted until the reason for failure has been determined and corrective action taken. The qualifying activity shall be notified of any test failures encountered.

4.5 Test methods.**4.5.1 Examination of product.**

Each finished bun of material shall be visually inspected for consistency of cell structure, color, complete reticulation, obvious voids, or surface imperfections and the dimensional tolerances specified in [3.8](#) prior to final packaging. Criteria for rejection of buns shall be any exterior surface defects that could seriously affect the end function of the product, such as:

- a. Excessive cleaves, voids, or splits.
- b. Holes larger than 1/2 inch in diameter and 1/2 inch in depth, not to exceed four per bun and no closer together than 2 feet.
- c. Level of non-reticulation not to exceed 0.40 percent of the total surface area or 0.07 percent of the total volume, based on the standard size bun.

4.5.2 Color test.

Testing for color shall be by visual analysis. The material shall be of a uniform color. Any unusual color variations over the foam surface shall be cause for rejection, especially distinct surface darkening due to dirt, contamination, or surface deterioration or any color mottling. The color of the product is not limited except that it shall not be blue, orange, yellow, or red.

4.5.3 Density test.

One test specimen shall be tested in accordance with *ASTM D3574 (Test A)*. Specimen size shall be 3 inches x 7 inches x 10 inches, such that the 3-inch dimension is in the direction of the width (see [6.5.2](#)) and the 7-inch dimension is in the direction of rise ([6.5.3](#)) of the test section. The results shall be reported to the nearest 0.1 pound per cubic foot (lb/ft³).

4.5.3.1 Density uniformity.

The density uniformity of the product shall be demonstrated during qualification testing by sampling a bun of product at a minimum of ten locations throughout the bun (height) top and bottom and tested for density per [4.5.3](#). The specimen size shall be 3 inches x 4 inches x 10 inches. The 4-inch dimension shall be in the height direction. The density variation shall be determined and all values reported for both buns of material.

4.5.4 Porosity (air pressure drop) test.

The pore size determination shall be by the air pressure drop technique specified herein. Two specimens per sample shall be run for production tests and three specimens per sample for qualification tests. The cylindrical specimen shall be 10 inches in diameter by 1 ±0.02 inch

MIL-PRF-87260B(USAF)

thick, where the 1-inch dimension is in the height direction of the test section. For production tests the porosity test specimen shall be taken with the top and bottom 3 inches of the test section. For qualification testing, the three specimens shall be taken from the same location but from the upper, middle, and lower portions of the bun height. Pressure drop measurements shall be made using a porosity test rig (see [FIGURE 1](#)) which has been properly calibrated. Calibration shall be conducted on a daily basis using either a special pressure drop screen or orifice plate in order to determine the reference setting for the orifice differential manometer, or a correction factor that accounts for density variations of the air. Prior to sample testing, the manometer which indicates the air flow shall be adjusted to zero with no airflow. The specimen shall then be inserted into the sample holder until it is properly seated into the cutout. The blower shall be started and the airflow set to coincide with the daily reference calibration setting on the orifice differential manometer. Next, read the sample pressure drop (uncorrected) to the nearest 0.005 inch on the 4-inch manometer (designated sample differential). The value shall then be corrected for thickness (if other than 1.00 inch thickness) by dividing it by the measured sample thickness. This corrected air pressure drop shall then be compared to the porosity curve ([FIGURE 2](#)) in order to determine the average pore size for the sample specimens. The sample pressure drop shall be reported.

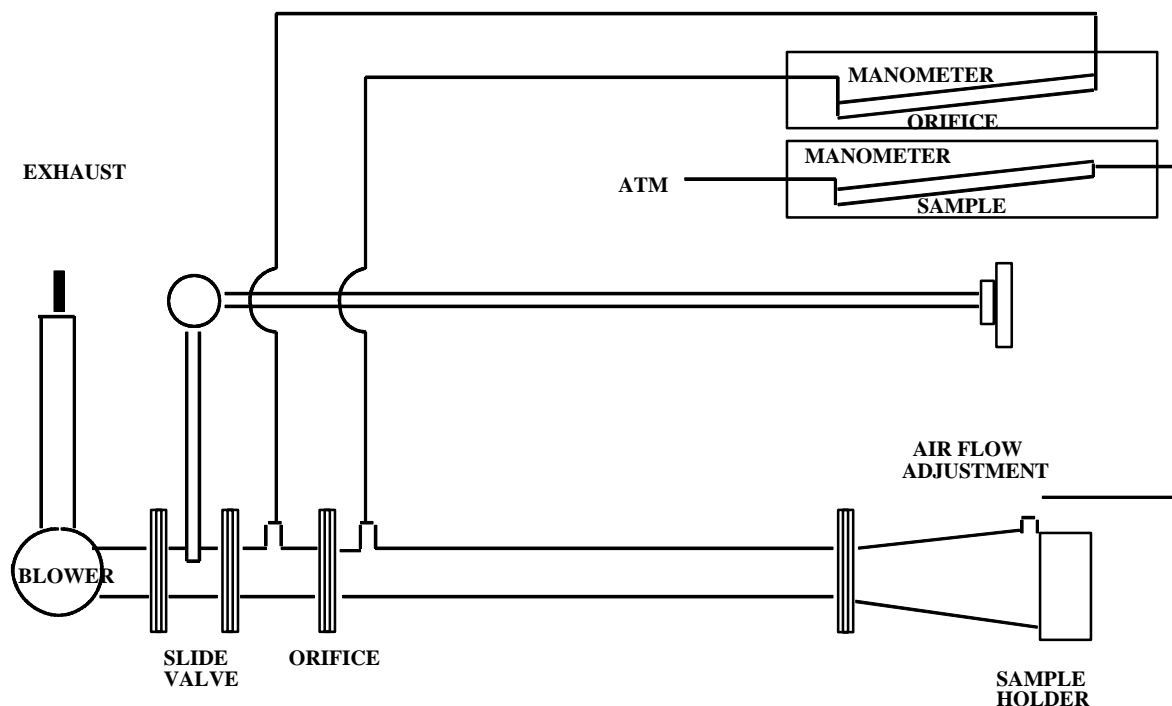


FIGURE 1. Typical porosity (air flow) test apparatus.

MIL-PRF-87260B(USAF)

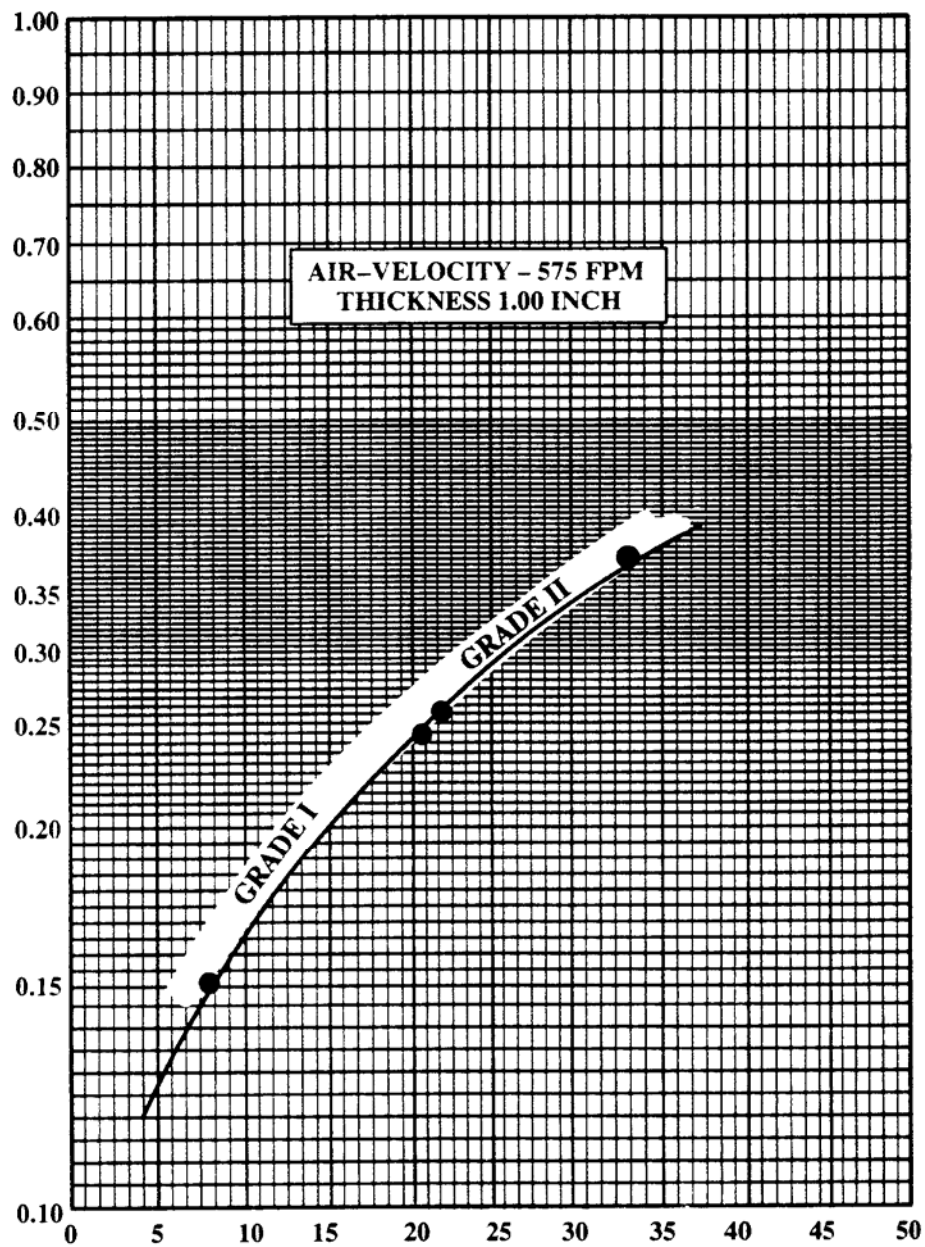


FIGURE 2. Porosity curve.

NOTE: The porosity values shown on [FIGURE 2](#) are assigned for reference only and do not necessarily relate directly to the actual number of pores per lineal inch.

MIL-PRF-87260B(USAF)

4.5.5 Tensile strength and ultimate elongation tests.

Tension tests, including tensile strength and ultimate elongation, shall be conducted in accordance with *ASTM D3574 (Test E)*, except that the initial jaw separation shall be 2.0 inches. A 2.5-inch jaw separation may be used for consistency with other products, however, data must be provided in the qualification report showing the correlation between the two grip lengths. For elongation measurements there are two approved techniques, benchmarks and crosshead travel. The rate of travel of the power actuated grip shall be 20 ± 1 inches per minute and shall be uniform at all times. The specimen die to be used shall be in accordance with *Figure 1 of ASTM D3574 entitled "Die For Stamping Tension Specimens."* The approximate size of the dry specimen shall be 5.5 inches x 1 inch x 0.50 inch thick. The specimen thickness dimension shall be between 0.500 and 0.600 inches. For all but qualification, three specimens per sample shall be tested. If any value deviates more than 20.0 percent from the average, two additional specimens shall be tested and the average of all five reported. For qualification, 10 specimens shall be tested and all values reported. In addition, a copy of the recorded traces shall be included in the test report. The tensile strength shall be reported in pounds per square inch, and ultimate elongation in percent. Tension specimens shall be taken from the upper half of the test section, and the orientation shall be such that the 1/2-inch dimension (5.5 inches) is always in the machine direction (see [6.5.4](#)). For special fuel-wet tension tests, the specimens shall be measured on the dry specimen prior to test fluid exposure and recorded for later usage. When testing the specimen, it should be removed from the fluid prior to tension tests, drained, and immediately tested. Do not allow the sample to air dry, as it will affect the test results. Original dry specimen data shall be used as a baseline for calculating the percent loss in wet tensile strength for tests in [4.5.16b](#).

4.5.6 Tear resistance test.

Three specimens shall be tested in accordance with *ASTM D3574 (Test F)* using a crosshead speed of 20 ± 1 inches per minute. Dry specimen size shall be 6 inches x 1 inch x 1 inch where the 6-inch dimension is in the machine direction and the slit cut is parallel to the direction of rise. The tear resistance shall be reported in pounds per lineal inch of thickness. Specimens shall be cut from 1 inch thick slabs of material. For fuel-wet tear resistance tests ([4.5.16b](#)), the specimen shall be tested immediately after removal from the fluid. Original dry specimen data shall be used as a baseline for calculating the percent loss in wet tear resistance.

4.5.7 Constant deflection compression set test.

Three specimens shall be tested in accordance with *Method B of ASTM D3574 (Test D)* at a 50 percent deflection. Sample size shall be 4 inches x 4 inches x 3 inches. The three specimens shall be cut from the same 3-inch thick slab located in the middle of the test section (bun height) and adjacent to the specimens used in the compression load deflection test specified in [4.5.8](#). The specimens shall be tested (compressed) in the direction of rise (3-inch dimension). Results for all specimens shall be reported in percent of original thickness.

4.5.8 Compression load deflection (CLD) test.

Three specimens shall be tested in accordance with *ASTM D3574 (Test C)* at the 25 percent and 65 percent deflection level after 1 minute at each deflection point. Specimen size shall be 4 inches x 4 inches x 3 inches and taken near the middle of the bun height such that the 3-inch dimension is in the direction of material rise. The rate of compression (deflection) shall be 2 inches per minute. New material shall be aged for a minimum of 96 hours following thermal

MIL-PRF-87260B(USAF)

reticulation prior to compression load deflection testing. Tests shall be conducted in the direction of material rise. Prior to testing, the specimens shall be pre-flexed twice to 80 percent compression. A copy of the recorded traces for each test shall be included in the test report and results reported in pounds per square inch. For wet compression load deflection properties, the test specimen shall be tested immediately after removal from the fluid. The original dry specimen data shall be used as a baseline for determining the percent loss in wet compression load deflection properties per [4.5.16b](#).

4.5.9 Fuel displacement test.

One sample per test shall be run using Grade JP-5 or JP-8 turbine fuel conforming to *MIL-DTL-5624* or *MIL-DTL-83133* respectively, and the average reported as the fuel displacement. The test shall be conducted at standard conditions using a standard 1,000 milliliters (ml) capacity cylinder having 10 to 20 ml graduations. Each specimen shall be cut into a cylindrical shape having a diameter approximately equal to that of the graduated cylinder and a length sufficient to fill the test cylinder to the 900 ml mark. Specimens shall be cut in the direction of the material rise (bun height). Fuel shall be added to the 900 ml mark in the graduated cylinder and the specimen slowly added until it is completely immersed. The specimen shall be immersed for a period of 24 hours to obtain maximum swelling effects. The new fluid level shall be noted and the increase in milliliters shall be recorded. The size of each specimen shall be measured and recorded. The displacement shall be calculated as follows:

$$\text{Percent volume displacement} = \frac{\text{milliliters increase} \times 100}{\text{original fluid volume}}$$

4.5.9.1 Theoretical fuel displacement.

The theoretical volume displacement of the material as calculated from the following formula and based on the material density specified in [4.5.3](#) shall be reported:

$$\text{Percent volume displacement} = \frac{\text{material density (lbs/ft}^3\text{)} \times 100}{\text{density of polymers (lbs/ft}^3\text{)}}$$

4.5.10 Fuel and water retention test.

4.5.10.1 Fuel retention tests.

Fuel retention shall be determined on a 6 inches x 6 inches x 6 inches specimen using Grade JP-5 or JP-8 turbine fuel conforming to *MIL-DTL-5624* or *MIL-DTL-83133* respectively, having a specific gravity of 0.788 to 0.845. For qualification testing the retention values shall be determined throughout the porosity range of the product and the data plotted as a function of air pressure drop. A minimum of three different retention specimens at three different porosity locations (lower, middle, and upper porosity ranges) shall be tested for each type of material. At each test location, four retention specimens shall be cut from the center of the respective test section (bun height) directly adjacent to each other. These shall be identified at the top surface,

MIL-PRF-87260B(USAF)

and two each shall be identified for the fuel retention test specified herein and the water retention test specified in 4.5.10.2. A porosity test specimen shall be taken directly above and below the retention specimen and tested per 4.5.4. All retention and porosity (air pressure drop) specimens shall be properly labeled on the top surface and submitted to the coordinating activity. One fuel retention specimen from each of the three test locations shall be tested in accordance with the following procedure, and all applicable data shall be recorded:

a. The specimen shall be preconditioned at a temperature of 72 ±5°F for a minimum of 30 minutes, weighed to the nearest 0.1 gram, and the dimensions measured in accordance with ASTM D3574. The test fluid shall be pre-filtered through an 0.8-micron filter as specified in 4.5.14 and then adequately preconditioned at the test temperature. Just prior to use, the fluid shall be tested for specific gravity (density) and temperature.

b. The retention test apparatus shall be sized to approximately 7 inches x 7 inches x 10 inches and shall have a means of draining the fuel from the bottom at the rate of 500 ±50 cubic centimeters/minute. The draining drop rate in this particular apparatus should approximate 0.5 inch per minute. The test fluid shall be charged into the container to a level that corresponds to approximately 0.5 inch above the top of the specimen.

c. The specimen shall then be slowly placed into the container such that the specimen is oriented in the direction of rise (bun height) and supported off the bottom of the container by two glass rods and spaced 0.5 inch from all sides of the container. Fuel shall then be drained at the prescribed rate until flow ceases and the specimen is then allowed to drain in this position for an additional 2 minutes.

d. The specimen shall then be carefully removed from the container and weighed to the nearest 0.1 gram. Care should be taken not to spill the fluid from the bottom surface of the specimen when removing from the test rig. Using the specimen weights before and after fluid-wetting in grams, specimen volume in cubic centimeters, and fuel density in grams per cubic centimeter, the percent volume retention shall be calculated as follows:

$$\text{Percent retention} = \frac{(\text{wet specimen weight} - \text{dry specimen weight}) \times 100}{\text{specimen volume} \times \text{density of fuel}}$$

e. All values, including test fluid temperature and porosity values (top/bottom) shall be reported. The above procedure shall then be repeated, using the same specimens, and report results.

4.5.10.2 Water retention test.

Using one unused water retention specimen specified in 4.5.10.1, the volume percent retention shall be determined using the same procedure. This shall be repeated for at least one other porosity location. The test fluid shall be unused distilled water that has been tested for temperature and density just prior to use. CAUTION: Do not run more than two tests per batch of water.

4.5.10.3 Test data and sample requirements.

Report test data on all samples tested including the air pressure drop of the specimens directly above and below the retention samples. A minimum of three specimens from various locations

MIL-PRF-87260B(USAF)

within the porosity range for a material shall be evaluated and a curve of fuel retention versus air pressure drop established. The data (curve) shall be extrapolated to the upper air pressure drop limit and the projected fuel retention limit shall be established and reported. A fuel retention sample for each data point along with the tested porosity samples shall be identified and submitted to the qualifying activity (see 6.3.1).

4.5.11 Extractable material test.

The extractable material test shall be conducted on one test specimen. The specimen size shall be 1 inch x 1 inch x 2 inches and cut by means of a saw or die. Preconditioning of the specimen shall include drying at 200 ±5°F for 15 minutes and cooling the specimen in a desiccator for a minimum of 30 minutes. Immediately following the preconditioning, the specimen shall be weighed to the nearest 0.10 milligram. The specimen shall then be placed in a 60 ml volume Soxhlet extraction tube which is connected to a water-jacketed condenser. Several standard boiling stones and 125 ml of Type III test fluid conforming to SAE AMS-2629 shall be added to a 250 ml Florence flask and the flask attached to the extraction tube. The heating unit shall be activated and the fluid allowed to reflux for a period of 3 hours. Following reflux, the specimen shall be removed, dried at 200 ±5°F for 15 minutes, cooled in a desiccator for 30 minutes and then weighed. The percentage of extractable material shall be calculated as follows:

$$\text{Percent extractable} = \frac{(\text{original specimen weight} - \text{final weight}) \times 100}{\text{original specimen weight}}$$

4.5.12 Volume swell test.

One specimen for each test fluid shall be tested for volume changes after immersion for 7 days at 72 ±5°F in Type I test fluid conforming to SAE AMS-2629, Type III test fluid conforming to SAE AMS-2629, and Grade JP-5 conforming to MIL-DTL-5624, and JP-8 turbine fuel conforming to MIL-DTL-83133. Sample size shall be 6 inches x 6 inches x 6 inches. The samples shall be taken from the same approximate location in the test section as the retention test specimens specified in 4.5.10. Dry and wet measurements shall be made on the test specimens in accordance with ASTM D3574. Following immersion, the specimens shall be removed and immediately measured wet for the final volume. All values for the specimen including original and wet volumes shall be reported and the percent volume increase from the original and wet measurements shall be calculated.

4.5.13 Low temperature flexibility test.

Three specimens, 2 inches x 1/2 inch x 12 inches shall be preconditioned in air along with a 3-inch diameter rod to a temperature of -55 ±5°F for 1 hour. Each specimen shall be cut such that the 12-inch dimension is in the machine direction. At the end of the conditioning period and without removing the specimens from the chamber, each specimen shall be bent around the rod. Any evidence of breaking or cracking of foam strands shall be cause for failure.

MIL-PRF-87260B(USAF)

4.5.14 Entrained solid contamination tests.

Solid contamination tests shall be conducted on a hot-wire-cut cylindrical specimen having dimensions of 9.25 inches in diameter and 8 inches in height. The 8-inch dimension shall be cut in the direction of rise (bun height). For material having more than 8 inches in bun height, the specimen shall be taken from the lower portion of the test section. Testing shall be conducted using a U.S. Testing Company model 6523 dry cleaning machine or equivalent, having a tumbler rotation speed of 45 revolutions per minute (rpm). The specimen shall be positioned in the center of the tumbler. The test cycle shall be 5 minutes using a 4-liter charge of Type I fluid conforming to *SAE AMS-2629* which has been pre-filtered through a 0.8-micron Millipore Filter Corporation filter, or equivalent. Upon completion of the test cycle, the specimen shall be positioned slightly above the fluid level and allowed to drain for 5 minutes prior to removal. The test fluid shall then be tested for level of solid contamination in accordance with *ASTM D2276, Appendix A3* (laboratory filtration) or *T.O. 42B-1-1, sections 5-23, 5-24, and 5-47 through 5-52*. Following filtration of the test fluid and just prior to removal of the filter pad from the apparatus, the filter and contamination shall be neutralized of static charge with an Amstat Industries Model 2U500 air deionizer, or equivalent (see [6.10](#)). This step reduces the loss of particles from the filter pad during transfer to the drying oven. Each filter used shall be dried at $200 \pm 5^\circ\text{F}$ for a minimum of 15 minutes and then cooled for a minimum of 15 minutes. A minimum of one control filter shall be run for each set of samples. Test results shall be reported in milligrams per cubic foot of material.

4.5.15 Steam autoclave exposure test.

Testing shall be conducted in accordance with *ASTM D3574 (Test J)* steam autoclave test for 10 hours at $250 \pm 5^\circ\text{F}$. Tension tests as specified in [4.5.5](#) shall be conducted on five control specimens and five exposed specimens. Prior to testing, exposed specimens shall be post-dried for 30 minutes at $160 \pm 5^\circ\text{F}$ and then cooled at room temperature for 30 minutes. The results for tensile strength and elongation shall be reported before and after exposure and the average percent change in tensile strength calculated.

4.5.16 Fluid immersion tests.

Fluid exposure tests shall be conducted on the baffle material under the following exposure conditions:

- a. Grade JP-8 turbine fuel conforming to *MIL-DTL-83133* for 12 weeks at $160 \pm 5^\circ\text{F}$. Tension and Electrical Resistivity Specimens, per [4.5.5](#) and [4.5.4](#) respectively, shall also be exposed to the test environment.
- b. Grade JP-8 turbine fuel conforming to *MIL-DTL-83133* for 4 weeks at $72 \pm 5^\circ\text{F}$. To determine the wet property values for the material, tension, CLD, and tear resistance specimens, per [4.5.5](#), [4.5.8](#), and [4.5.6](#) respectively, shall be exposed to the test environment, and then tested fuel-wetted for all properties.

4.5.16.1 Tension tests.

Tests in accordance with [4.5.16a](#) shall be conducted in loosely capped standard quart jars using approximately 900 ± 25 ml of fluid for each nine specimens for each sampling frequency. Specimens shall be taken from the upper half of the test section as specified in [4.5.5](#). The testing frequency for each condition shall be every 4 weeks. The JP-8 fuel shall be changed at each 4-week interval. Testing shall include three specimens dry tension-tested as specified in

MIL-PRF-87260B(USAF)

4.5.5, three specimens tested in accordance with the steam autoclave exposure test as specified in 4.5.15, and three additional specimens shall be tension-tested while fuel-wetted in accordance with 4.5.5. Prior to dry tension and steam autoclave testing, the specimens shall be rinsed in petroleum ether, dried for 30 minutes at $160 \pm 5^\circ\text{F}$, and cooled at room temperature. Wet tension tests shall be run immediately after removal from the test fluid. All values for original, final (dry and wet), and percent change of tension properties shall be reported. The wet tensile property change shall be calculated using the initial (time = 0) wet tension values as a baseline.

4.5.16.2 Electrical resistivity test.

In addition to the tension tests of 4.5.16.1 electrical resistivity tests shall be conducted using specimens defined in 4.5.4. A total of six specimens shall be exposed to the fuel and at each 2-week interval a specimen shall be removed, washed in petroleum ether, dried, and then tested per 4.5.22. The specimens shall also be tested initially, prior to immersion into the fuel. At each 4-week interval the fuel shall be changed. All values for original and final electrical resistivity shall be reported along with the test conditions (temperature and humidity) that existed during the electrical resistivity tests. The electrical resistivity tests shall be in accordance with the procedures for production specimens defined in 4.5.22.

4.5.16.3 Immersion tests.

Fuel (JP-8) immersion tests in accordance with 4.5.16b shall be accomplished in containers sufficient to provide complete immersion for all tension, CLD, and tear specimens. Following the 4-week exposure at $72 \pm 5^\circ\text{F}$ (room temperature), three specimens shall be immediately tested for wet tension properties in accordance with 4.5.5, wet CLD properties in accordance with 4.5.8, and wet tear resistance properties in accordance with 4.5.6. All values and averages shall be reported and the percent change from the original dry test properties shall be calculated. The original dry property data determined in 4.5.5, 4.5.8, and 4.5.6 should be used to establish the percent change in properties due to fuel-wetting.

4.5.17 Hydrolytic stability tests.

The following tests shall be conducted to characterize the hydrolytic stability of the material:

4.5.17.1 Humidity exposure.

Tension and electrical resistivity specimens shall be exposed to $200 \pm 5^\circ\text{F}$ and 95 ± 5 percent relative humidity for 12 weeks or until failure, whichever occurs first.

4.5.17.1.1 Tension tests.

Tension tests shall be conducted in loosely capped standard glass quart jars using 50 ml of distilled water for each 900 ml of container volume and a maximum of nine tension specimens for each sampling jar. Specimens shall be supported above the water and the water level shall be maintained throughout the test. A minimum of three specimens for each exposure time shall be tensile tested. In addition, electrical resistivity tests shall be conducted using specimens defined in 4.5.4. Sufficient specimens shall be used to allow samples to be removed at least once a week in order to determine the failure point for the conductive treatment or coating. All resistivity specimens shall be measured initially (prior to exposure) to establish property changes due to testing. The test specimens shall be removed, drained free of water, blotted dry,

MIL-PRF-87260B(USAF)

and then air dried at $160 \pm 5^\circ\text{F}$ for a minimum of 30 minutes prior to any preconditioning for resistivity evaluations. The resistivity specimens shall then be tested (following preconditioning) in accordance with [4.5.22](#). Water used in the containers shall be maintained and changed on a weekly basis. All values for original and final electrical resistivity shall be reported along with the test conditions (temperature/percent relative humidity) that the specimens were exposed to during the resistivity measurements.

4.5.17.2 Water immersion.

Tension specimens shall be immersed in pure distilled water at $160 \pm 5^\circ\text{F}$ for 12 weeks or until failure.

4.5.17.2.1 Testing frequency.

Testing frequency and water changes shall be at 4-week intervals. Three specimens shall be tested for each sampling frequency. Testing shall be conducted in loosely capped jars using 900 ± 25 ml of water for each of the nine tension specimens. Water used in the containers shall be changed on a weekly basis.

4.5.17.3 Dry heat.

Tension and electrical resistivity specimens shall be exposed to dry heat at $250 \pm 5^\circ\text{F}$ for 8 weeks or until failure.

4.5.17.3.1 Testing frequency.

The testing frequency shall be at 2-week intervals. A minimum of three tension specimens shall be tested for each sampling frequency. Additional specimens shall be exposed for electrical resistivity evaluations. The specimens shall be in accordance with [4.5.4](#) and shall be tested, as a minimum, on a weekly basis. Prior to the start of the test, the electrical resistivity specimens shall be tested and the test conditions recorded per [4.5.22](#). All test data shall be reported including test conditions (temperature/relative humidity) for the electrical resistivity evaluations.

4.5.17.4 Data requirements.

All tension and electrical resistivity test data obtained under [4.5.17](#) shall be reported per [4.5.5](#) as well as percent loss in tensile strength. In addition, the tensile strength and electrical resistivity values shall be plotted as a function of exposure time.

4.5.18 Explosion suppression and flame arrestor characteristics.

The explosion suppression and flame arrestor characteristics of the material shall be defined using a small scale flame tube type apparatus having a minimum total volume of 5 cubic feet and a 100 square inch cross-sectional area. The following parameters shall be satisfied in all the testing:

- a. Stoichiometric propane/air mixture (4.5 to 5.2 volume percent propane) verified by bomb sampling
- b. Spark ignition source having a minimum of 0.25 millijoules energy
- c. Dry arrestor material

MIL-PRF-87260B(USAF)

d. Minimum instrumentation shall include: pressure rise, combustion temperature indication, and visual, photographic, or photocell indication of flame penetration downstream of arrestor.

e. Combustion relief area shall be 80.0 percent of cross-sectional area or greater. The material used for the testing shall be taken from a given bun that has been sufficiently tested to establish its air pressure drop (porosity) and density characteristics. The material shall always be oriented in the test apparatus to permit flame penetration in the direction of porosity testing (direction of rise or bun height).

f. Where practical, the material used shall be in the lower half of the air pressure drop range. For example: Grade I shall be between 0.150 and 0.200 inch of water and Grade II shall be between 0.260 and 0.310 inch of water.

4.5.18.1 Material sizing.

The material shall always be slightly oversized, 2.0 percent maximum, when installed and restraints used to avoid arrestor movement during testing. The combustible mixture on each test shall be verified by bomb sampling and shall meet the following minimum criteria for pressure rise:

$$P_{min} = (8P_0) \times 0.7, \text{ where } P_0 = \text{initial pressure of system in psia}$$

The following definitions shall apply (see [FIGURE 3](#)).

psid = differential pressure rise starting point maximum overpressure point during the combustion process.

V_a = arrestor volume

V_c = combustion (ignition volume)

V_r = relief volume = *V_a* + *V_v*

V_t = total volume of apparatus = *V_c* + *V_r*

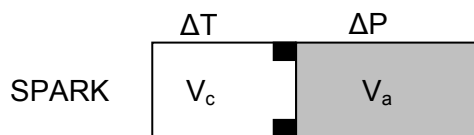
V_v = void volume downstream of arrestor

T_m = minimum arrestor thickness required to prevent flame propagation *V_c* to *V_v*

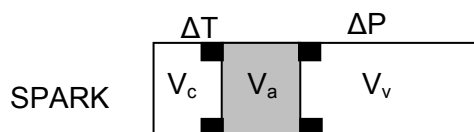
4.5.18.2 Additional tests.

The following testing shall be conducted and all data shall be reported for each test condition (see [FIGURE 3](#) for typical flame arrestor orientation).

MIL-PRF-87260B(USAF)



a. Typical set-up for single void ignitions.



b. Typical set-up for arrestor thickness tests.

FIGURE 3. Flame arrestor apparatus**4.5.18.2.1 Single void ignitions.**

Single void ignitions shall be conducted at 3 psig (17.7 pounds per square inch absolute (psia)) initial pressure with the following minimum number of percent combustion volumes (percent V_v) of:

- a. Grade I: 10, 15, 20, and 30 volume percent
- b. Grade II: 20, 30, 35, and 40 volume percent

A minimum of two tests shall be conducted for a given condition and all data such as bomb sample and system pressure rise, test temperature, extent and location of arrestor damage, and any other related information shall be submitted to the coordinating activity. A plot of pressure rise (psid) versus (percent) combustion volume shall be submitted for each initial pressure condition. Repeat test may be conducted on the material provided the damaged (burned) arrestor is replaced after each test. All tests shall be conducted at standard laboratory conditions.

4.5.18.2.2 For Grade II (fine pore material only).

Determine the amount of arrestor (thickness) required to prevent flame propagation from V_c to V_v when the combustion volume (V_c) is 9.1 and 16.7 volume percent and the initial pressures are 0 and 3 psig. Testing shall be conducted at 1 inch (thickness) intervals until the minimum arrestor thickness (T_m) is determined. Then a minimum of two tests shall be conducted to verify the minimum thickness. All data including system and bomb pressure rise, test temperatures, extent and location of arrestor damage, and any other related information shall be reported. The ratio of arrestor volume to combustion volume (V_a/V_c) shall be calculated for each minimum arrestor thickness (T_m).

MIL-PRF-87260B(USAF)

4.5.19 Corrosion and adhesion test.

One 4-inch by 3-inch specimen cut such that the 4-inch dimension is in the direction of material rise shall be exposed in contact with 7075 aluminum alloy per *SAE AMS-QQ-A-250* or *ASTM B209* for 14 days at $75 \pm 5^\circ\text{F}$ (room temperature) and 95 ± 5 percent relative humidity. Color photographs shall be taken of all specimens and plates both before and after exposure.

4.5.19.1 Roughness test.

Three sets of metallic plates shall be used having a surface finish of 5-15 microinches obtained by lapping. The roughness shall be determined by a profilometer or equivalent instrument. The roughness reading is the arithmetical average of the deviations in the surface expressed in microinches measured normal to the surface. For these tests the surface finish should be prepared per [4.5.19.1.1](#).

4.5.19.1.1 Surface finish.

A surface finish of 5-15 microinches measured perpendicular to the lay at a roughness-width cutoff rating of 0.030 inch and a maximum roughness-width rating of 0.015 inch. One set shall be clamped together with the baffle material specimen such that the baffle material is compressed from 4 to 3 inches in thickness in contact with the polished surfaces. This set along with one set of extra plates (controls) shall then be exposed for 14 days at room temperature and 95 ± 5 percent relative humidity in a sealed container or humidity cabinet. In addition, the third set of plates shall be used to run a comparison test on the vendor's basic blue polyether foam having the same porosity as the conductive material. Test specimens should be oriented such that the metal plates are vertical in order to minimize moisture condensation and pooling on the plate surfaces. At the termination of the test, there shall be no adhesion of the baffle material to the metal plates, nor any evidence of pitting, erosion, corrosion, or bad discoloration as a result of the material contact, as determined by the following procedures. The basis for the comparison shall be the exposed set of control plates.

a. The surfaces of the plates that were in contact with the material shall be inspected for such things as discoloration, deposits, and pitting. If any of these conditions exists, the surface of the plates shall be washed in precipitation naphtha. Deposits determined as urethane materials or elements, which can be removed by this process, shall be construed as adhesion.

b. If any other marks remain on the surface of the plates after being washed in precipitation naphtha as specified in [4.5.19.1.1a](#), the surfaces shall be lightly polished with a nonabrasive cloth buff. Any pits or eroded marks remaining after this process shall be construed to be corrosion. Discoloration or staining (marks that do not physically affect the surface of the plates and that easily wash or buff off) shall not be considered detrimental.

4.5.19.2 Test report.

All test data, including the photographs, shall be included in the test report. In addition, the test plates shall also be submitted to the qualifying activity.

MIL-PRF-87260B(USAF)

4.5.20 Electrical resistivity permanence.**4.5.20.1 Water immersion.**

Electrical resistivity specimens, per 4.5.4, shall be immersed in pure distilled water at $120 \pm 5^\circ\text{F}$ for 4 weeks or until failure. A 3-week exposure at $160 \pm 5^\circ\text{F}$ may be substituted for this temperature (see 4.5.17.2).

4.5.20.1.1 Electrical resistivity specimens.

Sufficient electrical resistivity specimens shall be included to allow for specimen removal at least once a week. The specimens shall be tested for initial resistivity prior to water exposure to establish a baseline. At least one specimen shall be removed from the test environment once weekly, and upon removal it shall be dried at $160 \pm 5^\circ\text{F}$ until dry and then preconditioned to the laboratory conditions for at least 30 minutes. Following preconditioning, the specimen shall be tested per 4.5.22 and reinserted into the test environment. The distilled water shall be replaced on a weekly basis. All initial and final electrical resistivities shall be reported, as well as all test conditions (temperature and humidity) during resistivity measurements.

4.5.21 Fuel compatibility test.

MIL-DTL-83133 (Grade JP-8) fuels shall be exposed to the conductive material for 4 weeks at $72 \pm 5^\circ\text{F}$ and then tested for the properties listed below. The fuels shall also be tested prior to the exposure as indicated. The ratio of conductive material to fuel shall be approximately 1:1. The foam color/pigment change (loss) shall be documented following exposure to verify the requirement of section 3.4.

Fuel property	Control Test	After Exposure
a. Saybolt color	YES	YES
b. Total acid number	YES	YES
c. Aromatics and olefins	YES	NO
d. Mercaptan sulfur and total	YES	NO
e. Distillation	YES	NO
f. Density	YES	NO
g. Copper strip corrosion	YES	YES
h. Thermal stability	YES	YES
i. Existent gum	YES	YES
j. Filtration time	YES	YES
k. Water reaction	YES	YES
l. Water separation (may use MicroSep or MiniSonic in place of WSIM)	YES	YES
m. Fuel electrical conductivity	YES	YES

MIL-PRF-87260B(USAF)

4.5.22 Electrical resistivity test.

All volume resistivity measurements of the conductive material shall be conducted in accordance with the basic requirements of *ASTM D257*. A standard set of electrodes shall be used for all measurements and shall be constructed from stainless steel in accordance with the guidelines of *FIGURE 4*. The measurements shall be made on qualification and production specimens per *4.5.4*, and on full size buns through the bun thickness. For qualification, the resistivity and resistivity uniformity shall be established by evaluating a standard bun of the finished material at several locations (four minimum) by using production specimens, per *4.5.4*, from various positions in the bun (top, middle, bottom of the bun height). For production tests, the resistivity specimens shall be taken from the top, middle, and bottom 3 inches of the test section (per *4.5.4*). Qualification and production specimens shall be tested at standard laboratory conditions. The full size bun measurements may be conducted in the manufacturing area after final trimming has been accomplished, with temperature and humidity recorded. These measurements shall be conducted without cutting the bun, but by measuring the resistance through the bun thickness. The electrical resistivity values shall then be extrapolated to standard conditions (72 ±5°F, 55 ±5 percent relative humidity). The extrapolation shall be accomplished with a calibration curve(s) for the product as a function of relative humidity (see *4.5.22.3* for the required correlation data).

4.5.22.1 Test equipment and preconditioning environment.

A suggested electrical resistivity test set-up employs a Beckman Megohmmeter Model L-8 (or equivalent) with a variable resistance (10⁶ to 10¹³ ohms) and a variable supply voltage (1-1000 volts), in combination with the stainless steel electrodes on *FIGURE 4*. The production test specimen size shall be the same as that for air pressure drop (see *4.5.4*) and shall be taken from the same locations (top, middle, bottom). One set may be used for both air pressure drop and electrical resistivity. If the resistance of the 1-inch test specimen is below the lower limit (10⁶ ohms) of the megohmmeter being used, it may be necessary to use more than one specimen. By stacking two or three 1-inch thick specimens together a value of resistance may be obtained that can be measured on the meter. Production tests shall be conducted in a laboratory environment following a minimum of 1 hour of preconditioning in the test environment. For full size bun testing, the bun should also be preconditioned for at least 1 hour prior to testing, and the temperature and humidity in the production test location recorded along with the measured resistance and calculated resistivity.

4.5.22.2 Test procedure and calculation.

The test equipment (electrodes/megohmmeter) shall be set up as shown on *FIGURE 4* with the test sample (specimen or bun) placed between the electrodes and the electrodes centered directly over each other to produce a vertical "field" between the plates. The megohmmeter voltage shall be set to 500 volts and the variable resistance adjusted until a steady-state resistance reading is obtained on the meter. Allow the meter to stabilize for 1 minute and record the resistance for the specimen along with the temperature, relative humidity, and the thickness of the specimen. Calculate the sample volume resistivity as follows:

$$ev = \frac{\text{measured resistance (ohms)} \times 155.7 \text{ (cm}^2\text{)}}{\text{sample thickness (inches)} \times 2.54 \text{ (cm/in)}}$$

MIL-PRF-87260B(USAF)

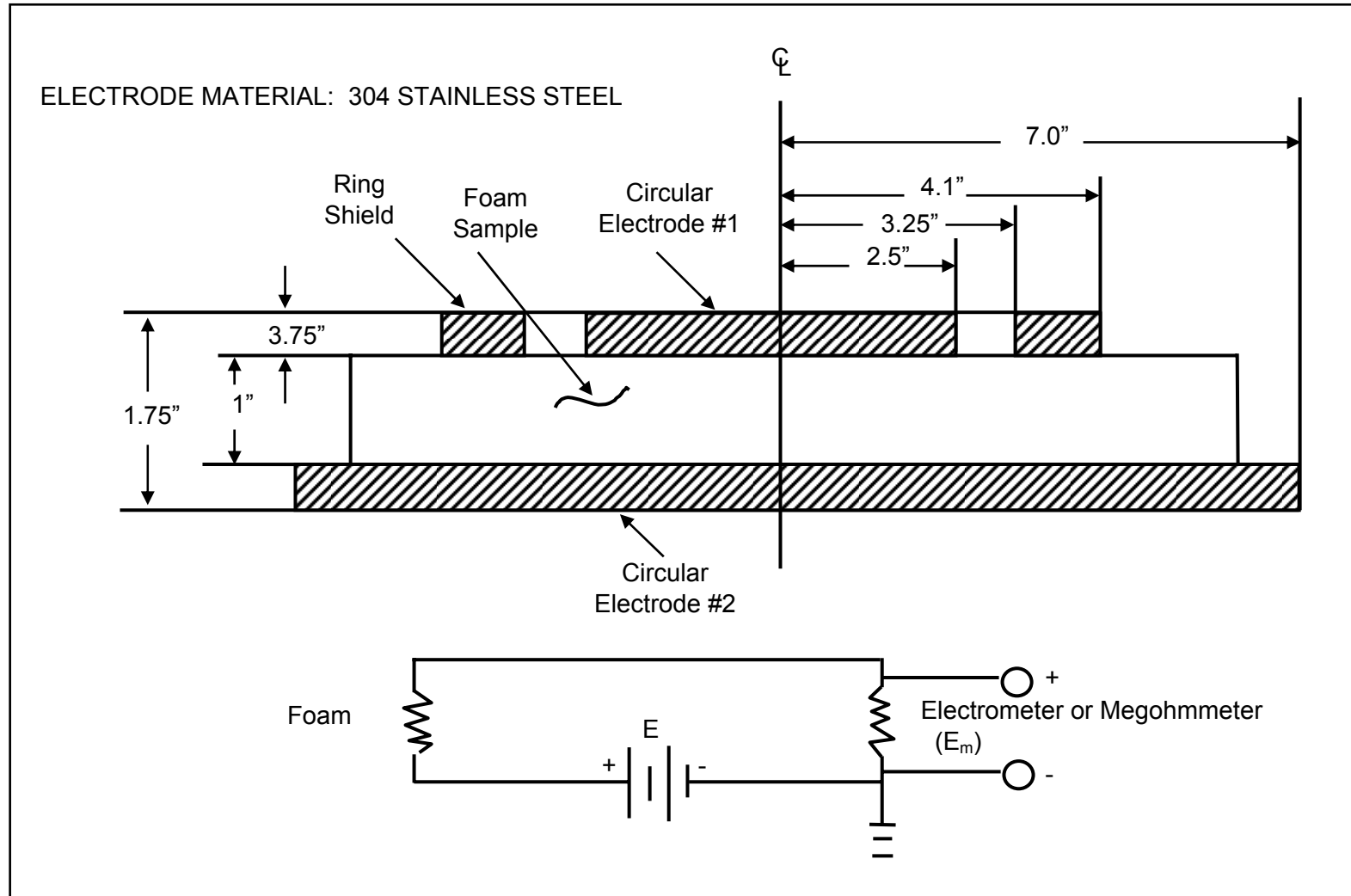


FIGURE 4. Dimensional and electrical diagrams for measuring volume resistivity of conductive materials.

MIL-PRF-87260B(USAF)

4.5.22.3 Correlation test data requirements.

A correlation study shall be conducted for the conductive material(s) to show the effect of temperature and humidity on the volume resistance and resistivity using the procedure and equipment in 4.5.22.1 and 4.5.22.2. As a minimum, the test data should include tests at 72 ±5°F and relative humidity of 10, 50, and 80 ±5 percent, and material thickness of 1, 4, 8, and 12 inches. In addition, electrical resistivity tests shall be run on 1-inch thick specimens at 140, 60, 32, 10, 0, -15, -25, -30 and -40 ±5°F and all data reported including relative humidity conditions (if possible). Record all values for resistance, resistivity, temperature, humidity, and material sample overall dimensions (if other than that per 4.5.4). If the conductive material is found to vary more than two orders of magnitude as a result of the humidity changes, a correlation curve or extrapolation factor should be developed for use on all production bun tests where the test conditions will likely be other than 72 ±5°F and 50 ±5 percent humidity. The correction will enable the qualifying activity to know the resistivity value at standard laboratory conditions (72°F and 50 percent relative humidity). A plot of resistivity versus temperature shall also be generated to demonstrate the material's resistivity properties throughout the anticipated end use temperature range.

4.5.23 Electrostatic fuel impingement test.

A fuel impingement test shall be conducted on the conductive material to demonstrate its electrostatic compatibility. The testing shall be done in a small scale rig (55-gallon drum or equivalent). The basic test requirements to be met include:

- a. The test tank shall be filled with the conductive material except for a 1 to 2-inch gap (ullage) at the nozzle inlet area.
- b. Fuel requirements: *MIL-DTL-83133* Grade JP-8 having the following electrical conductivities shall be used for the evaluations: 0 to 10, 50, 100, 200, 500, and 800 cu. The base fuel should be from the same batch and the conductivity level varied through the use of static dissipation additives. The approved additives are listed in *MIL-DTL-83133*. Information relating to mixing of the additives into fuel can be found in *T.O. 42B-1-1*.
- c. Test temperatures: Fuel test temperatures shall include, but not be limited to, 140, 60, 30, 10, 0, -15, -25, -30 and -40 ±2°F. The test fuel temperature is defined as the starting test temperature, but is not necessarily the fuel temperature at the test end point.
- d. The test tank shall be fitted with a straight 1-inch diameter inlet nozzle. The flow rates simulated shall be up to 150 gallons per minute (gpm) with an approximate velocity as high as 61 feet per second (fps). The flow rate/velocity shall then be systematically adjusted until the associated discharge activity appears and/or ceases. The critical flow rate/velocity required for discharge activity shall be identified and compared to the candidate material's critical flow rate/velocity for a given standardized fuel. The absence of discharge activity, at the 150 gpm flow rate, may require the straight inlet nozzle's diameter be reduced to allow higher fuel velocities. Any changes in the fuel's electrical conductivity shall be noted and recorded. In addition to discharge activity, charge transfer levels, associated field strength, and induced voltages on an isolated conductor shall be measured.
- e. Baseline fuel impingement tests shall also be conducted on the *MIL-DTL-83054* blue (Type IV or V) using the same batch of fuel that is used on the candidate material. This test data will be used for comparison to the data for the candidate conductive material.

MIL-PRF-87260B(USAF)

f. Appropriate *MIL-DTL-83133* fuel property tests (including existent gum, particulate matter, and filtration time) shall be performed periodically to aid in identifying test facility contamination. Existent gum test should also be run periodically to control the charge tendency of the fuel that may be affected by the extractable from the candidate materials.

g. A suggested test setup for the 55-gallon test rig is shown in [APPENDIX A](#) of this specification.

4.5.23.1 Test data requirements.

The critical flow rates associated with the presence of discharges from flow charge generation shall be reported for the candidate material and the blue and orange baseline materials. The associated field strengths, induced voltages, charge transfer levels, and variances in fuel conductivity shall also be reported. In addition, any vapor ignitions that occur shall be documented and the burned material retained for examination.

4.5.24 Infrared spectrum analysis test.

The conductive material shall be characterized (identified) by an infrared spectrometer using a frustrated multiple internal reflectance (FMIR) technique. The spectrum shall be of such detail as to clearly distinguish it from the baseline polyether material. A reference polyether spectrum shall be included for comparison. The following criteria shall be satisfied where applicable:

a. The baseline of the spectrum determined at 5 microns wavelength shall be a minimum of 95 percent transmittance.

b. The scan speed shall be such as to obtain optimum resolution.

c. A 45° KRS-5 prism, Perkin-Elmer Corporation P/N 186-1595, or equivalent, and an FMIR attachment, Perkin-Elmer Corporation P/N 186-0174, or equivalent, shall be used to maintain the specimen for analysis. All equipment used as well as details of the test procedure and instrument settings (scan speed, slot setting, etc.) shall be identified for future reference.

4.5.25 Electrochemical corrosion test.

This test is intended to identify foam materials that may act as an electrolyte in high humidity environments and in the presence of dissimilar metals. The electrochemical corrosion potential of the conductive material shall be evaluated by placing it in contact with dissimilar metals at 120 ±5°F and 100 percent relative humidity for 7 days. Measurements of current flow and corrosion/adhesion shall be made at specified, periodic intervals. For comparison, a similar set of cells, using the vendor's baseline blue polyether (*MIL-DTL-83054*) shall also be tested.

4.5.25.1 Testing requirements.

Testing shall be conducted in a condensing humidity cabinet (bath) at 120 ±5°F for 7 days with measurements of initial current flow (after approximately 1 hour of exposure) and then at 1, 3, 5, and 7 days. Visual inspection and photographing of the plates for corrosion and adhesion shall be performed initially and then at 7 days. Test equipment required for the current flow measurements shall include a Keithley Electrometer Model 616, or equivalent. The meter used shall be set to the 10⁻⁶ ampere scale. Bonding of the test panels shall be with the standard number 18 or 20 American wire gage (AWG) wire attached to each plate with "alligator" clamps or with copper-nickel print (adhesive) and epoxy overcoat. The test cells shall consist of a "sandwich" of two standard stock finished metal panels (6 inches x 3 inches x stock thickness),

MIL-PRF-87260B(USAF)

and the conductive material (6 inches x 3 inches x 1 inch). The comparison cells shall have blue polyether foam of similar porosity substituted in place of the conductive material. The metals shall conform to the following specifications:

- a. 305 stainless steel per *SAE AMS-QQ-P-35* or *ASTM A967*.
- b. Bare 7075 aluminum per *SAE AMS-QQ-A-250* or *ASTM B209*.
- c. Bare 2024 aluminum per *SAE AMS-QQ-A-250* or *ASTM B209*.

The test cell set-up and bonding schematic is shown in [FIGURE 5](#).

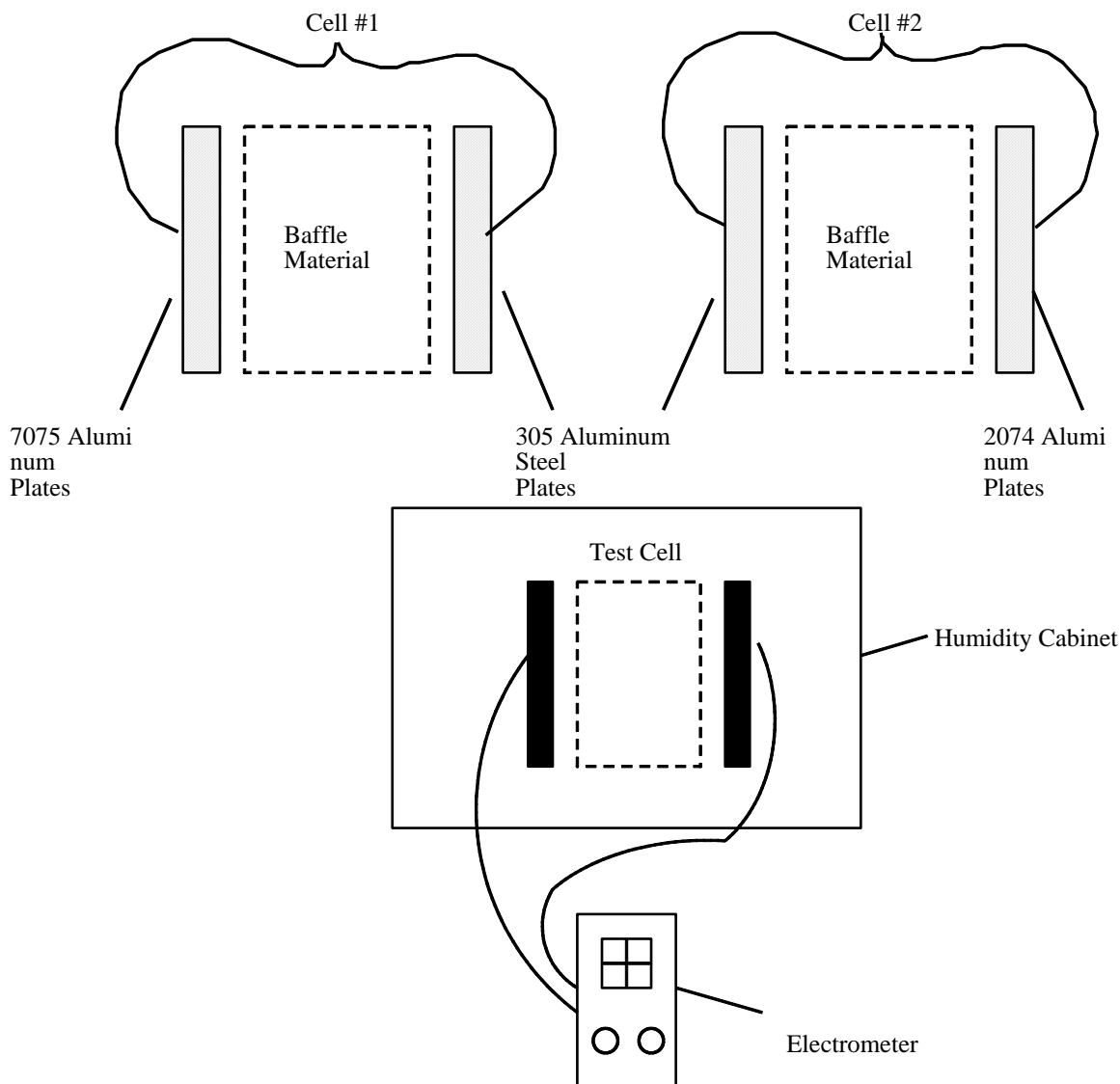


FIGURE 5. Electromechanical test set-up.

MIL-PRF-87260B(USAF)

4.5.25.2 Test metal panels.

Test metal panels shall be pre-cleaned and photographed prior to the test initiation. The pre-cleaning shall consist of the following steps:

- a. Degrease the plates in a trichloroethane vapor pit.
- b. Soak the plates in *MIL-PRF-87937* Type I alkaline detergent (10:1 solution) for 15-20 minutes until water “breaks-free.”
- c. Wash (deoxidize) in alcoholic/phosphoric deoxidizer for 2 minutes at room temperature per *ASTM F1110*.
- d. Wash in running tap water.
- e. Rinse with distilled water.
- f. Force dry with air or nitrogen.

4.5.25.3 Test exposure.

The test exposure shall be accomplished with the plates initially dry. The cells shall be exposed in the vertical position to allow for the drainage of condensing water. The cells may be held together with nylon string or rubber bands to maintain constant contact between metal and foam. The current flow measurements should be made while the specimens are at equilibrium in the humidity cabinet. This requires that the bonding wires be run external to the cabinet for hook-up to the electrometer.

4.5.25.4 Test intervals.

At seven intervals the plates shall be inspected and photographed for any adhesion, corrosion, erosion, pitting, and discoloration or staining. The following procedure shall be used to evaluate the plates:

- a. The surfaces of the plates that were in contact with the foam material shall be inspected for discoloration, deposits, and pitting. If any of these conditions exist, the surface of the plates shall be washed in precipitation naphtha. Deposits determined as urethane materials or elements of the baffle material, which can be removed by this process, shall be construed as adhesion.
- b. If any other marks remain on the surface of the plates after being washed in precipitation naphtha, the surface shall be lightly polished with a nonabrasive cloth buff. Any pits or eroded marks remaining after this process shall be construed as corrosion. Discoloration or staining (marks that do not physically affect the surface of the plates and that easily wash or buff off) shall not be considered detrimental.

4.5.25.5 Qualification report.

All test results and photographs shall be included in the qualification report. In addition, the test specimens shall be forwarded to the qualifying activity for review and inspection.

4.5.26 Marker legibility test.

A marker legibility test shall be conducted to identify a suitable marker that can be used for individual foam kit component identification before installation into the aircraft fuel system. The marker shall be fuel compatible and shall be of a legible color (either white or black) with

MIL-PRF-87260B(USAF)

respect to the candidate material color. The test may be conducted in conjunction with the JP-8 fuel immersion test. A resistivity specimen, per 4.5.4, shall be marked with the selected marking pen, and then exposed to the 160°F JP-8 immersion environment for 4 weeks. The identification marking used shall be "590-51B" and shall be 1/2 inch to 2 inches in height. Color photographs shall be taken of the specimens both before and after exposure. The exposed specimens and the photographs shall be submitted to the qualifying activity for evaluation.

4.5.26.1 Test report.

The test data and photographs shall be included in the qualification report. In addition, all specimens tested shall be submitted to the qualifying activity.

4.5.27 Electrostatic charge dissipation test.

The material shall be evaluated for its ability to dissipate static charges as a function of time. The charge dissipation test measures the time required for a test specimen to dissipate a 5000 volt charge to 10 percent of the initial value (500 volts). The static decay times of the material(s) shall be determined at temperatures of -25, 10, and 72°F.

4.5.27.1 Test equipment.

The test equipment necessary to determine the status decay properties should include the following:

- a. A static generator (voltage generator capable of supplying 5000 volts) used to apply the charge to the foam sample under test.
- b. Isoprobe non-contact voltmeter (used to monitor the voltage on the sample under test).
- c. Environmental chamber, capable of controlling humidity down to 10 percent and temperatures from -40 to +80°F.

4.5.27.2 Specimen preparation, conditioning and mounting.

- a. Test specimen shall be 18.0 inches x 18.0 inches x 1.0 inch ± 0.1 inch. The 1 inch dimension shall be cut along the direction of foam rise.
- b. Prior to testing, specimens shall be conditioned at least 24 hours in an atmosphere uniformly maintained at room temperature and a relative humidity of 12 ± 5 percent.
- c. Each specimen, when tested, shall be mounted vertically between the electrodes and the wing nuts shall be tightened in such a manner to insure intimate contact with the electrode surfaces without causing visible distortion or compression of the specimen.
- d. For qualification, the static decay and static decay uniformity shall be established by evaluating a standard bun of the finished material at several locations (four minimum), and from various positions in the bun (top, middle, and bottom of the bun height).

4.5.27.3 Specimen test procedure.

The following procedure shall be used to determine the voltage decay times at each of the specified temperature conditions:

MIL-PRF-87260B(USAF)

- a. Place the specimen in an environmental chamber hung from an insulation support with a volume resistance greater than 1×10^{16} ohms.
- b. Attach the voltage lead to the specimen using a stainless steel clamp with a minimum surface area of 2 inches square on each side of the specimen.
- c. Place the non-contact voltmeter within 3/8 to 1/2 inch of the specimen at least 8 inches from the clamp.
- d. Allow test chamber to condition samples for a minimum of 24 hours at test conditions.
- e. Apply the voltage to the foam using the static generator.
- f. Monitor the voltage on the non-contact voltmeter, when 5000 volts is achieved, remove the voltage source and apply a ground to the clamp.
- g. Measure the time required for the voltage to dissipate from 5000 volts to 500 volts. A chart recorder connected to the external output of the non-contact voltmeter can be used to accurately measure the dissipation time.

4.5.27.4 Report.

Report sample designation (run or lot number), conditioning time, charge cutoff, and three static decay results for each polarity (plus or minus), for each specimen at each designated temperature. In addition, report test temperature/humidity, decay voltage and charging voltage, dissipation time, and test equipment details. Data shall be reported for each designated test condition.

4.5.28 Aircraft service test evaluation.

A full scale aircraft service test evaluation of the electrically conductive material shall be conducted for a minimum of 6 months prior to approval of a material for the Qualified Products List (QPL). The test shall be conducted under controlled conditions and samples of the product shall be removed from the aircraft fuel system for evaluation of critical properties. These critical properties shall include, but shall not be limited to, electrical resistivity, tensile strength, contamination, and compression load deflection. The installation shall include, as a minimum, one complete fuel tank on a given aircraft, and should preferably include the entire fuel tankage. Test samples (various foam kit pieces) of the product shall be identified in several fuel tank locations and tested for initial electrical resistivity (through the piece thickness) prior to fuel tank installation. The selected foam kit pieces will be removed at 3 to 6-month intervals depending on the service test duration. Additional foam kit pieces shall also be provided to replace those pieces that are removed for critical property testing. The material vendor shall conduct laboratory evaluations of the test specimens if requested. A detailed service test plan shall also be provided to the qualifying activity for approval prior to initiation of the service evaluation. If possible the testing should be conducted in a severe (cold) climate where electrostatic generation, discharges, ignitions were prevalent with blue polyether foam. Consideration should also be given to demonstrating over-the-wing (gravity) refueling if available on the aircraft. Guidelines for designing a foam kit for use in fuel tanks are provided in *SAE AIR-4170*.

4.6 Inspection of preparation for delivery.

Inspection of the preservation, packaging, packing, and marking for shipment shall be in accordance with the requirements of *section 5*.

MIL-PRF-87260B(USAF)

5. PACKAGING

5.1 Packaging.

For acquisition purposes, the packaging requirements shall be as specified in the contract or order (see 6.2). When packaging of materiel is to be performed by DoD or in-house contractor personnel, these personnel need to contact the responsible packaging activity to ascertain packaging requirements. Packaging requirements are maintained by the Inventory Control Point's packaging activities within the Military Service or Defense Agency, or within the military service's system commands. Packaging data retrieval is available from the managing Military Department's or Defense Agency's automated packaging files, CD-ROM products, or by contacting the responsible packaging activity.

6. NOTES

(This section contains information of a general or explanatory nature that may be helpful, but is not mandatory.)

6.1 Intended use.

The materials covered by this specification are intended for use in aircraft and ground vehicles fuel tanks using gasoline or kerosene type fuels at temperatures from $-55 \pm 5^{\circ}\text{F}$ to $160 \pm 5^{\circ}\text{F}$ for explosion and fire suppression. There are two classes of foam: Class 1 can be used from $+10^{\circ}$ to $+160^{\circ}\text{F}$; Class 2 can be used from -25° to $+160^{\circ}\text{F}$. Temperatures greater than 160°F and high humidity conditions will shorten the service life. These materials are intended to replace all the currently used foams qualified under *MIL-DTL-83054*. The advantages of these conductive materials include potentially longer service life and reduced incidents of electrostatic charge generation, discharge, and vapor ignitions that were experienced with the previous polyester/polyether materials.

6.2 Acquisition requirements.

Acquisition documents should specify the following:

- a. Title, number, and date of the specification.
- b. Class and grade of material required (see 1.2) and size (see 3.8).
- c. Location and conditions for testing and Government inspection and acceptance (see 4.4).
- d. Location of markings (see 3.9)
- e. Level of packing required

6.3 Qualification.

With respect to products requiring qualification, awards will be made only for products which are, at the time of award of contract, qualified for inclusion in Qualified Products List QPL - 8 7260 whether or not such products have actually been so listed by that date. The attention of the contractors is called to these requirements, and manufacturers are urged to arrange to have the products that they propose to offer to the Federal Government tested for qualification in order that they may be eligible to be awarded contracts or orders for the products

MIL-PRF-87260B(USAF)

covered by this specification. Information pertaining to qualification of products may be obtained from Mr. Alan Fletcher, AFRL/MLSA, 2179 12th Street, Wright-Patterson AFB OH 45433-7718, email alan.fletcher@wpafb.af.mil.

6.3.1 Qualifying activity.

The organization responsible for qualification approval for material covered by this specification:

Air Force Research Laboratories
ATTN: AFRL/MLSA
2179 12th Street, B652 R142
Wright-Patterson AFB, OH 45433-7718
e-mail: alan.fletcher@wpafb.af.mil

6.3.2 Qualification tests.

The qualification tests contained in this specification are considered adequate to insure that the explosion suppression material procured under this specification is satisfactory for the intended purpose provided the materials are similar to polyurethane reticulated foam. The qualifying activity may specify additional testing on materials submitted for qualification approval with chemical and mechanical properties not anticipated in the preparation of this specification. These tests may include, but will not necessarily be limited to, the following: ballistic response (gunfire), fuel flow and pump down, venting, icing, and fuel system compatibility.

6.4 Supersession data.

The inherently electrically conductive explosion suppression materials conforming to this specification supersedes all previous non-conductive explosion suppression foam baffle materials governed by *MIL-DTL-83054*, for new procurement applications.

6.5 Definitions.

For the purpose of this specification, the following definitions will apply:

6.5.1 Run of material.

Any continuous batch of product or a machine run produced over any continuous time period, the maximum run time being a 4-hour period. When production is interrupted for 2 or more hours, this will constitute a new run.

6.5.2 Test section width.

The standard width direction on a bun (40 or 44 inches).

6.5.3 Direction of rise.

The height direction relative to the standard bun (the 4, 8, or 12-inch direction).

6.5.4 Machine direction.

The lengthwise direction during production or the longest dimension relative to the standard bun size.

MIL-PRF-87260B(USAF)

6.5.5 Lot.

Fifteen machine runs of product.

6.6 Classes.

The class distinction noted in the specification is intended to separate the available products by their ability to dissipate static electricity at low fuel temperatures. Testing has indicated that below the low temperature fuel limits stated (Class 1 is +10°F and Class 2 is -25°F) the material may accumulate excessive static charge when impinged by high velocity fuel and as such may result in discharge and vapor ignition. This in itself is not a safety problem for single point refueling since the foam suppresses the resulting explosion internally; however, in the case where over the wing (gravity) type refueling is used, then a potential hazardous situation exists to refueling personnel and equipment since the fuel vapor can be ignited as it exits the filler opening. Requirements for interchangeability of products are as follows:

- a. Class 2 products are interchangeable (substitutable) for Class 1 materials; however, only if it is the same grade (example: Grade I for Grade I and Grade II for Grade II).
- b. Class 1 products are not interchangeable for Class 2 materials unless it is determined that the low temperature requirements can be waived or gravity refueling is restricted to emergency operations.

6.7 Installation guidelines for reticulated foam in aircraft fuel tanks and dry bays.

Guidelines for the proper design, fabrication, and installation of various reticulated foams can be found in *SAE AIR 4170*.

6.8 Service life.

The service life of the foam under this military specification begins on the date of foam manufacture. Service life limitation is a cumulative effect of its storage life exposure and its exposures in aircraft fuel tanks. Service life for foam baffles are also dependent upon how often the foam is removed and replaced in an aircraft. Although it is anticipated that foams under this specification will have a long, useful service life, determination of the practical limit of the service life by vigilant periodic inspection and testing, is the responsibility of each aircraft's maintenance and engineering organizations.

6.9 Fabricated parts.

Fabricated foam parts will be free of contaminates, specifically lint and debris.

6.10 Air deionizer.

An available source for the model 2U500 air deionizer specified in [4.5.14](#) is Amstat Industries, Inc., 3012 N. Lake Terrace, Glenview IL 60026-1335, phone 1-800-783-9999, web site <http://www.amstat.com/solutions/staticmaster.html>.

6.11 Marker identifier.

Suggested markers to be considered for testing are as follows:

- a. Diagraph GP-X, Diagraph Marking & Stenciling Products Group, 5307 Meadowland Parkway, Marion IL 62959, phone 1-800-521-3047, web site <http://www.diagrammsp.com/> or equivalent.

MIL-PRF-87260B(USAF)

- b. Commercial marking pen ink per A-A-208.

6.12 Product identification label card.

A suggested source for the fastener label is Avery Dennison, Fastener Division, 224 Industrial Road, Fitchburg MA 01420 (P/N 08909), phone 1-800-225-5913, web site <http://www.fastener.averydennison.com/>, or equivalent.

6.13 Subject term (key word) listing.

- Baffle material
- Ballistic foam
- Combat protection and survivability
- Electrostatically conductive foam
- Fire and explosion
- Fuel tank explosion protection
- Inerting
- Passive system
- Reticulated polyurethane foam
- Safety foam
- Slosh attenuation
- Survivability enhancement

6.14 Changes from previous issues.

The margins of this specification are marked with vertical lines to indicate where changes from the previous issue were made. This was done as a convenience only and the Government assumes no liability whatsoever for any inaccuracies in these notations. Bidders and contractors are cautioned to evaluate the requirements of this document based on the entire content irrespective of the marginal notations and relationship to the last previous issue.

MIL-PRF-87260B(USAF)

APPENDIX A**GUIDELINES FOR SMALL SCALE
FUEL IMPINGEMENT EVALUATION****A.1. SCOPE****A.1.1 Scope.**

This appendix contains guidelines for the small scale fuel impingement evaluation of reticulated foam products. The equipment, procedures, and data requirements specified in this appendix may be used for evaluating the electrostatic charge generation and ignition characteristics of reticulated foams (explosion suppression material) by impinging fuel at high velocities onto the material. Measurements of voltage buildup on an isolated conductor within the tank are used as criteria for evaluation of a product. Electrical activity/discharges/vapor ignitions are other criteria also used to evaluate the acceptability of the product. This appendix is not a mandatory part of the specification. The information contained herein is intended for guidance only.

A.2. APPLICABLE DOCUMENTS

DEPARTMENT OF DEFENSE SPECIFICATIONS

MIL-DTL-5624	Turbine Fuel, Aviation, Grades JP-4 AND JP-5
MIL-DTL-83054	Baffle and Inerting Material, Aircraft Fuel Tank
MIL-DTL-83133	Turbine Fuels, Aviation, Kerosene Types, NATO F-34(JP-8), NATO F-35, AND JP-8 + 100

AMERICAN SOCIETY FOR TESTING AND MATERIALS (ASTM)

ASTM D1655	Standard Specification for Aviation Turbine Fuels
------------	---

MIL-PRF-87260B(USAF)
APPENDIX A

A.3. TEST EQUIPMENT AND MATERIALS

The following minimum equipment is suggested for proper conduct of the testing:

- a. Three standard 55-gallon unlined steel drums, modified per the schematics.
- b. Fuel inlet: Standard pipe nozzle (0.82 inch inside diameter).
- c. Fuel lines and fittings to provide transfer of fuel at a maximum flow rate of 65 gpm (60 \pm 5 gpm) to the test tank (drum).
- d. Fuel transfer pump and valving to adjust flow rate to the specified rate of 60 \pm 5 gpm (37 \pm 5 fps at inlet).
- e. Thermometer and portable fuel conductivity meter for checking temperature and electrical conductivity of the test fuel before and after each test.
- f. Equipment for measuring the ambient temperature and relative humidity around the test area.
- g. Steel probe and teflon isolator for use in measuring voltage buildup within the tank.
- h. Electrostatic voltmeter and recorder (printer) and/or mag tape recorder for measuring and recording voltage buildup vs. time in the tank.
- i. Equipment for measuring the electrical resistivity of the test foam material prior to and after a test series on a product.
- j. Fuel conforming to *MIL-DTL-83133* or *ASTM D1655*, Grade JP-8 or Jet A-1 respectively is the preferred fuel; however, JP-5, (*MIL-DTL-5624*) or Jet A (*ASTM D1655*) are approved alternates. The primary test fuel should be "free" of conductivity additive (antistatic additive) and have a conductivity of less than 10 cu. In addition, testing with ASA additives at 50, 100, 200, 500, and 800 conductivity units should be accomplished by mixing the fuel/additive to the desired level.
- k. Test foam: Baseline orange polyester and blue polyether ESM per *MIL-DTL-83054*, *Types IV and V*, as well as the conductive candidate material in the proper porosity range(s). All drums shall be filled with foam for safety purposes.
- l. Photographic equipment for documenting the test setup and fuel vapor ignitions (flame out vent) including burned/singed foam.

A.4. TEST PARAMETERS AND DETAILS

The basic test setup is shown on [FIGURE A-1](#) utilizing three standard (uncoated) 55-gallon drums as the basic tank and fuel collector reservoir. The fuel should be transferred with an electrically driven fuel transfer pump from the reservoir outlet to the test tank inlet located at the top of the upper tank. The suggested flow rate is 60 \pm 5 gpm with corresponding inlet velocity of 37 \pm 5 fps at the pipe inlet nozzle (3/4 inch outside diameter). A ullage (gap) of one to two inches should be provided at the fuel inlet between the nozzle and foam in the tank. All tanks should be filled with foam (cylinders) cut to the drum inside diameter and 6, 8, or 12 inches high (or combinations thereof). The test drum foam should be dry at the beginning of each sequence of tests in order to provide a worse case condition. The fuel should be fresh and should not contain any trace of electrical conductivity type additives and be verified by test just prior to and after each evaluation. The preferred fuel type is JP-8 or Jet A-1; however, tests can be run

MIL-PRF-87260B(USAF)
APPENDIX A

using JP-5 or Jet A, as long as the properties of the fuel are documented and within specification limits (*MIL-DTL-83133*, *MIL-DTL-5624*, or *ASTM D1655* as applicable). A minimum of three test runs lasting five to twenty minutes shall be run on each foam type and configuration. The three runs can be run consecutively starting with dry foam at the beginning of each sequence (three tests). A 5-minute (minimum) down time shall be maintained between test runs for charge relaxation, appropriate data measurements, and test tank should be done at a low temperature and relative humidity to enhance static electricity generation. Specifically, temperatures below 50°F and 50 percent relative humidity are desirable; however, if these conditions cannot be attained, then it will not be a limiting factor for conduct of the testing. In order to evaluate the potential effects of temperature on the static electricity generation, it is recommended that several fuel temperatures be evaluated and an optimum set of conditions chosen for the final evaluation of the conductive foam. These include the following temperatures: 160, 60, 30, 10, 0, -15, -25, -30°F. The test fuel temperature is defined as the temperature at which the bulk fuel shall be at the start of a test run. If an ignition occurs during a test, the drum shall be opened and the damaged foam replaced and photographed.

A.5. TEST FOAM TYPES AND CONFIGURATIONS

Baseline foam tests should be run on orange and blue *MIL-DTL-83054*, Types II, IV or V polyester and foams to demonstrate the electrical charge generation/ignition characteristics of the material and measurement capability of the test rig instrumentation. Two test configurations shall be run: (a) fully packed, and (b) “over the wing” refueling inlet simulation. The test tank configurations are shown on [FIGURE A-2](#) and [FIGURE A-3](#) for reference. The “over the wing” void should be four inches in diameter and centered under the nozzle inlet. A one- to two-inch ullage shall also be provided just above the foam cylinders in the test tank. Fuel impingement tests should be run on each porosity type of foam being qualified, that is, coarse and fine pore or Types I and II respectively. Also, both configurations, fully packed and “over the wing” voiding, shall be evaluated on each type of foam being qualified unless it can be shown that the fully packed condition is a worse case.

A.6. TEST DATA AND DOCUMENTATION

Prior to any conduct of testing, a test procedure shall be written and submitted for approval. Photographs of the test setup and instrumentation shall be included if available. Typical test data requirements are outlined in the attached data sheets ([FIGURE A-4](#) and [FIGURE A-5](#)). Copies of data traces shall also be provided with adequate labeling and identification.

MIL-PRF-87260B(USAF)
APPENDIX A

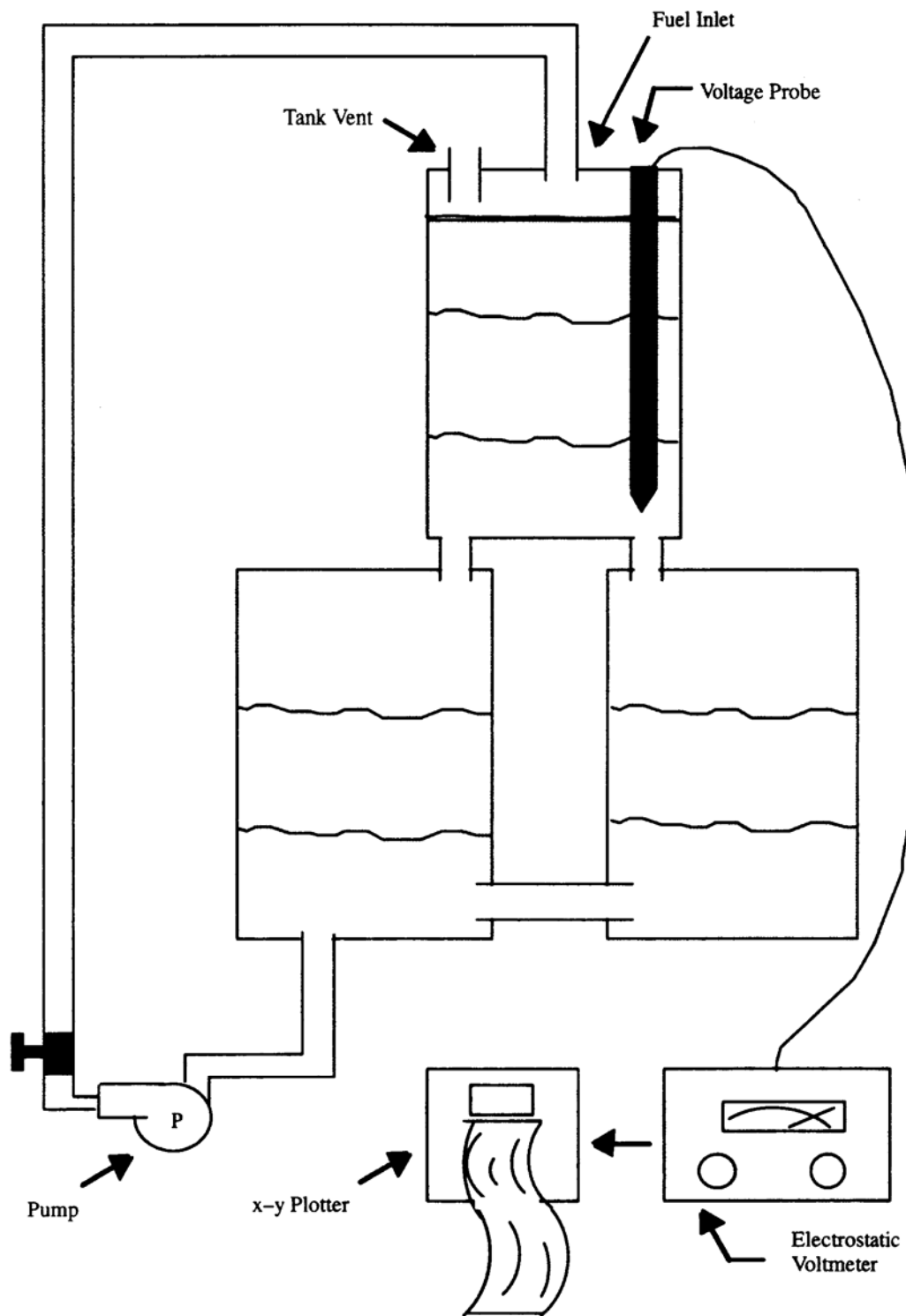


FIGURE A-1. Fuel impingement test set-up schematic.

MIL-PRF-87260B(USAF)
APPENDIX A

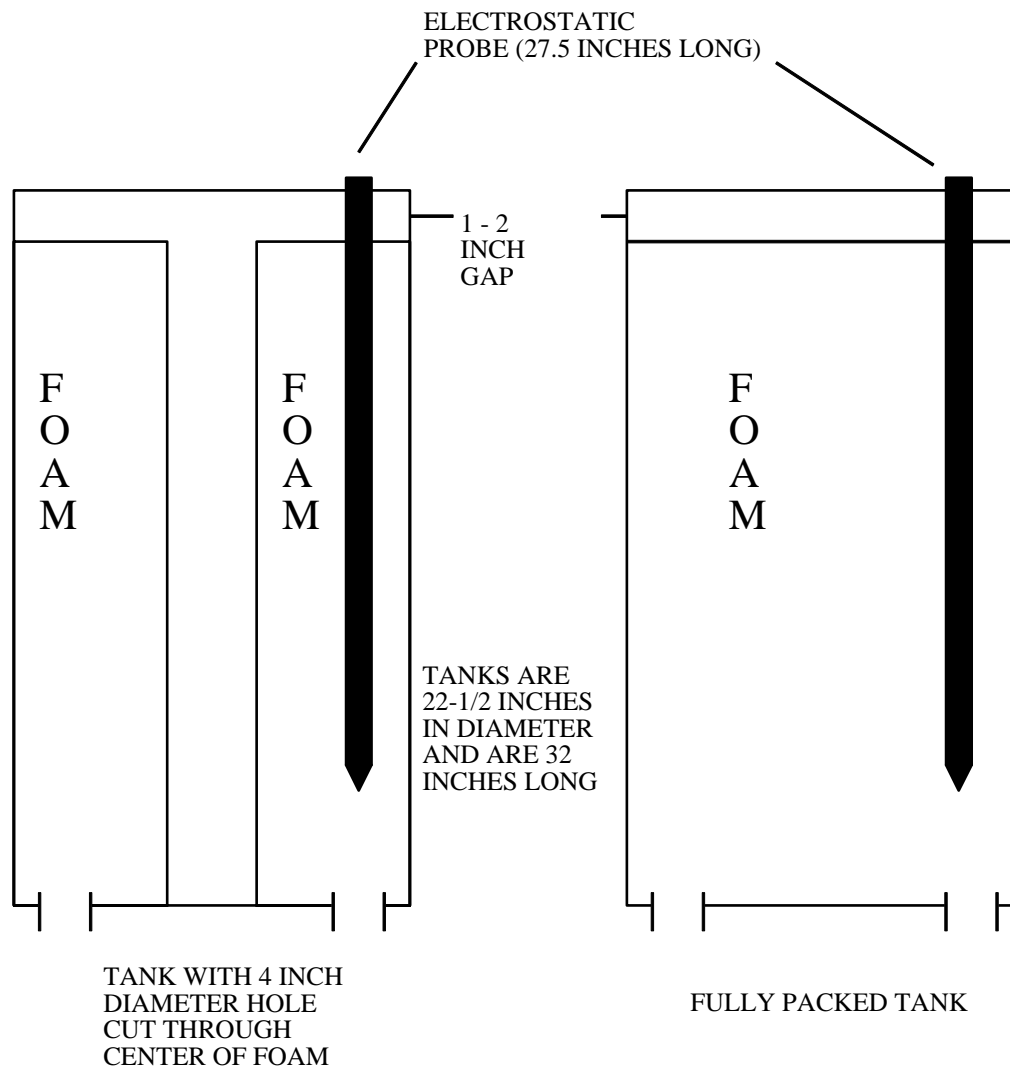


FIGURE A-2. Schematic of two different impingement scenarios.

MIL-PRF-87260B(USAF)
APPENDIX A

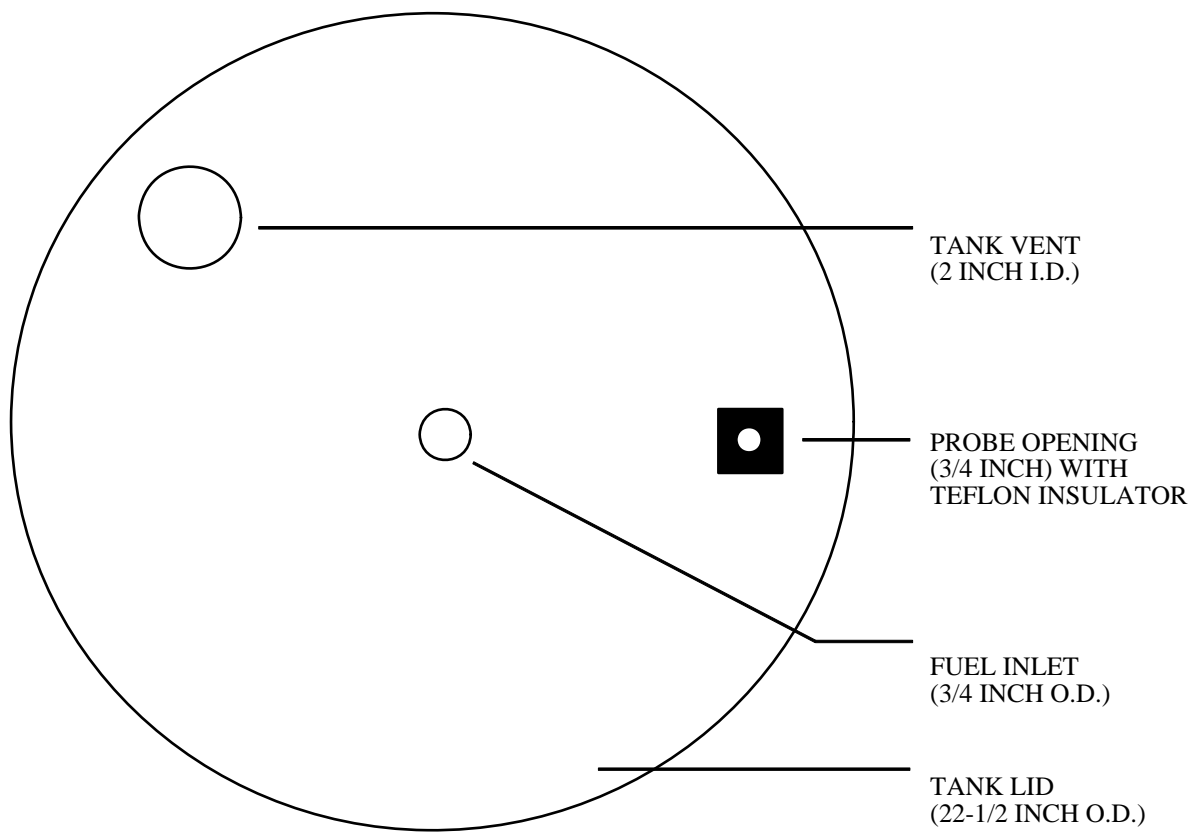


FIGURE A-3. Tank lid for the uppermost tank of the fuel impingement test rig.

MIL-PRF-87260B(USAF)
APPENDIX A

FUEL IMPINGEMENT SUMMARY TEST DATA SHEET			
Test # _____	Date _____	Time _____	
Foam Type: Dry or Wet			
Fuel Type: Conductivity			
Fuel Test Temp:	Start _____	End _____	
Ambient Conditions:			
Temperature	(% RH _____)		
Type Conductivity Additive used _____ (PPM _____)			
Test Run Data Summary: Test Run Time			
Maximum Voltage _____		(Time _____)	
Sparks? Electrical Activity			
Vapor Ignition (explain)			
Visual Inspection/Observations:			
Foam Kit Data	Part Number	Thickness	Resistivity
Top Piece			
Center Lower			
Center Upper			
Bottom			

FIGURE A-4. Fuel impingement summary test data sheet.

MIL-PRF-87260B(USAF)
APPENDIX A

FUEL IMPINGEMENT TEST DATA SHEET				
TEST # _____	DATE: _____	FOAM TYPE:		
		DRY: _____ WET: _____		
<u>TIME</u>	<u>VOLTAGE</u>	<u>FLOW</u>	<u>FUEL TEMP</u>	<u>NOTES</u>
				<u>FOAM LAYOUT</u>
				ULLAGE VOID:
				PN _____
				PN _____
				PN _____
				PN _____
				TEST CONDITIONS:
				<u>OAT</u> <u>%RH</u> <u>TIME</u>
				_____ START
				_____ END
				FUEL CONDUCTIVITY DATA:
				_____ <u>TEMP</u> <u>CU</u>
				START _____
				END _____
				ADDITIVE TYPE

				ULLAGE VOID

				TEST ENGINEER

FIGURE A-5. Fuel impingement test data sheet.

MIL-PRF-87260B(USAF)

CONCLUDING MATERIAL

Custodian:
Air Force-11

Preparing Activity:
Air Force - 11
(Project 9330-2007-001)

NOTE: The activities listed above were interested in this document as of the date of this document. Since organizations and responsibilities can change, you should verify the currency of the information above using the ASSIST Online database at <http://assist.daps.dla.mil/>.